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TECHNICAL REPORT RD-77-5

AN EXPERIMENTAL INVESTIGATION OF THE AERODYNAMIC
CHARACTERISTICS OF NOSE MOUNTED CANARD
CONFIGURATIONS AT SUPERSONIC MACH NUMBERS
(1.5 THROUGH 4.5)

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Experimental aerodynamic investigations were conducted at the AEDC, Tunnel A, supersonic wind tunnel to determine the aerodynamic characteristics of several nose mounted canard wing configurations. Canard position, geometry, angular orientation, and model nose shape were varied systematically to determine their effects on model aerodynamics. The model was tested at angles of attack of -5 to 13 degrees and Mach numbers of 1.5 to 4.5. Aerodynamic loads were measured separately on four canards and tail panels, and the total model. The (Continued)		

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20. ABSTRACT

vapor screen method was used to trace the paths of the vortices shed from the canards. Roll control effectiveness of canards, both with a planar tail and a ring tail was investigated. This test was designated as V41A-C1A.

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TABLE OF CONTENTS

	PAGE
INDEX OF TABLES	2
INDEX OF MODEL FIGURES	2
INDEX OF DATA FIGURES	3
INTRODUCTION	8
TEST FACILITY	9
MODELS	10
INSTRUMENTATION AND PRECISION	12
TEST CONDITIONS AND PROCEDURES	15
DATA REDUCTION	17
TEST UNCERTAINTIES	18
REFERENCES	20
NOMENCLATURE	21
TABLES	24
MODEL FIGURES	31
PLOTTED DATA	40

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INDEX OF TABLES

TABLE		PAGE
1	Data Set/Run Number Collation Summary	24

INDEX OF MODEL FIGURES

FIGURE		PAGE
1	Model Photographs	
	a. Model Set-up in Tunnel "A"	31
	b. Ring Tail	32
2	Model Details	
	a. Canard, Tail, and Vapor Screen Stations	33
	b. Details of Various Noses	34
	c. Planar Tail Panel and Ring Tail Panel	35
	d. Canards C1, C3, and C6	36
3	Axis Systems and Positive Sign Convention; Canards and Tail Fins	37
4	Axis Systems and Positive Sign Convention; Unrolled Body Axis System	38
5	Sign Convention for Canard Deflection Angles	39

INDEX OF DATA FIGURES

TITLE	CONDITIONS VARYING	PLOT SCHEDULE	PAGE
BODY ALONE LONGITUDINAL CHARACTERISTICS TAIL OFF	CONFIG, MACH	(A)	1-2
BODY-TAIL LONGITUDINAL CHARACTERISTICS PHITAL = 0	PHICND, MACH	(A)	3-4
BODY-TAIL LONGITUDINAL CHARACTERISTICS TAIL OFF PHICND = 0	DCND2, 4	(A)	5-5
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4	(A)	6-6
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(A)	7-14
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(A)	15-16
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 45, PHICND = 0	DCND2, 4, MACH	(A)	17-21
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 45, PHICND = 45	DCND1, 2, 3, 4, MACH	(A)	22-25
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4	(A)	26-27
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4	(A)	28-28
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	CONFIG, DCND2, 4, MACH	(A)	29-31
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4 MACH	(A)	32-33

INDEX OF DATA FIGURES (Continued)

TITLE	CONDITIONS VARYING	PLOT SCHEDULE	PAGE
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(A)	34-35
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(A)	36-37
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	DCND2, 4, MACH	(A)	38-39
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(A)	40-41
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(A)	42-45
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(A)	46-47
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, TAIL OFF, PHICND = 0	DCND2, 4	(B)	48-50
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, TAIL OFF, PHICND = 45	DCND1, 2, 3, 4	(B)	51-53
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(B)	54-77
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	78-83
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 45, PHICND = 0	DCND2, 4, MACH	(B)	84-98

INDEX OF DATA FIGURES (Continued)

TITLE	CONDITIONS VARYING	PLOT SCHEDULE	PAGE
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 45, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	99-110
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4	(B)	111-116
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4	(B)	117-119
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	CONFIG., DCND2, 4, MACH	(B)	120-128
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	129-134
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(B)	135-140
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	141-146
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	DCND2, 4, MACH	(B)	147-152
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	153-158
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(B)	159-170
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(B)	171-176

INDEX OF DATA FIGURES (Continued)

TITLE	CONDITIONS VARYING	PLOT SCHEDULE	PAGE
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(C)	177-200
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	201-206
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 45, PHICND = 0	DCND2, 4, MACH	(C)	207-221
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 45, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	222-233
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4	(C)	234-240
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4	(D)	241-242
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	CONFIG, DCND2, 4, MACH	(C)	243-251
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	252-257
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(C)	258-263
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	264-269
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	DCND2, 4, MACH	(C)	270-275

INDEX OF DATA FIGURES (Concluded)

TITLE	CONDITIONS VARYING	PLOT SCHEDULE	PAGE
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	276-281
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 0	DCND1, 2, 3, 4, MACH	(C)	282-293
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL, PHITAL = 0, PHICND = 45	DCND1, 2, 3, 4, MACH	(C)	294-299
CANARD AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS, PHITAL = 0, DCND1 = 0	ALPHA, DCND2, 3, 4	(B)	300-302
TAIL AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS, PHITAL = 0, DCND1 = 0	ALPHA, DCND2, 3, 4	(C)	303-305

PLOT SCHEDULE:

- (A) C_{NNR} , C_{mNR}
- (B) C_{NC1} , 2, 3, 4', C_{HM1} , 2, 3, 4', C_{BM1} , 2, 3, 4
- (C) C_{NT1} , 2, 3, 4', C_{HM1} , 2, 3, 4', C_{BM1} , 2, 3, 4
- (D) C_{HM1} , 2, 3, 4', C_{BM1} , 2, 3, 4

INTRODUCTION

In recent years, the Army has developed a strong interest in adding a terminal guidance capability to existing and new missile systems for situations where high accuracy is desired. Nose mounted canard controls are an attractive candidate for terminally guided missiles because of their simplicity and large control force to control package weight ratio.

An accurate method is needed to predict the aerodynamic characteristics of a missile with canards for preliminary design purposes and to help shorten expensive wind tunnel tests in advanced design. Comparisons between results from the prediction method currently used for canard configurations and available wind tunnel data has shown this method is generally inaccurate, particularly for nose mounted canards. This inaccuracy is possibly due to some of the simplifying assumptions, used in developing the theory, being violated by the geometry and conditions of interest.

The purpose of this study is to develop an accurate method of predicting the aerodynamics of canard configurations by developing a systematic set of wind tunnel data for general canard configurations and using this data to improve the currently used aerodynamic prediction technique. A supersonic wind tunnel test was conducted at AEDC on a sting supported model with nose mounted, remotely deflected canards. The primary points of interest in this test were the effect of model roll position, canard size, and canard position on control forces developed. The model was unique in that forces were measured separately on the four

canards, the four tail panels, and the total model, thereby yielding the component forces needed to improve the mathematical model used to predict canard configuration aerodynamics. The vapor screen method was used to trace the paths of the vortices being shed from the canards (Ref. 1). Secondary objectives of this test were to investigate the feasibility of using canards for roll control with both a planar tail and ring tail and to investigate the effect of nose mounted antennas on canard forces. This test extends the Mach number range of a similar transonic test (Ref. 2) and a supersonic test (Ref. 3).

TEST FACILITY

Tunnel A is a continuous, closed-circuit, variable density wind tunnel with an automatically driven flexible-plate nozzle and a 40- by 40-in. test section. The tunnel can be operated at Mach numbers from 1.5 to 6.0 at maximum stagnation pressures from 29 to 200 psia, respectively, and stagnation temperatures up to 750°R ($M_{\infty} = 6.0$). Minimum operating pressures range from about one-tenth to one-twentieth of the maximum at each Mach number. The tunnel is equipped with a model injection system which allows removal of the model from the test section while the tunnel remains in operation. A description of the tunnel and airflow calibration information may be found in Ref. 4.

MODELS

Components of the test configurations are the following:

B	Basic body
C1	Small canards in aft position
C3	Small canards in forward position
C6	Large "Navy" canards in aft position
N1	Sharp nose
N2	Blunt nose
N4	Prong nose with four antennas
N5	Prong nose with two antennas
N6	Prong nose with no antennas
T1	Planar tails
TR	Ring tail

Figure 1 shows the Army canard control model as installed in Tunnel A. The baseline model Fig. 2a was a 52-in.-long, 5-in.-diam. tangent ogive cylinder with canards and tail fins. The configuration variables included three noses (Fig. 2b), two tails (Fig. 2c) and three sets of canards (Fig. 2d).

A study of the aerodynamic contributions of the individual model components included testing the sharp (N1) and blunt (N2) noses with the body alone (B). The planar tails (T1, Fig. 2c) and canards (C1, Fig. 2d) were added to BN1 to obtain the baseline model (BN1C1T1). In this configuration, the canard and tail panels are aligned as shown in Fig. 2a.

A similar configuration was tested with the T1 tail section rotated 45 deg. with respect to the rest of the model. The planar tails were maintained at zero attitude (not deflected) and were balance-supported for all configurations throughout the test program.

The "ring tail" (TR, Fig. 2c) was tested in an attempt to obtain better roll control without affecting the stability of the vehicle. As shown in Fig. 2c, the vertical support struts were not attached to the body. With this arrangement, the two horizontal tail balances provided measurements of the total force (but not the moments) on the ring tail. A model fabrication error resulted in an incidence angle of approximately 1 deg. in the pitch attitude of the ring tail relative to the model centerline. Since the model was otherwise symmetrical, this was treated as an incremental model load and accounted for by a coefficient shift included in the data correction procedure.

Two of the three sets of canards (C1 and C3, Fig. 2d) had the same planform area and shape but different attachment angles because of the different longitudinal mounting positions on the model (Fig. 2a). The third set of canards (C6, Fig. 2d) was much larger than C1 and C3 and was tested only with the "prong" nose (N4, N5, N6, Fig. 2b) and the planar tails. All canards were balance supported with provisions for independent deflections (δc) from -5 to 15 deg., produced by remotely operated electric motors internal to the model. The sign convention for the planform surfaces is shown in Fig. 3.

Model base pressures were measured at two positions in the vertical plane equally spaced above and below the model centerline.

INSTRUMENTATION AND PRECISION

The Tunnel A stilling chamber pressure is measured with a 15-, 60-, 150-, or 300-psid transducer referenced to a near vacuum. Based on periodic comparisons with secondary standards, the uncertainty (a bandwidth which includes 95 percent of the residuals) of these transducers is estimated to be within ± 0.2 percent of reading or ± 0.015 psia, whichever is greater. Stilling chamber temperature is measured with a copper-constantan thermocouple with an uncertainty of $\pm 3^\circ\text{F}$, based on repeat calibrations.

Base pressures were measured with 15-psid transducers referenced to a near vacuum. These transducers have an estimated uncertainty of ± 0.15 percent of reading or 0.003 psi, whichever is greater.

Main Balance

Model forces and moments were measured with a six-component, moment-type strain-gage balance supplied and calibrated by VKF. Prior to the test, static loads in each plane and combined static loads were applied to the balance to simulate the range of loads and center-of-pressure locations anticipated during the test. The following uncertainties represent the bands of 95 percent of the measured residuals, based on differences between the applied loads and the corresponding values calculated from the balance calibration equations included in the final data reduction. The range of check loads applied and the measurement uncertainties follow.

<u>Component</u>	<u>Balance Design Loads</u>	<u>Calibration Load Range</u>	<u>Range of Check Loads</u>	<u>Measurement Uncertainty</u>
Normal force, lb*	± 700	± 700	± 175	±1.50
Pitching moment, in.-lb	±3,645	±3,645	±2,075	±5.00
Side force, lb*	± 700	± 350	± 100	±0.75
Yawing moment, in.-lb	±3,645	±1,822	±1,040	±2.00
Rolling moment, in.-lb	± 320	± 104	± 20	±0.75
Axial force, lb	0 to 125	0 to 95	0 to 50	±1.00

*About balance forward moment bridge

The transfer distance from the balance forward moment bridge to the model moment reference location was 4.15 in. along the longitudinal axis and was measured with an estimated precision of ±0.01 in.

Model pitching moments were referenced to a point at model station 26.00 on the centerline of the model (Fig. 2a).

Canard And Tail Balances

Canard and tail panel loads were measured with eight 3-component, moment type, strain-gage balances supplied by USAMC and calibrated by VKF. The same calibration procedure was used with these eight fin balances as was previously described for the main balance. The range of check loads applied and the measurement uncertainties follow.

<u>Component</u>	<u>Balance Design Loads</u>	<u>Calibration Load Range</u>	<u>Range of Check Loads</u>	<u>Measurement Uncertainty</u>
Canard Balances				
Normal force, lb	± 40	± 40	±40	±0.40
Root bending moment, in.-lb	± 35	± 35	±35	±0.35
Hinge moment, in.-lb	± 25	± 25	±10	±0.25
Tail Fin Balances*				
Normal force, lb	± 60	± 60	±30	±0.60
Root bending moment, in.-lb	±130	±130	±50	±1.30
Hinge moment, in.-lb	±100	±100	±35	±1.66

*For the ring-tail (TR) configuration, only the normal-force uncertainty quotations apply to the tail fin balances (Nos. 2 and 4). Reliable moment measurements could not be obtained on the fin balances with this configuration.

The transfer distances from the balance forward moment bridges to the planform reference locations were measured with an estimated precision of ±0.01 in. and are listed below.

	<u>Forward Moment Bridge to Hinge Line, in.</u>	<u>Balance Centerline to Root Chord, in.</u>
Canard Balance		
No. 1	0.00	-0.25
No. 2	↓	↓
No. 3		
No. 4		
Tail Fin Balance		
No. 1	1.12	-0.58
No. 2	1.09	-0.57
No. 3	1.10	-0.57
No. 4	1.12	-0.57

TEST CONDITIONS AND PROCEDURES

Test Conditions

The test was conducted at Mach numbers 1.5, 2.5, 3.0, and 4.5. The nominal test conditions at each Mach number are given below.

M_∞	p_0 , psia	T_0 , °R	q_∞ , psia	p_∞ , psia	$Re_L \times 10^{-6}$
1.5*	4	560	1.9	1.2	5.4
2.5	15	570	3.8	0.9	11.9
3.0	22	570	3.8	0.6	13.6
4.5	78	600	3.8	0.3	20.8

*Only canard panel loads were measured.

A test summary showing all configurations tested and the variables for each is presented in Table 1.

Test Procedures

After the model was injected into the tunnel, one of three procedures was used to obtain the test data. The first consisted of rotating the model to the desired roll angle and pitching to the maximum negative angle of attack. Each data channel was then recorded at a rate of 60 samples per second as the model was continuously driven to the maximum positive angle of attack at a rate of approximately 0.25 deg. per second. Each calculated data point for this type of operation was based on the average of 16 consecutive samples. These "averaged points" were subsequently interpolated to provide data at precisely the desired angles of attack.

Data obtained in this mode of operation are generally referred to as "continuous sweep" data.

In the second procedure, generally referred to as the "pitch-pause" mode, the model paused at each desired angle in the angle-of-attack sequence and fifty samples of each data channel were recorded. Each data point for this type of operation is the average of these 50 samples.

The third data-taking procedure consisted of pitching the model to a specified angle of attack and taking 50 samples of each data channel at 15-deg. increments of roll angle in the range from 0 to 90 deg.

Periodically during the test, the continuous sweep data were checked by repeating a test sequence in the "pitch-pause" mode. These data agreed within the repeatability listed.

The measured total axial force was corrected to zero base drag using the measured base pressures and the base cross-sectional area. The initial value of base drag (obtained at $\alpha \approx -4$ deg.) was used for all angles of attack in the sweep data because the continuous sweep technique does not allow time for base pressure stabilization. Shortly after the test began it was found that during injection into the free stream, model loads were higher than anticipated. The operating procedure therefore had to be changed so that p_0 was reduced for each injection and retraction, and for many cases at the lowest M_∞ the injection and retraction had to be made at a higher Mach number. This reduced the amount of data which could be obtained during the test entry. Flow-field shadowgraphs and vapor screen photographs were taken at selected test conditions.

DATA REDUCTION

Main balance force and moment data were reduced in both the body and missile (non-rolling body) axis systems. Dimensional quantities employed in the reduction of the main balance data are listed below:

SREF = 19.635 sq. in.	XMRP = 26.0 in.
LREF = 5.0 in.	YMRP = 0.0 in.
BREF = 5.0 in.	ZMRP = 0.0 in.

Canard and tail root bending moments are referenced to their respective root chord lines while the hinge moments of the canard and tail are referenced to their hinge lines.

The axis system and sign convention used for the model main balance data is shown in Figure 4; for the canard and tail balances refer to Figures 3 and 5.

A review of the tunnel monitor force and moment data obtained on the various configurations showed evidence of offsets in level due to irregularities in model fabrication and to local nonuniformities in the test section flow. Repeat test runs were made to confirm the cause of the offsets and to establish data reduction procedures to account for the resulting shifts in the levels of the force and moment coefficients. The data consistently indicated the presence of incremental forces acting on localized regions of the model. This produced incremental shifts in the force and moment coefficients, but there was no evidence of a change in the variation of the coefficients with angle of attack. Since the models

with undeflected control surfaces were designed as symmetrical configurations, the extraneous local model loads were accounted for by adding increments to the force and moment coefficients so that they were zero at zero model attitude. The increments established for the symmetrical configurations were used for the corresponding nonsymmetrical configurations that resulted from deflection of the control surfaces. Axial-force coefficients were not shifted.

TEST UNCERTAINTIES

An evaluation of the influence of random measurement errors is presented in this section to provide a measure of the uncertainty of the final test results. Although evaluation of the systematic measurement error (bias) is not included, it should be noted that the instrumentation precision values used in this evaluation represent a total uncertainty including both systematic and random (two standard deviation bandwidth) error contributions.

Test Condition Uncertainty

Uncertainties in the basic tunnel parameters p_0 and T_0 and the two standard deviation uncertainty in Mach number determined from test section flow calibrations were used to estimate uncertainties in the other free-stream properties, using the Taylor series method of error propagation.

Test Condition Uncertainty (\pm), percent						
M_∞	M_∞	P_0	T_0	P_∞	q_∞	Re_ℓ
1.5	1.3	0.2	0.5	2.9	0.3	0.8
2.5	0.8	0.2	0.5	3.0	1.5	1.3
3.0	0.7	0.2	0.5	3.0	1.7	1.3
4.5	0.4	0.2	0.5	2.5	1.6	1.3

Test Data Uncertainty

The main balance force and moment data are presented in the missile-axis system with pitching moments referenced to the midpoint of the model length (MS 26.00).

The balance and base pressure uncertainties were combined with uncertainties in the tunnel parameters, using the Taylor series method of error propagation, to estimate the uncertainties of the aerodynamic coefficients presented below.

Measured Coefficient Uncertainty (\pm), percent				
M_∞	C_N	C_m	C_ℓ	C_A
2.5	1.70	1.72	*	4.31
3.0	1.91	2.24	*	2.72
4.5	1.80	2.96	*	4.63

*For these cases, the corresponding repeatability values listed below are to be used since the maximum rolling-moment coefficients are approximately equal to balance precision.

The uncertainties of the aerodynamic coefficients were also computed using only the balance and base pressure uncertainties and the nominal test conditions. This calculation is based on the assumption that the free-stream flow nonuniformities introduce a bias type of uncertainty that is constant for all test runs at a particular Mach number. The values therefore represent the data repeatability expected and are especially useful for detailed discrimination purposes in parametric model studies.

Measured Coefficient Repeatability (\pm)				
M_∞	C_N	C_m	C_ℓ	C_A
2.5	0.020	0.021	0.002	0.017
3.0	0.020	0.022	0.002	0.015
4.5	0.020	0.022	0.002	0.014

The uncertainty in model angle of attack and roll, as determined from tunnel sector calibrations and possible errors in model deflection calculations, is estimated to be ± 0.1 deg.

REFERENCES

1. Spahr, J. R. and Dickey, R. R. "Wind Tunnel Investigations of the Vortex Wakes and Downwash Field Behind Triangular Wings and Wing Body Combinations at Supersonic Speeds." NACA RM A53010, 1953.
2. Burt, J. R., Jr. "An Experimental Investigation of the Aerodynamic Characteristics of Several Nose Mounted Canard Configurations at Transonic Mach Numbers." Technical Report RD-75-2, 30 August 1974.

3. Burt, James R., Jr. "An Experimental Investigation of the Aerodynamic Characteristics of Several Nose Mounted Canard Configurations at Supersonic Mach Numbers." Technical Report RD-75-17, U. S. Army Missile Command, 30 January 1975.
4. Test Facilities Handbook (Tenth Edition). "von Kármán Gas Dynamics Facility, Vol. 3." Arnold Engineering Development Center, May 1974.
5. Burt, J. R., Jr. "Pre-test Information for a Supersonic Investigation of a Model With Nose Mounted Canard Wings (AEDC)." Internal Technical Note RD-76-1.

NOMENCLATURE

<u>SYMBOL</u>	<u>MNEMONIC</u>	<u>DEFINITION</u>
α	ALPHA	total pitch plane angle of attack, deg.
ϕ_C	PHICND	total model roll angle, deg.
ϕ_T	PHITAL	tail roll angle relative to canard, deg.
$\delta_{C1,2,3,4}$	DCND1,2,3,4	canard deflection relative to missile centerline, deg.
$C_{N_{NR}}$	CNP	total model normal force coefficient, N/QSREF, non-rolling axis system
$C_{m_{NR}}$	CLMP	total model pitching moment coefficient, M/QSREFLREF, non-rolling axis system
$C_{N_{T1,2,3,4}}$	CNT1,2,3,4	tail panel normal force coefficient, NT/QSREF
$C_{N_{C1,2,3,4}}$	CNC1,2,3,4	canard panel normal force coefficient, NC/QSREF
$C_{HM_{C1,2,3,4}}$	CLMHC1,2,3,4	canard hinge moment coefficient, MHC/QSREFLREF

NOMENCLATURE (Continued)

<u>SYMBOL</u>	<u>MNEMONIC</u>	<u>DEFINITION</u>
$CBM_{C1,2,3,4}$	CLMRC1,2,3,4	canard root bending moment coefficient, MBC/QSREFLREF
C_{ℓ}	CBL	model rolling moment coefficient, LB/QSREFBREF, non-rolling axis system or missile axis system
C_A	CA	model forebody axial force coefficient, FA/QSREF, non-rolling axis system and missile axis system
$CHM_{T1,2,3,4}$	CLMHT1,2,3,4	tail hinge moment coefficient, MHT/QSREFLREF
$CBM_{T1,2,3,4}$	CLMRT1,2,3,4	tail root bending moment coefficient, MBT/QSREFLREF
$C_{Y_{NR}}$	CYP	total model side force coefficient, Y/SREF, non-rolling axis system
$C_{n_{NR}}$	CYNP	total model yawing moment coefficient, NB/QSREFBREF, non-rolling axis system
q_{∞}	Q	dynamic pressure, psia
S_{ref}	SREF	non-dimensionalizing reference area, sq. in.
l_{ref}	LREF	non-dimensionalizing reference length, in.
b_{ref}	BREF	non-dimensionalizing reference length, in.
N	N	total normal force on model, lbs.
M_m	M	total pitching moment on model, in.-lbs.
N_T	NT	normal force on tail panel, lbs.
N_C	NC	normal force on canard, lbs.
HM_C	MHC	canard hinge moment, in.-lbs.
BM_C	MBC	canard root bending moment, in.-lbs.
M_{ℓ}	LB	total model rolling moment about body/missile axes, in.-lbs.

NOMENCLATURE (Concluded)

<u>SYMBOL</u>	<u>MNEMONIC</u>	<u>DEFINITION</u>
A	FA	model forebody axial force, lbs.
HM_T	MHT	tail panel hinge moment, in.-lbs.
BM_T	MBT	tail panel root bending moment, in.-lbs.
Y	Y	model side force, lbs.
M_n	NB	model yawing moment about body/missile axes, in.-lbs.
M, M_∞	MACH MRP	free stream Mach number moment reference point
P_0		stilling chamber total pressure, psia.
T_0		stilling chamber total temperature, deg. Rankine
p_∞		free stream static pressure, psia
Re_x	RN/L	free stream Reynolds number, 1/ft.
MS		missile station, in.
A_b		missile base area
V		free stream velocity, ft./sec.
CPX_C		canard panel center of pressure relative to hinge line, in.
CPX_T		tail panel center of pressure relative to hinge line, in.
CPY_C		canard panel center of pressure relative to root chord, in.
CPY_T		tail panel center of pressure relative to root chord, in.
$b_c(x)$		distance from canard hinge line to MRP, in.
$b_T(x)$		distance from tail hinge line to MRP, in.

TABLE 1.

TEST : AEDC V41A-C1A		DATA SET / RUN NUMBER COLLATION SUMMARY												DATE : 08-25-76				
DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS				TEST RUN NUMBERS
			α	ϕ_C	$\delta C1$	$\delta C2$	$\delta C3$	$\delta C4$	ϕ_T						1.5	2.5	3.0	
RXH001	BN1	A		0	OFF	OFF	OFF	OFF	OFF	OFF			2					
002	BN1TI	T		0	OFF	OFF	OFF	OFF	0				T				174	71
003	BN1TI			45	OFF	OFF	OFF	OFF	0				T				232	73
004	BN1C1			0	0	-3	0	-3	OFF				1				233	74
005	T			T	T	0	T	0	T				T				166	
006				T	T	3	T	3	T				T				161	
007				T	T	6	T	6	T				T				165	
008				T	T	9	T	9	T				T				164	
009				T	T	15	T	15	T				T				163	
010				45	-3	-3	-3	-3					T				162	
011				T	0	0	0	0					T				172	
012				T	3	3	3	3					T				167	
013				T	6	6	6	6					T				171	
014				T	9	9	9	9					T				170	
015				T	15	15	15	15					T				169	
016	BN1C1TI			0	0	-3	0	-3	0				2				168	
017	T			T	0	T	0	T	0				3				139	83
018	T			T	1	T	1	T	1				1				157	76
CNP	CLMP	CNT1	CNT2	CNT3	CNT4	CNC1	CNC2	CNC3	CNC4				1				82	
CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMRC1	CLMRC2	CLMRC3	CLMRC4	CBL	CA				CNC4	MACH	ALPHA	10		
CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMRT1	CLMRT2	CLMRT3	CLMRT4	CYP	CYNP				CA	MACH	ALPHA	10		
CNP	CLMP	CA	CYP	CYNP	CBL								CYNP	MACH	ALPHA	10		
Coefficients																		
R=A: $\alpha(A): -5^\circ \rightarrow 13^\circ$																		
R=B: $\phi_C(A): 0^\circ \rightarrow 90^\circ$																		
R=C: $\alpha(B): -5^\circ \rightarrow 5^\circ$																		
R=D: IDVAR(1) IDVAR(2) NDV																		

R=A:
R=B:
R=C:
R=D:

TABLE 1. (Continued)

TEST: AEDC V41A-C1A

DATA SET/RUN NUMBER COLLATION SUMMARY

DATE: 08-25-76

DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES	NO. OF RUNS	MACH NUMBERS	TEST RUN NUMBERS
RXH019	BN1C1T1	A	ϕ_C 0 0 3 0 3 0 3 0 $\delta C1$ T T 6 T 6 T 9 $\delta C2$ T T 9 T 15 T 15 $\delta C3$ T T 15 T 3 3 -3 $\delta C4$ T T 6 T 15 T 15 ϕ_T 0 0 3 0 3 0 3 0	2	1.5 2.5 3.0 4.5	81 80 79 78 84 85 86 87 88 96 89 94 93 92 91 105 98 104
020		T		T		
021				T		
022				T		
023				T		
024				T		
025				T		
026				T		
027				T		
028				T		
029				T		
030				T		
031				T		
032				T		
033	T			T		
034	BN1C1T1			T		
035				T		
036	T			T		
R=A: CNP	CLMP	CNT1	CNT2	CNT3	CNT4	CNC1
R=B: CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMRC1	CLMRC2	CLMRC3
R=C: CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMRT1	CLMRT2	CLMRT3
R=D: CNP	CLMP	CA	CYP	CYNP	CBL	CBL

TABLE 1. (Continued)

TEST: AEDC V41A-C1A		DATA SET/RUN NUMBER COLLATION SUMMARY														DATE: 08-25-76	
DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS			
			ϕ	δC	$\delta C1$	$\delta C2$	$\delta C3$	$\delta C4$	ϕT					1.5	2.5	3.0	4.5
RXH037	BN1C1T1	A	0	0	0	3	0	3	45				2			123	103
038		T	0	0	0	6	0	6	T							122	102
039			0	0	0	9	0	9								121	101
040		V	0	0	0	15	0	15								120	100
041		B	45	5	-5	-5	5	5					1				114
042		A	T	-3	-3	-3	-3	-3					2			132	113
043		T		0	0	0	0	0					2			126	106
044				1	1	1	1	1					1				112
045				3	3	3	3	3					2			131	111
046				6	6	6	6	6								130	110
047				9	9	9	9	9								129	109
048		V	15	15	15	15	15	15								128	108
049	BN1C3T1	A	0	0	0	0	0	0	0				1			234	
050		T	T	T	1	T	1	1	T							239	
051				3	3	3	3	3								238	
052				6	6	6	6	6								237	
053				9	9	9	9	9								236	
054		V	15	15	15	15	15	15								235	
R=A: CNP	CLMP	CNT1	CNT2	CNT3	CNT4	CNT4	CNT1	CNC1	CNC2	CNC3	CNC4	CNC4		MACH		ALPHA	10
R=B: CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMHC1	CLMHC2	CLMHC2	CLMHC3	CLMRC4	CBL	CA	CA	CA		MACH		ALPHA	10
R=C: CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMHT1	CLMHT2	CLMHT2	CLMHT3	CLMRT4	CYP	CYNP	CYNP	CYNP		MACH		ALPHA	10
R=D: CNP	CLMP	CA	CYP	CYNP	CBL	CBL								MACH		ALPHA	6
														IDVAR(1)	IDVAR(2)	NDV	

26

Coefficients

 $\alpha(A): -5^\circ \rightarrow 13^\circ$ $\phi_c(A): 0^\circ \rightarrow 90^\circ$ $\alpha(B): -5^\circ \rightarrow 5^\circ$

TABLE 1. (Continued)

TEST: AEDC V41A-C1A

DATA SET/RUN NUMBER COLLATION SUMMARY

DATE: 08-25-76

DATA SET IDENTIFIER		CONFIGURATION	SCHD.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS					TEST RUN NUMBERS
				α	ϕ_c	$\delta C1$	$\delta C2$	$\delta C3$	$\delta C4$	ϕT					1.5	2.5	3.0	4.5		
RXH055	BN1C3T1			B	0	5	-5	-5	5	0				1			242			
056				B	0	2	-2	-2	2	T				T			241			
057				B	0	-3	3	3	-3								240			
058				A	45	0	0	0	0								243			
059	▼			A	45	15	15	15	15	▼				▼			244			
060	BN2			A	0	OFF	OFF	OFF	OFF	OFF				2			175	69		
061	BN2C1T1			A	0	0	-3	0	-3	0				1			156			
062	T			T	T	T	0	T	0	T				T			150			
063							3		3								155			
064							6		6								154			
065							9		9								153			
066	▼			▼	▼	▼	15	▼	15					▼			152			
067	BN4C6T1			A	0	0	0	0	0					2	211		199			
068	T			T	T	T	3	T	3					T	210		202			
069					▼	▼	9	▼	9					▼	209		207			
070					▼	▼	15	▼	15					▼	208		200			
071					-100	0	9	0	9					1			201			
▼072	▼			▼	45	0	0	0	0	▼				2	212		203			
R=A:	CNP	CLMP	CNT1	CNT2	CNT3	CNT4	CNC1	CNC2	CNC3	CNC4	MACH	ALPHA	10							
R=B:	CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMRC1	CLMRC2	CLMRC3	CLMRC4	CBL	CA	MACH	ALPHA	10							
R=C:	CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMRT1	CLMRT2	CLMRT3	CLMRT4	CYP	CYNP	MACH	ALPHA	10							
R=D:	CNP	CLMP	CA	CYP	CYNP	CBL					MACH	ALPHA	6							
											IDVAR(1)	IDVAR(2)	NDV							

27

NOTE: At $M_\infty = 1.5$ the nose shock reflection crossed the model near the base, and tests were made only to obtain canard panel loading.

TABLE 1. (Continued)

TEST: AEDC V41A-C1A		DATA SET/RUN NUMBER COLLATION SUMMARY										DATE: 08-25-76											
DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS					TEST RUN NUMBERS				
			ϕ_c	$\delta C1$	$\delta C2$	$\delta C3$	$\delta C4$	ϕ_T						1.5	2.5	3.0	4.5						
RXH073	BN4C6T1	A	45	3	3	3	3	0					2	215		206							
074		T	45	9	9	9	9	T						214		205							
075		V	45	15	15	15	15							213		204							
076	BN5C6T1	A	0	0	0	0	0							216		191							
077		T	T	T	3	T	3							220		194							
078				9	9	9	9							219		193							
079				V	15	V	15						V	218		192							
080			V	15	15	15	15						1	217									
081			45	0	0	0	0						2	221		195							
082			T	3	3	3	3							224		198							
083				9	9	9	9							223		197							
084		V	V	15	15	15	15							222		196							
085	BN6C6T1	A	0	0	0	0	0							225		183							
086		T	T	T	3	T	3							228		182							
087				9	9	9	9							227		181							
088			V	V	15	V	15							226		180							
089			45	0	0	0	0							229		184							
090		V	45	6	6	6	6	V					V	231		187							
R=A: CNP	CLMP	CNT1	CHT2	CNT3	CNT4	CNT4	CNC1	CNC2	CNC3	CNC4				MACH		ALPHA	10						
R=B: CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMHC4	CLMRC1	CLMRC2	CLMRC3	CLMRC4	CBL	CA				MACH		ALPHA	10						
R=C: CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMHT4	CLMRT1	CLMRT2	CLMRT3	CLMRT4	CYP	CYNP				MACH		ALPHA	10						
R=D: CNP	CLMP	CA	CYP	CYP	CYNP	CBL								MACH		ALPHA	6						
Coefficients																			IDVAR(1)	IDVAR(2)	NDV		
$\alpha(A): -5^\circ \rightarrow 13^\circ$																							
$\phi_c(A): 0^\circ \rightarrow 90^\circ$																							
$\alpha(B): -5^\circ \rightarrow 5^\circ$																							

NOTE: At $M_\infty = 1.5$ the nose shock reflection crossed the model near the base, and tests were made only to obtain canard panel loading.

TABLE 1. (Continued)

TEST : AEDC V41A-C1A		DATA SET/RUN NUMBER COLLATION SUMMARY												DATE : 08-25-76				
DATA SET IDENTIFIER	CONFIGURATION	SCHD.	PARAMETERS/VALUES										NO. OF RUNS	MACH NUMBERS				
			φC	δC1	δC2	δC3	δC4	φT						1.5	2.5	3.0	4.5	
RXH091	BN6C6T1	A	45	15	15	15	15	0					2	230		186		
092	BN6C6T1	A	180	0	0	0	0	T					1			188		
093	BN1C1TR	A	0	0	0	0	0					2		16		2		
094					3		3								20	6		
095					9		9								19	5		
096					15		15								18	4		
097		B		5	-5	-5	5								46	8		
098		B		2	-2	-2	2								21	7		
099		B		0	0	0	0					1		42				
100		A		0	0	0	0							41				
101		A												58				
102		A										2		60		67		
103		A												61		62		
104		A	45	0	0	0	0							56		14		
105				3	3	3	3							54		12		
106				9	9	9	9							53		11		
107				15	15	15	15							52		10		
CNP	CLMP	CNT1	CNT2	CNT3	CNT4	CNC1	CNC2	CNC3	CNC4					MACH	ALPHA	10		
CLMHC1	CLMHC2	CLMHC3	CLMHC4	CLMRC1	CLMRC2	CLMRC3	CLMRC4	CBL	CA					MACH	ALPHA	10		
CLMHT1	CLMHT2	CLMHT3	CLMHT4	CLMRT1	CLMRT2	CLMRT3	CLMRT4	CYP	CYNP					MACH	ALPHA	10		
CNP	CLMP	CA	CYP	CYNP	CBL									MACH	ALPHA	6		
IDVAR(1) IDVAR(2) IDV																		

R=A:
R=B:
R=C:
R=D:

Coefficients

29

Coefficients

 $\alpha(A): -5^\circ \rightarrow 13^\circ$ $\phi_c(A): 0^\circ \rightarrow 90^\circ$ $\alpha(B): -5^\circ \rightarrow 5^\circ$

NOTE: At $M_\infty = 1.5$ the nose shock reflection crossed the model near the base, and tests were made only to obtain canard panel loading.

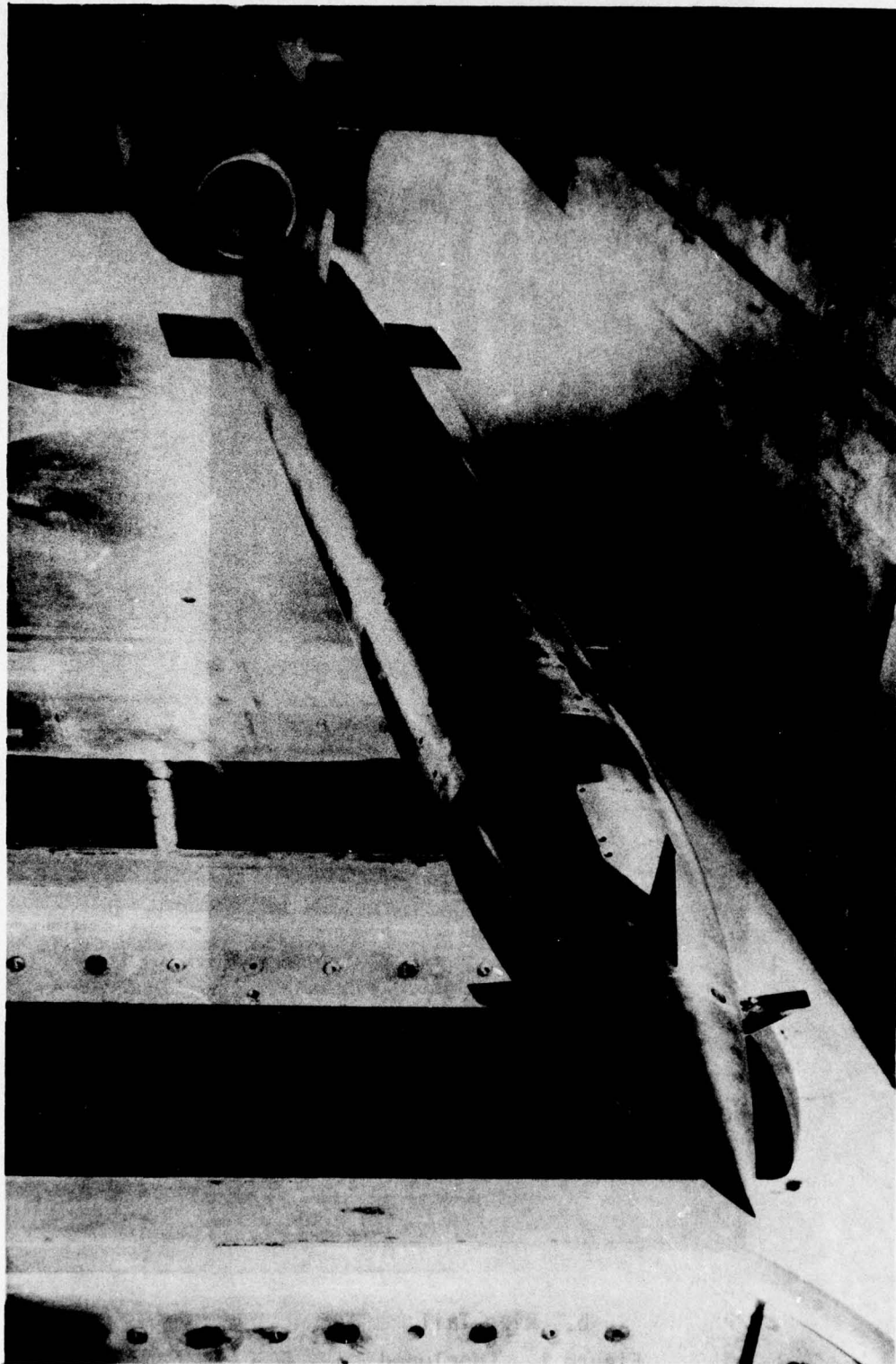
TABLE 1. (Concluded)

[illegible]
$$\alpha(A): -5^\circ \rightarrow 13^\circ$$

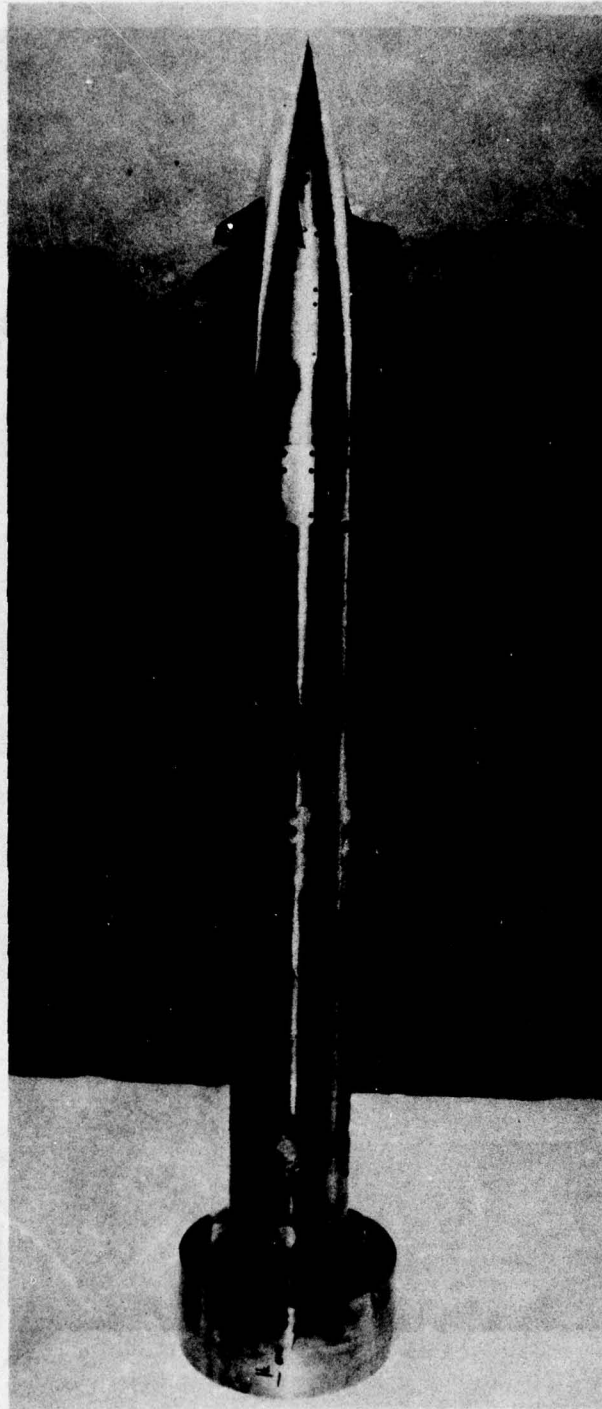
$$\alpha(B): -5^\circ \rightarrow 5^\circ$$
 $\phi_C(A): 0^\circ \rightarrow 90^\circ$

Coefficients

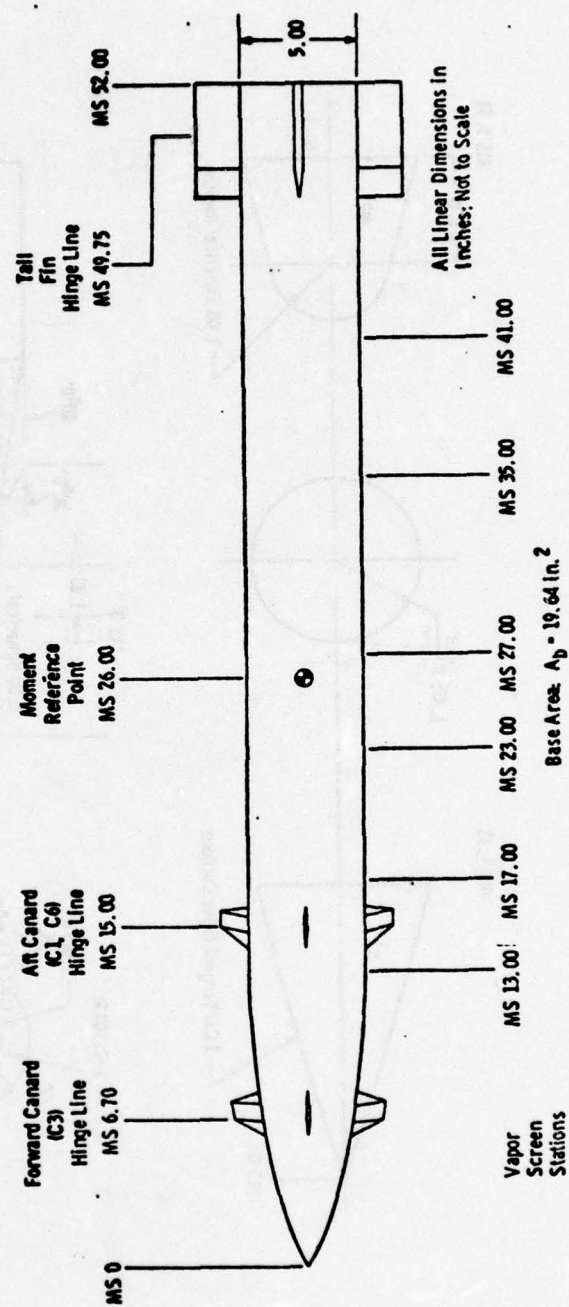
IDVAR(1)	IDVAR(2)	IDV
----------	----------	-----



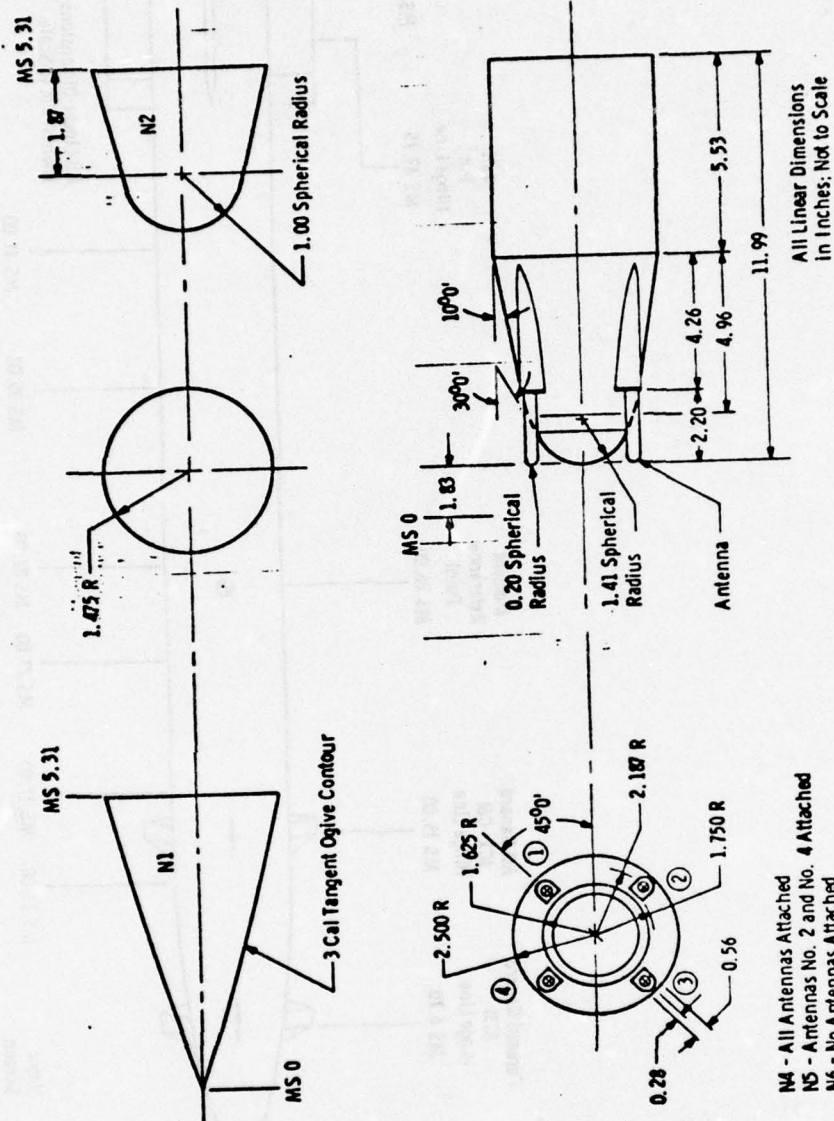
a. Model Set-up in Tunnel "A"
Figure 1. Model Photographs



b. Ring Tail
Figure 1. Concluded

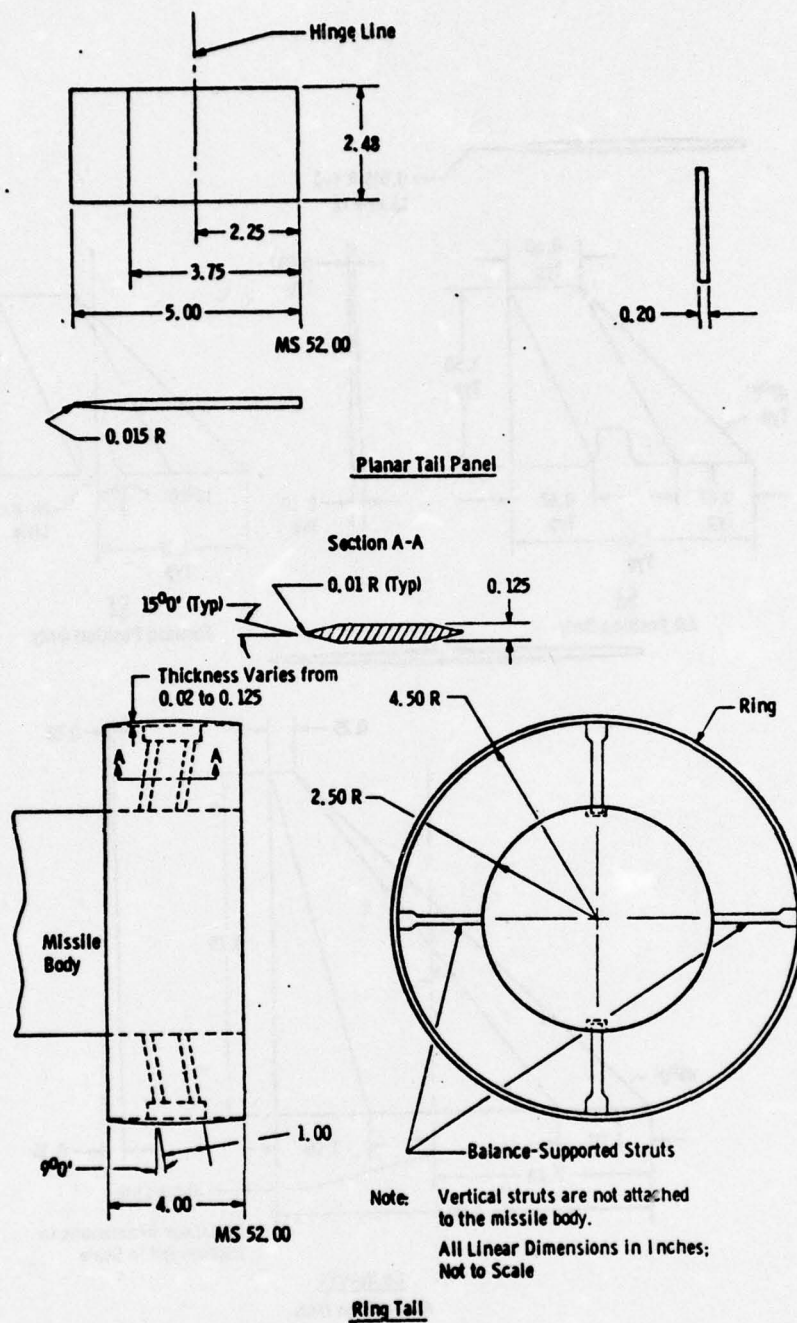


a. Canard, Tail, and Vapor Screen Stations
Figure 2. Model Details

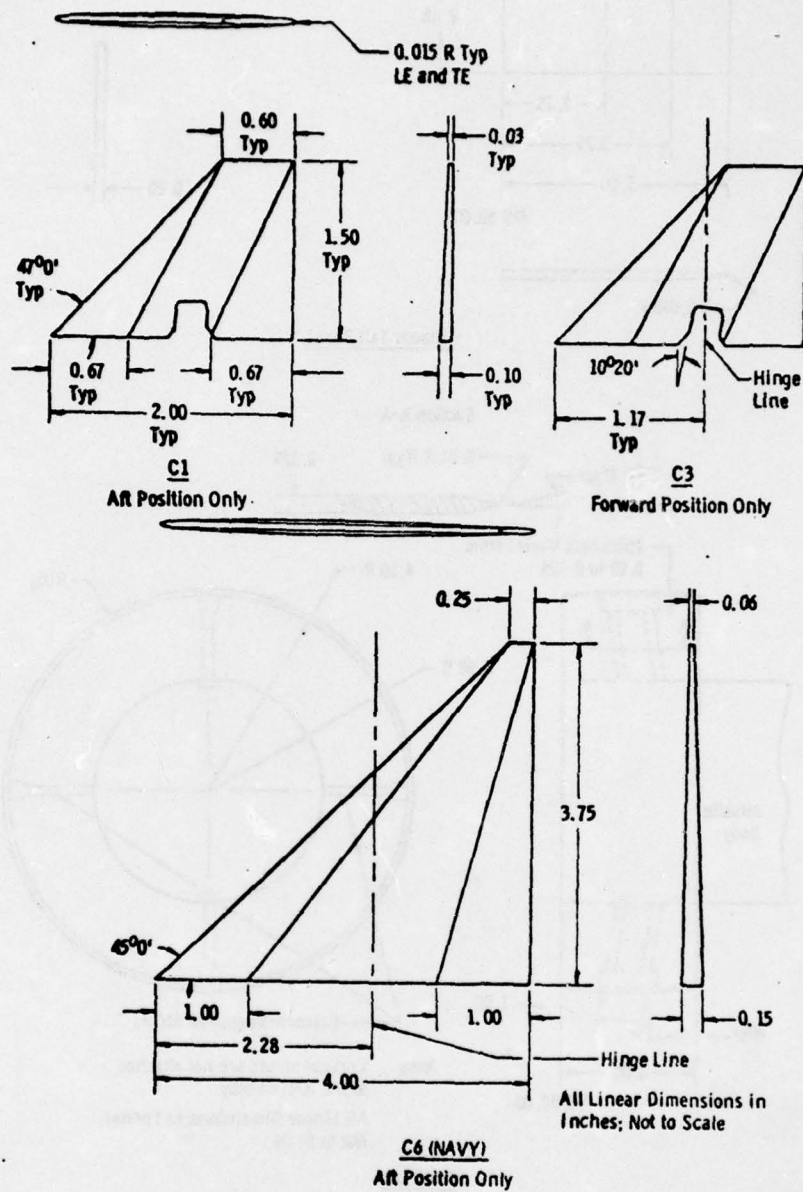


N1 - All Antennas Attached
 N2 - Antennas No. 2 and No. 4 Attached
 N3 - No Antennas Attached

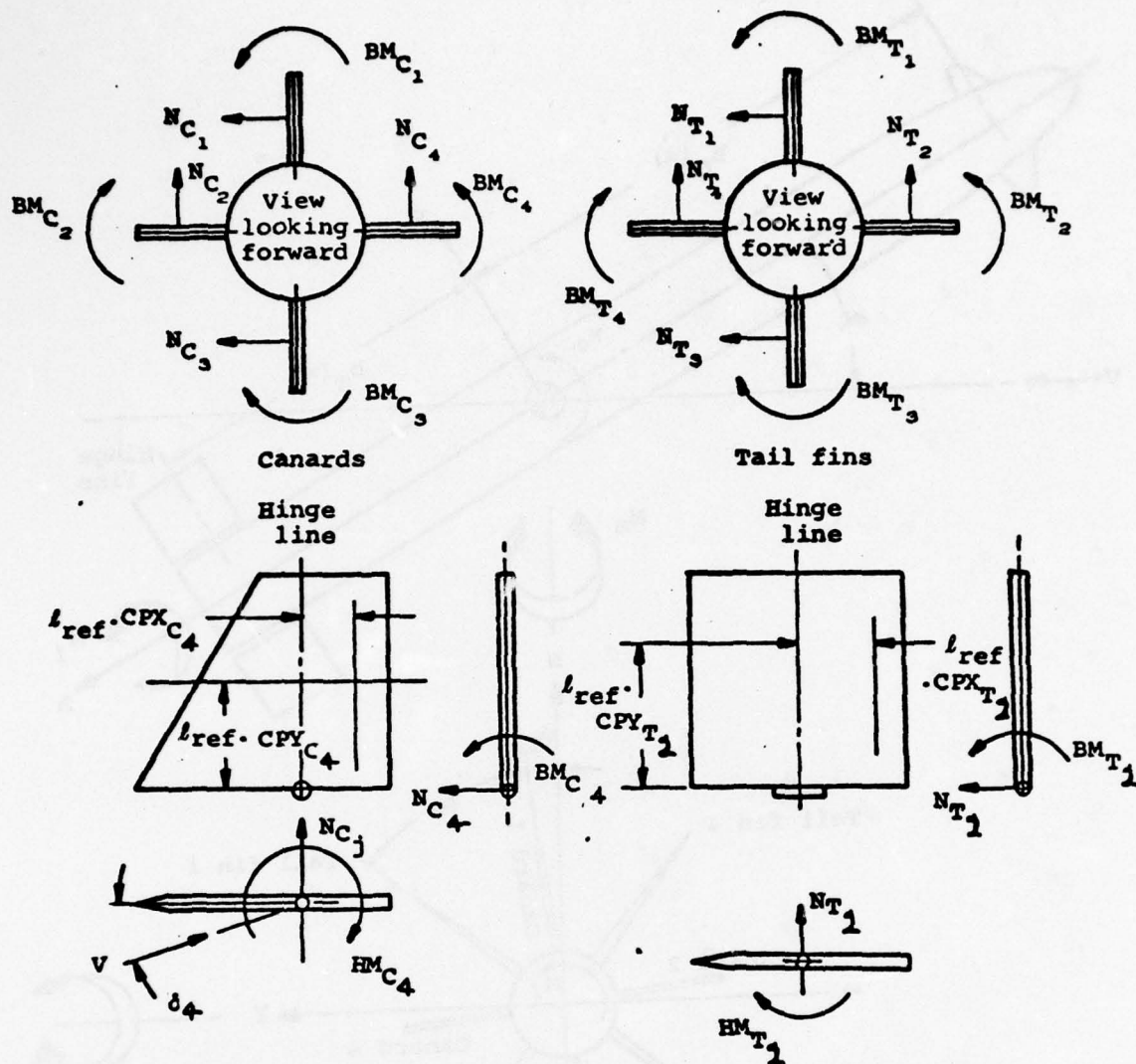
b. Details of Various Noses
 Figure 2. Continued



c. Planar Tail Panel and Ring Tail Panel
Figure 2. Continued



d. Canards C1, C3, and C6
Figure 2. Concluded



Note 1: Normal forces are measured perpendicular to the panel platform.

Note 2: Tail panels are numbered clockwise and canard panels counter-clockwise.

Figure 3. Axis Systems and Positive Sign Convention; Canards and Tail Fins

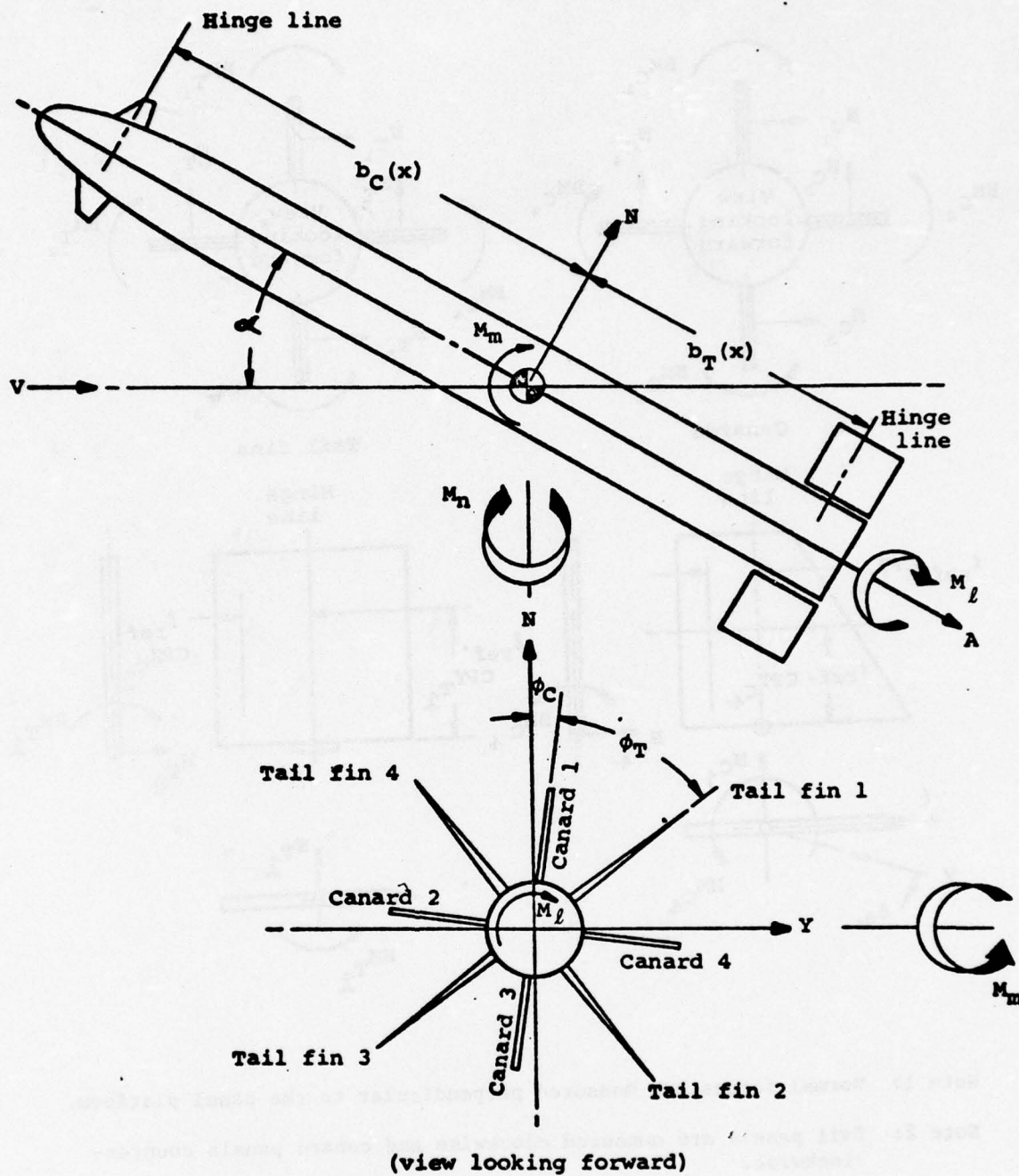
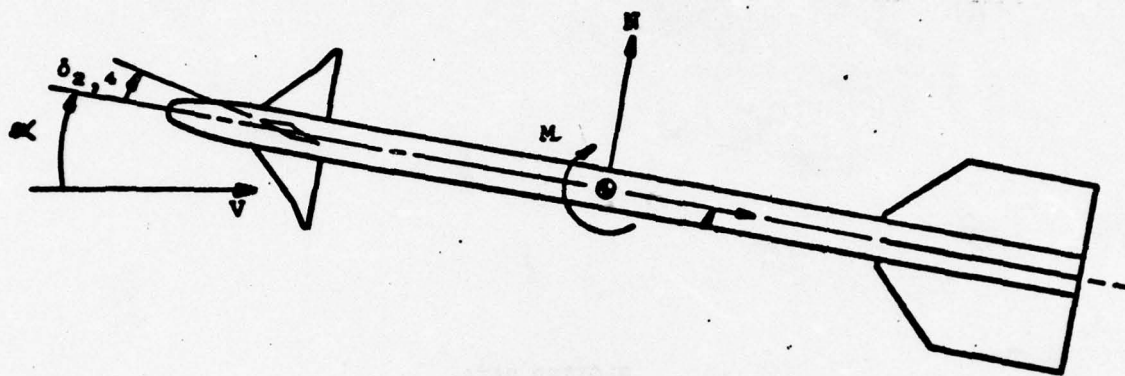
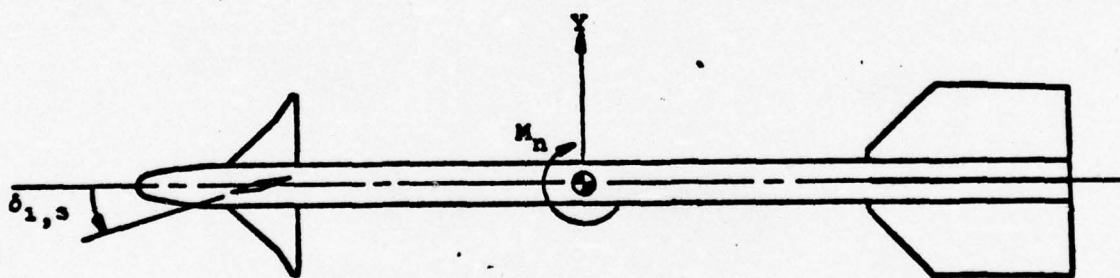


Figure 4. Axis Systems and Positive Sign Convention;
Unrolled Body Axis System

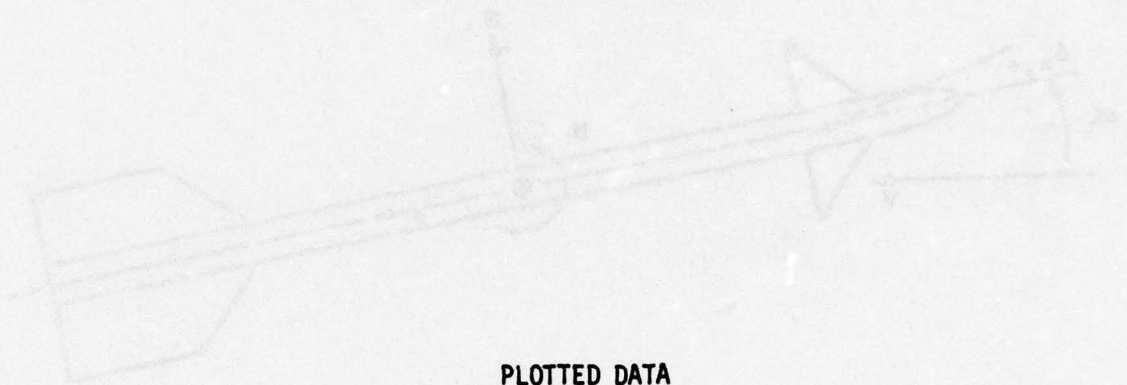


(a) Pitch plane.

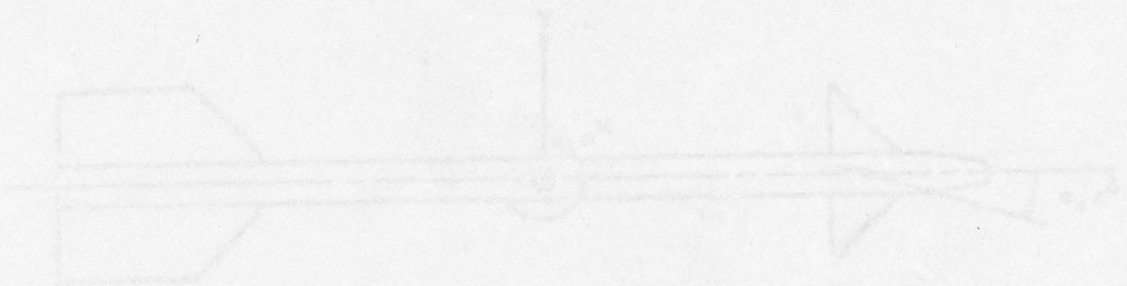


(b) Yaw plane.

Figure 5. Sign Convention for Canard Deflection Angles



PLOTTED DATA



Tabulations of the plotted data and corresponding source data are available from Data Management Services Operations.

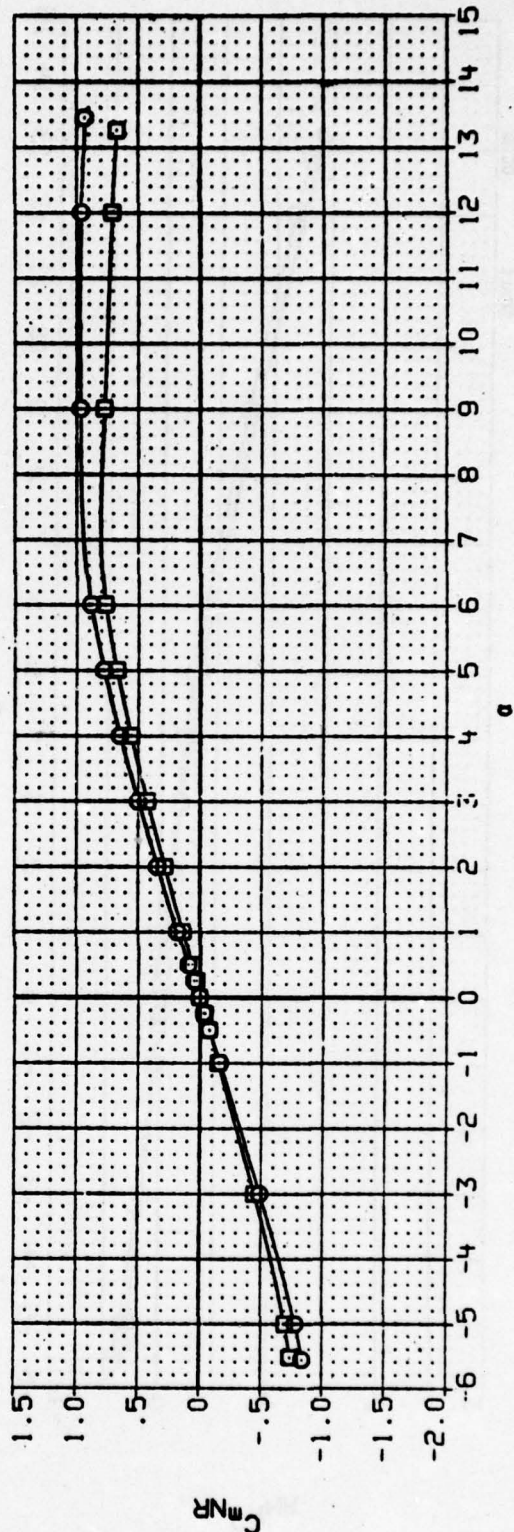
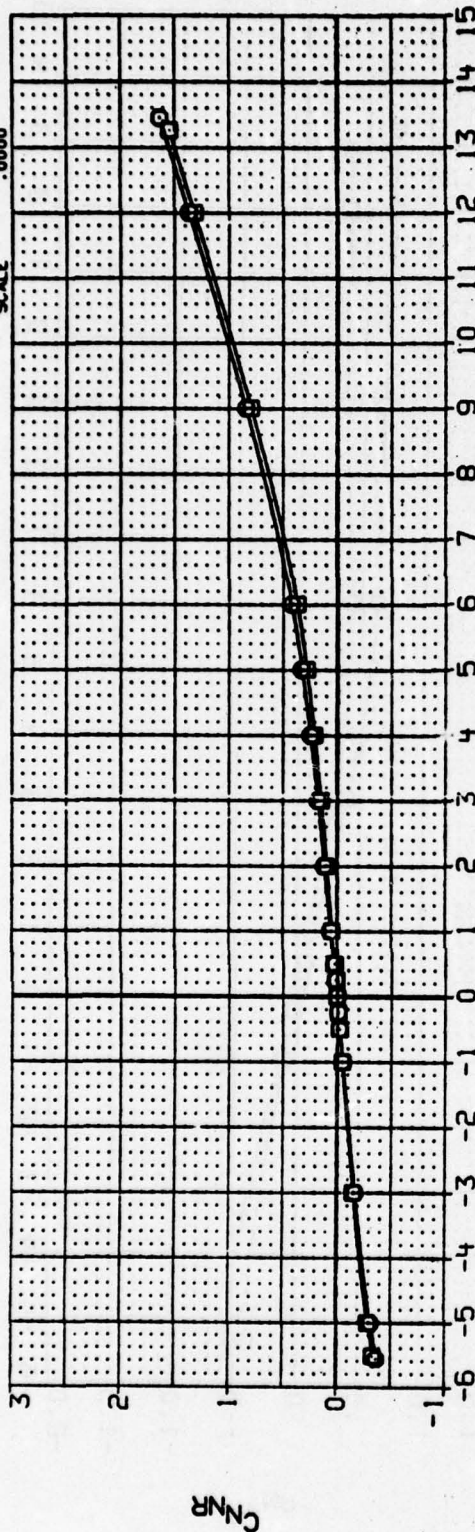
DATA SET SYMBOL
(DXH001)
(DXH060)

CONFIGURATION DESCRIPTION

AEDC W1A-C1A, CANARD CONTROL, BN1
AEDC W1A-C1A, CANARD CONTROL, BN2

PHICND
.000
.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



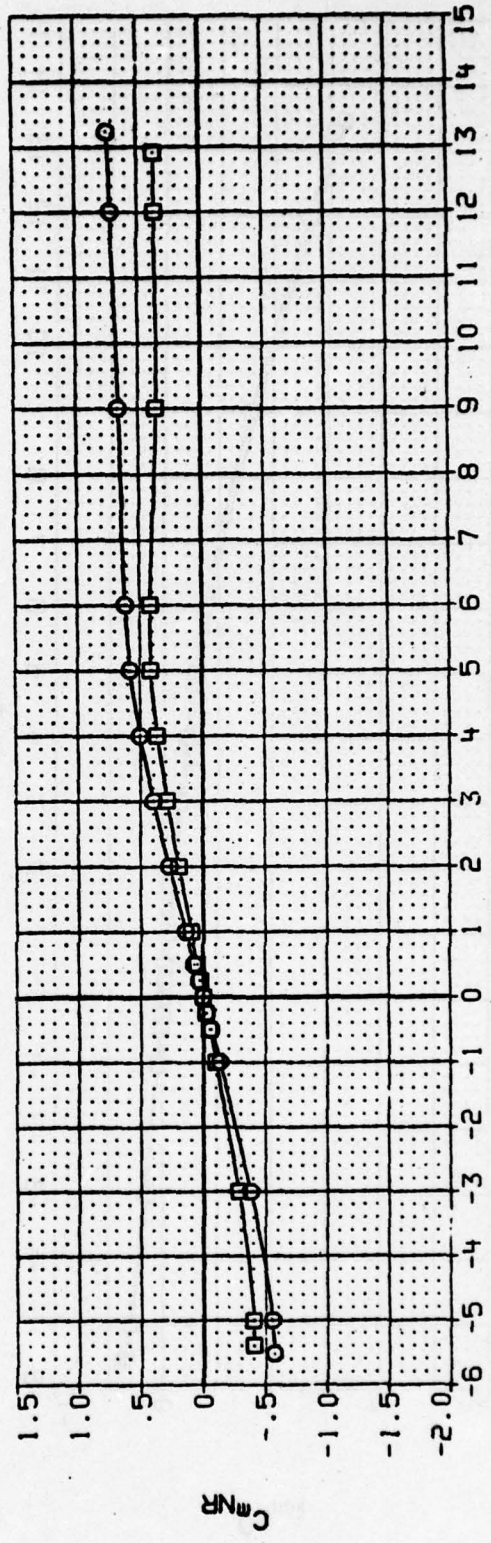
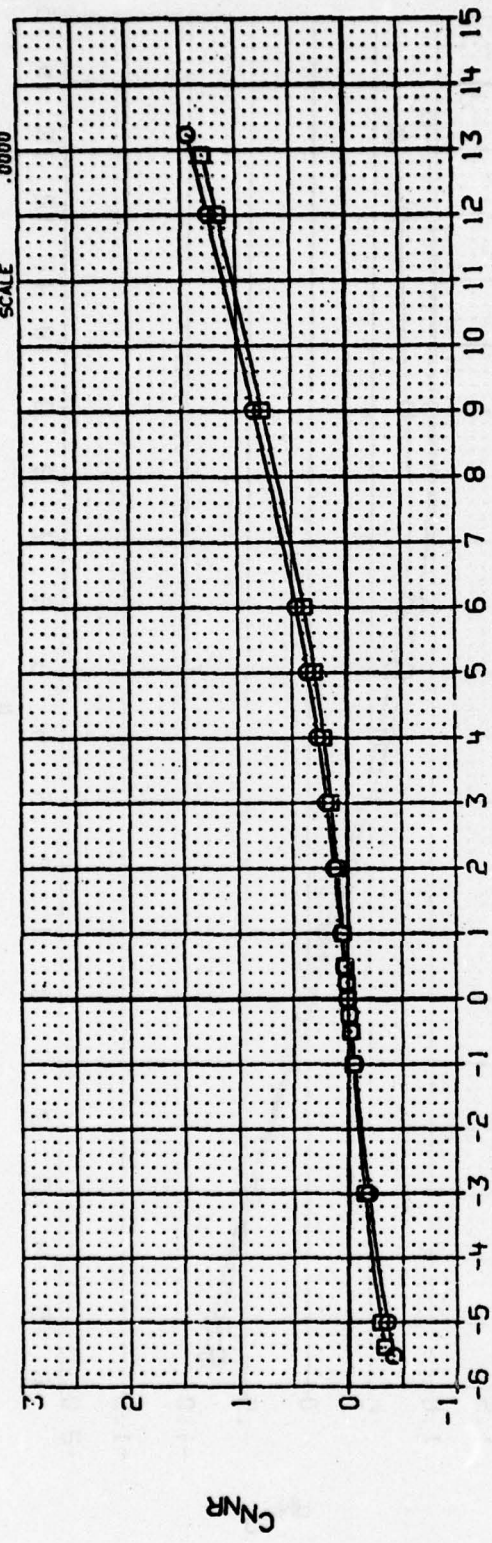
BODY ALONE LONGITUDINAL CHARACTERISTICS
TAIL OFF

(A) MACH = 3.01

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000

PHIND
 .000
 .000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH001) AEDC V41A-CIA, CANARD CONTROL, BN1
 (DXH060) AEDC V41A-CIA, CANARD CONTROL, BN2

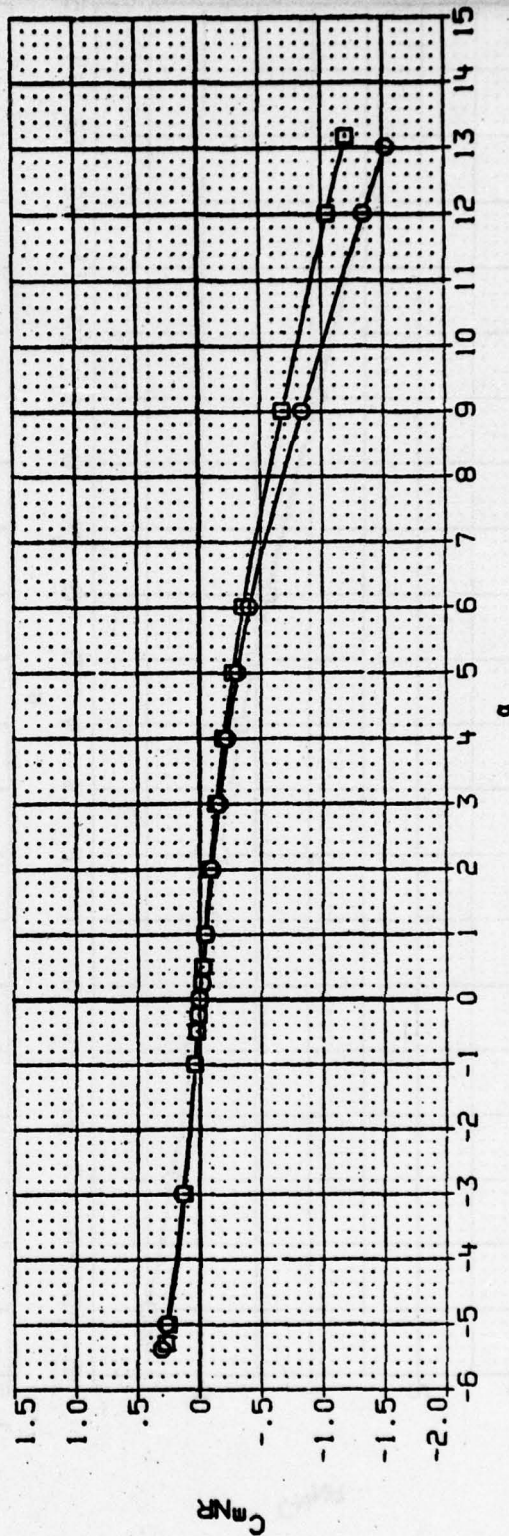
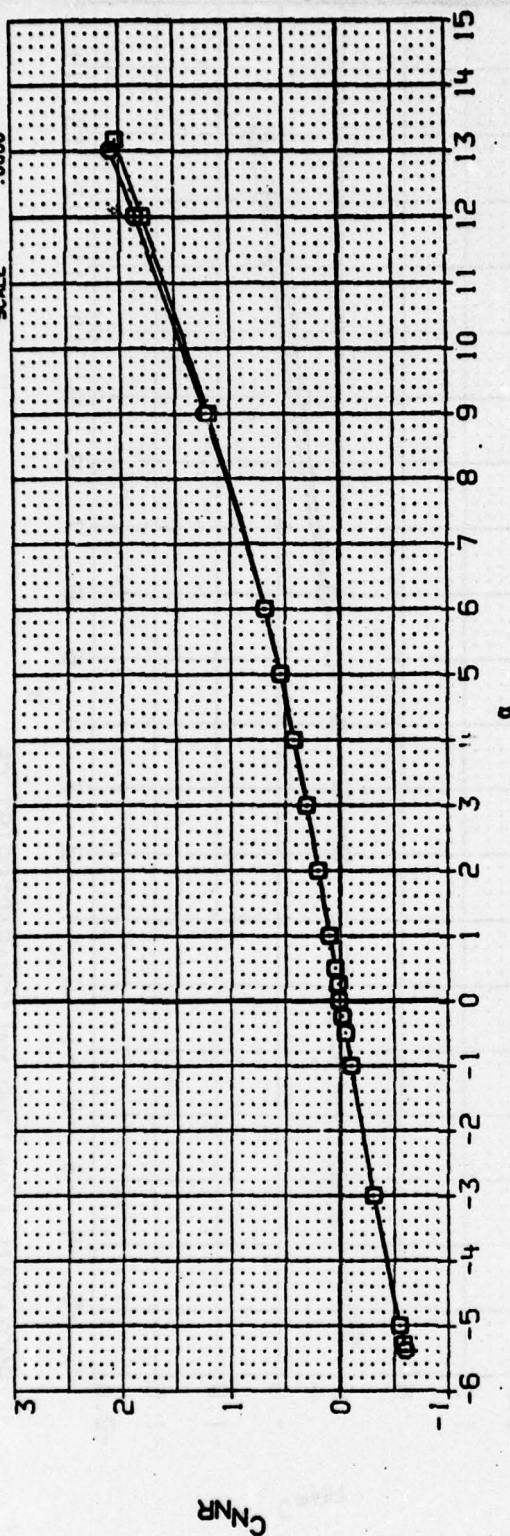


BODY ALONE LONGITUDINAL CHARACTERISTICS
 TAIL OFF
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX-002) ☐ AEDC V-1A-C1A, CANARD CONTROL, BNIT
 (DX-003) ☐ AEDC V-1A-C1A, CANARD CONTROL, BNIT

PHICND
 .000
 45.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 9.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



BODY-TAIL LONGITUDINAL CHARACTERISTICS

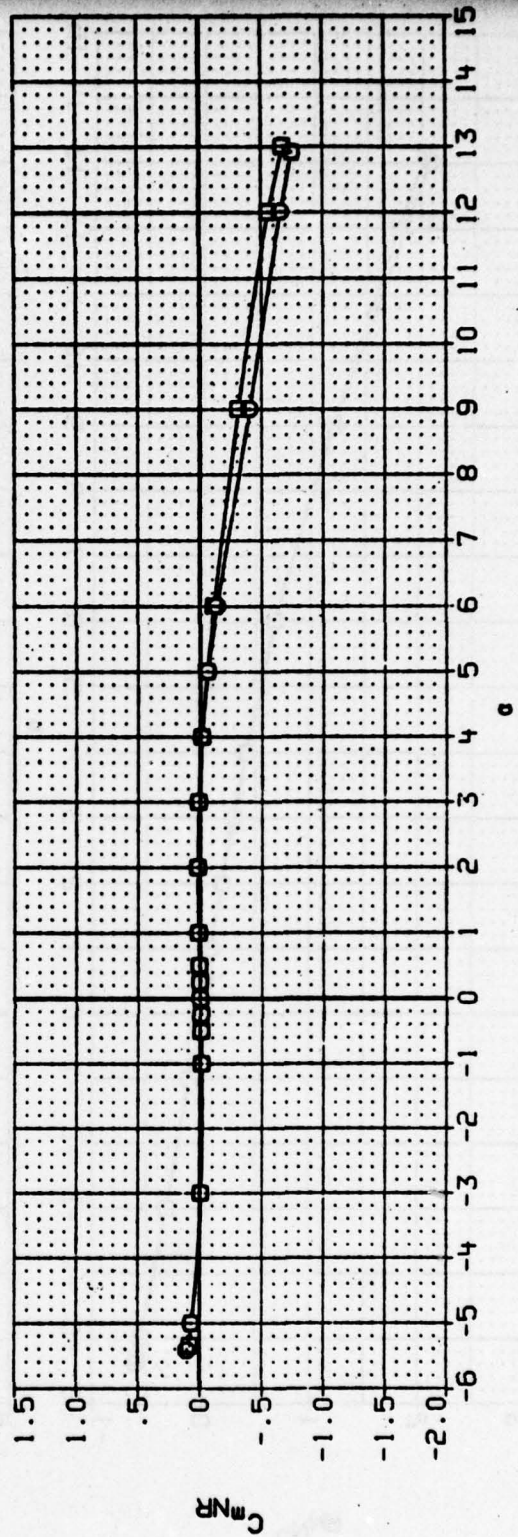
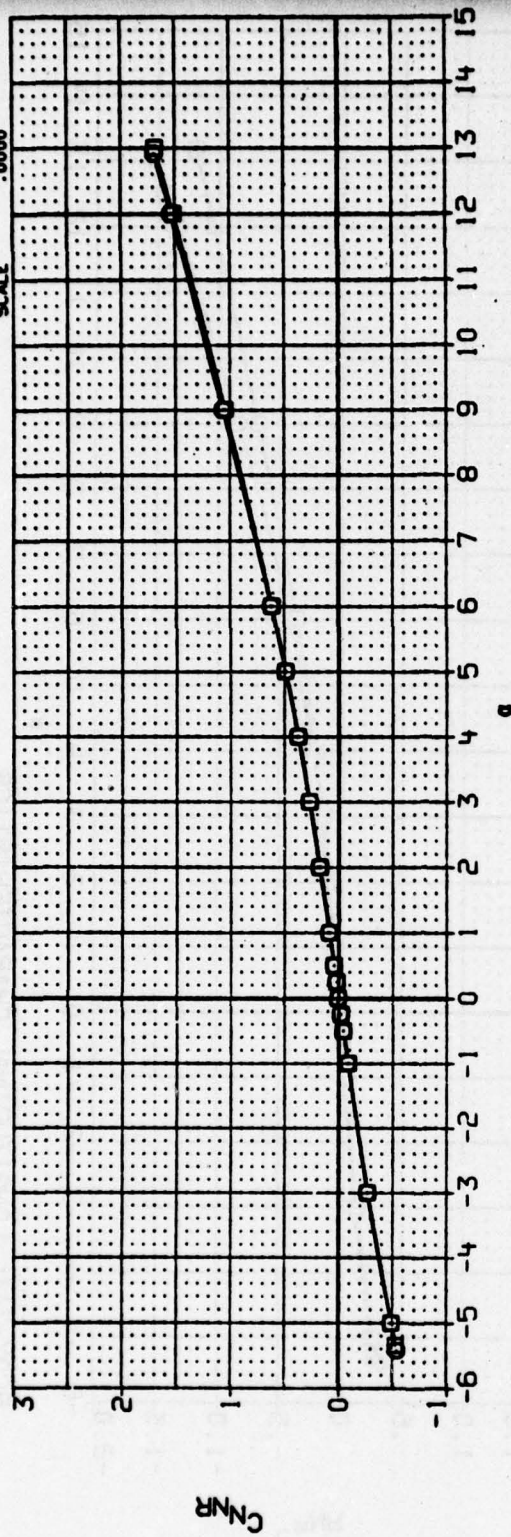
PHITAIL=0
 (A) MACH = 3.01

DATA SET SYMBOL
(DXH002)
(DXH003)

CONFIGURATION DESCRIPTION
AEDC V1A-CIA, CANARD CONTROL, BN111
AEDC V1A-CIA, CANARD CONTROL, BN111

PHICND
.000
45.000

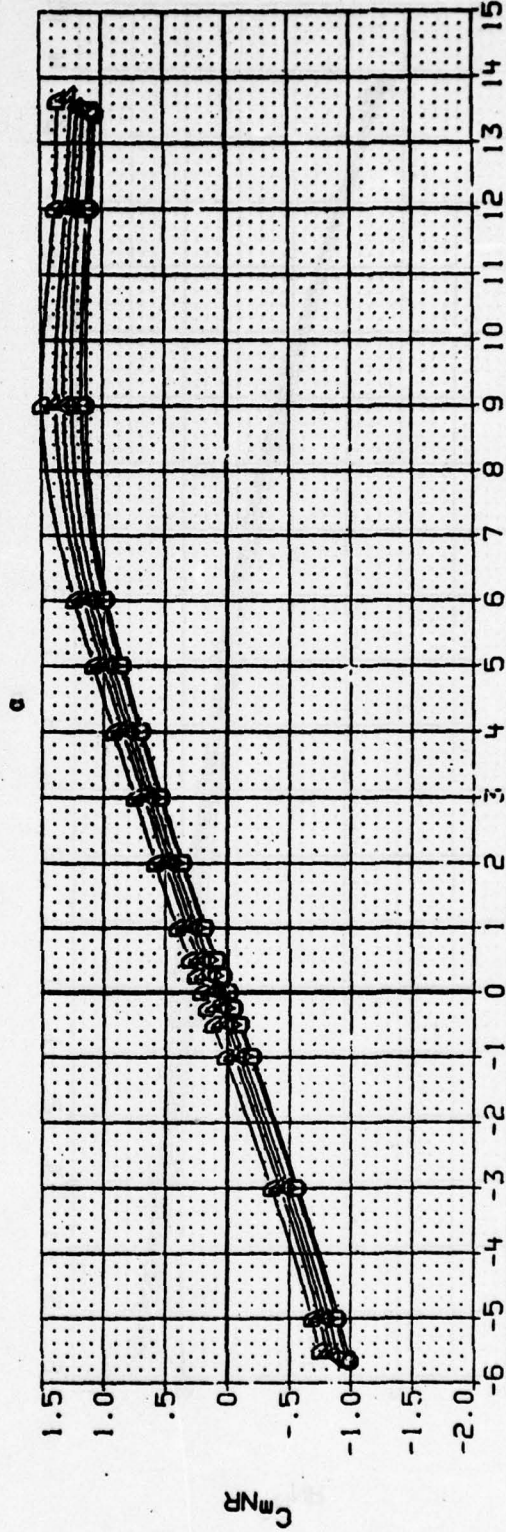
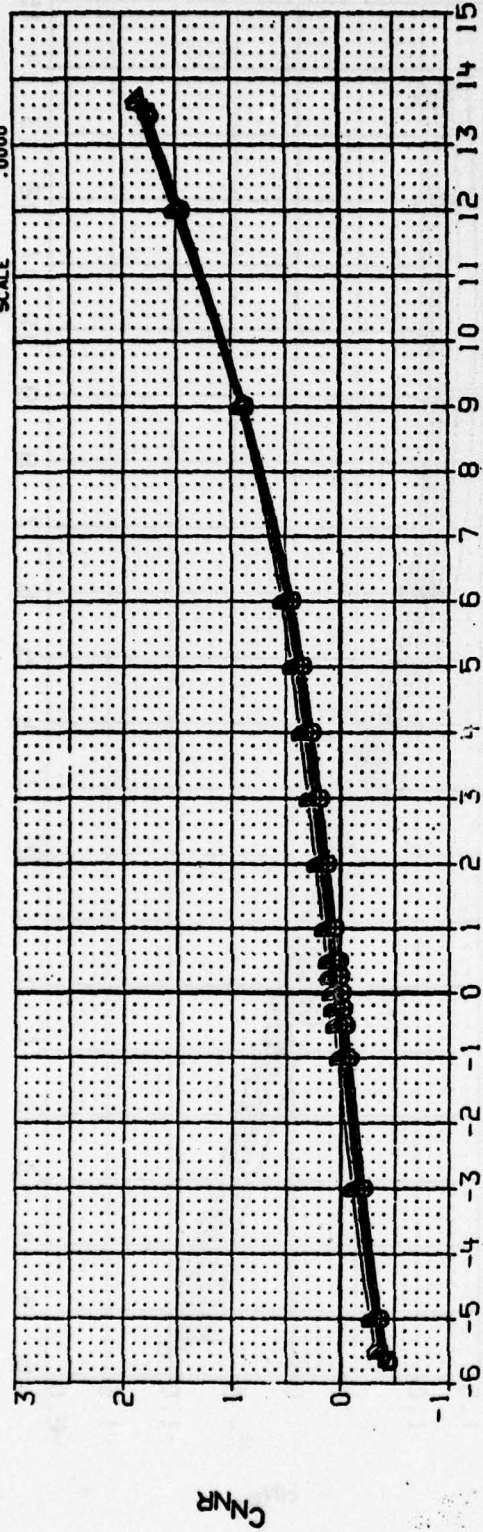
REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XREF 26.0000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000



BODY-TAIL LONGITUDINAL CHARACTERISTICS

PHITAIL=0
(B) MACH = 4.52

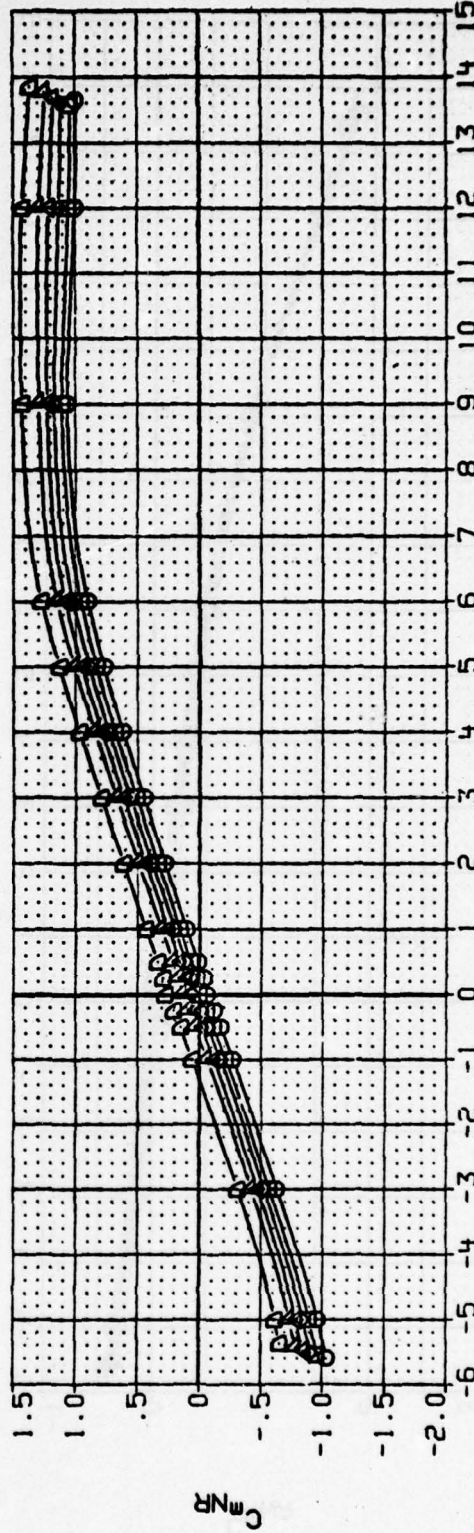
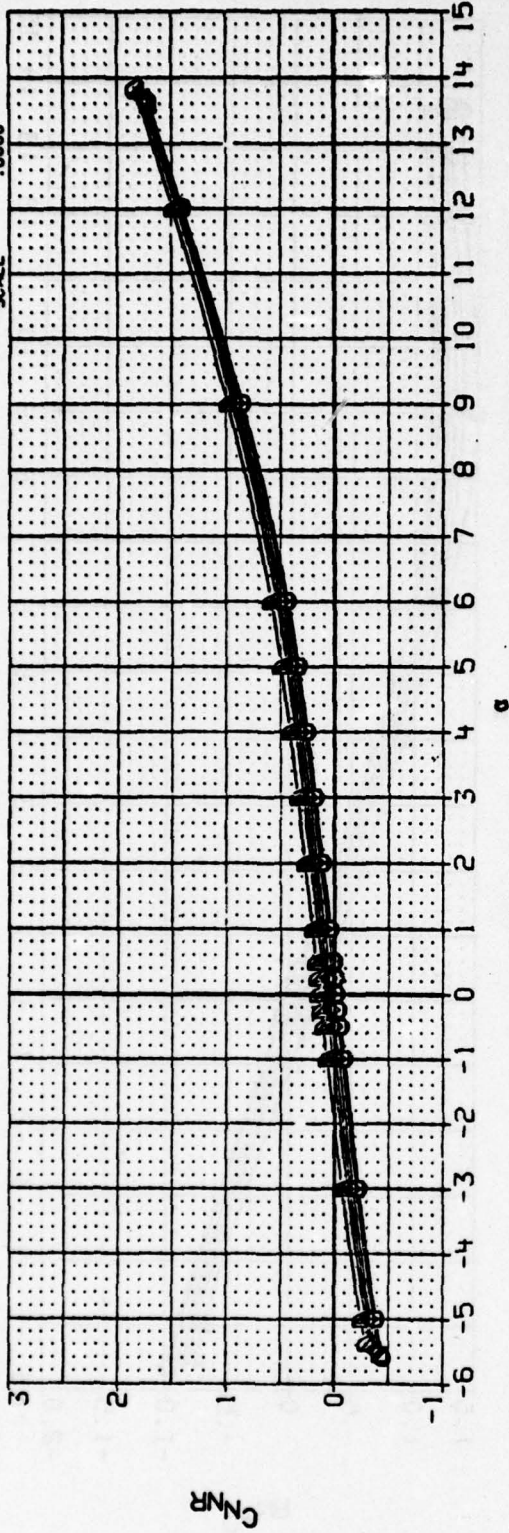
DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH004)	○	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	-3.000	.000	-3.000	SREF 19.6350 SQ. IN.
(DXH005)	◇	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH006)	◊	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	3.000	.000	3.000	BREF 5.0000 IN.
(DXH007)	△	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	6.000	.000	6.000	XREF 26.0000 IN.
(DXH008)	▽	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	9.000	.000	9.000	YREF .0000 IN.
(DXH009)	◻	AEDC V41A-C1A, CANARD CONTROL, BN1C1	.000	15.000	.000	15.000	ZREF .0000 IN.



BODY-TAIL LONGITUDINAL CHARACTERISTICS
TAIL OFF PHICND=0

(A) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH010)	□	AEDC W1A-C1A, CANARD CONTROL, BNICI	-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ. IN.
(DXH011)	□	AEDC W1A-C1A, CANARD CONTROL, BNICI	3.000	3.000	3.000	3.000	LREF 5.0000 IN.
(DXH012)	◇	AEDC W1A-C1A, CANARD CONTROL, BNICI	3.000	3.000	3.000	3.000	BREF 5.0000 IN.
(DXH013)	◇	AEDC W1A-C1A, CANARD CONTROL, BNICI	6.000	6.000	6.000	6.000	XPRP 26.0000 IN.
(DXH014)	◇	AEDC W1A-C1A, CANARD CONTROL, BNICI	9.000	9.000	9.000	9.000	YPRP .0000 IN.
(DXH015)	◇	AEDC W1A-C1A, CANARD CONTROL, BNICI	15.000	15.000	15.000	15.000	ZPRP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=0 PHICND=45
 (A) MACH = 3.01

DATA NOT AVAILABLE	DATA NOT AVAILABLE	DATA NOT AVAILABLE	DATA NOT AVAILABLE
AEDC V41A-C1A, CANARD CONTROL, ENICITI			
DATA NOT AVAILABLE	DATA NOT AVAILABLE	DATA NOT AVAILABLE	DATA NOT AVAILABLE

DATA SET	SYMBOL
(DXH016)	○
(DXH017)	□
(DXH018)	◇
(DXH019)	△

[illegible]

WHEEL	CONFIGURATION DESCRIPTION
0	DATA NOT AVAILABLE
1	AEDC V41A-C1A, CANARD CONTROL, ENICITI
2	DATA NOT AVAILABLE
3	DATA NOT AVAILABLE
4	DATA NOT AVAILABLE

DCND!
 .000
 .000
 .000
 .000

DCND2
-3.000
.000
1.000
3.000

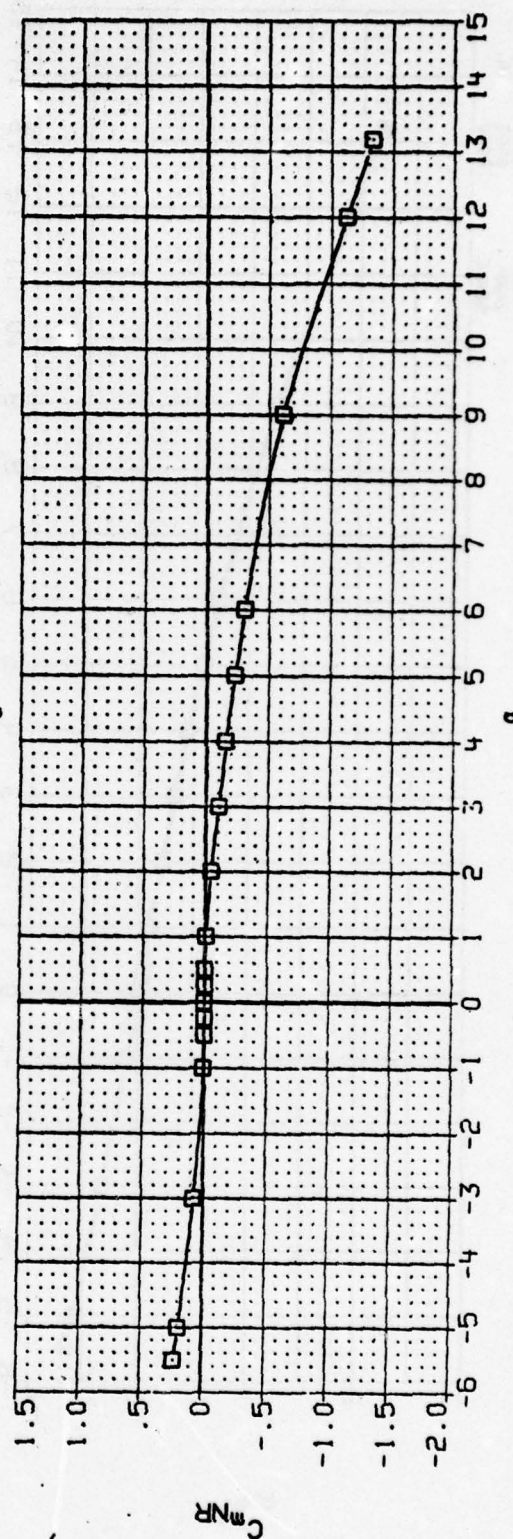
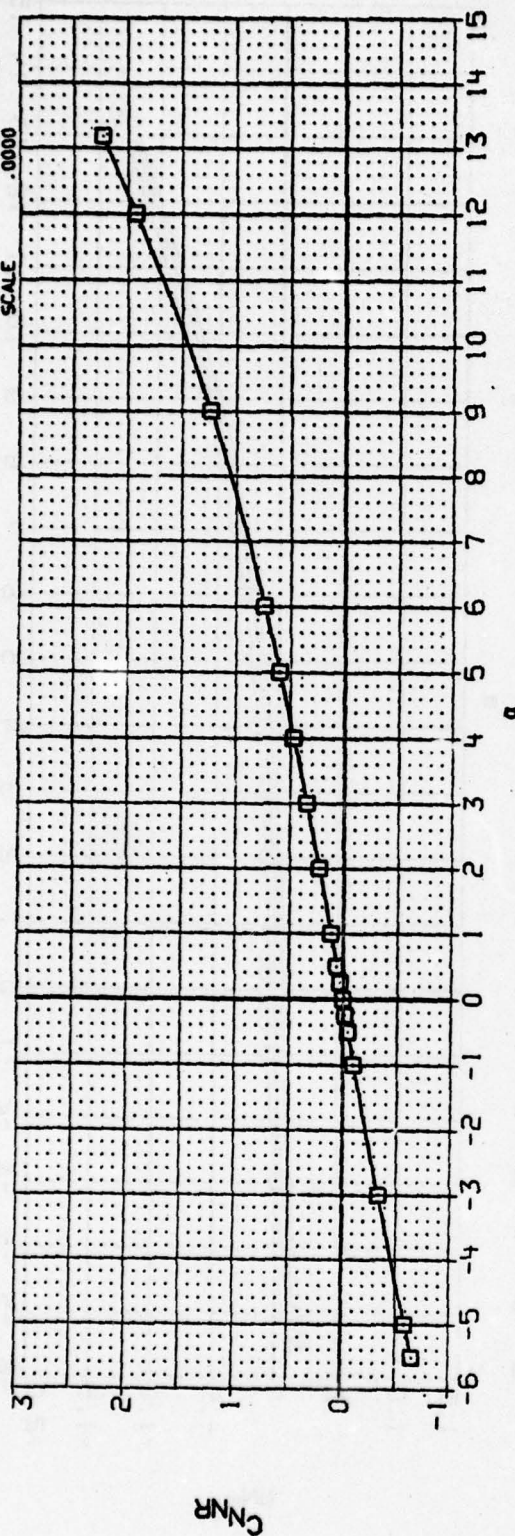
DCND3
.000
.000
.000
.000

3.000
.000
1.000
3.000

REFERENCE INFORMATION	
SREF	19.6350 IN.
LREF	5.0000 IN.
BREF	5.0000 IN.
XREF	26.0000 IN.

REFERENCE INFORMATION	
SREF	19.6350 SQ. IN.
LREF	5.0000 IN.
BREF	5.0000 IN.
XPRP	26.0000 IN.

REF
SREF
LREF
BREF
XMRP



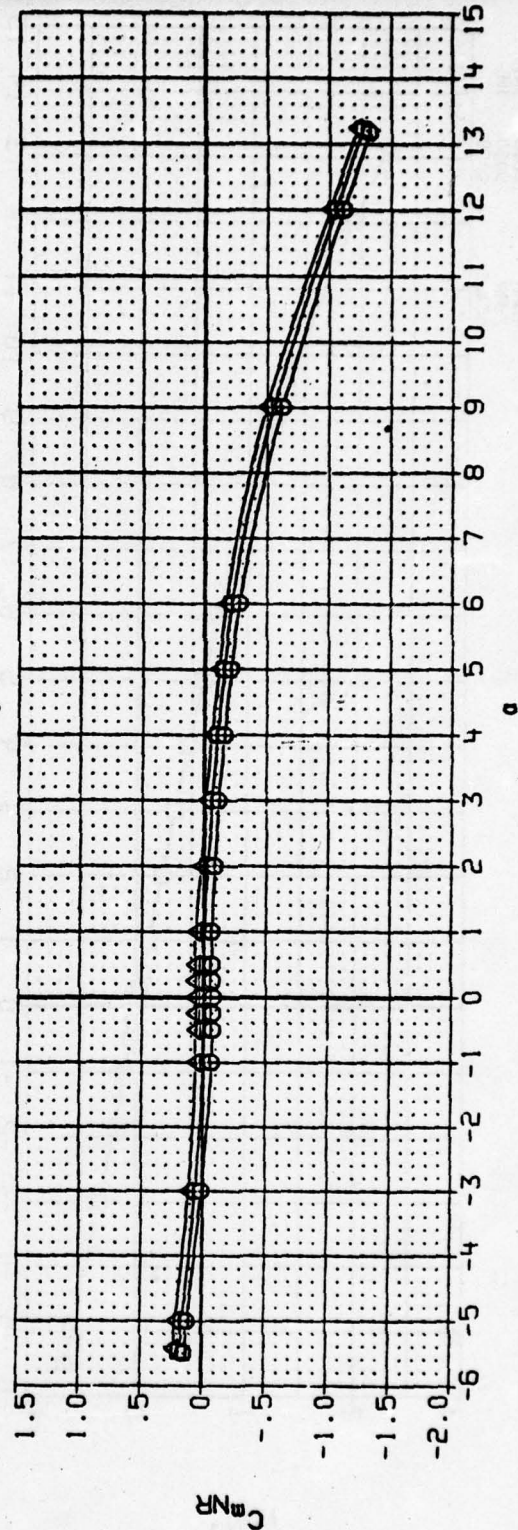
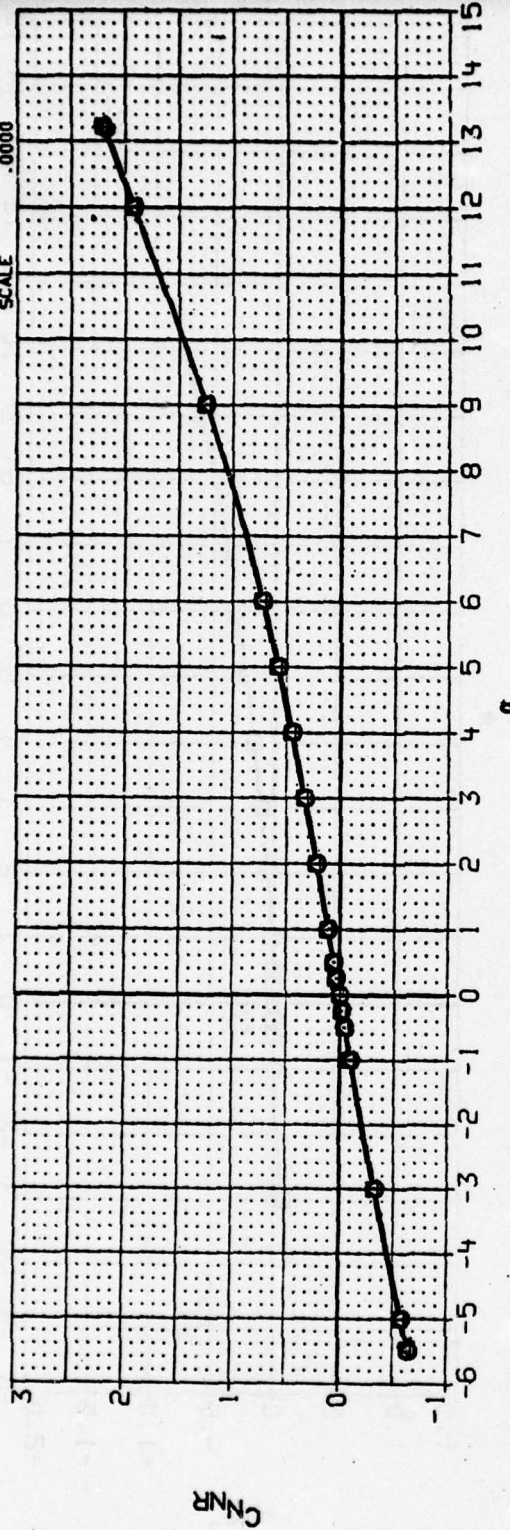
(A) MACH = 2.50
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHITAL=0 PHICND=0

DATA SET SYMBOL
(DXH016)
(DXH017)
(DXH018)
(DXH019)

CONFIGURATION DESCRIPTION
AEDC W1A-CIA, CANARD CONTROL, ENICITI
AEDC W1A-CIA, CANARD CONTROL, ENICITI
DATA NOT AVAILABLE
AEDC W1A-CIA, CANARD CONTROL, ENICITI

DCND1 DCND2 DCND3 DCND4
.000 -3.000 .000 -3.000
.000 .000 .000 .000
.000 1.000 .000 1.000
.000 3.000 .000 3.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 25.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000

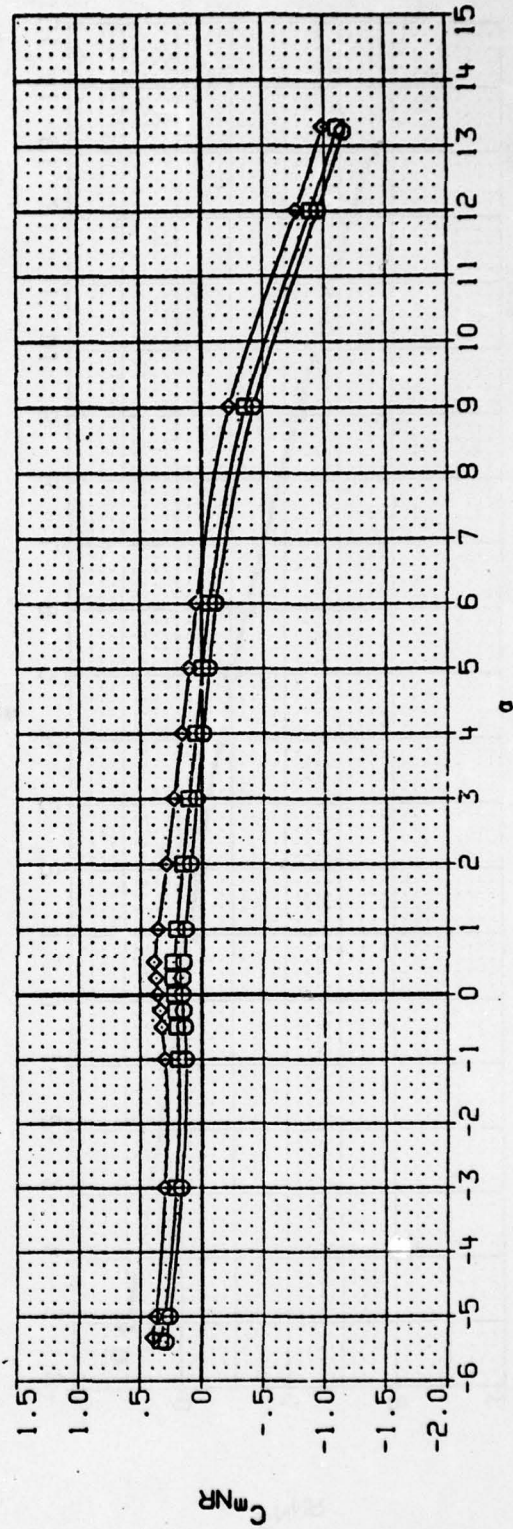
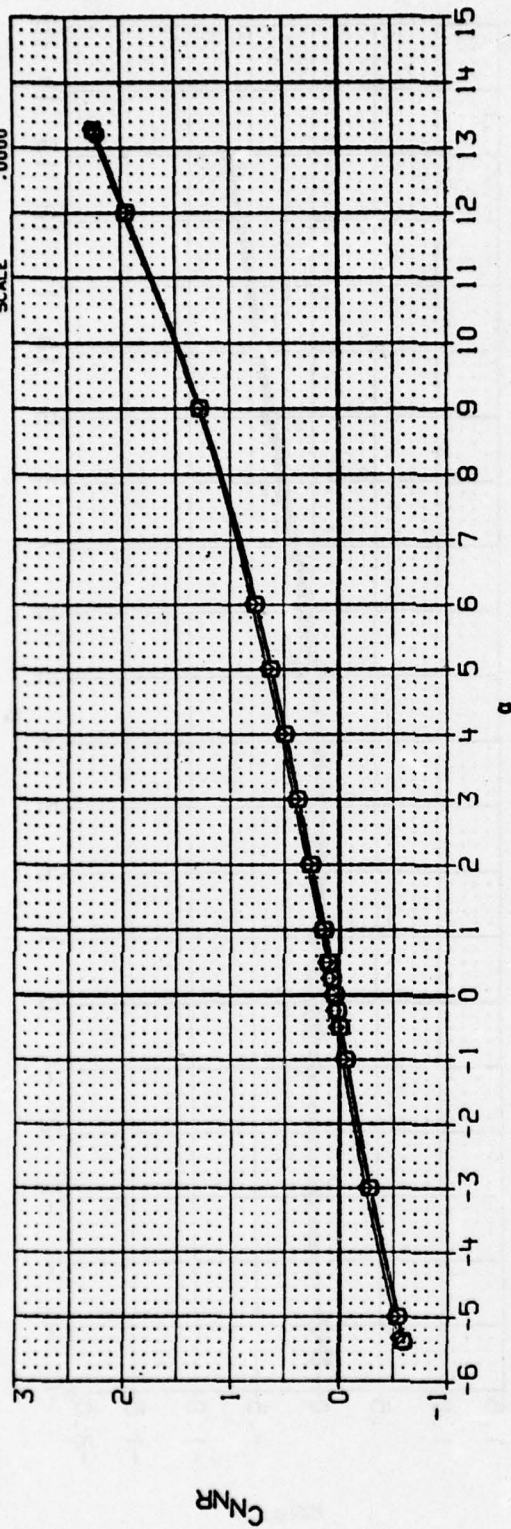


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHICAL=0 PHICND=0
(B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH020) AEDC V1A-CIA, CANARD CONTROL, BNICITI
 (DXH021) AEDC V1A-CIA, CANARD CONTROL, BNICITI
 (DXH022) AEDC V1A-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

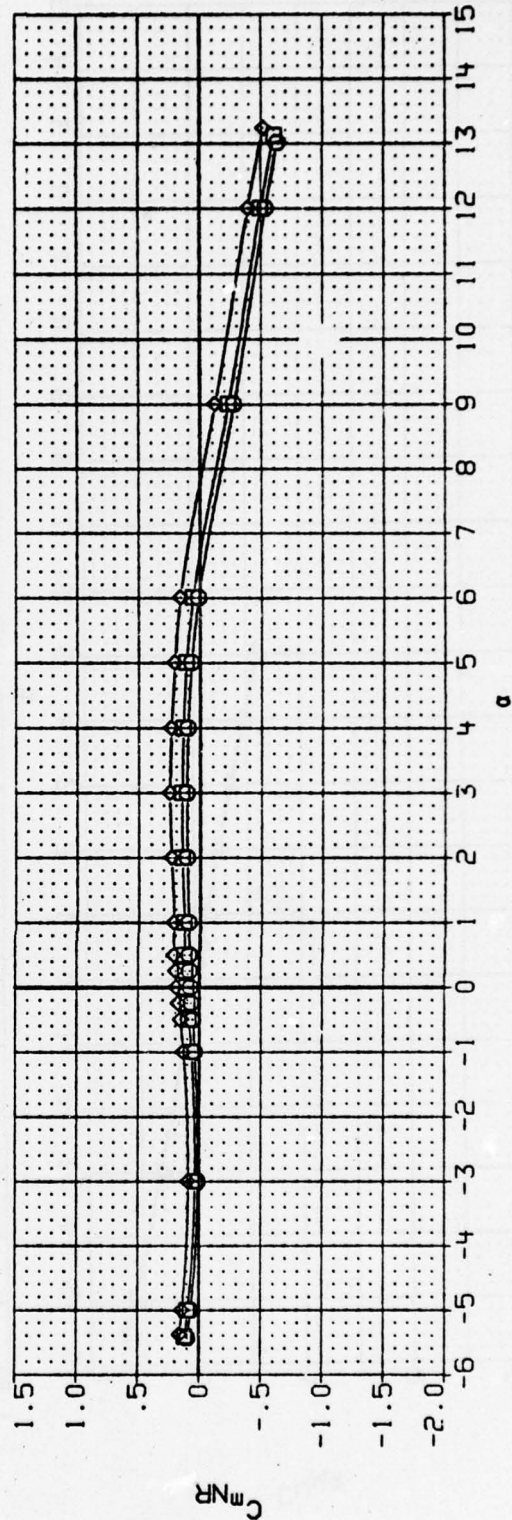
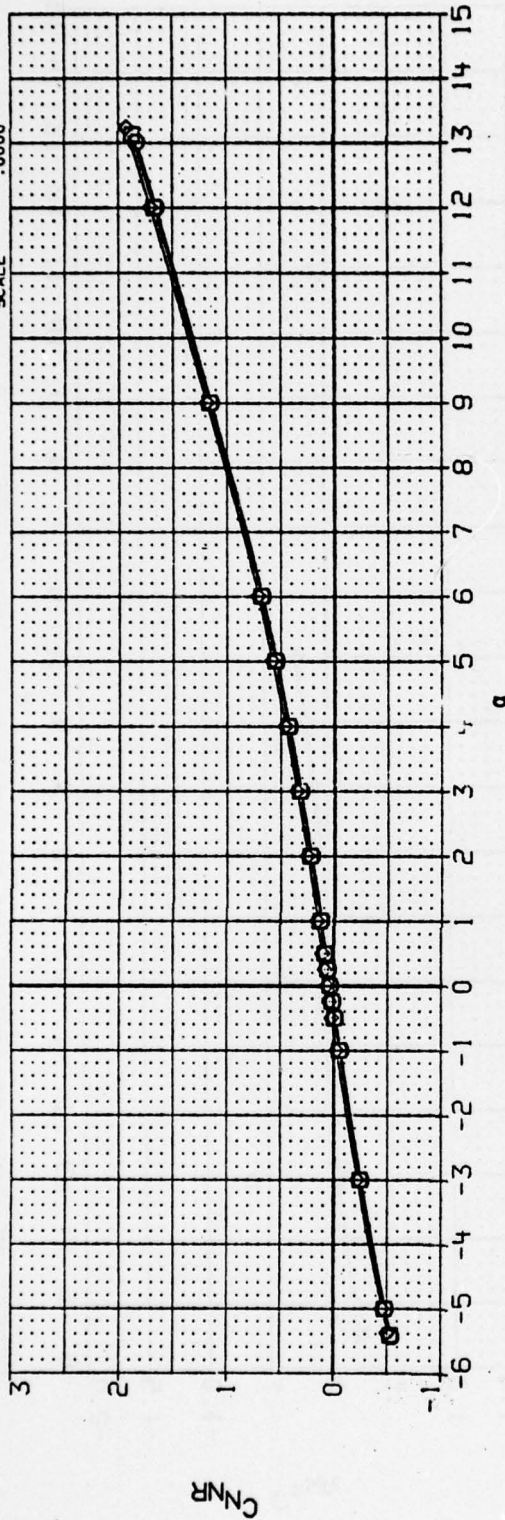
REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=0 PHICND=0

(A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH020)	○	AEDC V41A-C1A, CANARD CONTROL, BNIC111	.000	6.000	.000	6.000	SREF 19.6350 50. IN.
(DXH021)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC111	.000	9.000	.000	9.000	LREF 5.0000 IN.
(DXH022)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC111	.000	15.000	.000	15.000	BREF 5.0000 IN.
							XRRP 26.0000 IN.
							YRRP .0000 IN.
							ZRRP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

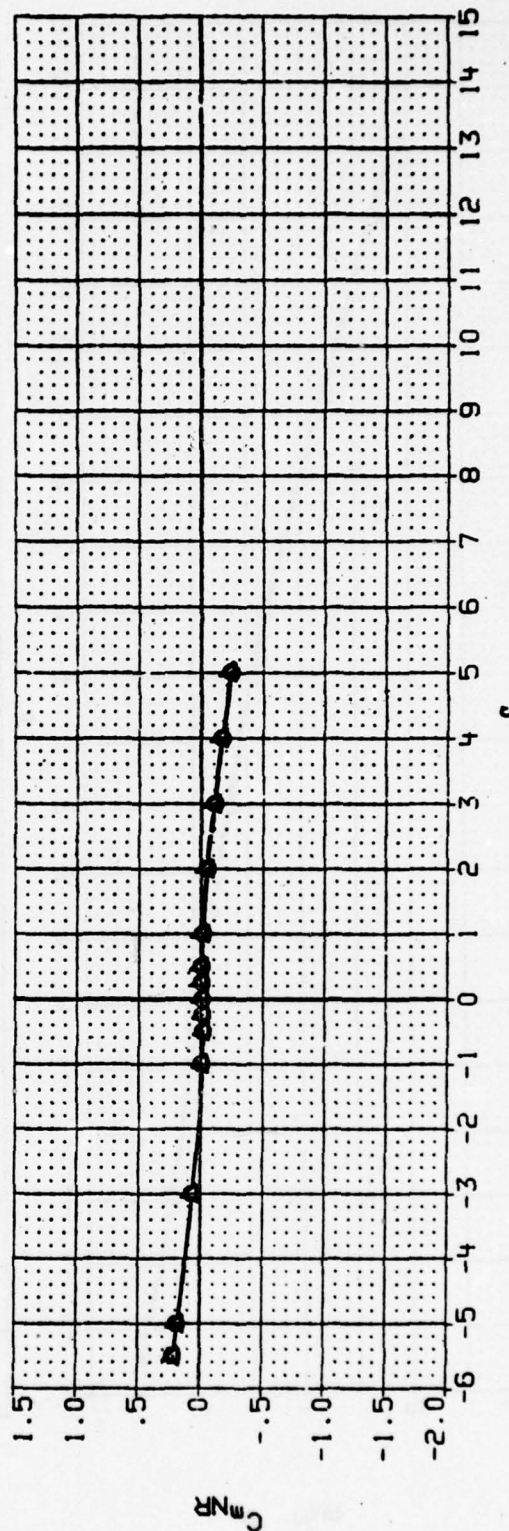
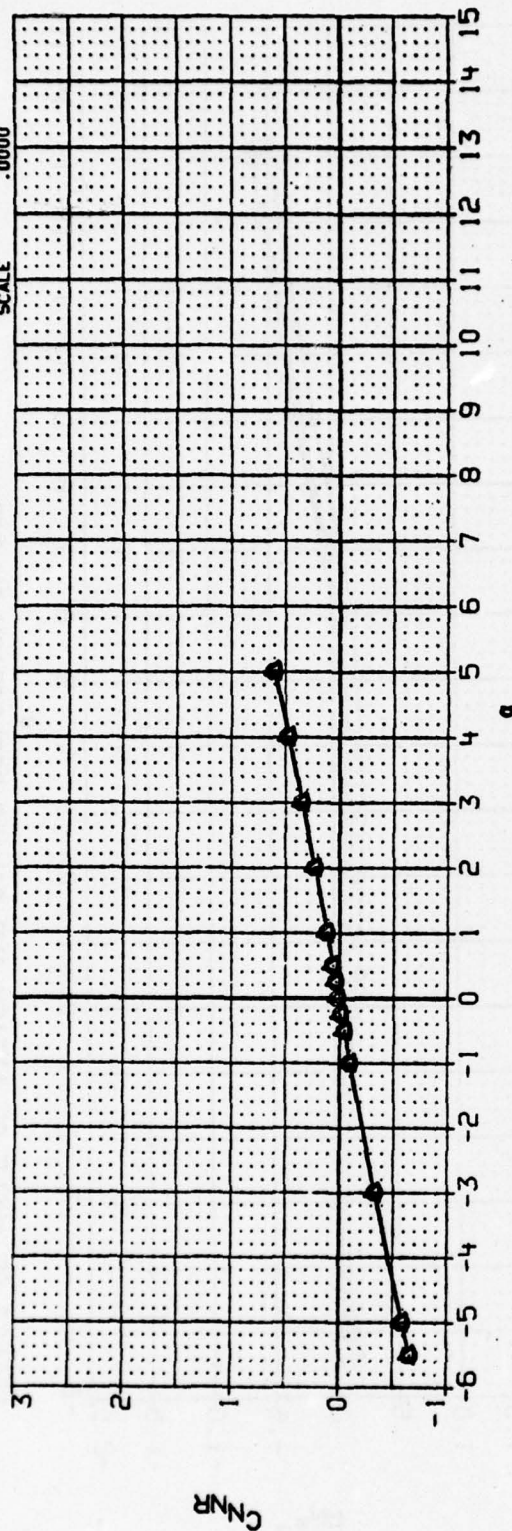
PHITAL=0 PHICND=0

(B) MACH = 4.52

AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
DATA NOT AVAILABLE
DATA NOT AVAILABLE
AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

	DCND1	DCND2	DCND3	DCND4
1	-3.000	3.000	3.000	-3.000
2	.500	-.500	-.500	.500
3	1.000	-1.000	-1.000	1.000
4	2.000	-2.000	-2.000	2.000
5	5.000	-5.000	-5.000	5.000

REFERENCE INFORMATION	
SREF	19.6350 50.
LREF	5.0000 IN.
BREF	5.0000 IN.
DOFP	26.0000 IN.
WFP	.0000 IN.
ZWFP	.0000 IN.
SCALE	.0000



(A)MACH = 2.50
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHITAL=0 PHICND=0

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH023) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1T1

(DXH024) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1T1

(DXH025) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1T1

(DXH026) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1T1

(DXH027) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 3.000 -3.000

1.500 -1.500 -1.500 1.500

2.000 -2.000 -2.000 2.000

5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

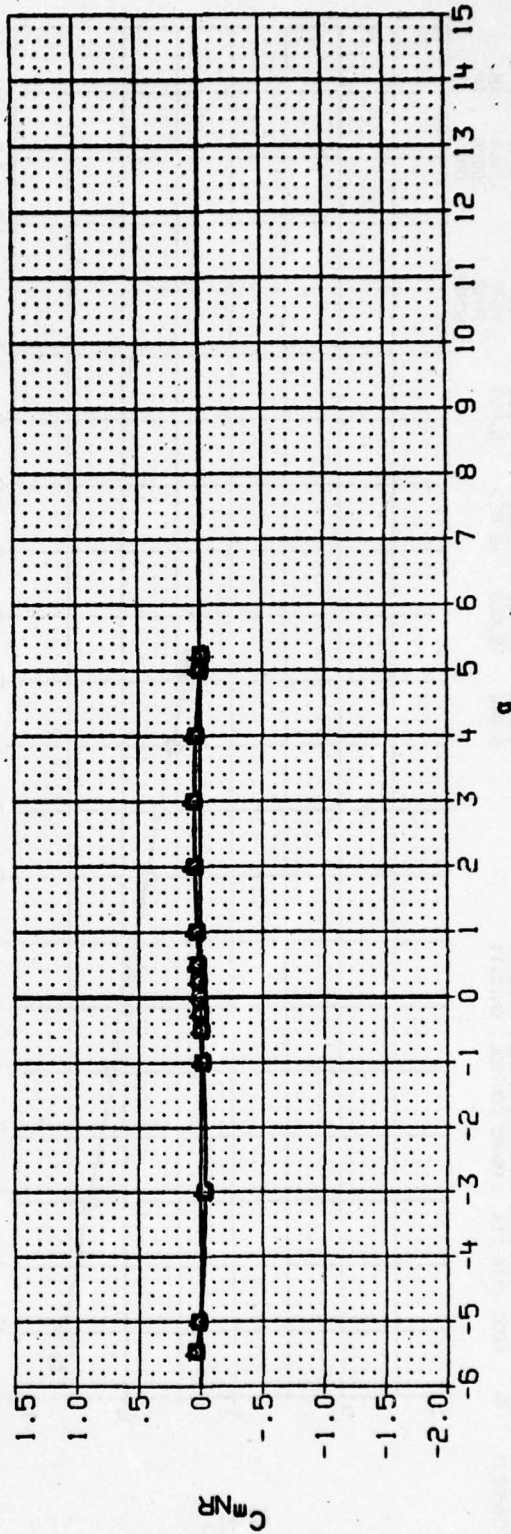
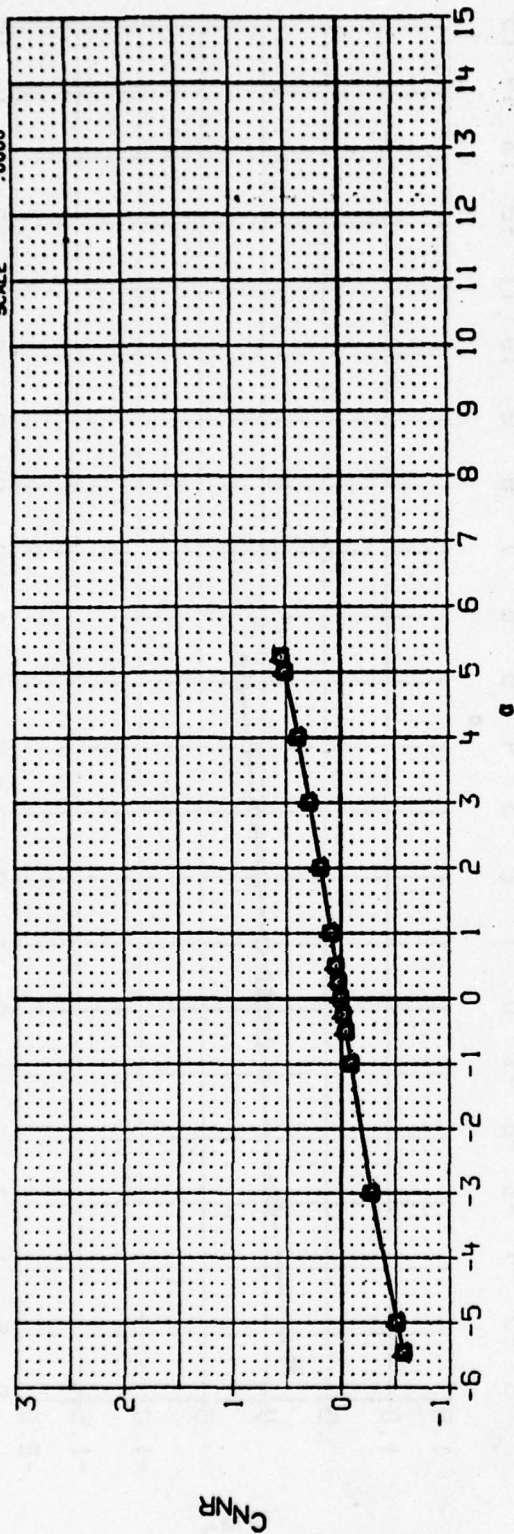
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000

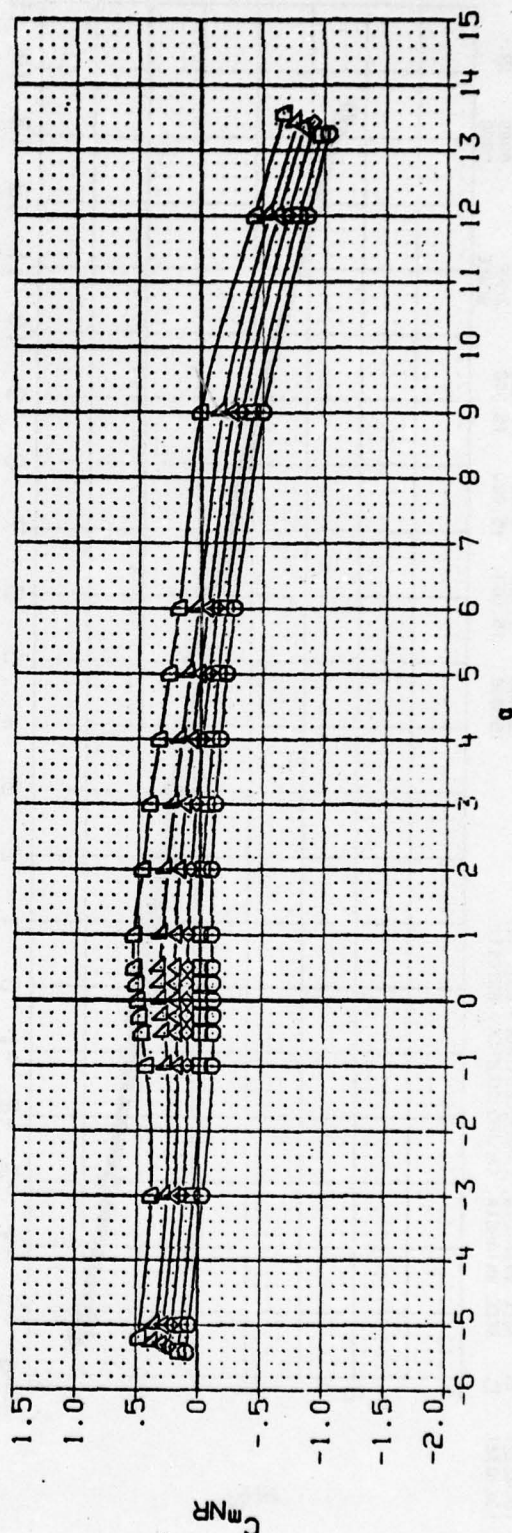
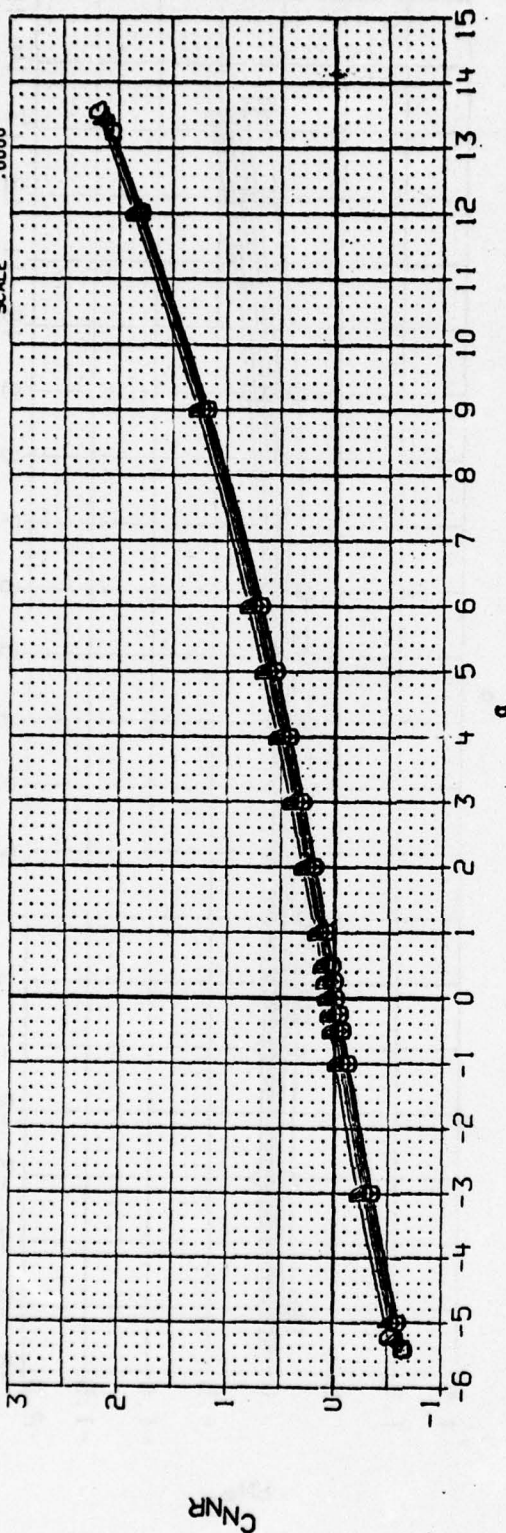


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHICAL=0 PHICND=0

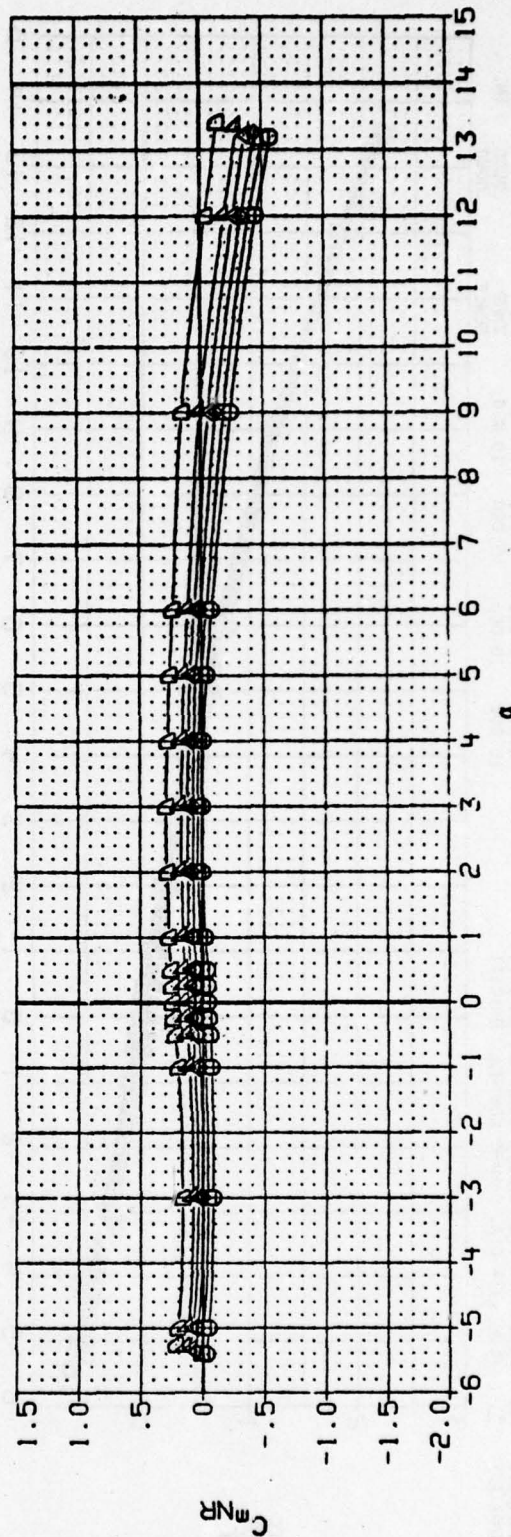
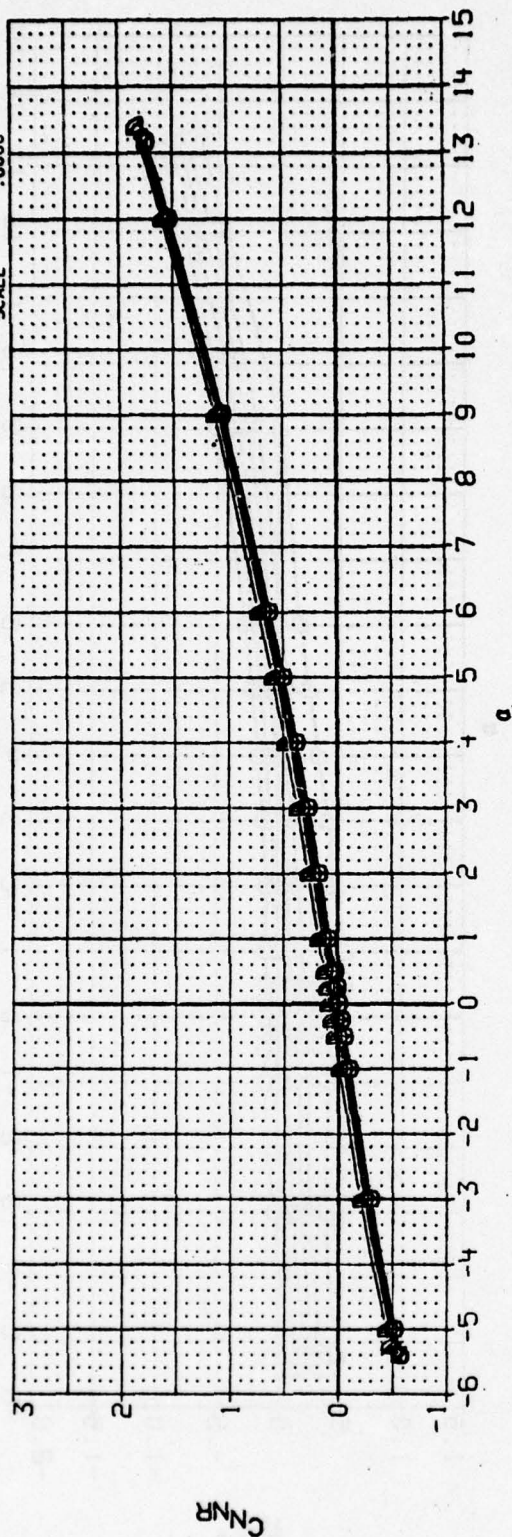
(C)MACH = 4.52

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH028)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ. IN.
(DXH029)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH030)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	3.000	3.000	3.000	3.000	BREF 5.0000 IN.
(DXH031)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	6.000	6.000	6.000	6.000	XPRP 26.0000 IN.
(DXH032)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	9.000	9.000	9.000	9.000	YPRP .0000 IN.
(DXH033)	AEDC V41A-CIA, CANARD CONTROL, BNICITI	15.000	15.000	15.000	15.000	ZPRP .0000 IN.
						SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=0 PHICND=45
 (A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH028)	□	AEDC W1A-CIA, CANARD CONTROL, BNICITI	-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ. IN.
(DXH029)	□	AEDC W1A-CIA, CANARD CONTROL, BNICITI	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH030)	◇	AEDC W1A-CIA, CANARD CONTROL, BNICITI	3.000	3.000	3.000	3.000	BREF 5.0000 IN.
(DXH031)	△	AEDC W1A-CIA, CANARD CONTROL, BNICITI	6.000	6.000	6.000	6.000	XMRP 26.0000 IN.
(DXH032)	△	AEDC W1A-CIA, CANARD CONTROL, BNICITI	9.000	9.000	9.000	9.000	YMRP .0000 IN.
(DXH033)	◇	AEDC W1A-CIA, CANARD CONTROL, BNICITI	15.000	15.000	15.000	15.000	ZMRP .0000 IN.
							SCALE .0000



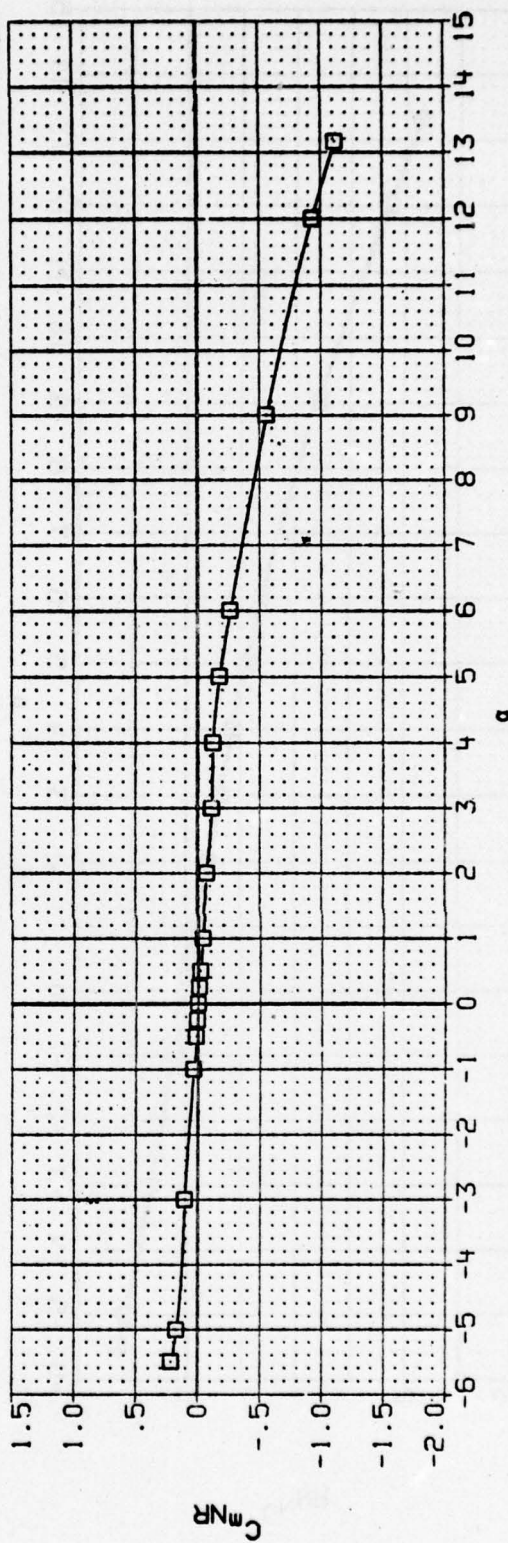
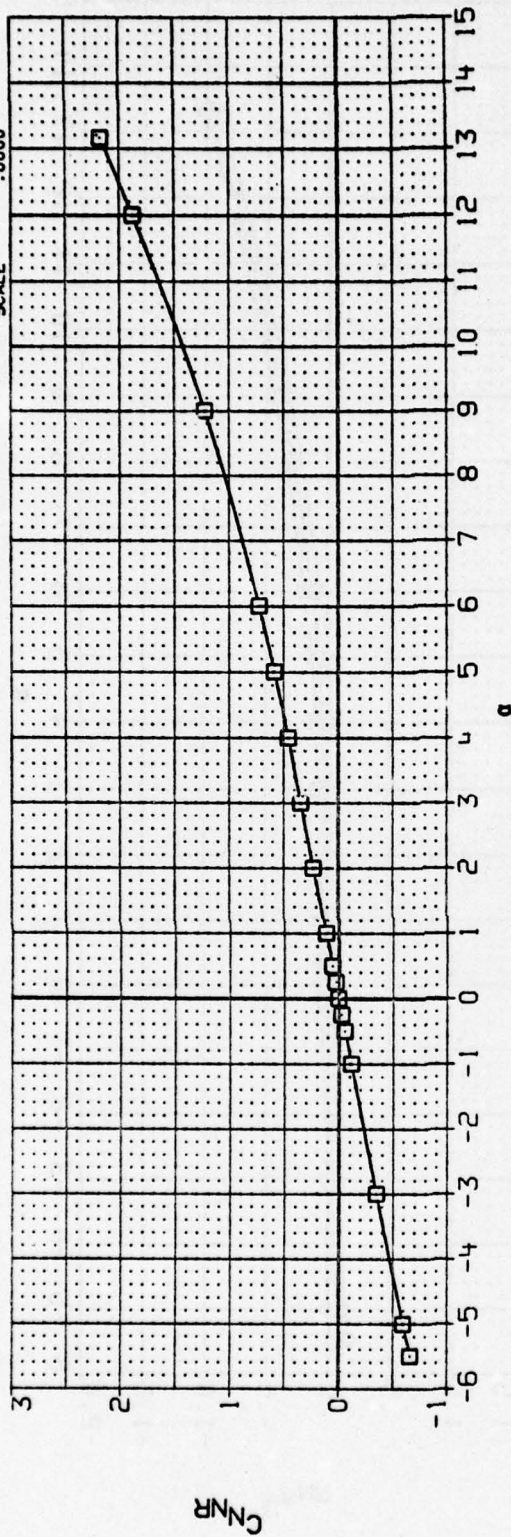
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 $\text{PHICAL} = 0$ $\text{PHICND} = 45$

(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH034) DATA NOT AVAILABLE
 (DXH035) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (DXH036) DATA NOT AVAILABLE
 (DXH037) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000

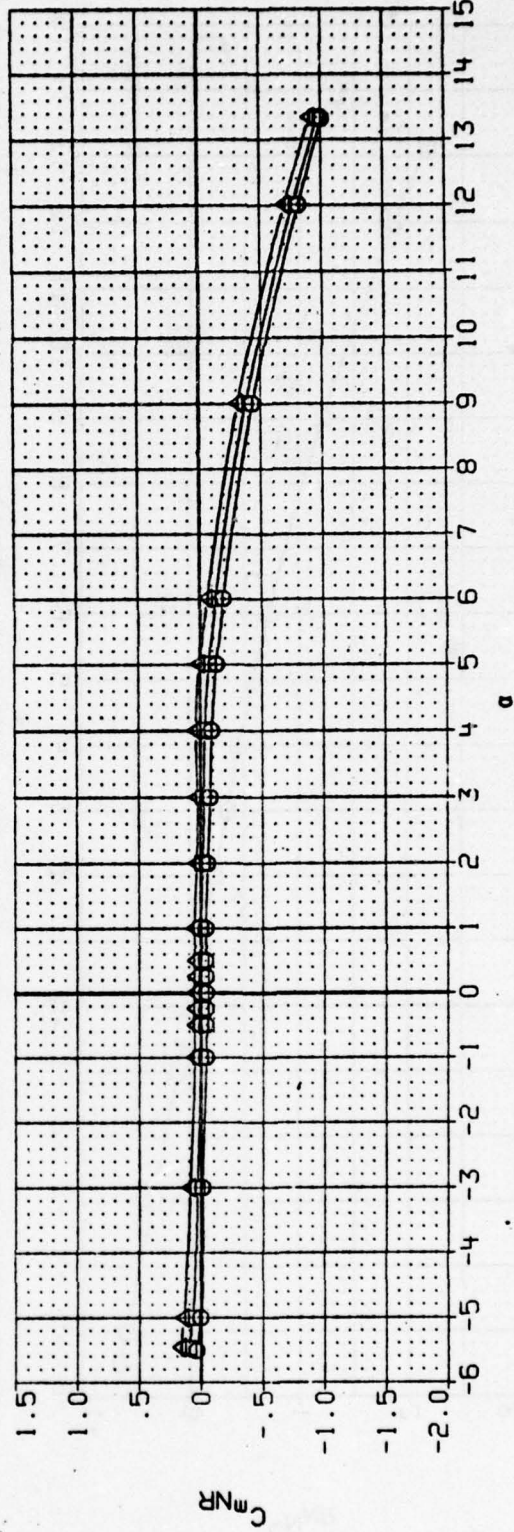
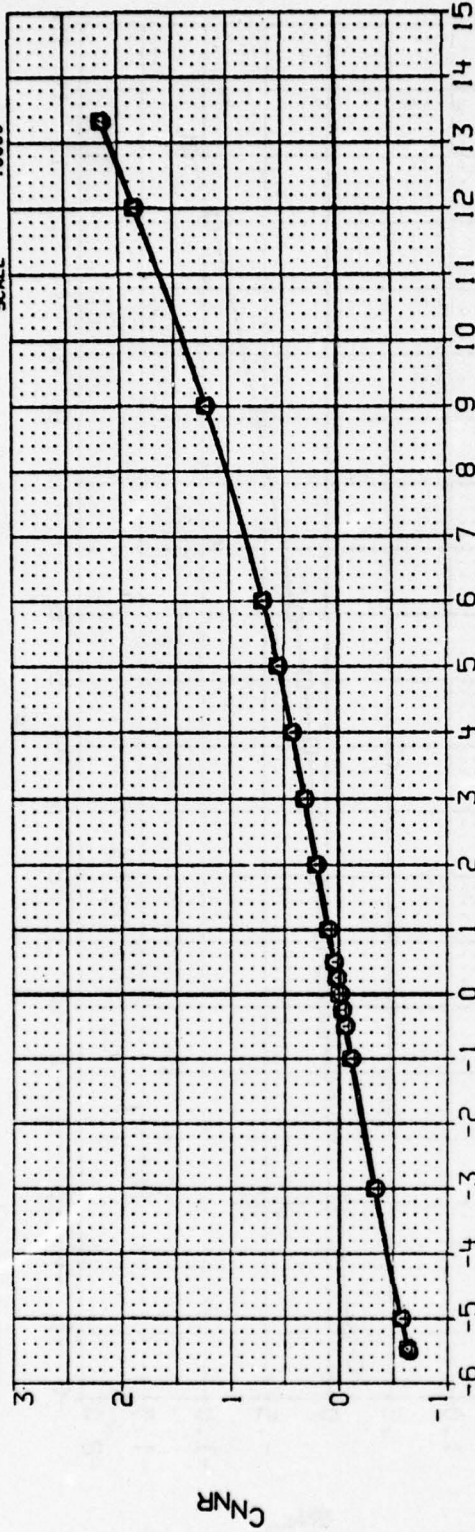


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=45 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH034) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (DXH035) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (DXH036) \square DATA NOT AVAILABLE
 (DXH037) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC111

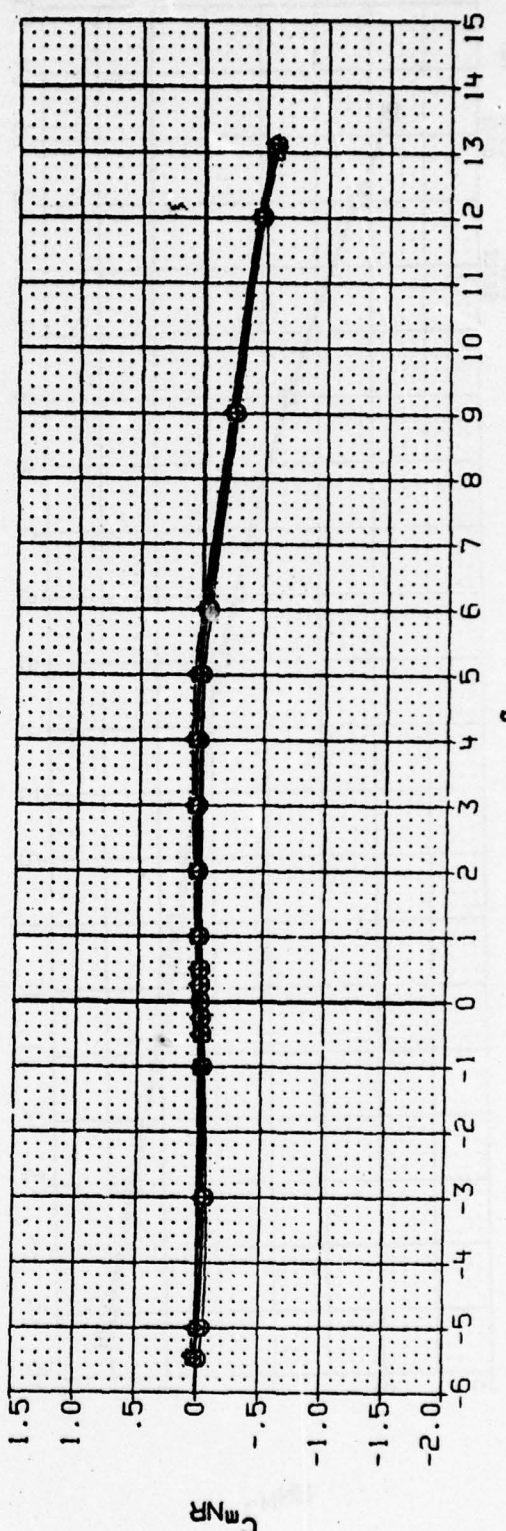
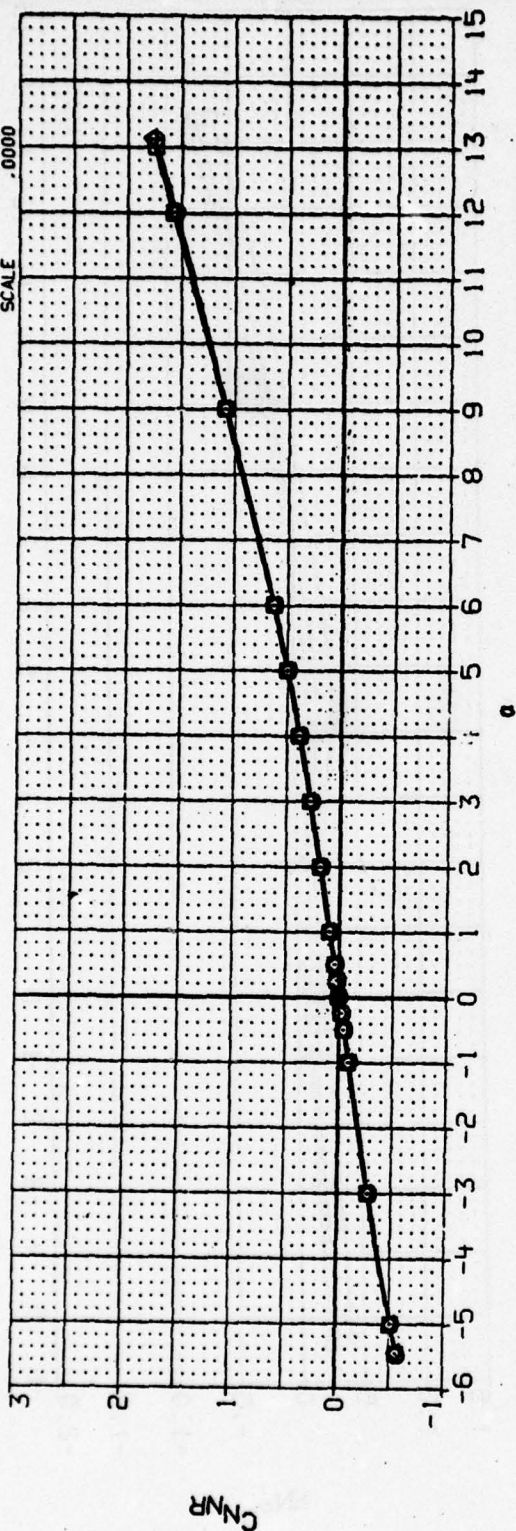
DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



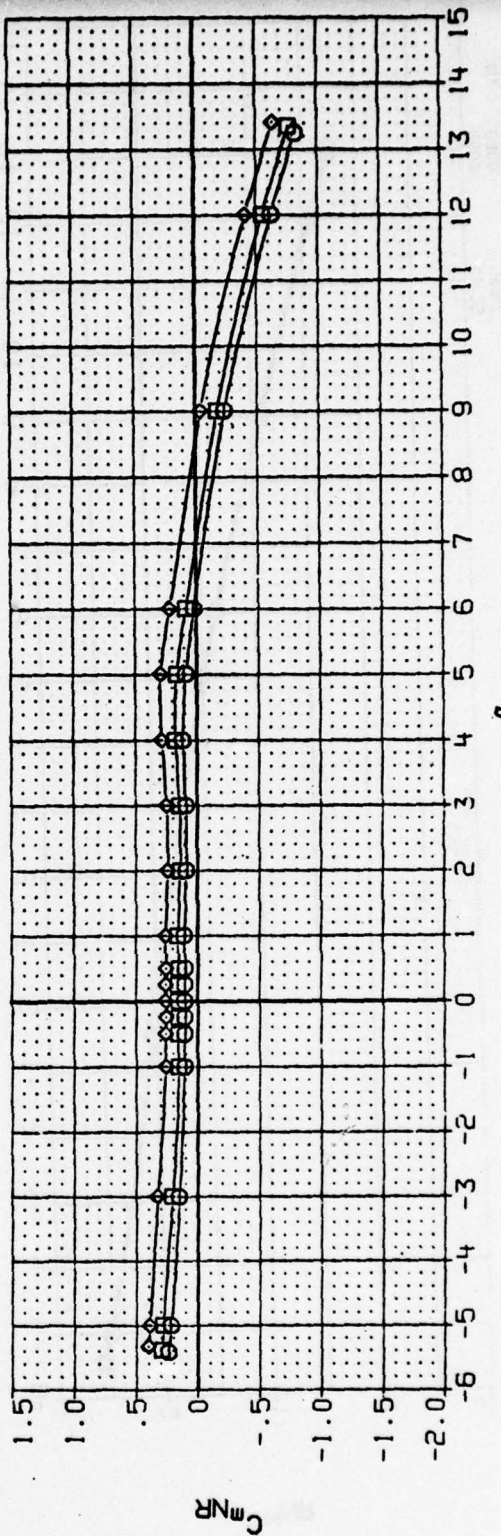
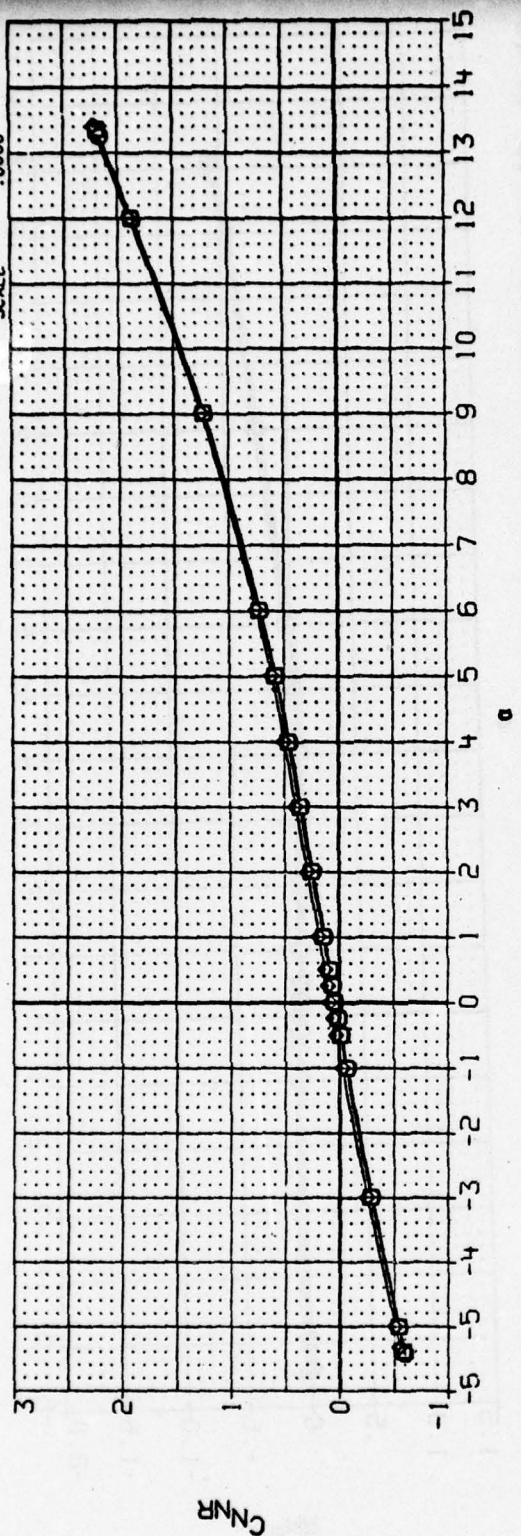
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=45 PHICND=0
 (B)MACH = 3.00

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH034)	AEDC V4IA-CIA, CANARD CONTROL, BNIC1T1	.000	-3.000	.000	-3.000	SREF 19.6350 SO. IN.
(DXH035)	AEDC V4IA-CIA, CANARD CONTROL, BNIC1T1	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH036)	AEDC V4IA-CIA, CANARD CONTROL, BNIC1T1	.000	1.000	.000	1.000	BREF 5.0000 IN.
(DXH037)	AEDC V4IA-CIA, CANARD CONTROL, BNIC1T1	.000	3.000	.000	3.000	YPRP 26.0000 IN.
						ZPRP .0000 IN.
						SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 $\text{PHICAL}=45$
 $\text{PHICND}=0$
 $(C) \text{MACH} = 4.52$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH038)	AEDC V4A-CIA, CANARD CONTROL, BNICITI	.000	6.000	.000	6.000	SREF 19.6350 SO. IN.
(DXH039)	AEDC V4A-CIA, CANARD CONTROL, BNICITI	.000	9.000	.000	9.000	LREF 5.0000 IN.
(DXH040)	AEDC V4A-CIA, CANARD CONTROL, BNICITI	.000	15.000	.000	15.000	BREF 5.0000 IN.
						YMRP 25.0000 IN.
						ZMRP .0000 IN.
						SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=45 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

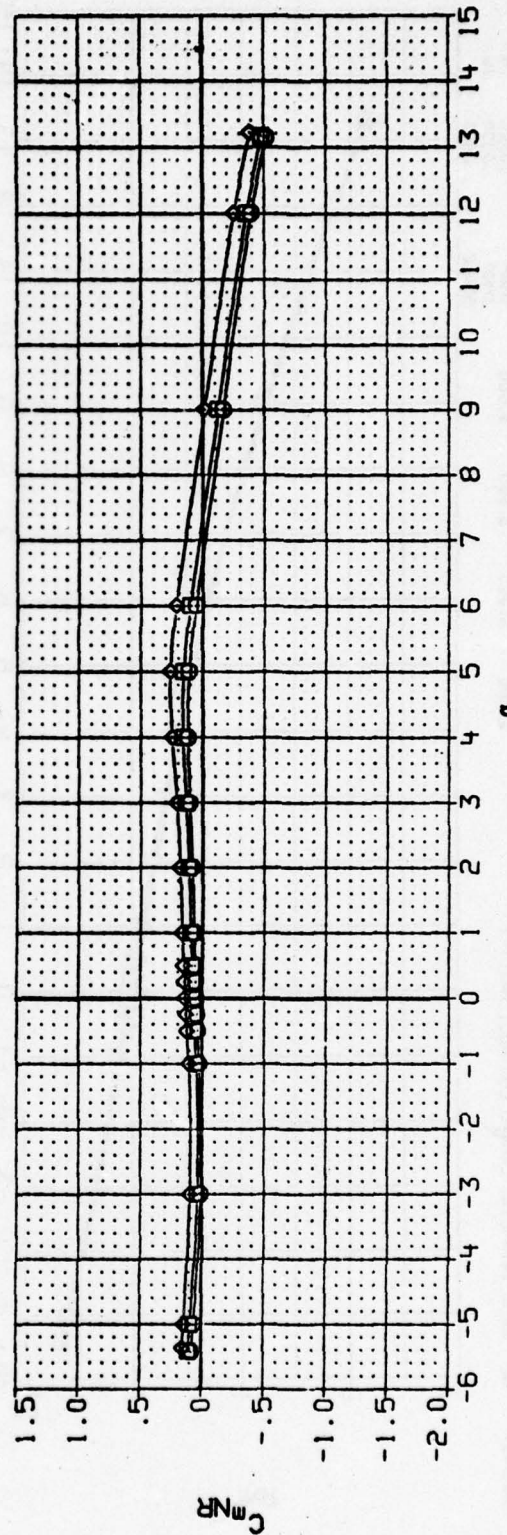
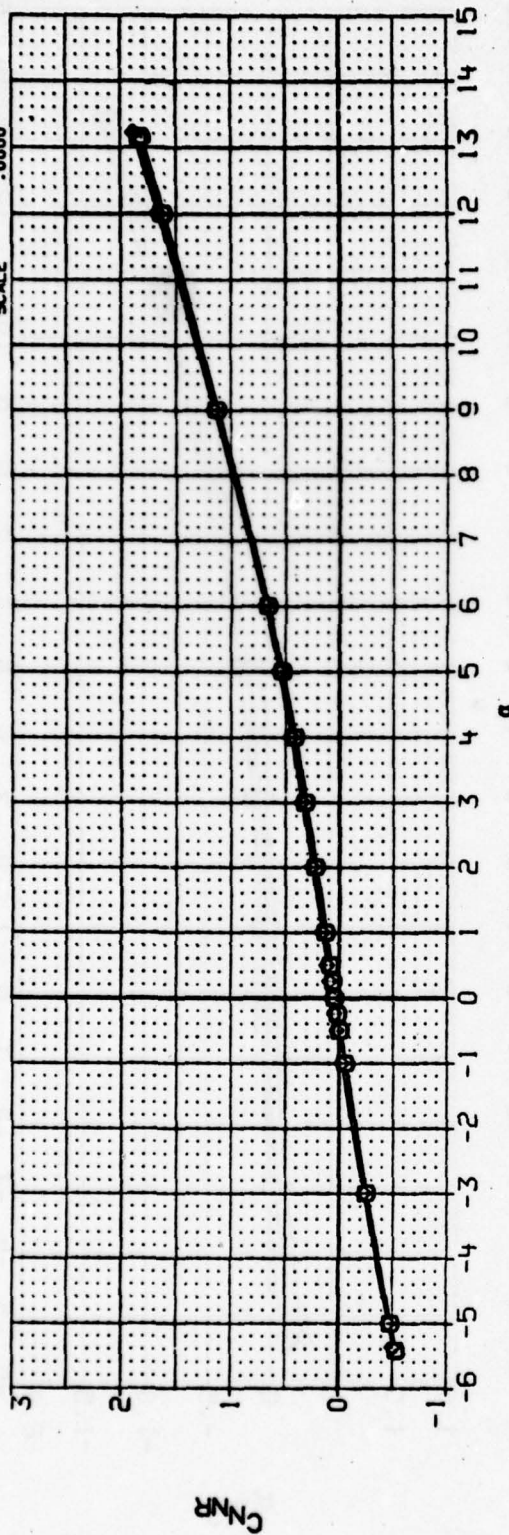
(DXH038) AEDC V1A-CIA, CANARD CONTROL, BNICITI
 (DXH039) AEDC V1A-CIA, CANARD CONTROL, BNICITI
 (DXH040) AEDC V1A-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

.000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=45 PHICND=0
 (B) MACH = 4.52

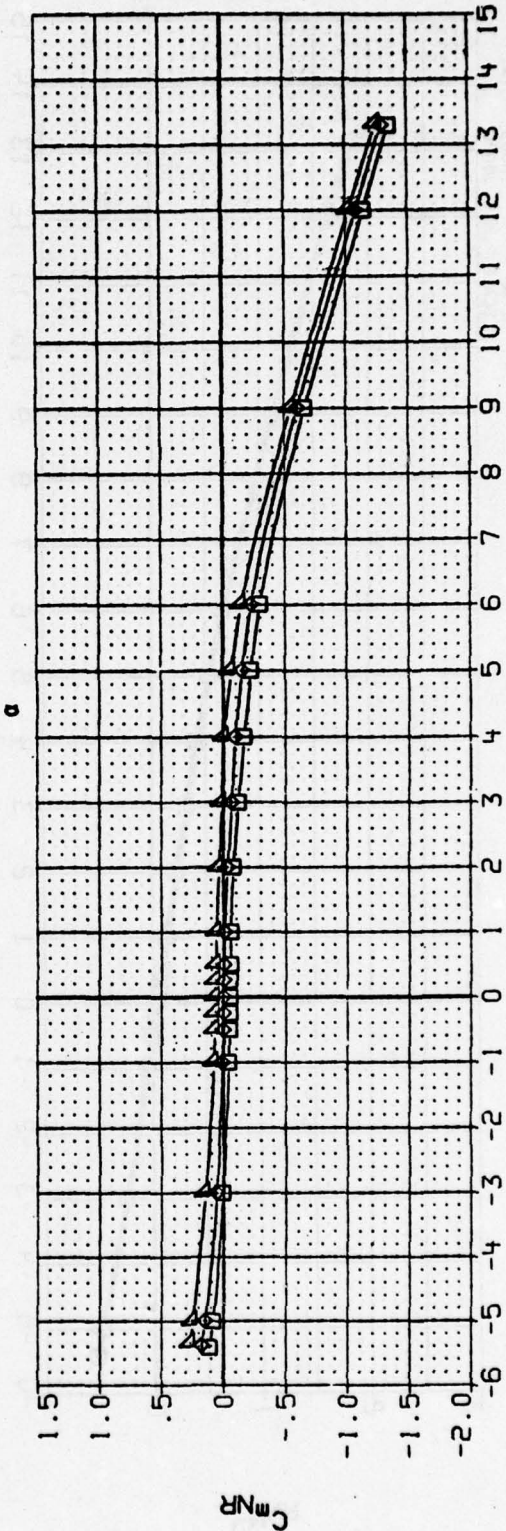
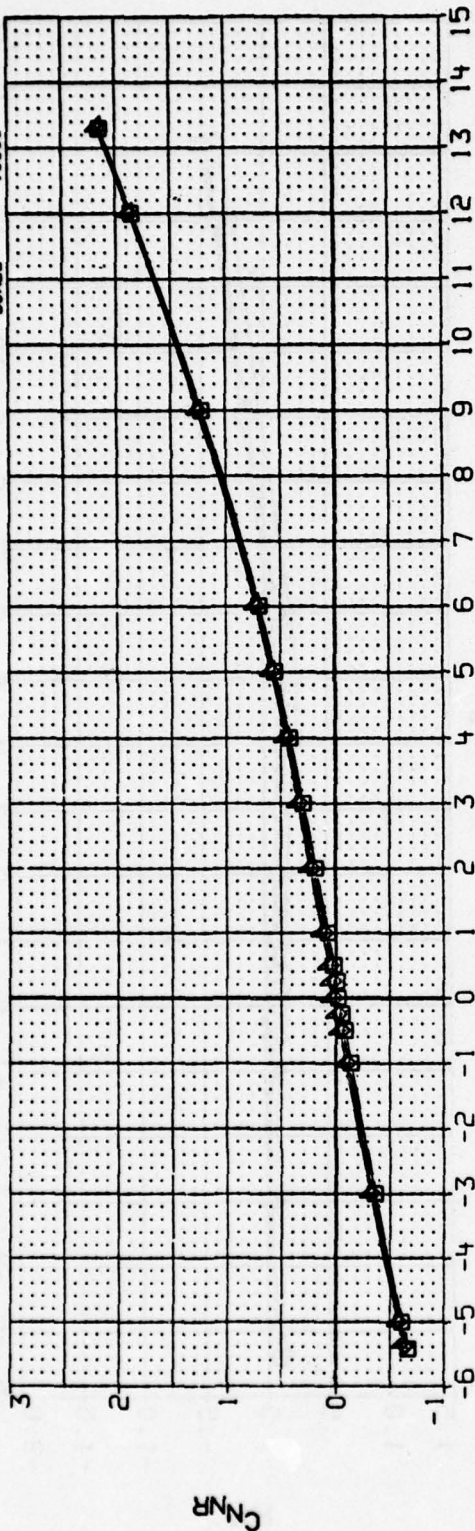
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH041)
(DXH042)
(DXH043)
(DXH044)
(DXH045)

DATA NOT AVAILABLE
AEDC W1A-CIA, CANARD CONTROL, BNICITI
AEDC W1A-CIA, CANARD CONTROL, BNICITI
DATA NOT AVAILABLE
AEDC W1A-CIA, CANARD CONTROL, BNICITI

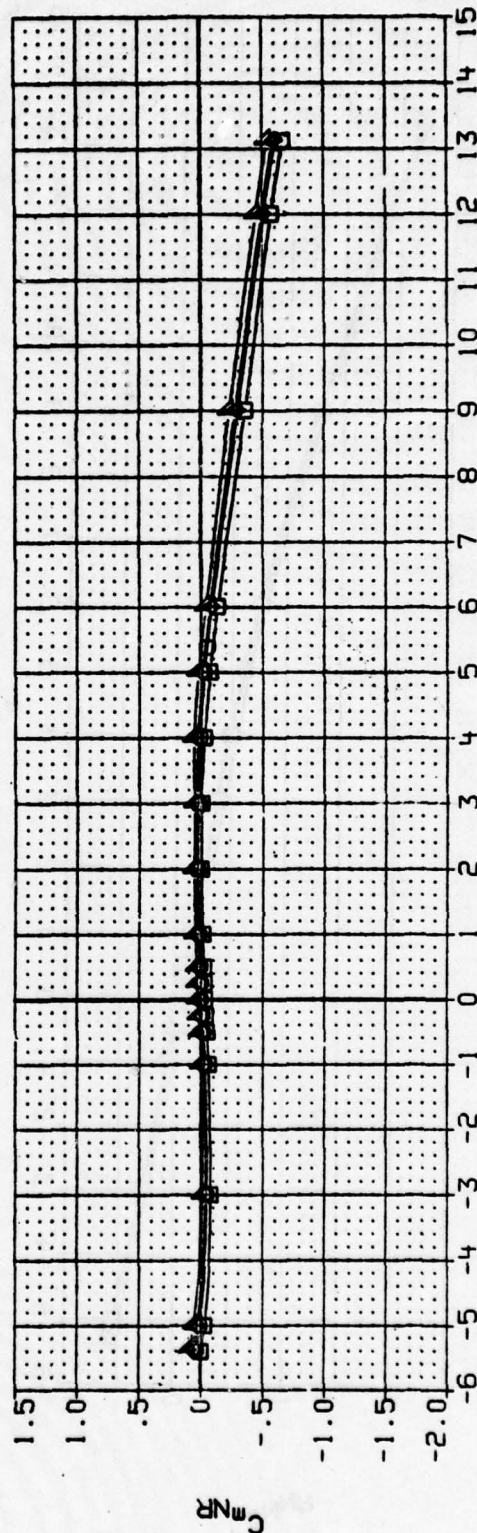
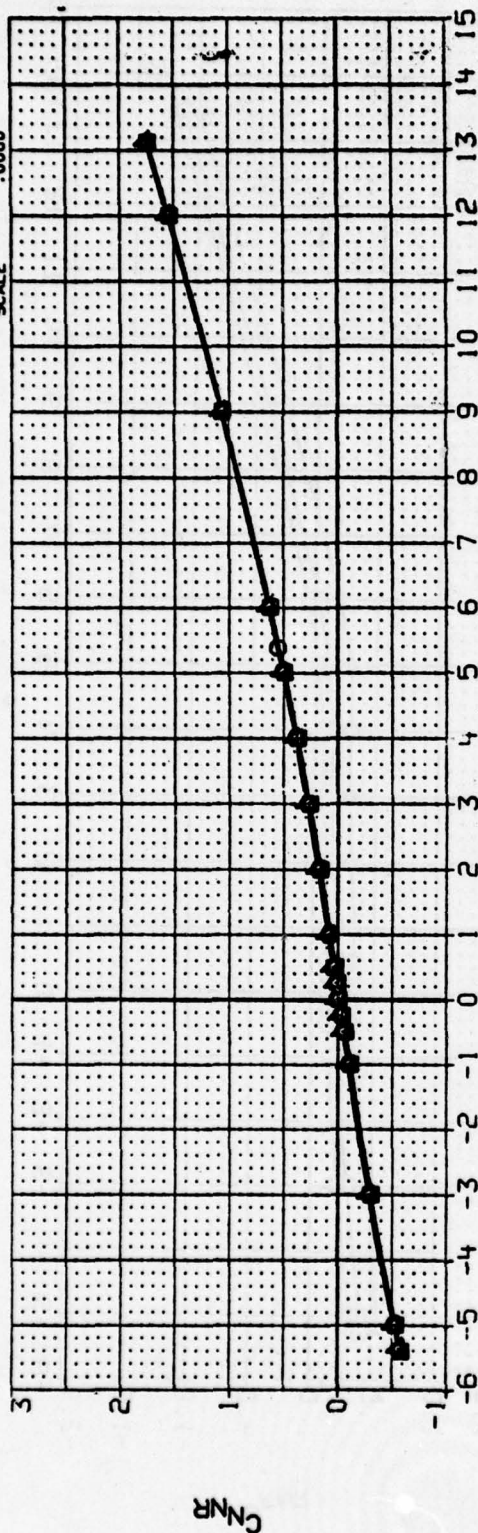
DCND1 DCND2 DCND3 DCND4
5.000 -5.000 -5.000 5.000
-3.000 -3.000 -3.000 -3.000
1.000 1.000 1.000 1.000
3.000 3.000 3.000 3.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XTRP 26.0000 IN.
YTRP .0000 IN.
ZTRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHICAL=45 PHICND=45
(A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH01)	□	AEDC V1A-CIA, CANARD CONTROL, BNICITI	5.000	-5.000	-5.000	5.000	SREF 19.6350 IN.
(DXH02)	□	AEDC V1A-CIA, CANARD CONTROL, BNICITI	-3.000	-3.000	-3.000	-3.000	LREF 5.0000 IN.
(DXH03)	△	AEDC V1A-CIA, CANARD CONTROL, BNICITI	1.000	1.000	1.000	1.000	BREF 5.0000 IN.
(DXH04)	△	AEDC V1A-CIA, CANARD CONTROL, BNICITI	3.000	3.000	3.000	3.000	XREF 26.0000 IN.
(DXH05)	△	AEDC V1A-CIA, CANARD CONTROL, BNICITI					YREF .0000 IN.
							ZREF .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHICAL=45 PHICND=45

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH046) ○ AEDC V41A-C1A, CANARD CONTROL, BNIC111

(DXH047) ◇ AEDC V41A-C1A, CANARD CONTROL, BNIC111

(DXH048) □ AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4

5.000 5.000 5.000 5.000

9.000 9.000 9.000 9.000

15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

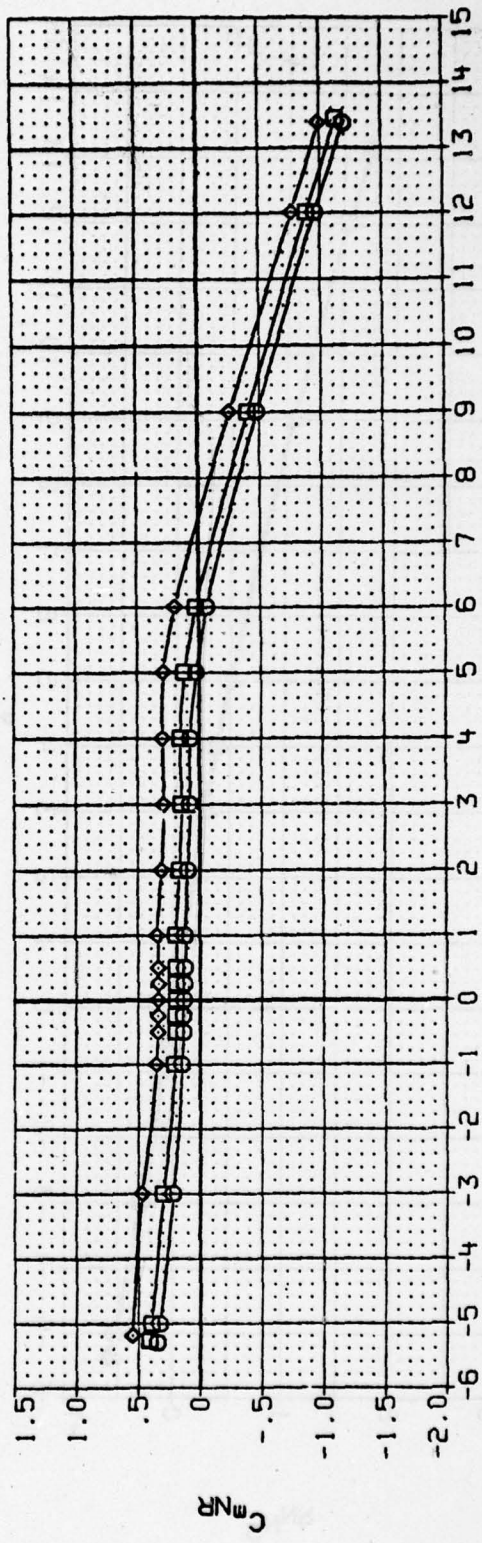
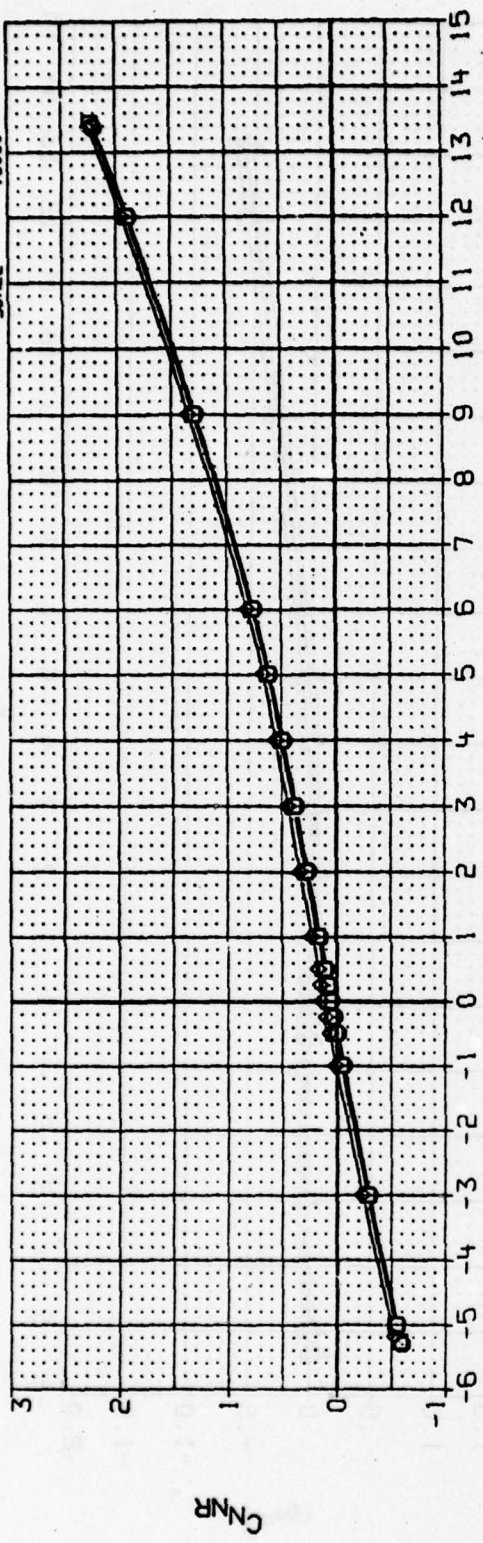
BREF 5.0000 IN.

XPRP 26.0000 IN.

YPRP .0000 IN.

ZPRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PH1AL=45 PH1CND=45

(A) MACH = 3.00

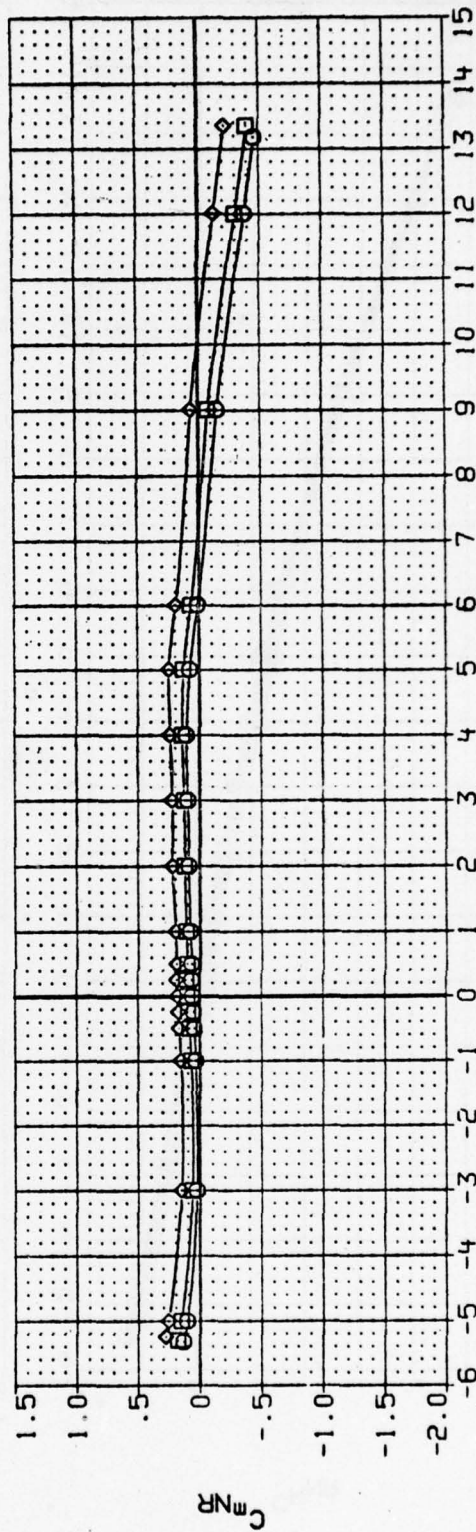
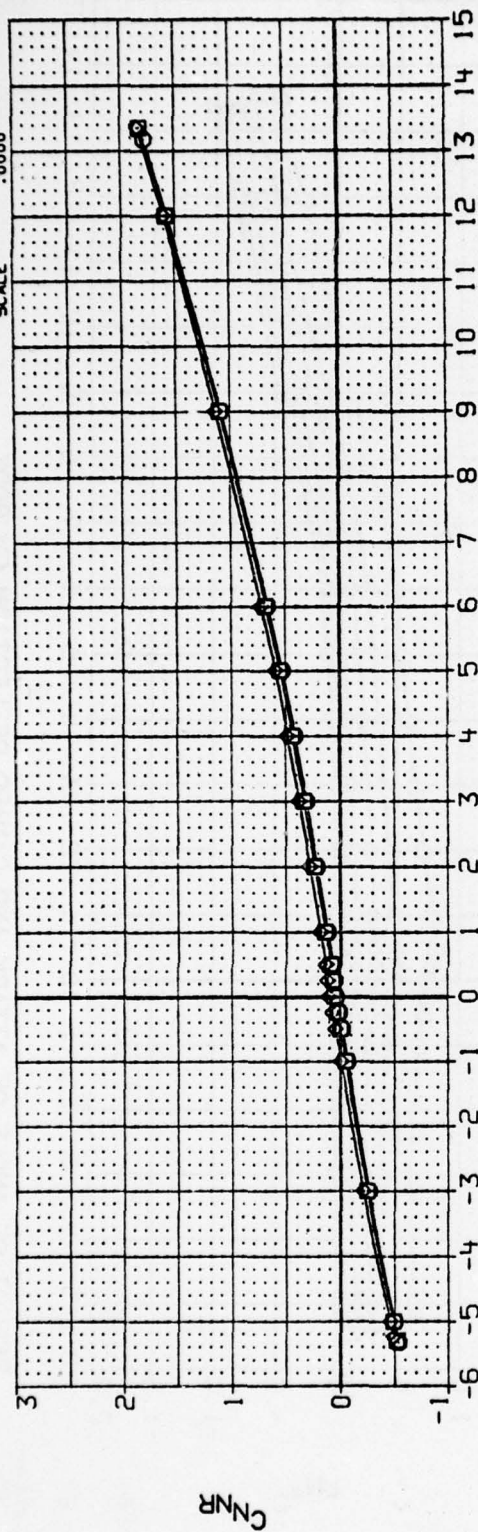
DATA SET SYMBOL
(DXH046)
(DXH047)
(DXH048)

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL
AEDC V41A-C1A, CANARD CONTROL
AEDC V41A-C1A, CANARD CONTROL

BNIC111
BNIC111
BNIC111

DCND1 DCND2 DCND3 DCND4
6.000 6.000 6.000 6.000
9.000 9.000 9.000 9.000
15.000 15.000 15.000 15.000

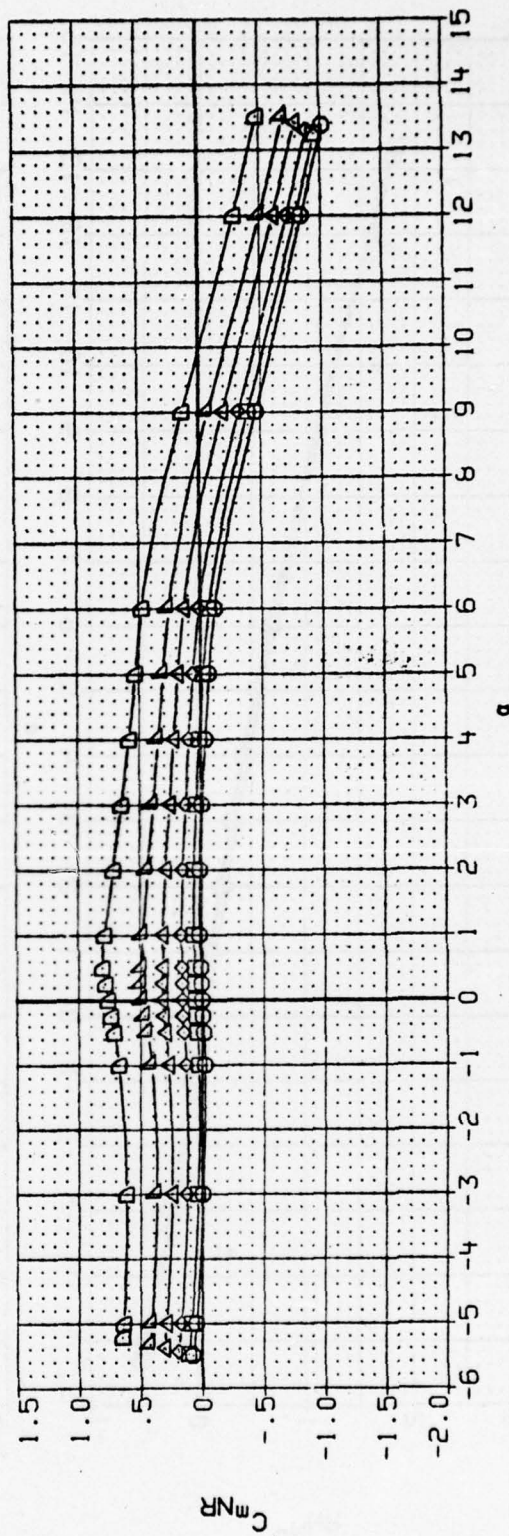
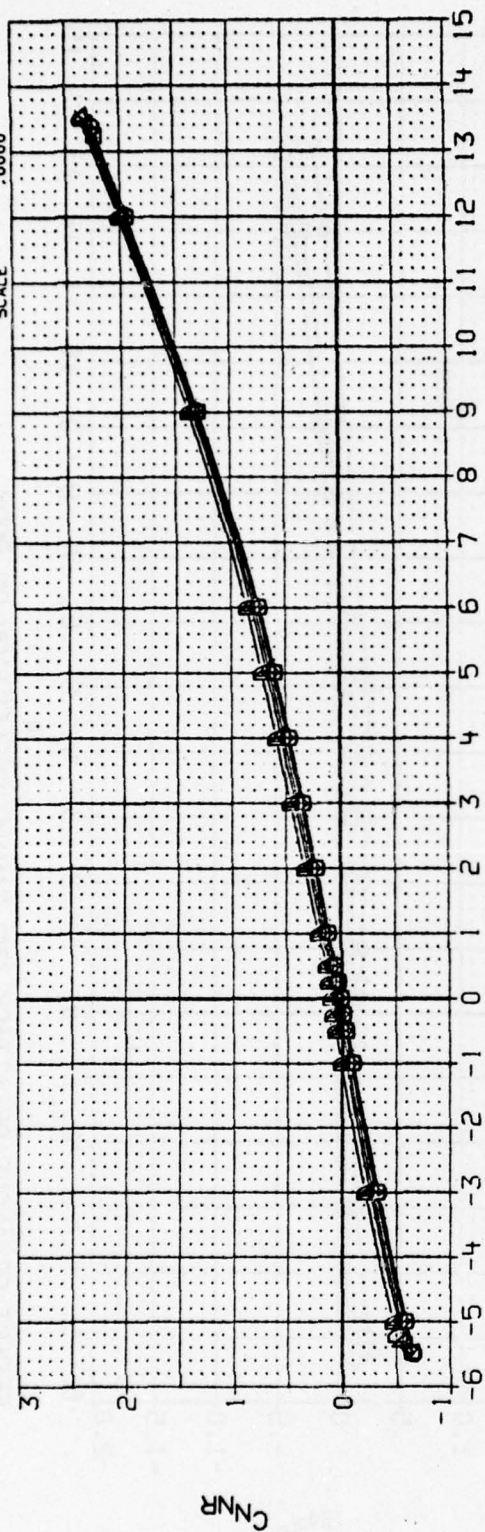
REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHICAL=45 PHICND=45
(B)MACH = 4.52

CONFIGURATION DESCRIPTION:

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION	\$0. IN.
.000	.000	.000	.000	SREF	19.6350
.000	.000	.000	1.000	LREF	5.0000
.000	1.000	.000	.000	BREF	5.0000
.000	3.000	.000	3.000	XREF	26.0000
.000	6.000	.000	6.000	YREF	.0000
.000	9.000	.000	9.000	ZREF	.0000
.000	15.000	.000	15.000	SCALE	.0000

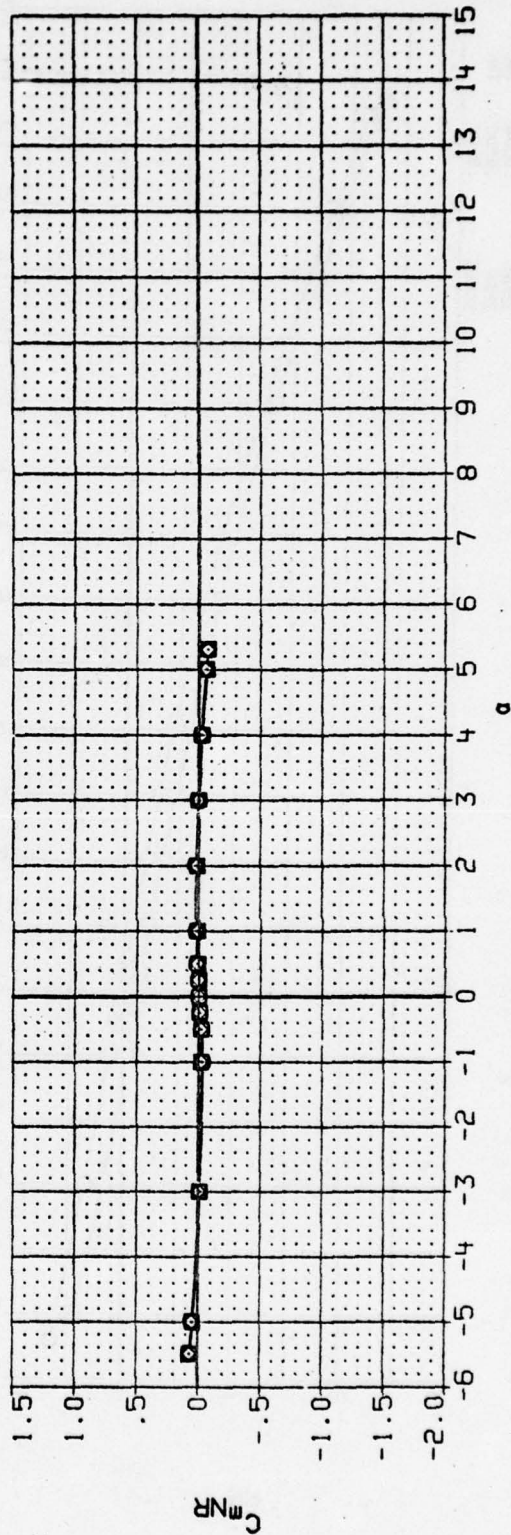
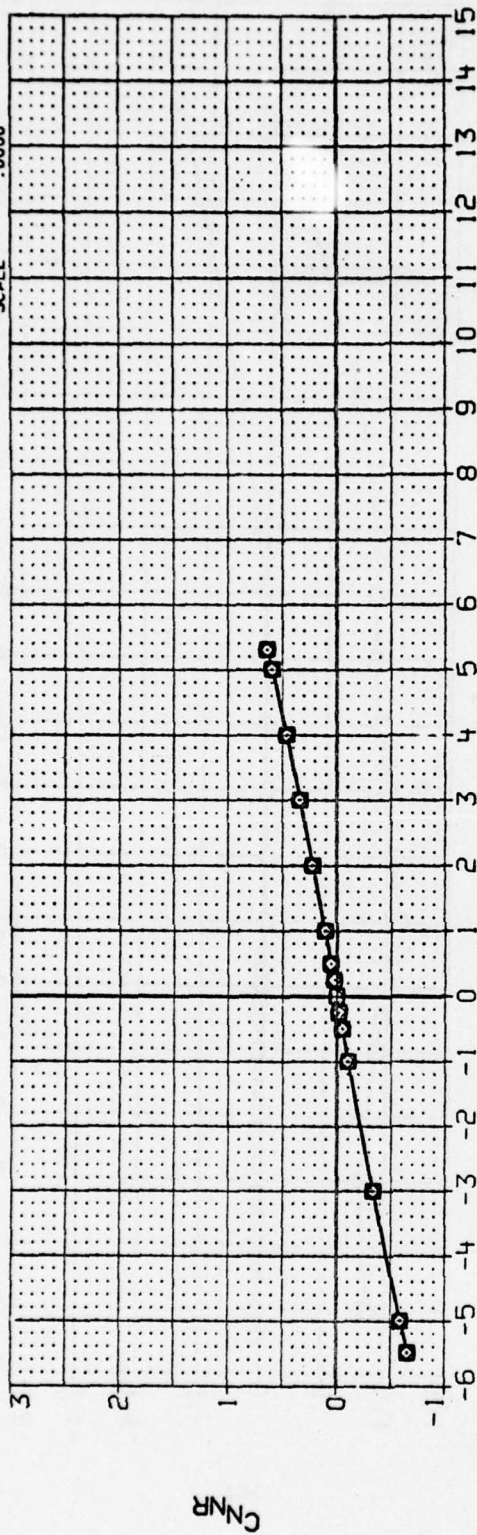


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHITAL=0 PHICND=0

$$(A)MACH = \frac{PHIAL-U}{3.01}$$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH055)	AEDC V1A-C1A, CANARD CONTROL, BNIC3T1	5.000	-5.000	-5.000	5.000	SREF 19.6350 50. IN.
(DXH056)	AEDC V1A-C1A, CANARD CONTROL, BNIC3T1	2.000	-2.000	-2.000	2.000	LREF 5.0000 IN.
(DXH057)	AEDC V1A-C1A, CANARD CONTROL, BNIC3T1	-3.000	3.000	3.000	-3.000	BREF 5.0000 IN.
						XTRP 26.0000 IN.
						YTRP .0000 IN.
						ZTRP .0000 IN.
						SCALE .0000

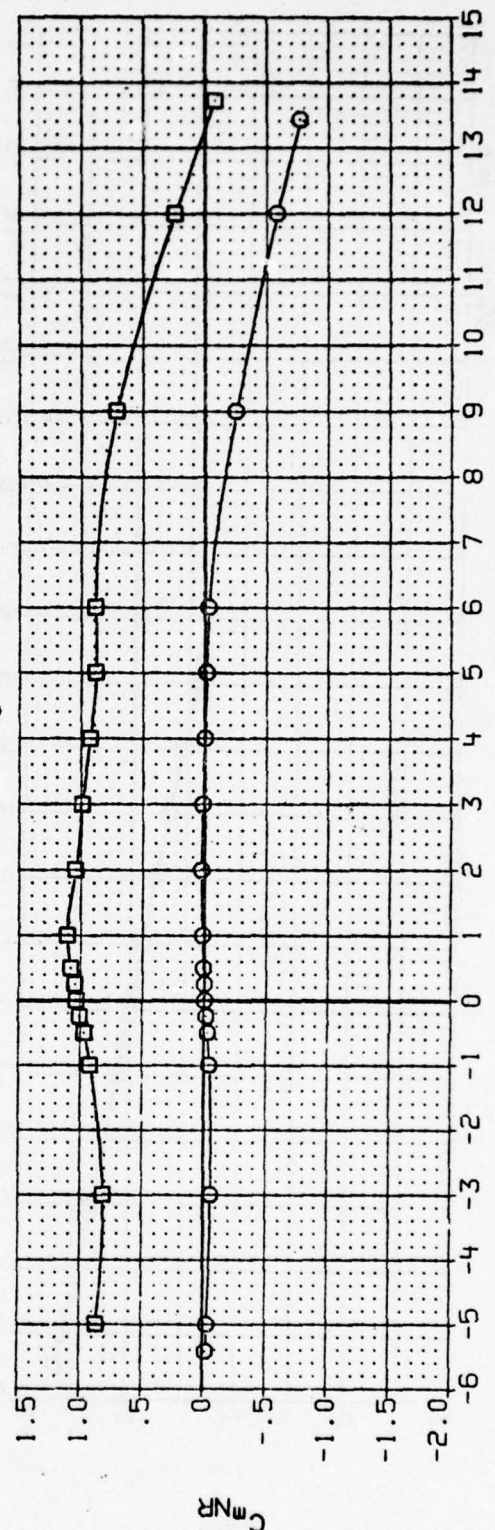
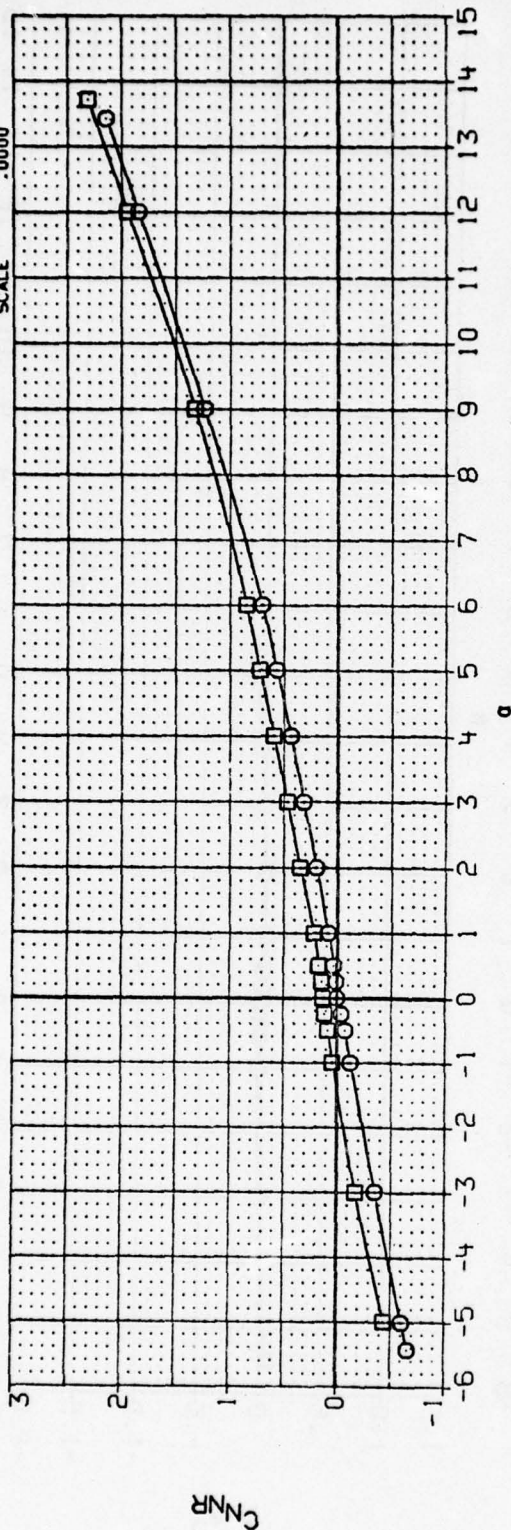


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=0 PHICND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH058) ☐ AEDC W1A-CIA, CANARD CONTROL, BNIC3T1
 (DXH059) ☐ AEDC W1A-CIA, CANARD CONTROL, BNIC3T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 15.000 15.000 15.000 15.000

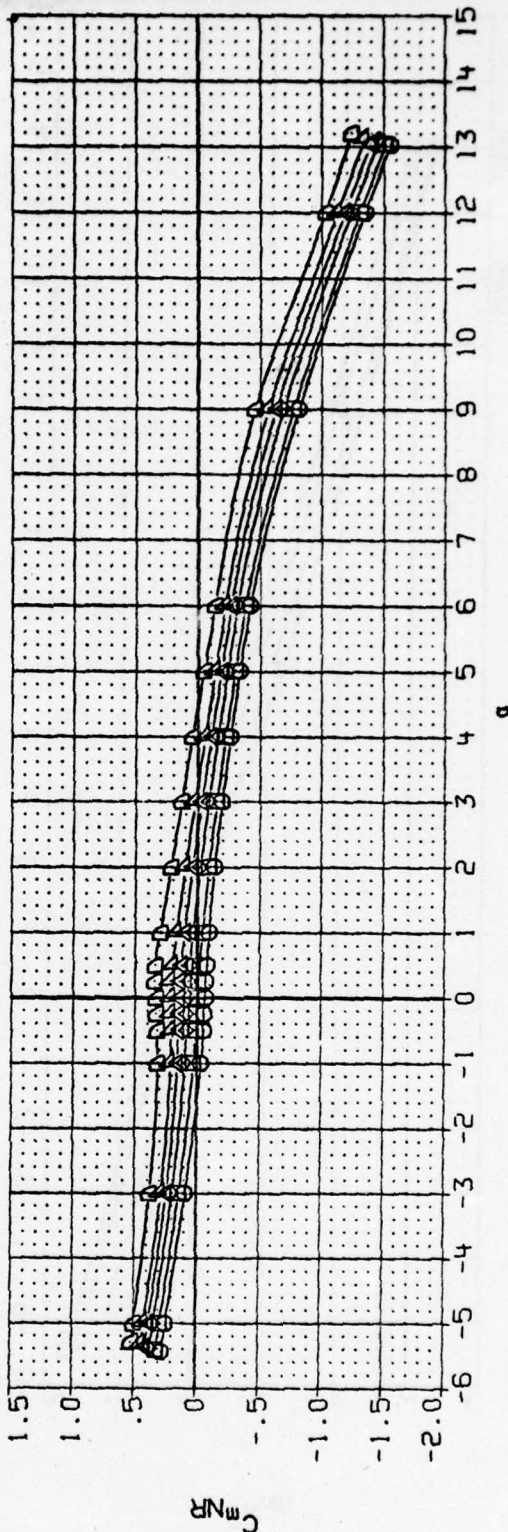
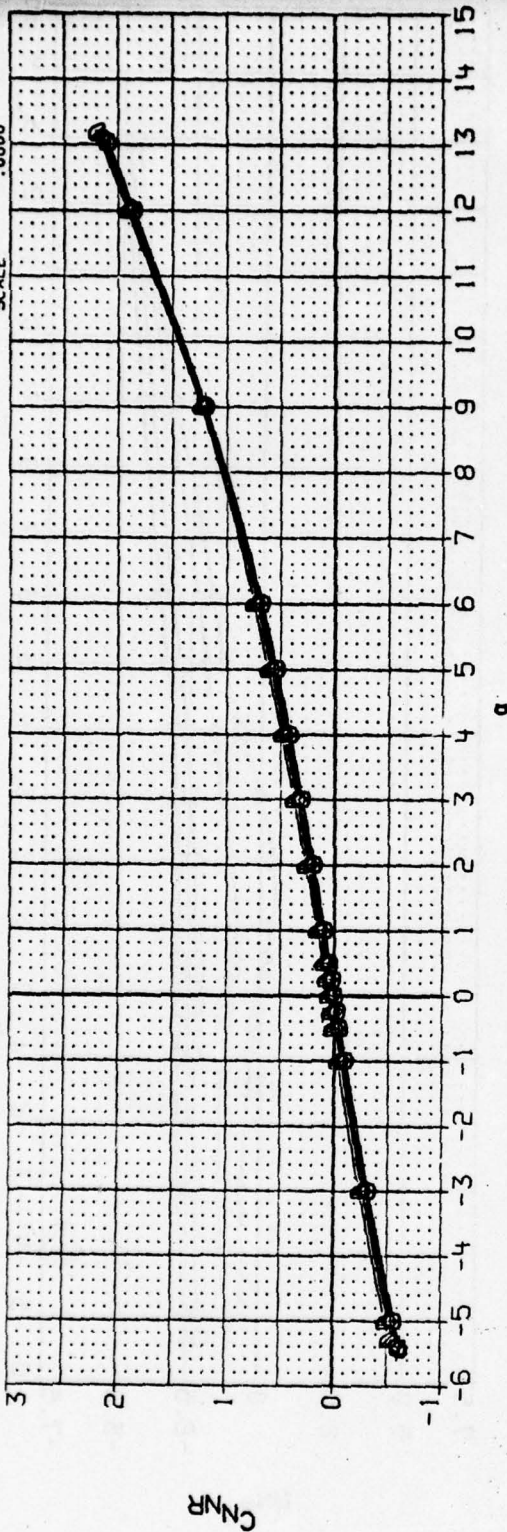
REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=0 PHICND=45

(A) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH061)	□	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	-3.000	.000	-3.000	SREF 19.6350 SQ. IN.
(DXH062)	◇	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH063)	△	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	3.000	.000	3.000	BREF 5.0000 IN.
(DXH064)	▽	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	6.000	.000	6.000	XMRP 26.0000 IN.
(DXH065)	□	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	9.000	.000	9.000	YMRP .0000 IN.
(DXH066)	□	AEDC V41A-C1A, CANARD CONTROL, BN2C111	.000	15.000	.000	15.000	ZMRP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHITAL=0 PHICND=0

(A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH067)
(DXH068)
(DXH069)
(DXH070)

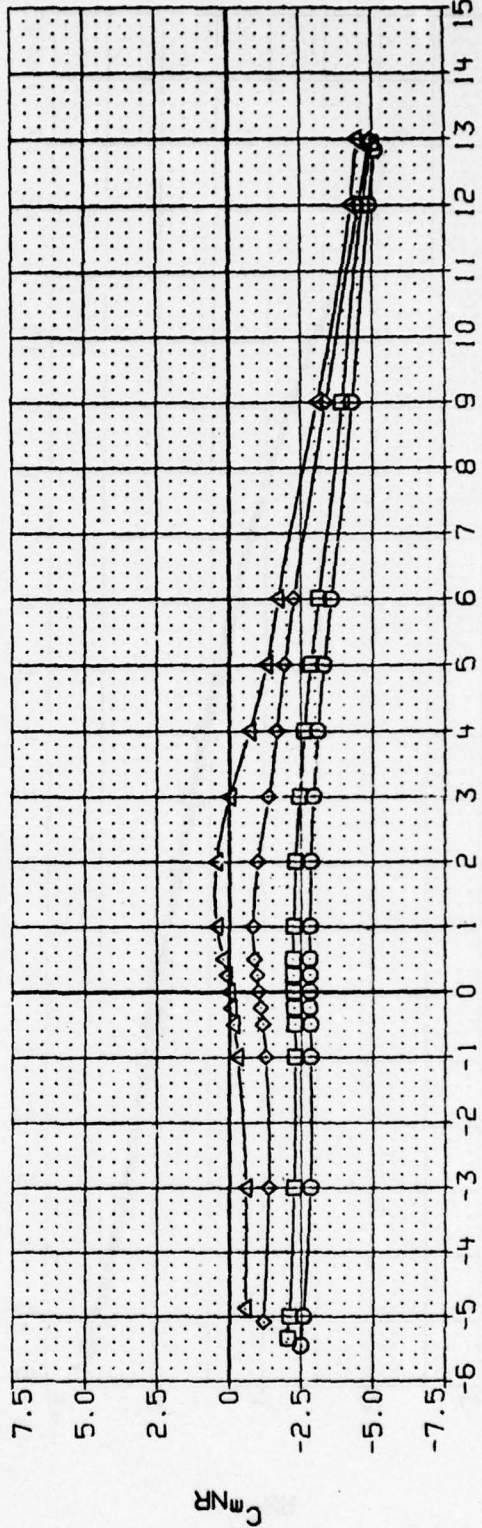
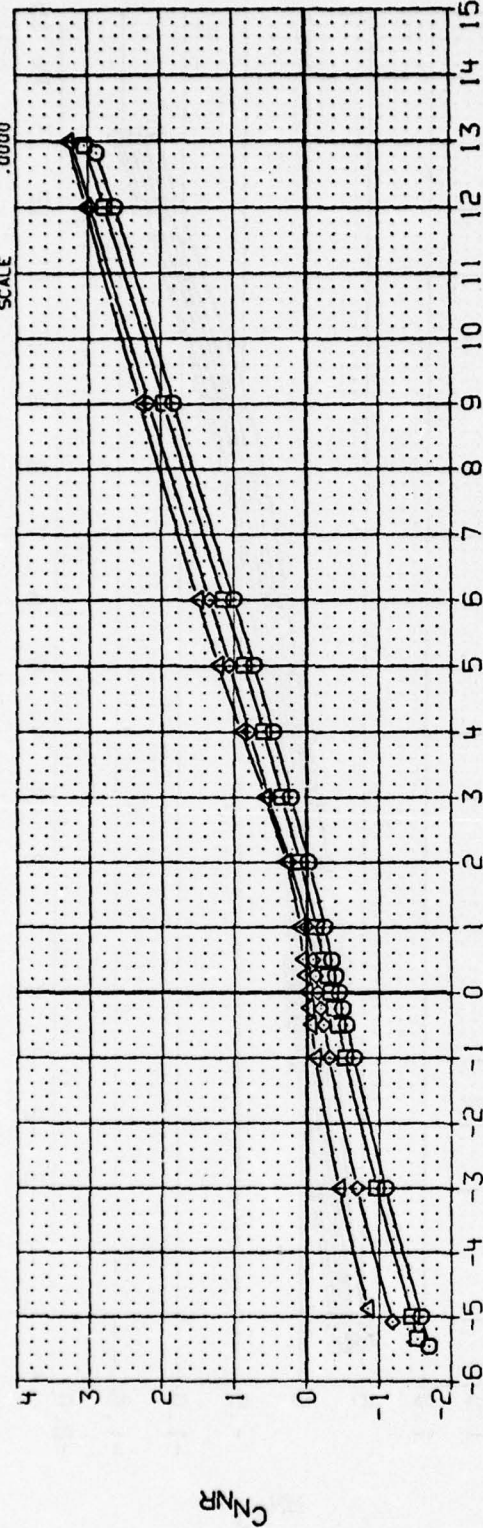
AEDC V1A-CIA, CANARD CONTROL, BNHC671
AEDC V1A-CIA, CANARD CONTROL, BNHC671
AEDC V1A-CIA, CANARD CONTROL, BNHC671
AEDC V1A-CIA, CANARD CONTROL, BNHC671

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000

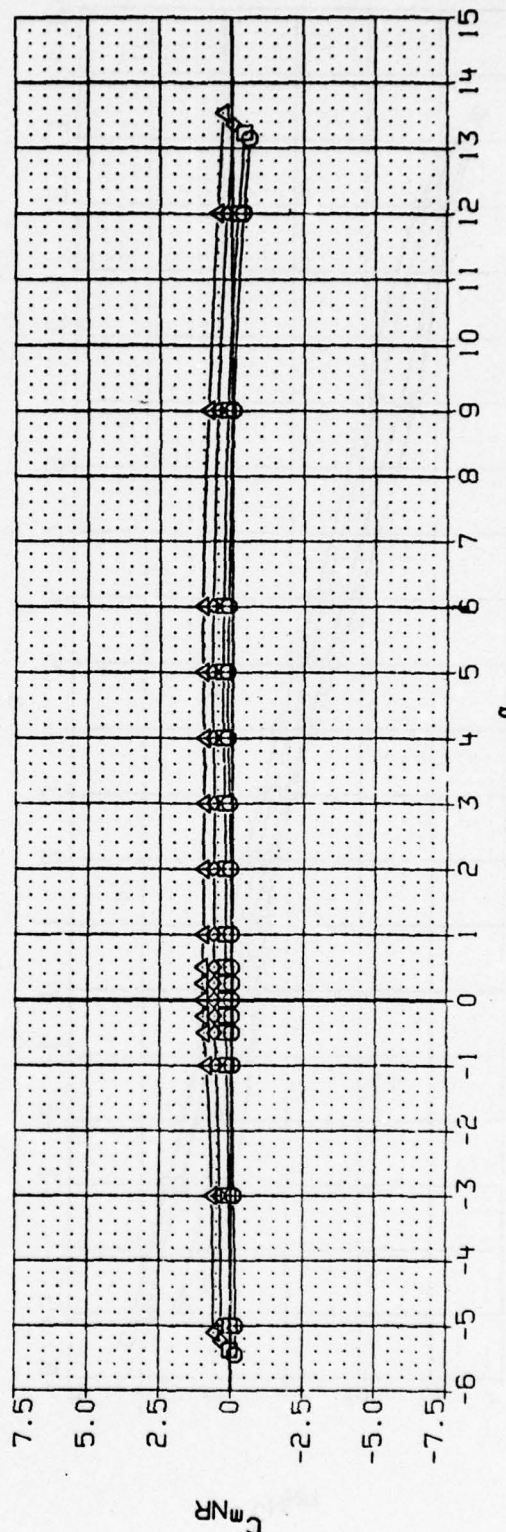
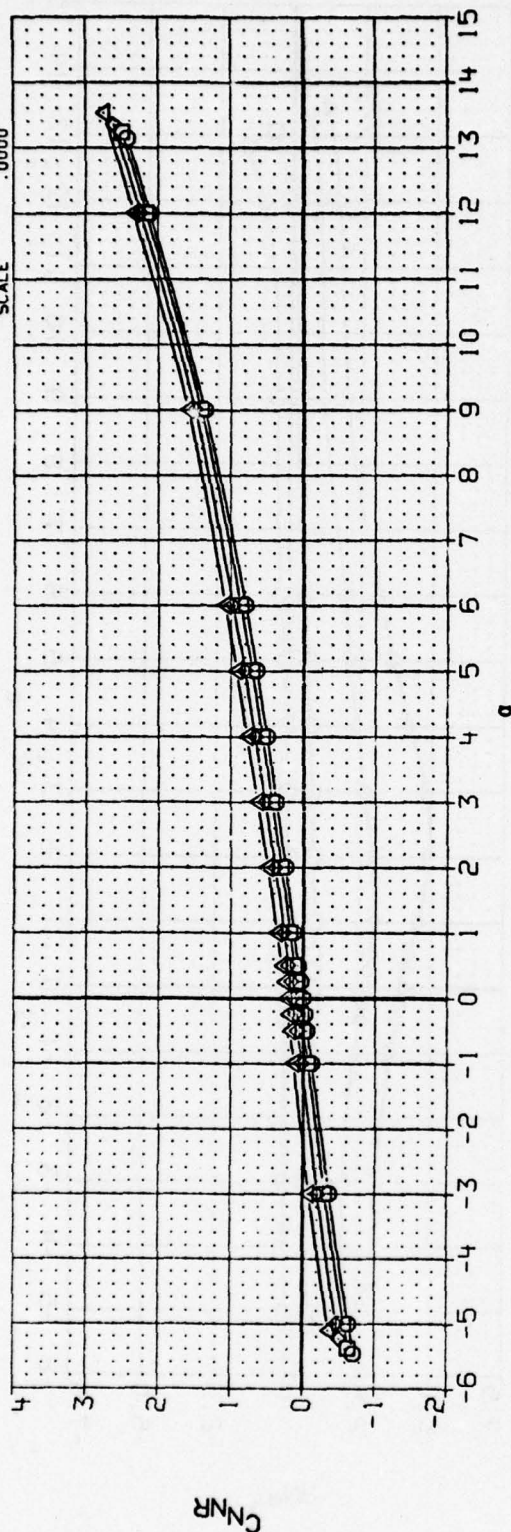
REFERENCE INFORMATION

SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.



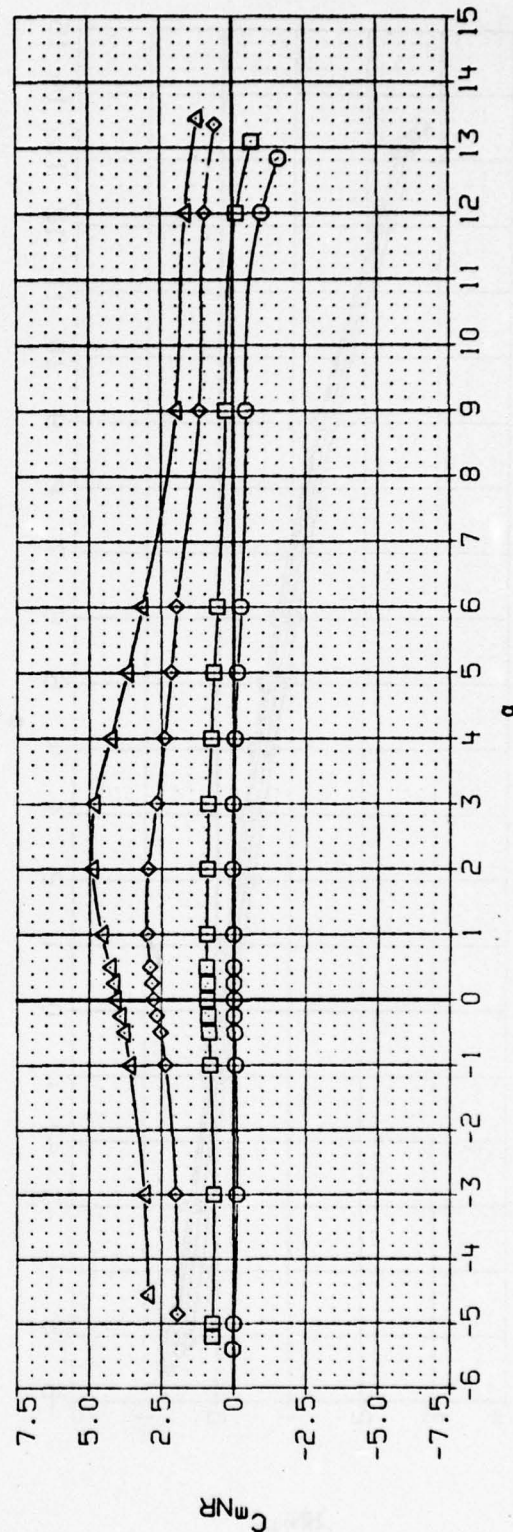
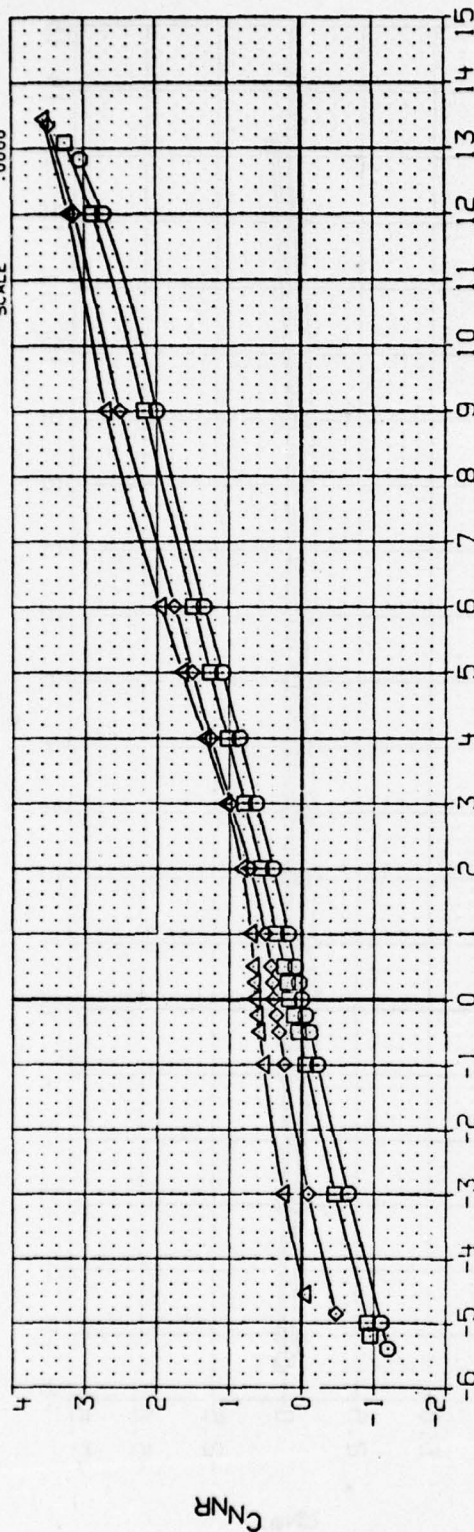
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHICAL=0 PHICND=0
(A) MACH = 1.51

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH057)	□	AEDC V41A-C1A, CANARD CONTROL, BN4CST1	.000	.000	.000	.000	SREF 19.6350 SQ. IN.
(DXH058)	□	AEDC V41A-C1A, CANARD CONTROL, BN4CST1	.000	3.000	.000	3.000	LREF 5.0000 IN.
(DXH059)	△	AEDC V41A-C1A, CANARD CONTROL, BN4CST1	.000	9.000	.000	9.000	BREF 5.0000 IN.
(DXH070)	△	AEDC V41A-C1A, CANARD CONTROL, BN4CST1	.000	15.000	.000	15.000	XMRP 26.0000 IN.
							YMRP .0000 IN.
							ZMRP .0000 IN.
							SCALE .0000



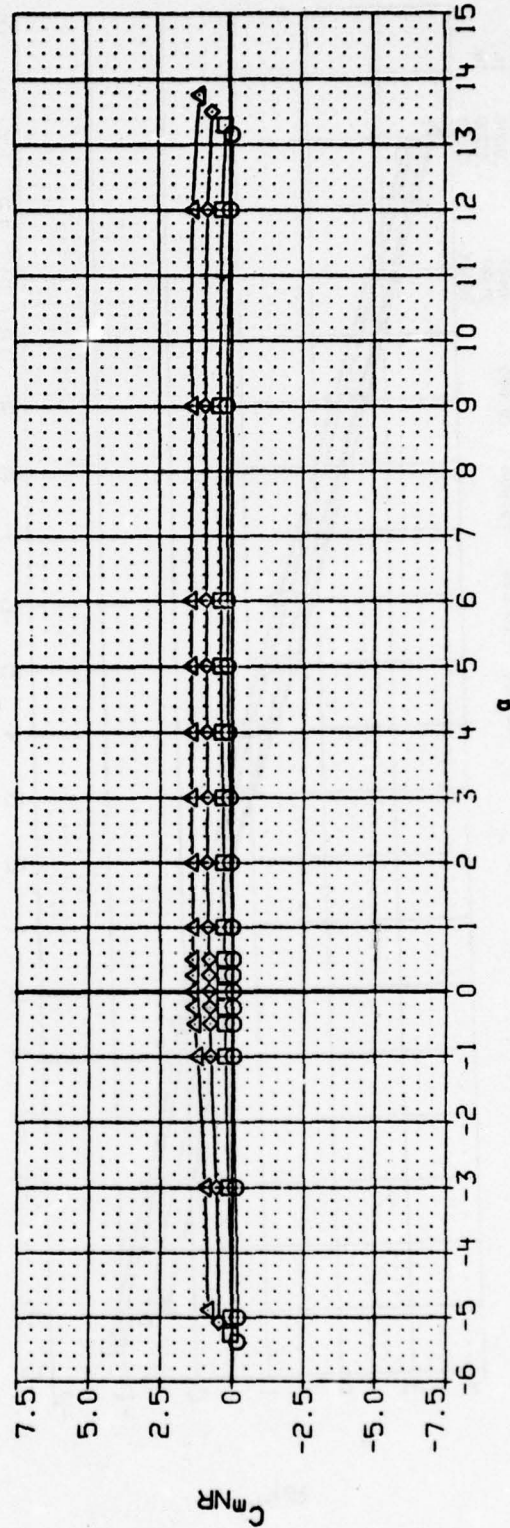
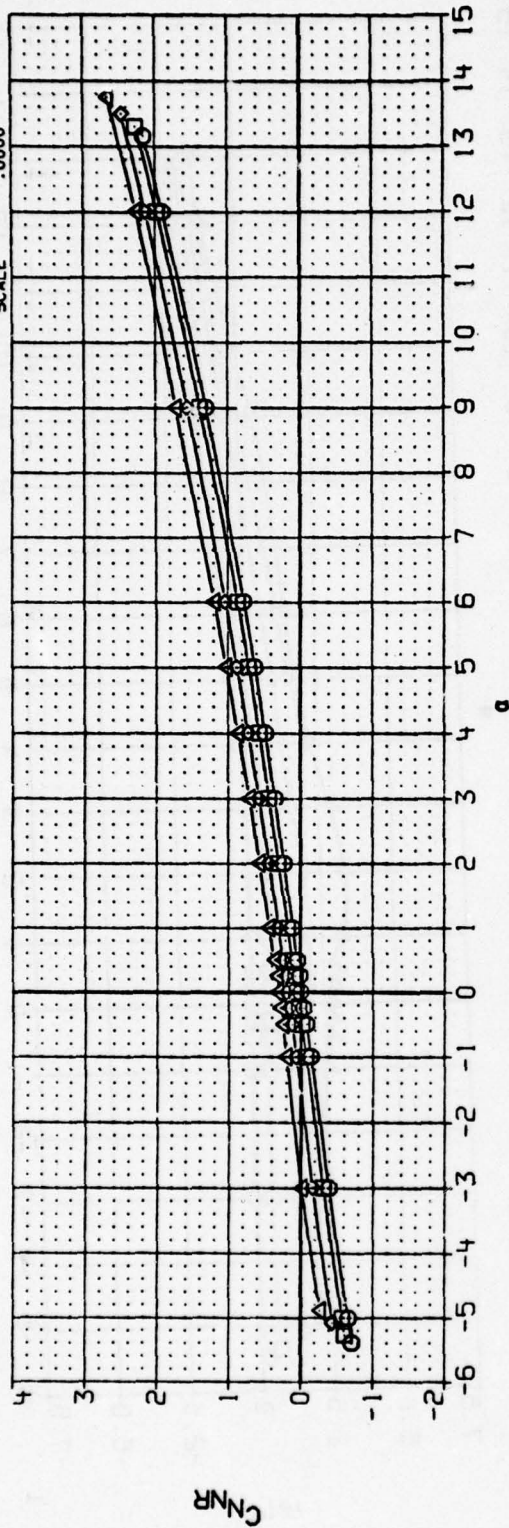
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=0 PHICND=0
 (B) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH072)	○	AEDC V41A-CIA, CANARD CONTROL, BN4CST1	.000	.000	.000	.000	SREF 19.6350 IN.
(DXH073)	□	AEDC V41A-CIA, CANARD CONTROL, BN4CST1	3.000	3.000	3.000	3.000	LREF 5.0000 IN.
(DXH074)	◇	AEDC V41A-CIA, CANARD CONTROL, BN4CST1	9.000	9.000	9.000	9.000	BREF 5.0000 IN.
(DXH075)	△	AEDC V41A-CIA, CANARD CONTROL, BN4CST1	15.000	15.000	15.000	15.000	XMRP 26.0000 IN.
							YMRP .0000 IN.
							ZMRP .0000 IN.
							SCALE .0000



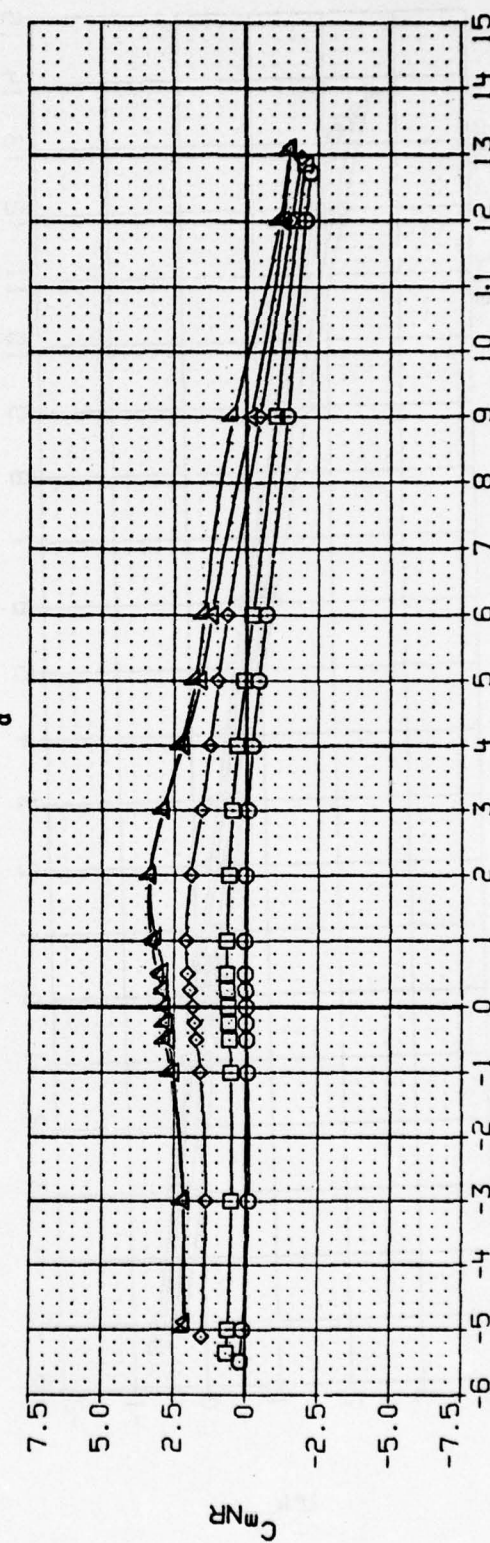
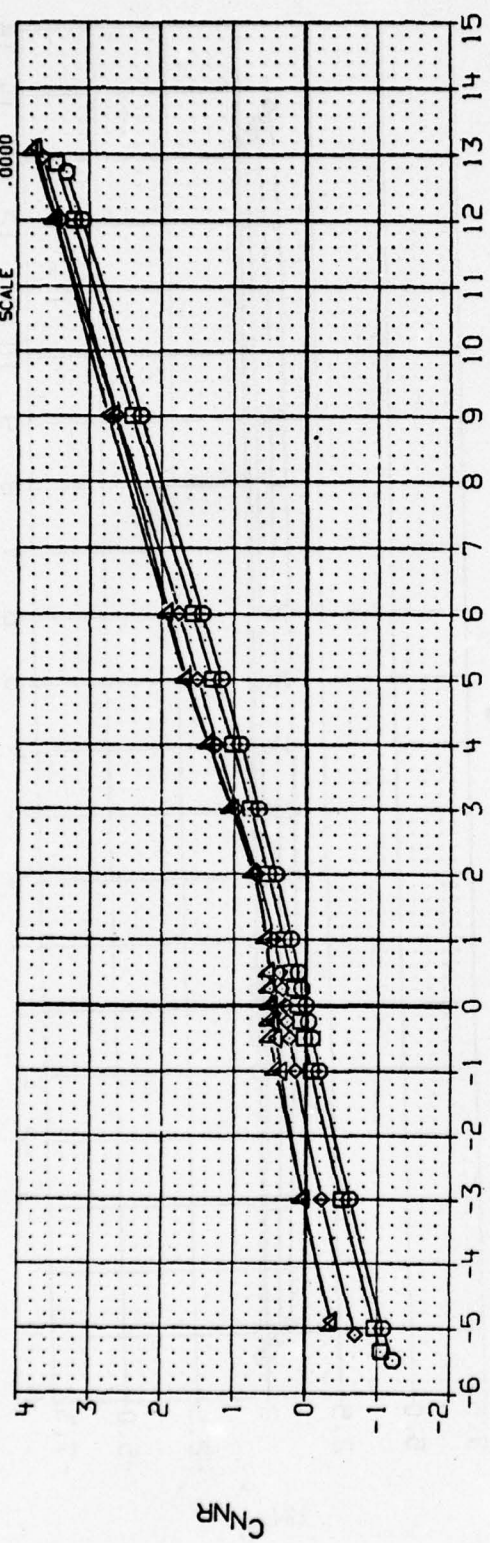
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 $\text{PHITAL} = 0$ $\text{PHICND} = 45$
 (A) MACH = 1.51

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH072)	□	AEDC V41A-C1A, CANARD CONTROL, BNHC6T1	.000	.000	.000	.000	SREF 19.6350 SQ. IN.
(DXH073)	◇	AEDC V41A-C1A, CANARD CONTROL, BNHC6T1	3.000	3.000	3.000	3.000	LREF 5.0000 IN.
(DXH074)	△	AEDC V41A-C1A, CANARD CONTROL, BNHC6T1	9.000	9.000	9.000	9.000	BREF 5.0000 IN.
(DXH075)	△	AEDC V41A-C1A, CANARD CONTROL, BNHC6T1	15.000	15.000	15.000	15.000	XMRP 26.0000 IN.
							YMRP .0000 IN.
							ZMRP .0000 IN.
							SCALE .0000



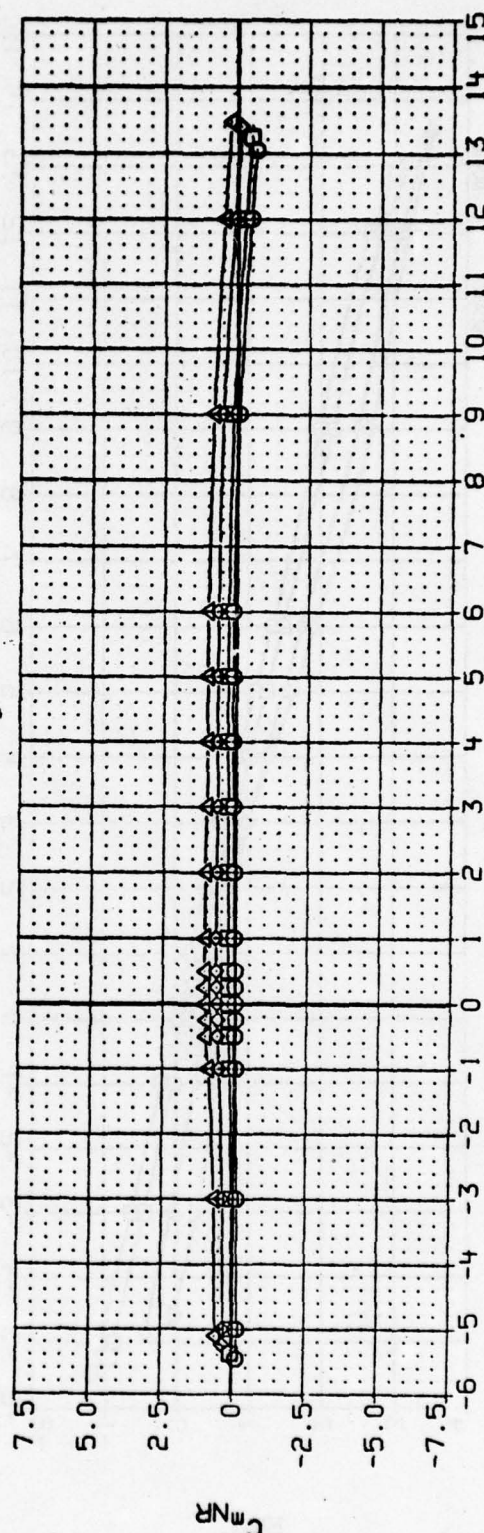
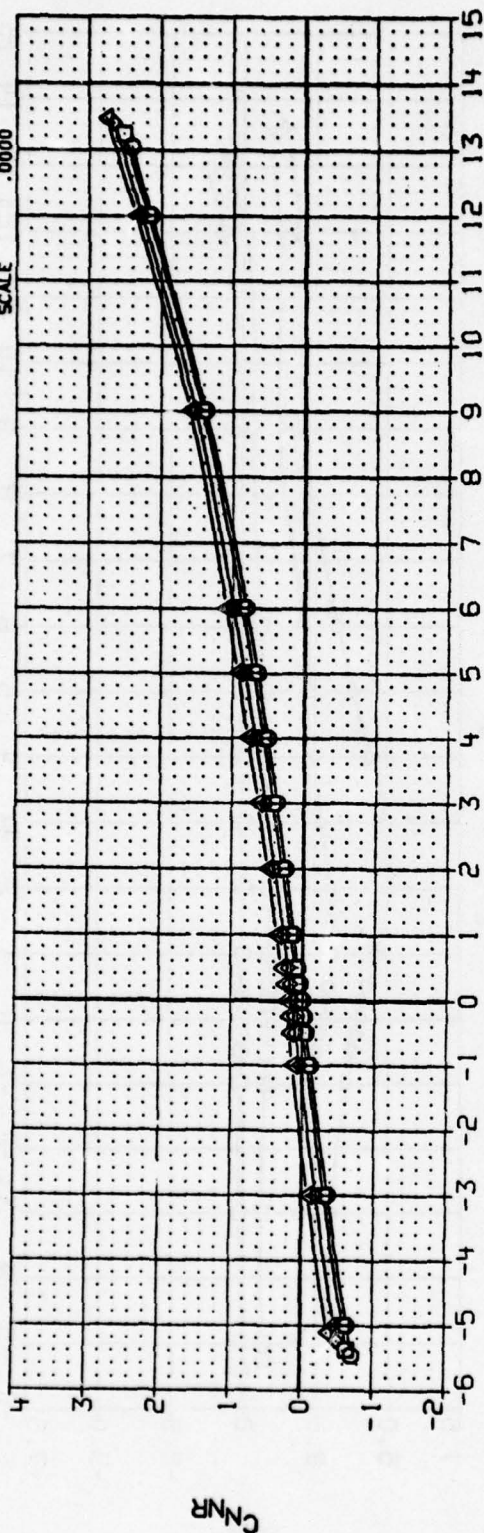
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHIAL=0 PHICND=45
 (B) MACH = 3.01

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION	
(DXH076)	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	SREF	19.6350 IN.
(DXH077)	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	LREF	5.0000 IN.
(DXH078)	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	BREF	5.0000 IN.
(DXH079)	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	XMRP	26.0000 IN.
(DXH080)	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	YMRP	.0000 IN.
		15.000	15.000	15.000	15.000	ZMRP	.0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 $\text{PHITAL}=0$ $\text{PHICND}=0$
 (A) MACH = 1.51

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH076)	□	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	.000	.000	.000	SREF 19.6350 SQ. IN.
(DXH077)	□	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	3.000	.000	3.000	LREF 5.0000 IN.
(DXH078)	□	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	9.000	.000	9.000	BREF 5.0000 IN.
(DXH079)	□	AEDC V41A-C1A, CANARD CONTROL, BN5C6T1	.000	15.000	.000	15.000	YMPP 26.0000 IN.
(DXH080)	□	DATA NOT AVAILABLE	15.000	15.000	15.000	15.000	ZMPP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 $\text{PHICAL}=0$ $\text{PHICND}=0$
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH081) AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH082) AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH083) AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH084) AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

3.000 3.000 3.000 3.000

5.000 9.000 9.000 15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

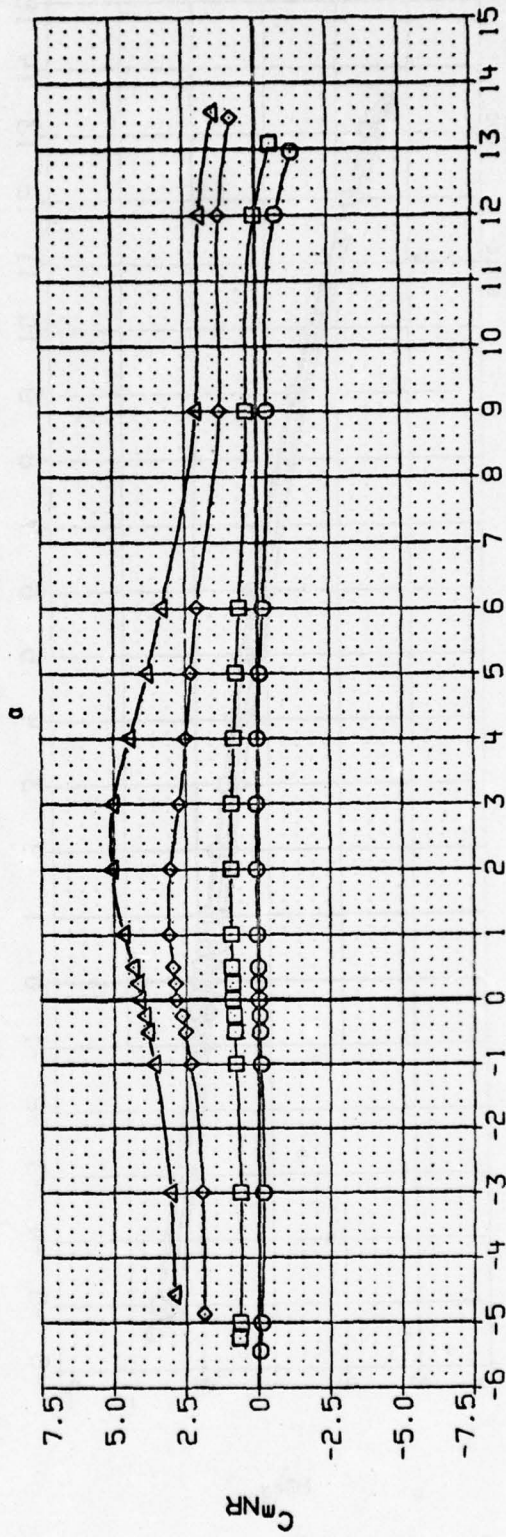
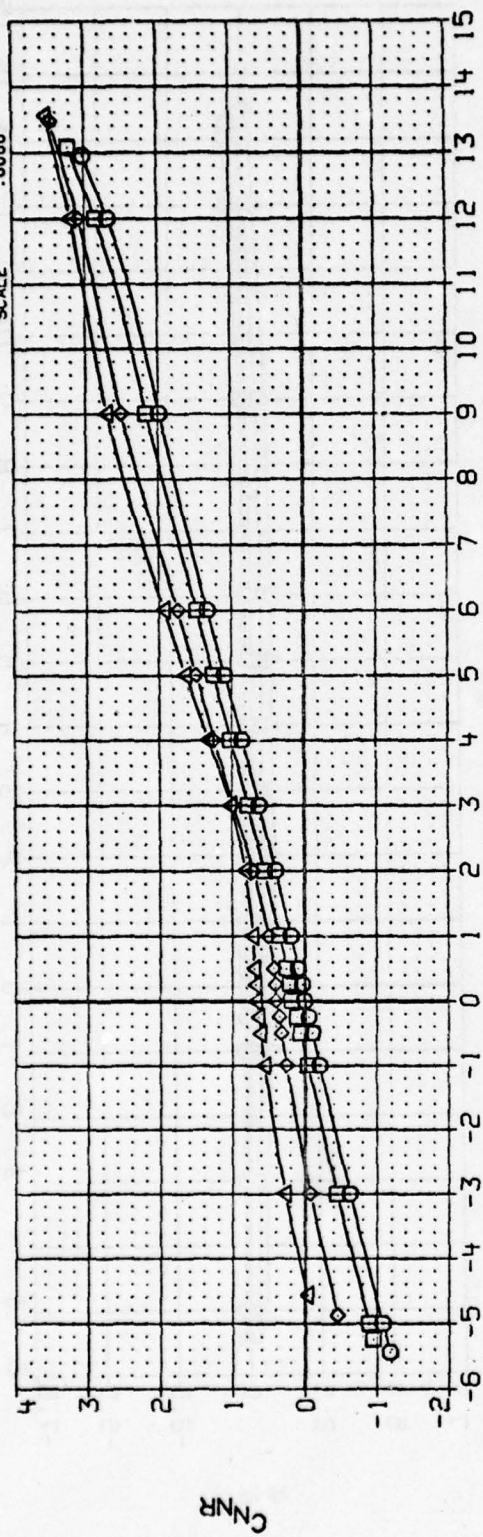
LREF 5.0000 IN.

BREF 5.0000 IN.

YMRP 26.0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHICAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH081) \square AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH082) \square AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH083) \triangle AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

(DXH084) \triangle AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

REFERENCE INFORMATION

SREF 19.6350 SO. IN.

LREF 5.0000 IN.

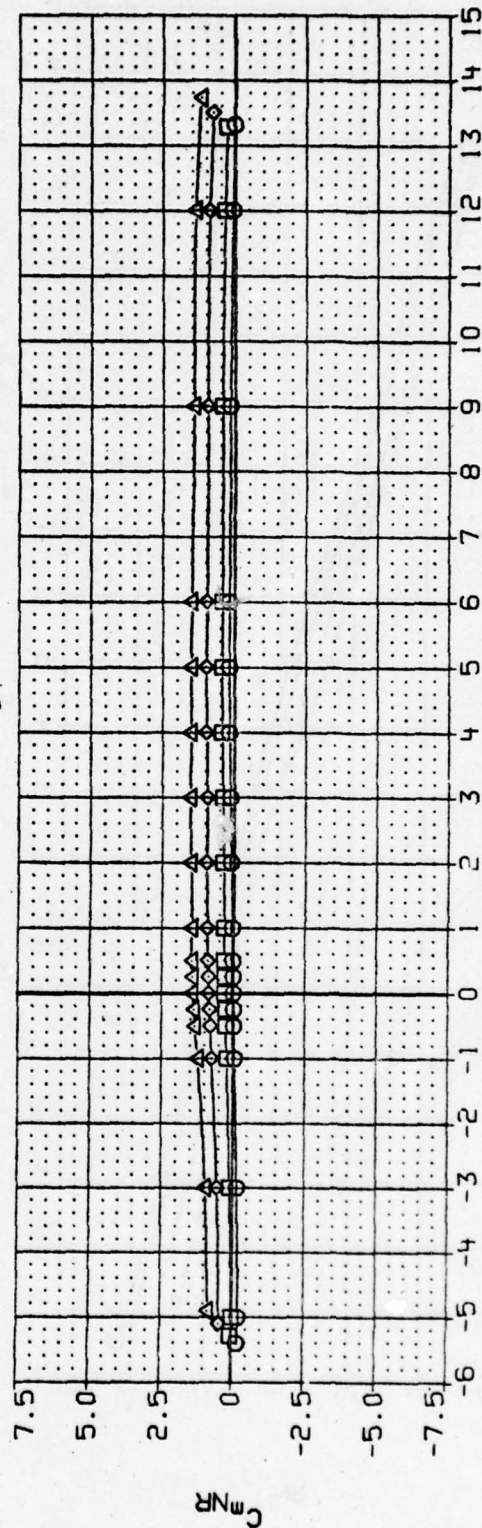
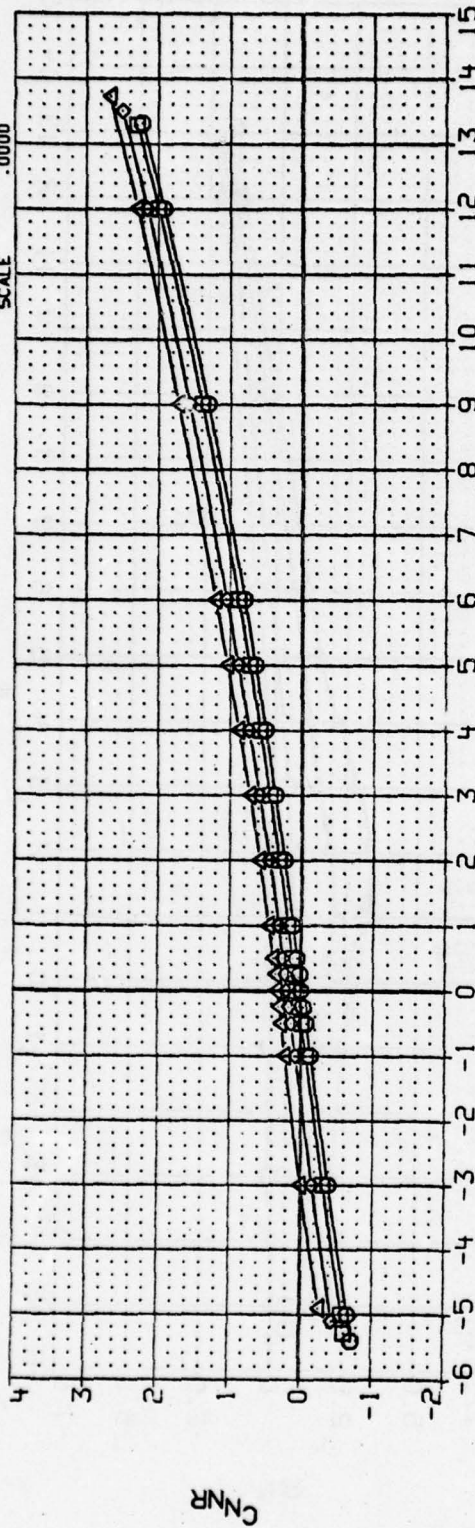
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000

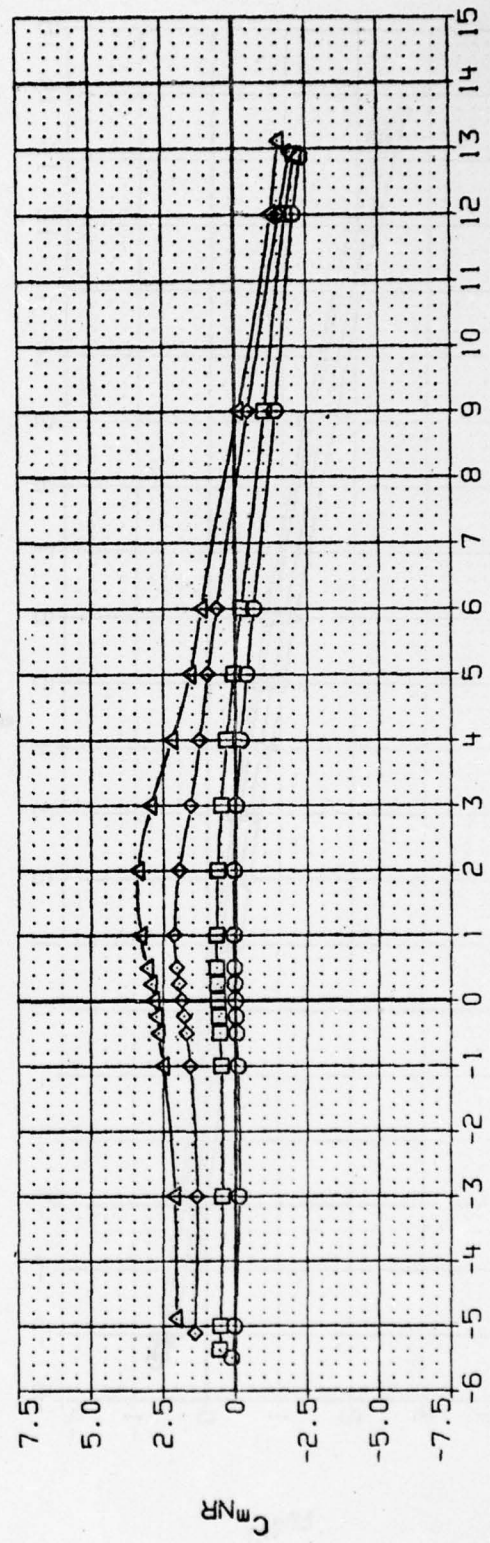
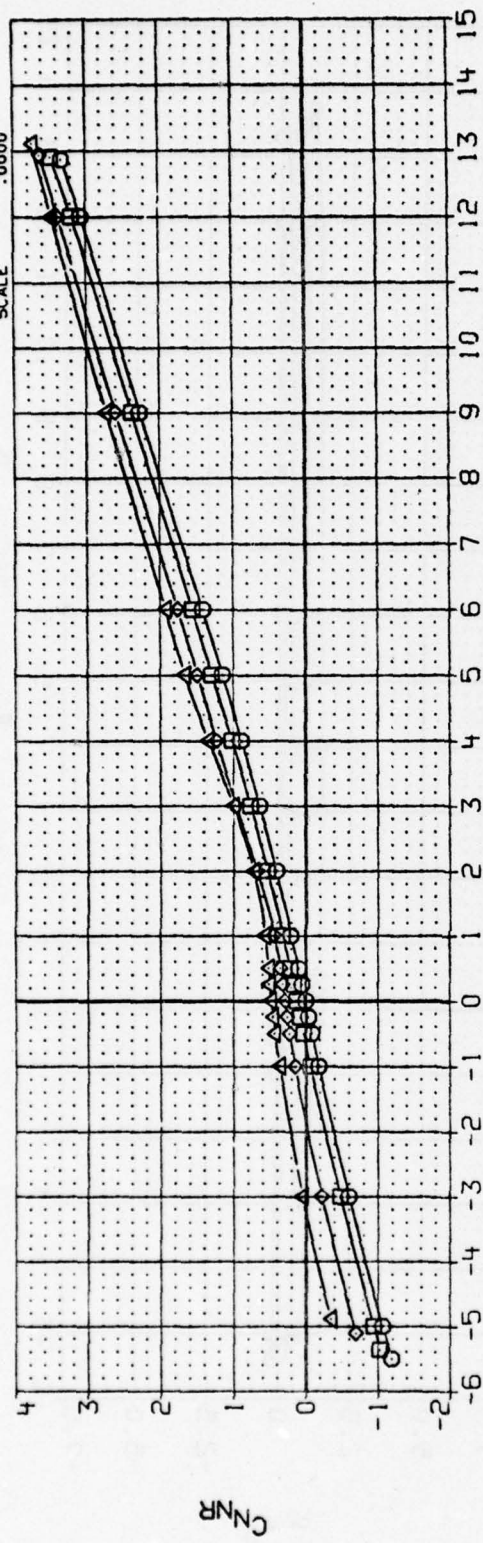


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHITAL=0 PHICND=45

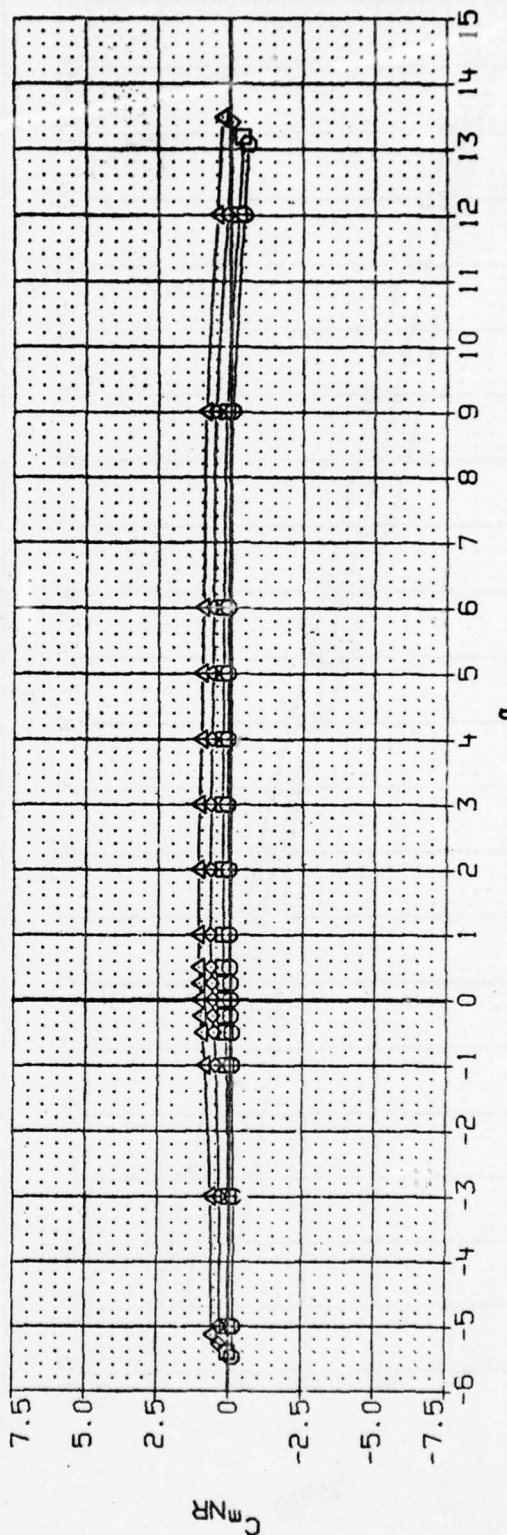
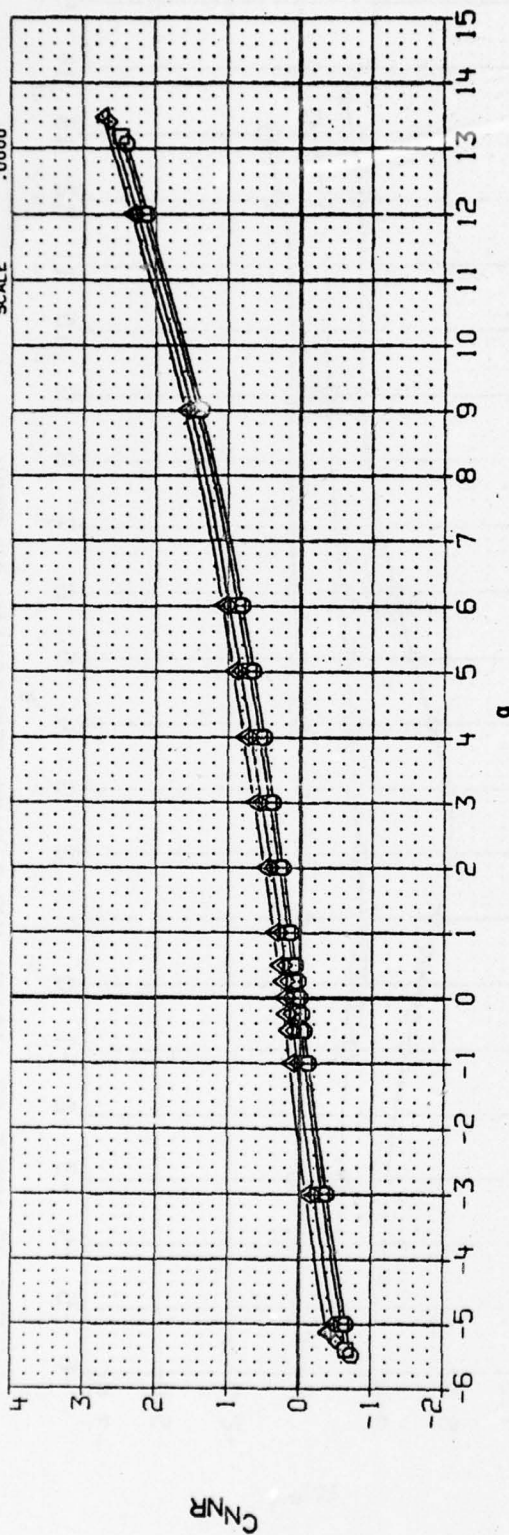
(B) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH085)	□	AEDC V41A-C1A, CANARD CONTROL, BN6C6T1	.000	.000	.000	.000	SREF 19.6350 SO. IN.
(DXH086)	□	AEDC V41A-C1A, CANARD CONTROL, BN6C6T1	.000	3.000	.000	3.000	LREF 5.0000 IN.
(DXH087)	◇	AEDC V41A-C1A, CANARD CONTROL, BN6C6T1	.000	9.000	.000	9.000	BREF 5.0000 IN.
(DXH088)	△	AEDC V41A-C1A, CANARD CONTROL, BN6C6T1	.000	15.000	.000	15.000	XREF 26.0000 IN.
							YREF .0000 IN.
							ZREF .0000 IN.
							SCALE .0000



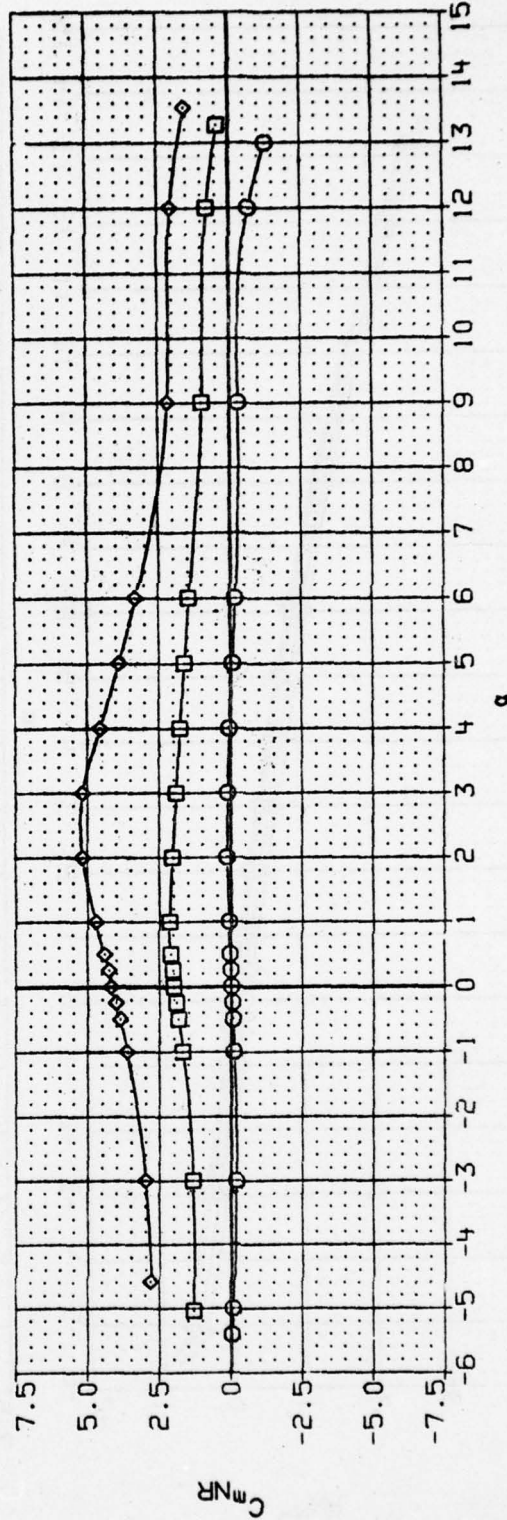
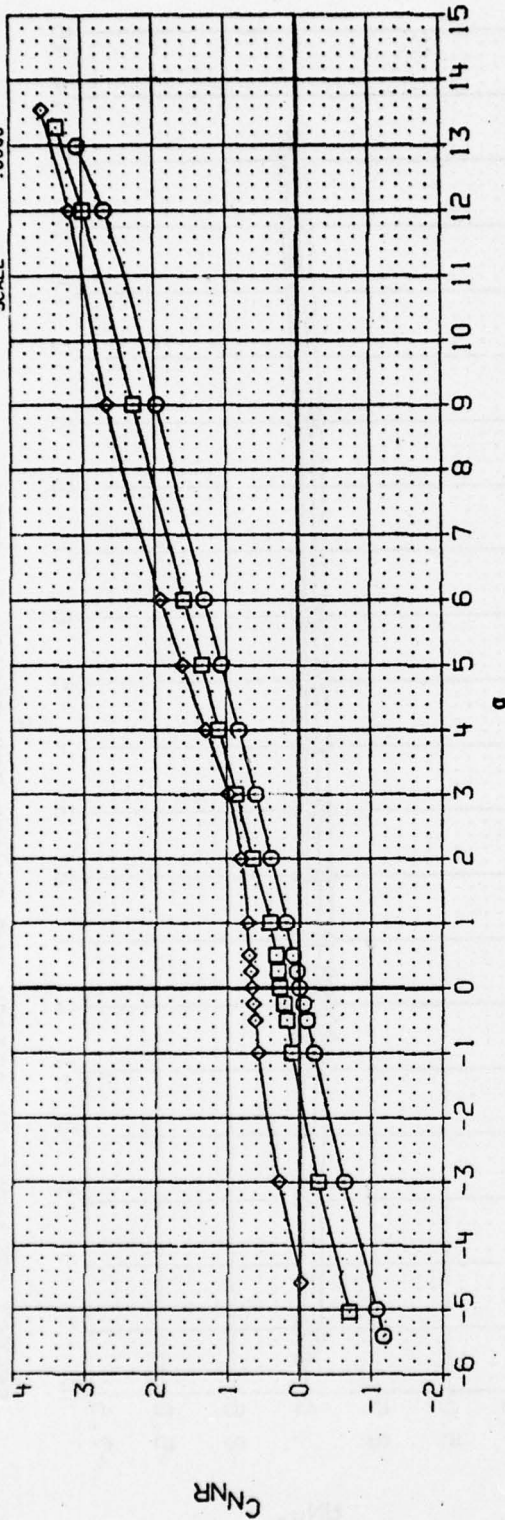
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=0 PHICND=0
 (A) MACH = 1.51

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH085)	○	AEDC V41A-C1A, CANARD CONTROL, BNSC6T1	.000	.000	.000	.000	SREF 19.6350 SQ. IN.
(DXH086)	□	AEDC V41A-C1A, CANARD CONTROL, BNSC6T1	.000	3.000	.000	3.000	LREF 5.0000 IN.
(DXH087)	◇	AEDC V41A-C1A, CANARD CONTROL, BNSC6T1	.000	9.000	.000	9.000	BREF 5.0000 IN.
(DXH088)	△	AEDC V41A-C1A, CANARD CONTROL, BNSC6T1	.000	15.000	.000	15.000	XMRP 26.0000 IN.
							YMRP .0000 IN.
							ZMRP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHICAL=0 PHICND=0
 (B)MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH089)	○	AEDC V41A-C1A, CANARD CONTROL, BNEC6T1	.000	.000	.000	.000	SREF 19.6350 50. IN.
(DXH090)	□	AEDC V41A-C1A, CANARD CONTROL, BNEC6T1	6.000	6.000	6.000	6.000	LREF 5.0000 IN.
(DXH091)	◇	AEDC V41A-C1A, CANARD CONTROL, BNEC6T1	15.000	15.000	15.000	15.000	BREF 5.0000 IN.
							XTRP 26.0000 IN.
							YTRP .0000 IN.
							ZTRP .0000 IN.
							SCALE .0000

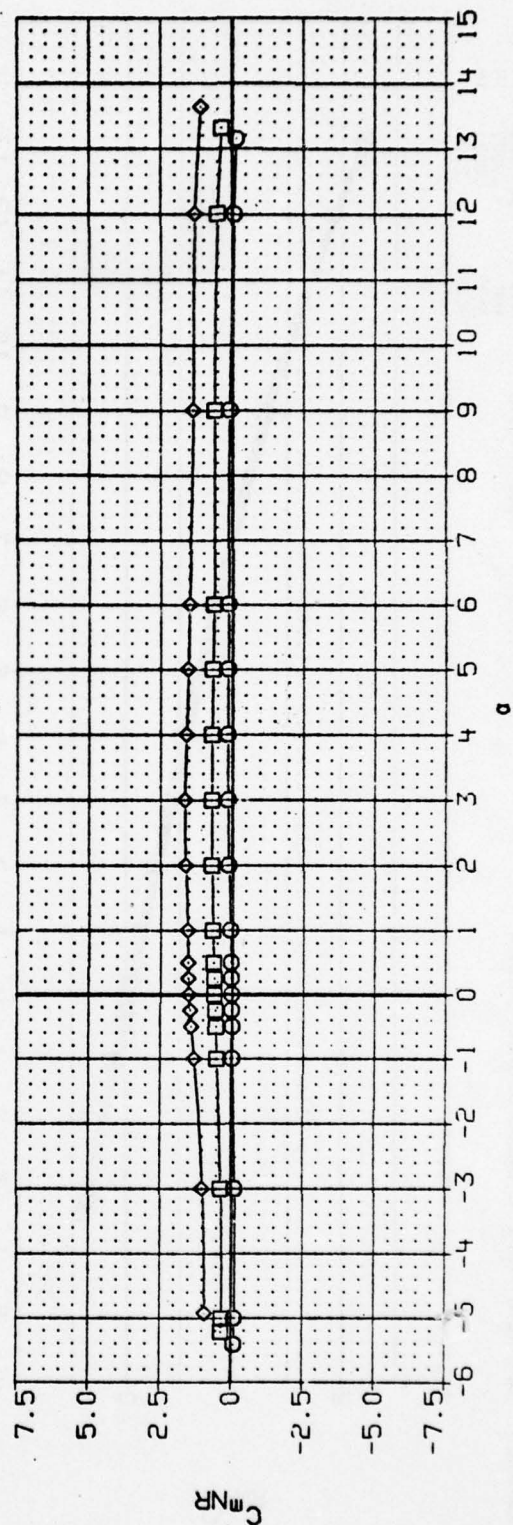
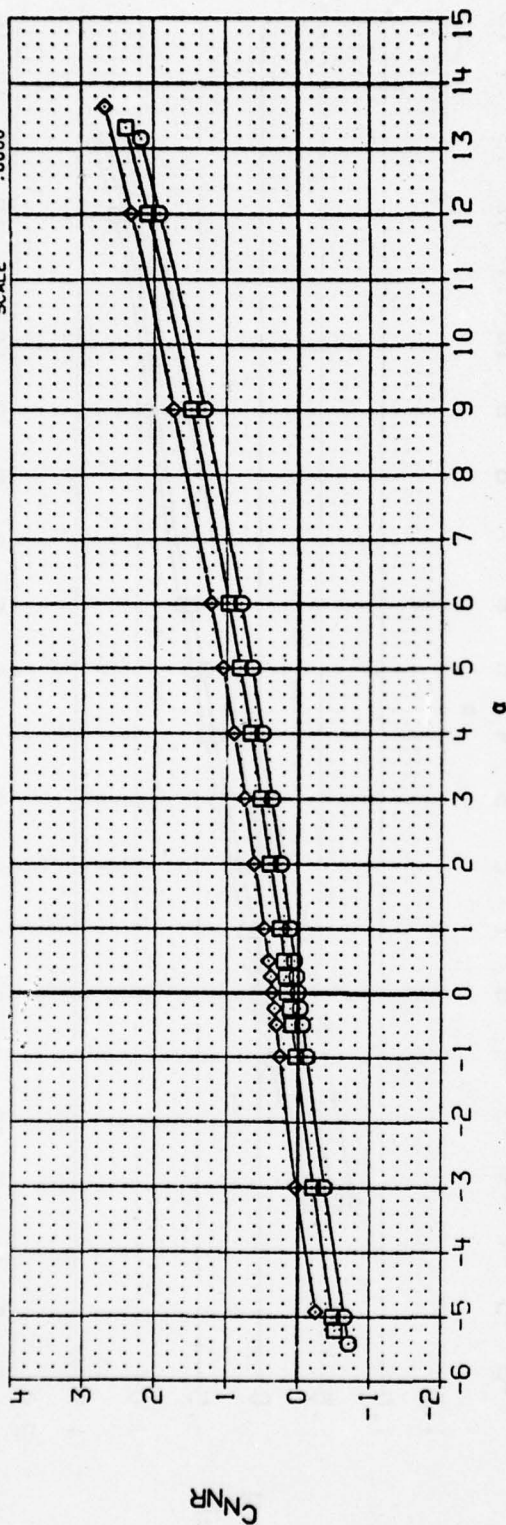


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DXH089) AEDC V41A-C1A, CANARD CONTROL, BNSC6T1
 (DXH090) AEDC V41A-C1A, CANARD CONTROL, BNSC6T1
 (DXH091) AEDC V41A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 6.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PHITAL=0 PHICND=45
 (B) MACH = 3.01

DATA SET SYMBOL
 (DXH093)
 (DXH094)
 (DXH095)
 (DXH096)

CONFIGURATION DESCRIPTION
 AEDC V41A-C1A, CANARD CONTROL, BNICITR
 AEDC V41A-C1A, CANARD CONTROL, BNICITR
 AEDC V41A-C1A, CANARD CONTROL, BNICITR
 AEDC V41A-C1A, CANARD CONTROL, BNICITR

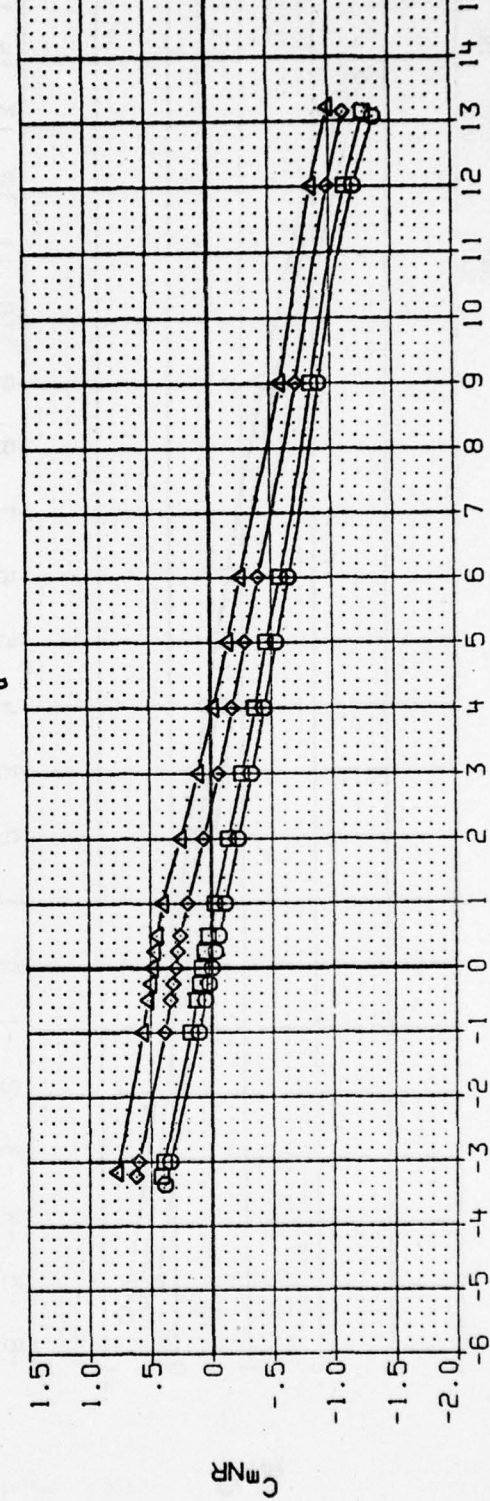
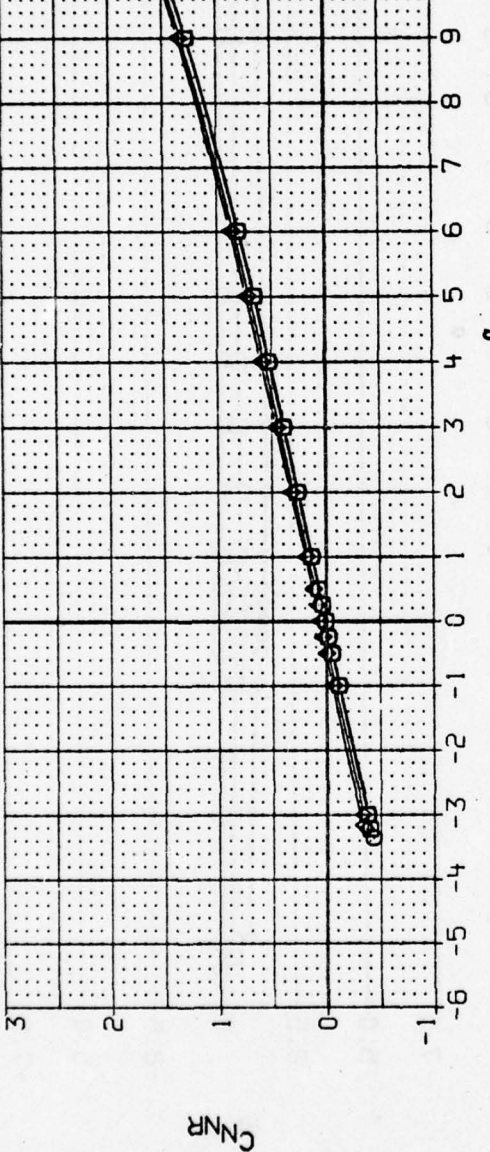
DCND1
 .000
 .000
 .000
 .000

DCND2
 .000
 3.000
 9.000
 15.000

DCND3
 .000
 .000
 .000
 .000

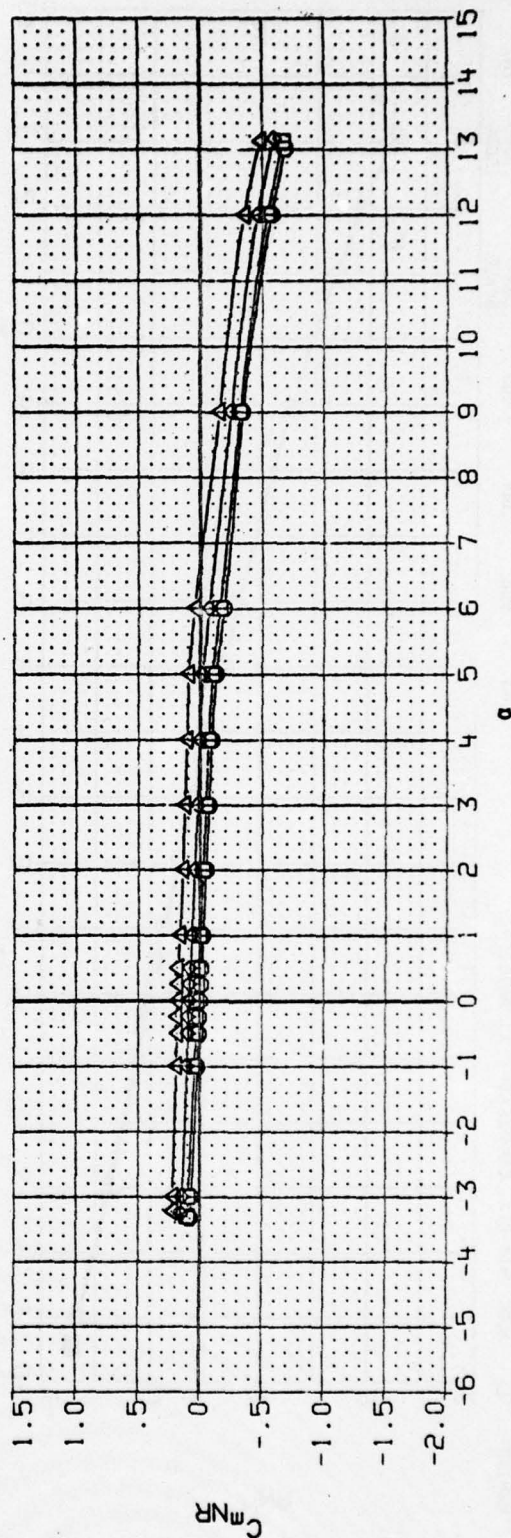
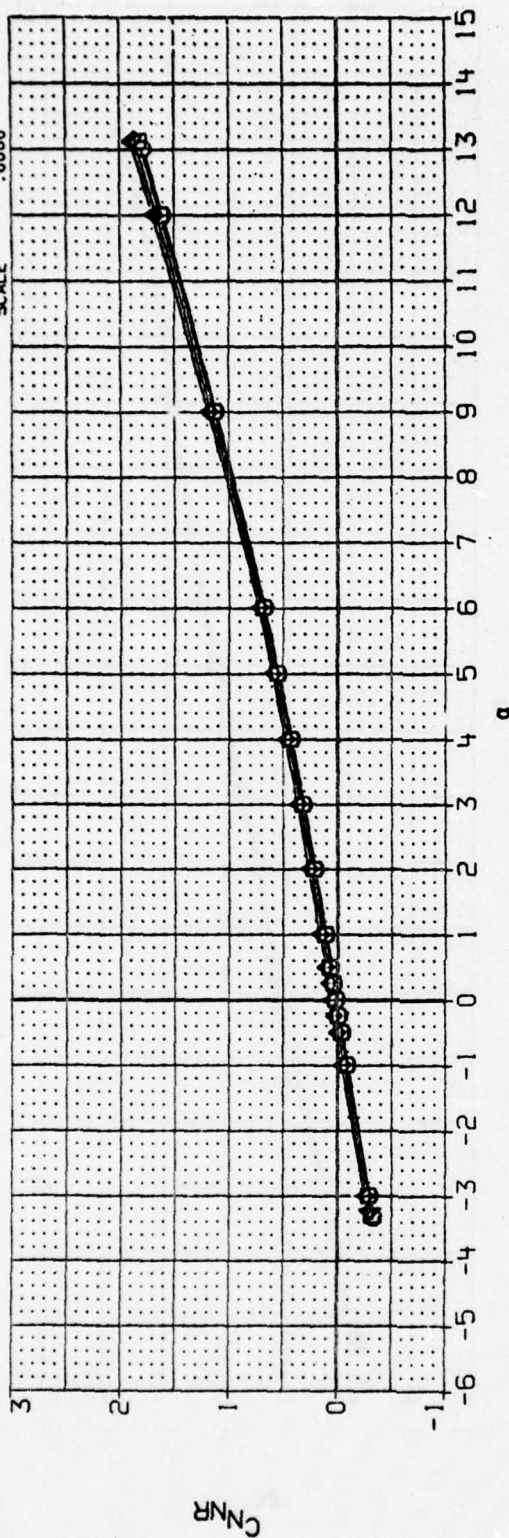
DCND4
 .000
 3.000
 9.000
 15.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
 PH1AL=0 PH1CND=0
 (A) MACH = 2.50

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(DXH093)	AEDC V4A-CIA, CANARD CONTROL, BNICITR	.000	.000	.000	.000	SREF 19.6350 50. IN.
(DXH094)	AEDC V4A-CIA, CANARD CONTROL, BNICITR	.000	.000	.000	.000	LREF 5.0000 IN.
(DXH095)	AEDC V4A-CIA, CANARD CONTROL, BNICITR	.000	.000	.000	.000	BREF 5.0000 IN.
(DXH096)	AEDC V4A-CIA, CANARD CONTROL, BNICITR	.000	.000	.000	.000	XPRP 26.0000 IN.
						YPRP .0000 IN.
						ZPRP .0000 IN.
						SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHICAL=0 PHICND=0

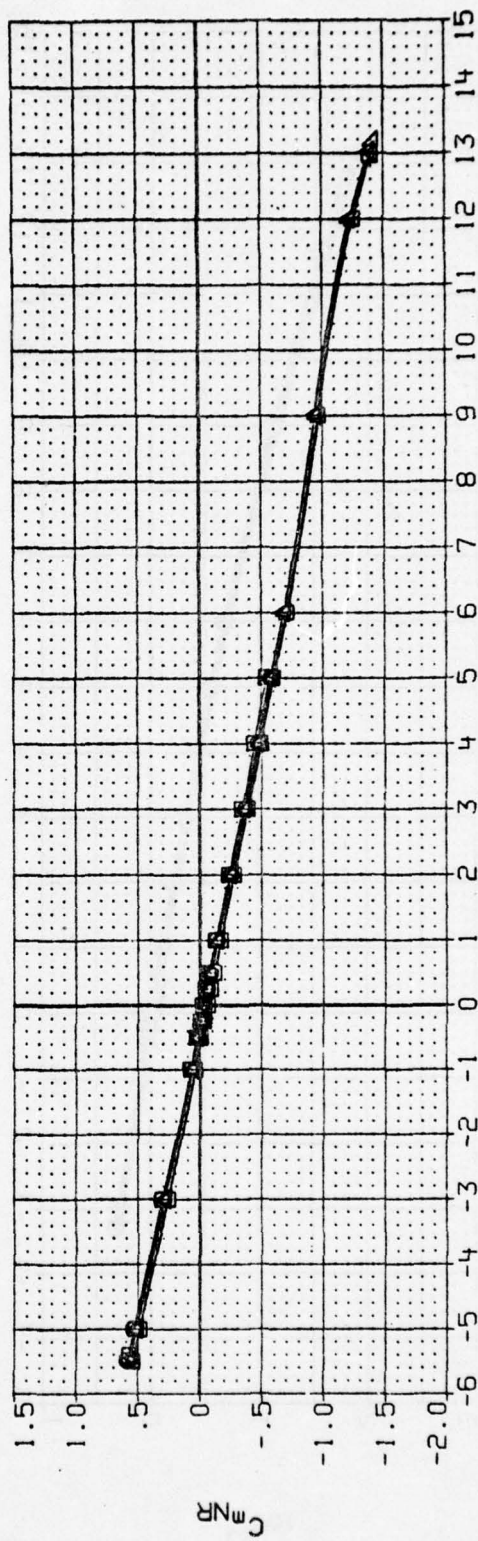
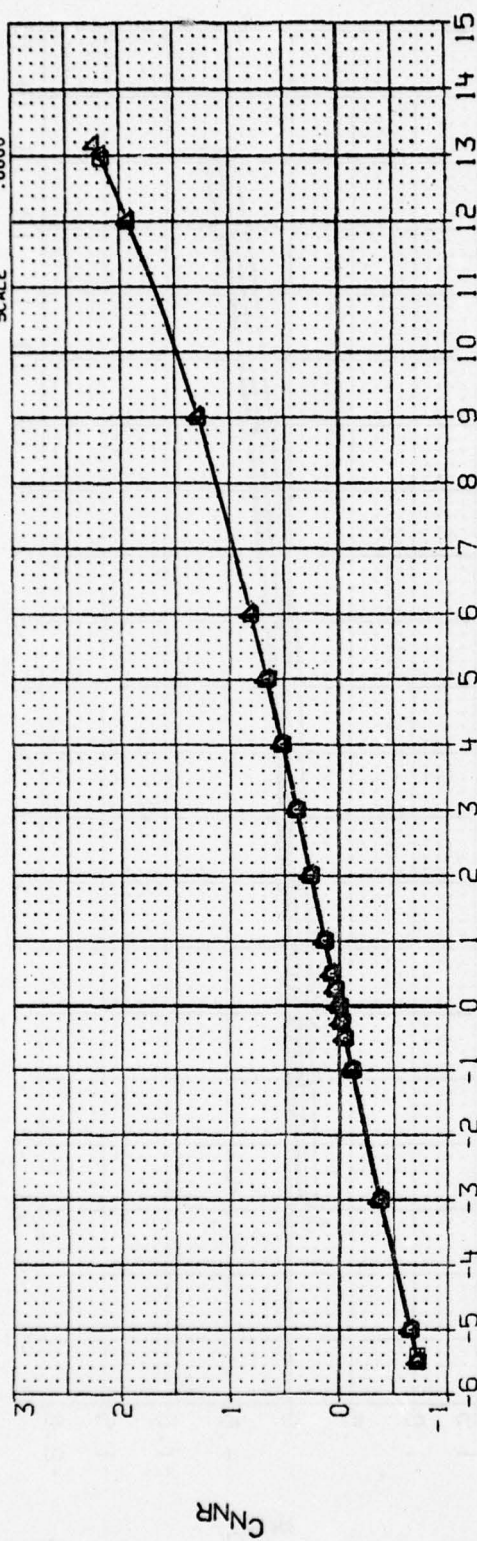
(B) MACH = 4.52

DATA SET SYMBOL

CONFIGURATION DESCRIPTION

DCND1 DCND2 DCND3 DCND4

REFERENCE INFORMATION



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHITAL=0 PHICND=0

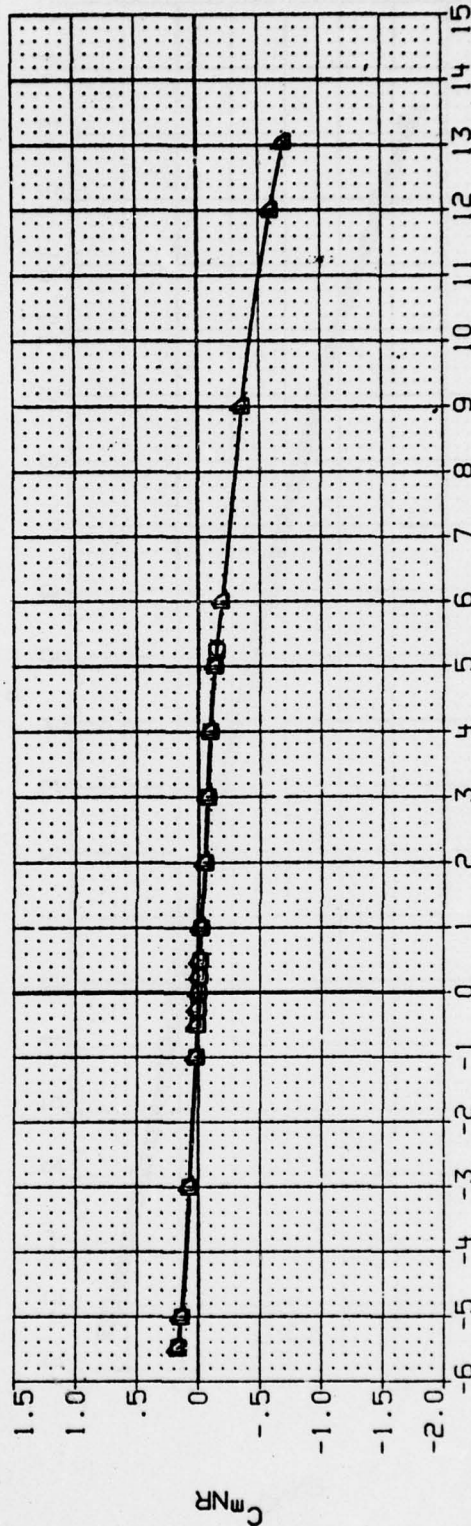
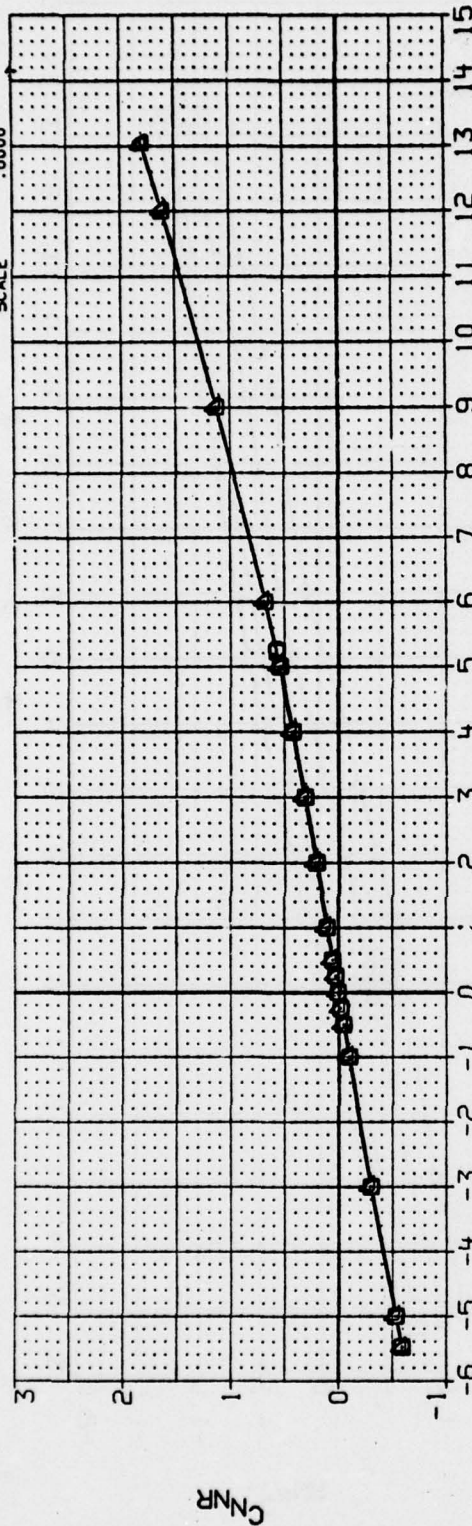
(A) MACH = 2.50

DATA SET SYMBOL
(DXH097)
(DXH098)
(DXH099)
(DXH101)
(DXH102)
(DXH103)

CONFIGURATION DESCRIPTION
AEDC V41A-CIA, CANARD CONTROL, BNICITR
AEDC V41A-CIA, CANARD CONTROL, BNICITR
DATA NOT AVAILABLE
DATA NOT AVAILABLE
AEDC V41A-CIA, CANARD CONTROL, BNICITR
AEDC V41A-CIA, CANARD CONTROL, BNICITR

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE

DCND1 5.000
DCND2 -5.000
DCND3 -2.000
DCND4 5.000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY
PHITAL=0 PHICND=0
(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXH104) AEDC V41A-C1A, CANARD CONTROL, BNICITR

(DXH105) AEDC V41A-C1A, CANARD CONTROL, BNICITR

(DXH106) AEDC V41A-C1A, CANARD CONTROL, BNICITR

(DXH107) AEDC V41A-C1A, CANARD CONTROL, BNICITR

DCND1 .000

DCND2 .000

DCND3 .000

DCND4 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

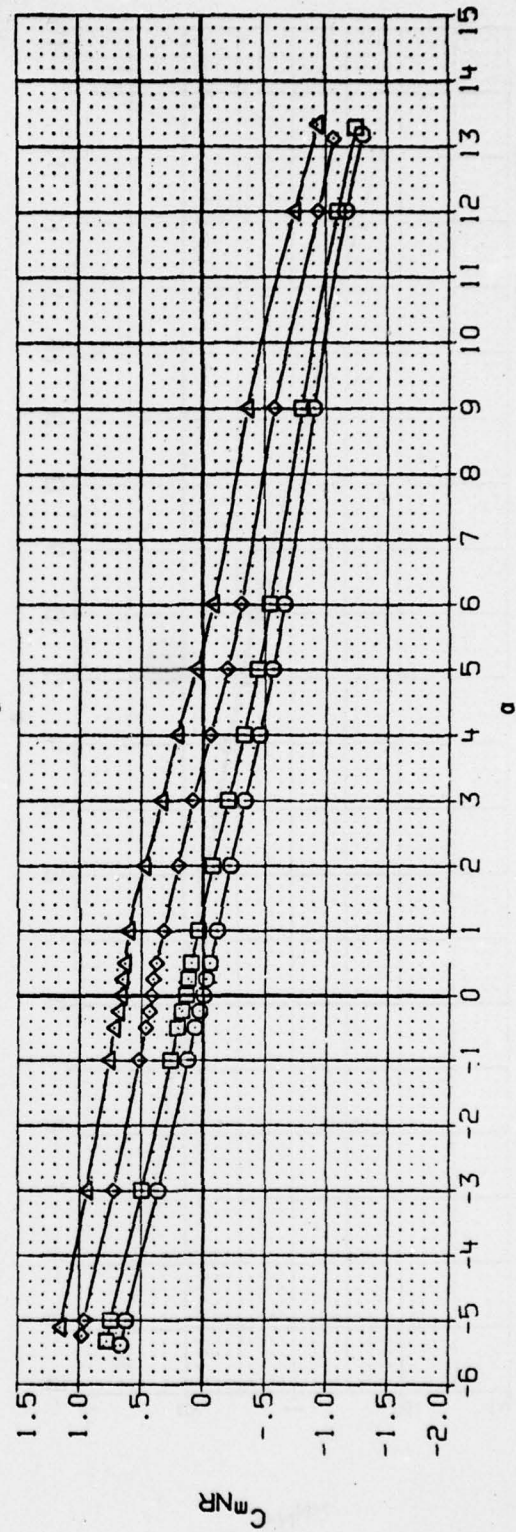
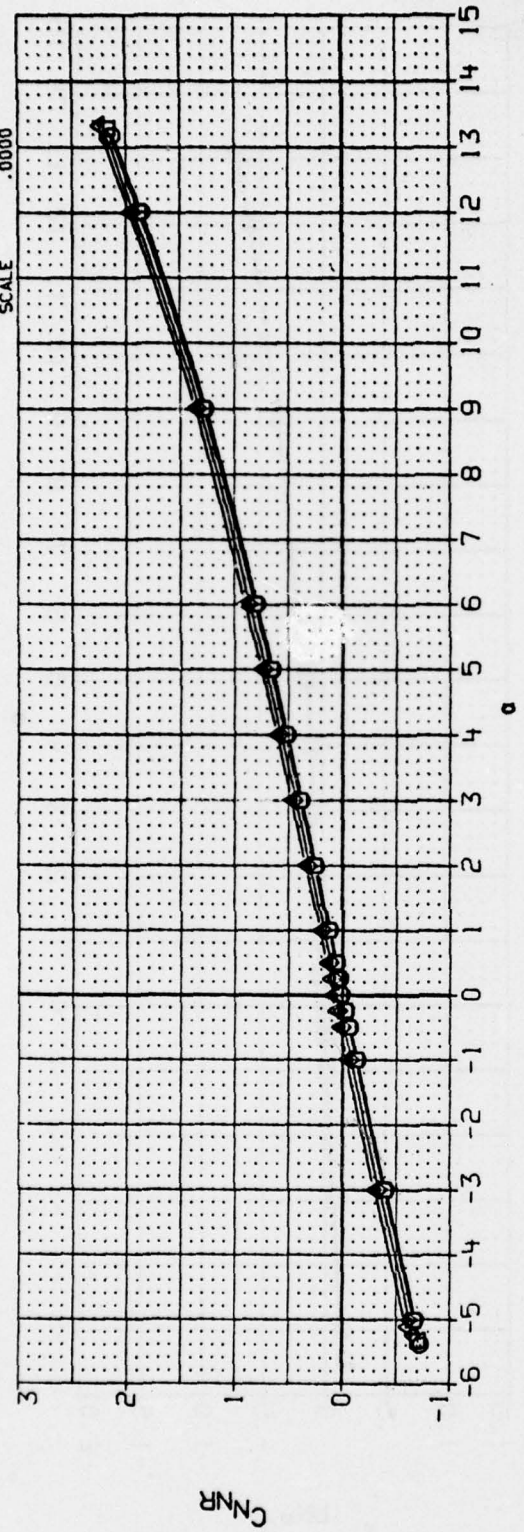
BREF 5.0000 IN.

XPRP 26.0000 IN.

YPRP .0000 IN.

ZPRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHITAL=0 PHICND=45

(A) MACH = 2.50

DATA SET SYMBOL

(DXH104) \square

(DXH105) \square

(DXH106) \diamond

(DXH107) \triangle

CONFIGURATION DESCRIPTION

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

3.000 3.000 3.000 3.000

9.000 9.000 9.000 9.000

15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 IN.

LREF 5.0000 IN.

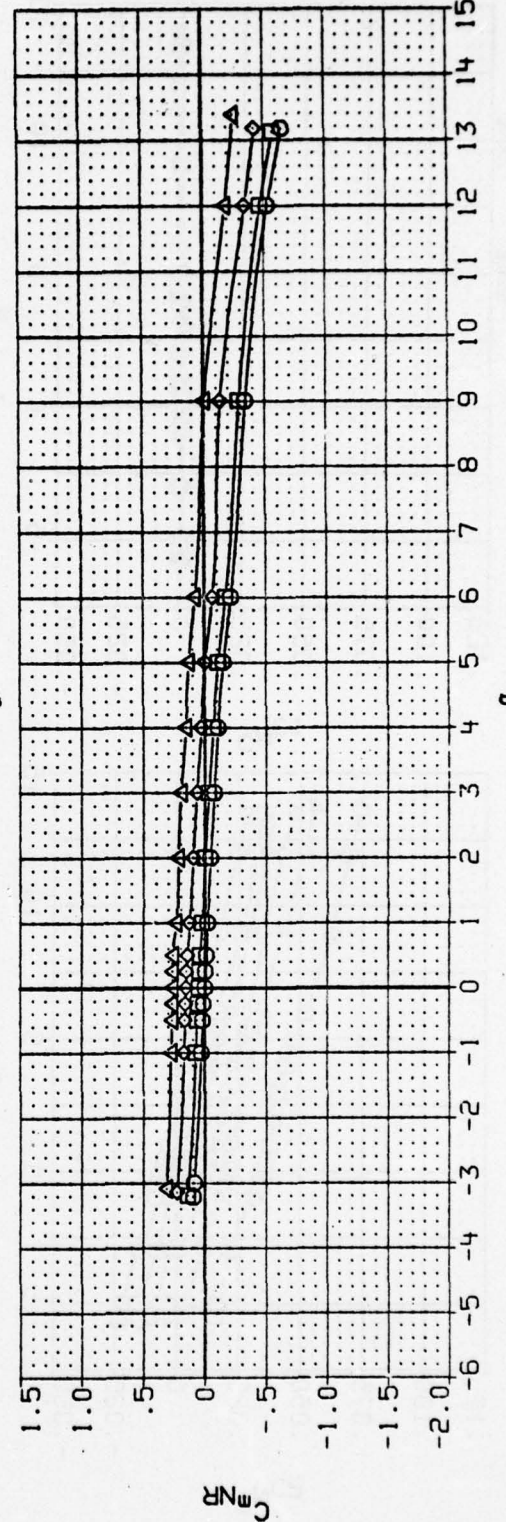
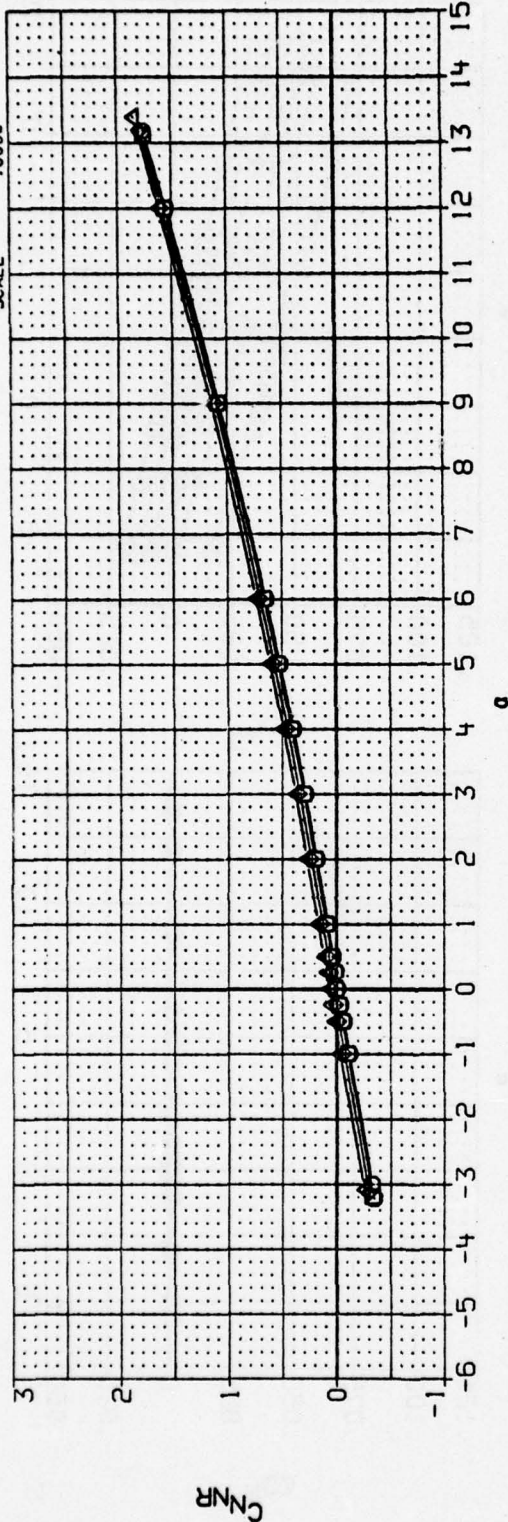
BREF 5.0000 IN.

XRRP 26.0000 IN.

YRRP .0000 IN.

ZRRP .0000 IN.

SCALE .0000

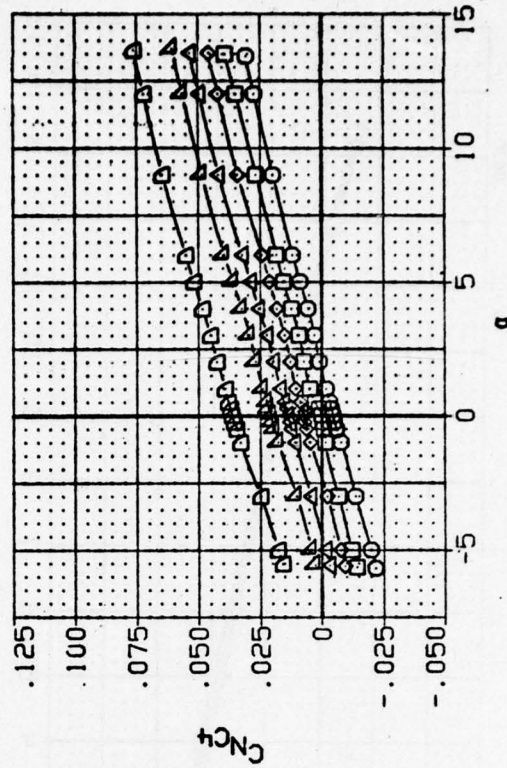
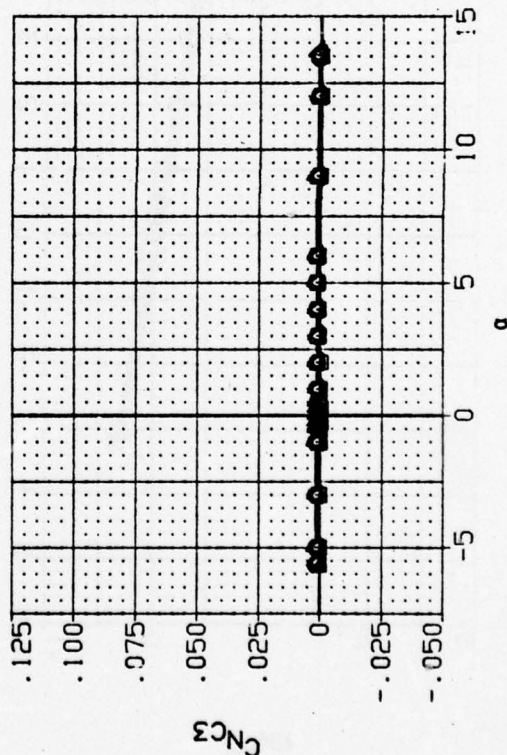
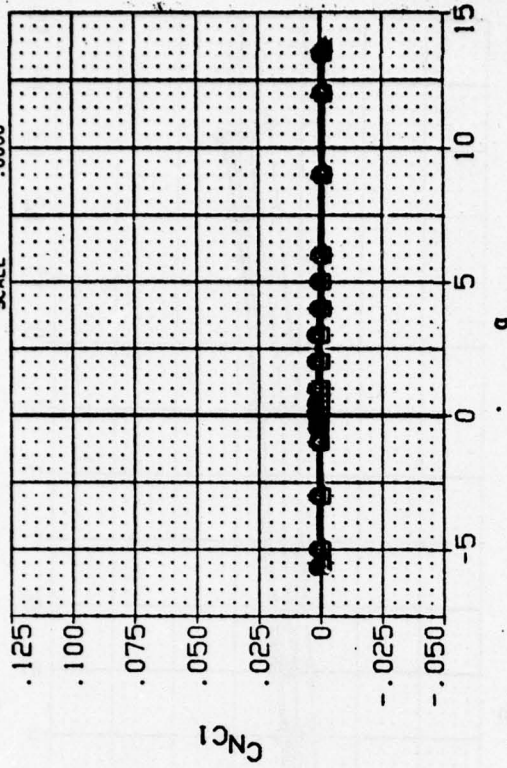
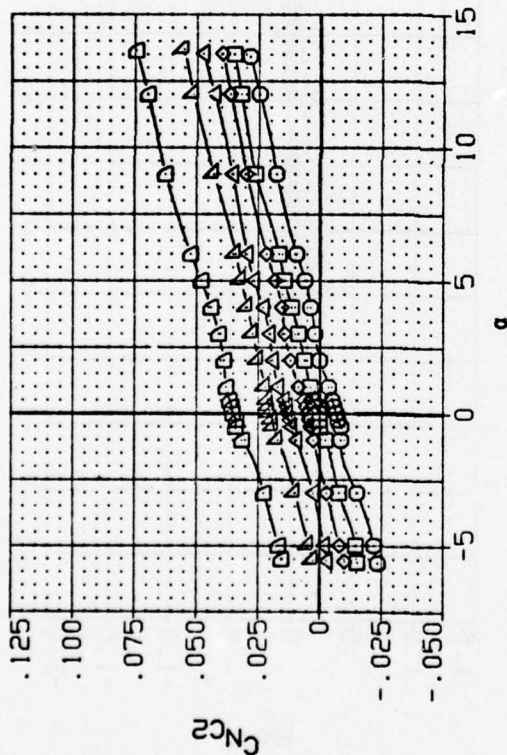


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON BODY

PHITAL=0 PHICND=45

(B) MACH = 4.52

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(AXH004)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	-3.000	.000	-3.000	SREF 19.6350 IN.
(AXH005)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	.000	.000	.000	LREF 5.0000 IN.
(AXH006)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	3.000	.000	3.000	BREF 5.0000 IN.
(AXH007)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	6.000	.000	6.000	XMRP 26.0000 IN.
(AXH008)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	9.000	.000	9.000	YMRP .0000 IN.
(AXH009)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	15.000	.000	15.000	ZMRP .0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
TAIL OFF PHICND=0
(A) MACH = 3.01

DATA SET SYMBOL

(BXH004) \square

(BXH005) \square

(BXH006) \square

(BXH007) \square

(BXH008) \square

(BXH009) \square

CONFIGURATION DESCRIPTION

AEDC V4A-CIA, CANARD CONTROL, BN1C1

AEDC V4A-CIA, CANARD CONTROL, BN1C1

AEDC V4A-CIA, CANARD CONTROL, BN1C1

AEDC V4A-CIA, CANARD CONTROL, BN1C1

AEDC V4A-CIA, CANARD CONTROL, BN1C1

AEDC V4A-CIA, CANARD CONTROL, BN1C1

DCND1 DCND2 DCND3 DCND4

.000 -3.000 .000 -3.000

.000 .000 .000 .000

.000 3.000 .000 3.000

.000 6.000 .000 9.000

.000 9.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

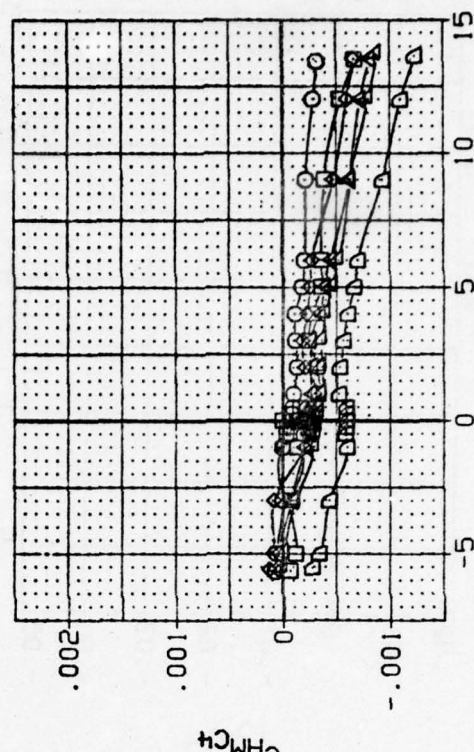
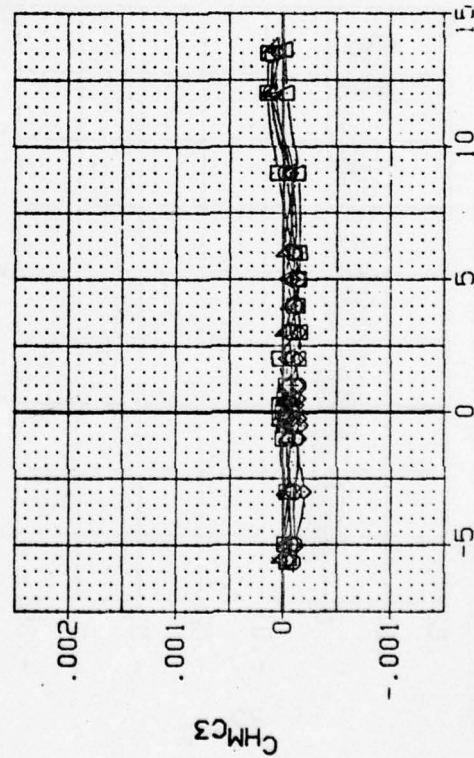
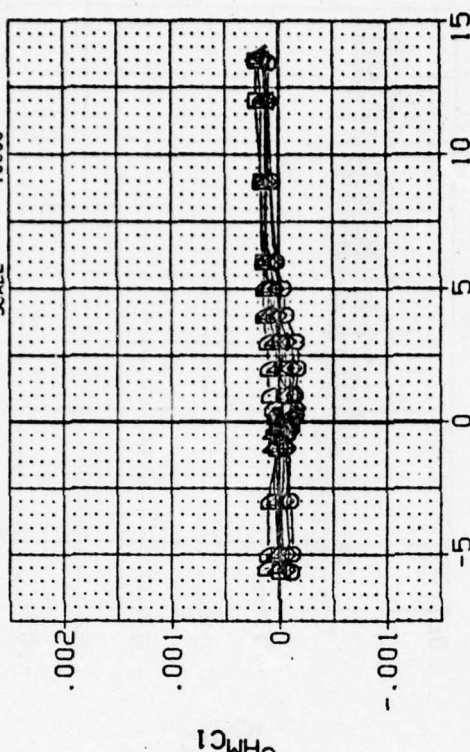
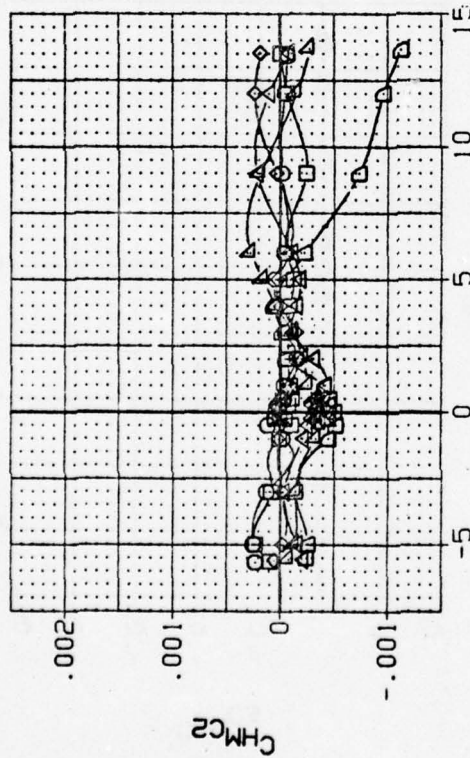
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

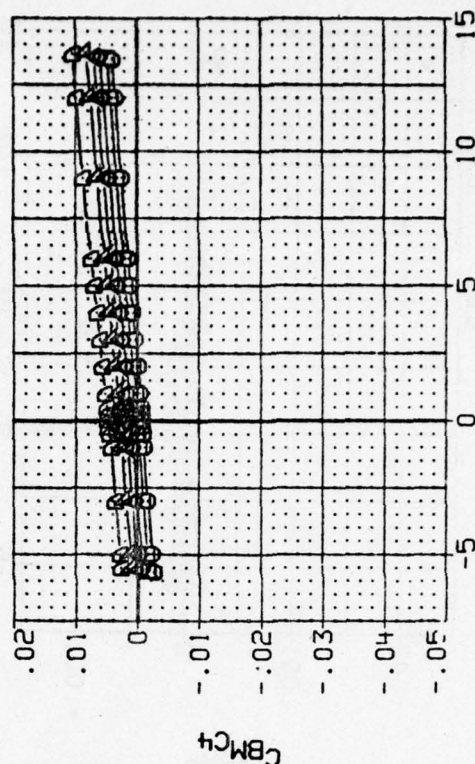
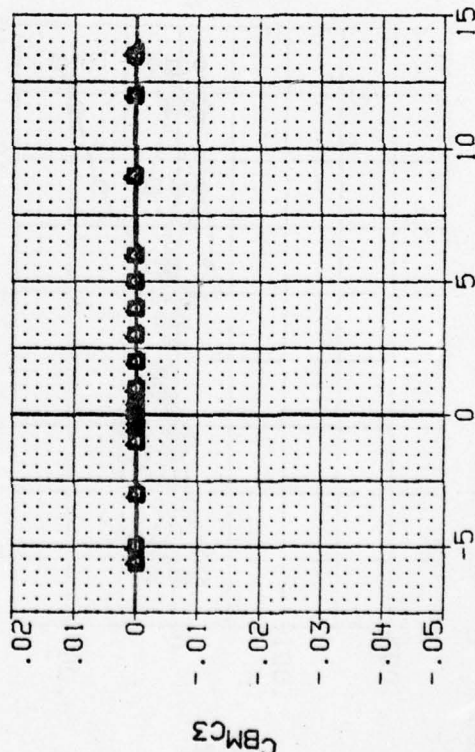
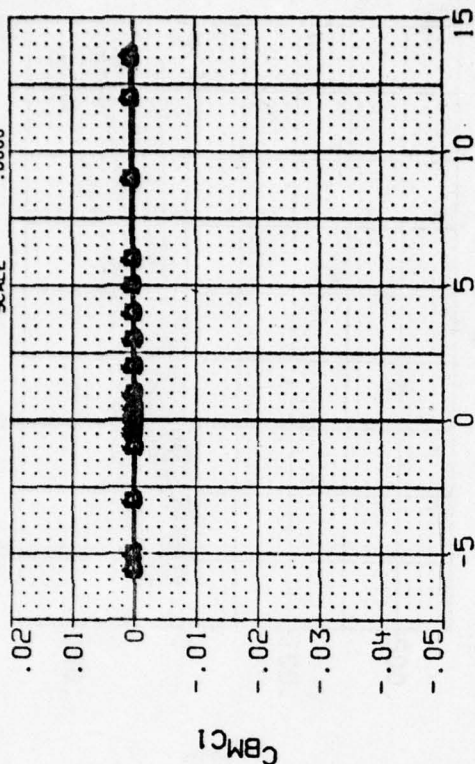
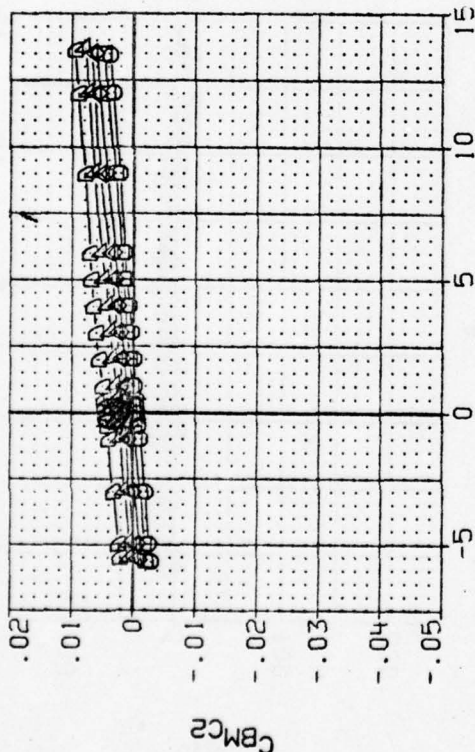
ZMRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
TAIL OFF PHICND=0
(A) MACH = 3.01

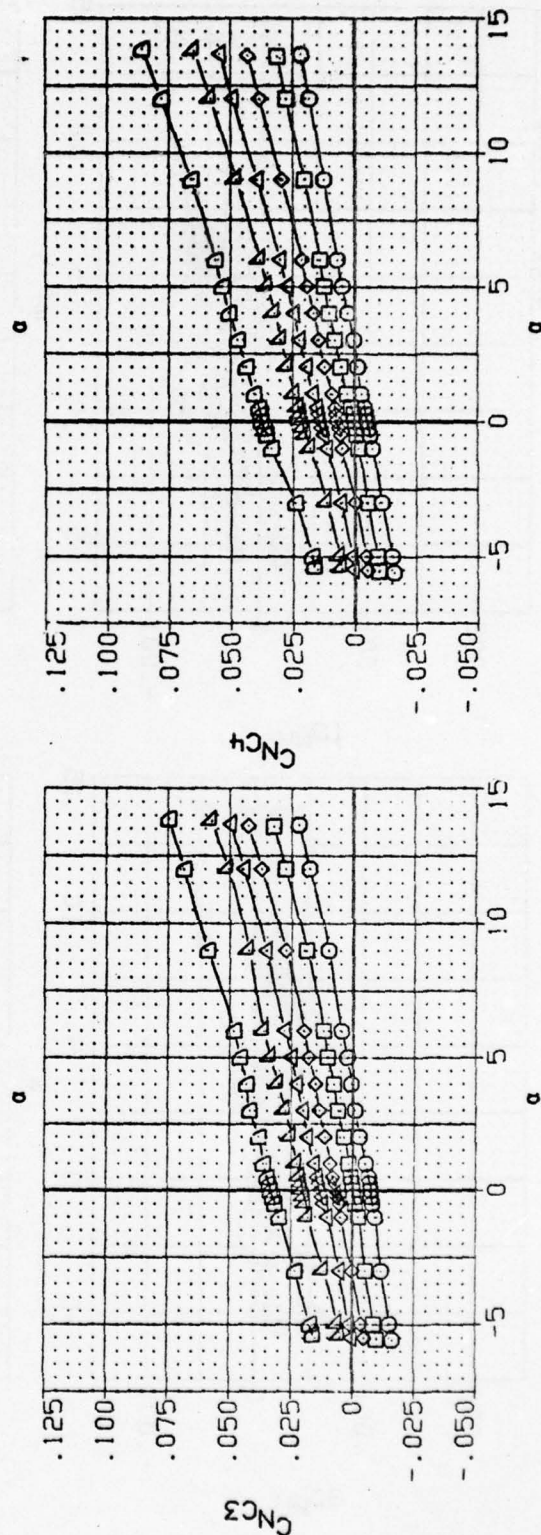
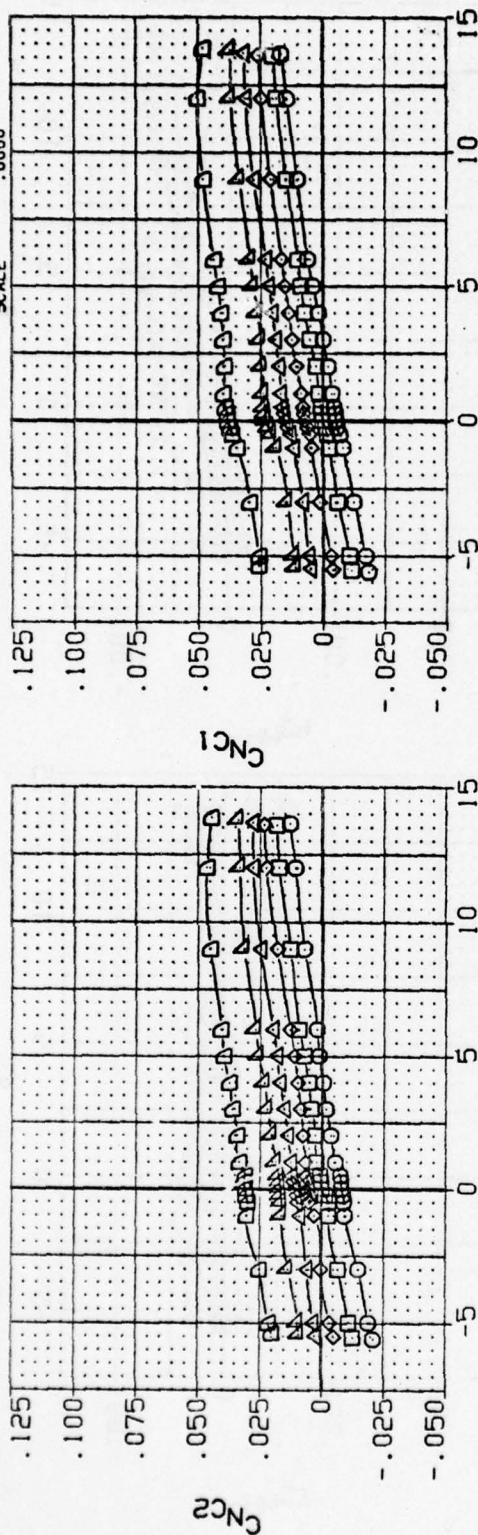
DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(BXH004)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	-3.000	.000	-3.000	SREF 19.6350 50. IN.
(BXH005)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	.000	.000	.000	LREF 5.0000 IN.
(BXH006)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	3.000	.000	3.000	BREF 5.0000 IN.
(BXH007)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	6.000	.000	6.000	XMRP 26.0000 IN.
(BXH008)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	9.000	.000	9.000	YMRP .0000 IN.
(BXH009)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1	.000	15.000	.000	15.000	ZMRP .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
TAIL OFF PHICND=0

(A) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(AXH010)	□	AEDC V41A-CIA, CANARD CONTROL, BNIC1	-3.000	-3.000	-3.000	-3.000	SREF 19.6350 50. IN.
(AXH011)	◇	AEDC V41A-CIA, CANARD CONTROL, BNIC1	.000	.000	.000	.000	LREF 5.0000 IN.
(AXH012)	◇	AEDC V41A-CIA, CANARD CONTROL, BNIC1	3.000	3.000	3.000	3.000	BREF 5.0000 IN.
(AXH013)	△	AEDC V41A-CIA, CANARD CONTROL, BNIC1	6.000	6.000	6.000	6.000	YPRP 26.0000 IN.
(AXH014)	△	AEDC V41A-CIA, CANARD CONTROL, BNIC1	9.000	9.000	9.000	9.000	YPRP .0000 IN.
(AXH015)	△	AEDC V41A-CIA, CANARD CONTROL, BNIC1	15.000	15.000	15.000	15.000	ZPRP .0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
TAIL OFF PHICND=45

(A) MACH = 3.01

DATA SET SYMBOL

(BXH010)
(BXH011)
(BXH012)
(BXH013)
(BXH014)
(BXH015)

CONFIGURATION DESCRIPTION

AEDC V41A-C1A, CANARD CONTROL, BNIC1
AEDC V41A-C1A, CANARD CONTROL, BNIC1
AEDC V41A-C1A, CANARD CONTROL, BNIC1
AEDC V41A-C1A, CANARD CONTROL, BNIC1
AEDC V41A-C1A, CANARD CONTROL, BNIC1
AEDC V41A-C1A, CANARD CONTROL, BNIC1

DCND1

-3.000
-3.000
3.000
6.000
9.000
15.000

DCND2

-3.000
-3.000
3.000
6.000
9.000
15.000

DCND3

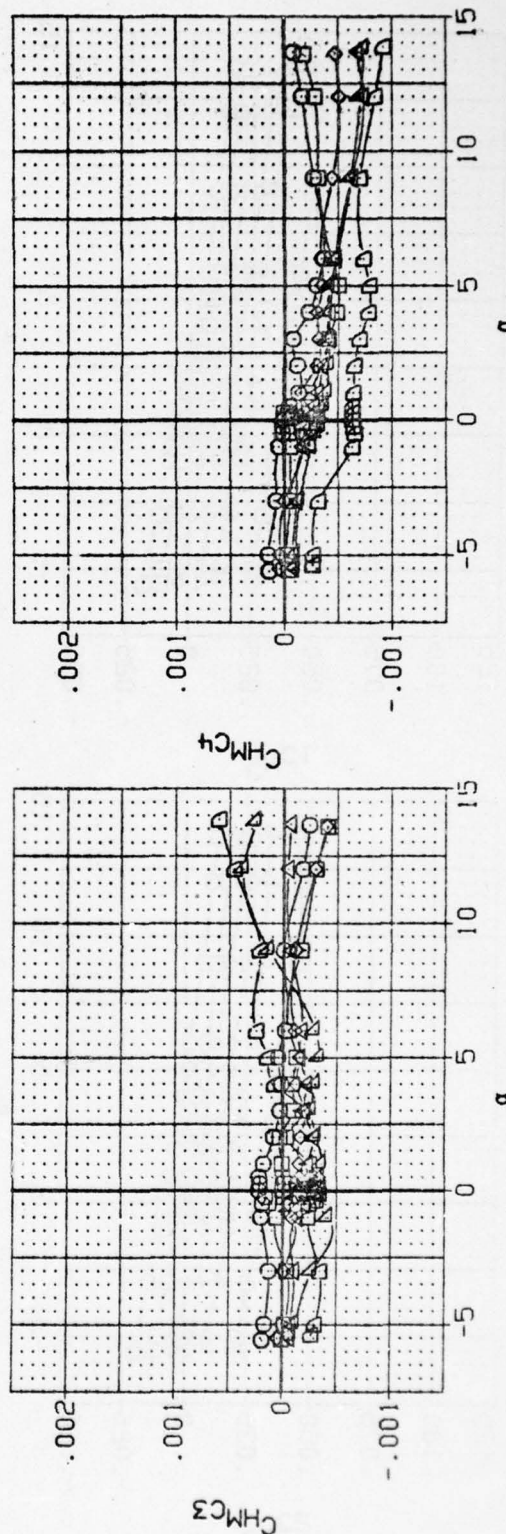
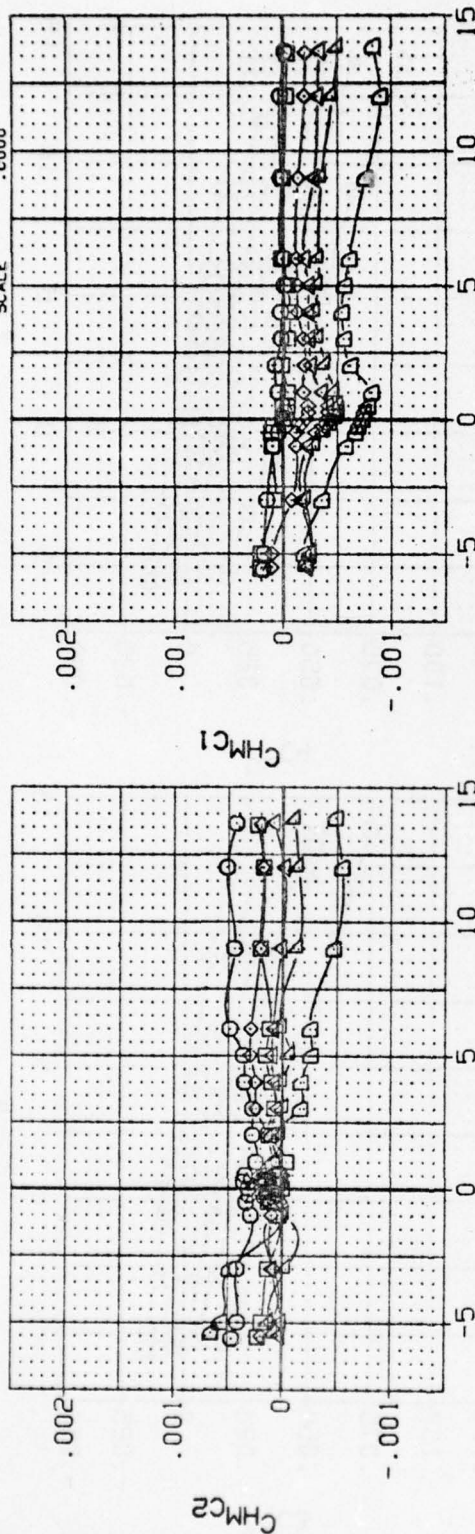
-3.000
-3.000
3.000
6.000
9.000
15.000

DCND4

-3.000
-3.000
3.000
6.000
9.000
15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
TAIL OFF PHICND=45
(A)MACH = 3.01

AD-A033 783

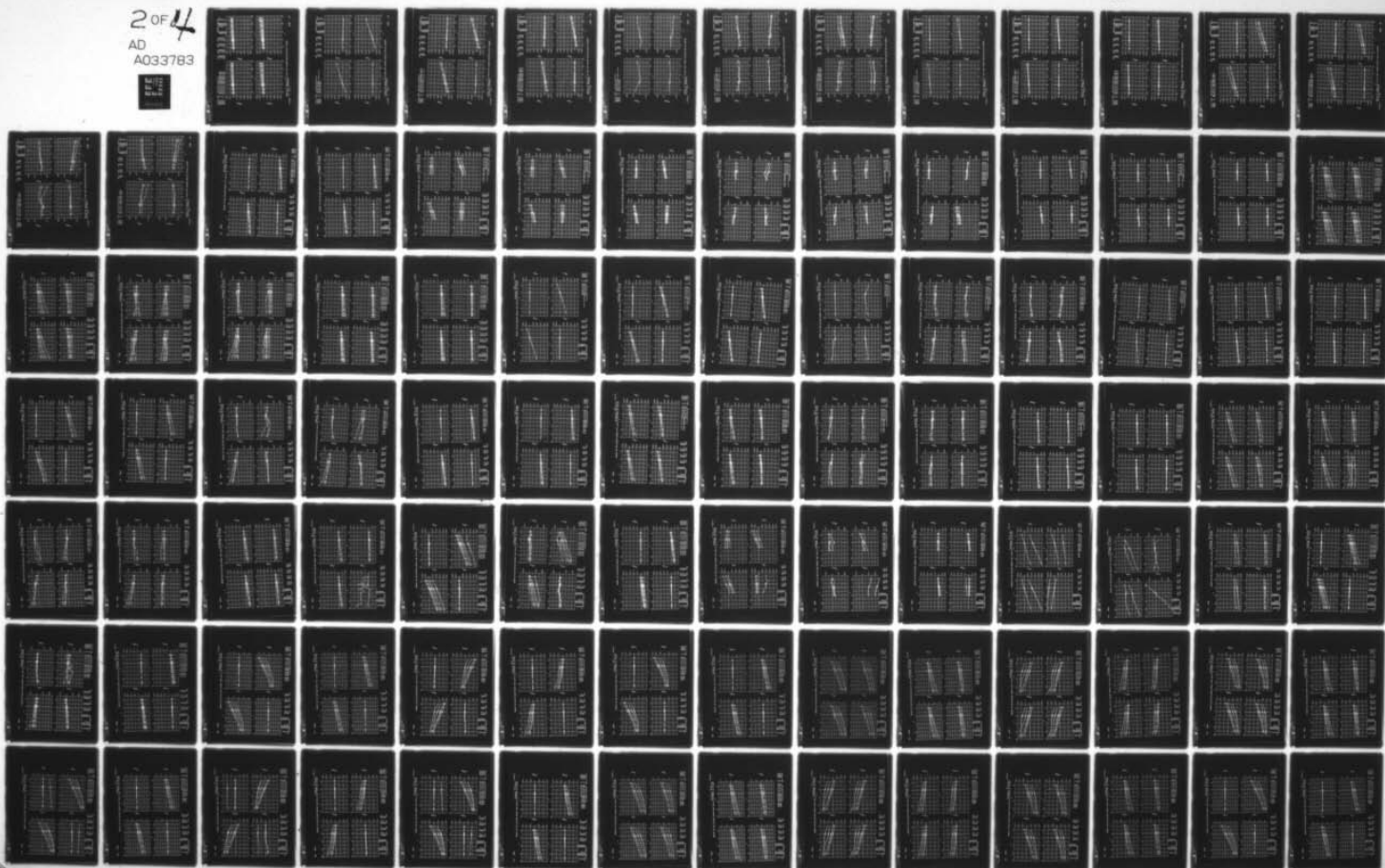
ARMY MISSILE RESEARCH DEVELOPMENT AND ENGINEERING LAB--ETC F/G 20/4
AN EXPERIMENTAL INVESTIGATION OF THE AERODYNAMIC CHARACTERISTIC--ETC(U)
OCT 76 J R BURT

UNCLASSIFIED

RD-77-5

NL

2 OF 4
AD
A033783



DATA SET SYMBOL

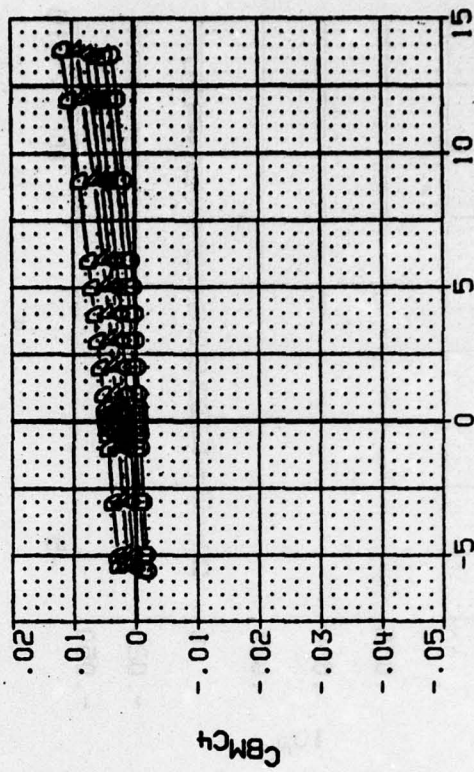
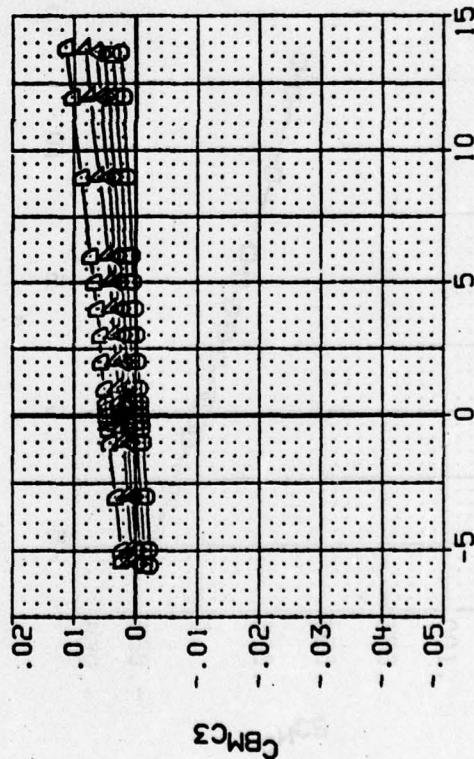
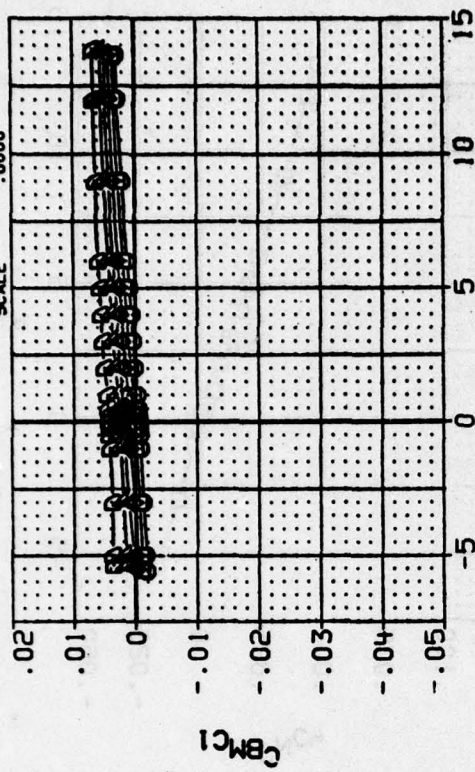
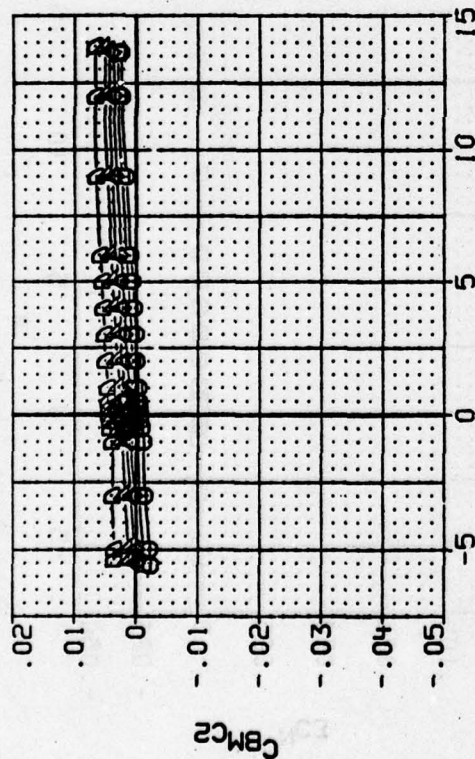
(BXH010) AEDC V41A-C1A, CANARD CONTROL, BNIC1
 (BXH011) AEDC V41A-C1A, CANARD CONTROL, BNIC1
 (BXH012) AEDC V41A-C1A, CANARD CONTROL, BNIC1
 (BXH013) AEDC V41A-C1A, CANARD CONTROL, BNIC1
 (BXH014) AEDC V41A-C1A, CANARD CONTROL, BNIC1
 (BXH015) AEDC V41A-C1A, CANARD CONTROL, BNIC1

CONFIGURATION DESCRIPTION

DCND1 DCND2 DCND3 DCND4
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000



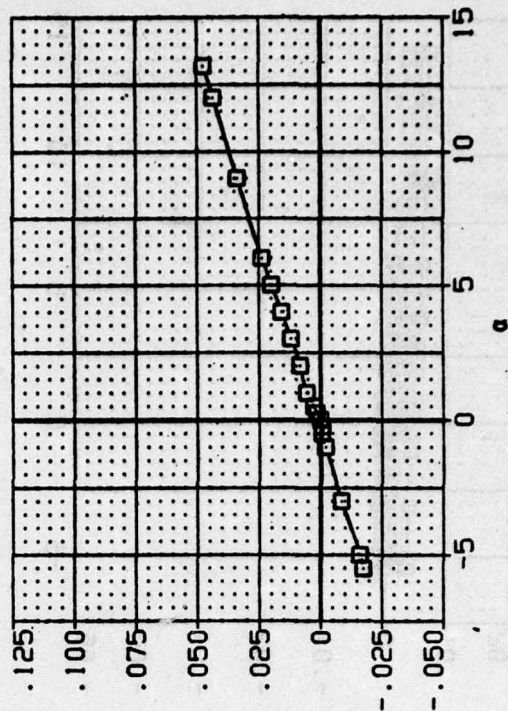
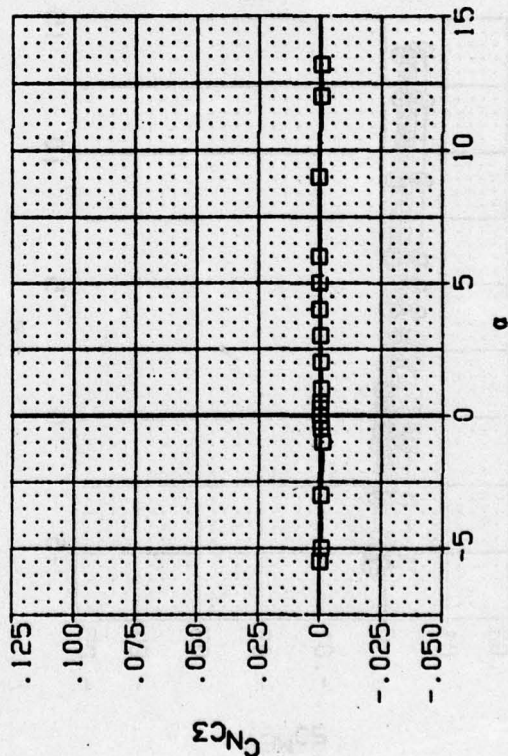
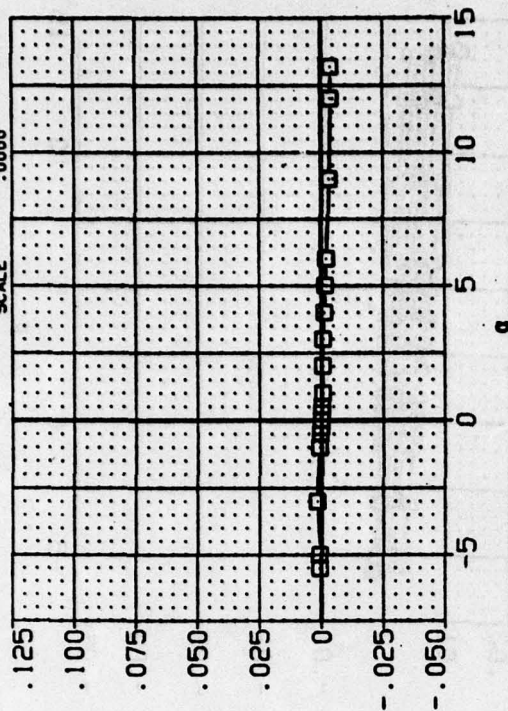
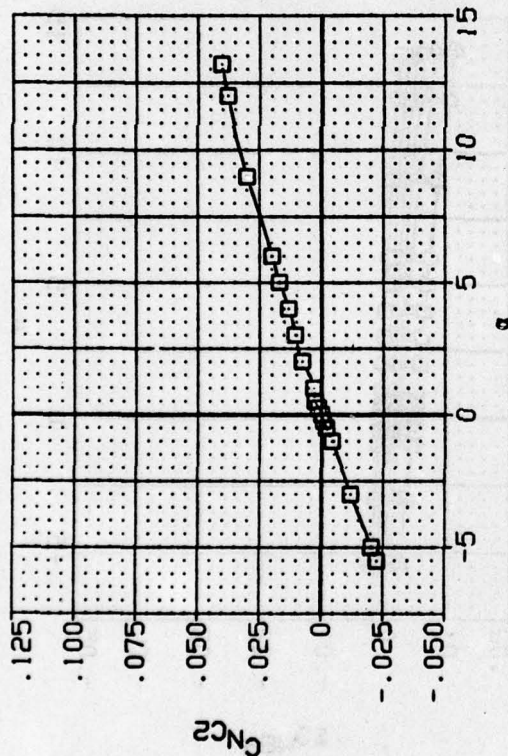
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 TAIL OFF PHICND=45

(A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH016) DATA NOT AVAILABLE
 (AXH017) AEDC W-1A-CIA, CANARD CONTROL, BNICITI
 (AXH018) DATA NOT AVAILABLE
 (AXH019) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YPRP 26.0000 IN.
 ZPRP .0000 IN.
 SCALE .0000

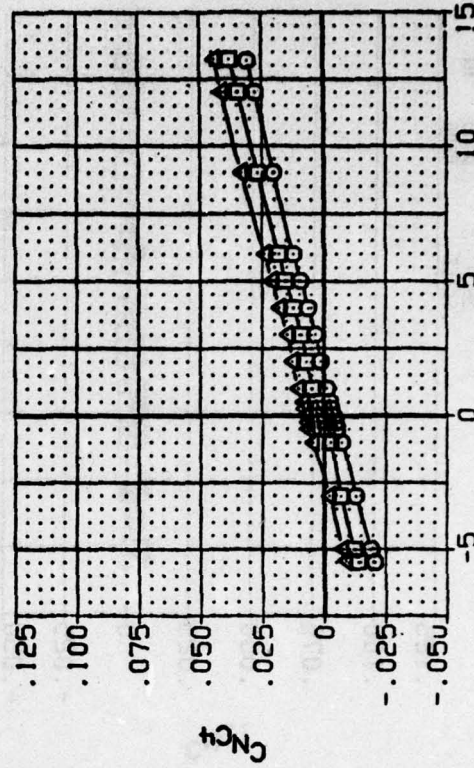
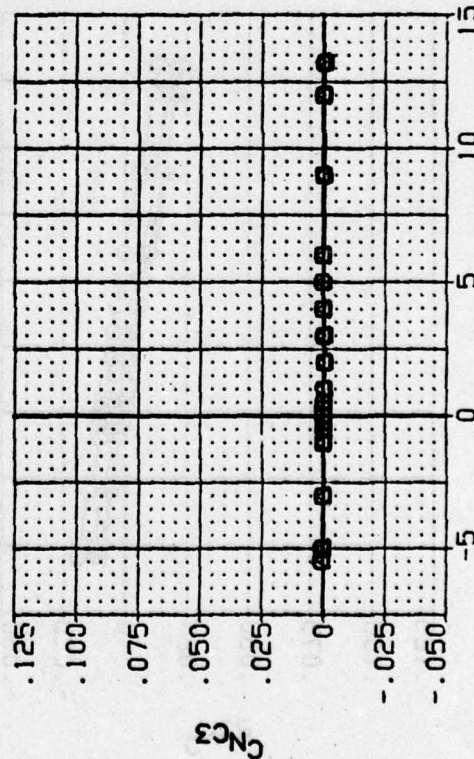
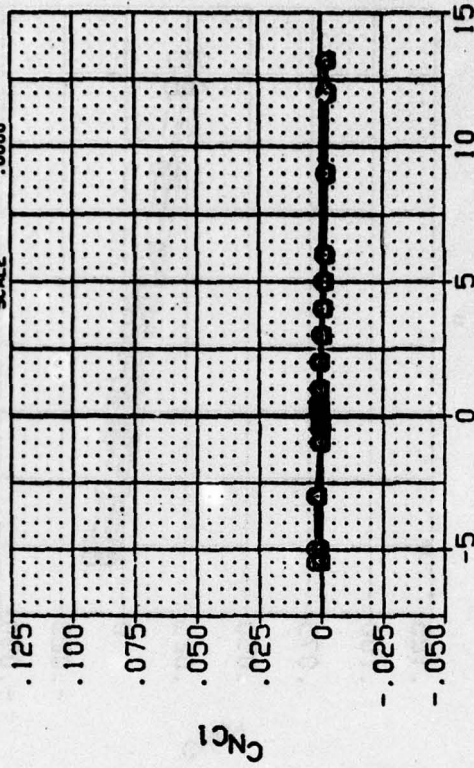
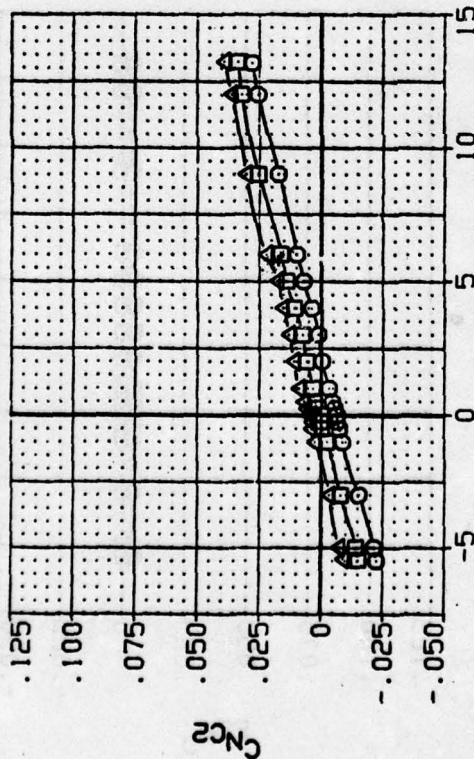


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHICAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH016) \square AEDC V41A-C1A, CANARD CONTROL, ENICITI
 (AXH017) \square AEDC V41A-C1A, CANARD CONTROL, ENICITI
 (AXH018) \diamond DATA NOT AVAILABLE
 (AXH019) \triangle AEDC V41A-C1A, CANARD CONTROL, ENICITI

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

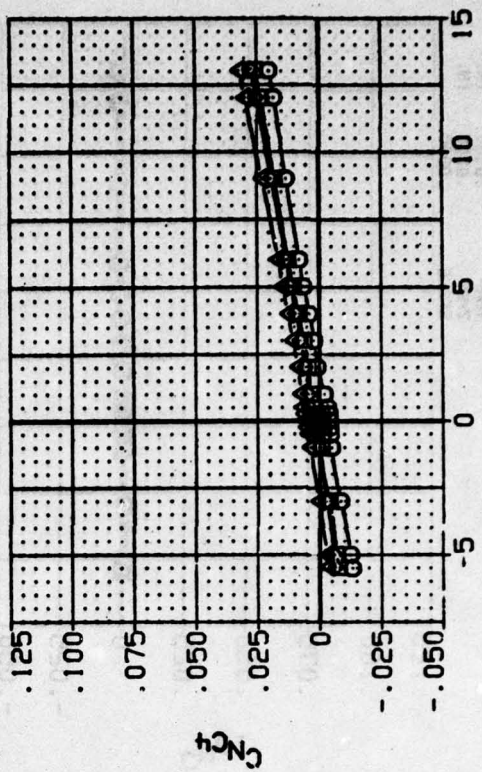
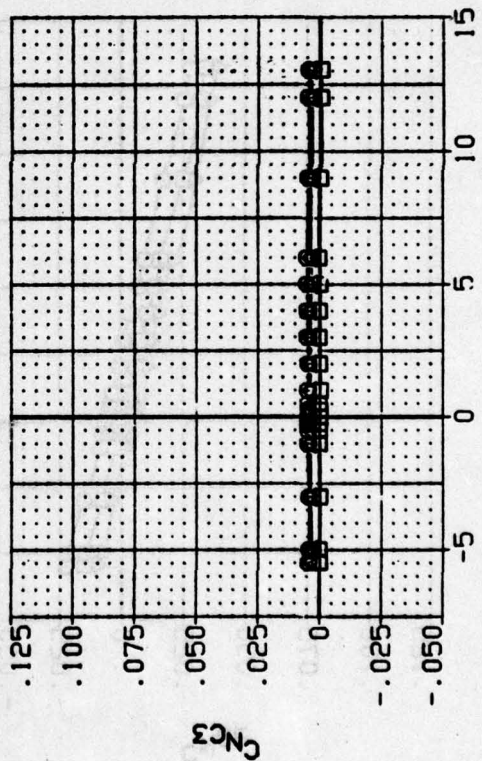
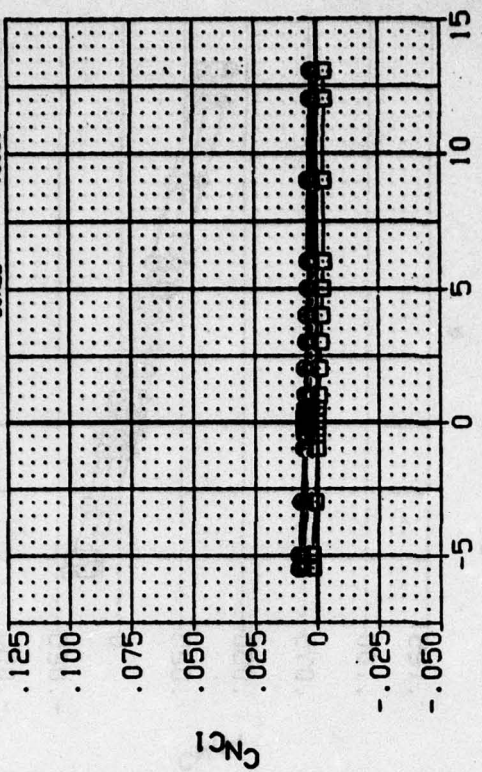
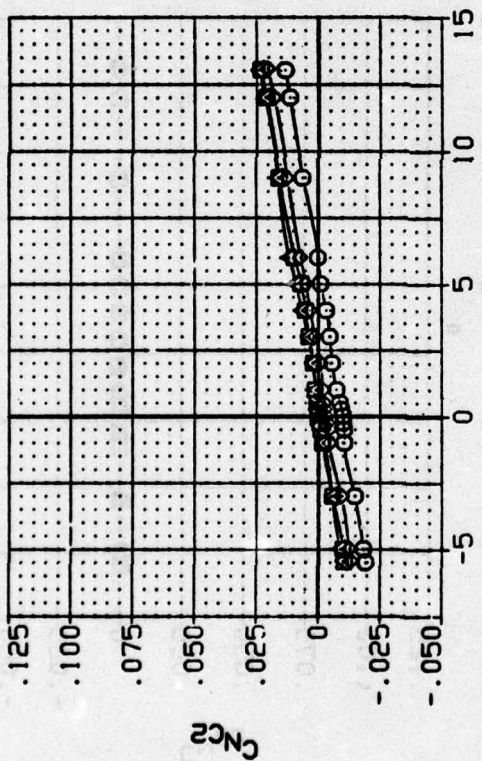
REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0

(B) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(AXH016)	□	AEDC W1A-C1A, CANARD CONTROL.	.000	-3.000	.000	-3.000	SREF 19.6350 SO. IN.
(AXH017)	□	AEDC W1A-C1A, CANARD CONTROL.	.000	.000	.000	.000	LREF 5.0000 IN.
(AXH018)	◇	AEDC W1A-C1A, CANARD CONTROL.	.000	1.000	.000	1.000	BREF 5.0000 IN.
(AXH019)	△	AEDC W1A-C1A, CANARD CONTROL.	.000	3.000	.000	3.000	XREF 26.0000 IN.
							YREF .0000 IN.
							ZREF .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0
 (C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION REFERENCE INFORMATION

(BXH016) DATA NOT AVAILABLE SREF 19.6350 50. IN.

(BXH017) AEDC V41A-C1A, CANARD CONTROL, BNIC11 LREF 5.0000 IN.

(BXH018) DATA NOT AVAILABLE BREF 5.0000 IN.

(BXH019) DATA NOT AVAILABLE XPRP 26.0000 IN.

YPRP .0000 IN.

ZPRP .0000 IN.

SCALE .0000

DCND1 DCND2 DCND3 DCND4

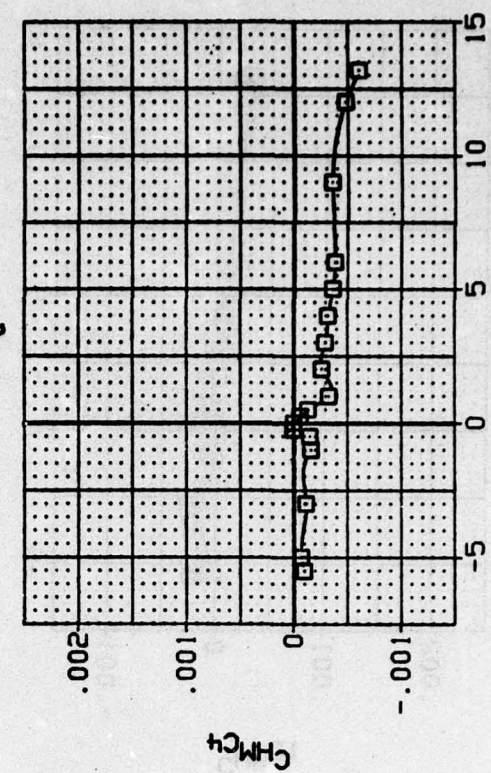
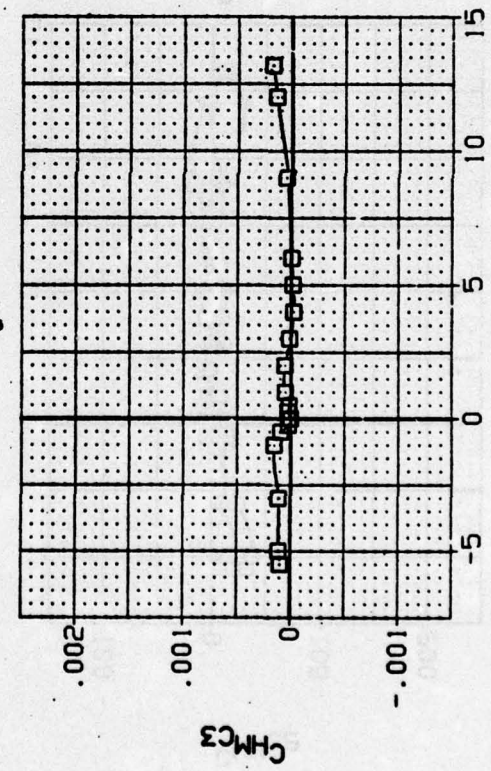
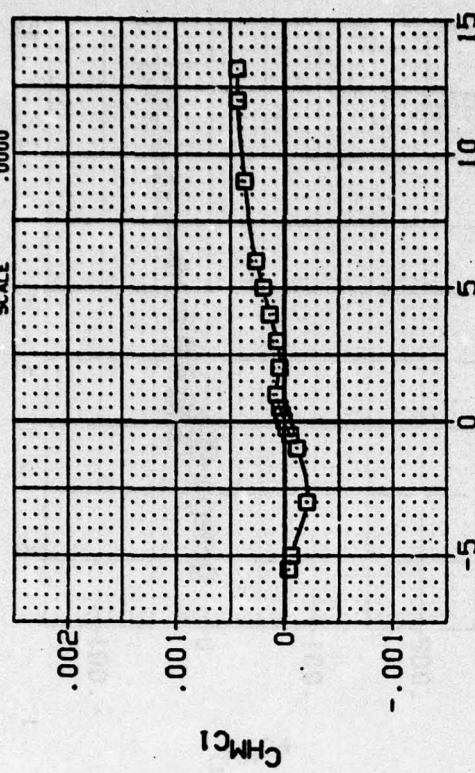
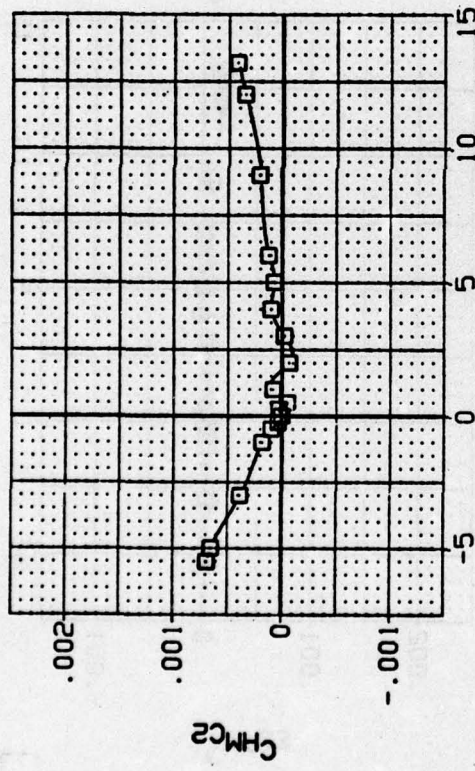
.000 -3.000 .000 -3.000

.000 .000 .000 .000

.000 .000 .000 .000

.000 1.000 .000 1.000

.000 3.000 .000 3.000



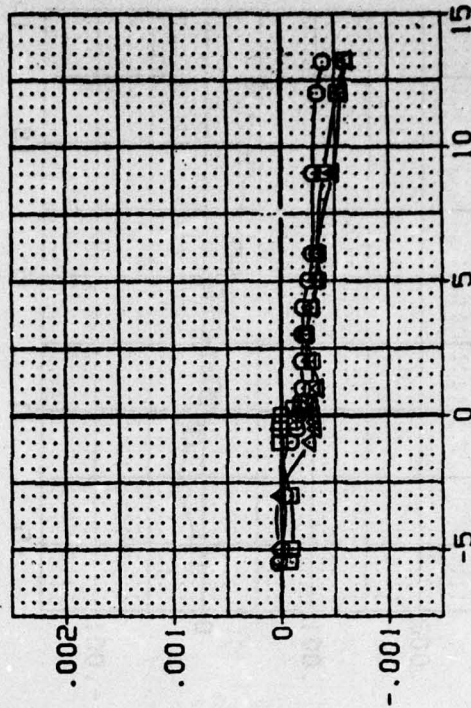
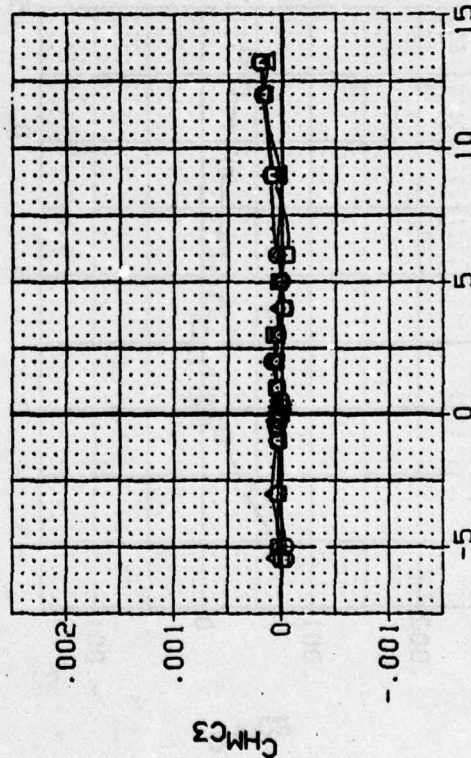
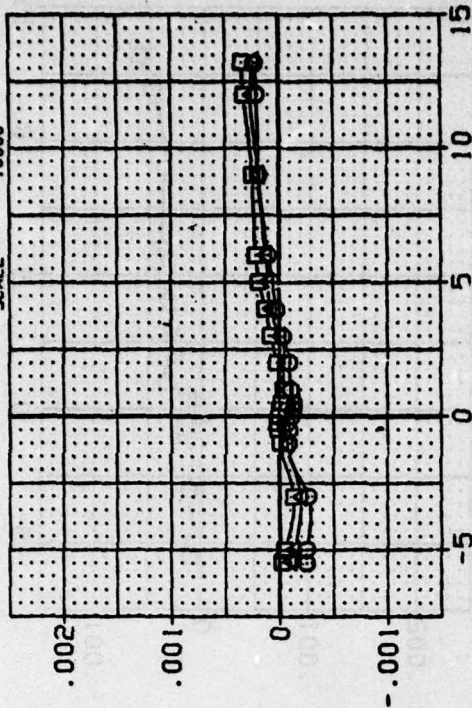
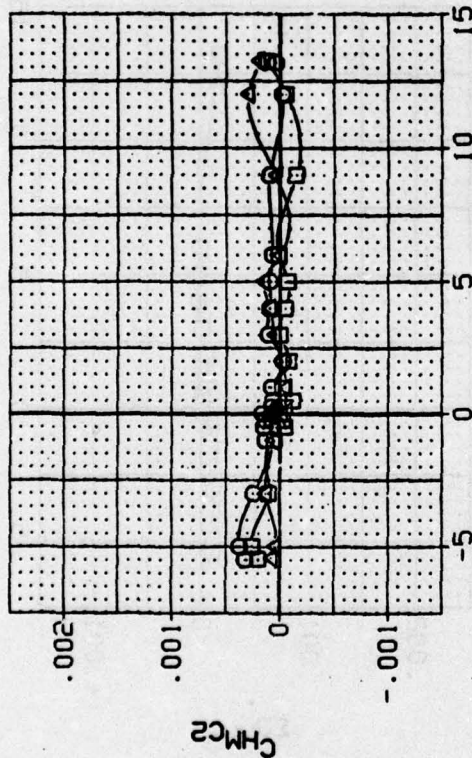
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=0$ $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL
(BXH016)
(BXH017)
(BXH018)
(BXH019)

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
DATA NOT AVAILABLE
AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
-3.000 -3.000 .000 -3.000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
SREF 5.0000 IN.
XREF 26.0000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0

(B) MACH = 3.00

DATA SET SYMBOL
(BX0016)
(BX0017)
(BX0018)
(BX0019)

CONFIGURATION DESCRIPTION
AEDC V4IA-CIA, CANARD CONTROL
AEDC V4IA-CIA, CANARD CONTROL
AEDC V4IA-CIA, CANARD CONTROL
AEDC V4IA-CIA, CANARD CONTROL

BNICITI
BNICITI
BNICITI
BNICITI

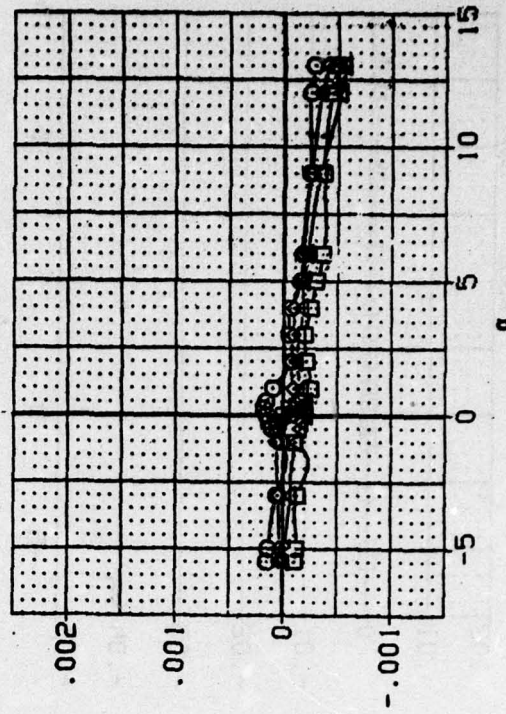
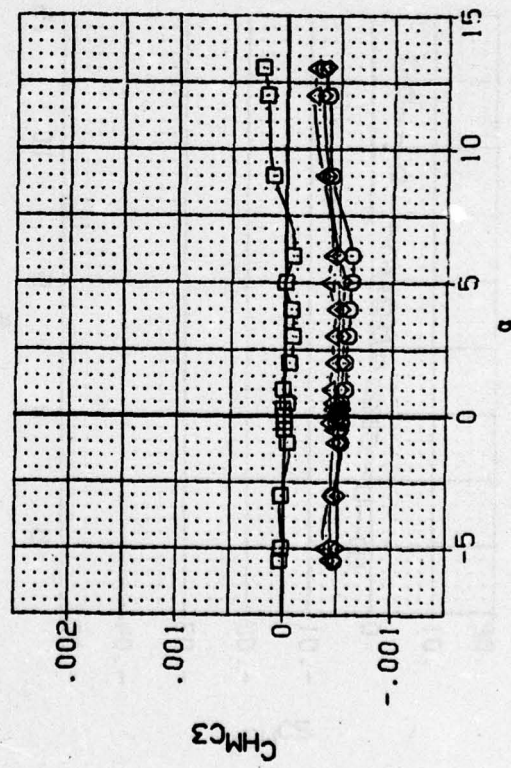
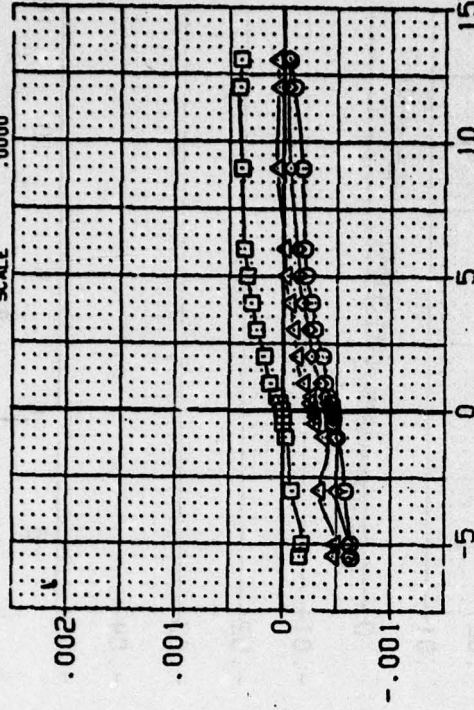
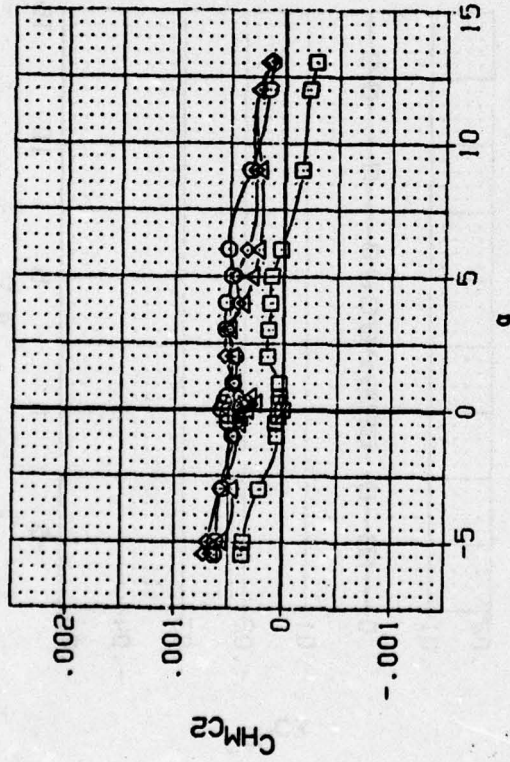
DCND1
-3.000
-3.000
-3.000
-3.000

DCND2
-3.000
-3.000
-3.000
-3.000

DCND3
-3.000
-3.000
-3.000
-3.000

DCND4
-3.000
-3.000
-3.000
-3.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0700 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0
(C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH016) DATA NOT AVAILABLE

(BXH017) AEDC V-1A-CIA, CANARD CONTROL, BNIC111

(BXH018) DATA NOT AVAILABLE

(BXH019) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4

.000 -3.000 .000 -3.000

.000 .000 .000 .000

.000 1.000 .000 1.000

.000 3.000 .000 3.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

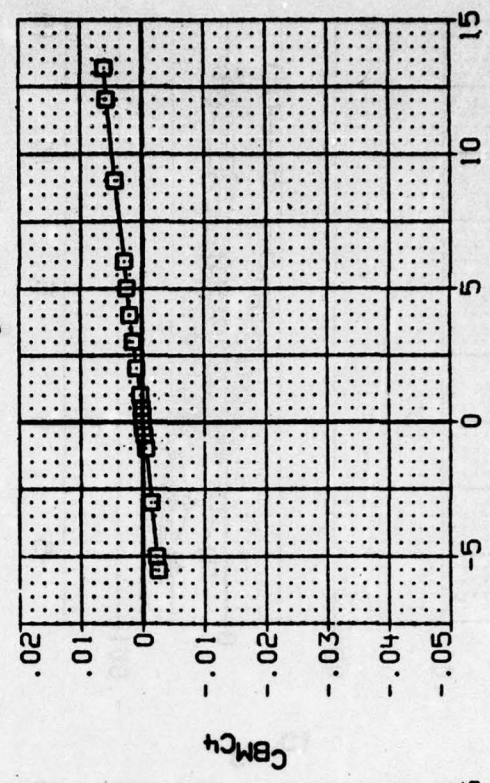
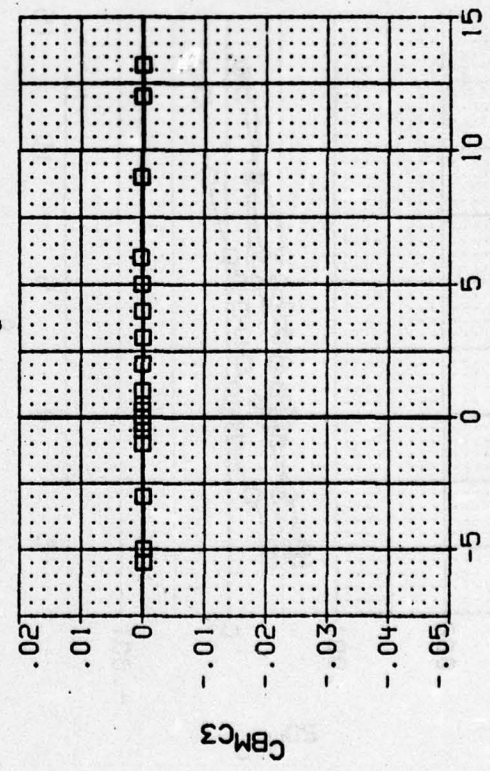
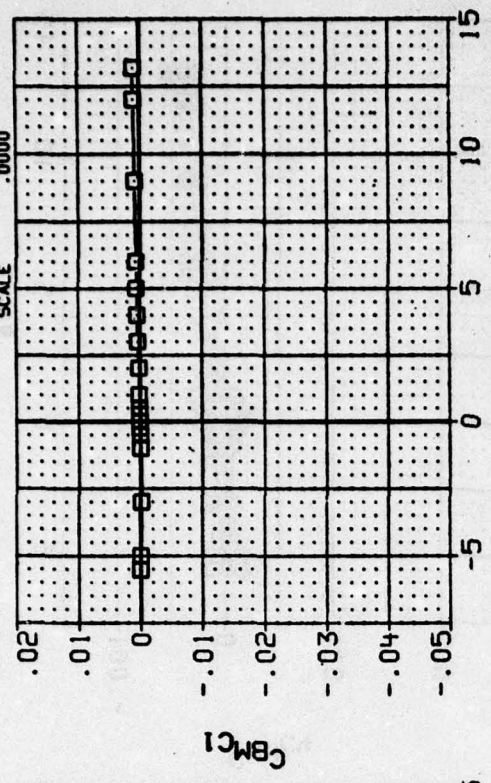
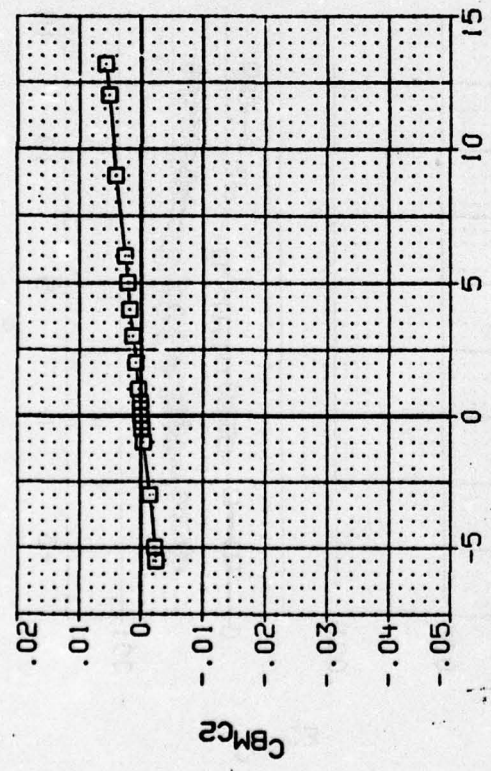
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(A) MACH = 2.50

DATA SET SYMBOL

(BX0016)
(BX0017)
(BX0018)
(BX0019)

CONFIGURATION DESCRIPTION

AEDC W1A-CIA, CANARD CONTROL, BNICITI
AEDC W1A-CIA, CANARD CONTROL, BNICITI
DATA NOT AVAILABLE
AEDC W1A-CIA, CANARD CONTROL, BNICITI

DCND1

-3.000
-3.000
1.000
3.000

DCND3

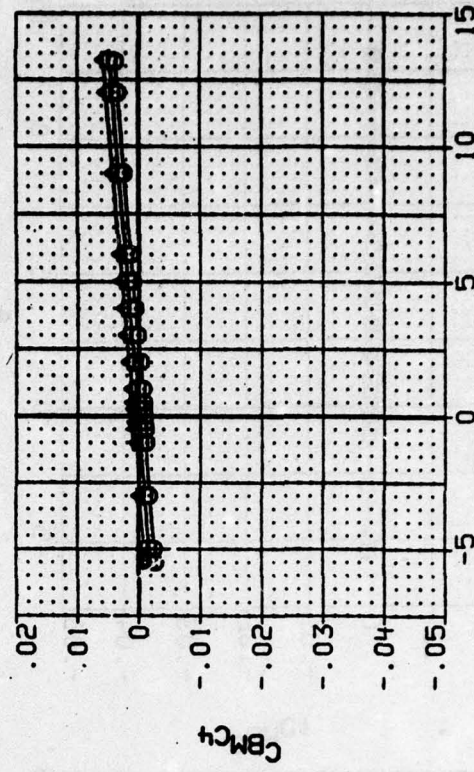
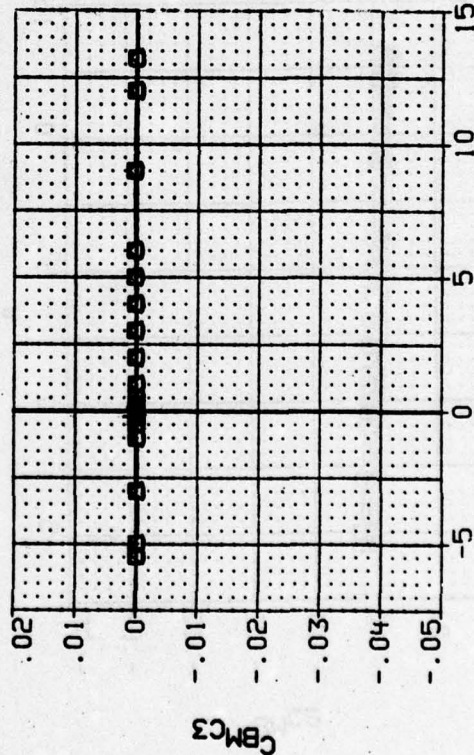
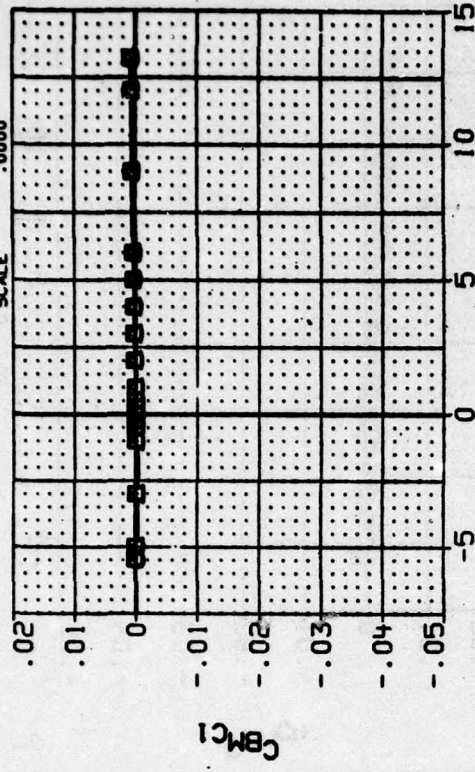
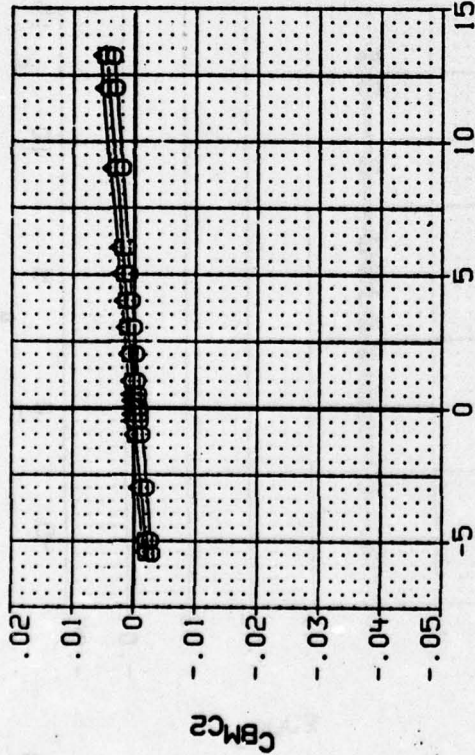
-3.000
-3.000
1.000
3.000

DCND4

-3.000
-3.000
1.000
3.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XREF 26.0000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0

(B)MACH = 3.00

DATA SET SYMBOL

(BXH016)
(BXH017)
(BXH018)
(BXH019)

CONFIGURATION DESCRIPTION

AEDC W1A-CIA, CANARD CONTROL, BNICITI
AEDC W1A-CIA, CANARD CONTROL, BNICITI
AEDC W1A-CIA, CANARD CONTROL, BNICITI
AEDC W1A-CIA, CANARD CONTROL, BNICITI

DCND1

.000
.000
.000
.000

DCND2

-3.000
.000
1.000
3.000

DCND3

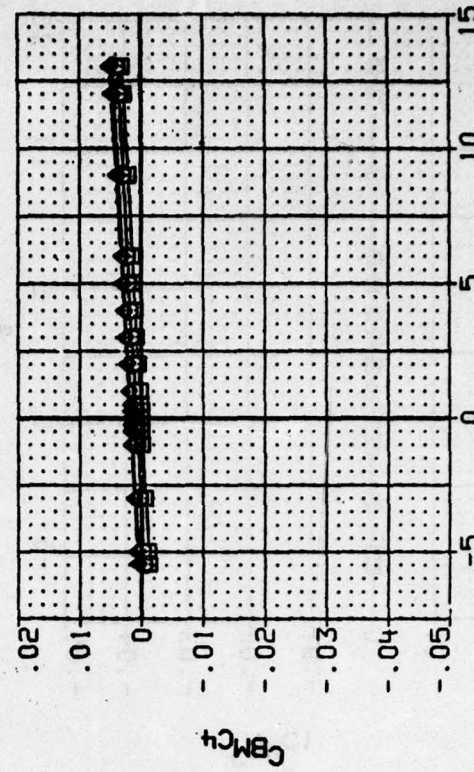
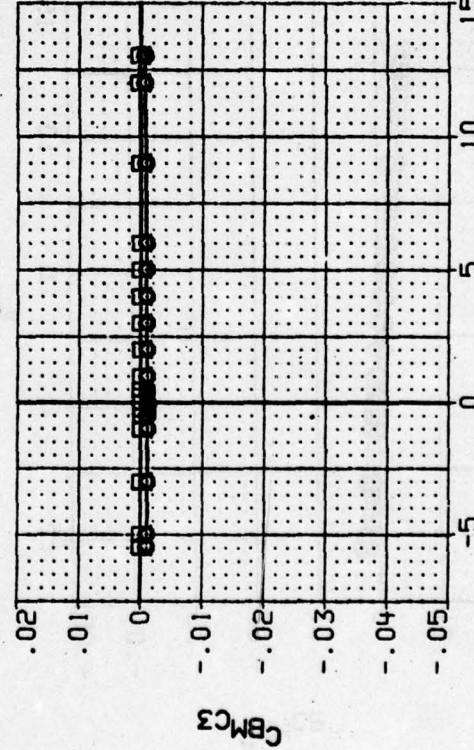
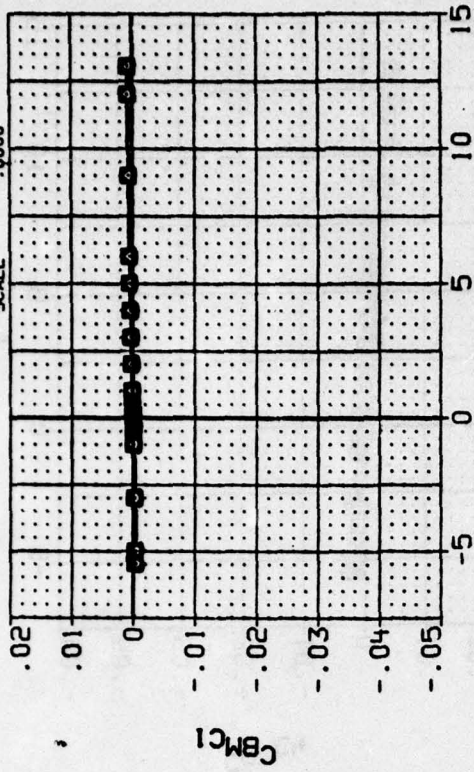
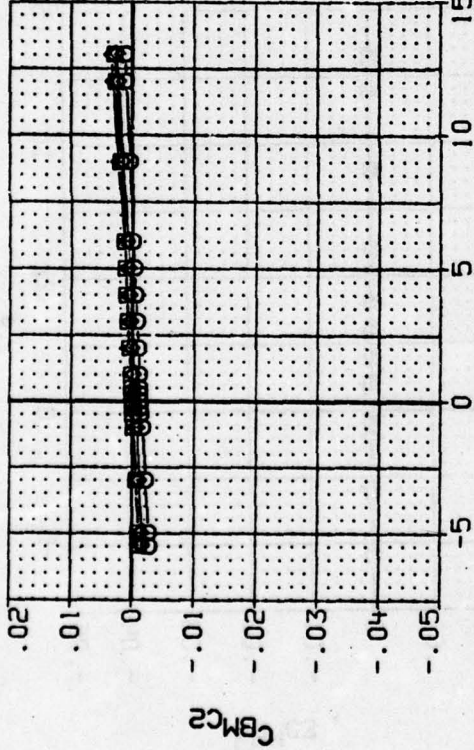
.000
.000
.000
.000

DCND4

-3.000
.000
1.000
3.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

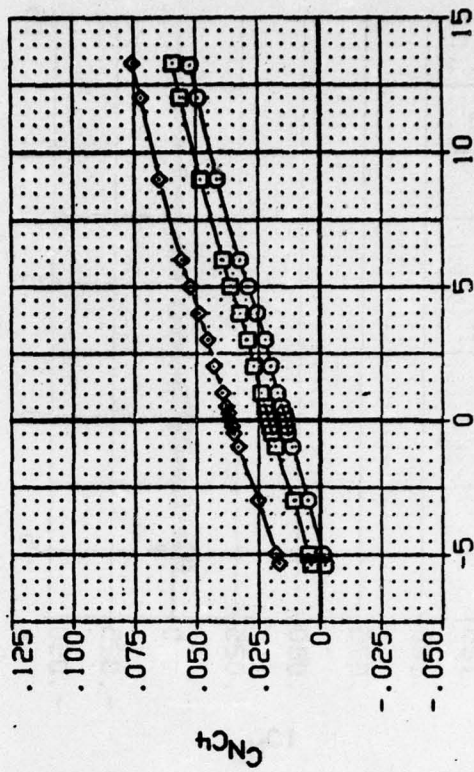
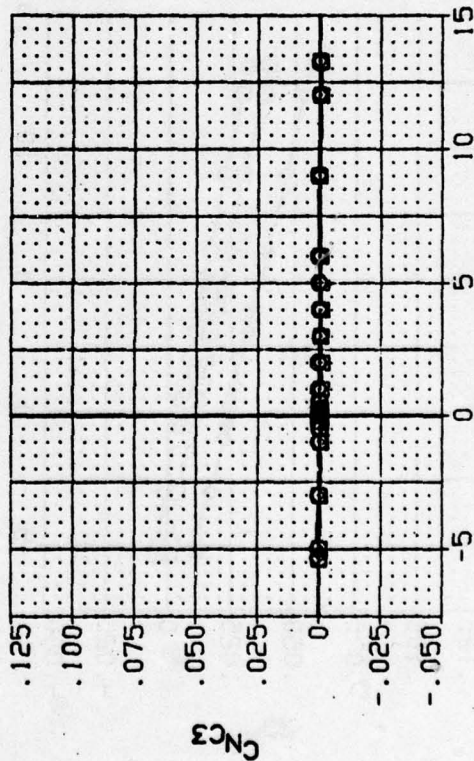
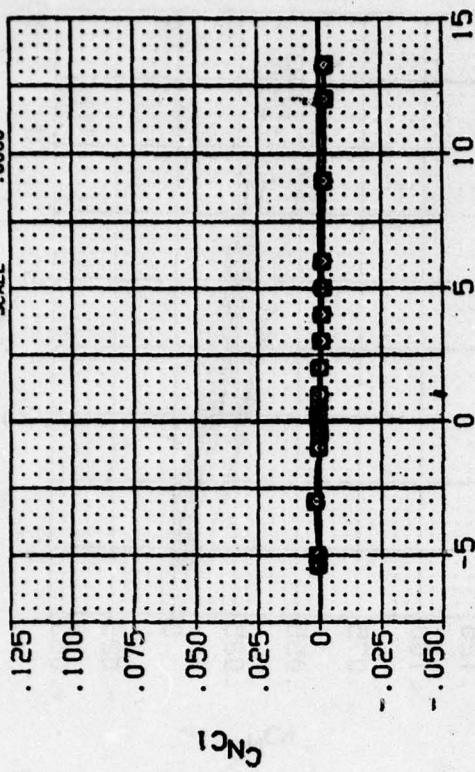
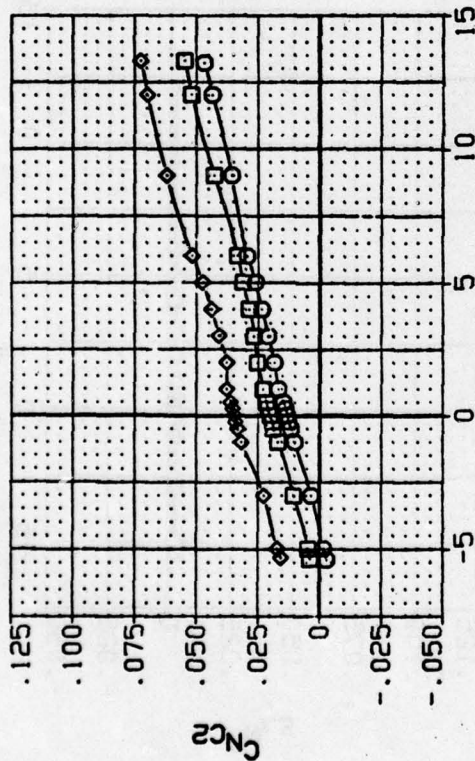
(C) MACH = 4.52

DATA SET SYMBOL
(AXH020) □
(AXH021) ○
(AXH022) ◇

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL, BNICITI
AEDC V41A-C1A, CANARD CONTROL, BNICITI
AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
.000 6.000 .000 6.000
.000 9.000 .000 9.000
.000 15.000 .000 15.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XREF 26.0000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000

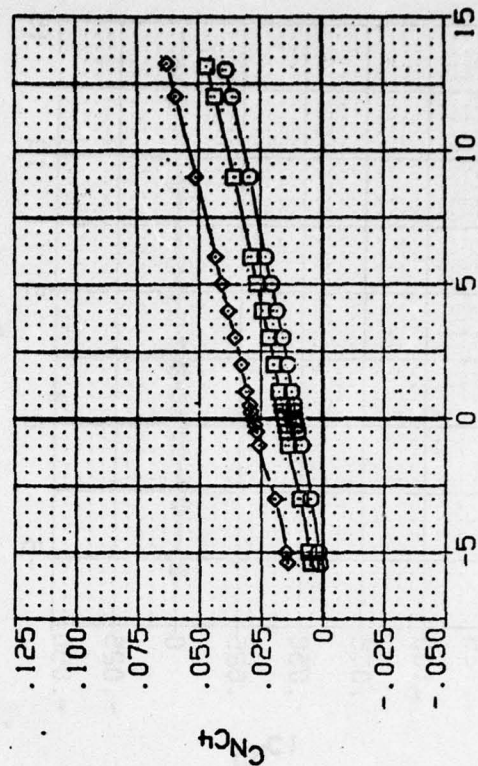
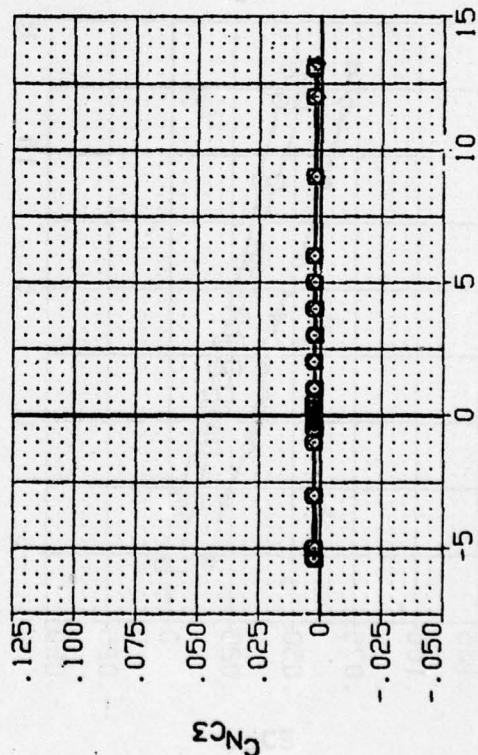
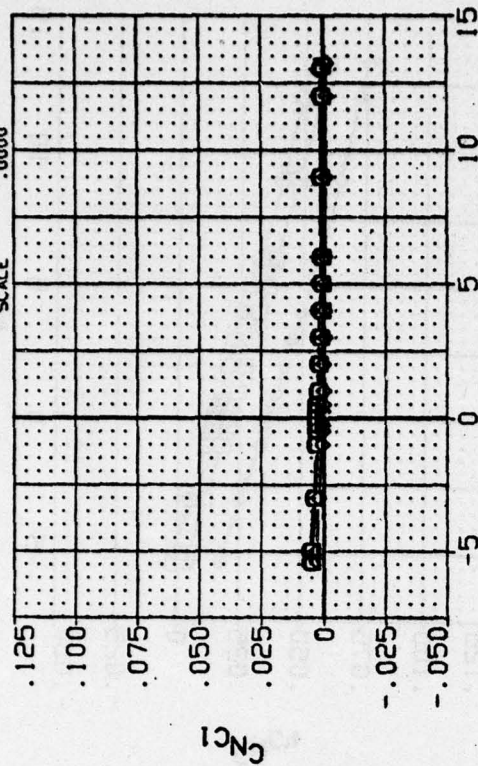
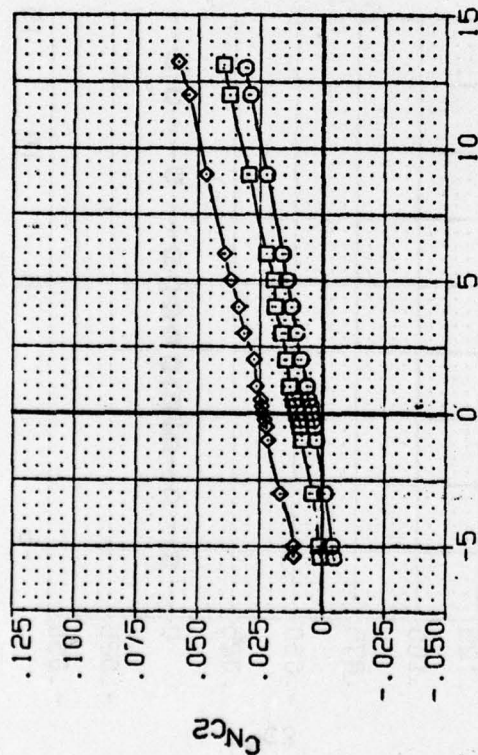


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHICAL=0 PHICND=0
(A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH020) □ AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH021) □ AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH022) ◇ AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



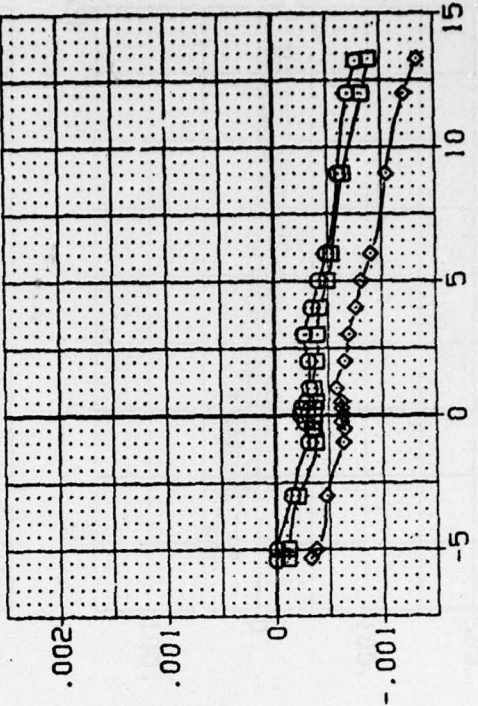
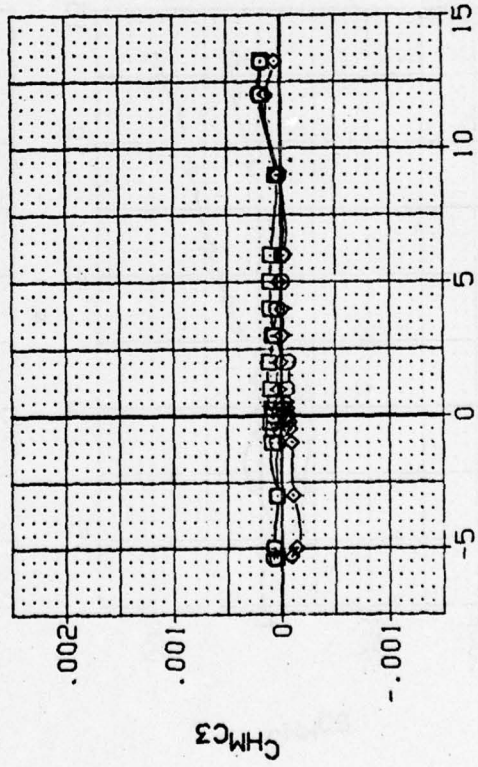
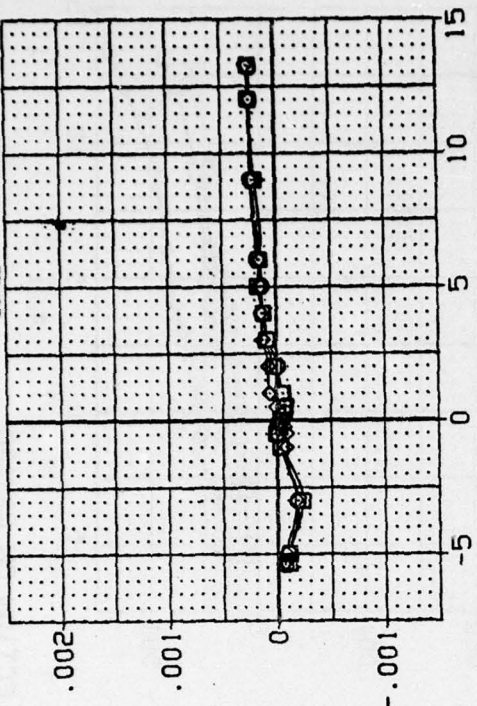
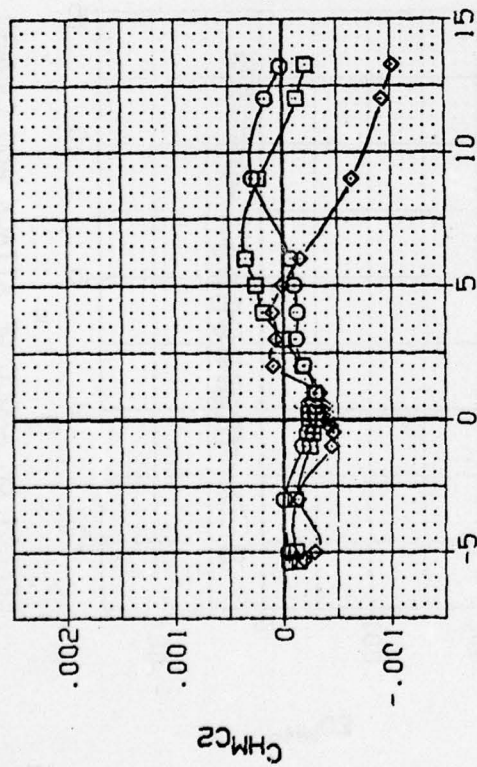
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL
(BXH020)
(BXH021)
(BXH022)

CONFIGURATION DESCRIPTION
AEDC V4A-C1A, CANARD CONTROL, BNIC11
AEDC V4A-C1A, CANARD CONTROL, BNIC11
AEDC V4A-C1A, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4
.000 6.000 .000 6.000
.000 9.000 .000 9.000
.000 15.000 .000 15.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
YMRP 26.0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHIAL=0 PHICND=0
(A) MACH = 3.00

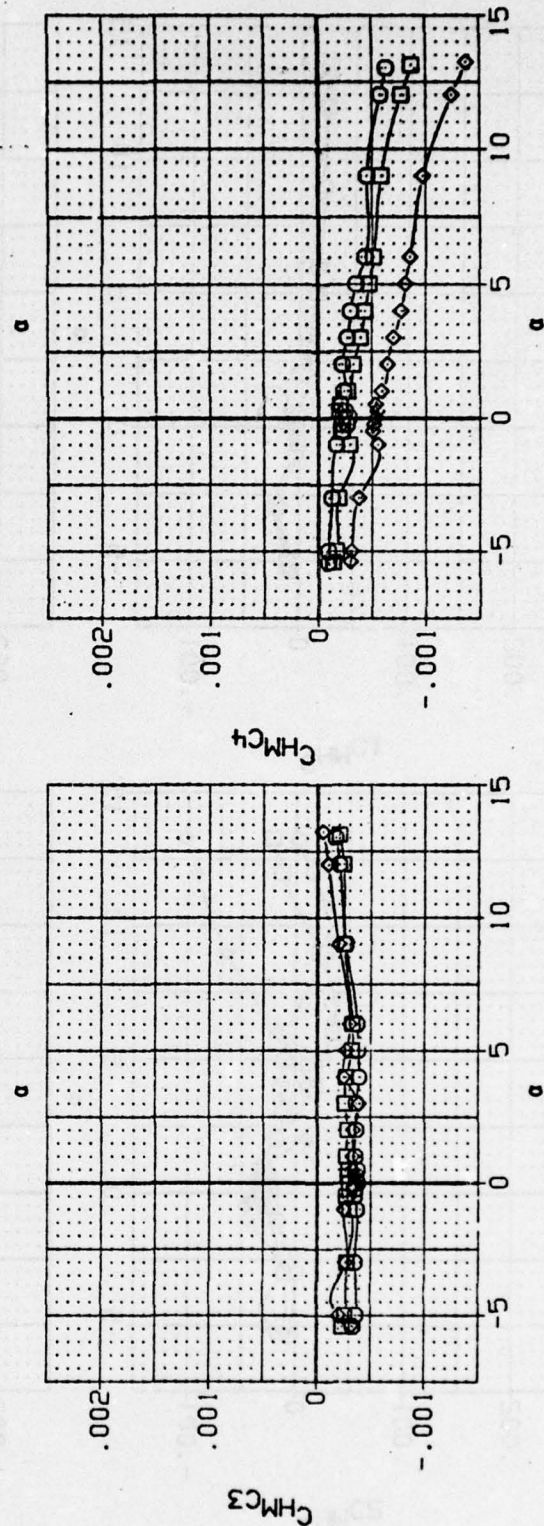
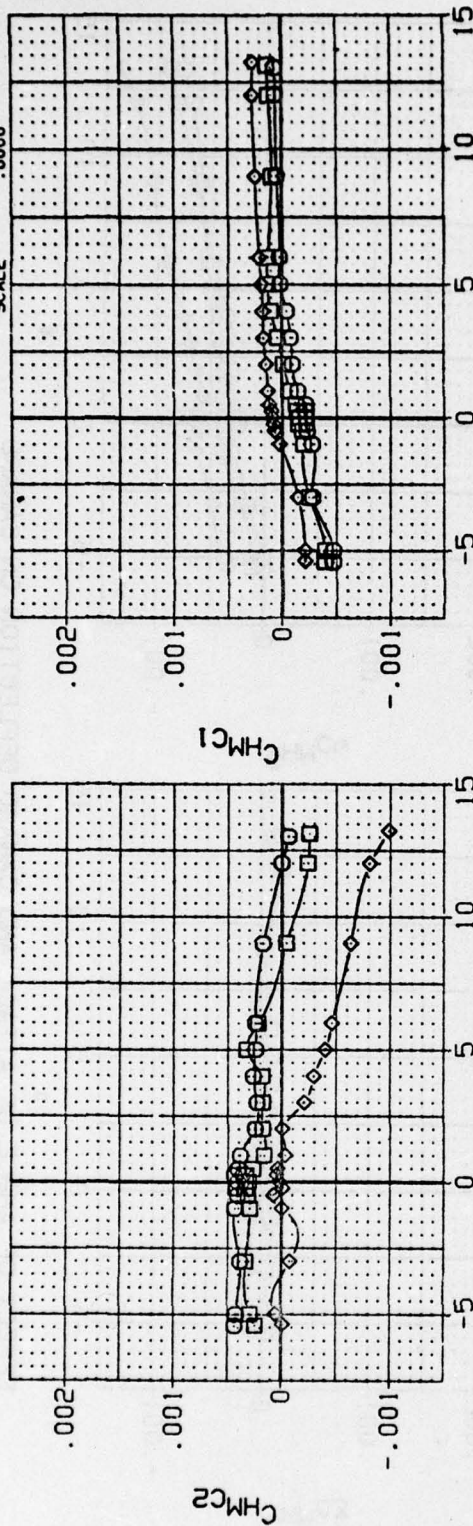
DATA SET SYMBOL
(BXH020)
(BXH021)
(BXH022)

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL.
AEDC V41A-C1A, CANARD CONTROL.
AEDC V41A-C1A, CANARD CONTROL.

BNICIT1
BNICIT1
BNICIT1

DCND1 DCND2 DCND3 DCND4
.000 6.000 .000 6.000
.000 9.000 .000 9.000
.000 15.000 .000 15.000

REFERENCE INFORMATION
SPREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000

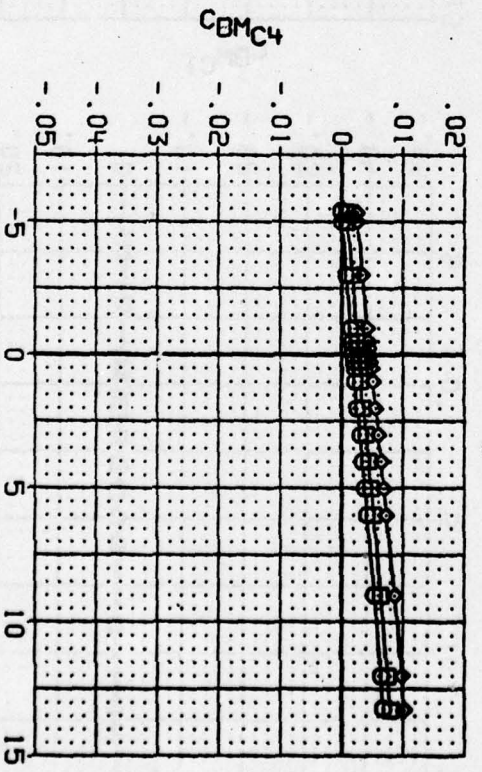
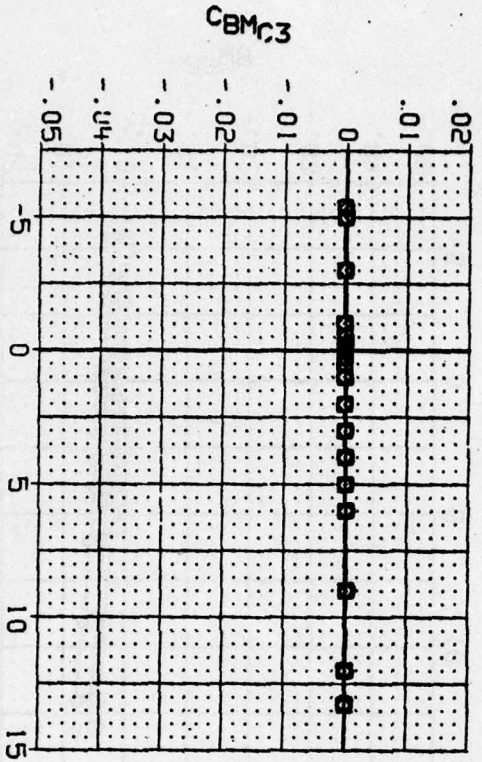
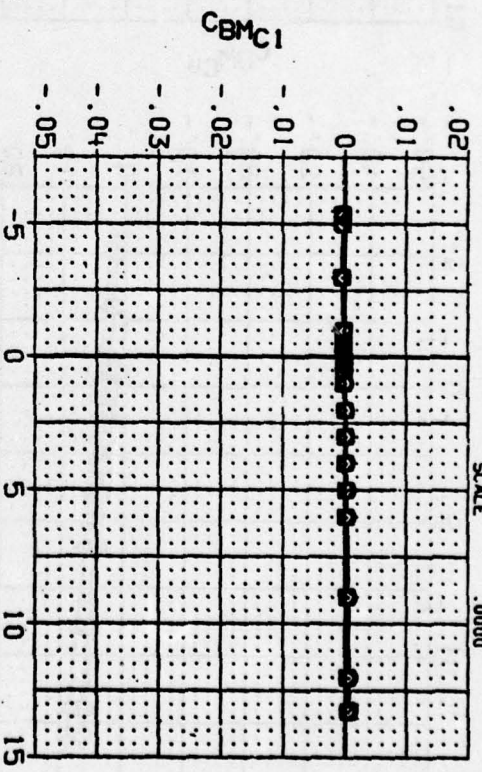
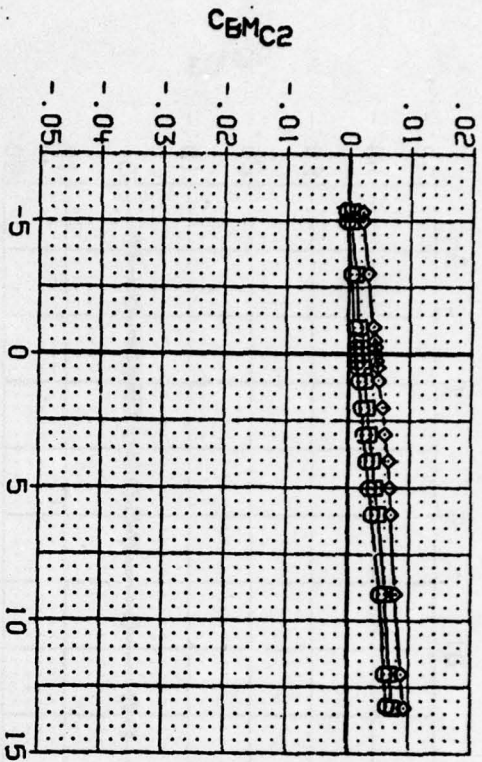


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHICAL=0 PHICND=0

(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (BXH021) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXH022) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 5.000 .000 5.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
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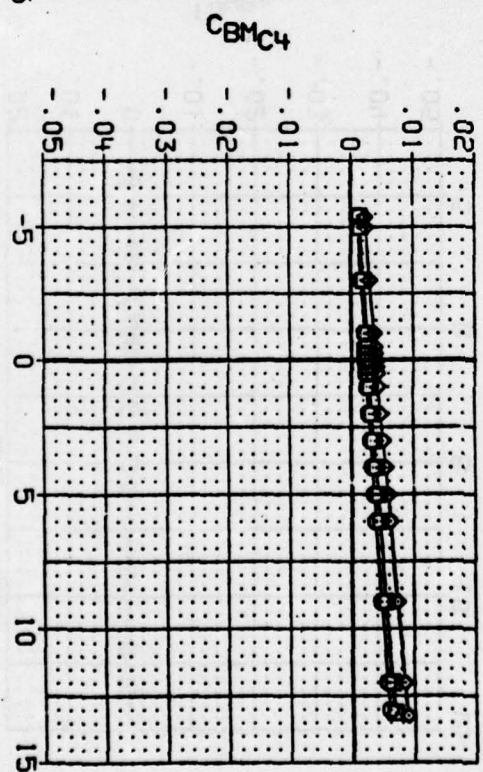
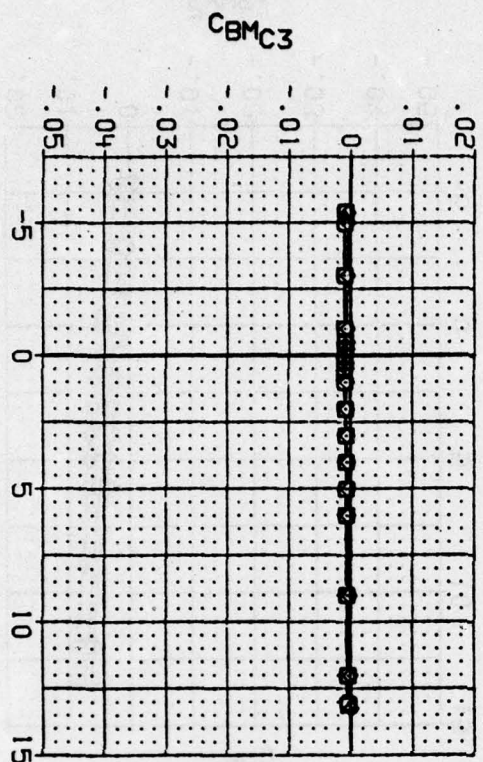
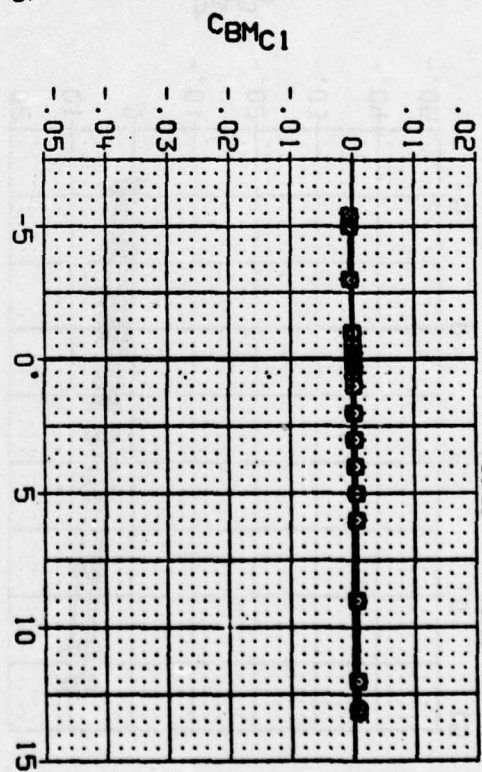
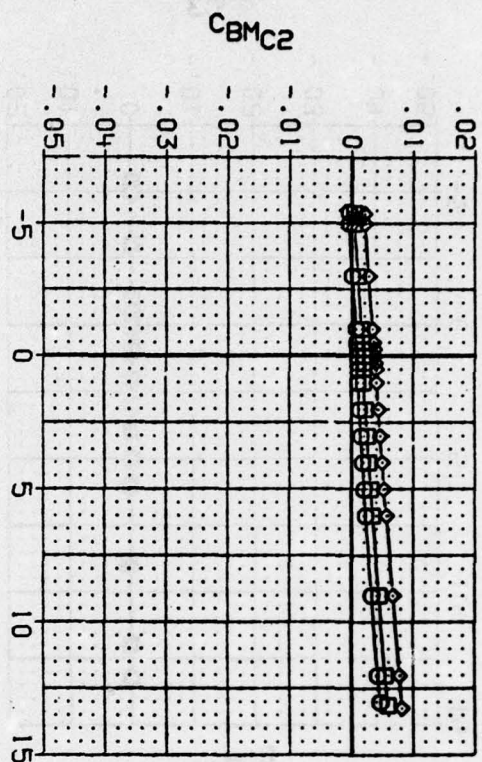


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{HITAL}=0$ $\Phi_{HICND}=0$
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH020) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (BXH021) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (BXH022) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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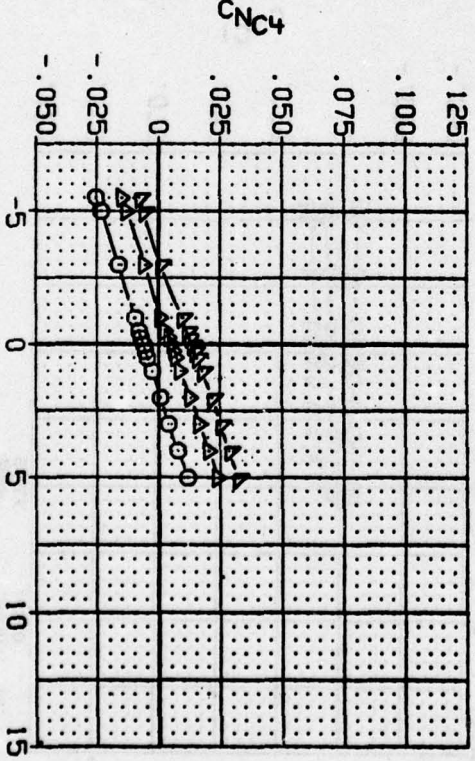
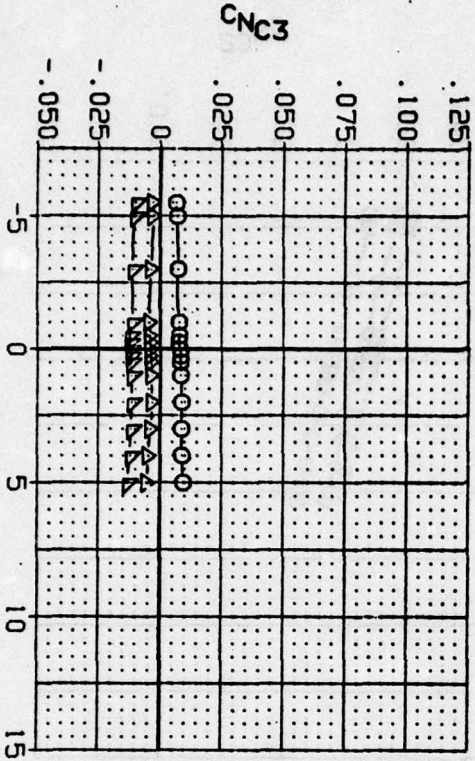
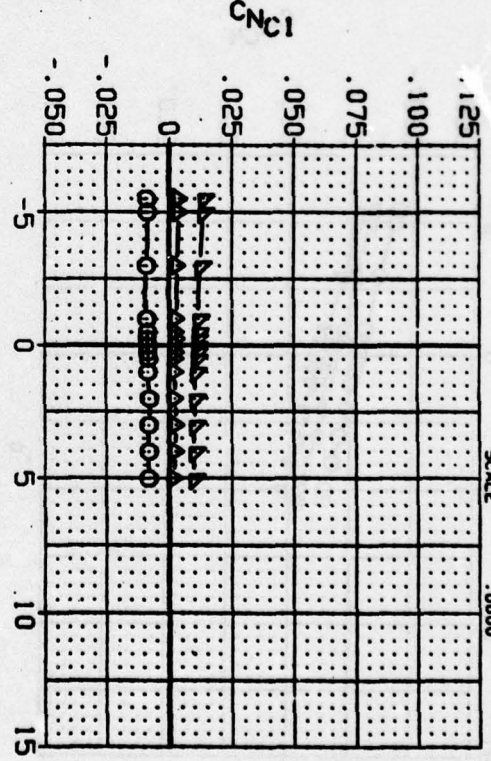
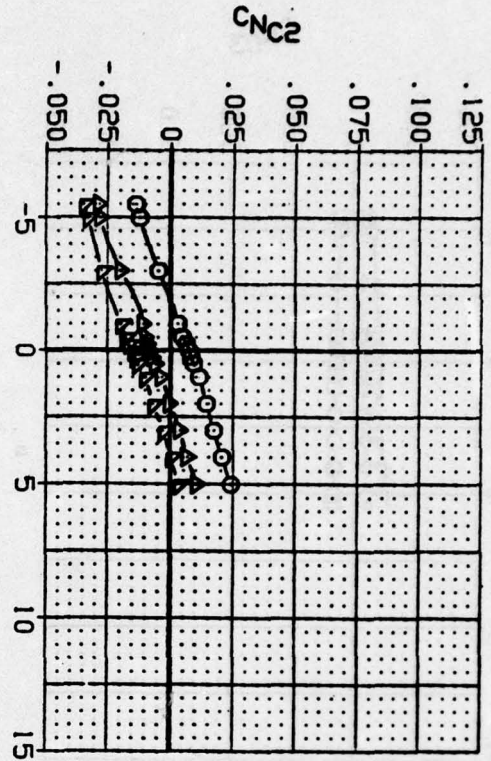


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=0$ $PHICND=0$
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH023) \square AEDC VAI-A-CIA, CANARD CONTROL, BNICITI
 (AXH024) \square DATA NOT AVAILABLE
 (AXH025) \square DATA NOT AVAILABLE
 (AXH026) \square AEDC VAI-A-CIA, CANARD CONTROL, BNICITI
 (AXH027) \square AEDC VAI-A-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 1.500 -1.500 -1.500 1.500
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
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 SCALE .0000

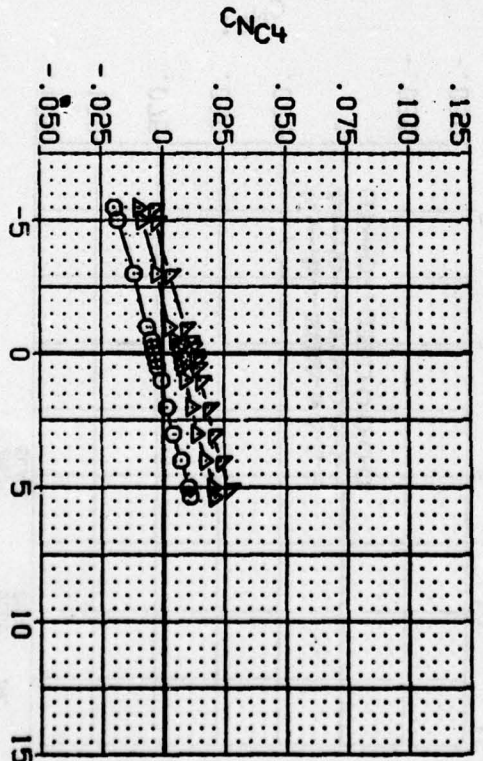
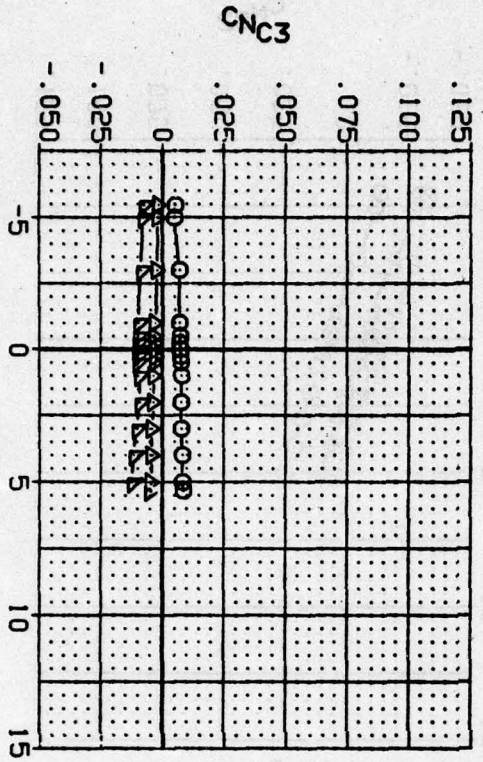
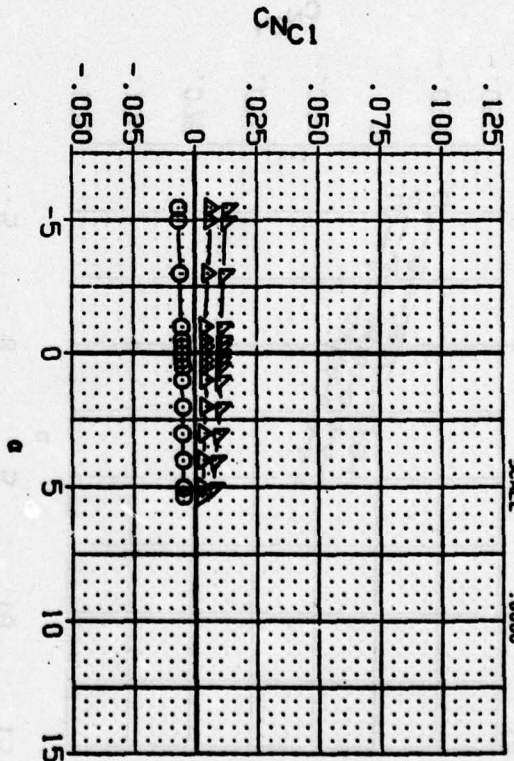
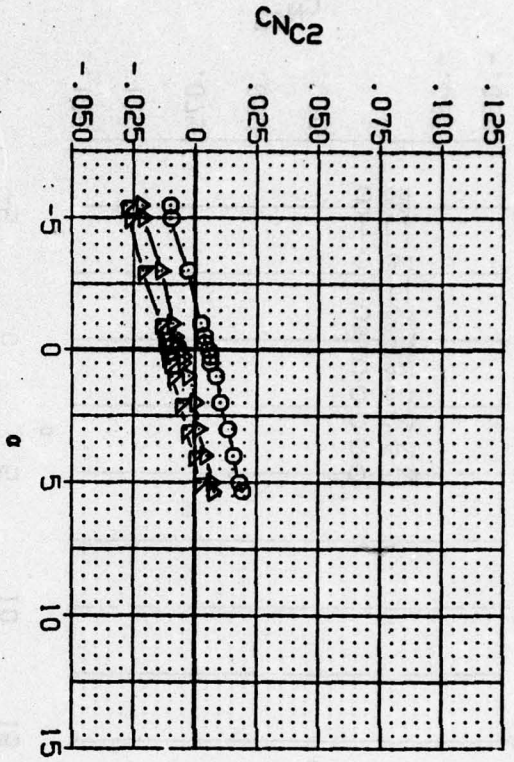


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (AXH024) \square DATA NOT AVAILABLE
 (AXH025) \square DATA NOT AVAILABLE
 (AXH026) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH027) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DOAND1 DOAND2 DOAND3 DOAND4
 -3.000 3.000 3.000 -3.000
 -5.000 -5.000 -5.000 5.000
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION:
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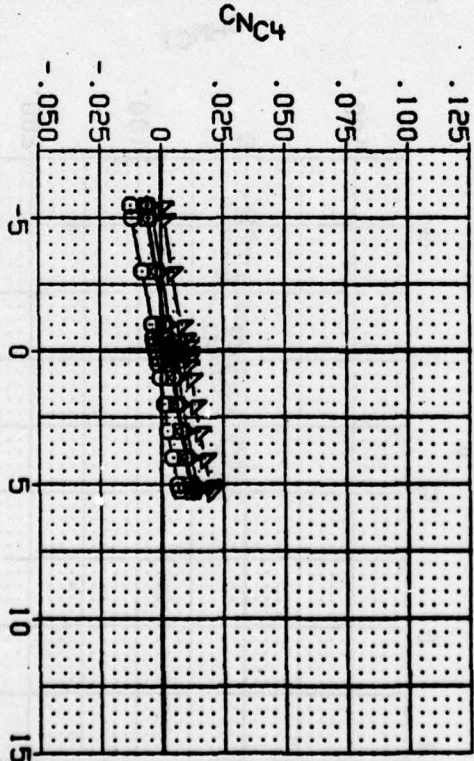
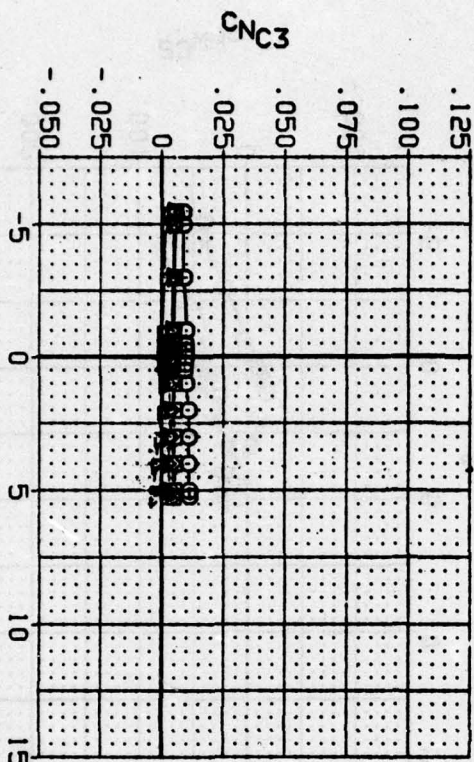
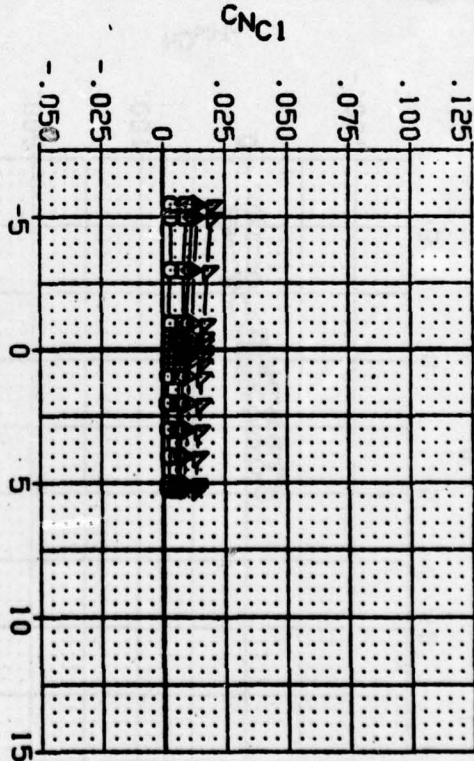
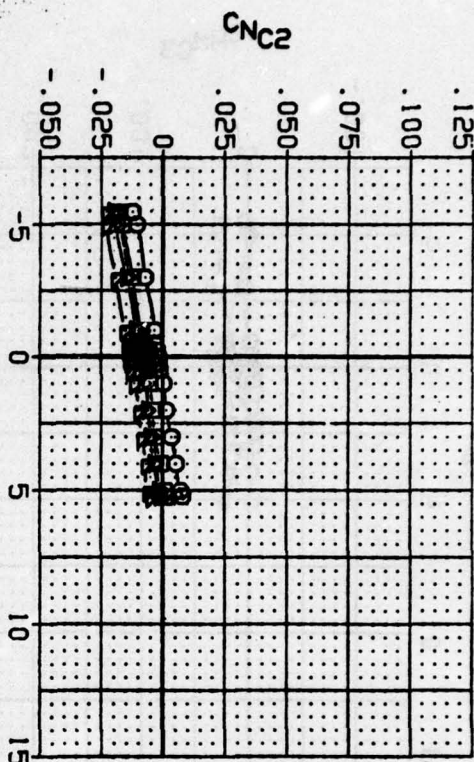
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0
 (B)MACH = 3.00

PAGE 70

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(AXH023)	○	AEDC W1A-C1A, CANARD CONTROL, BRICITI
(AXH024)	○	AEDC W1A-C1A, CANARD CONTROL, BRICITI
(AXH025)	○	AEDC W1A-C1A, CANARD CONTROL, BRICITI
(AXH026)	○	AEDC W1A-C1A, CANARD CONTROL, BRICITI
(AXH027)	○	AEDC W1A-C1A, CANARD CONTROL, BRICITI

DCND1	DCND2	DCND3	DCND4
-3.000	3.000	3.000	-3.000
.500	-.500	-.500	.500
1.000	-1.000	-1.000	1.000
2.000	-2.000	-2.000	2.000
5.000	-5.000	-5.000	5.000

REFERENCE INFORMATION	
SREF	19.6350
LBREF	5.0000
BRREF	5.0000
XRREF	25.0000
YRREF	.0000
SCALE	.0000
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IN.	IN.
IN.	IN.
IN.	IN.

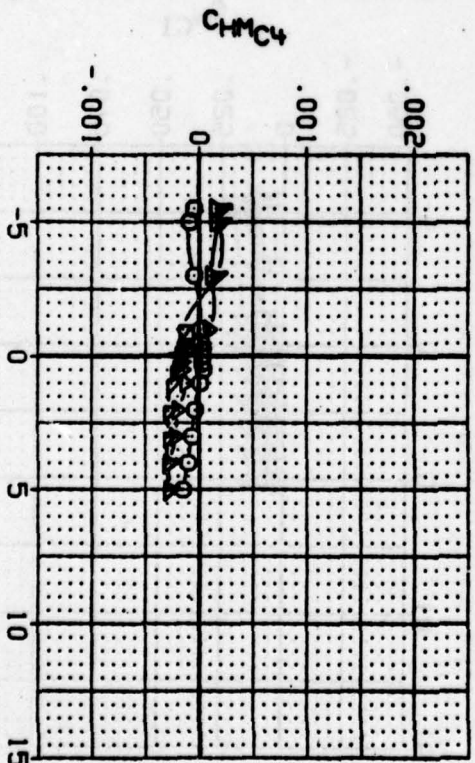
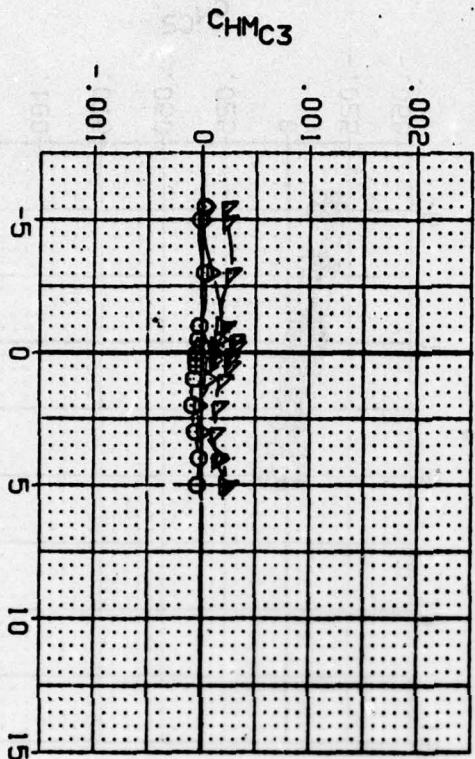
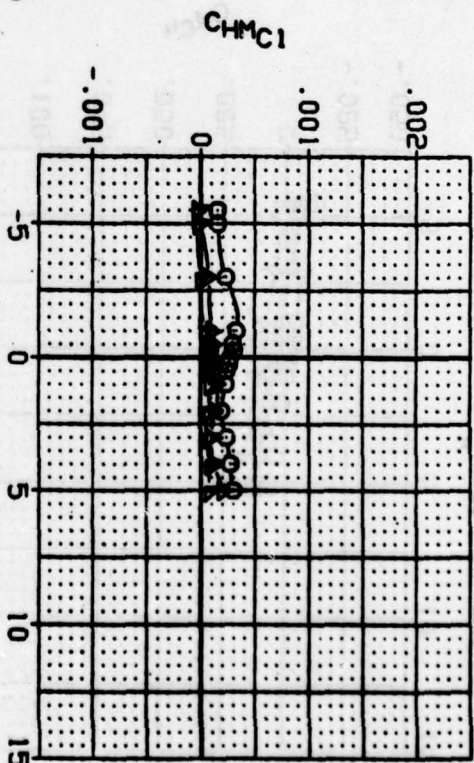
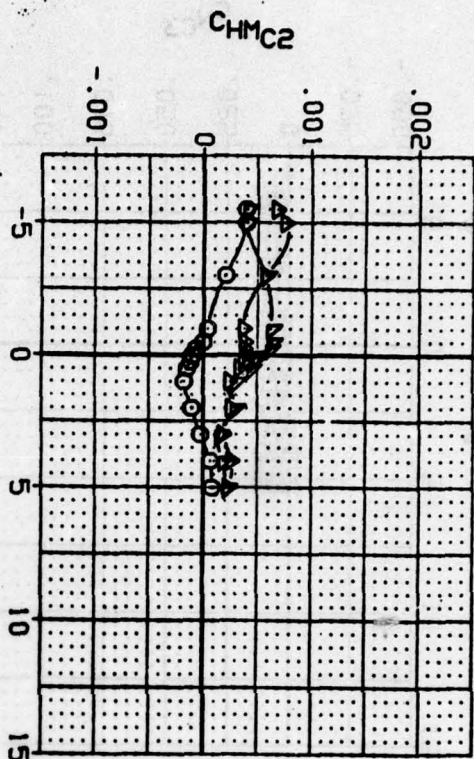


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0
 (C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (BX4024) ☐ DATA NOT AVAILABLE
 (BX4025) ☐ DATA NOT AVAILABLE
 (BX4026) ☐ AEDC WJIA-CIA, CANARD CONTROL, BNICITI
 (BX4027) ☒ AEDC WJIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 -1.500 -1.500 -1.500 -1.500
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
 SREF 19.6356 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXHQ23) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(BXHQ24) \diamond DATA NOT AVAILABLE

(BXHQ25) \triangle DATA NOT AVAILABLE

(BXHQ26) \circ AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(BXHQ27) ∇ AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 3.000 -3.000

-1.500 -1.500 -1.500 1.500

1.000 -1.000 -1.000 1.000

2.000 -2.000 -2.000 2.000

5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

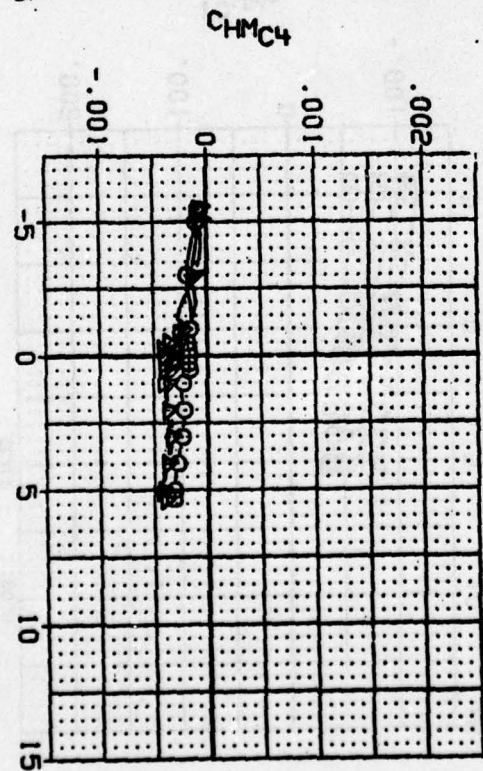
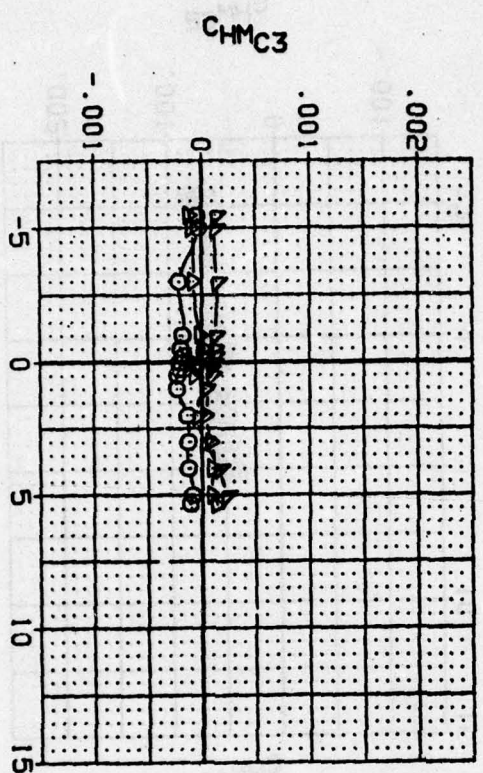
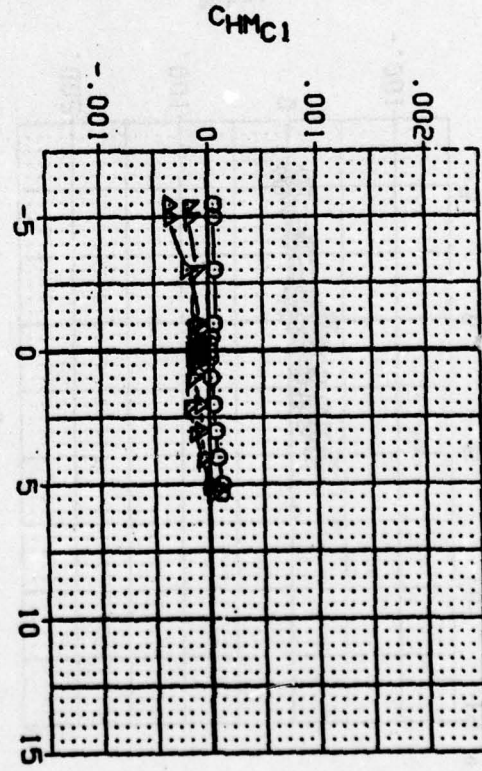
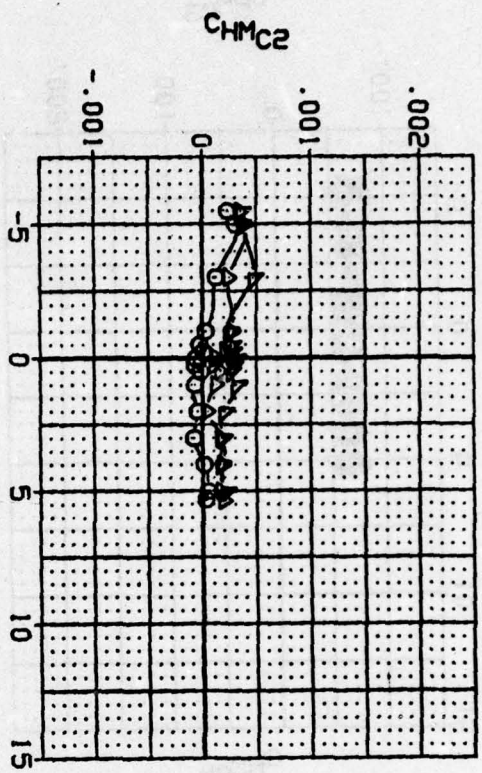
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YREF .0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHIAL=0 PHICND=0

(B)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

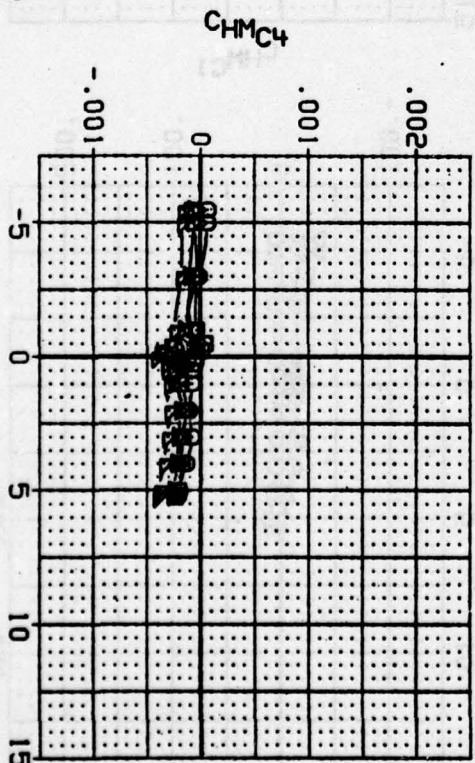
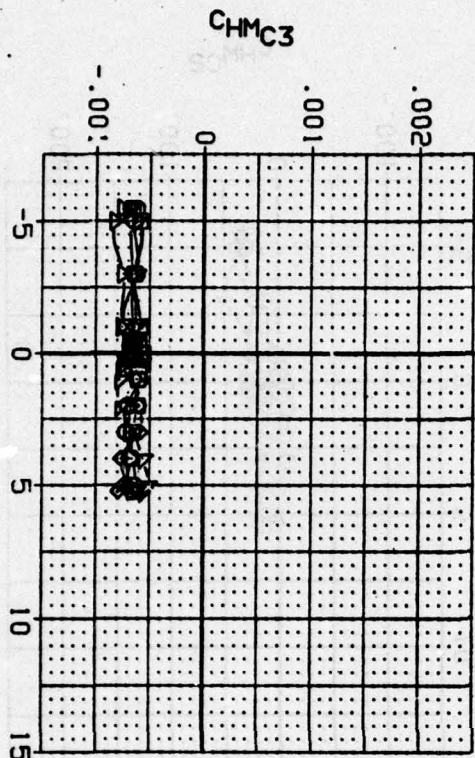
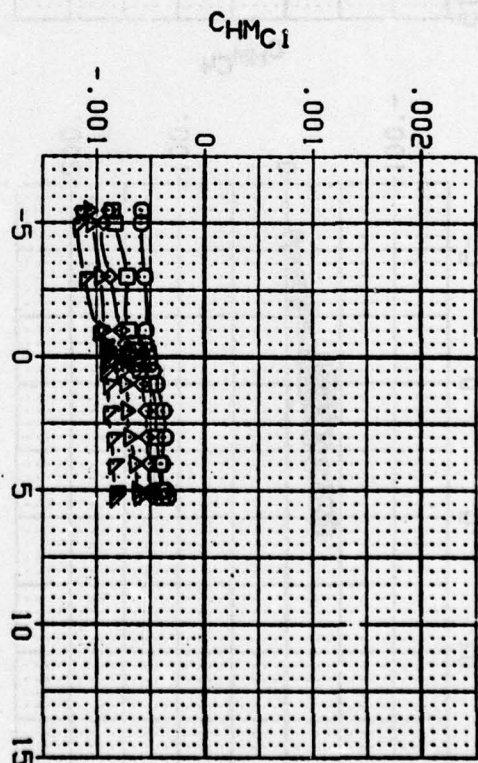
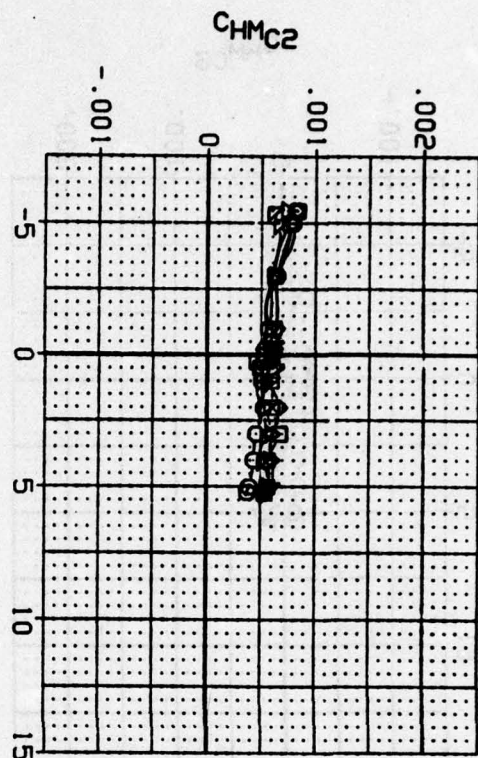
(BXH023) AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH024) AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH025) AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH026) AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH027) AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 -3.000 -3.000
 -1.500 -1.500 -1.500 -1.500
 1.000 -1.000 -1.000 -1.000
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 5.000 -5.000 -5.000 -5.000

REFERENCE INFORMATION

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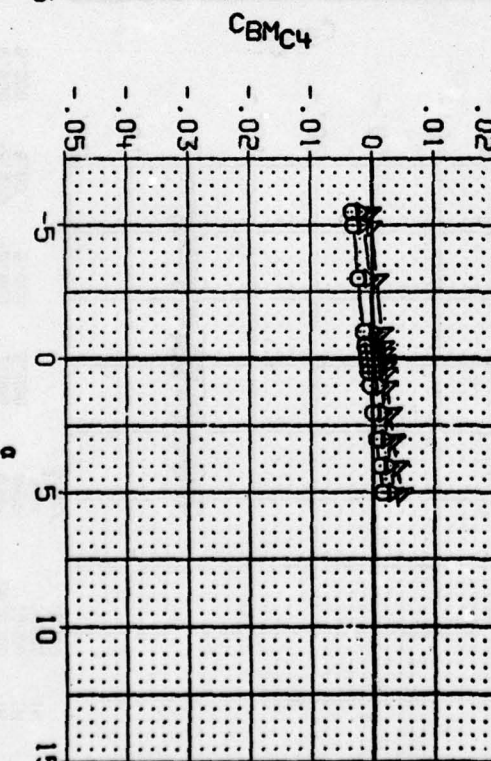


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(C)MACH # 4.52

REFERENCE INFORMATION				
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-500	-500	-500	5.000	LREF 5.0000 IN.
1.000	-1.000	1.000	2.000	BREF 5.0000 IN.
2.000	-2.000	-2.000	5.000	MREF 26.0000 IN.
5.000	-5.000	-5.000		YREF .0000 IN.



DATA SET SYMBOL CONFIGURATION DESCRIPTION

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(BXH024) DATA NOT AVAILABLE

(BXH025) DATA NOT AVAILABLE

(BXH026) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

(BXH027) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 3.000 -3.000

-1.500 -1.500 -1.500 -1.500

1.000 1.000 1.000 1.000

5.000 5.000 5.000 5.000

REFERENCE INFORMATION

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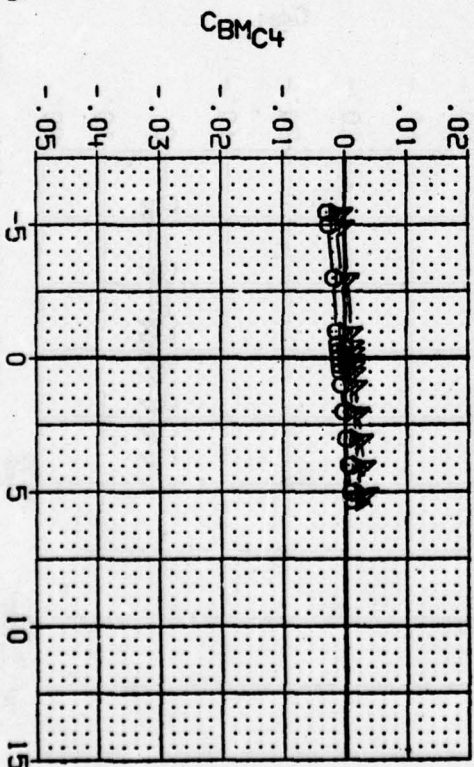
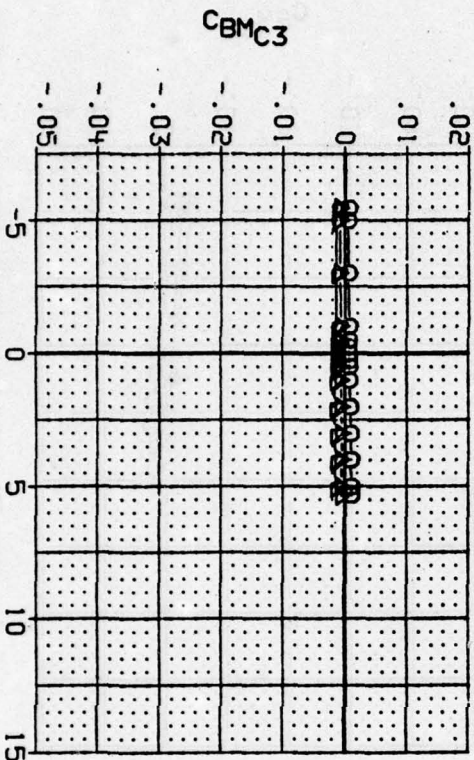
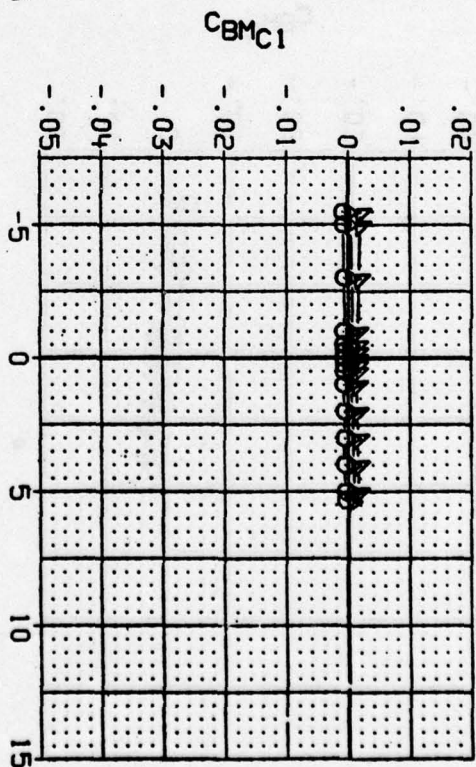
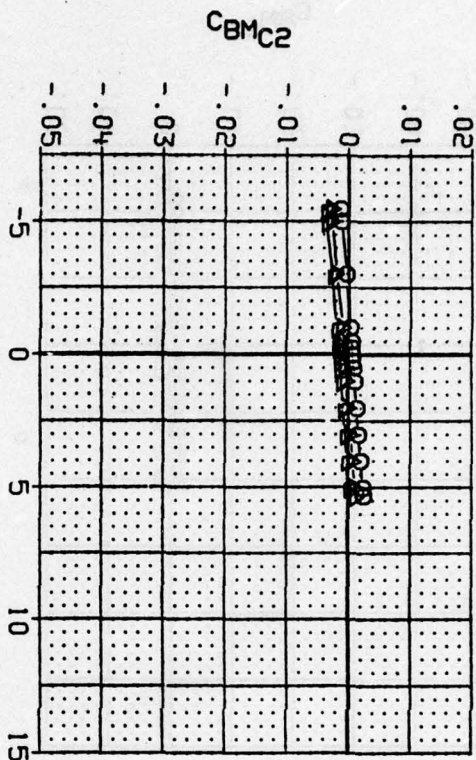
LREF 5.0000 IN.

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YREF 25.0000 IN.

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SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(B)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

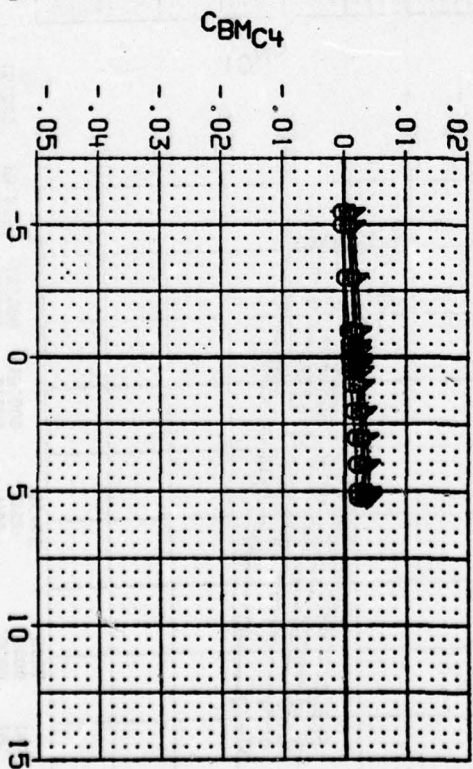
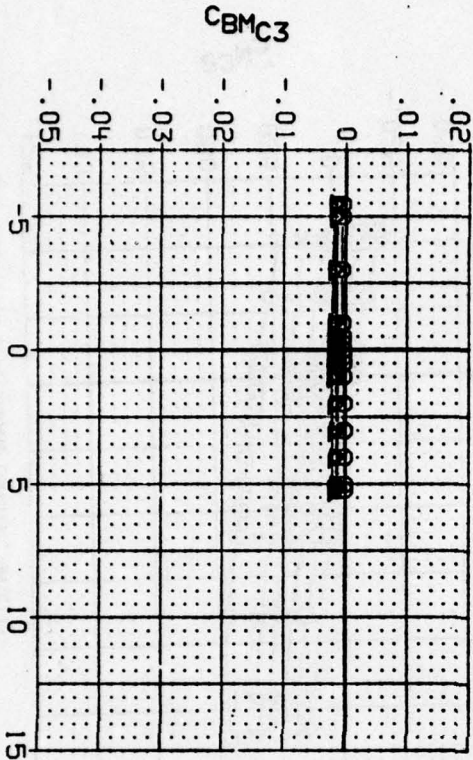
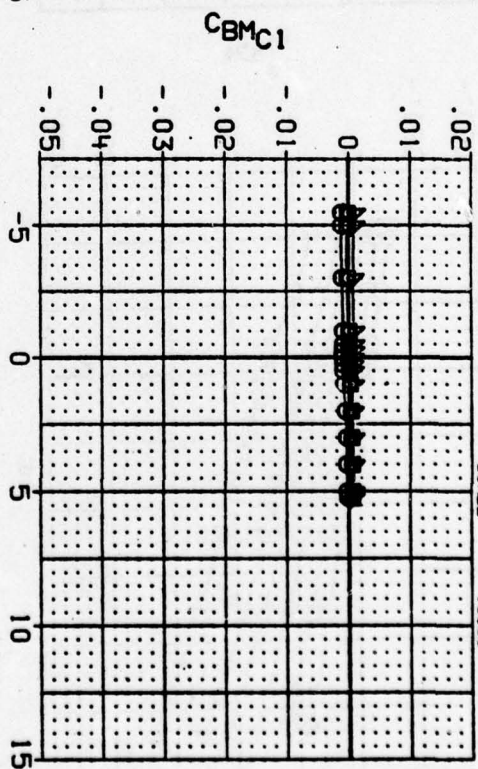
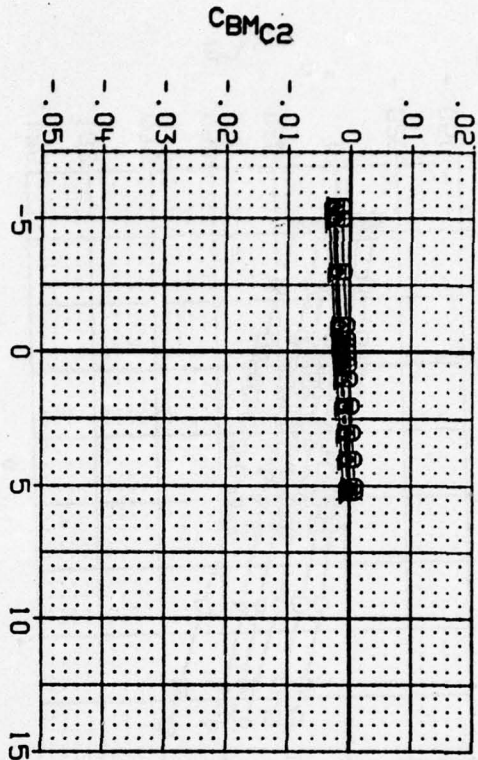
(BXH023) AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (BXH024) AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (BXH025) AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (BXH026) AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (BXH027) AEDC V41A-C1A, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 -3.000 -3.000
 -5.000 -5.000 -5.000 -5.000
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YREF 26.0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

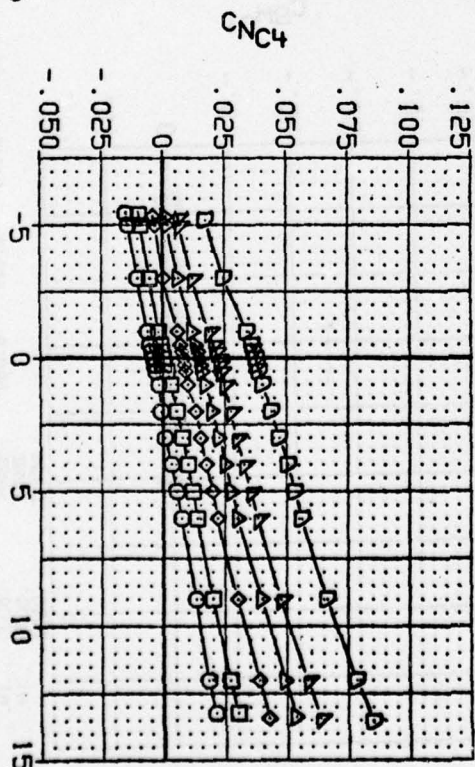
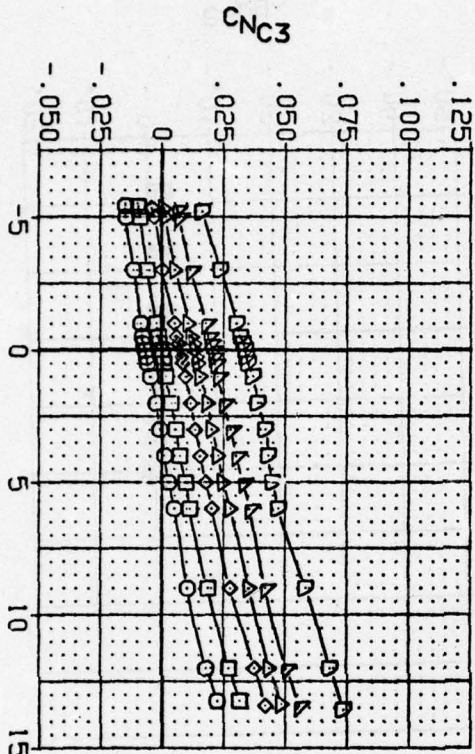
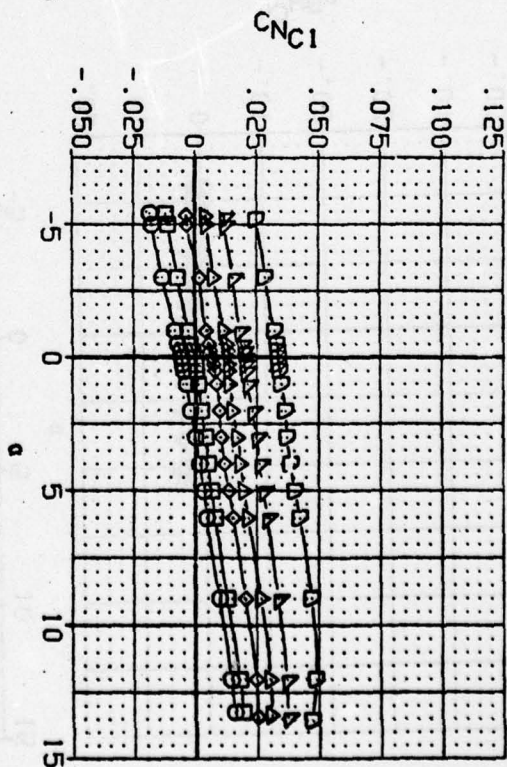
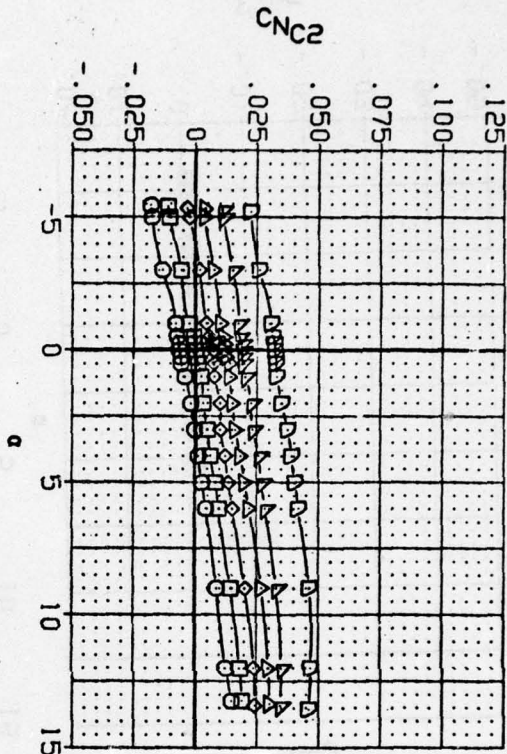
(AXH0281) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH0291) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH0301) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH0311) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH0321) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH0331) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

REFERENCE INFORMATION

DCND1 DCND2 DCND3 DCND4
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 3.000 3.000 3.000 3.000
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 15.000 15.000 15.000 15.000

SREF 19.6350 SQ. IN.
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SCALE .0000



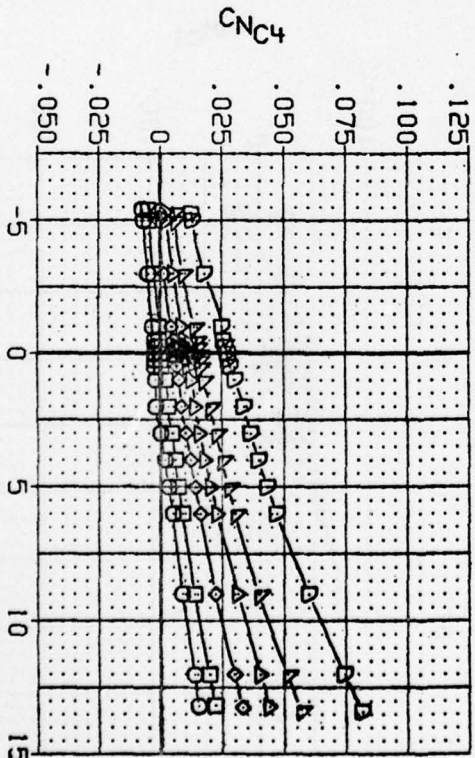
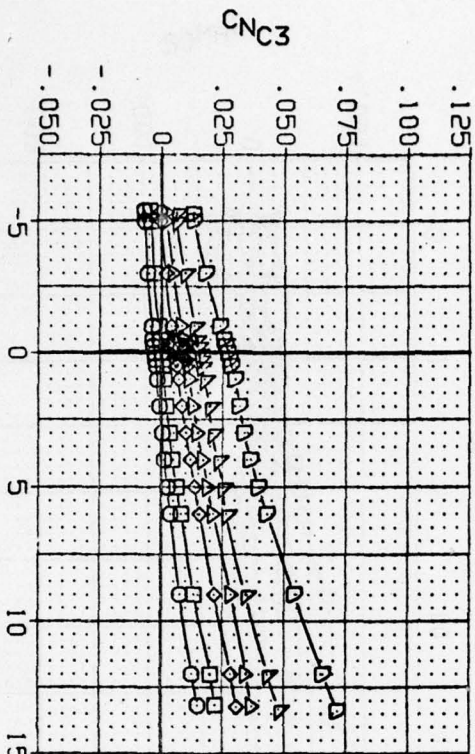
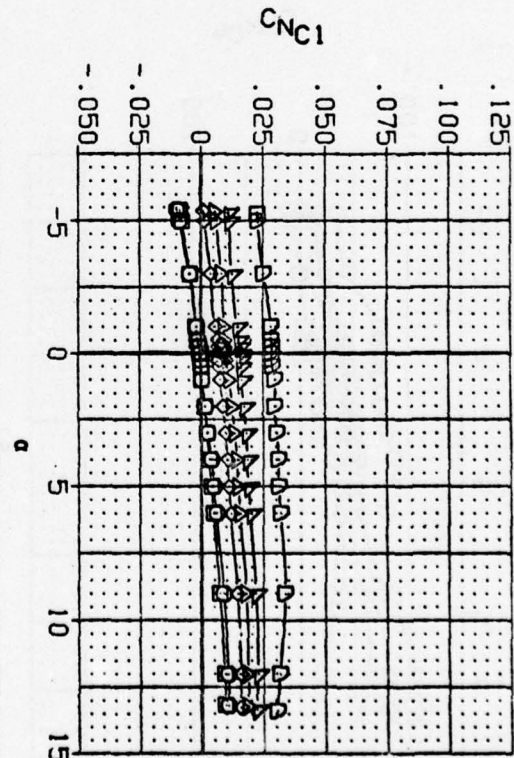
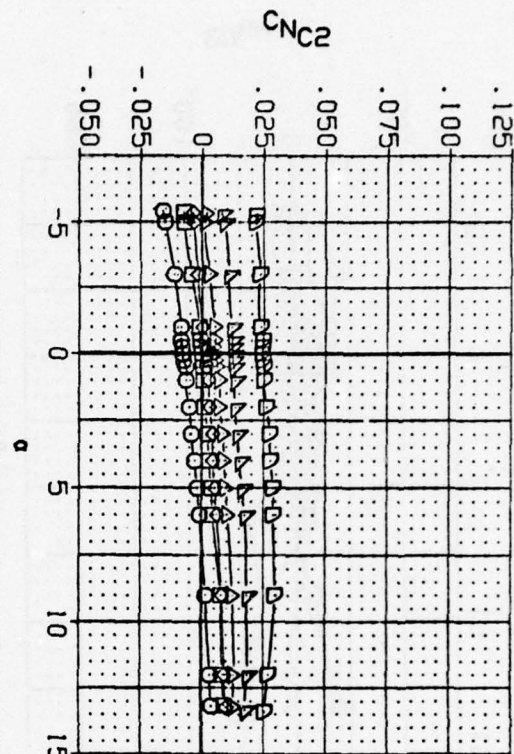
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(A)MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
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(AXH029)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH030)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH031)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH032)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH033)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ. IN.
.000	.000	.000	.000	LBRE 5.0000 IN.
3.000	3.000	3.000	3.000	BREF 5.0000 IN.
6.000	6.000	6.000	6.000	YHP 26.0000 IN.
9.000	9.000	9.000	9.000	ZHP .0000 IN.
15.000	15.000	15.000	15.000	SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

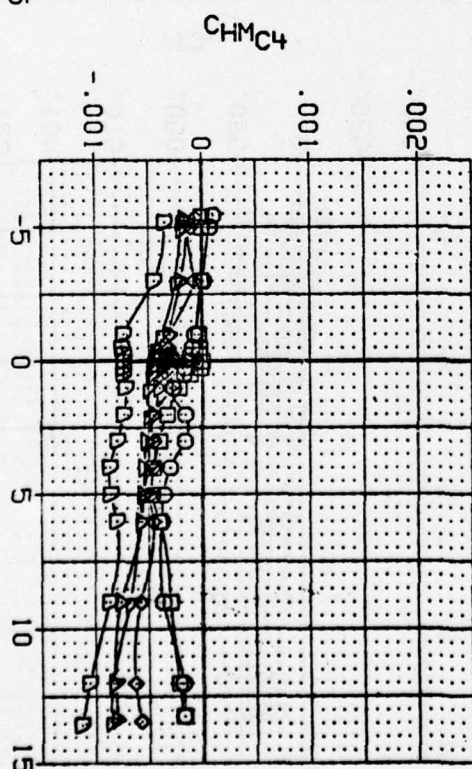
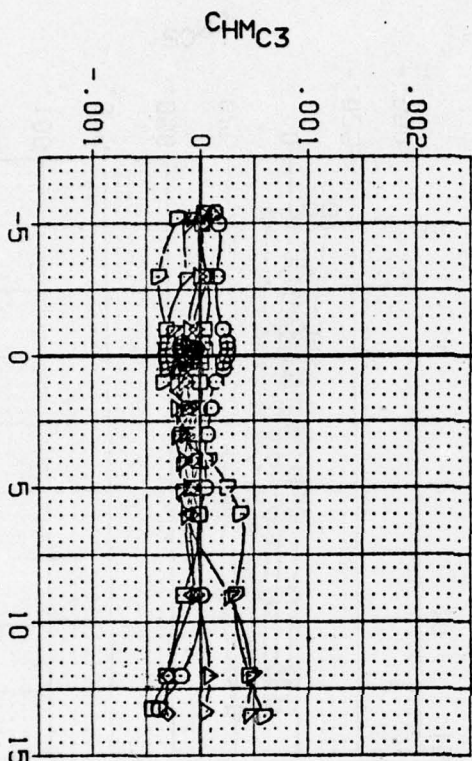
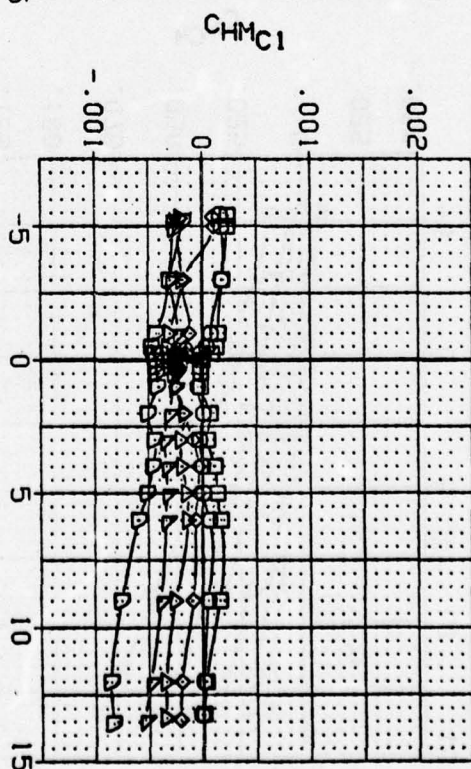
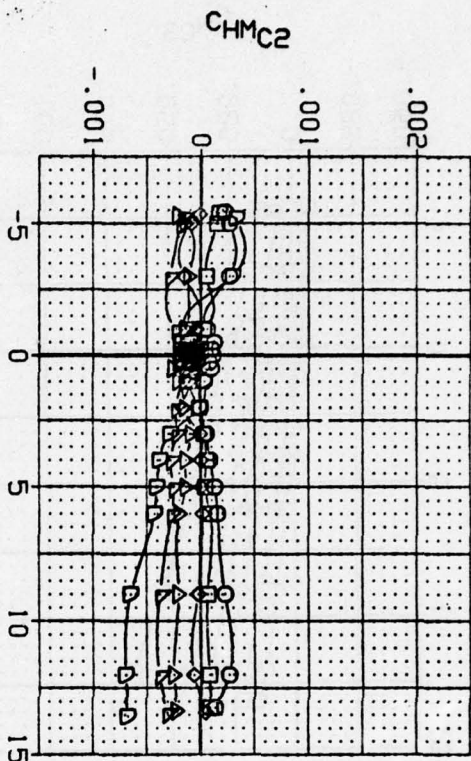
(BXH028)	□	AEDC V41A-C1A, CANARD CONTROL, BNICITI
(BXH029)	◇	AEDC V41A-C1A, CANARD CONTROL, BNICITI
(BXH030)	△	AEDC V41A-C1A, CANARD CONTROL, BNICITI
(BXH031)	▽	AEDC V41A-C1A, CANARD CONTROL, BNICITI
(BXH032)	○	AEDC V41A-C1A, CANARD CONTROL, BNICITI
(BXH033)	×	AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

-3.000	-3.000	-3.000	-3.000
.000	.000	.000	.000
3.000	3.000	3.000	3.000
6.000	6.000	6.000	6.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION

SREF	19.6350	SO. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=45
(A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

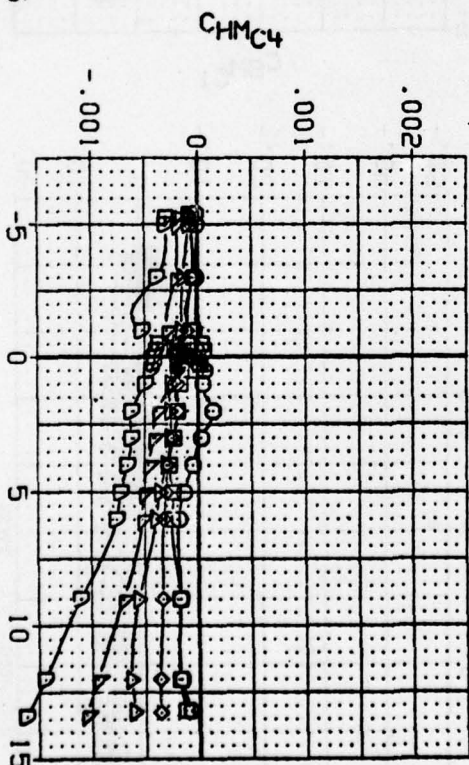
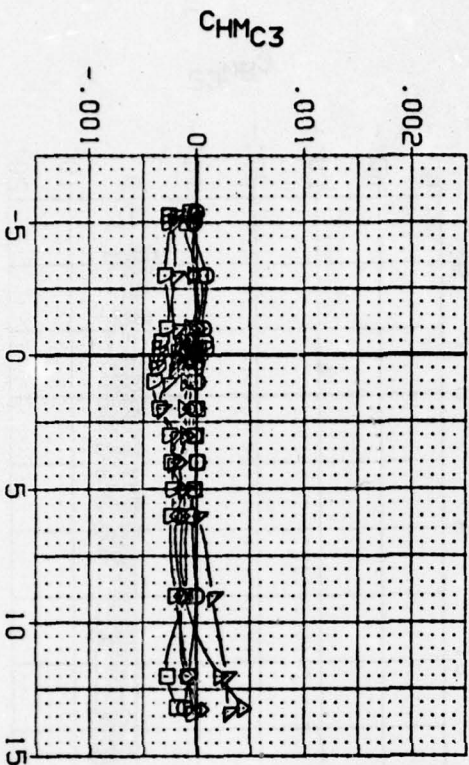
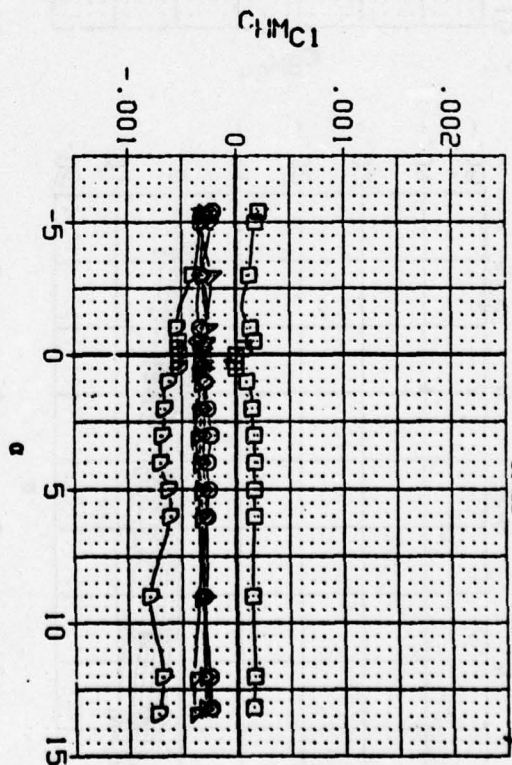
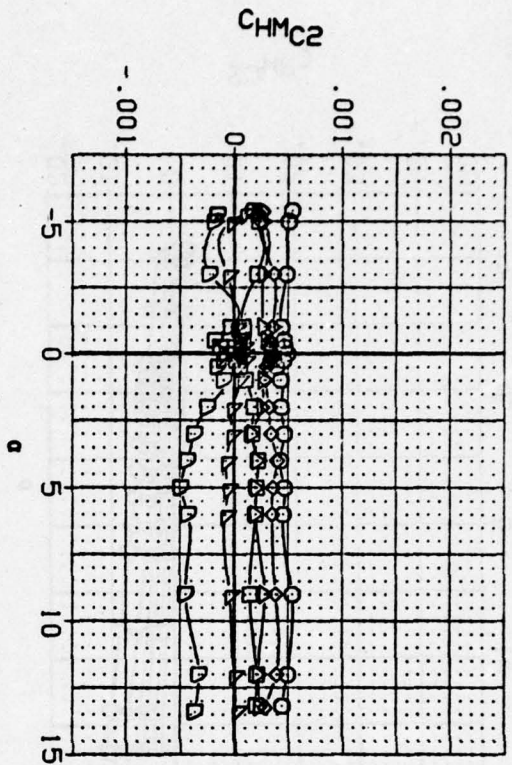
(BXH028) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (BXH029) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (BXH030) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (BXH031) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (BXH032) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (BXH033) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(B) MACH = 4.52

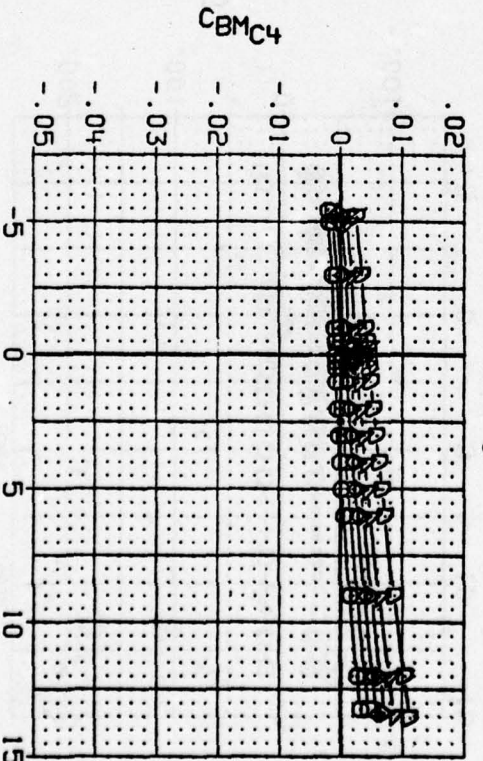
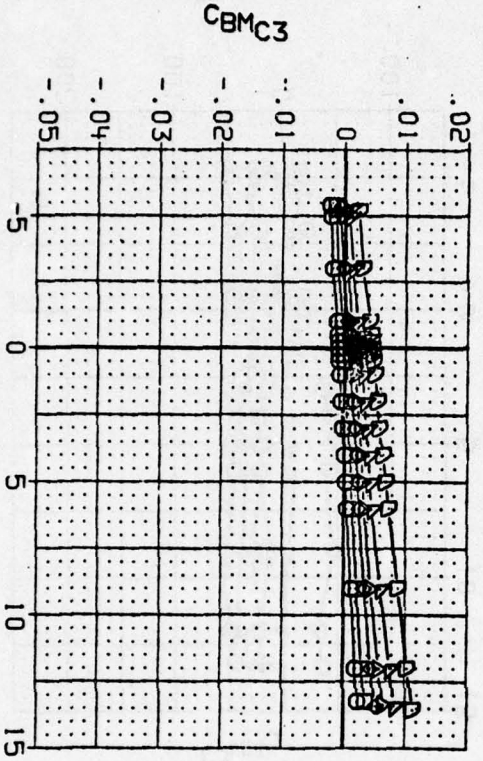
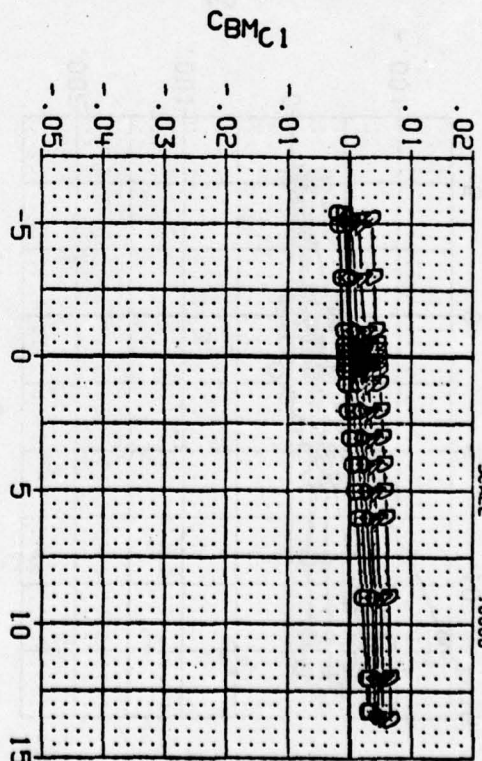
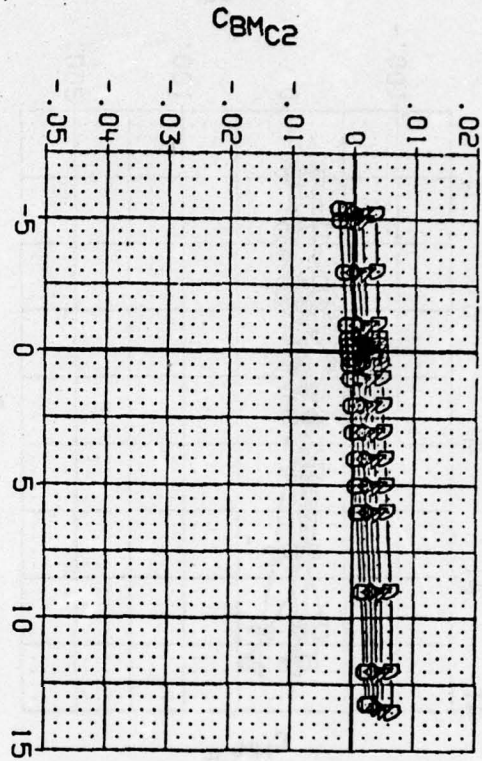
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH028)	□	AEDC V1A-C1A, CANARD CONTROL, BNICITI
(BXH029)	◇	AEDC V1A-C1A, CANARD CONTROL, BNICITI
(BXH030)	△	AEDC V1A-C1A, CANARD CONTROL, BNICITI
(BXH031)	▽	AEDC V1A-C1A, CANARD CONTROL, BNICITI
(BXH032)	○	AEDC V1A-C1A, CANARD CONTROL, BNICITI
(BXH033)	◇	AEDC V1A-C1A, CANARD CONTROL, BNICITI

REFERENCE INFORMATION

DCNO1	-3.000	DCNO2	-3.000	DCNO3	-3.000	DCNO4	-3.000
	.000		.000		.000		.000
	3.000		3.000		3.000		3.000
	6.000		6.000		6.000		6.000
	9.000		9.000		9.000		9.000
	15.000		15.000		15.000		15.000

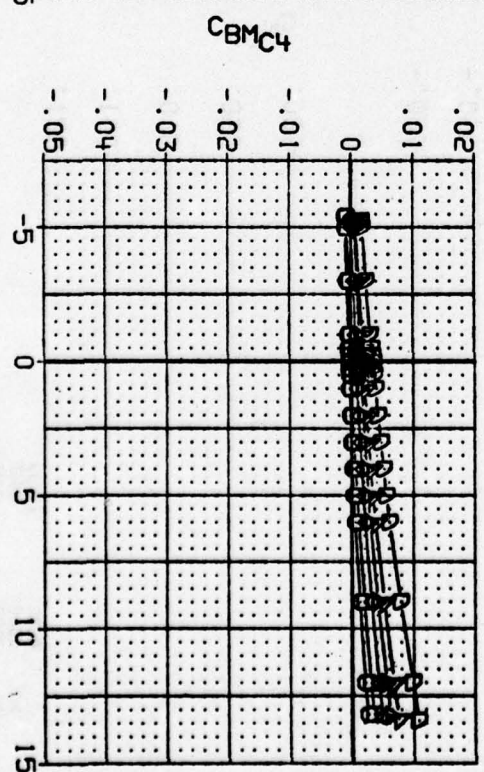
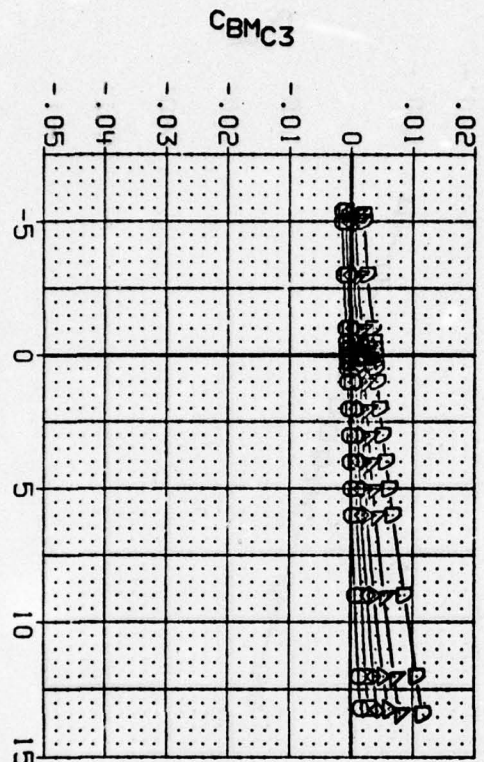
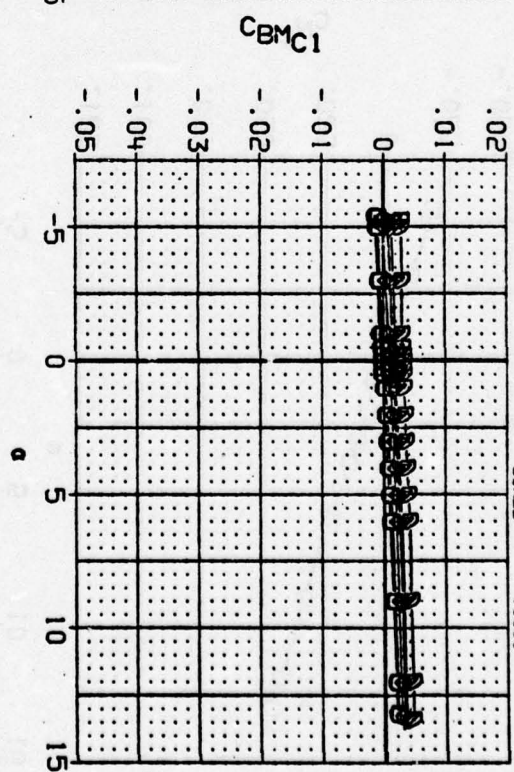
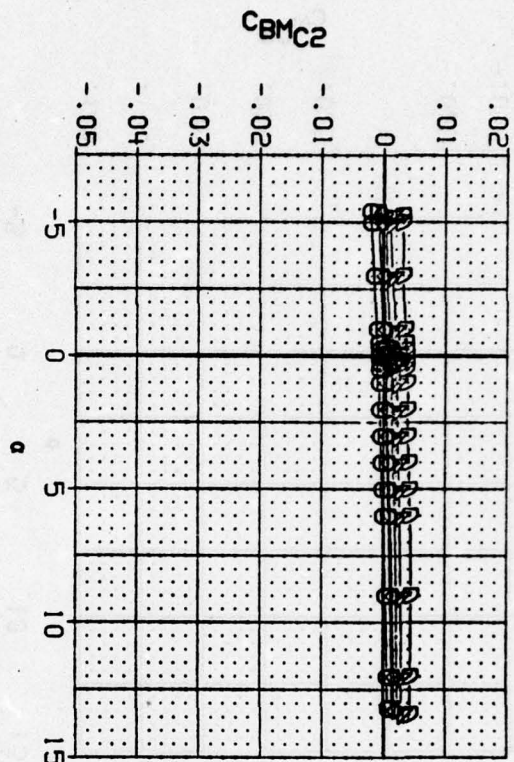
SREF	19.6350	SO.IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	25.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (A)MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(BXH028)	○	AEDC WJIA-CIA, CANARD CONTROL, BNICITI
(BXH029)	□	AEDC WJIA-CIA, CANARD CONTROL, BNICITI
(BXH030)	△	AEDC WJIA-CIA, CANARD CONTROL, BNICITI
(BXH031)	▽	AEDC WJIA-CIA, CANARD CONTROL, BNICITI
(BXH032)	◇	AEDC WJIA-CIA, CANARD CONTROL, BNICITI
(BXH033)	◊	AEDC WJIA-CIA, CANARD CONTROL, BNICITI

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SO. IN.
3.000	3.000	3.000	3.000	LBREF 5.0000 IN.
6.000	6.000	6.000	6.000	BRF 26.0000 IN.
9.000	9.000	9.000	9.000	YREF .0000 IN.
15.000	15.000	15.000	15.000	ZREF .0000 IN.
				SCALE .0000

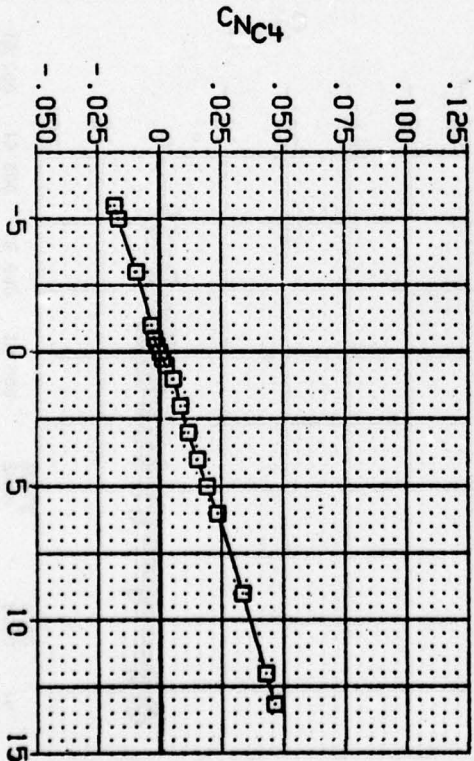
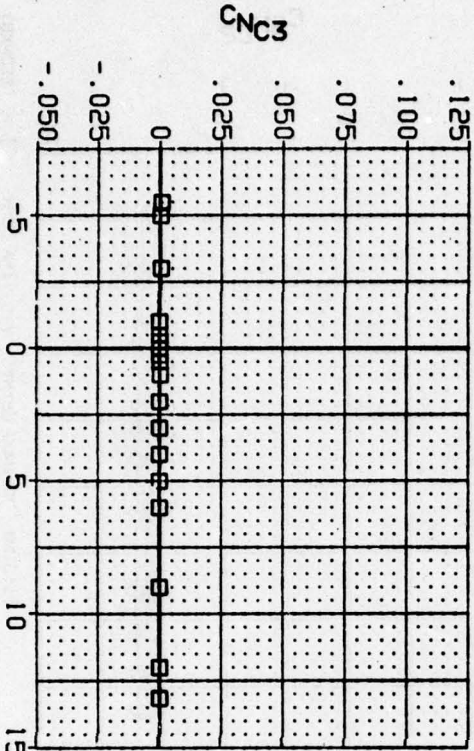
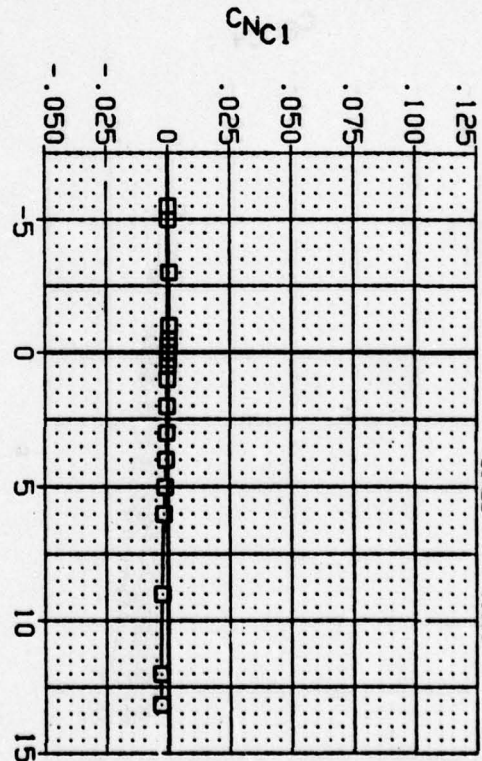
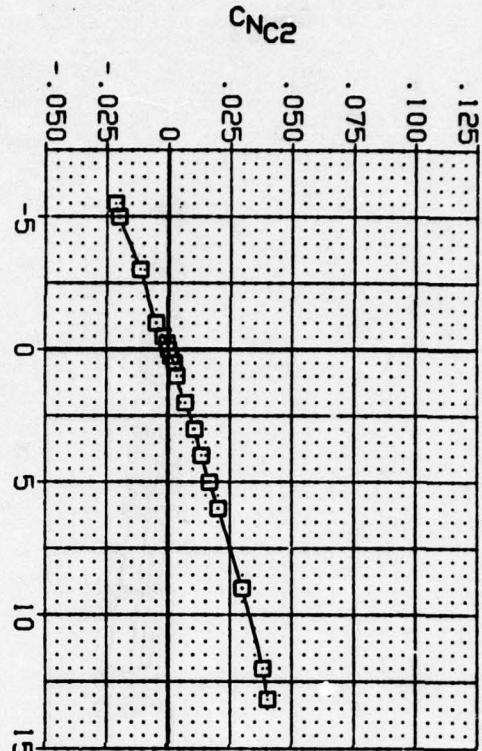


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH034) DATA NOT AVAILABLE
 (AXH035) AEDC WJA-C1A, CANARD CONTROL, BNIC11
 (AXH036) DATA NOT AVAILABLE
 (AXH037) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

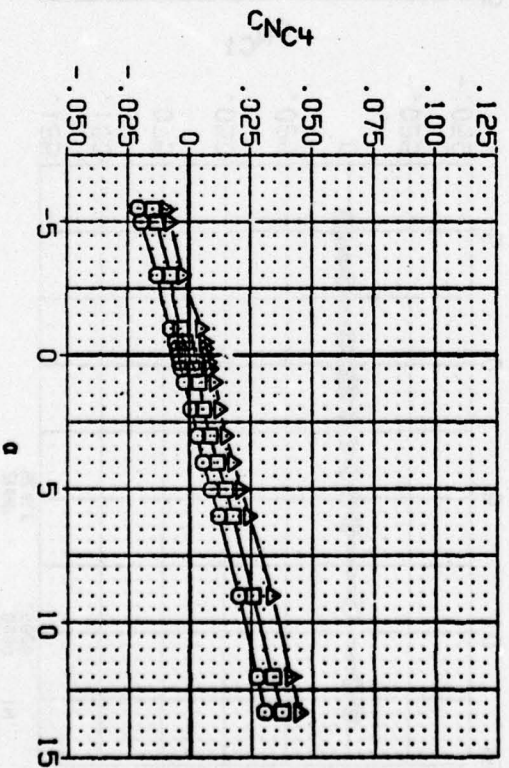
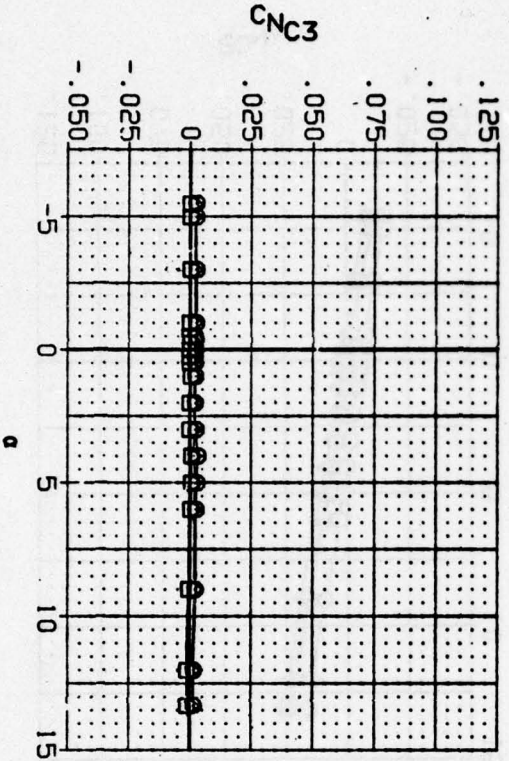
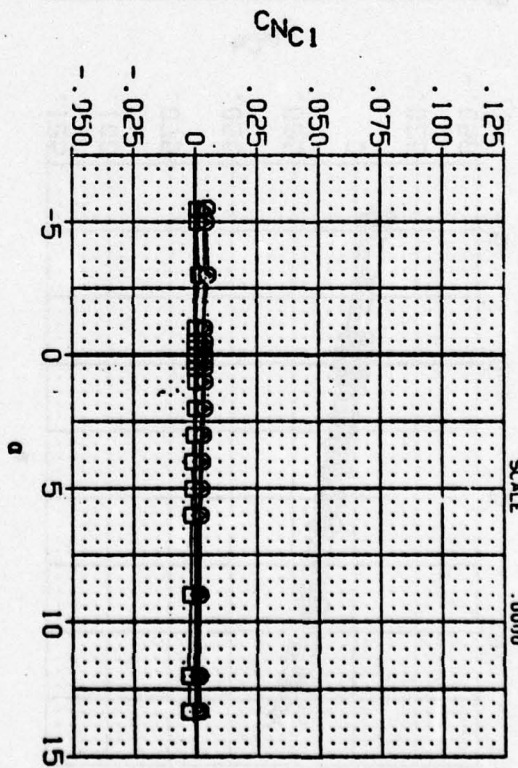
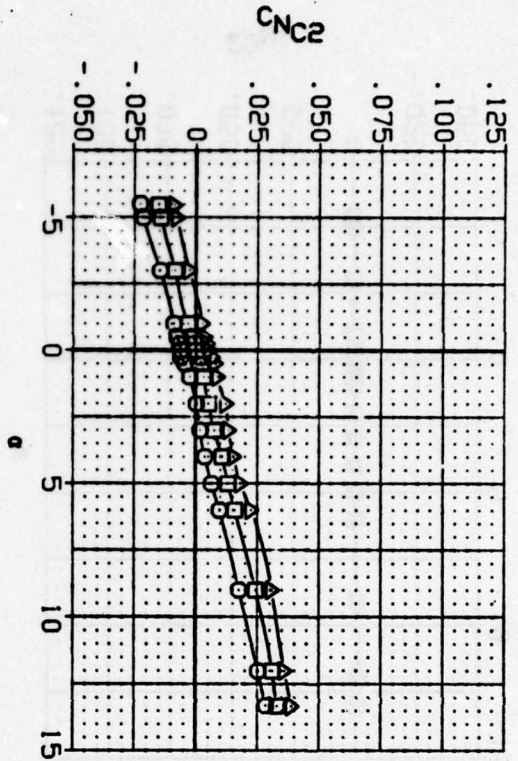


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH034) \square AEDC VVIA-CIA, CANARD CONTROL, BNIC11
 (AXH035) \square AEDC VVIA-CIA, CANARD CONTROL, BNIC11
 (AXH036) \square DATA NOT AVAILABLE
 (AXH037) Δ AEDC VVIA-CIA, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

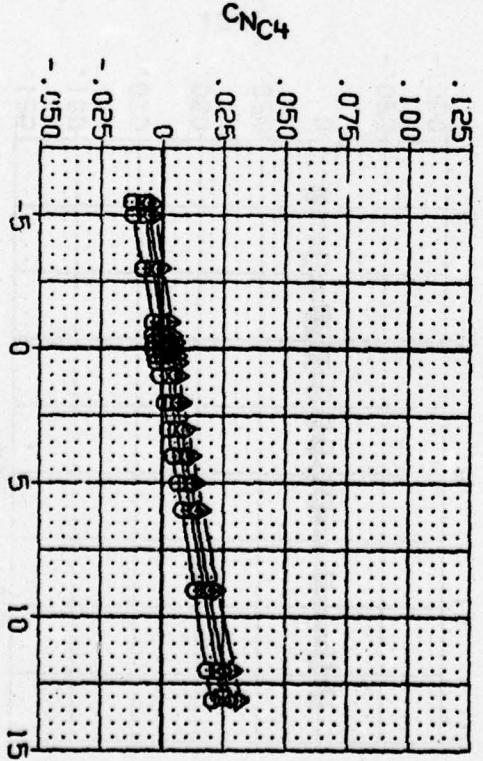
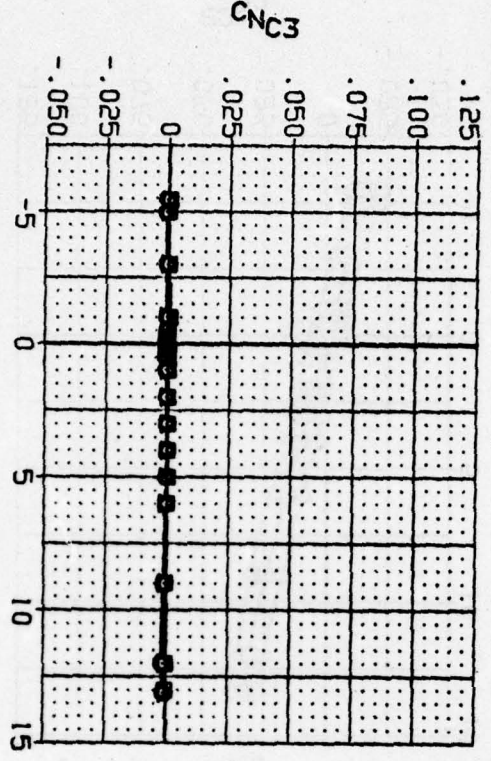
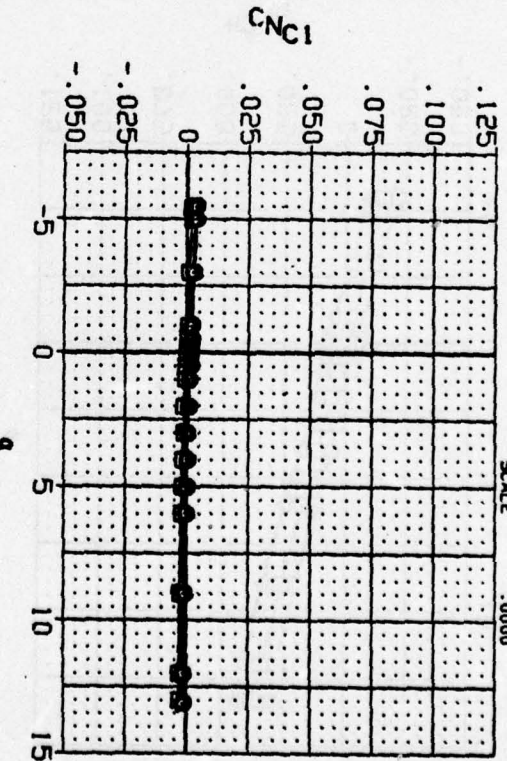
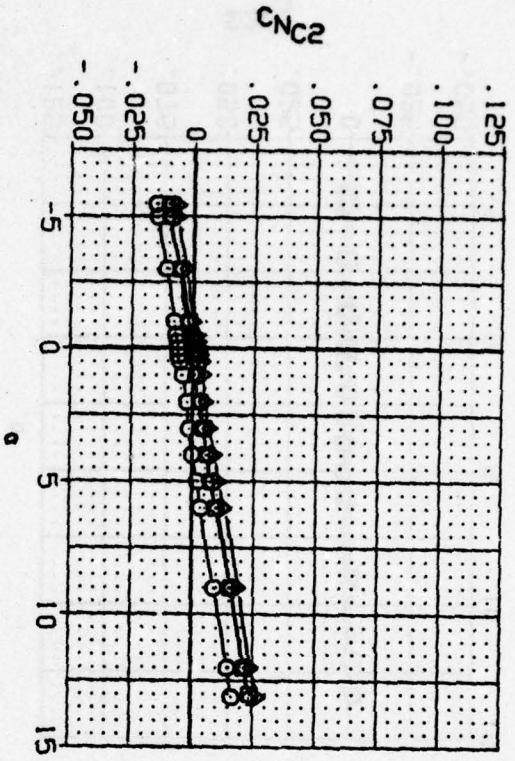


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=0
 (B)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (A0H034) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (A0H035) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (A0H036) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (A0H037) Δ AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

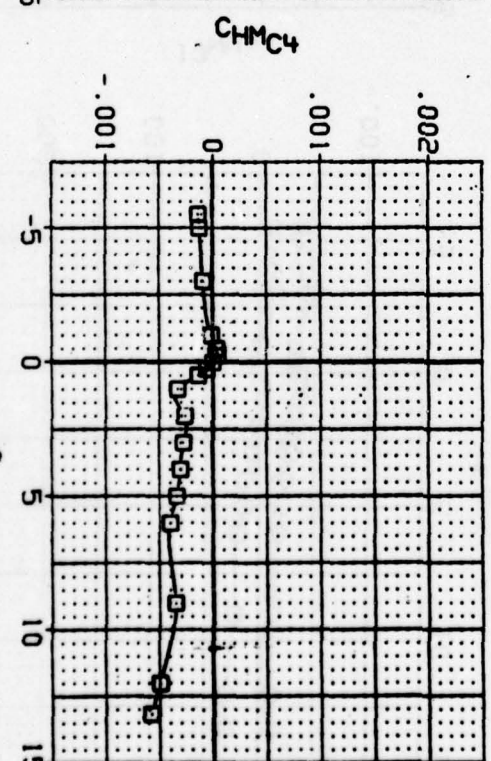
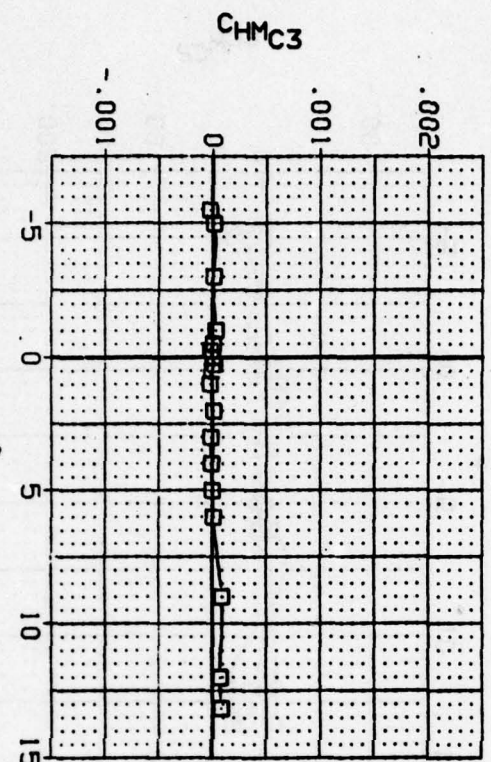
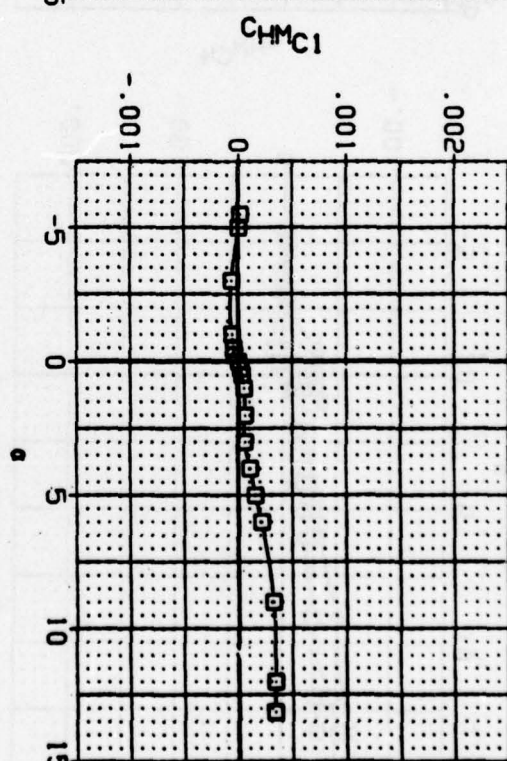
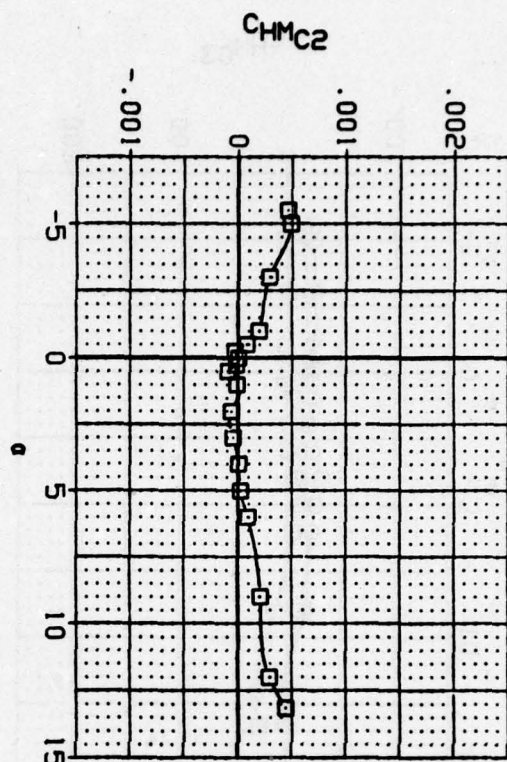


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH03N) DATA NOT AVAILABLE
 (BXH03S) AEDC VVIA-CIA, CANARD CONTROL, BNIC11
 (BXH03S) DATA NOT AVAILABLE
 (BXH037) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

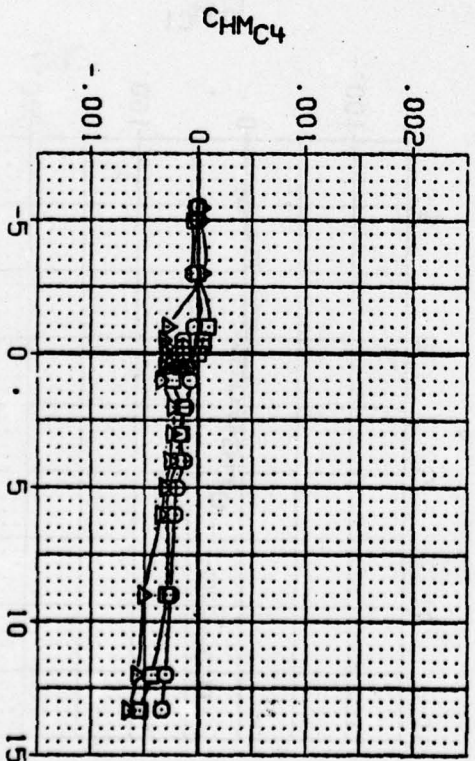
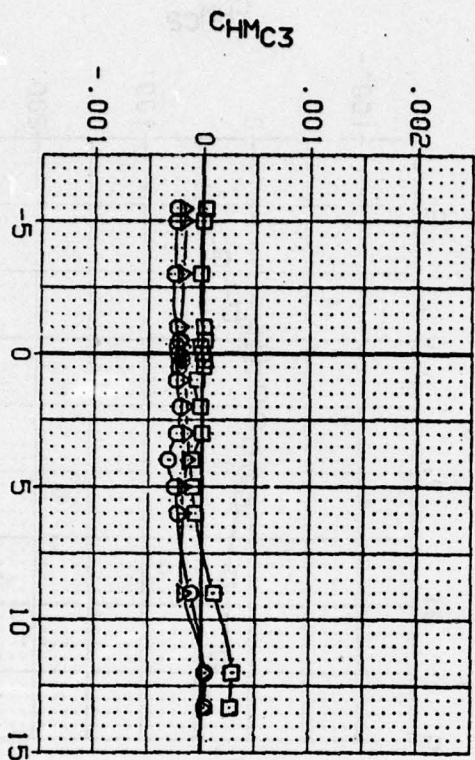
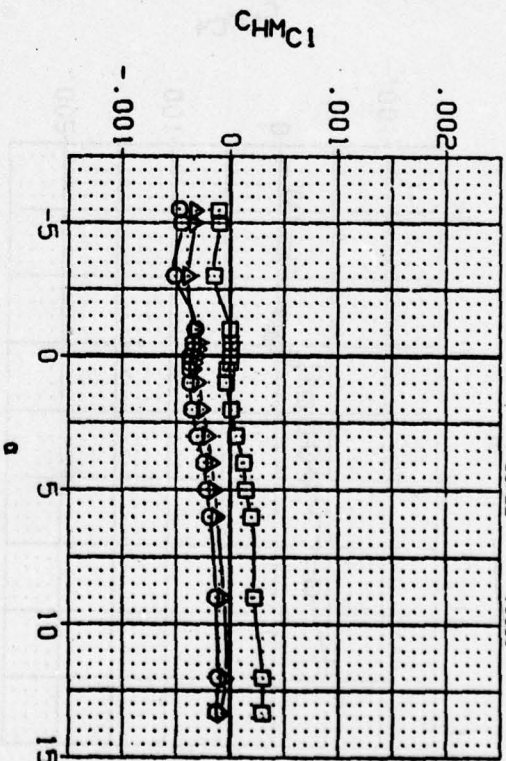
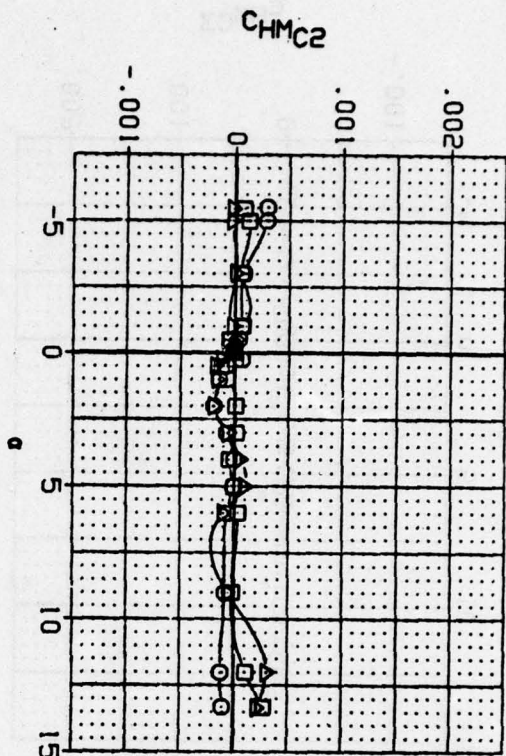


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=0
 (A)MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH03V) AEDC WJIA-CIA, CANARD CONTROL, BNICIT1
 (BXH03S) AEDC WJIA-CIA, CANARD CONTROL, BNICIT1
 (BXH036) DATA NOT AVAILABLE
 (BXH037) AEDC WJIA-CIA, CANARD CONTROL, BNICIT1

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SPEC 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

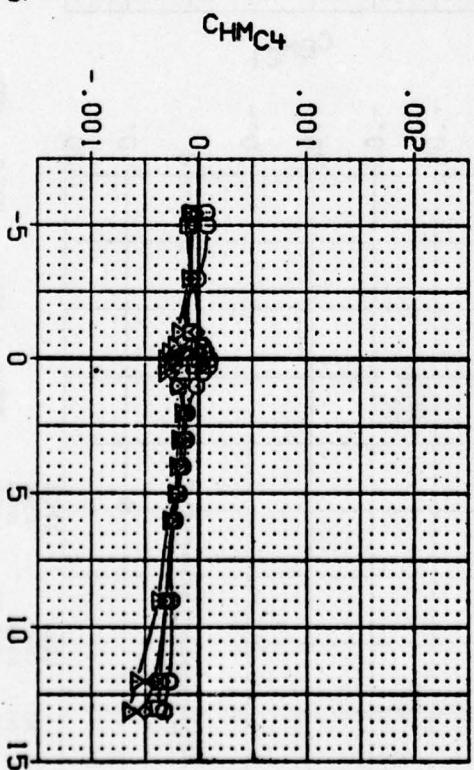
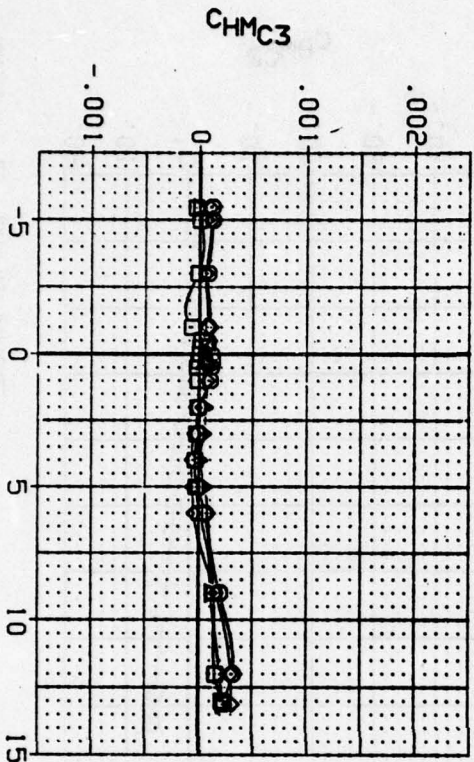
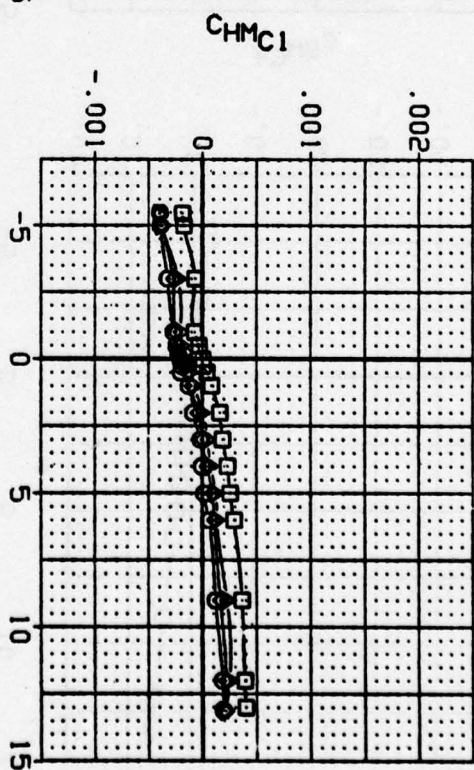
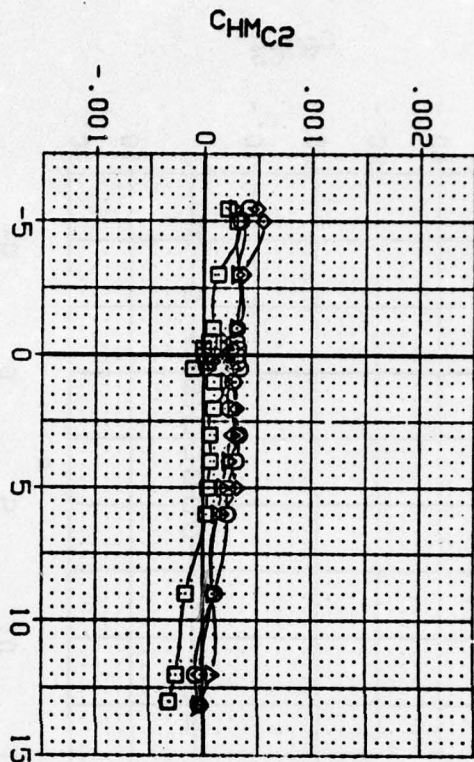


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (B)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH031) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (BXH033) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (BXH036) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (BXH037) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 .000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

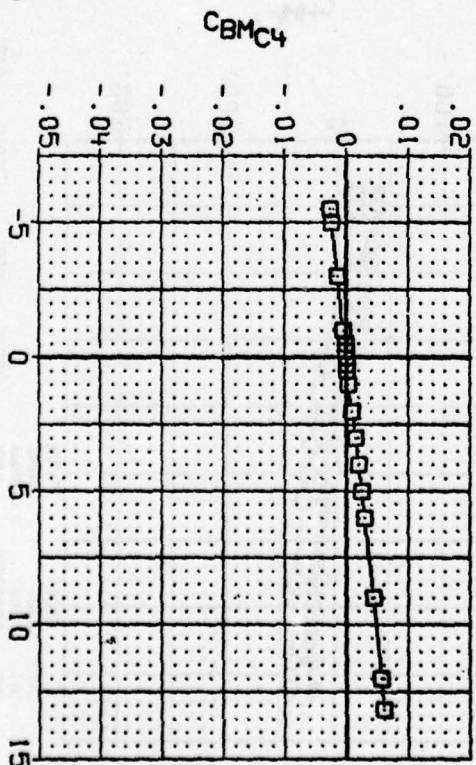
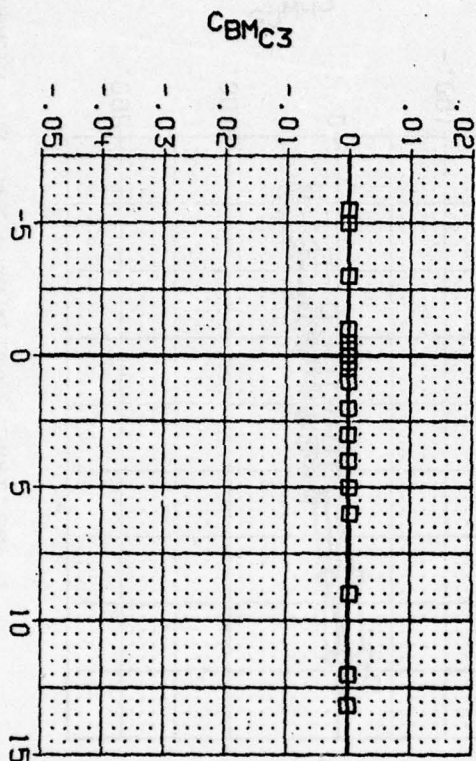
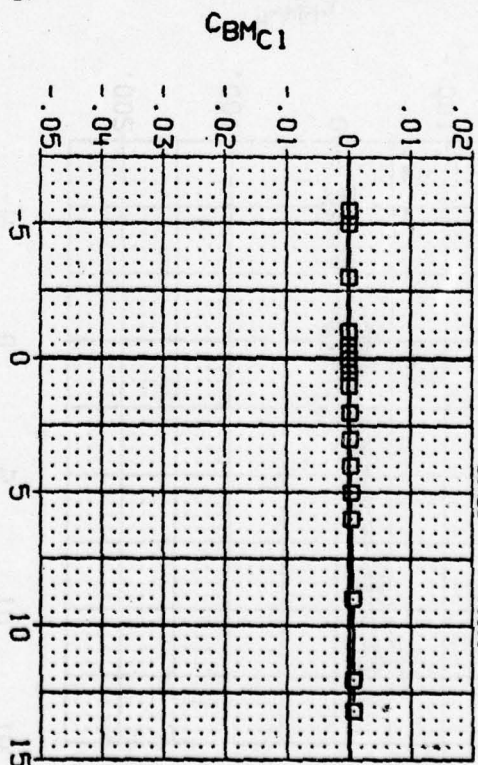
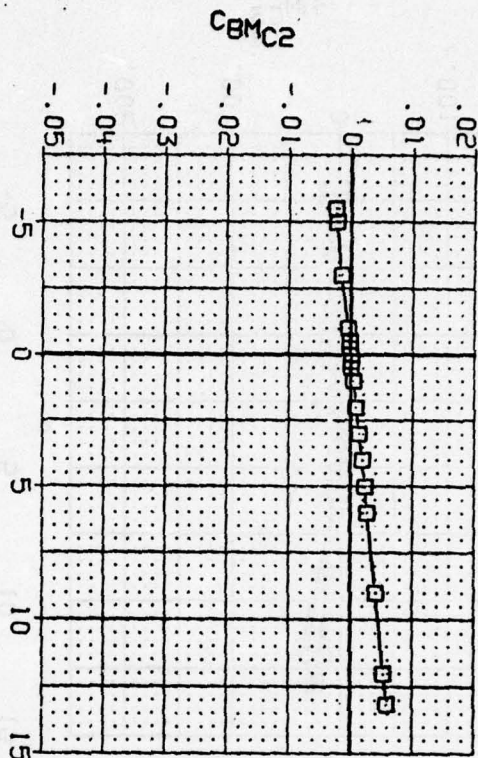


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH034) \square DATA NOT AVAILABLE
 (BXH035) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH036) \triangle DATA NOT AVAILABLE
 (BXH037) \triangle DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

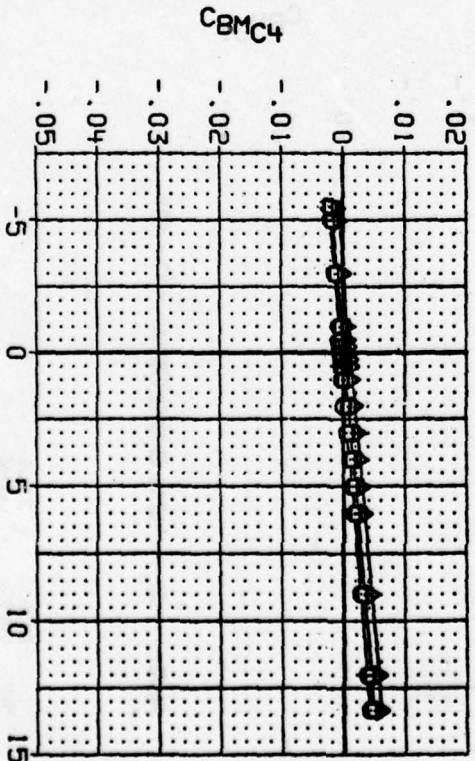
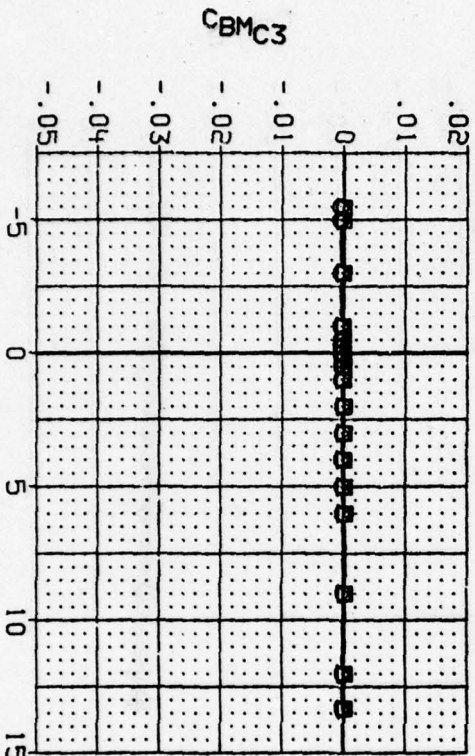
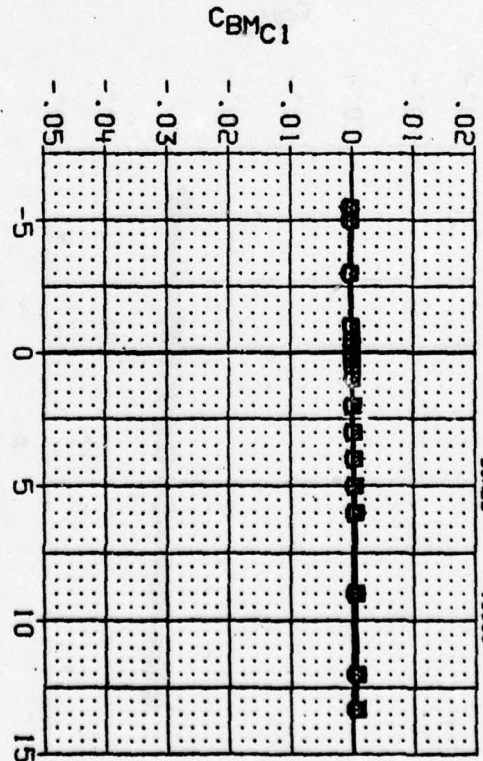
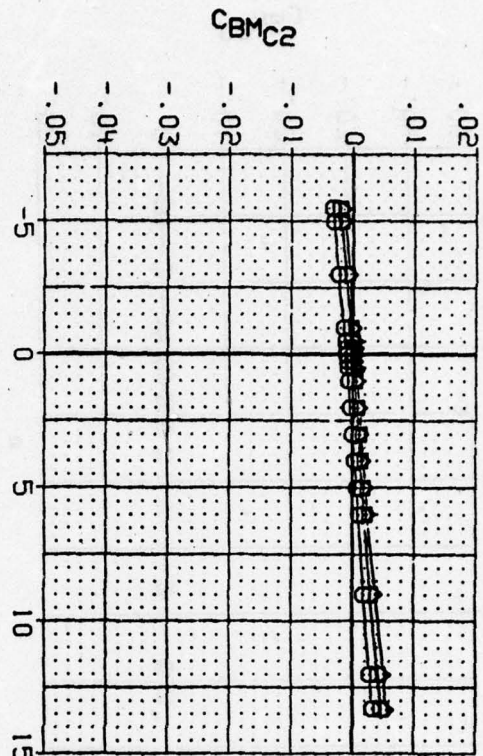


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=45$
 $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH031) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (BXH032) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (BXH033) \square DATA NOT AVAILABLE
 (BXH037) Δ AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

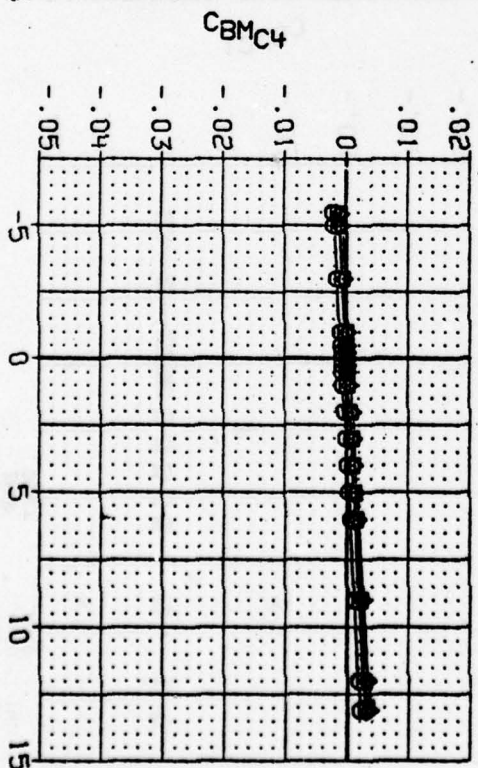
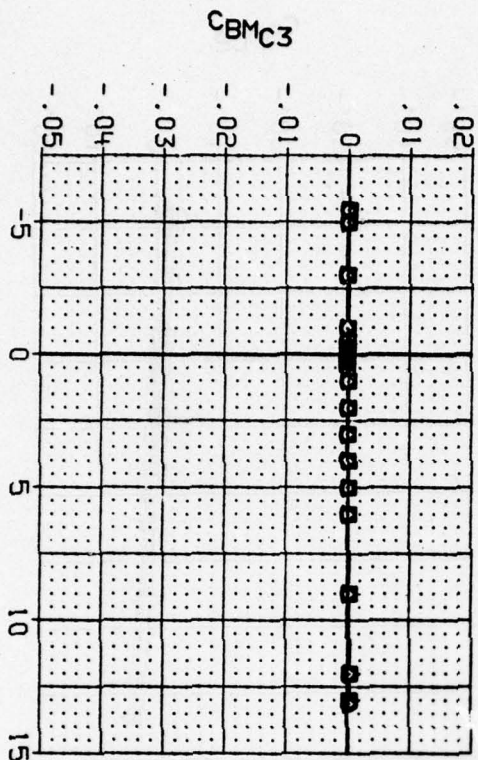
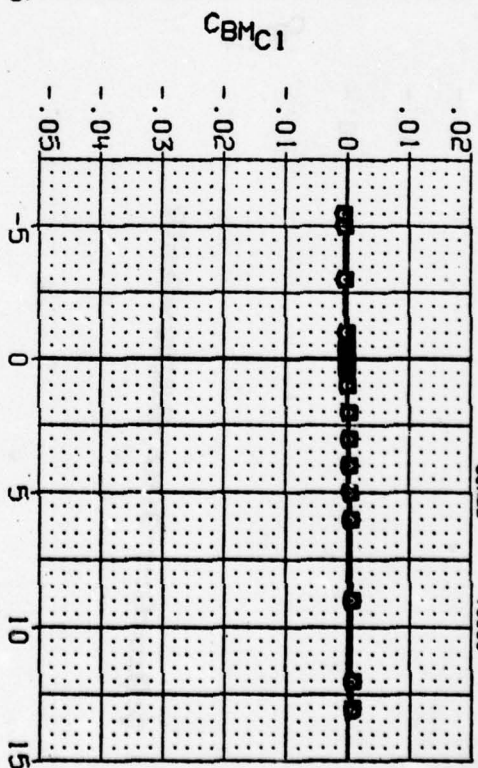
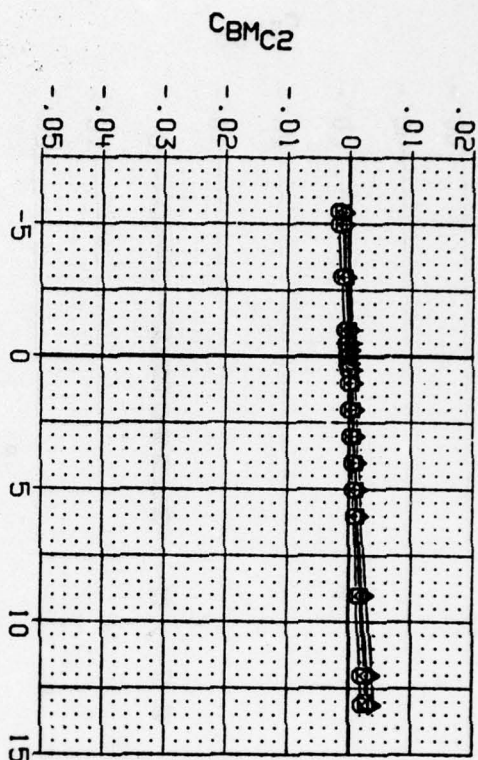


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH034) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH035) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH036) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (BXH037) Δ AEDC V41A-C1A, CANARD CONTROL, BNICITI

DN001 .000 DN002 -3.000 DN003 .000 DN004 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

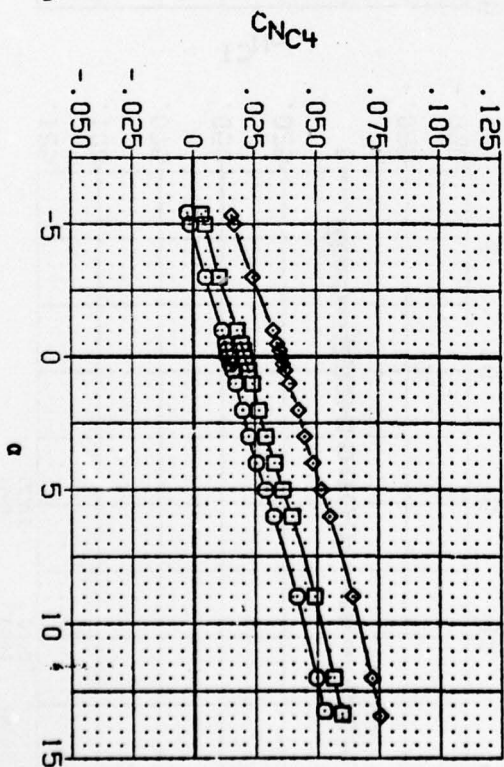
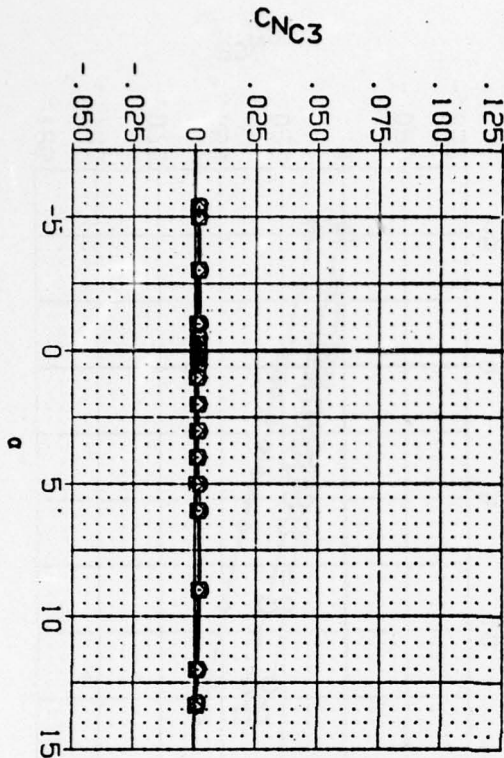
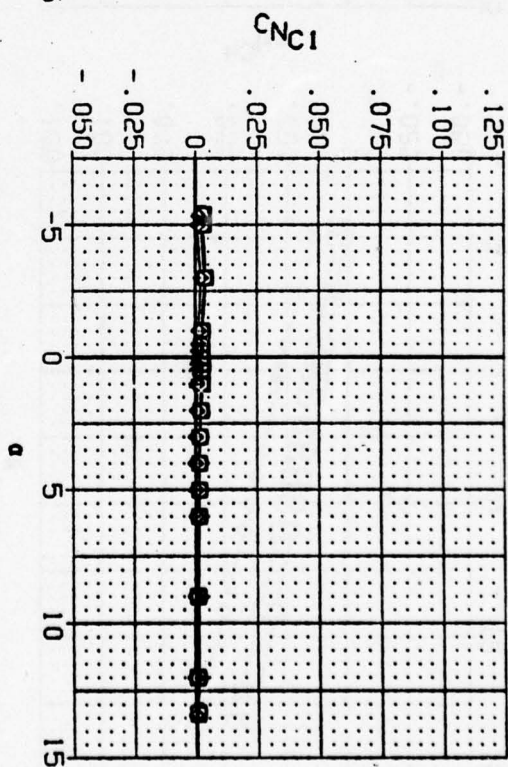
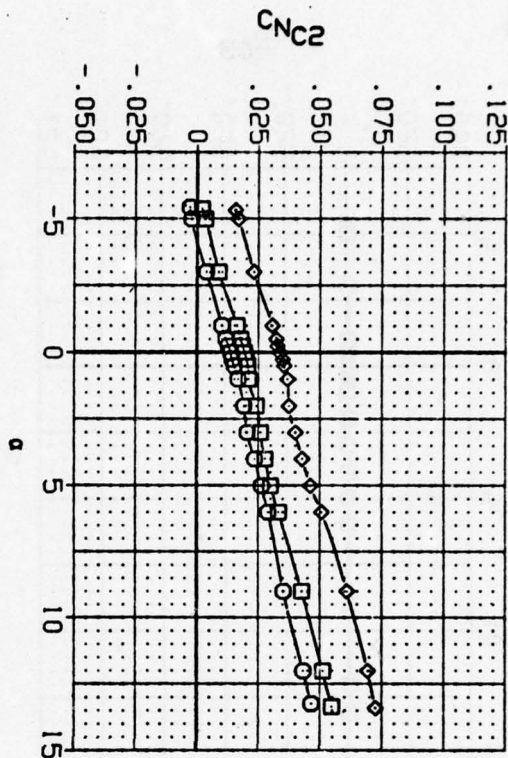


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=45$
 $PHICND=0$
 $(C)MACH = 4.52$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH038) 'AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH039) 'AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH040) 'AEDC VMA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
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 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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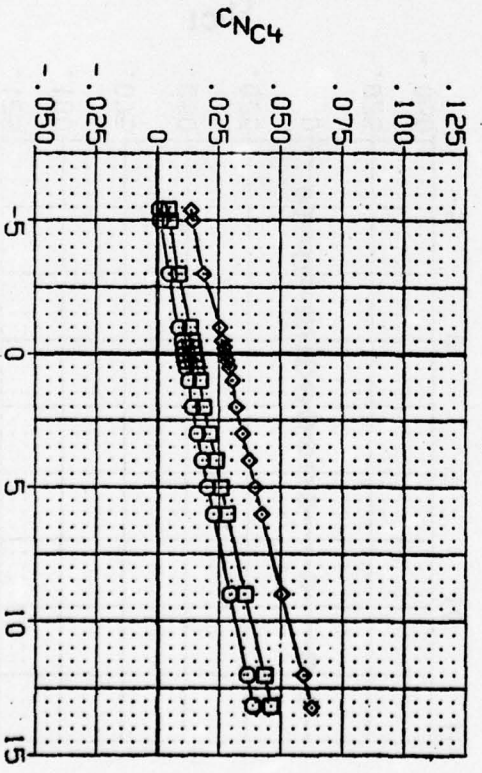
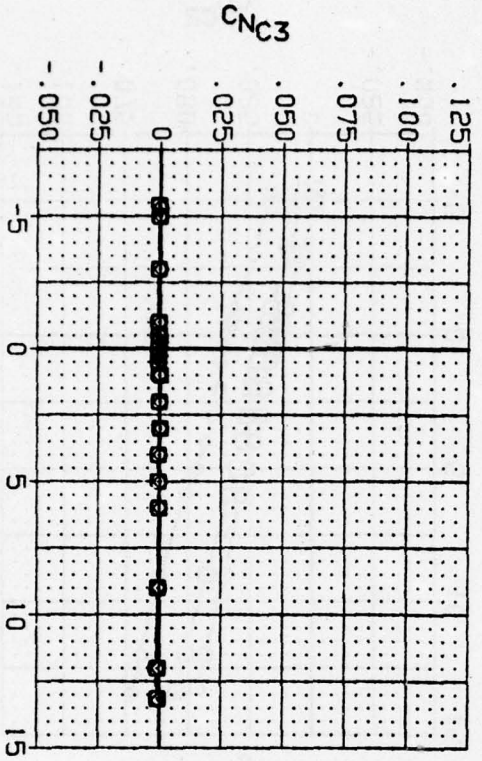
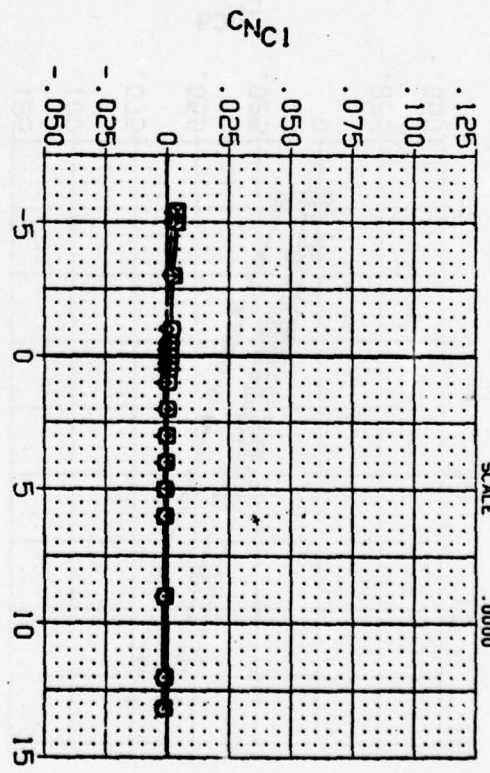
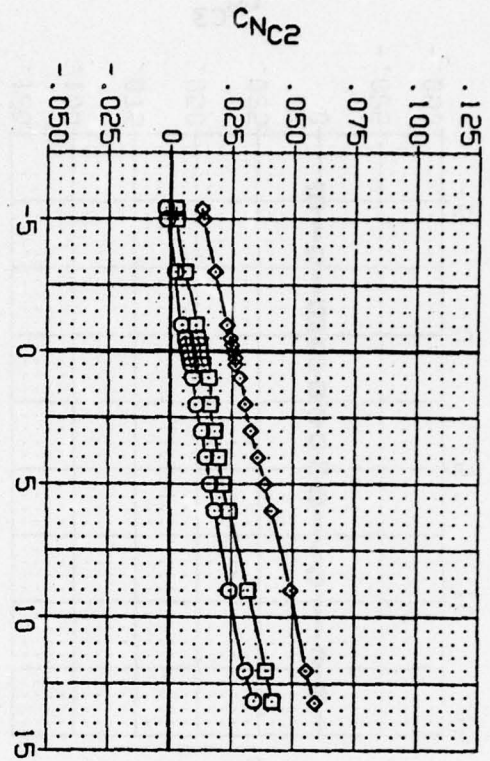


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (AXH023) \diamond AEDC WJIA-CIA, CANARD CONTROL, BNICITI
 (AXH040) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
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 .000 9.000 .000 9.000
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REFERENCE INFORMATION
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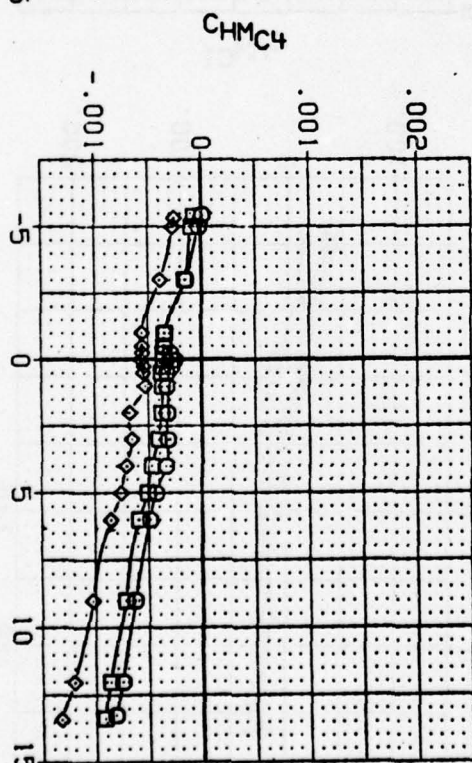
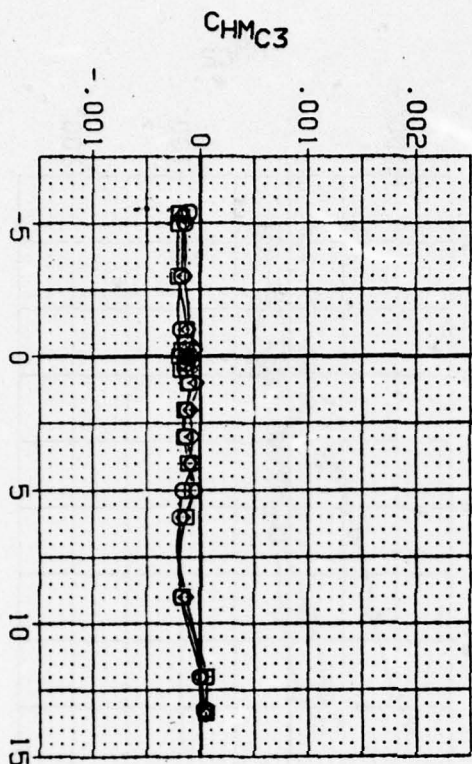
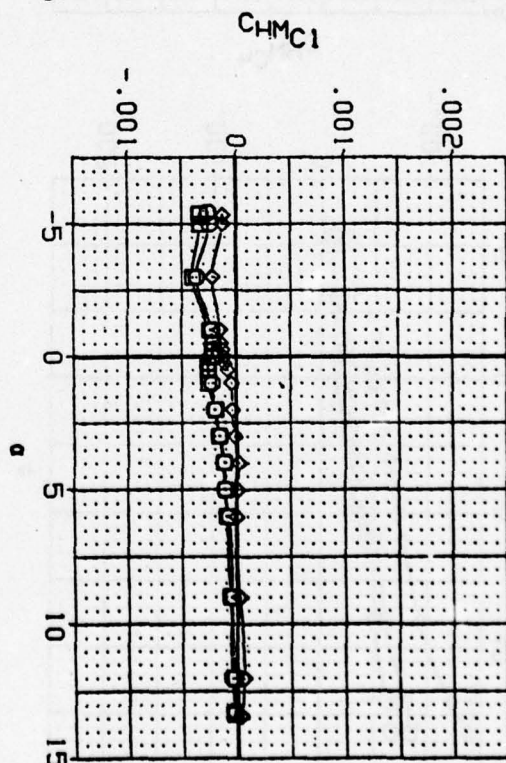
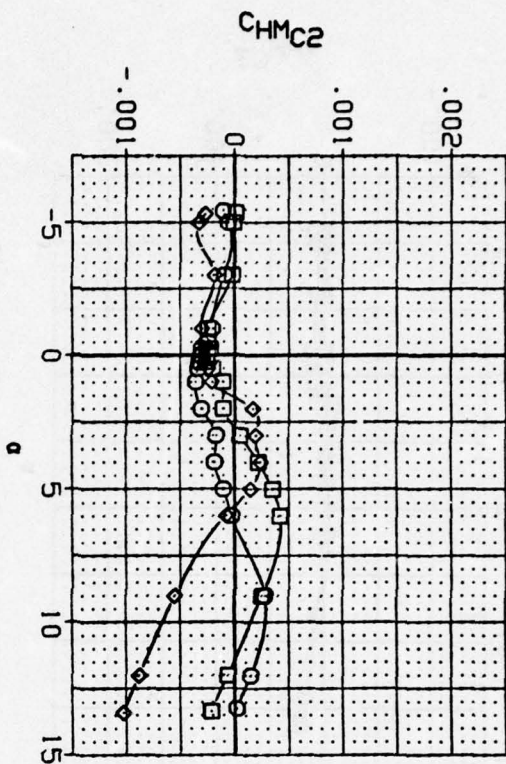


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH038) □ AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXH039) ○ AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXH040) ◇ AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
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 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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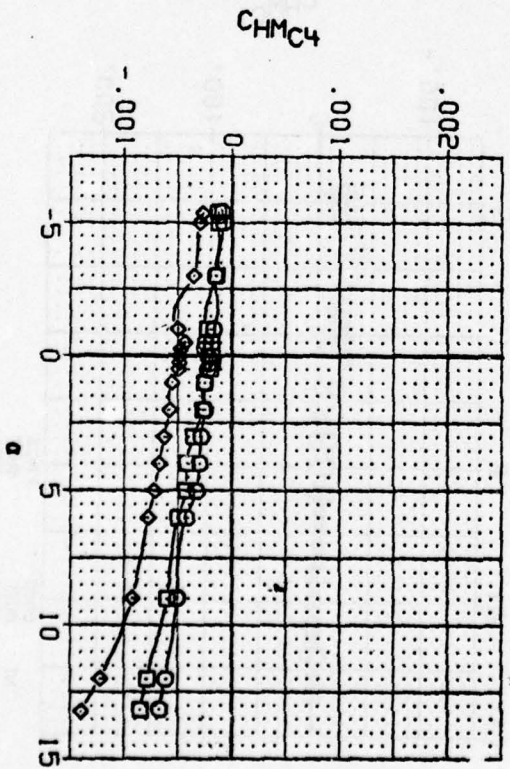
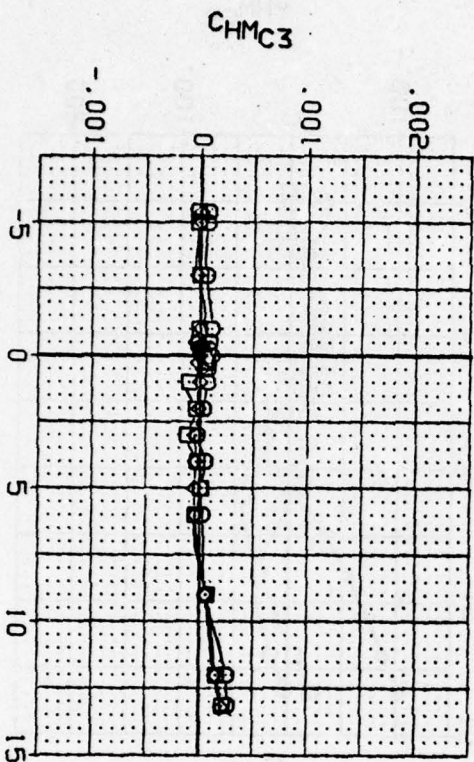
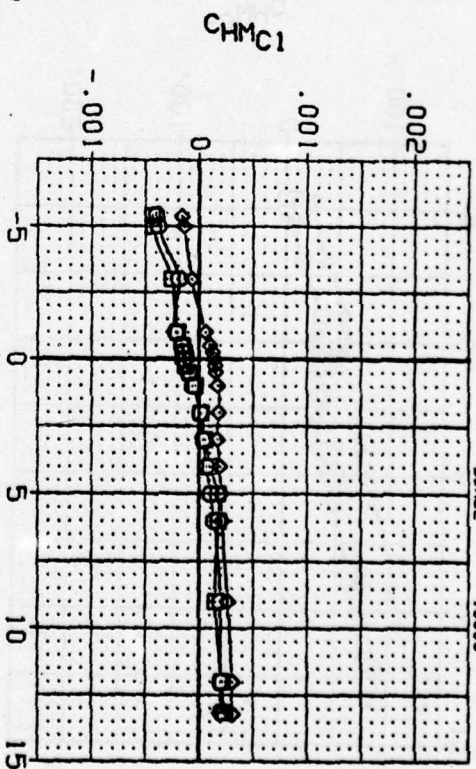
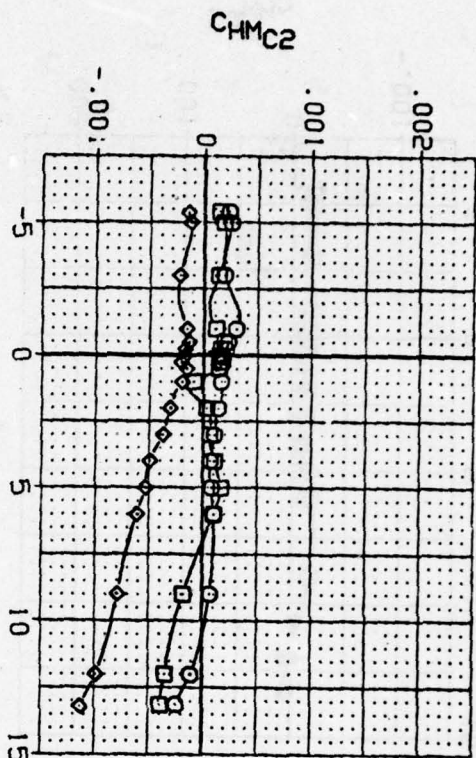


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=45$ $PHICND=0$
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXND38) AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXND39) AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXND40) AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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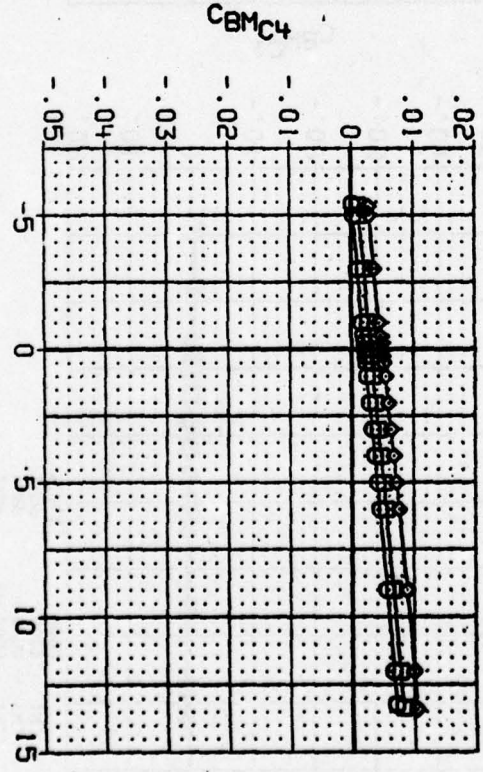
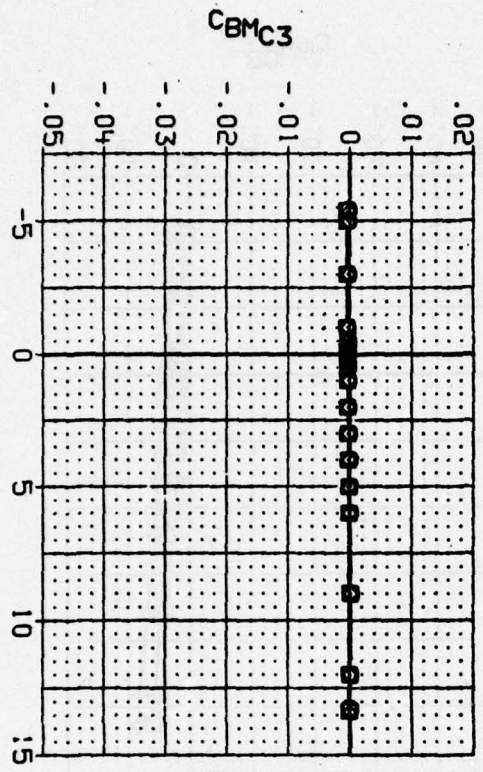
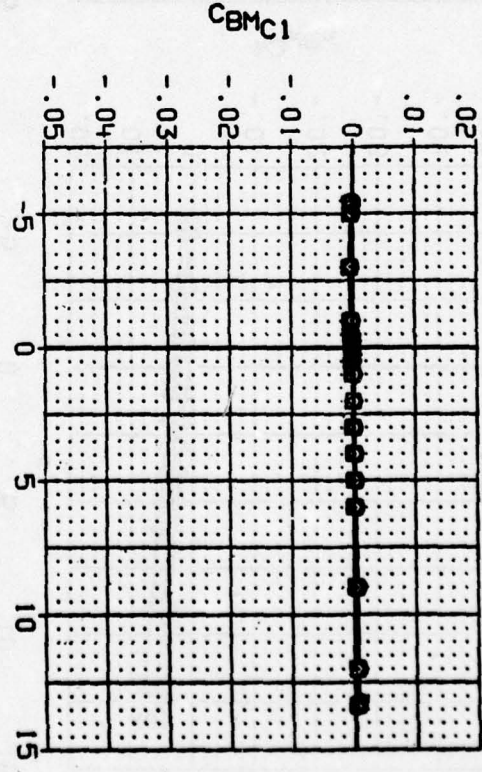
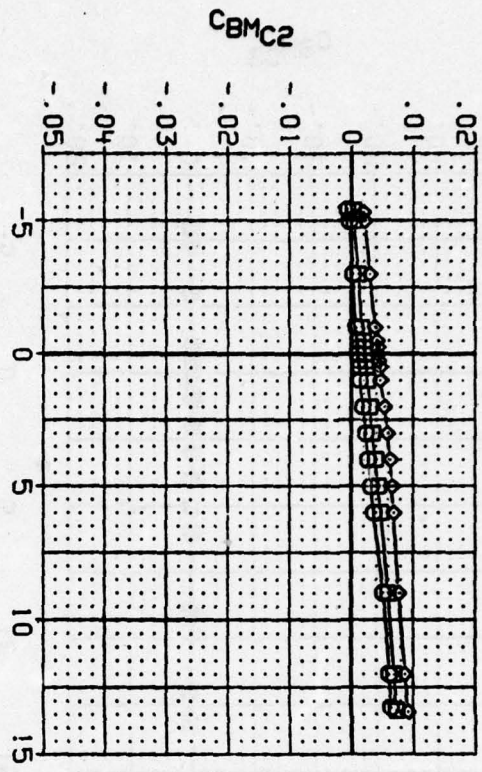


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=45$
 $PHICND=0$
 (B) MACH 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH038) ☐ AEDC V1A-C1A, CANARD CONTROL, BNIC11
 (BXH039) ☐ AEDC V1A-C1A, CANARD CONTROL, BNIC11
 (BXH040) ☒ AEDC V1A-C1A, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4
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 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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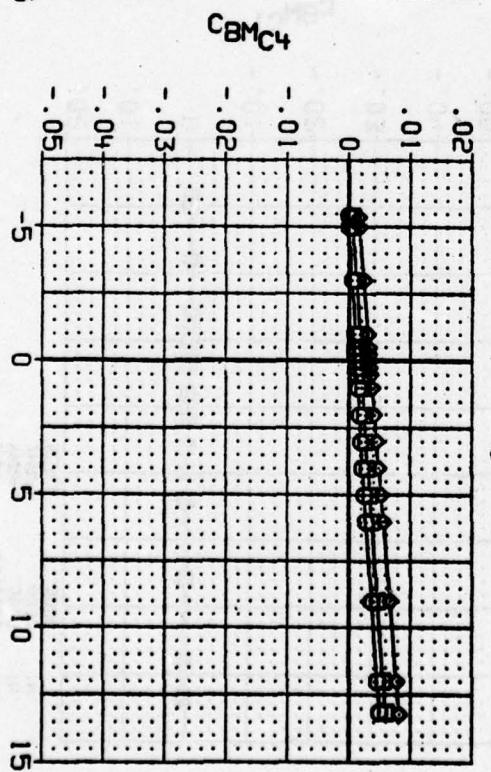
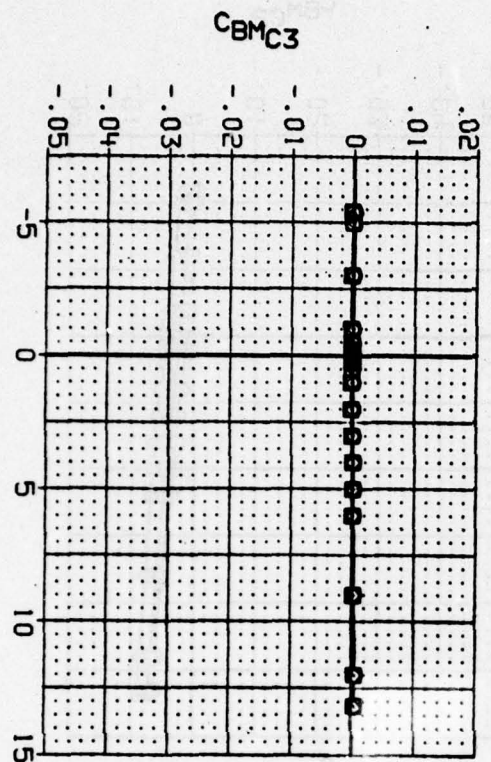
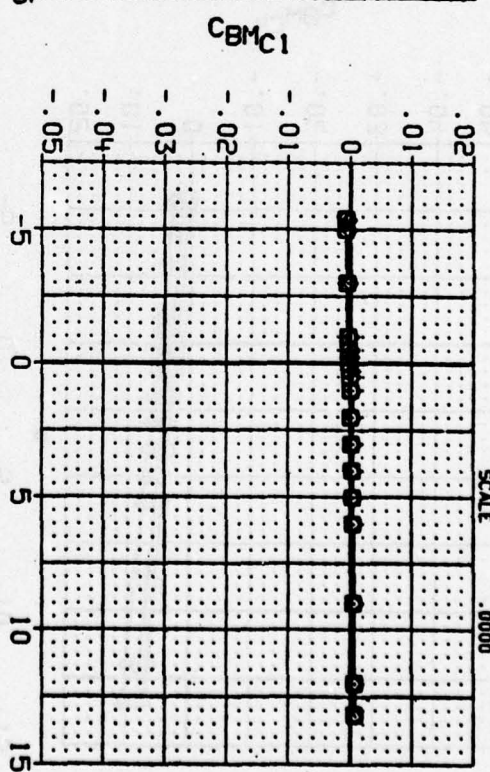
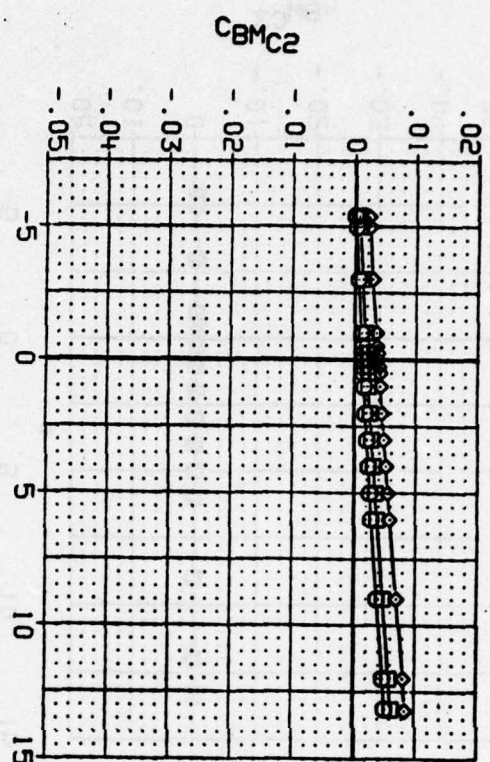


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=0
 (A)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (B04039) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (B040401) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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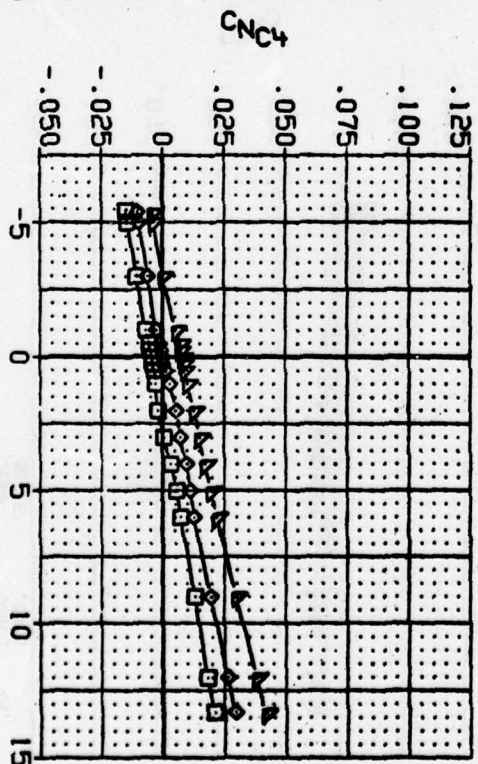
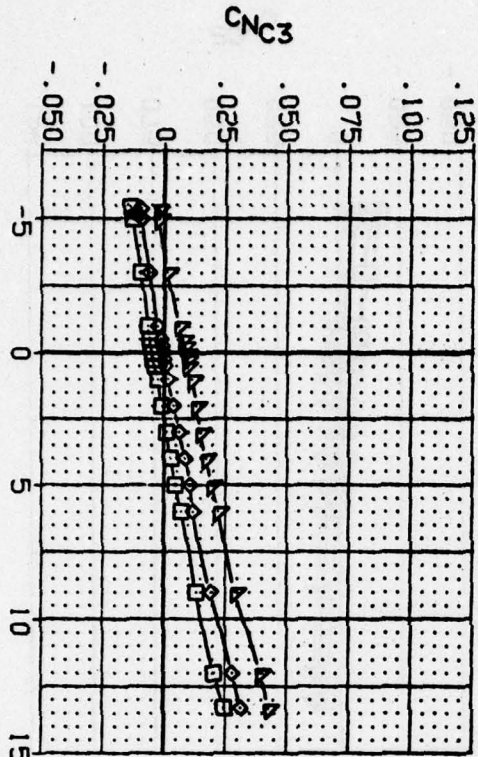
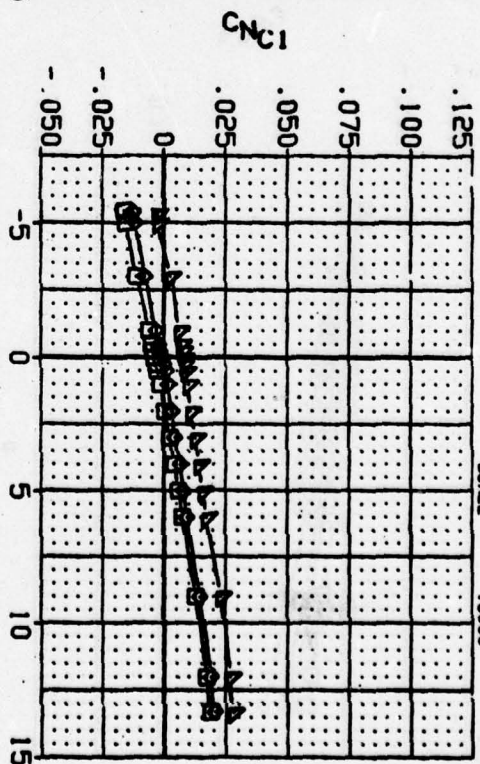
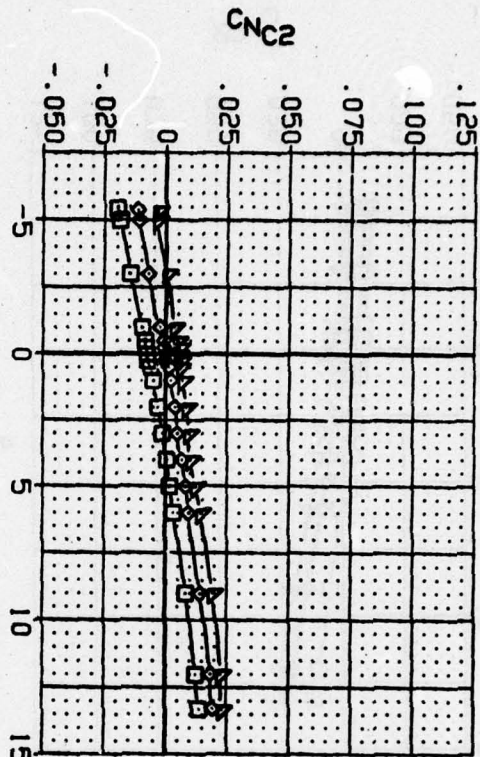


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (AXH0-2) ◇ AEDC WJIA-CIA, CANARD CONTROL, ENICITI
 (AXH0-3) △ AEDC WJIA-CIA, CANARD CONTROL, ENICITI
 (AXH0-4) △ AEDC WJIA-CIA, CANARD CONTROL, ENICITI
 (AXH0-5) △ AEDC WJIA-CIA, CANARD CONTROL, ENICITI

DCND1 DCND2 DCND3 DCND4
 -5.000 -5.000 -5.000 -5.000
 -3.000 -3.000 -3.000 -3.000
 1.000 1.000 1.000 1.000
 3.000 3.000 3.000 3.000

REFERENCE INFORMATION
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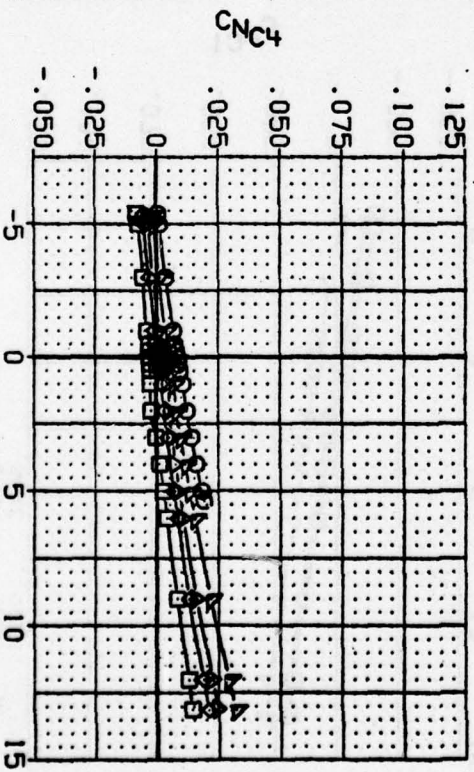
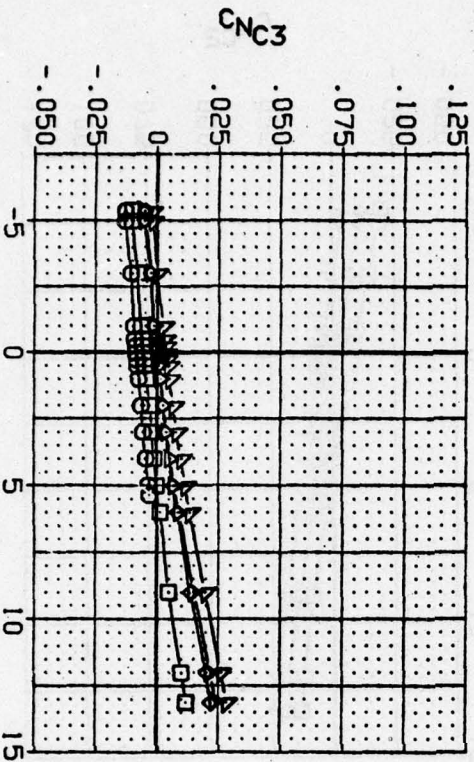
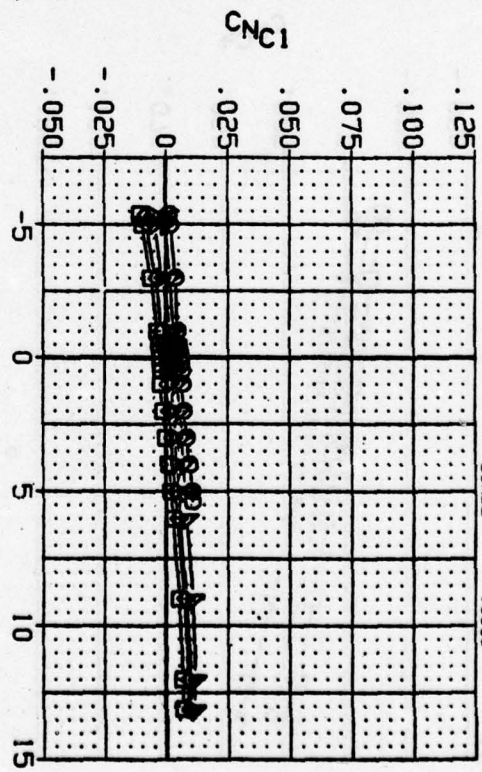
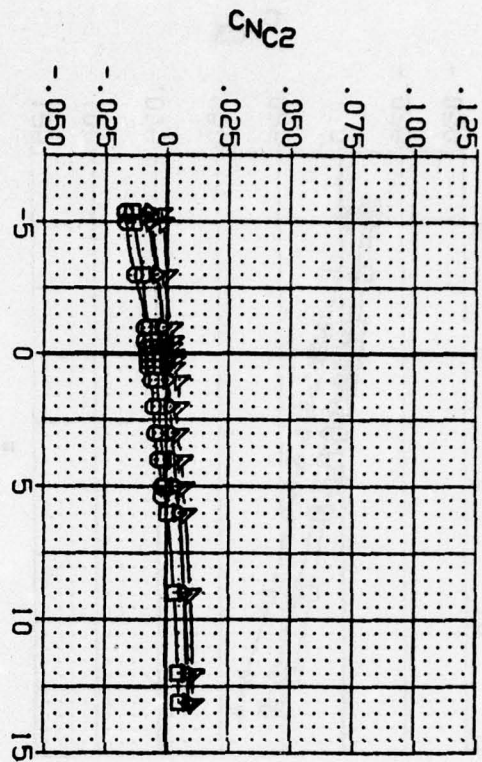


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH01) AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH02) AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH03) AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH04) AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH05) AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 5.000 -5.000 -5.000 5.000
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 1.000 1.000 1.000 1.000
 3.000 3.000 3.000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

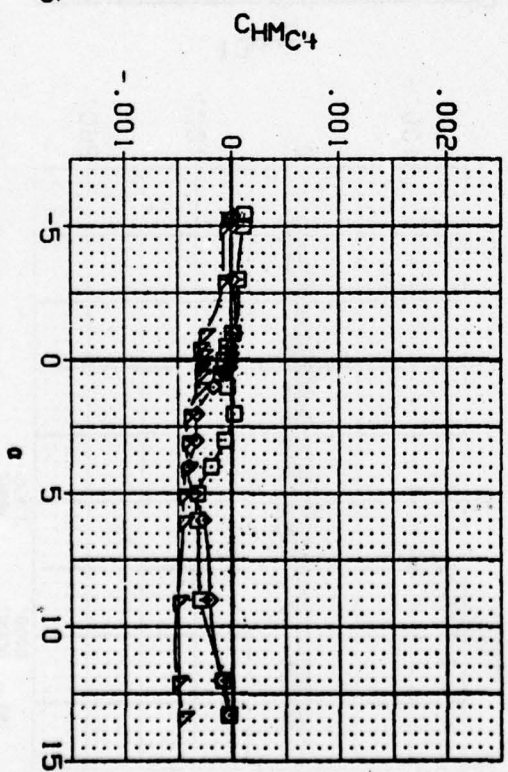
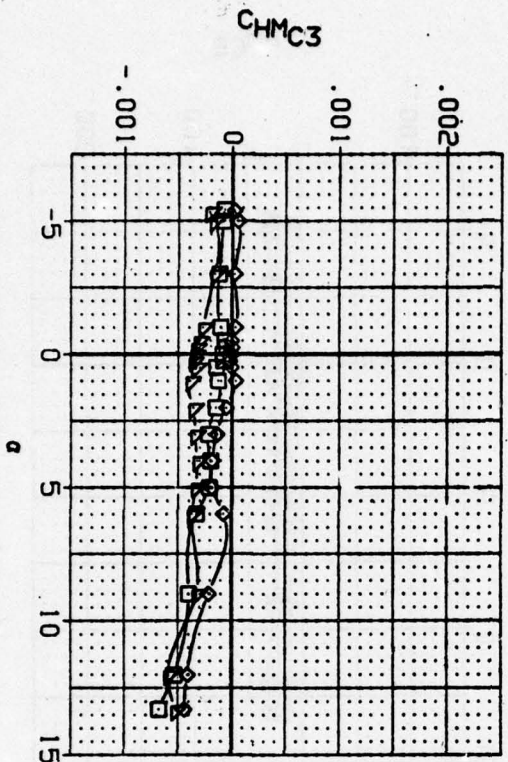
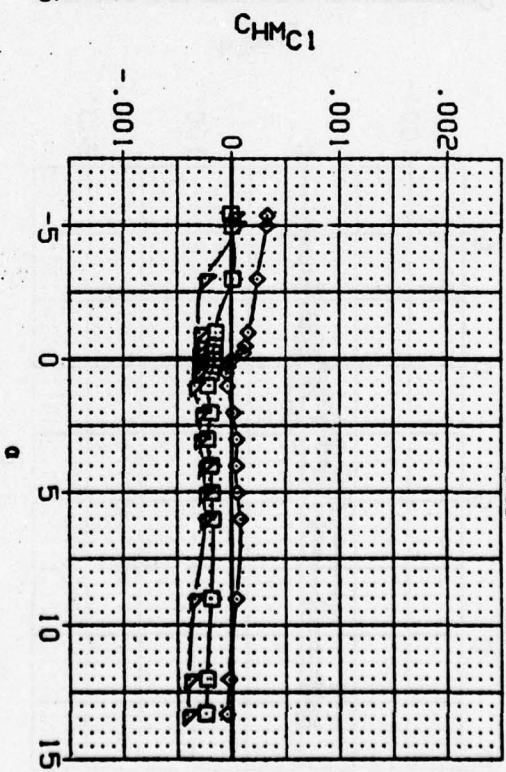
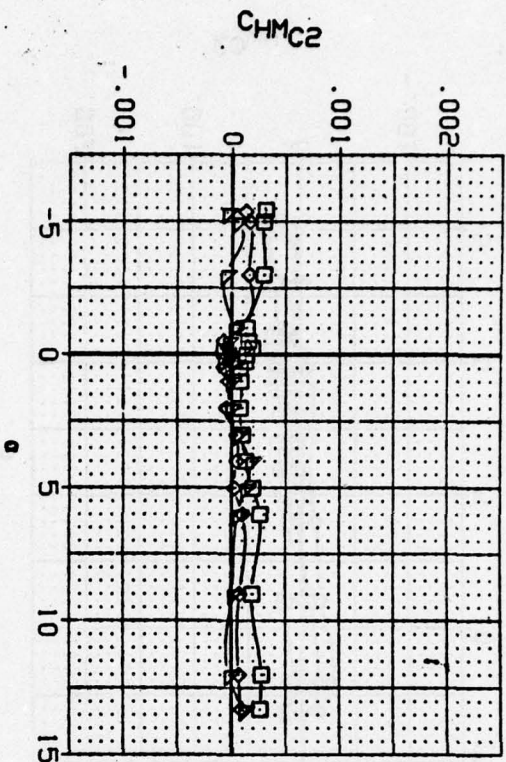
(BXH041) DATA NOT AVAILABLE
 (BXH042) AEDC WJIA-CIA, CANARD CONTROL, BNIC11
 (BXH043) AEDC WJIA-CIA, CANARD CONTROL, BNIC11
 (BXH044) DATA NOT AVAILABLE
 (BXH045) AEDC WJIA-CIA, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4

-5.000 -5.000 -5.000 -5.000
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHIAL=45 PHICND=45

(A)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

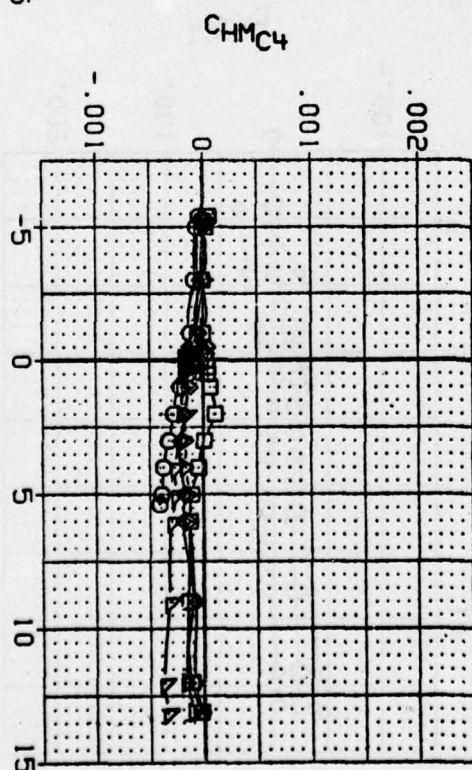
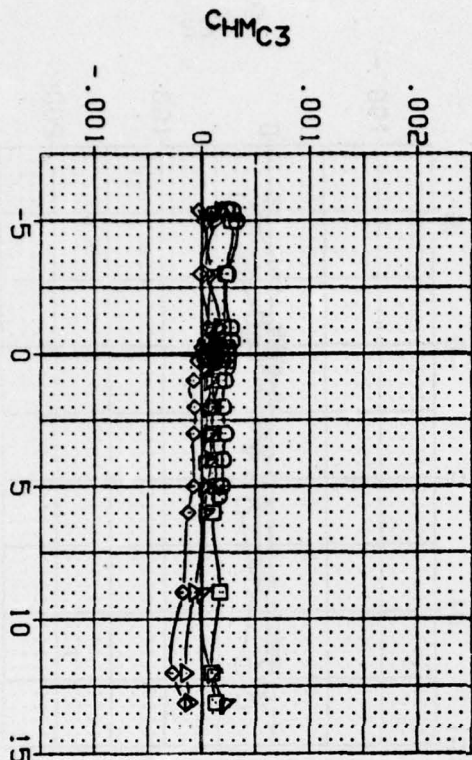
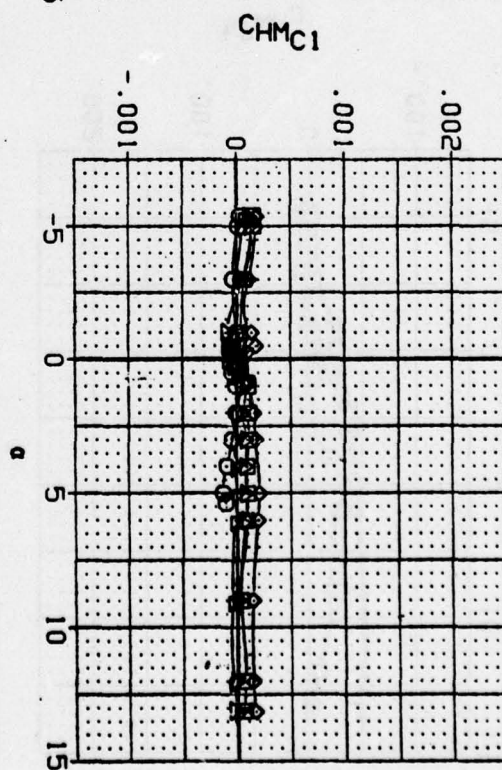
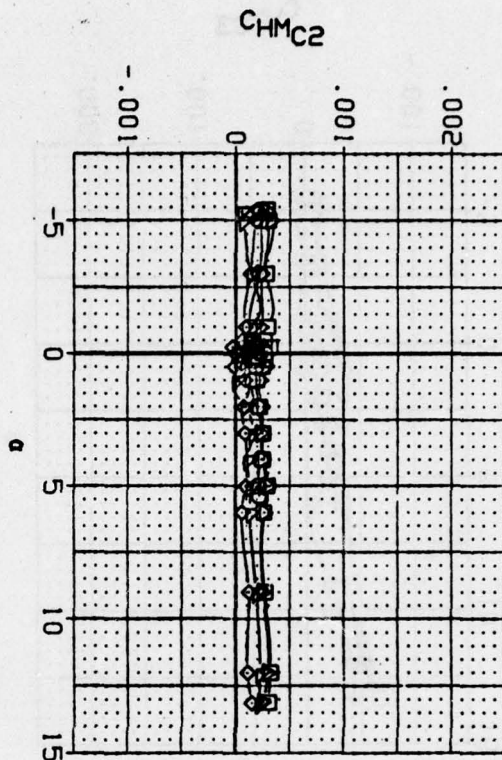
(BXH041)	○	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(BXH042)	□	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(BXH043)	△	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(BXH044)	◇	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(BXH045)	▽	AEDC W1A-C1A, CANARD CONTROL, BNICITI

CCND1 DCND2 DCND3 DCND4

3.000	-5.000	-5.000	5.000
-3.000	-3.000	-3.000	-3.000
1.000	1.000	1.000	1.000
3.000	3.000	3.000	3.000

REFERENCE INFORMATION

SREF	19.6350	50. IN.
REF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	0.0000	IN.
ZREF	0.0000	IN.
SCALE	0.0000	IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

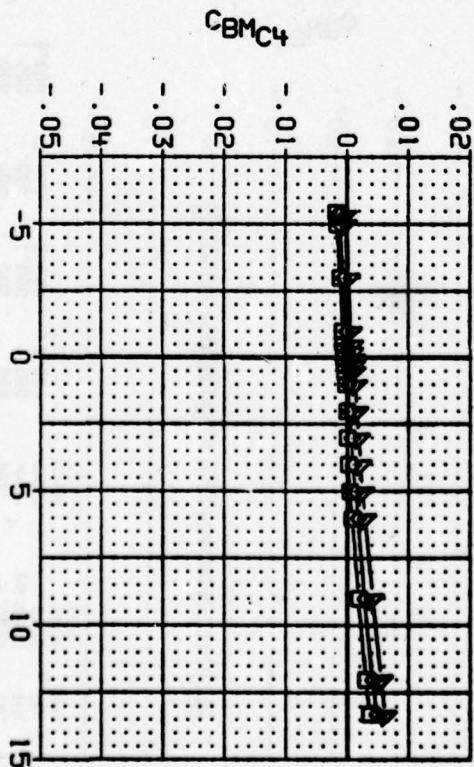
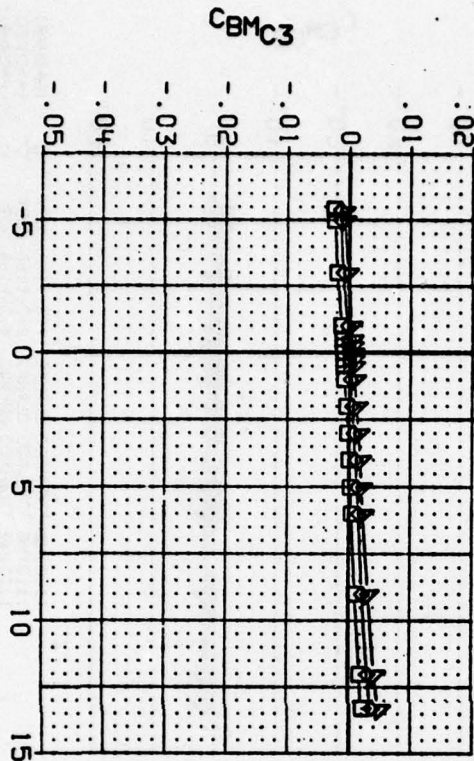
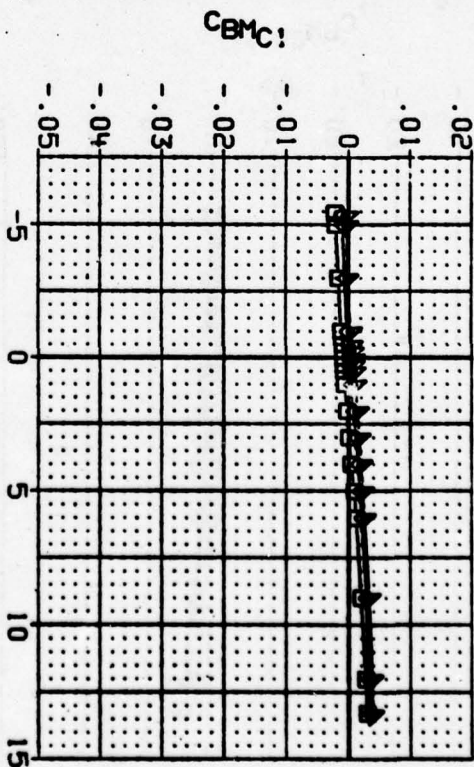
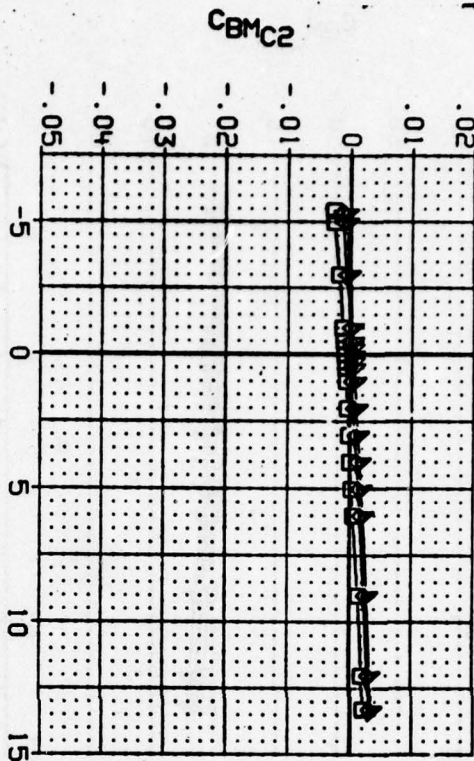
(BXH041) DATA NOT AVAILABLE
 (BXH042) AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXH043) AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (BXH044) DATA NOT AVAILABLE
 (BXH045) AEDC WIA-CIA, CANARD CONTROL, BNICITI

DOCD1 DOCD2 DOCD3 DOCD4

5.000 -5.000 -5.000 -5.000
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 1.000 1.000 1.000 1.000
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REFERENCE INFORMATION

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 LREF 5.0000 IN.
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 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

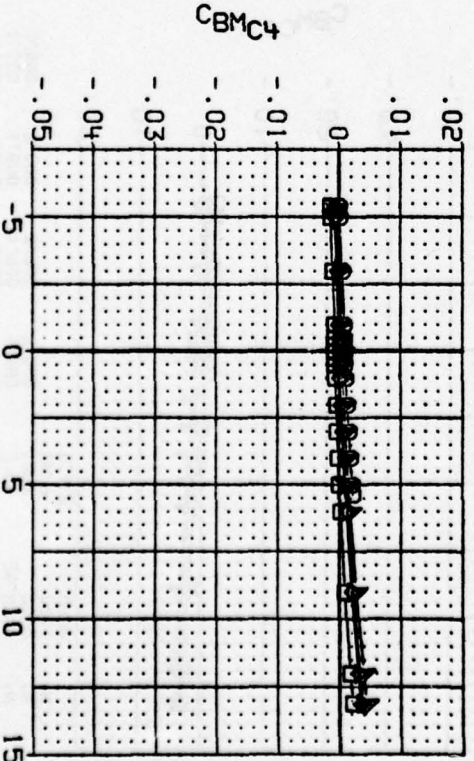
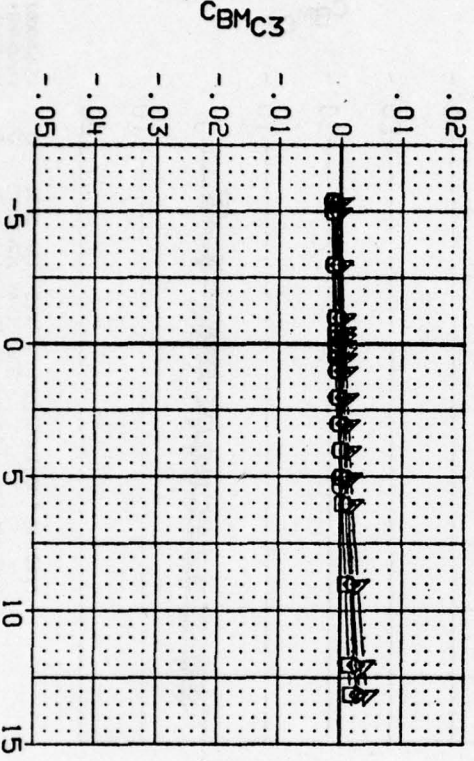
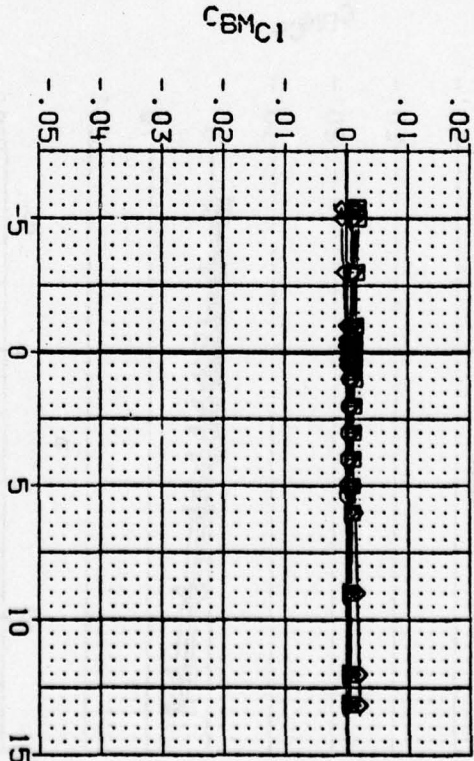
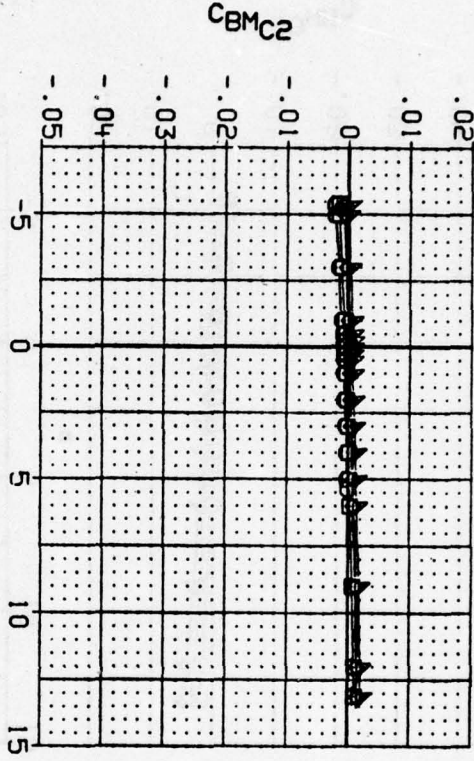
PHITAL=45 PHICND=45

(A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
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(BXH042)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC11
(BXH043)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC11
(BXH044)	▽	AEDC V41A-C1A, CANARD CONTROL, BNIC11
(BXH045)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC11

DCND1	DCND2	DCND3	DCND4
5.000	-5.000	-5.000	-5.000
-3.000	-3.000	-3.000	-3.000
.000	.000	.000	.000
1.000	1.000	1.000	1.000
3.000	3.000	3.000	3.000

REFERENCE INFORMATION	
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BREF	5.0000
MREF	26.0000
ZREF	.0000
SCALE	.0000
	IN.

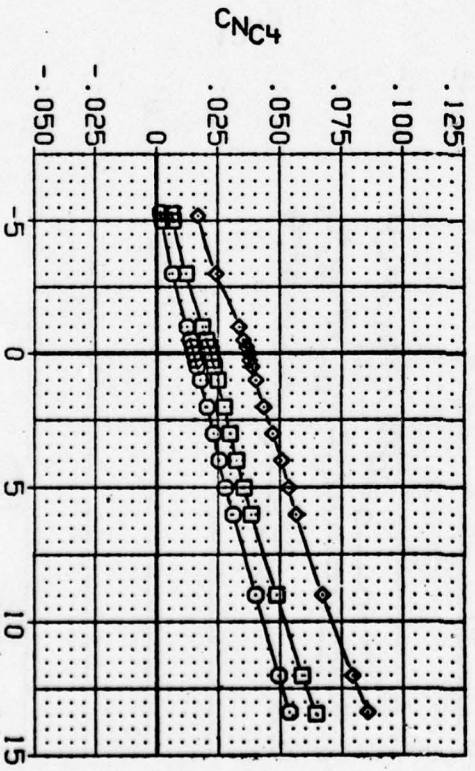
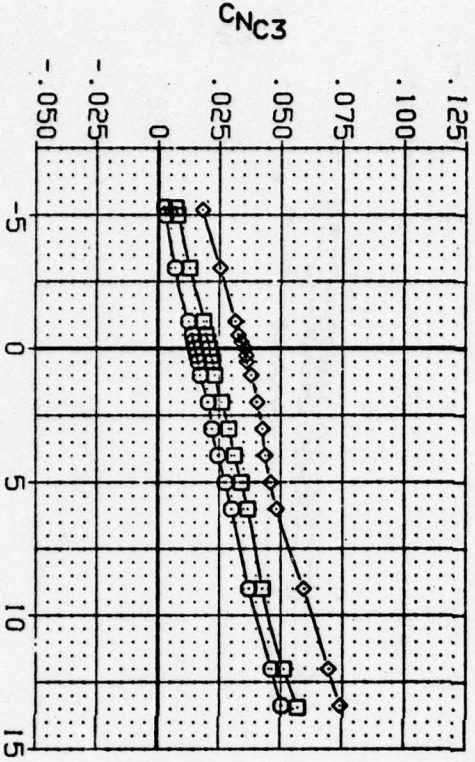
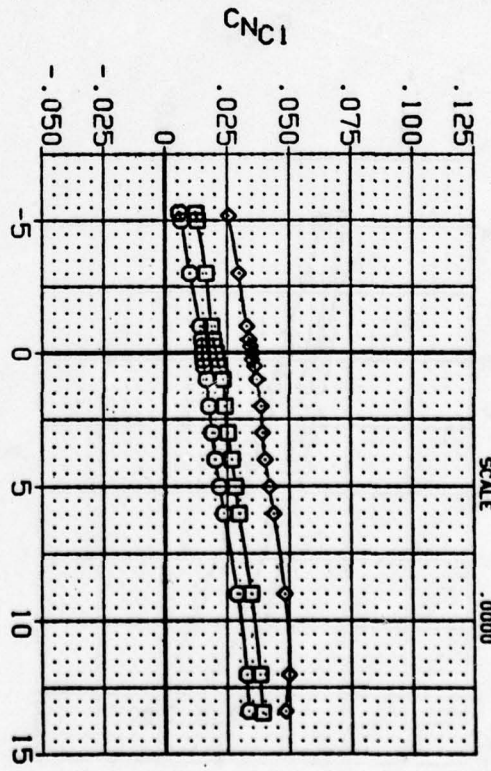
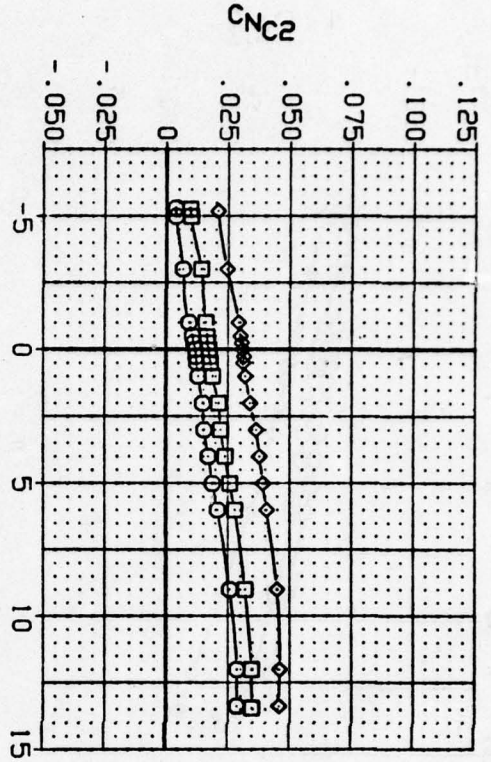


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45
 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH046) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH047) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH048) \diamond AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1 6.000 DCND2 6.000 DCND3 6.000 DCND4 6.000
 15.000 9.000 15.000 9.000 15.000 9.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
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 SCALE .0000

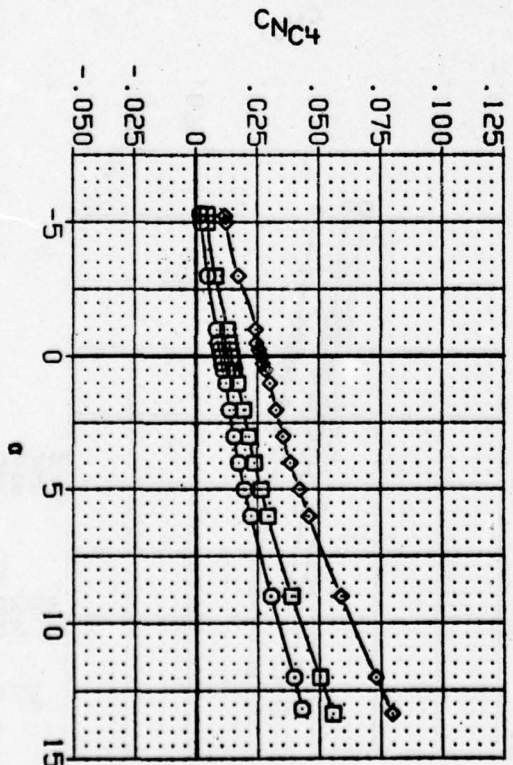
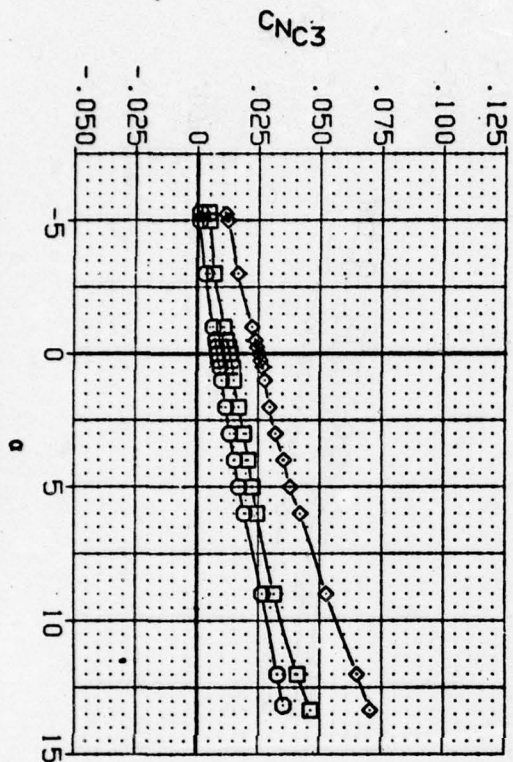
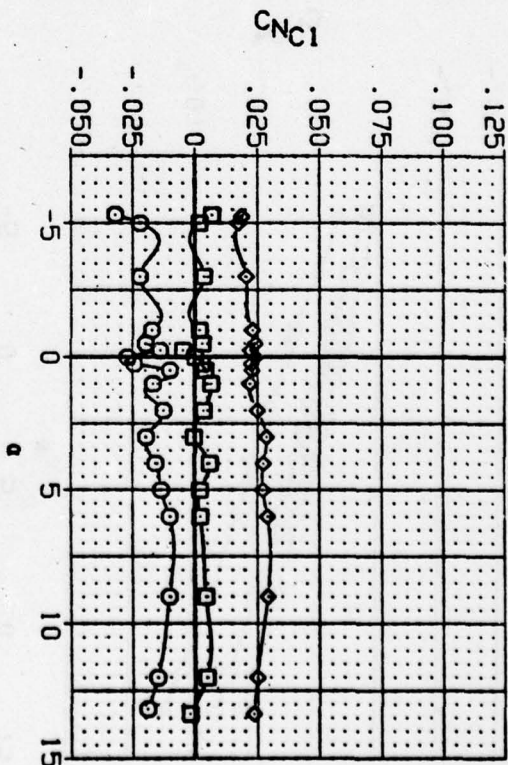
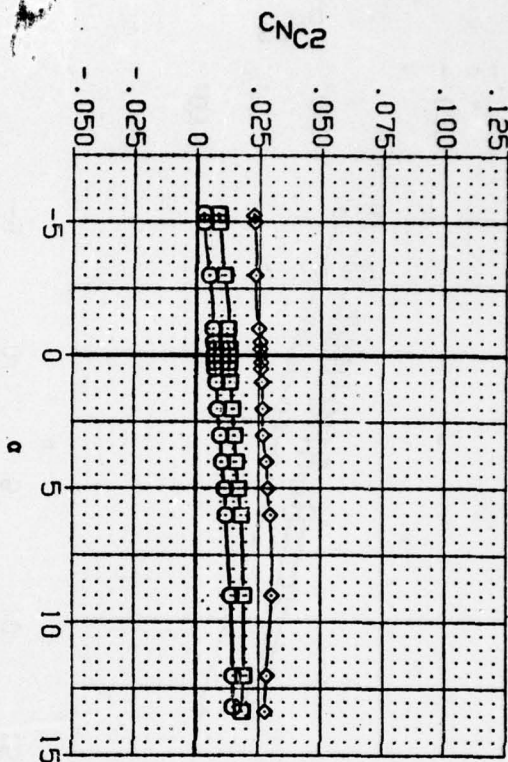


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHIAL=45$ $PHICND=45$
 $(A)MACH = 3.00$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH046) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (AXH047) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (AXH048) \diamond AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
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 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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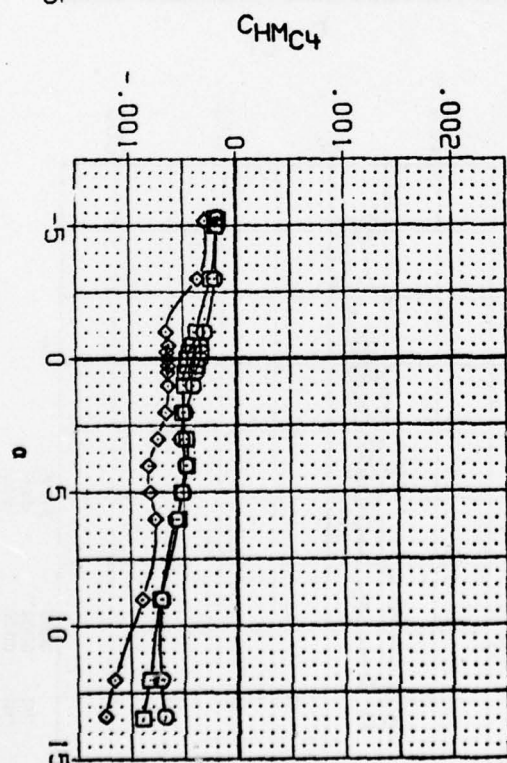
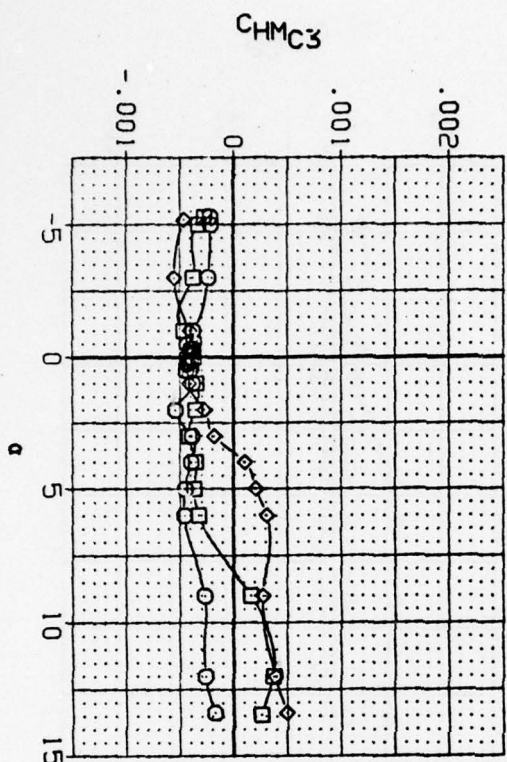
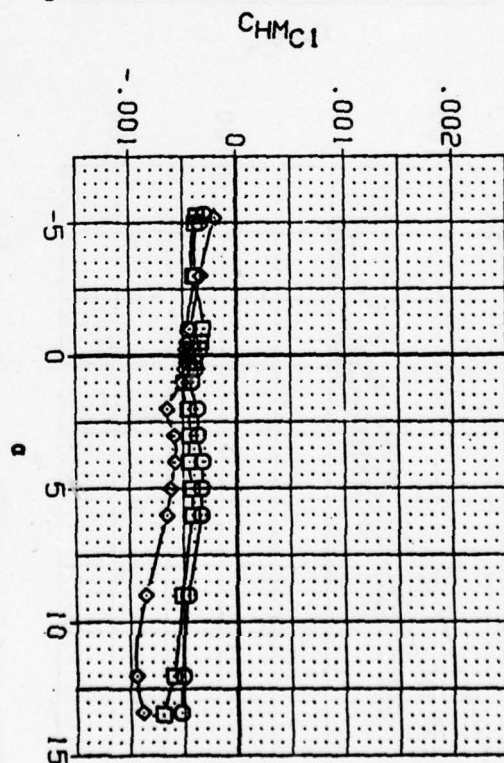
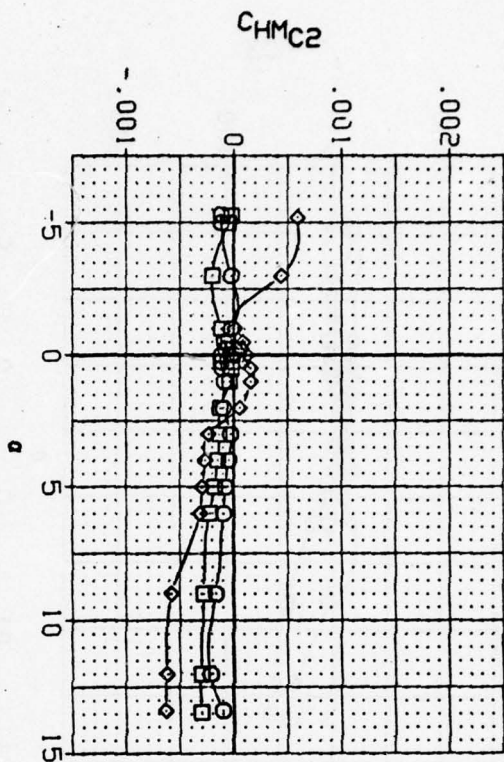


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{HIAL} = 45$ $\Phi_{HICND} = 45$
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH046) \square AEDC V1A-C1A, CANARD CONTROL, BNIC171
 (BXH047) \square AEDC V1A-C1A, CANARD CONTROL, BNIC171
 (BXH048) \diamond AEDC V1A-C1A, CANARD CONTROL, BNIC171

DCND1 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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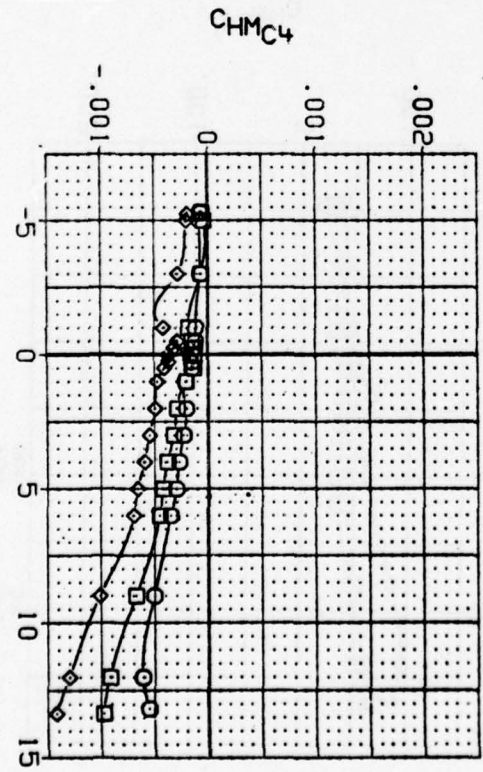
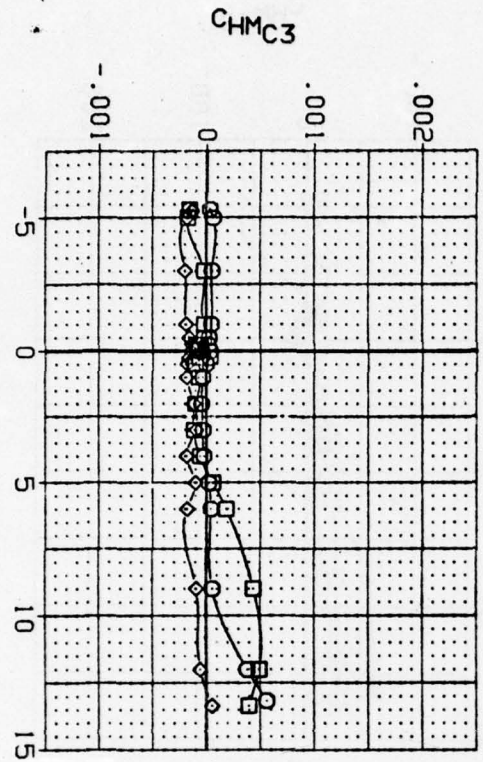
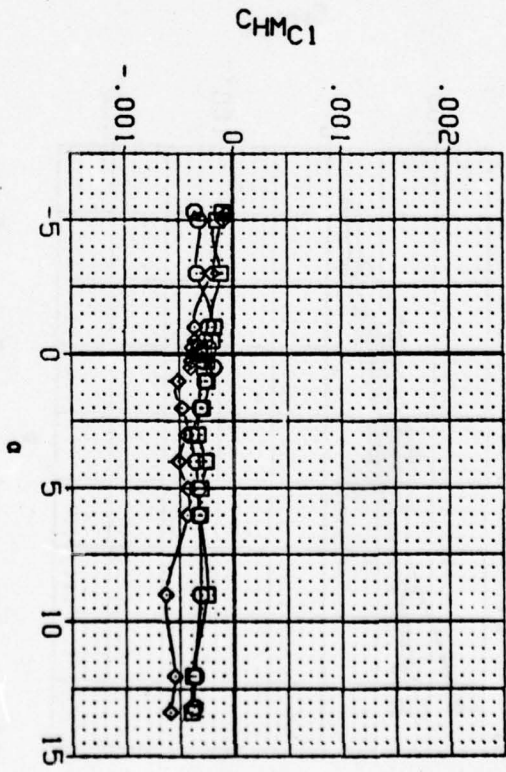
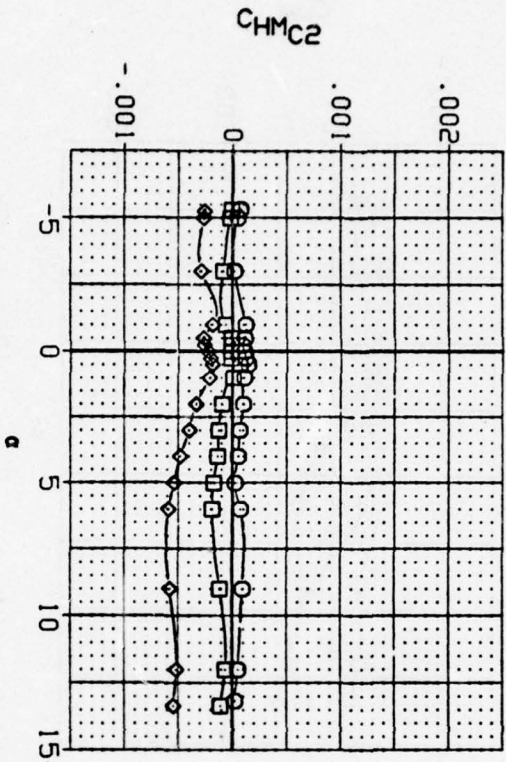


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH046) \square AEDC W1A-C1A, CANARD CONTROL, BNIC11
 (BXH047) \square AEDC W1A-C1A, CANARD CONTROL, BNIC11
 (BXH048) \diamond AEDC W1A-C1A, CANARD CONTROL, BNIC11

DCND1 5.000 6.000 9.000 15.000
 DCND2 5.000 6.000 9.000 15.000
 DCND3 5.000 6.000 9.000 15.000
 DCND4 5.000 6.000 9.000 15.000

REFERENCE INFORMATION
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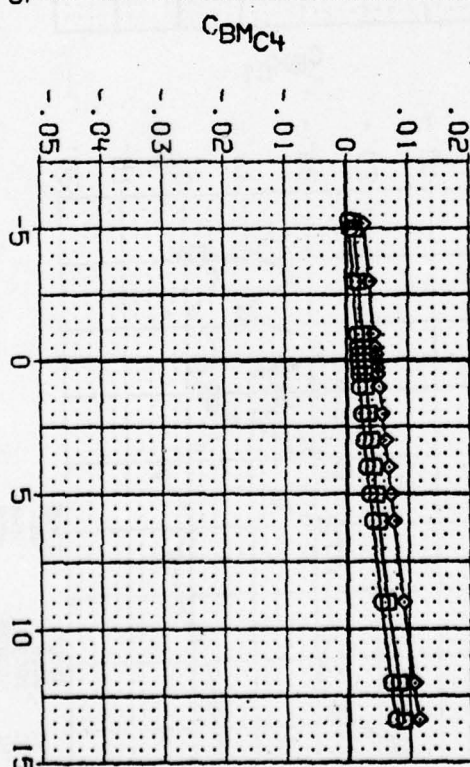
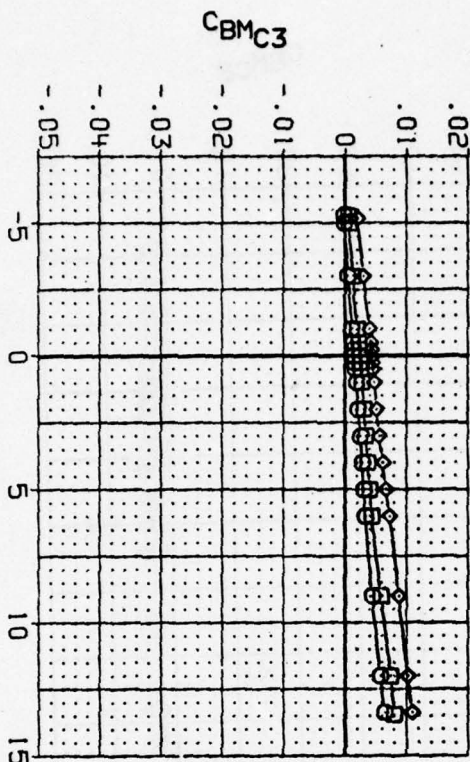
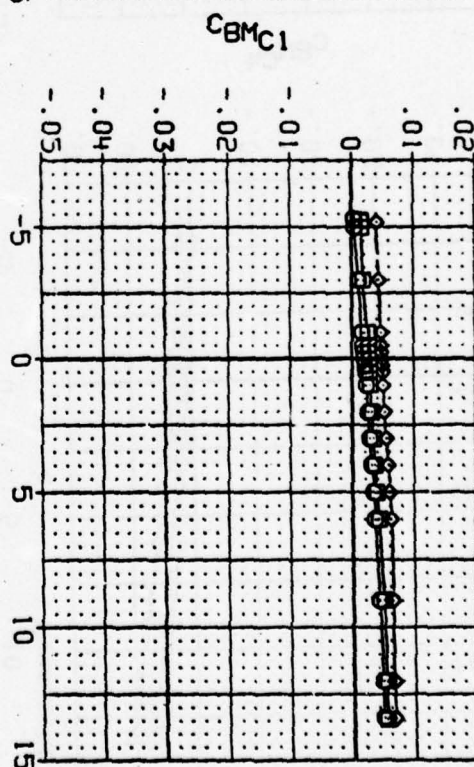
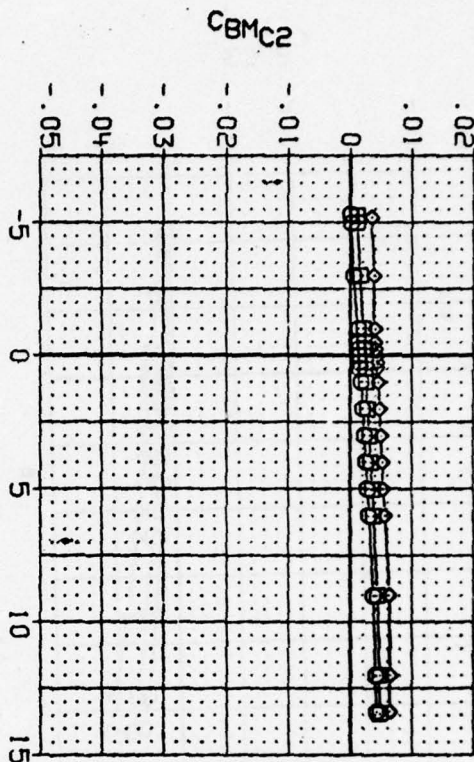


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH046) \square AEDC W1A-C1A, CANARD CONTROL, BNICIT1
 (BXH047) \square AEDC W1A-C1A, CANARD CONTROL, BNICIT1
 (BXH048) \diamond AEDC W1A-C1A, CANARD CONTROL, BNICIT1

DCND1 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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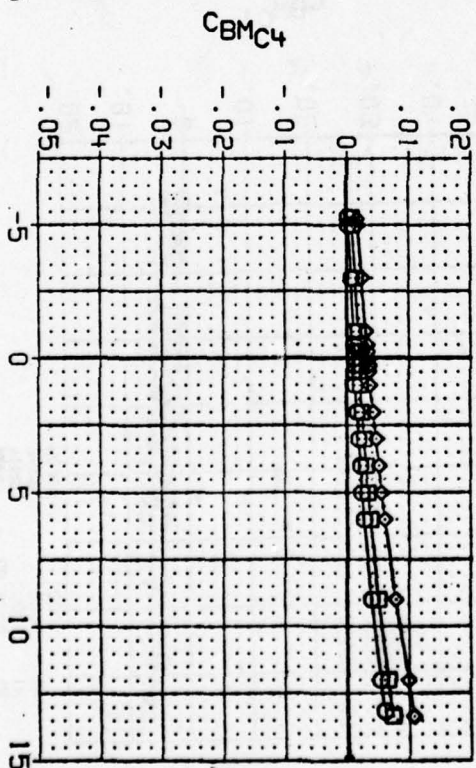
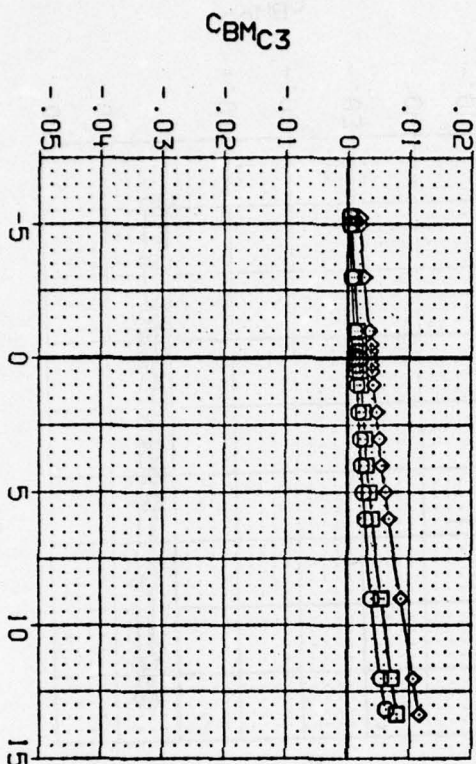
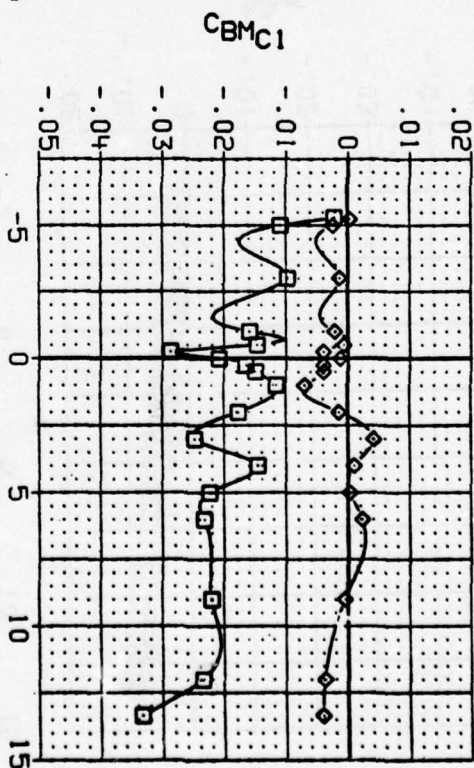
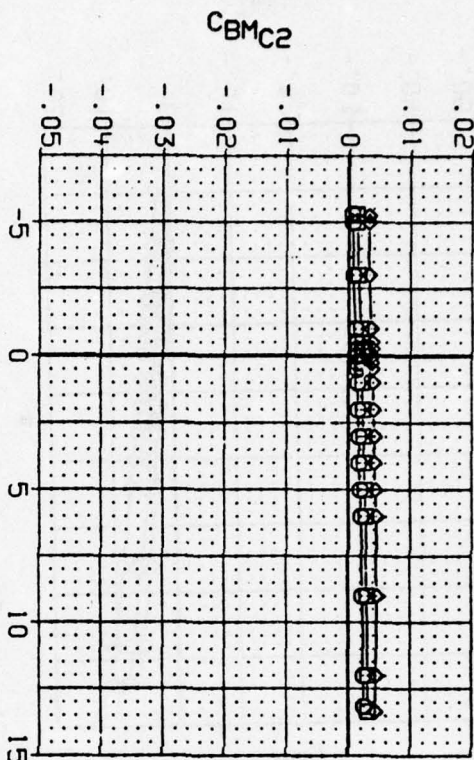


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{H1AL}=45$ $\Phi_{H1CND}=45$
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH046) AEDC WJIA-CIA, CANARD CONTROL, BNICITI
 (BXH047) AEDC WJIA-CIA, CANARD CONTROL, BNICITI
 (BXH048) AEDC WJIA-CIA, CANARD CONTROL, BNICITI

DCND1 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YREF 26.0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITIAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

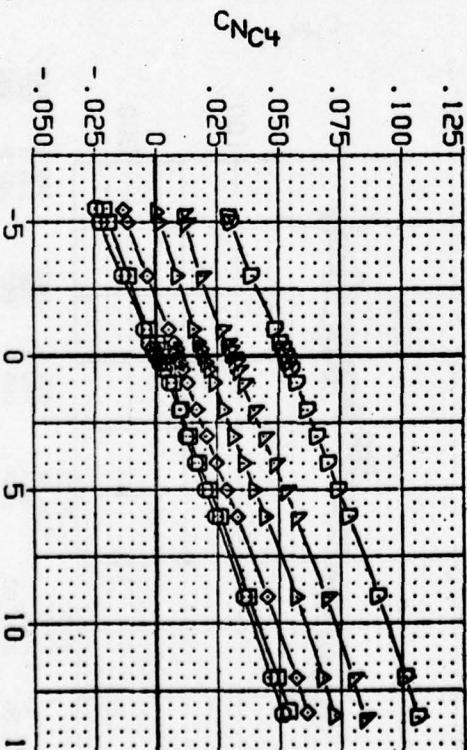
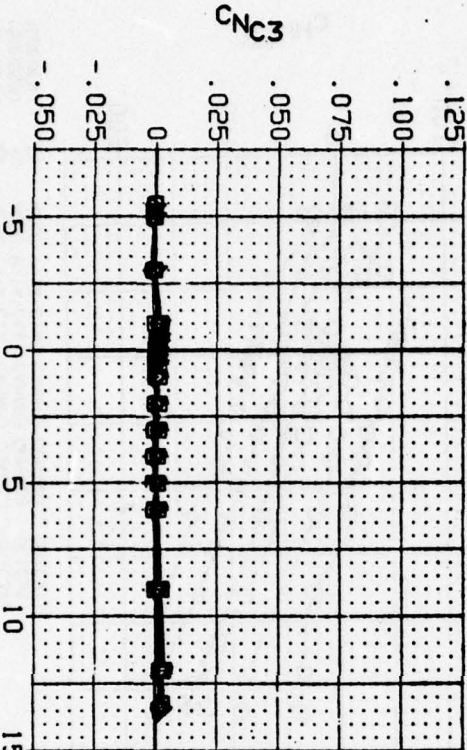
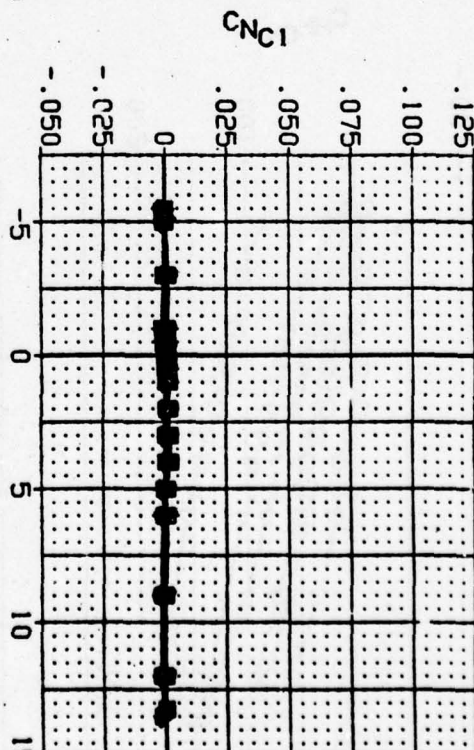
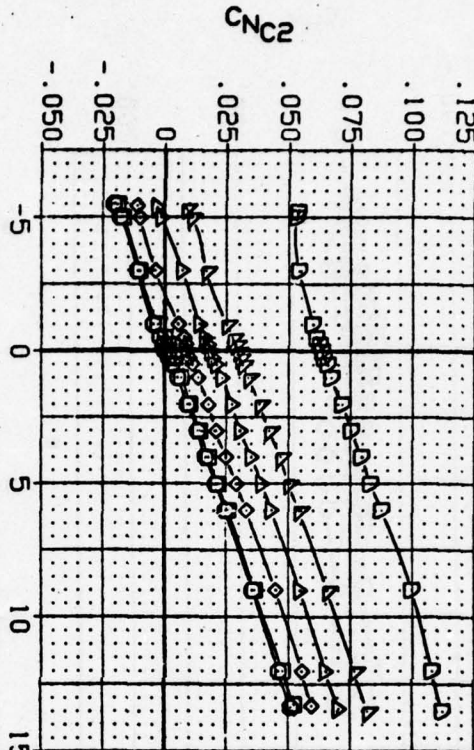
(AXH054) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH053) \triangle AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH052) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH051) \triangle AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH050) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH049) \triangle AEDC W1A-C1A, CANARD CONTROL, BNIC3T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 1.000 .000 .000
 .000 3.000 .000 .000
 .000 6.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION

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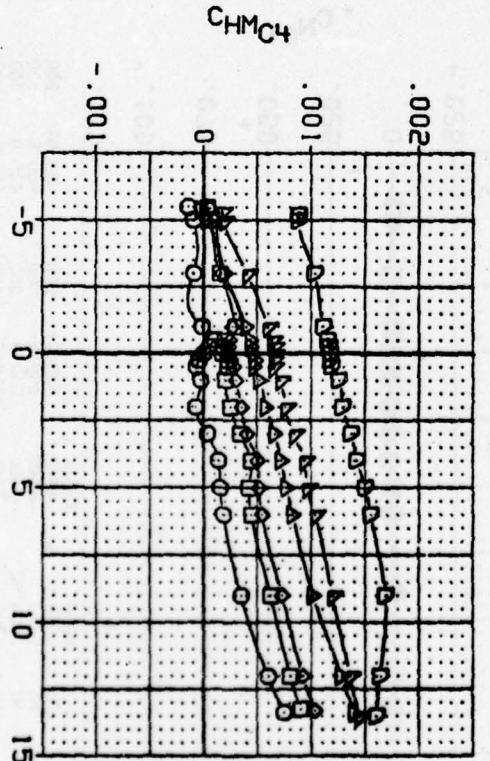


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

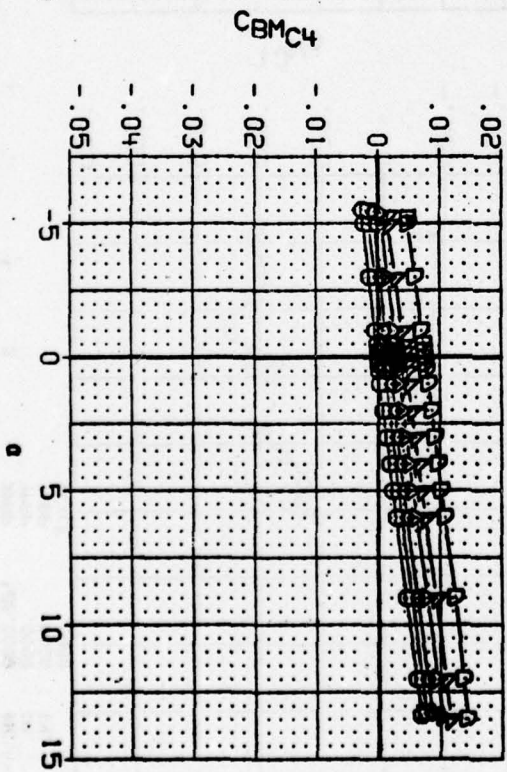
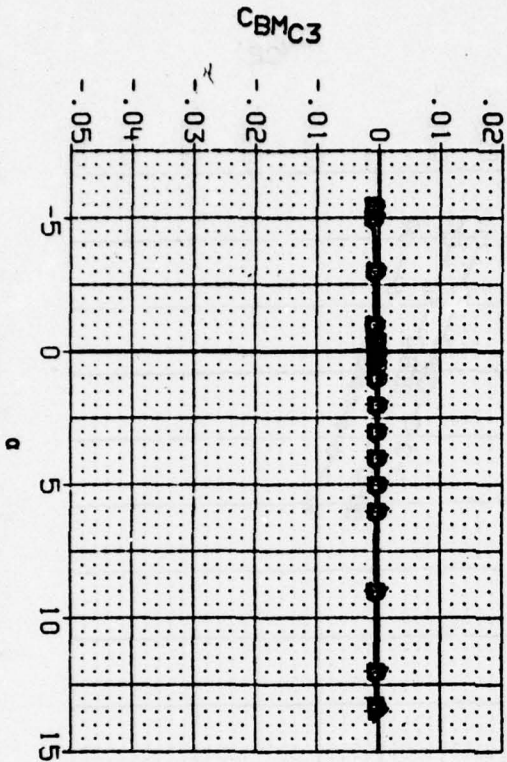
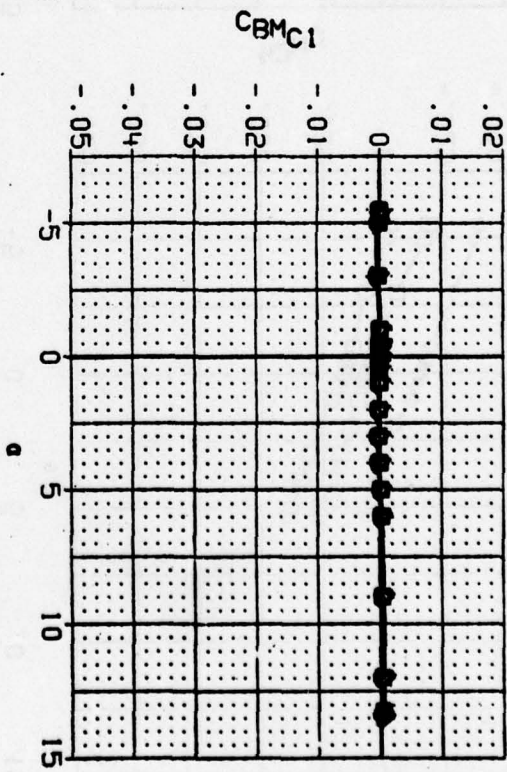
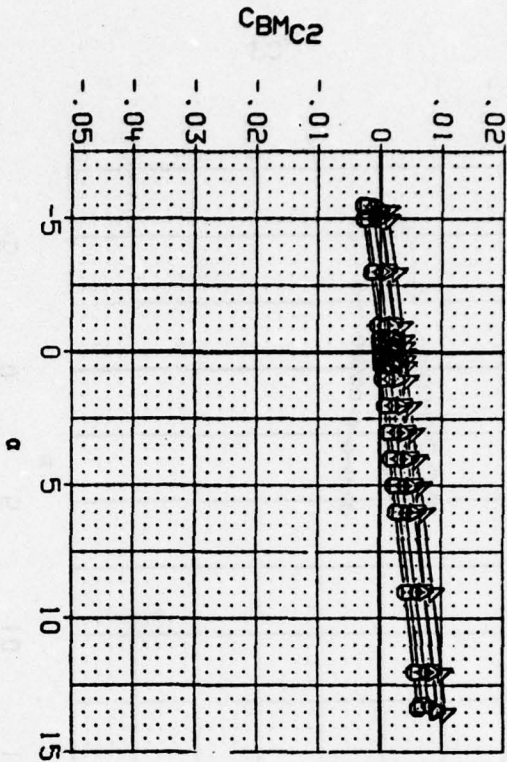
(A) MACH = 3.01

REFERENCE INFORMATION			
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.000	9.000	.000	9.000
.000	15.000	.000	15.000
SREF 19.6350 SQ. IN.			
LREF 5.0000 IN.			
BREF 5.0000 IN.			
MGRP 26.0000 IN.			
YGRP IN.			
ZGRP IN.			
SCALE .0000 IN.			



DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(BXH0491)	○	AEDC V41A-C1A, CANARD CONTROL, BN1C311
(BXH0501)	◇	AEDC V41A-C1A, CANARD CONTROL, BN1C311
(BXH0511)	△	AEDC V41A-C1A, CANARD CONTROL, BN1C311
(BXH0521)	▽	AEDC V41A-C1A, CANARD CONTROL, BN1C311
(BXH0531)	□	AEDC V41A-C1A, CANARD CONTROL, BN1C311
(BXH0541)	◇	AEDC V41A-C1A, CANARD CONTROL, BN1C311

DOAND1	DOAND2	DOAND3	DOAND4	REFERENCE INFORMATION
.000	.000	.000	.000	SREF 19.6350 50. IN.
.000	1.000	.000	.000	LREF 5.0000 IN.
.000	3.000	.000	.000	BREF 5.0000 IN.
.000	6.000	.000	.000	26.0000 IN.
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.000	15.000	.000	.000	26.0000 IN.
				SCALE .0000

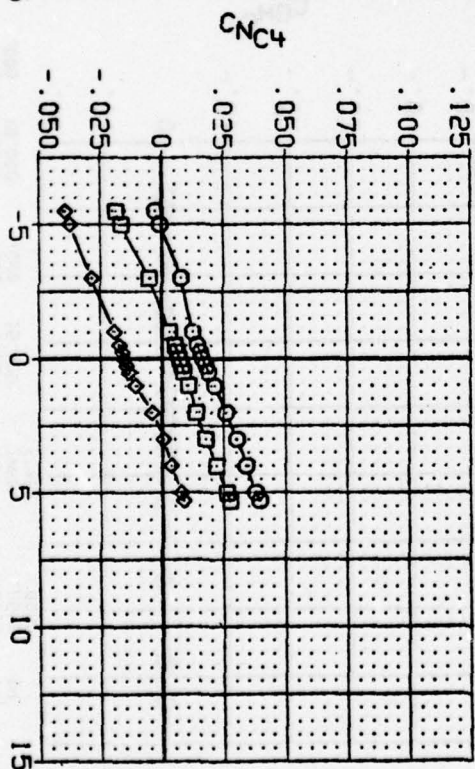
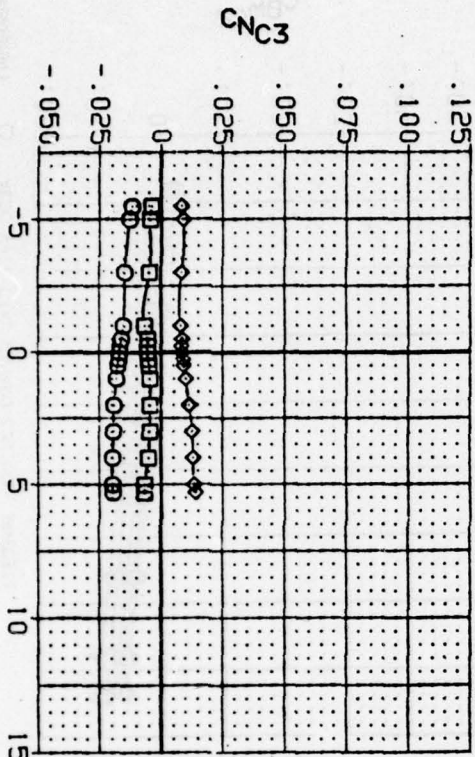
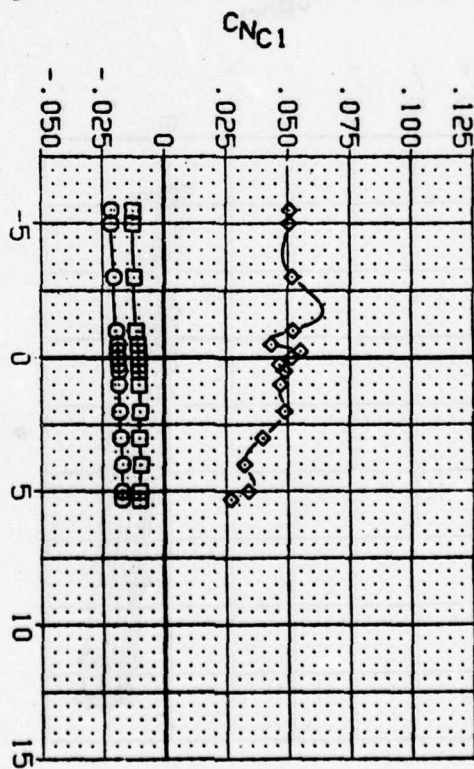
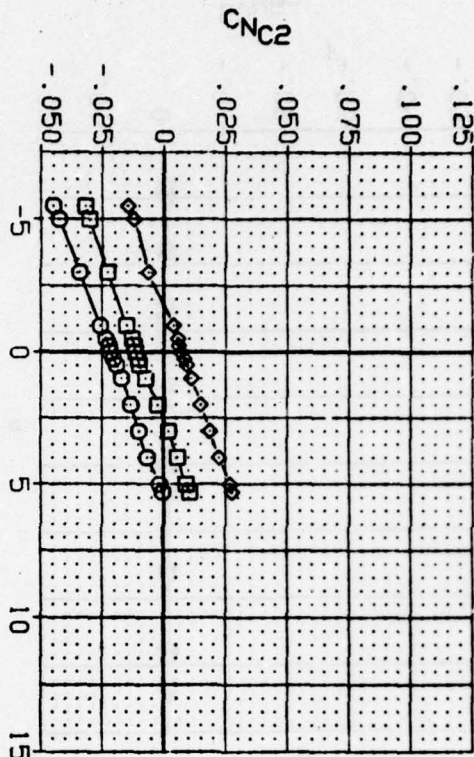


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH055) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH056) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH057) \diamond AEDC W1A-C1A, CANARD CONTROL, BNIC3T1

DCND1 DCND2 DCND3 DCND4
 5.000 -5.000 -5.000 5.000
 -3.000 -2.000 -2.000 2.000
 -3.000 3.000 3.000 -3.000

REFERENCE INFORMATION
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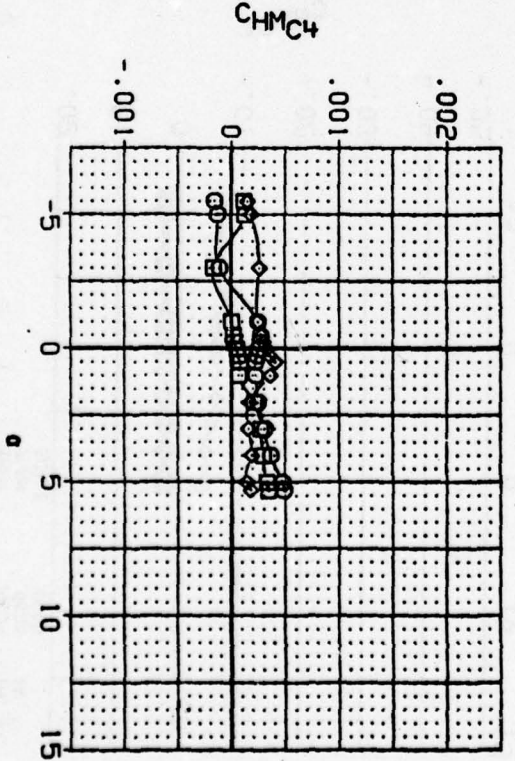
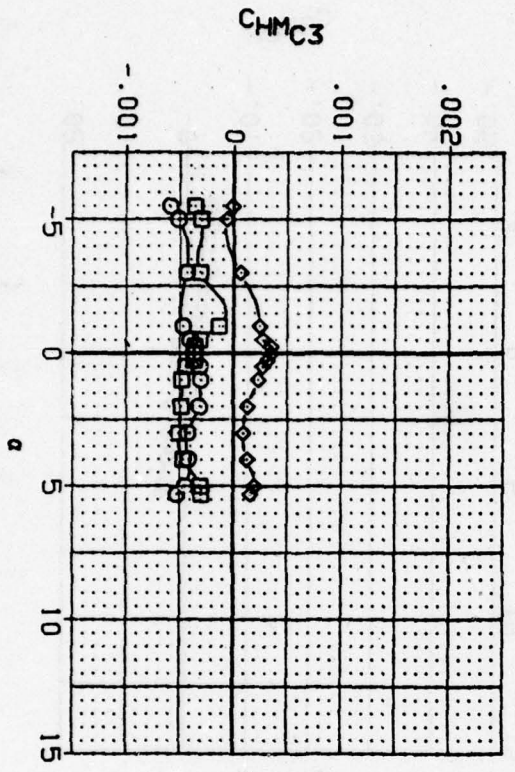
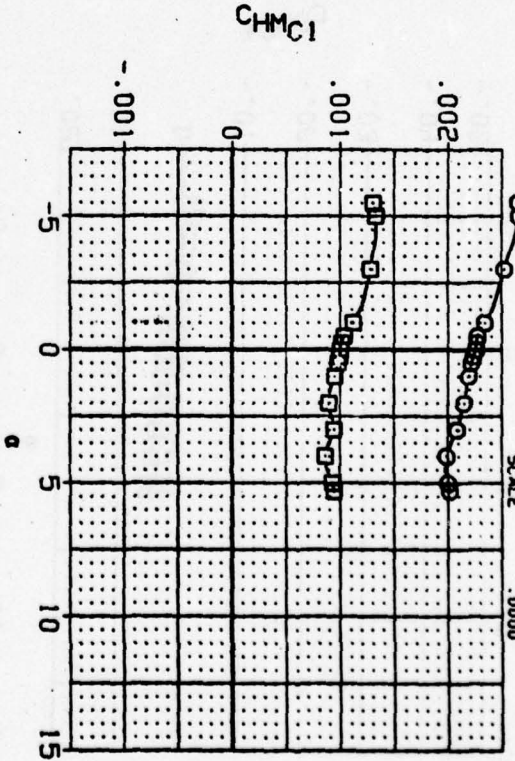
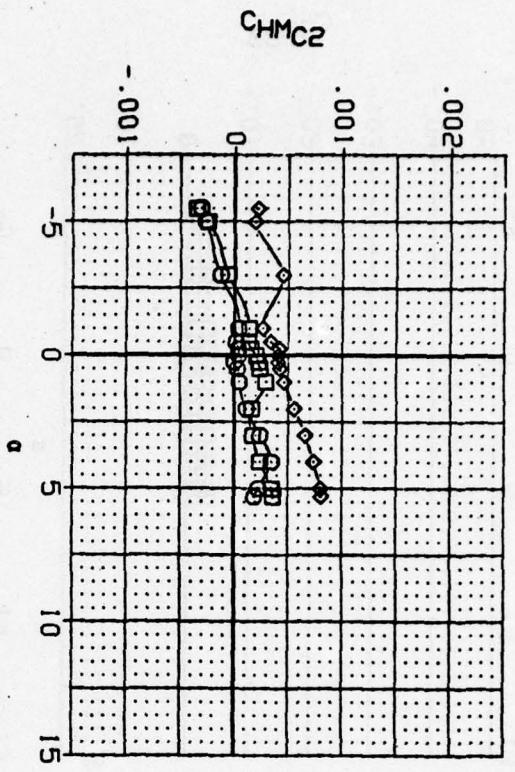


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{H1A}=0$ $\Phi_{H1CND}=0$
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BX4055) \square AEDC W1A-C1A, CANARD CONTROL, BN1C3T1
 (BX4056) \diamond AEDC W1A-C1A, CANARD CONTROL, BN1C3T1
 (BX4057) \square AEDC W1A-C1A, CANARD CONTROL, BN1C3T1

DCND1 5.000
 DCND2 -5.000
 DCND3 -2.000
 DCND4 3.000
 DCND5 5.000
 DCND6 2.000
 DCND7 -3.000

REFERENCE INFORMATION
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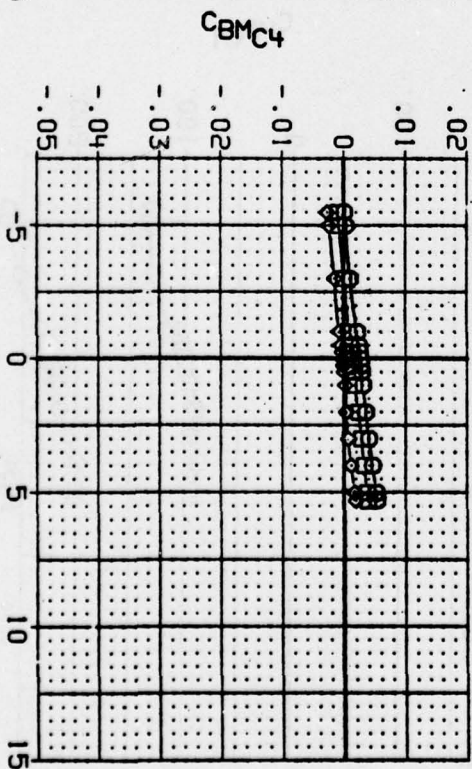
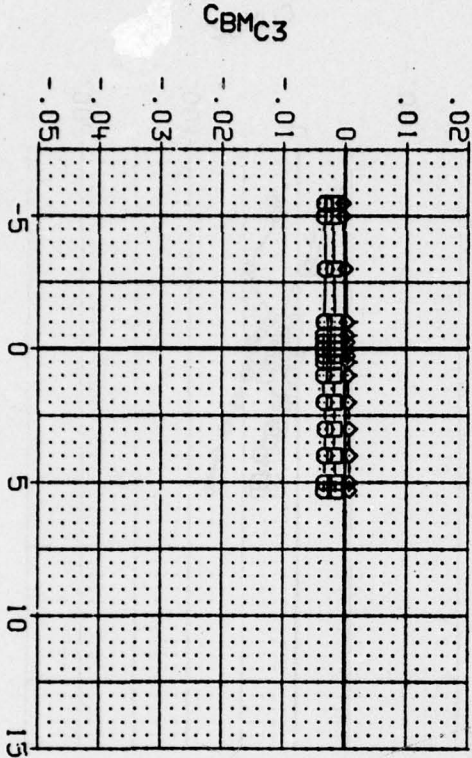
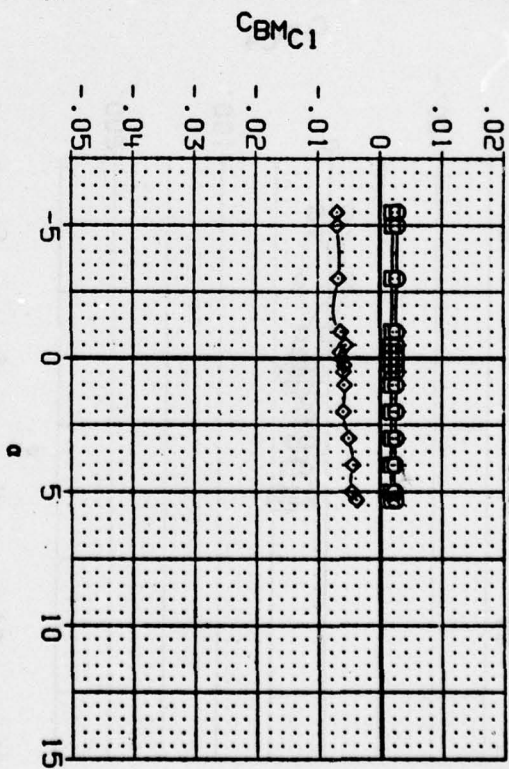
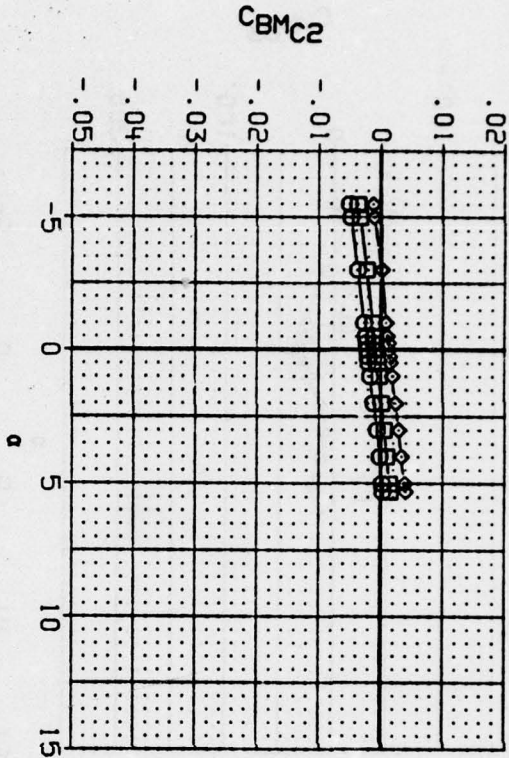


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PH1AL=0 PH1CND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH9551) \square AEDC WIA-CIA, CANARD CONTROL, BNIC3T1
 (BXH9561) \square AEDC WIA-CIA, CANARD CONTROL, BNIC3T1
 (BXH9571) \diamond AEDC WIA-CIA, CANARD CONTROL, BNIC3T1

DCND1 5.000 DCND2 -5.000 DCND3 -5.000 DCND4 5.000
 2.000 -2.000 -2.000 2.000
 -3.000 3.000 3.000 -3.000

REFERENCE INFORMATION
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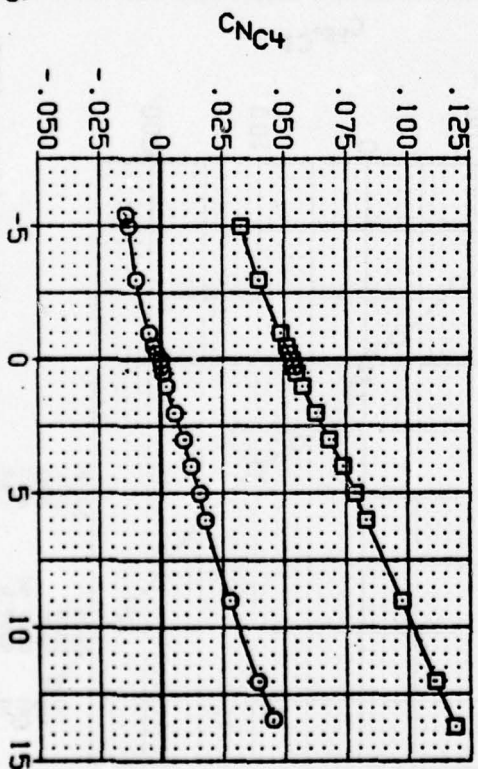
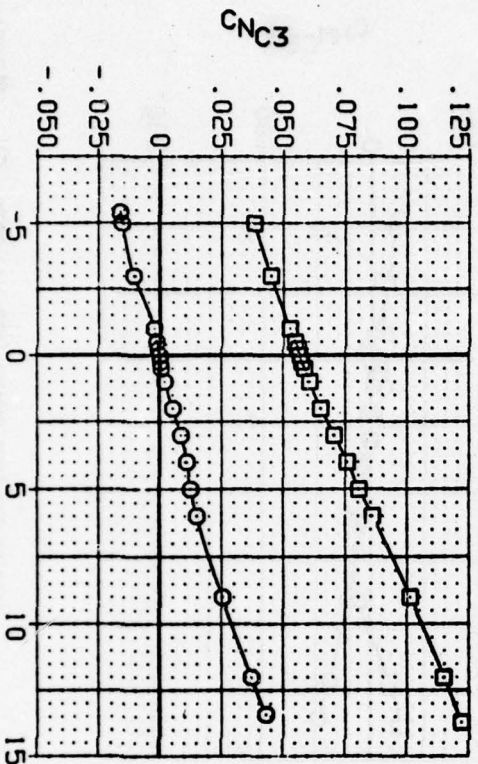
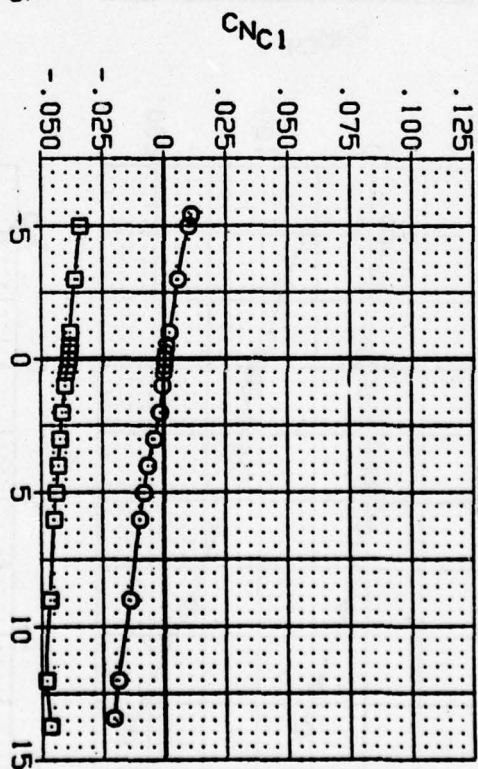
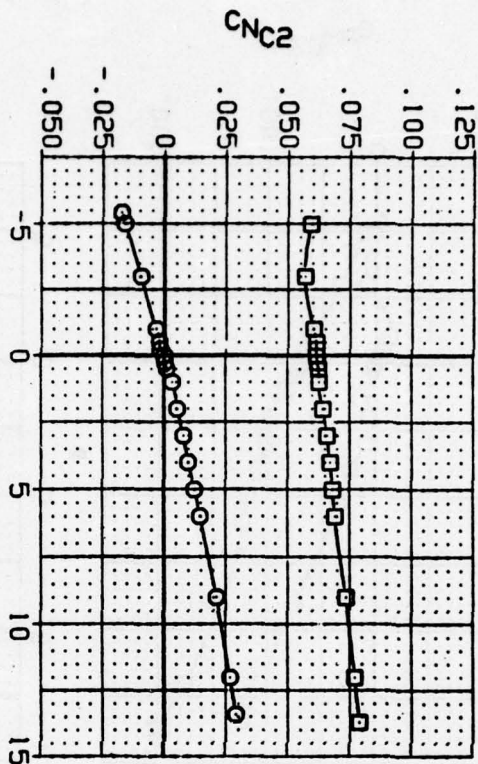


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHIAL=0$ $PHICND=0$
 $(A)MACH = 3.01$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (A)X058) ☐ AEDC V1A-C1A, CANARD CONTROL, ENIC311
 (A)X059) ☐ AEDC V1A-C1A, CANARD CONTROL, BNIC311

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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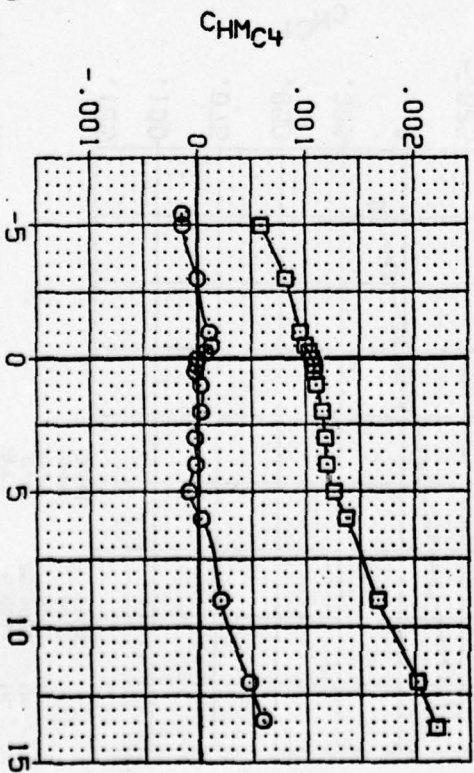
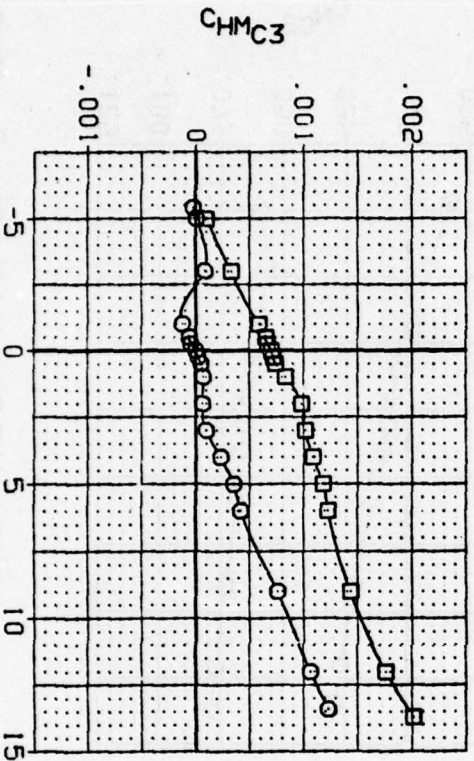
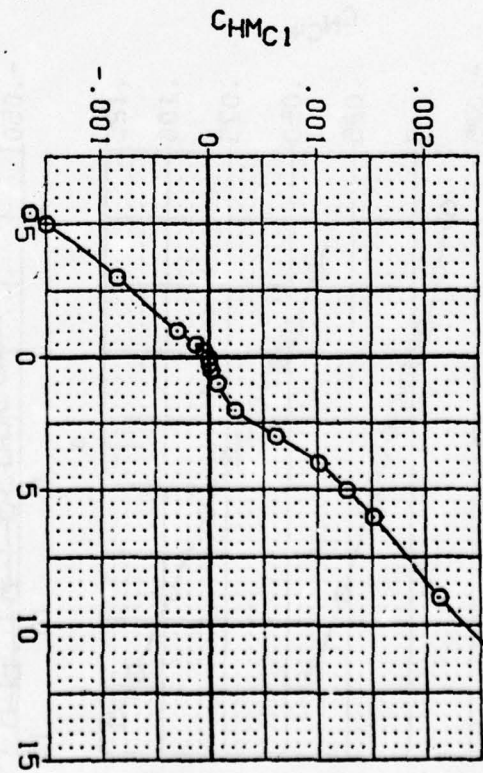
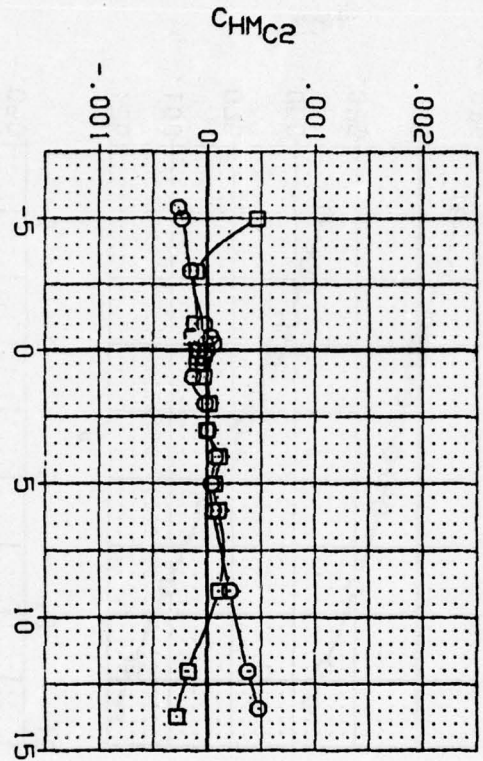


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHICAL=0 PHICND=45
 (A)MACH = 3.01

DATA SCT SYMBOL CONFIGURATION DESCRIPTION
 (BXH05B) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
 (BXH05J) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC3T1

DCND1 .003 DCND2 .000 DCND3 .000 DCND4 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 SCALE .0000

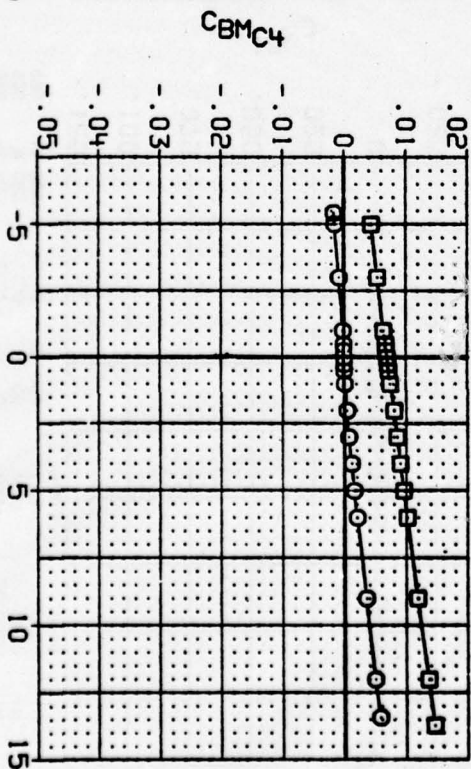
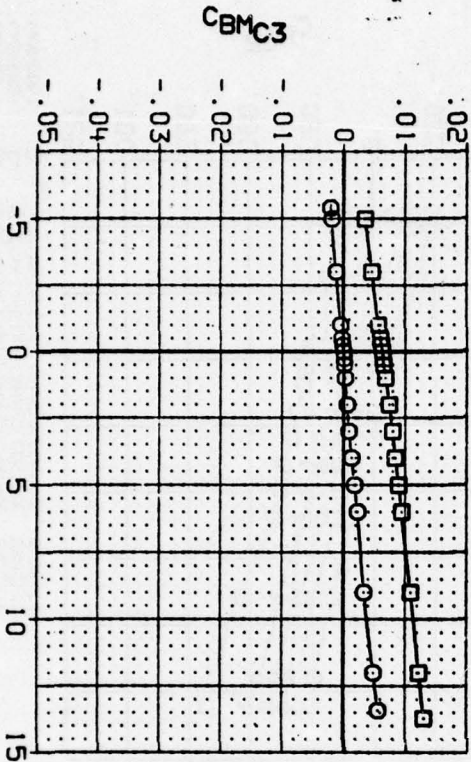
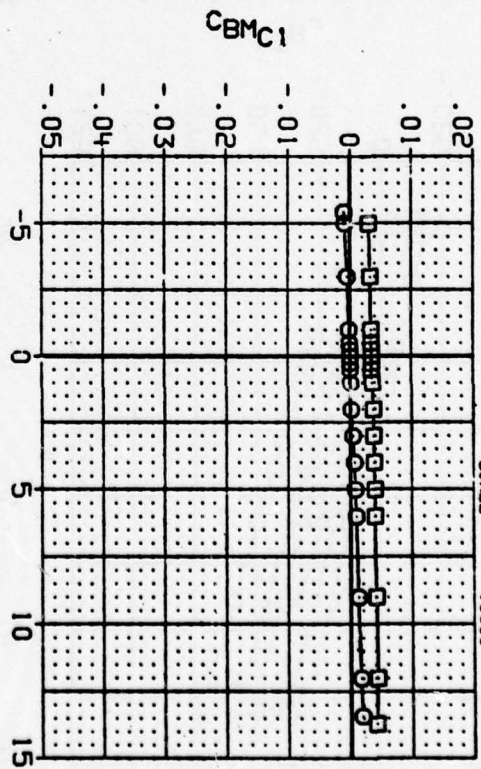
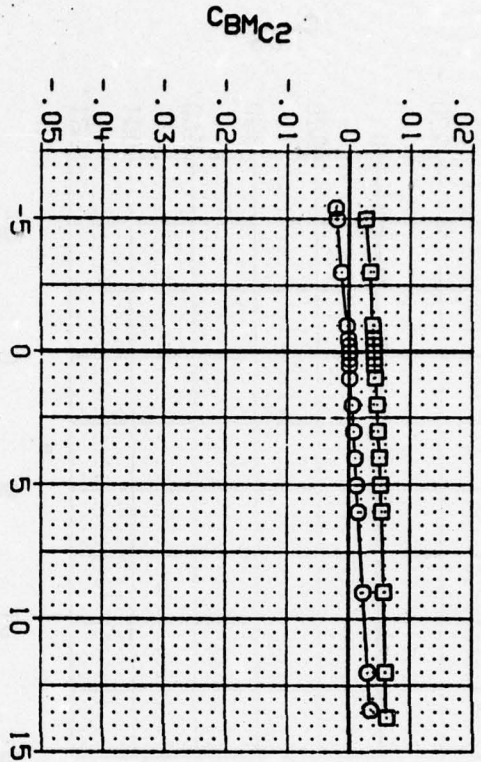


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=45
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH058) ☐ AEDC W1A-C1A, CANARD CONTROL, BN1C3T1
 (BXH059) ☐ AEDC W1A-C1A, CANARD CONTROL, BN1C3T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PH1AL=0 PHICND=45
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

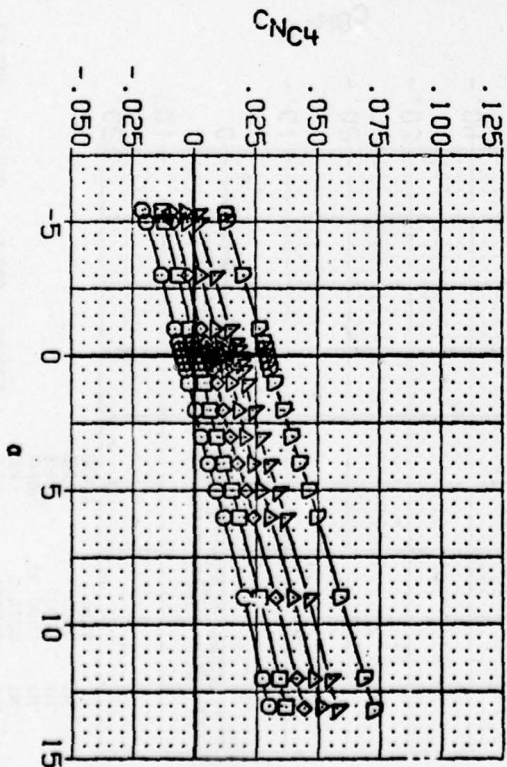
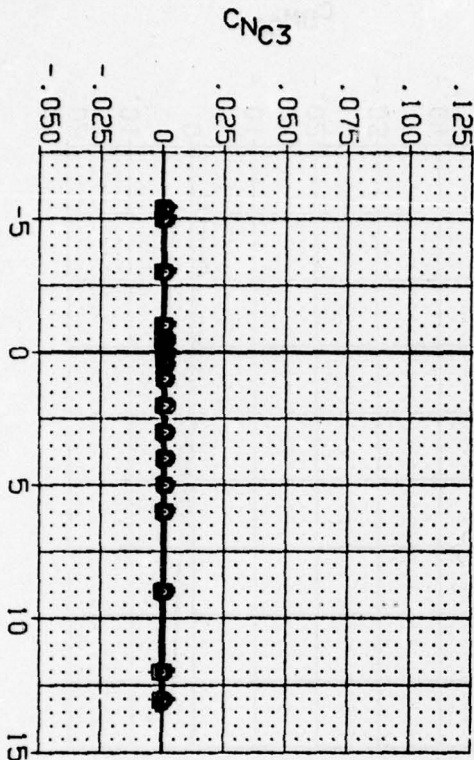
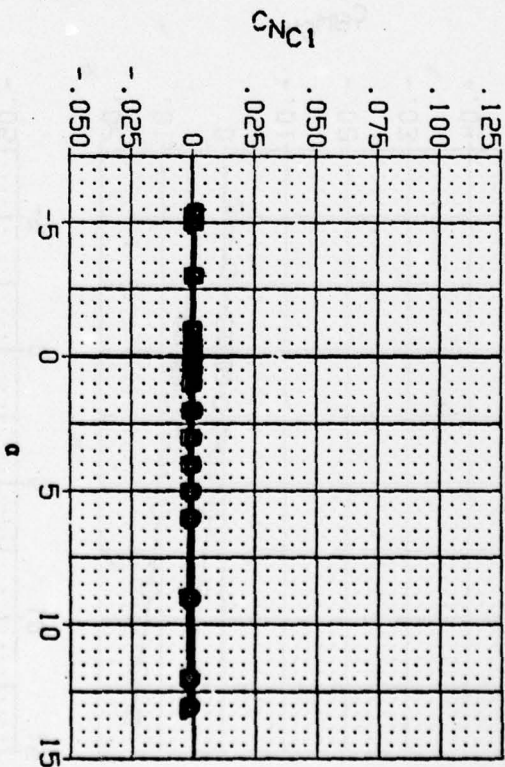
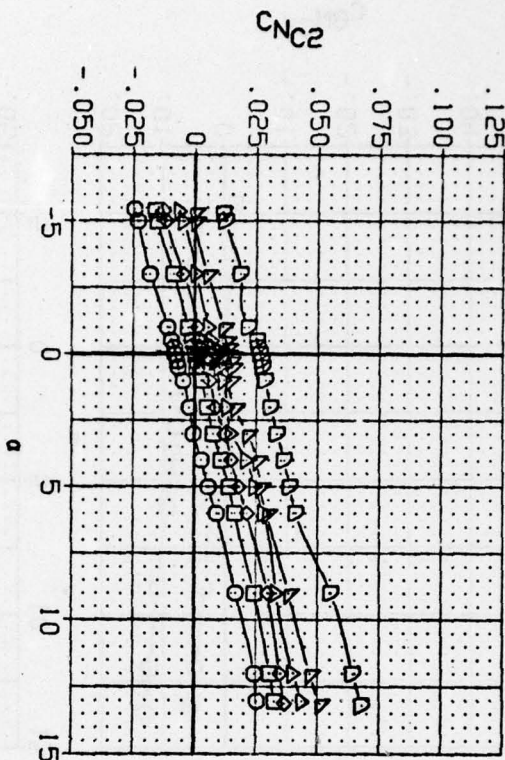
(AXH061)	○	AEDC V41A-C1A, CANARD CONTROL, BNC111
(AXH062)	□	AEDC V41A-C1A, CANARD CONTROL, BNC111
(AXH063)	△	AEDC V41A-C1A, CANARD CONTROL, BNC111
(AXH064)	▽	AEDC V41A-C1A, CANARD CONTROL, BNC111
(AXH065)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC111
(AXH066)	◊	AEDC V41A-C1A, CANARD CONTROL, BNC111

DCND1 DCND2 DCND3 DCND4

.000	-3.000	.000	-3.000
.000	.000	.000	.000
.000	3.000	.000	.000
.000	6.000	.000	.000
.000	9.000	.000	.000
.000	15.000	.000	.000

REFERENCE INFORMATION

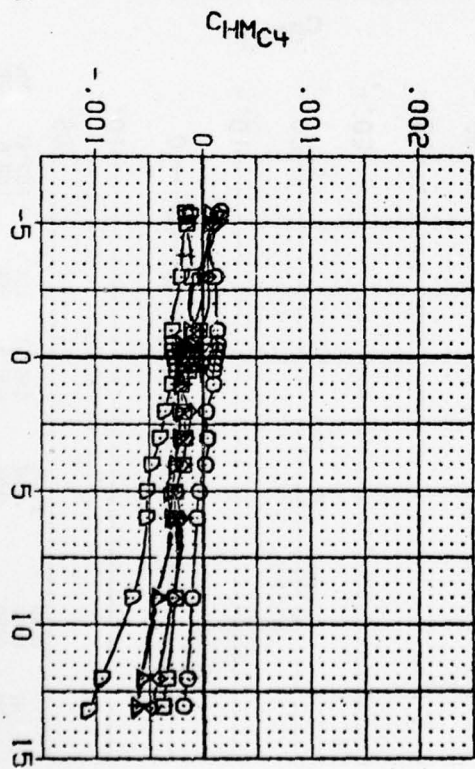
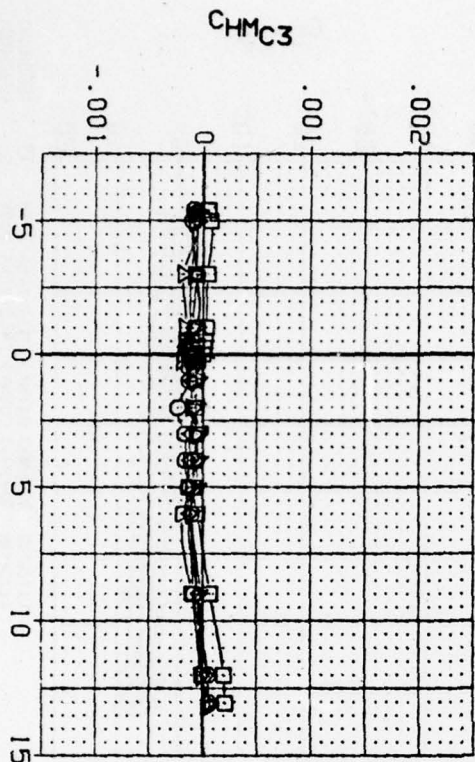
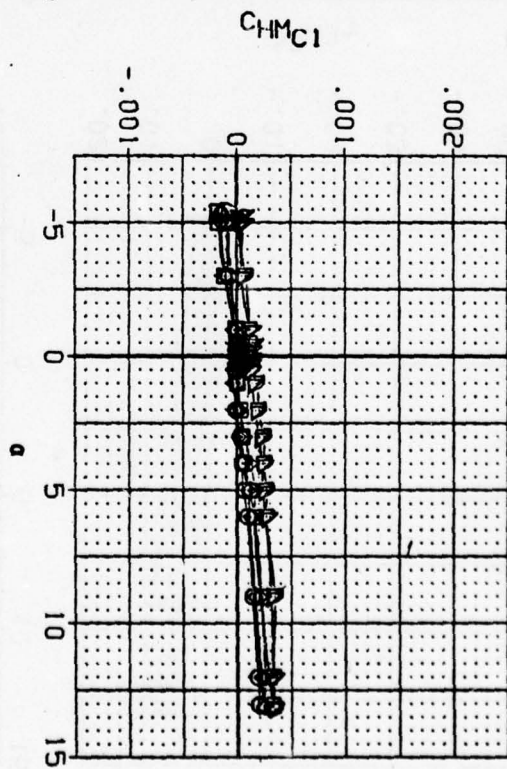
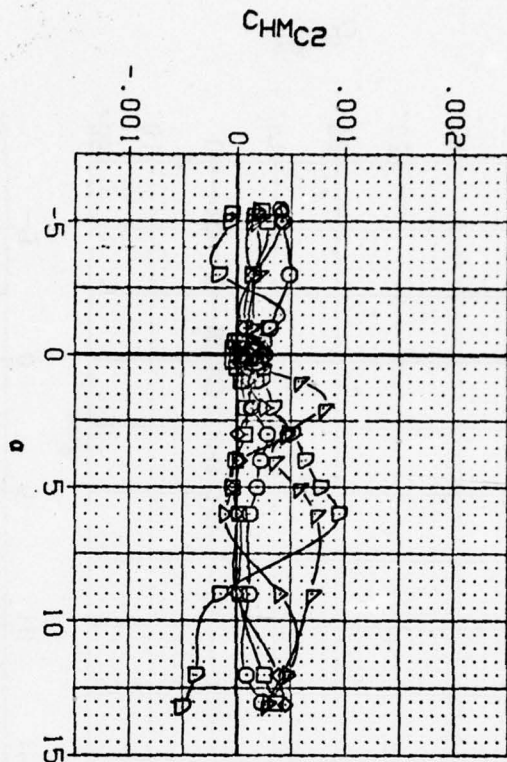
SREF	19.5350	SO. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{H1A}=0$ $\Phi_{H1CND}=0$
 (A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(BXH061)	□	AEDC V41A-C1A, CANARD CONTROL, BNC2C11
(BXH062)	○	AEDC V41A-C1A, CANARD CONTROL, BNC2C11
(BXH063)	△	AEDC V41A-C1A, CANARD CONTROL, BNC2C11
(BXH064)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC2C11
(BXH065)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC2C11
(BXH066)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC2C11

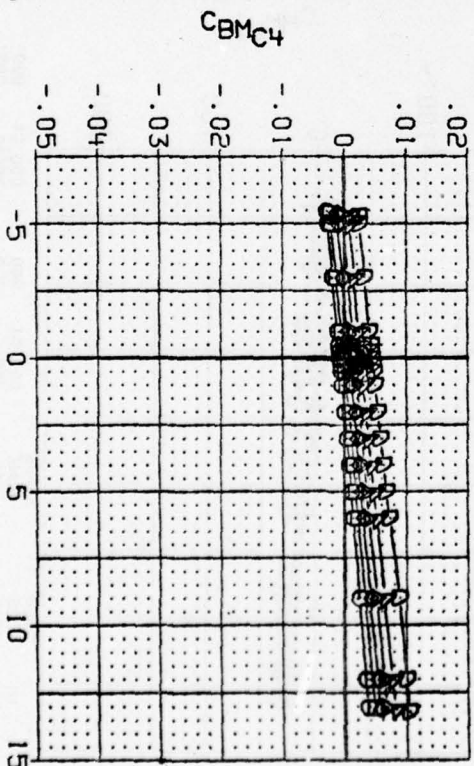
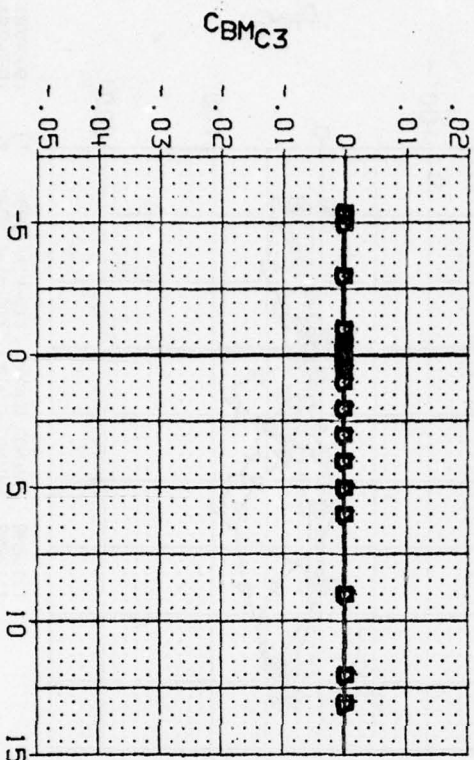
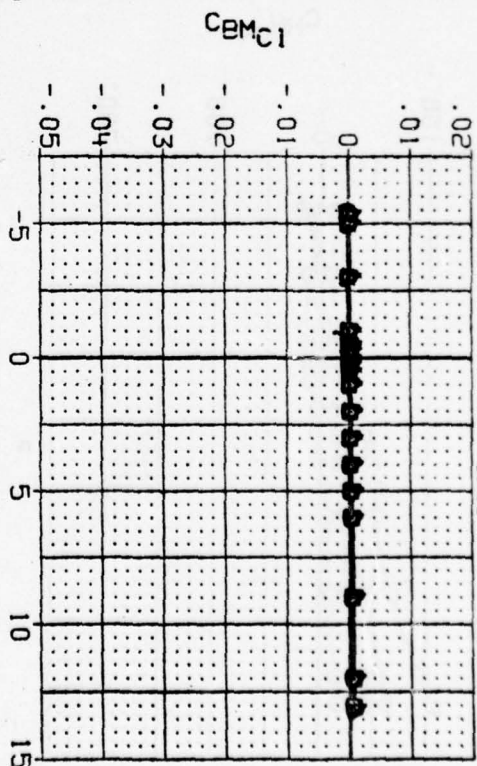
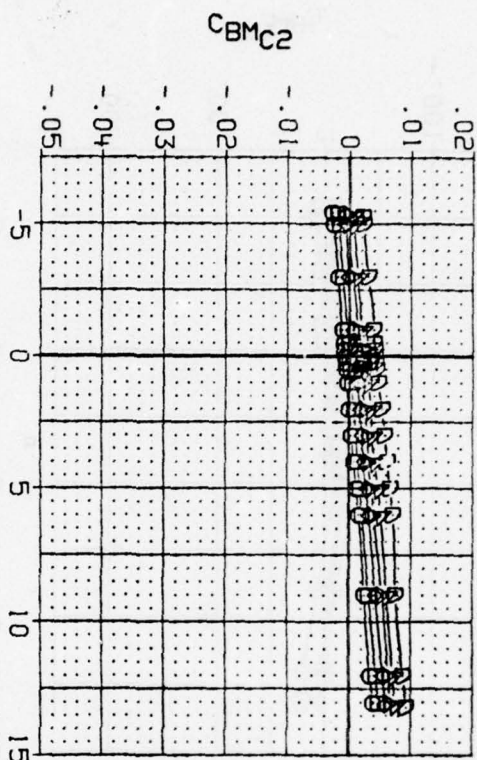
DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
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.000	.000	.000	.000	LREF 5.0000 IN.
.000	3.000	.000	3.000	BREF 5.0000 IN.
.000	6.000	.000	6.000	XREF 25.0000 IN.
.000	9.000	.000	9.000	YREF .0000 IN.
.000	15.000	.000	15.000	ZREF .0000 IN.
				SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0
 (A)MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(BXH061)	□	AEDC V41A-C1A, CANARD CONTROL, BNC2C111
(BXH062)	○	AEDC V41A-C1A, CANARD CONTROL, BNC2C111
(BXH063)	△	AEDC V41A-C1A, CANARD CONTROL, BNC2C111
(BXH064)	▽	AEDC V41A-C1A, CANARD CONTROL, BNC2C111
(BXH065)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC2C111
(BXH066)	◇	AEDC V41A-C1A, CANARD CONTROL, BNC2C111

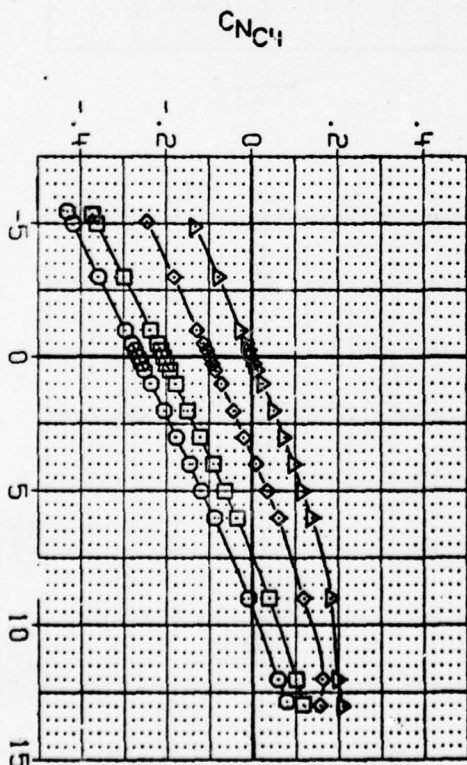
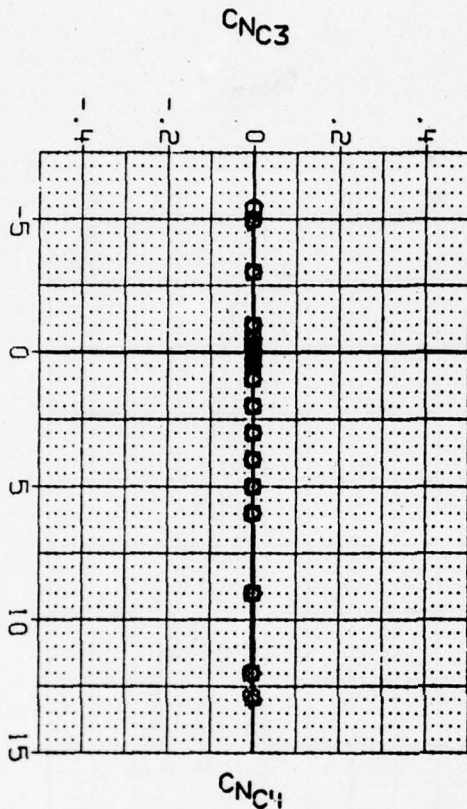
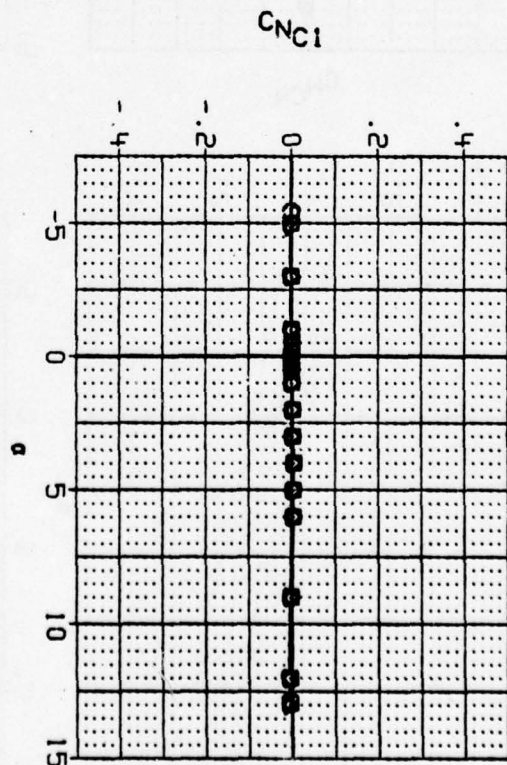
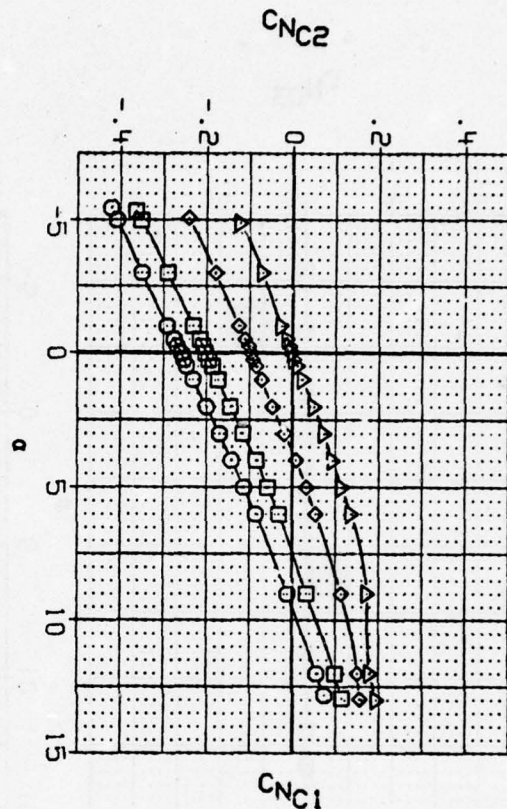
DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
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.000	.000	.000	.000	LREF 5.0000 IN.
.000	3.000	.000	3.000	BREF 5.0000 IN.
.000	5.000	.000	6.000	YREF 26.0000 IN.
.000	9.000	.000	9.000	ZREF .0000 IN.
.000	15.000	.000	15.000	SCALE .0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=0
 (A) MACH = 3.00

DCND1	DCND2	DCND3	DCND4
.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000

REFERENCE INFORMATION		
SREF	19.6350	50. IN.
LEEF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0
= 1.51
(A)MACH

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH067) \square AEDC V1A-C1A, CANARD CONTROL, BNNC6T1

(AXH068) \square AEDC V1A-C1A, CANARD CONTROL, BNNC6T1

(AXH069) \square AEDC V1A-C1A, CANARD CONTROL, BNNC6T1

(AXH070) \triangle AEDC V1A-C1A, CANARD CONTROL, BNNC6T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

.000 3.000 .000 .000

.000 9.000 .000 .000

.000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

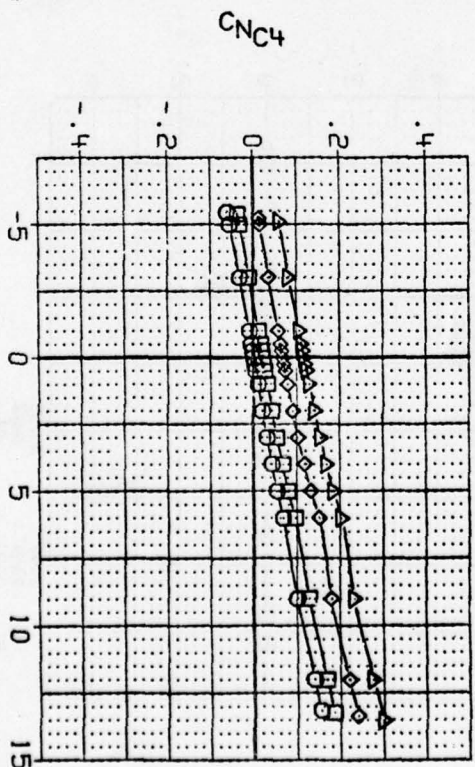
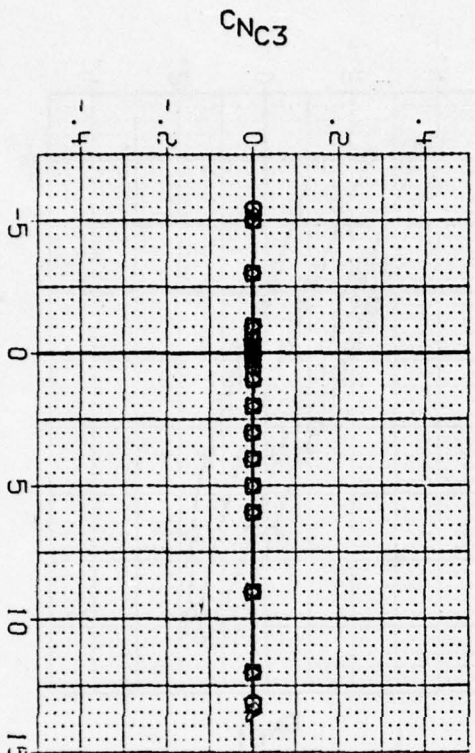
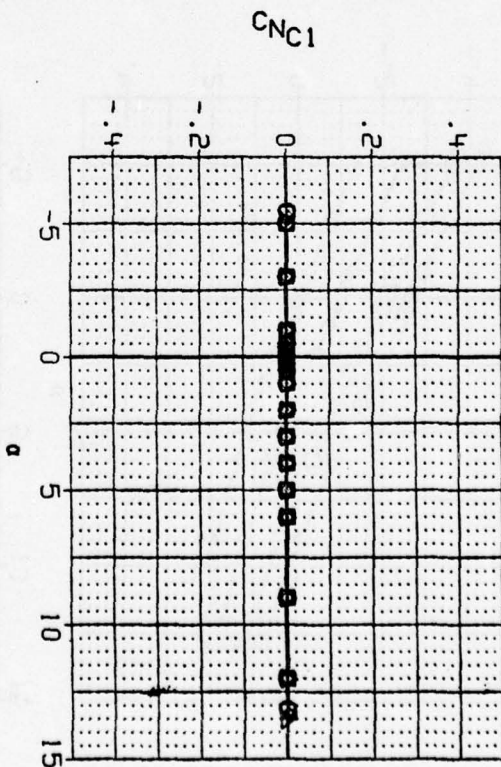
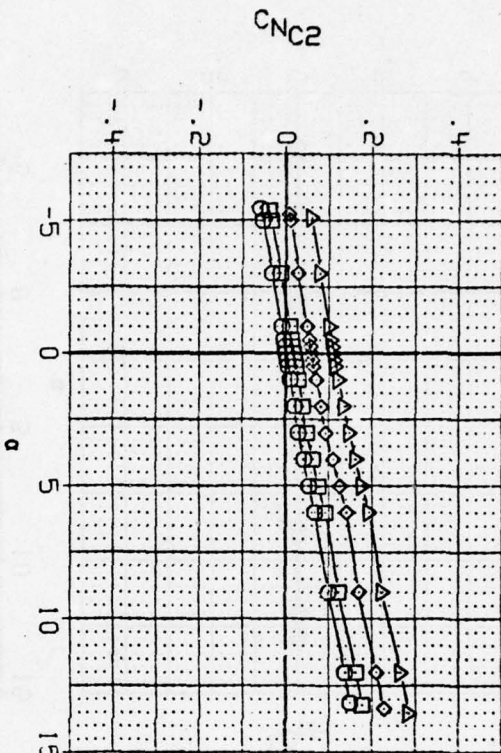
LREF 5.0000 IN.

BREF 5.0000 IN.

YMRP 26.0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

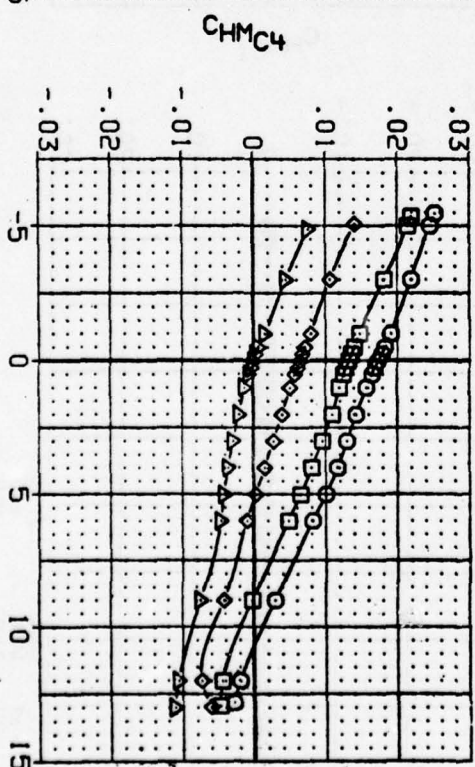
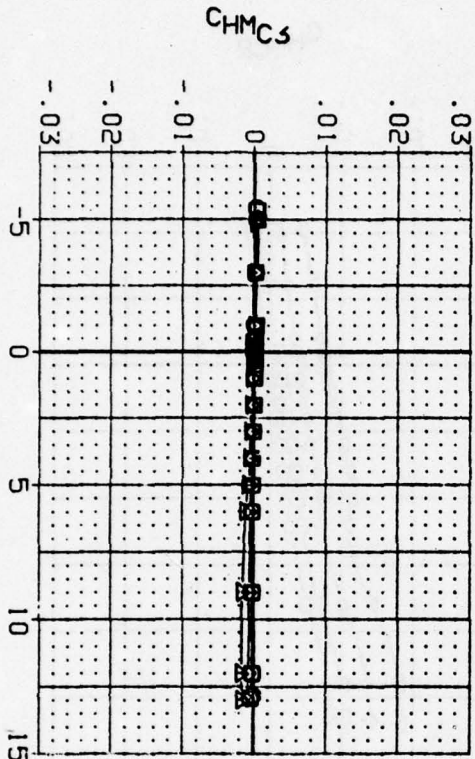
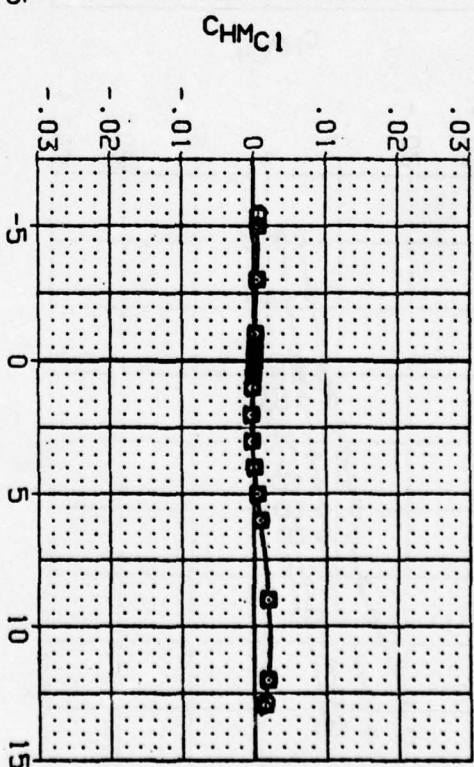
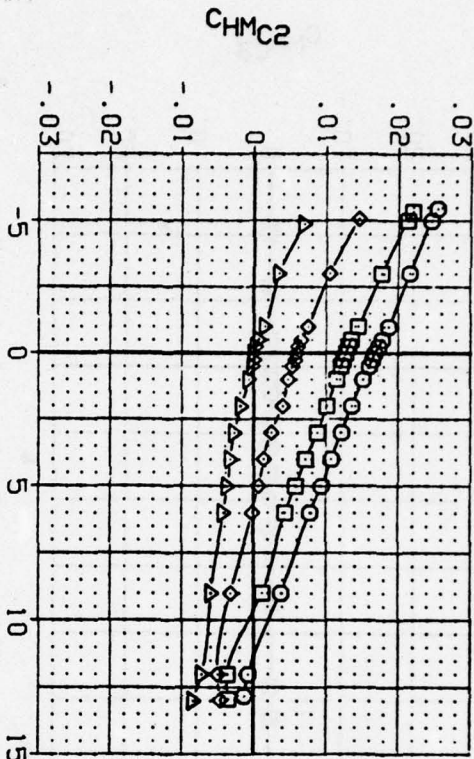
(BXH067) \square AEDC W1A-C1A, CANARD CONTROL, BNHC671
 (BXH068) \square AEDC W1A-C1A, CANARD CONTROL, BNHC671
 (BXH069) \square AEDC W1A-C1A, CANARD CONTROL, BNHC671
 (BXH070) \triangle AEDC W1A-C1A, CANARD CONTROL, BNHC671

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 PREF 5.0000 IN.
 YREF 26.0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

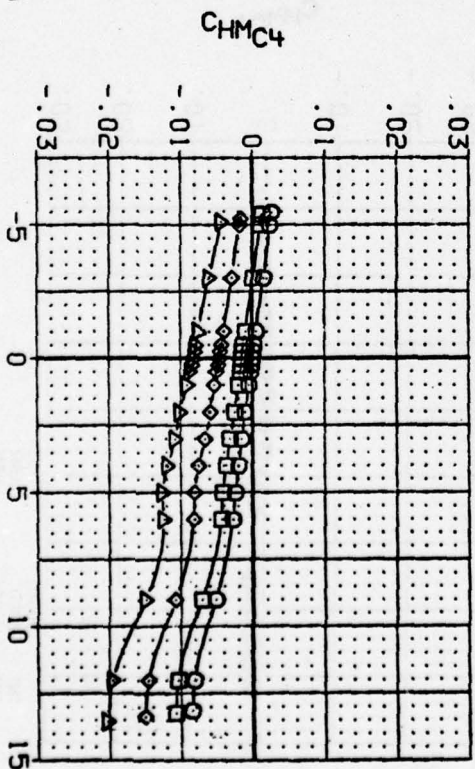
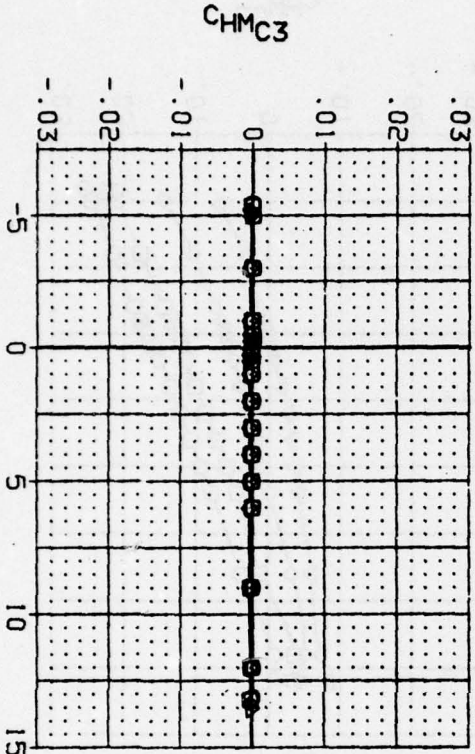
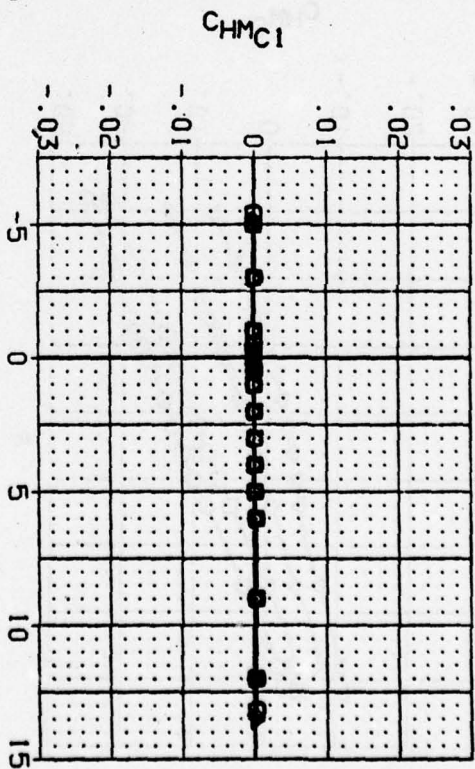
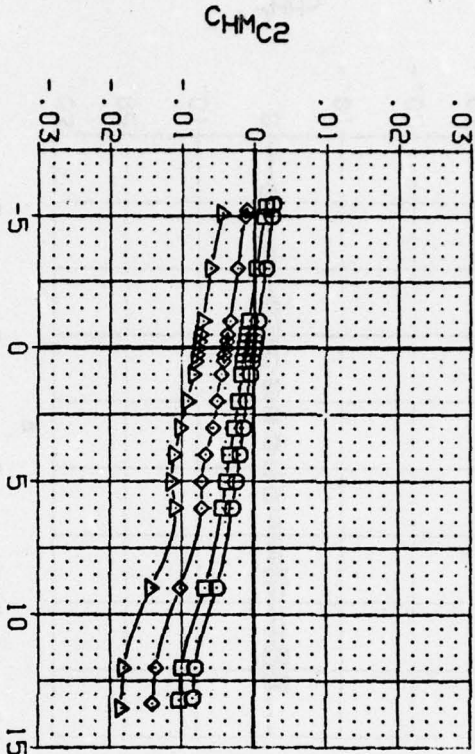
PHITAL=0 PHICND=0

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH067) \square AEDC W1A-C1A, CANARD CONTROL, BNMCS11
 (BXH068) \square AEDC W1A-C1A, CANARD CONTROL, BNMCS11
 (BXH069) \square AEDC W1A-C1A, CANARD CONTROL, BNMCS11
 (BXH070) \triangle AEDC W1A-C1A, CANARD CONTROL, BNMCS11

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

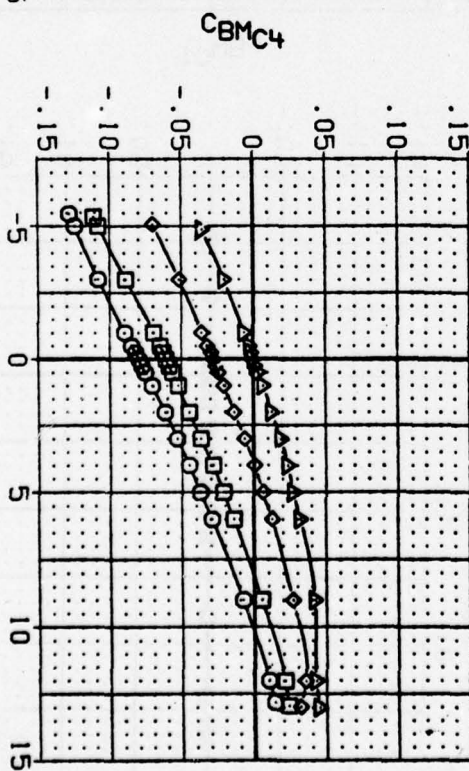
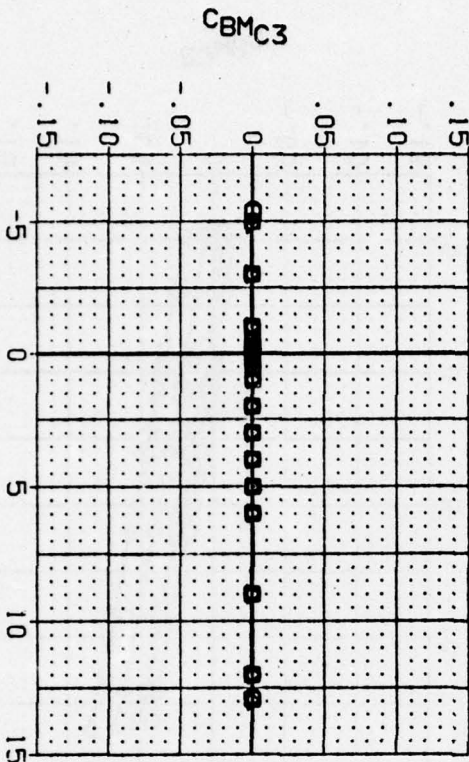
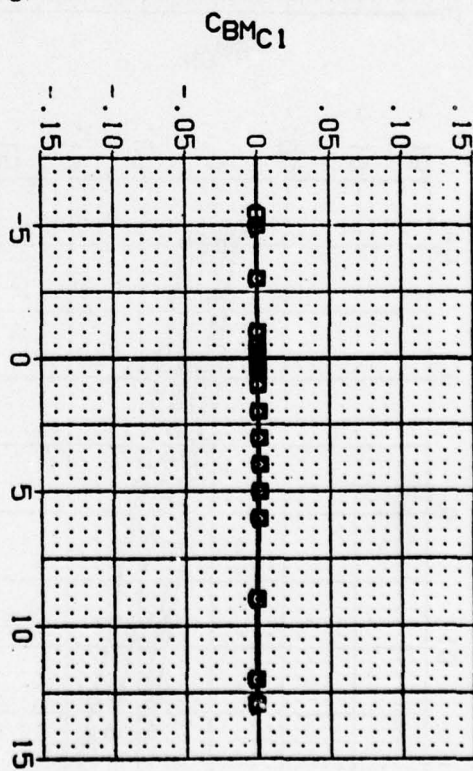
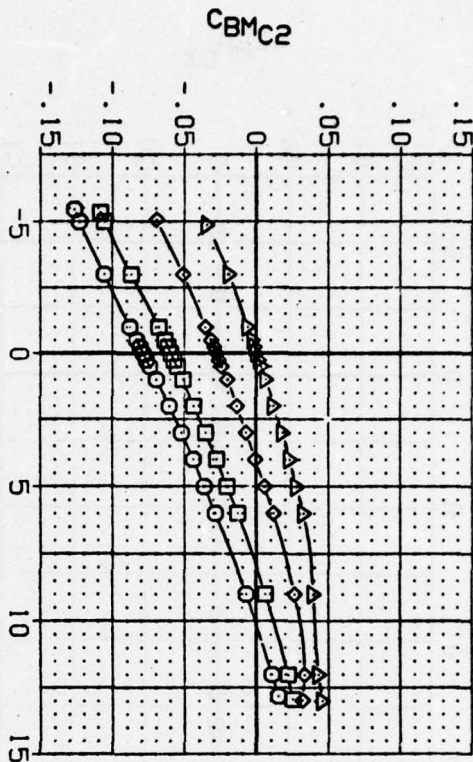


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHICAL=0 PHICND=0
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH067) \square AEDC W1A-C1A, CANARD CONTROL, BMH067
 (BXH068) \square AEDC W1A-C1A, CANARD CONTROL, BMH068
 (BXH069) \square AEDC W1A-C1A, CANARD CONTROL, BMH069
 (BXH070) \square AEDC W1A-C1A, CANARD CONTROL, BMH070
 (BXH071) \triangle AEDC W1A-C1A, CANARD CONTROL, BMH071

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{H1AL}=0$ $\Phi_{H1CD}=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

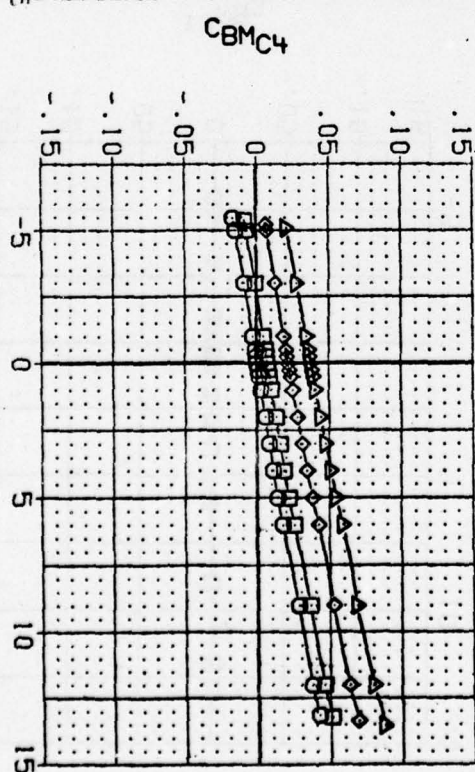
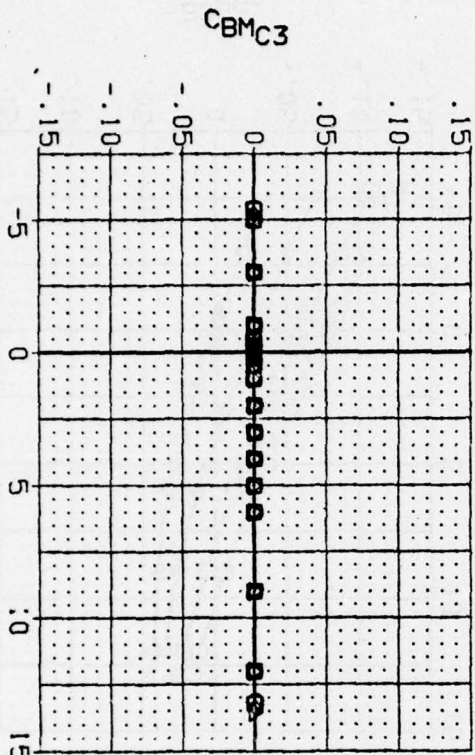
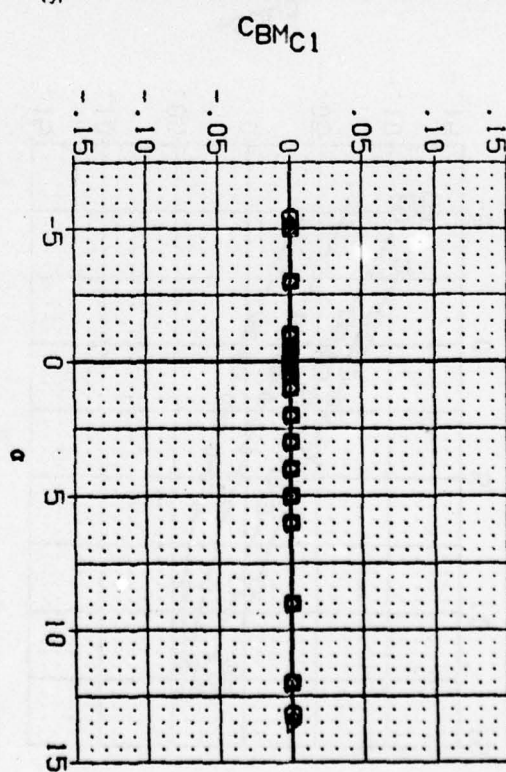
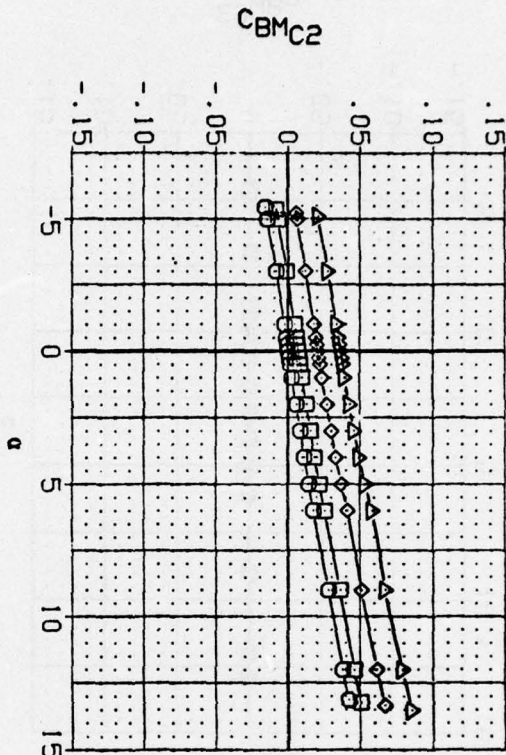
(BXH067) \square AEDC V41A-C1A, CANARD CONTROL, BMH0671
 (BXH068) \square AEDC V41A-C1A, CANARD CONTROL, BMH0681
 (BXH069) \square AEDC V41A-C1A, CANARD CONTROL, BMH0691
 (BXH070) \square AEDC V41A-C1A, CANARD CONTROL, BMH0701
 (BXH071) \square AEDC V41A-C1A, CANARD CONTROL, BMH0711

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$PHIAL=0$ $PHICND=0$

(B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

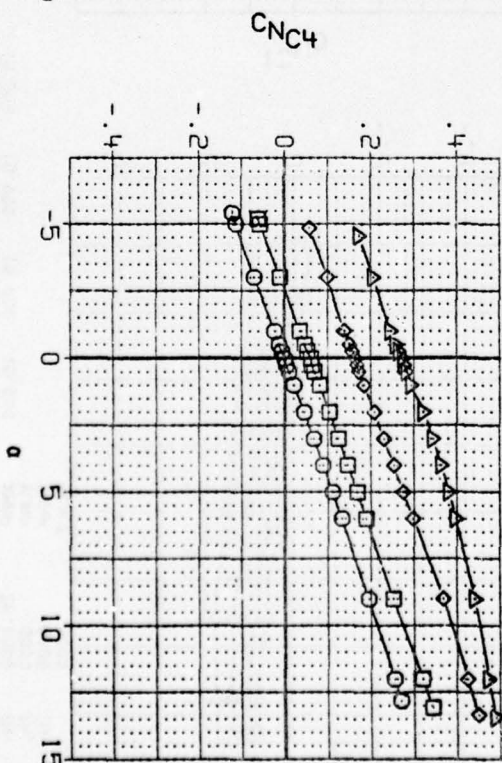
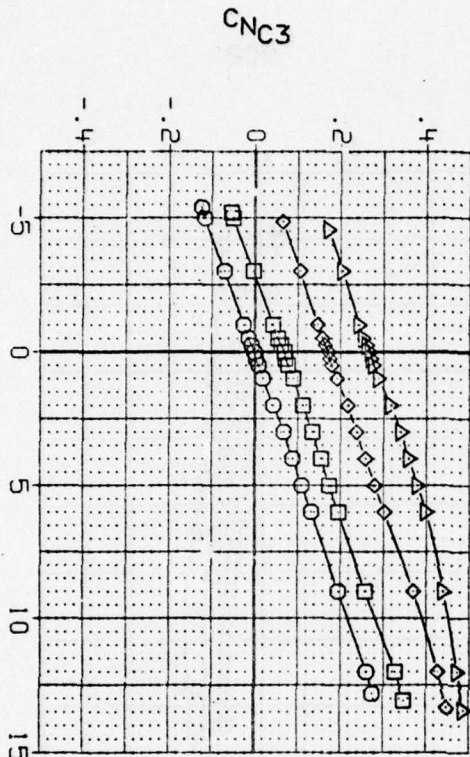
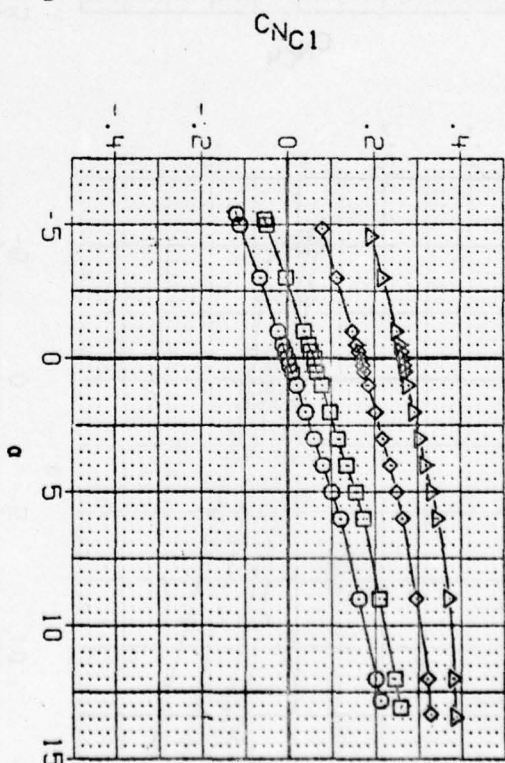
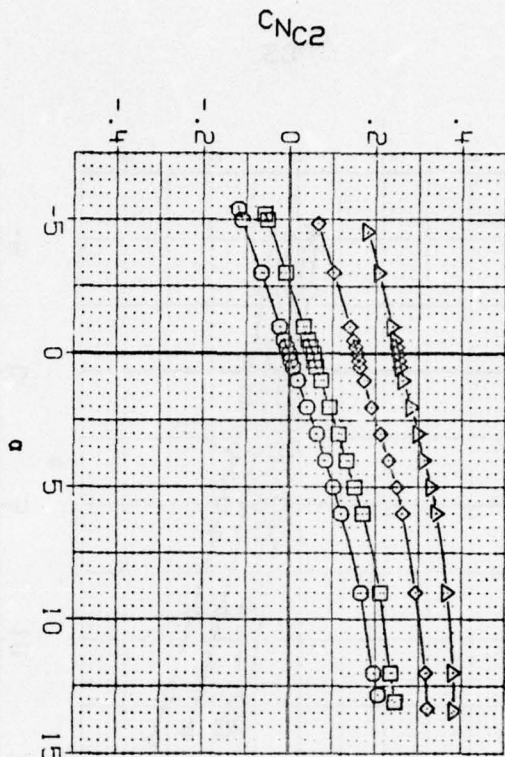
(AXH072)	○	AEDC V41A-C1A, CANARD CONTROL, BMM6511
(AXH073)	□	AEDC V41A-C1A, CANARD CONTROL, BMM6511
(AXH074)	△	AEDC V41A-C1A, CANARD CONTROL, BMM6511
(AXH075)	◇	AEDC V41A-C1A, CANARD CONTROL, BMM6511

DCND1 DCND2 DCND3 DCND4

.000	.000	.000	.000
3.000	3.000	3.000	3.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION

SREF	19.6350	50. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XMRP	26.0000	IN.
YMRP	.0000	IN.
ZMRP	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$PHITAL=0$ $PHICND=45$

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH072) \square AEDC V41A-C1A, CANARD CONTROL, BNHC511
 (AXH073) \square AEDC V41A-C1A, CANARD CONTROL, BNHC511
 (AXH074) \triangle AEDC V41A-C1A, CANARD CONTROL, BNHC511
 (AXH075) \triangle AEDC V41A-C1A, CANARD CONTROL, BNHC511

DCND1 .000
 .000
 3.000
 9.000
 15.000

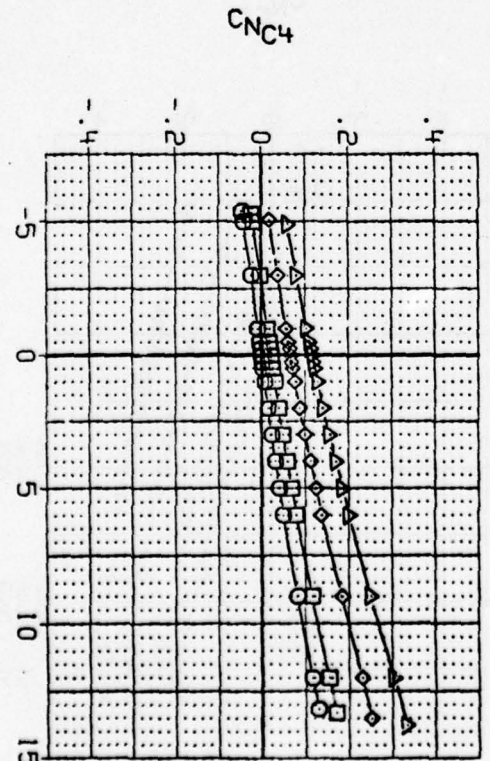
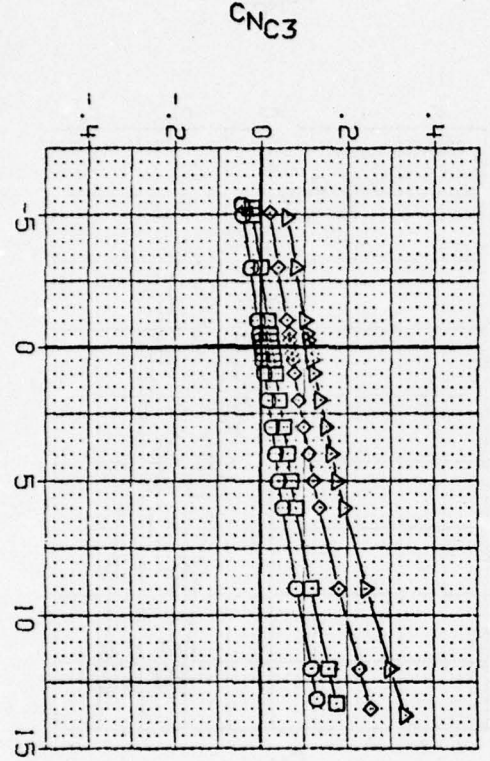
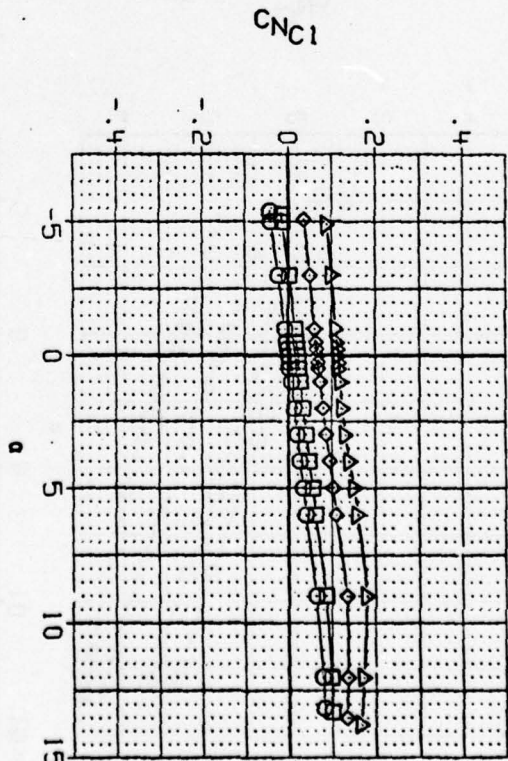
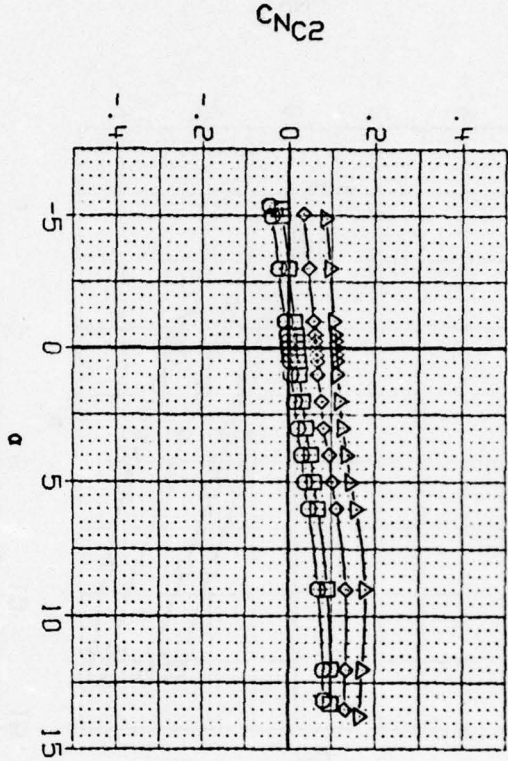
DCND2 .000
 .000
 3.000
 9.000
 15.000

DCND3 .000
 .000
 3.000
 9.000
 15.000

DCND4 .000
 .000
 3.000
 9.000
 15.000

REFERENCE INFORMATION

SREF 19.6350 50 IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

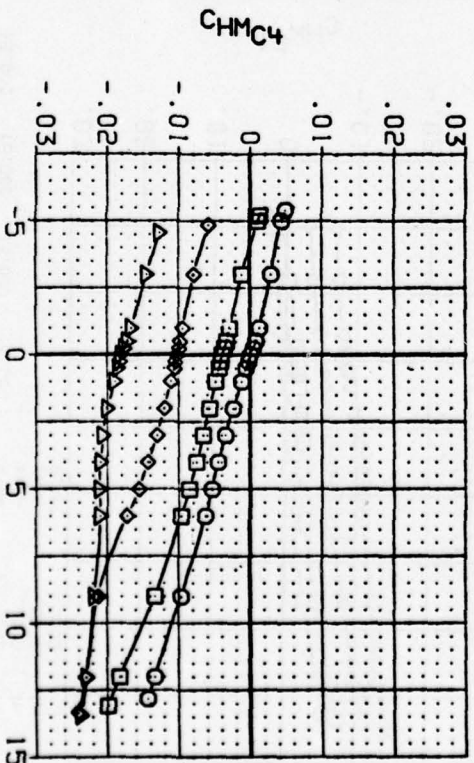
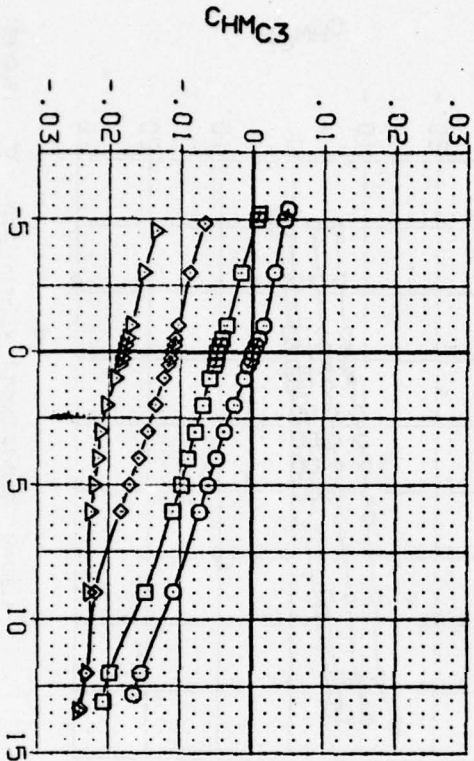
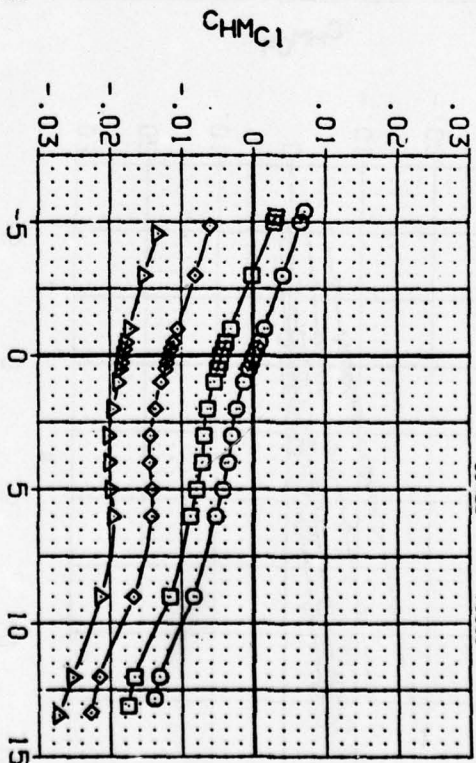
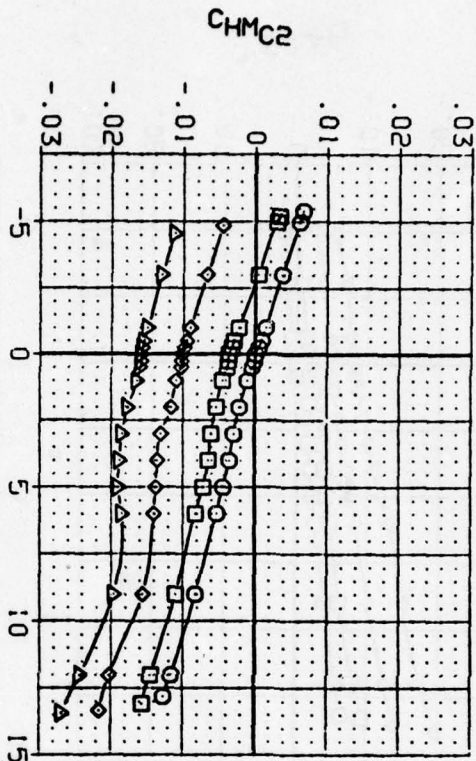
(BXH072) \square AEDC V/A-CIA, CANARD CONTROL, BNHC611
 (BXH073) \square AEDC V/A-CIA, CANARD CONTROL, BNHC611
 (BXH074) \triangle AEDC V/A-CIA, CANARD CONTROL, BNHC611
 (BXH075) \triangle AEDC V/A-CIA, CANARD CONTROL, BNHC611

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

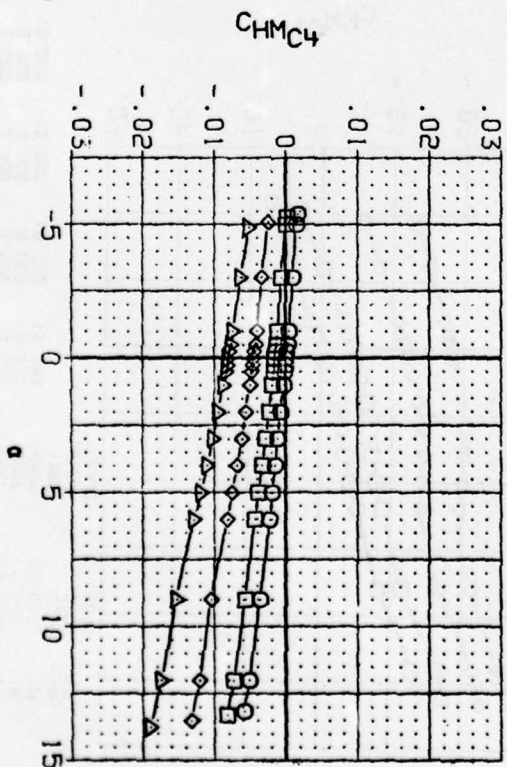
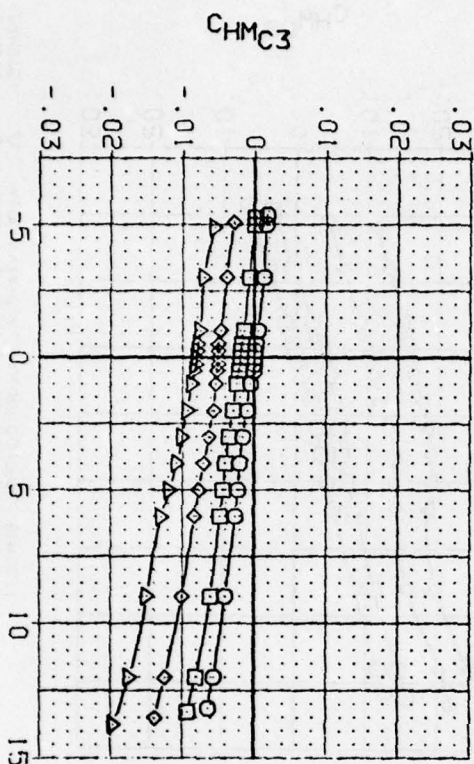
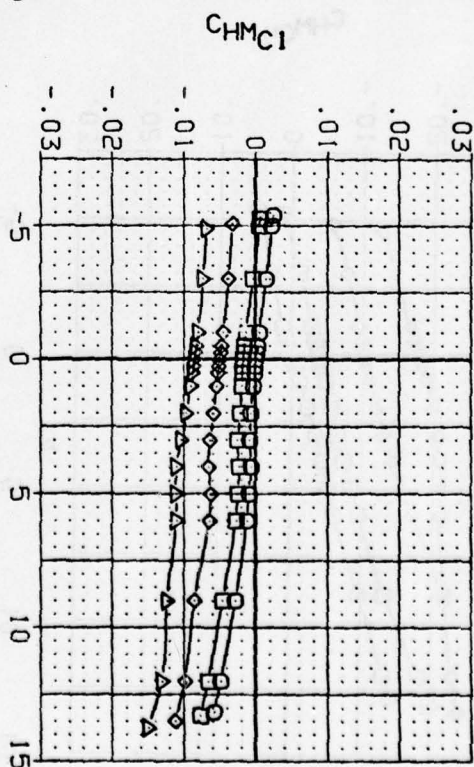
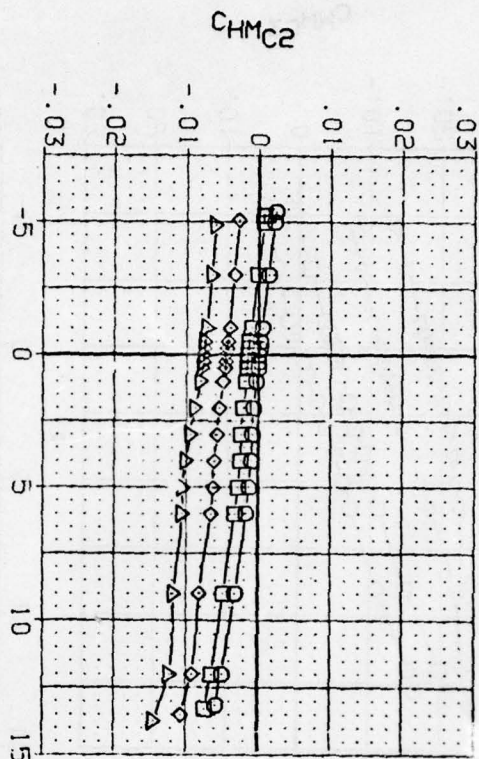
$\Phi_{H1AL}=0$ $\Phi_{H1CD}=45$

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (BXH33) AEDC V41A-C1A, CANARD CONTROL, BNA6C61
 (BXH34) AEDC V41A-C1A, CANARD CONTROL, BNA6C61
 (BXH35) AEDC V41A-C1A, CANARD CONTROL, BNA6C61

DCND1 .000
 DCND2 .002
 DCND3 .030
 DCND4 .000
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REFERENCE INFORMATION
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 SCALE .0000

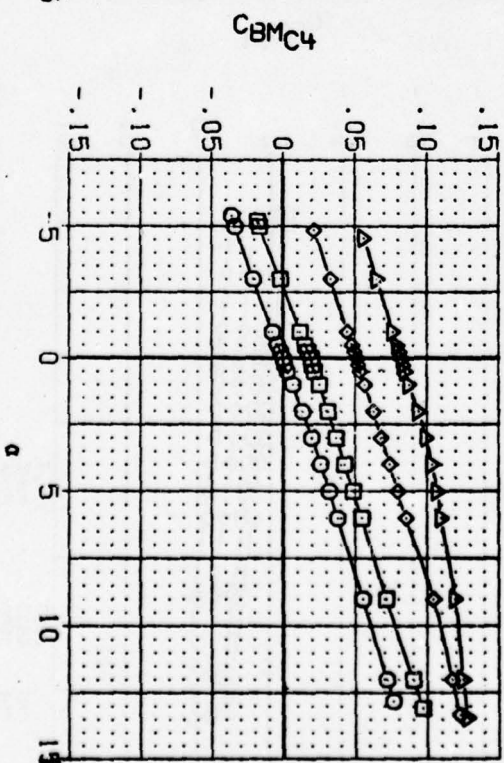
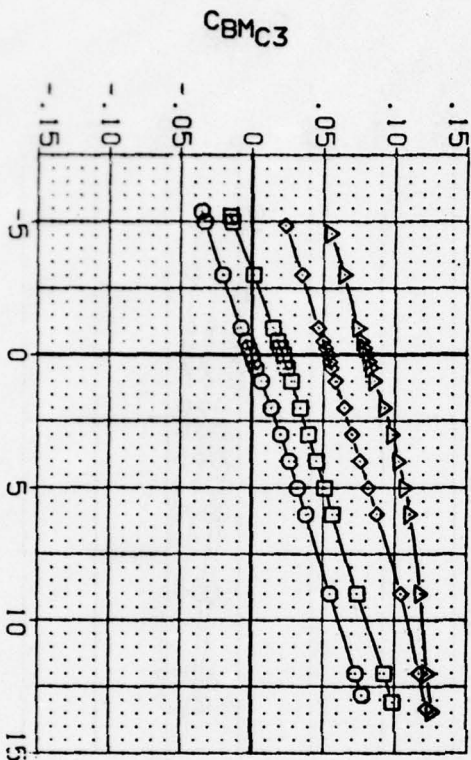
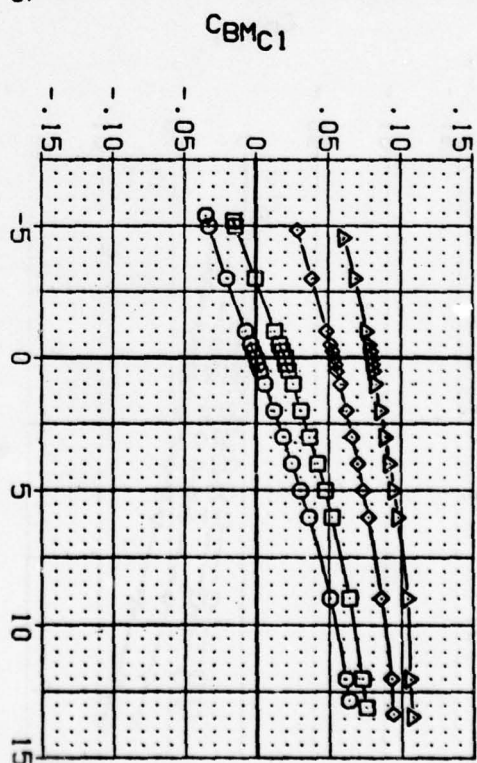
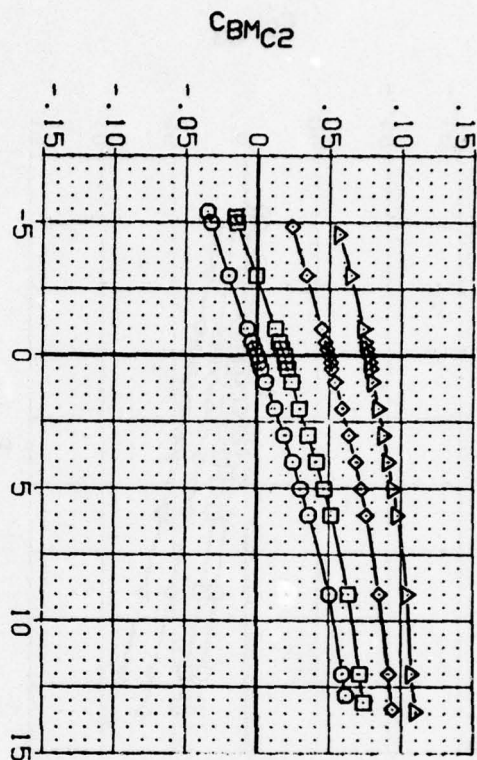


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (BXH073) \square AEDC V41A-C1A, CANARD CONTROL, BNHC6T1
 (BXH074) \square AEDC V41A-C1A, CANARD CONTROL, BNHC6T1
 (BXH075) \triangle AEDC V41A-C1A, CANARD CONTROL, BNHC6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
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 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
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 ZTRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH072) \square AEDC V41A-C1A, CANARD CONTROL, BKH6611

(BXH073) \square AEDC V41A-C1A, CANARD CONTROL, BKH6611

(BXH074) \square AEDC V41A-C1A, CANARD CONTROL, BKH6611

(BXH075) \triangle AEDC V41A-C1A, CANARD CONTROL, BKH6611

DCND1 DCND2 DCND3 DCND4

3.000 3.000 3.000 3.000

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15.000 15.000 15.000 15.000

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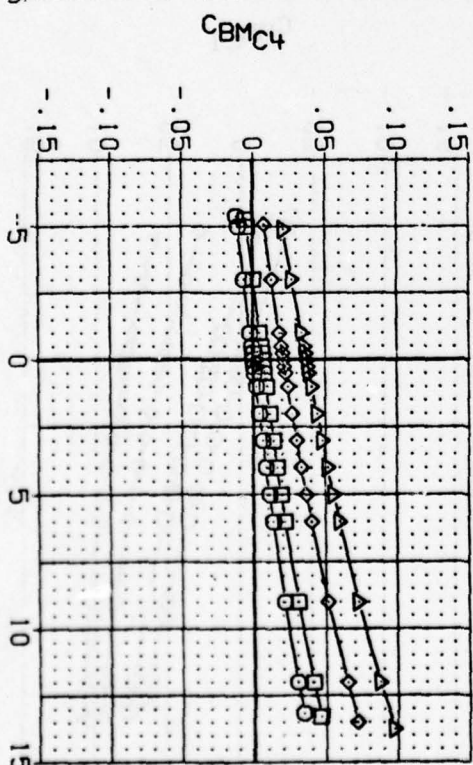
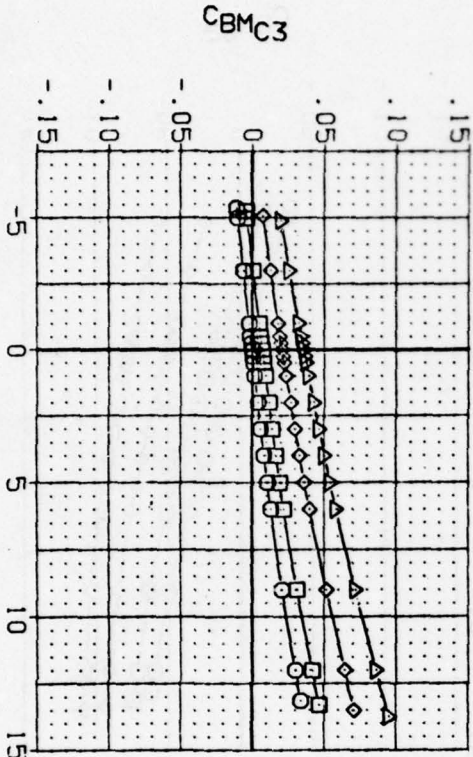
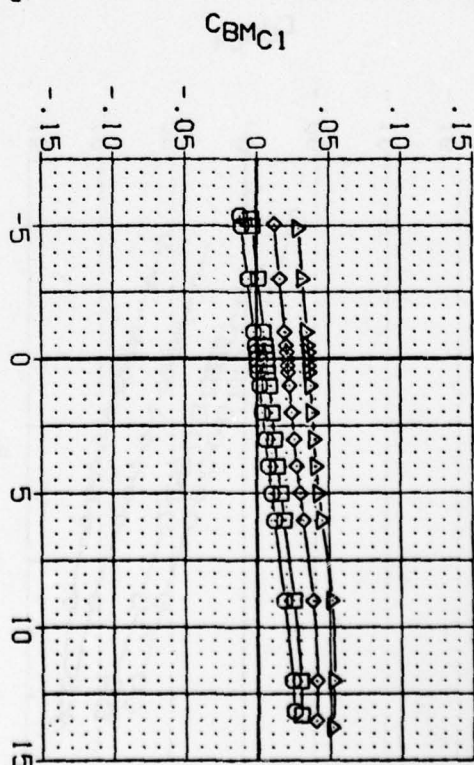
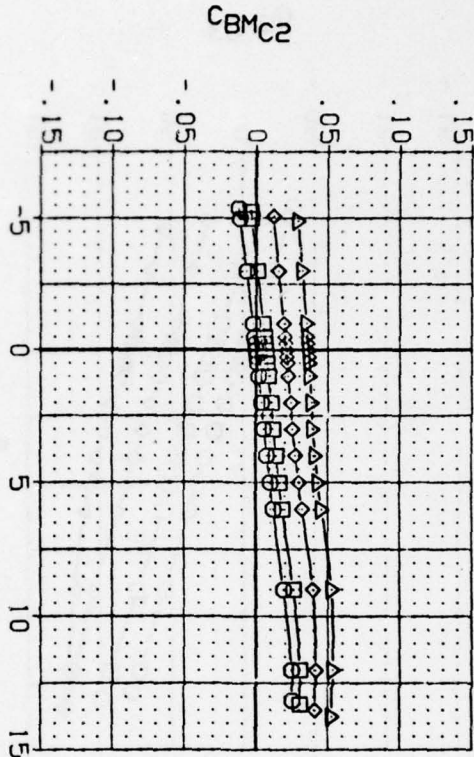
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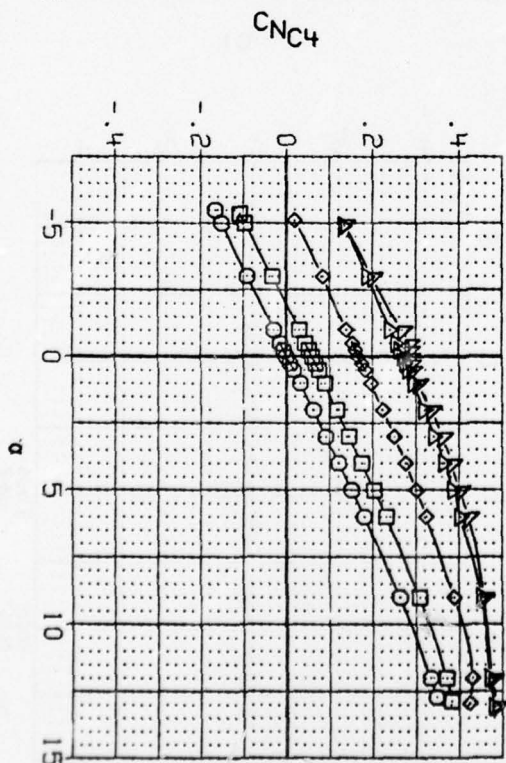
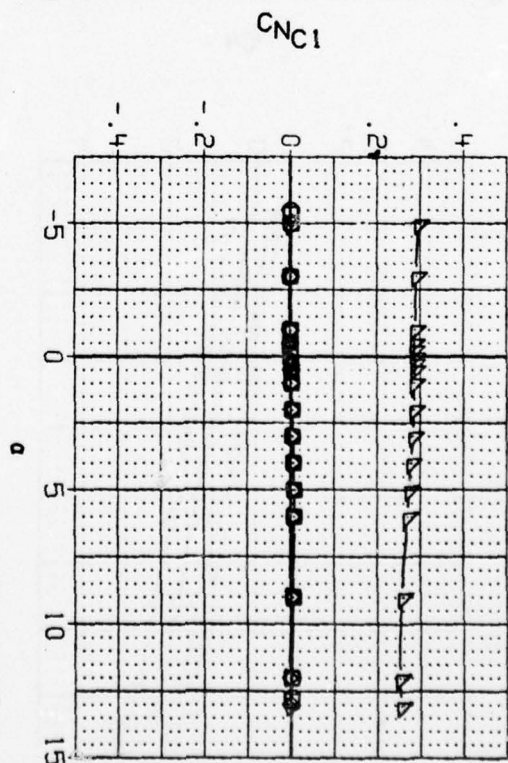


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$\Phi_{H1AL}=0$ $\Phi_{H1CND}=45$

(B)MACH = 3.01

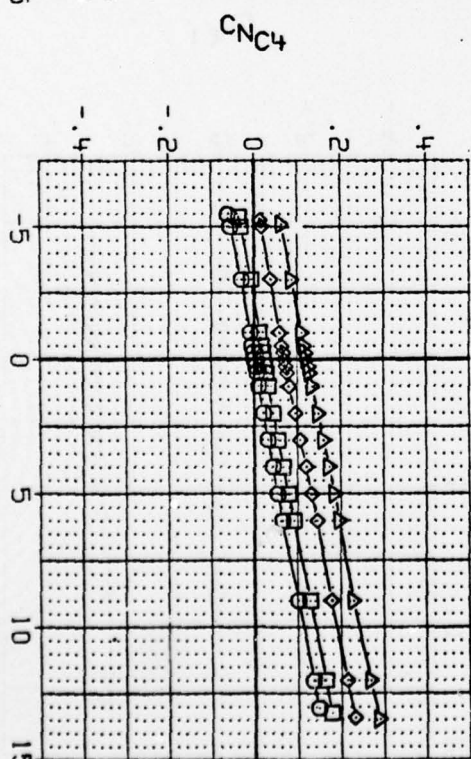
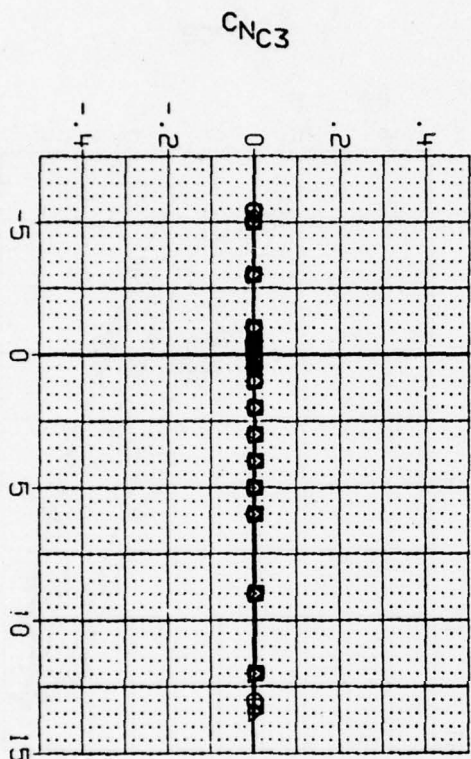
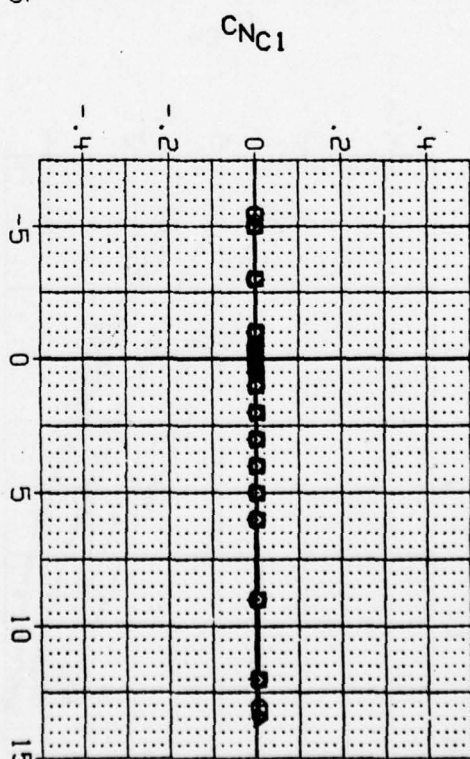
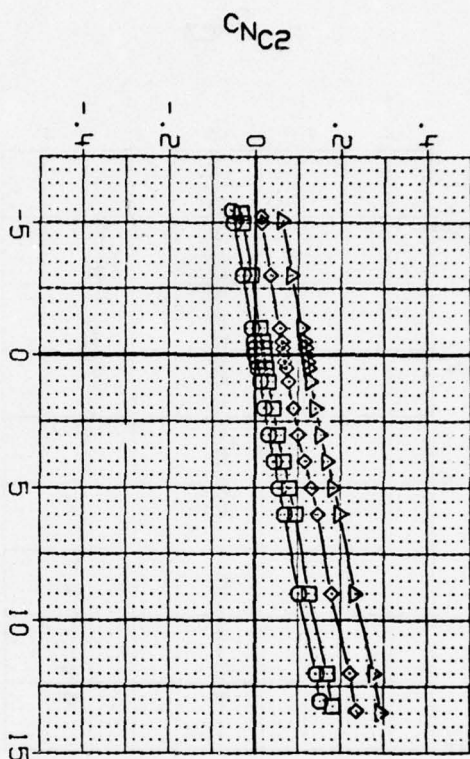
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BREF	5.0000 IN.
XMPR	25.0000 IN.
YMPR	.0000 IN.
ZMPR	.0000 IN.
SCALE	.0003



PAGE 135

DCND1	DCND2	DCND3	DCND4
.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000
15.000	15.000	15.000	15.000

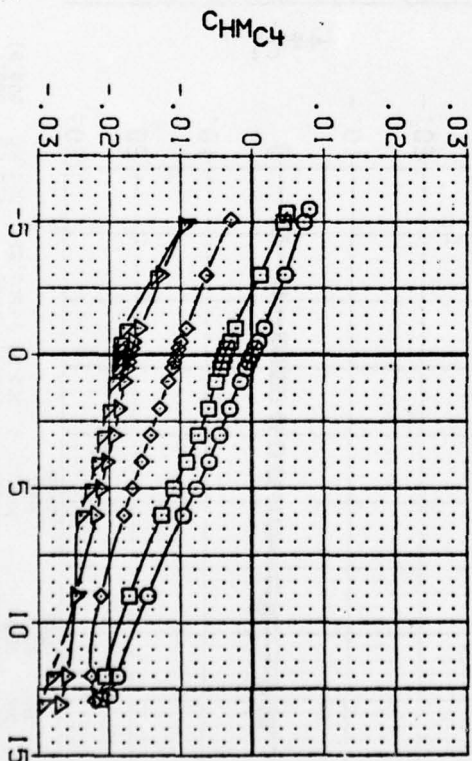
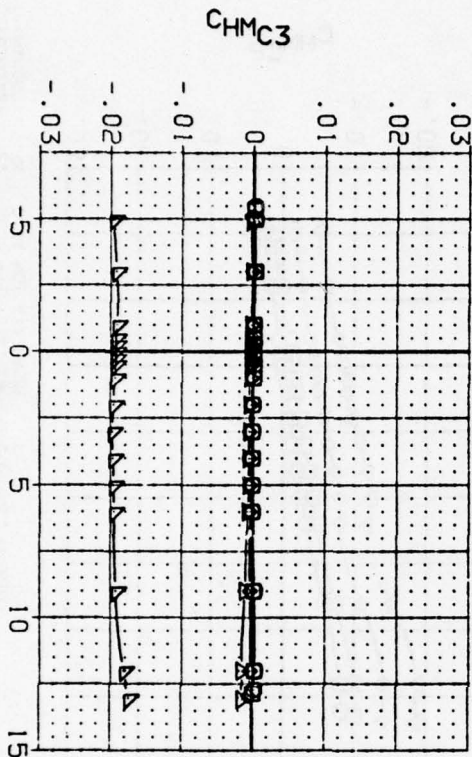
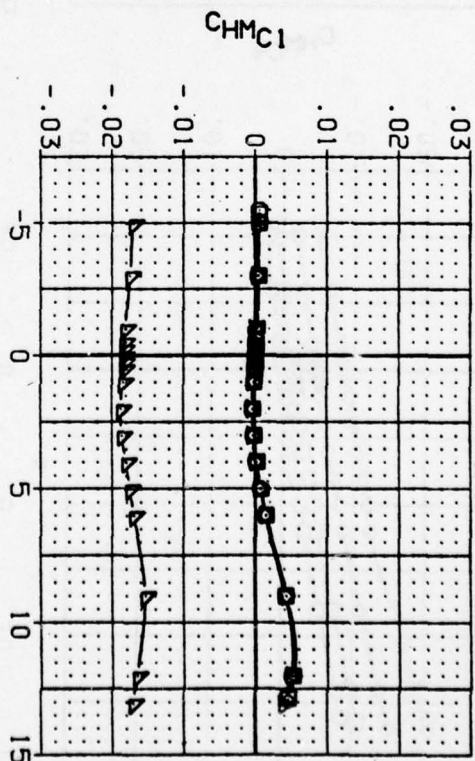
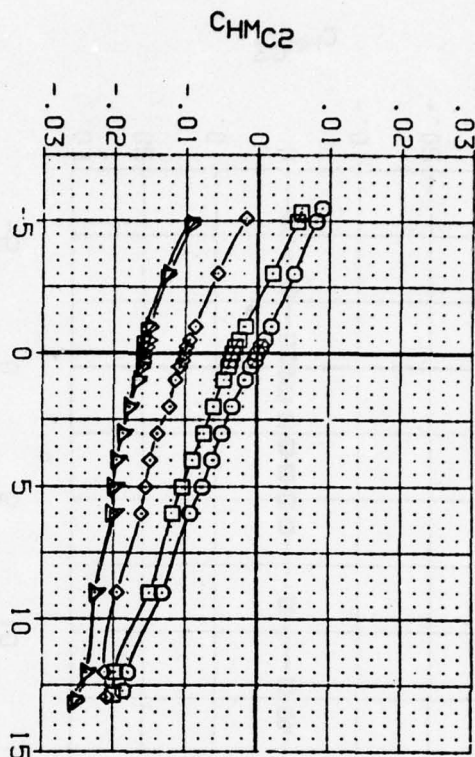
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XRRP	26.0000	IN.
YRRP	.0000	IN.
ZRRP	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0
= 3.01
(B)MACH

DCND1	DCND2	DCND3	DCND4
.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION		
SREF	19.6350	SO. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XMRP	26.0000	IN.
YMRP	.0000	IN.
ZMRP	.0000	IN.
SCALE	.0000	IN.

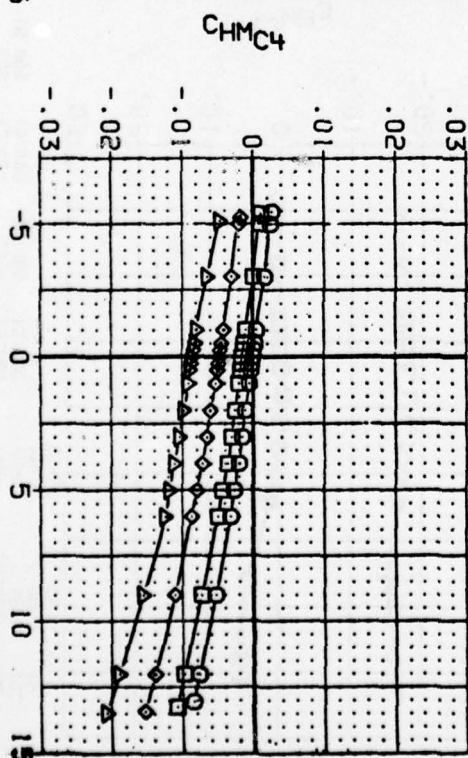
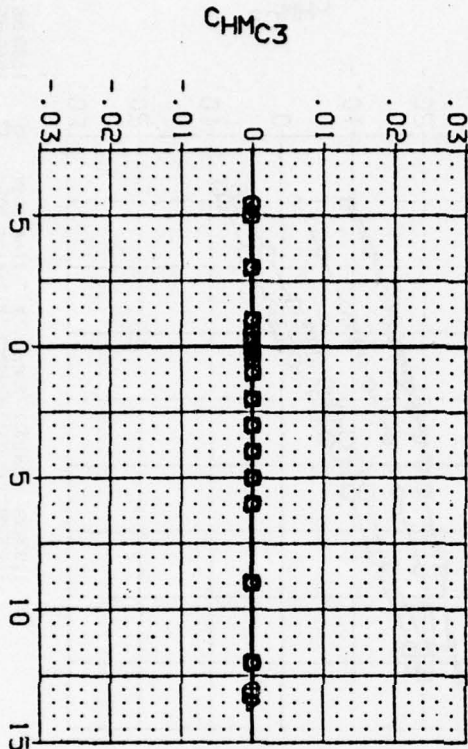
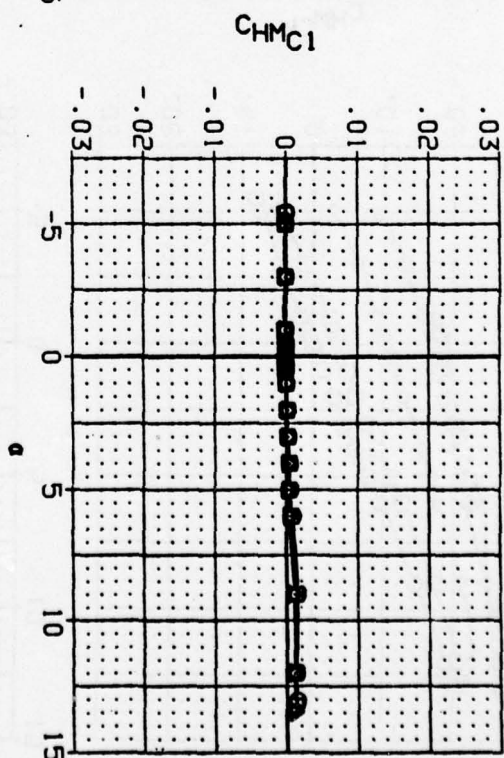
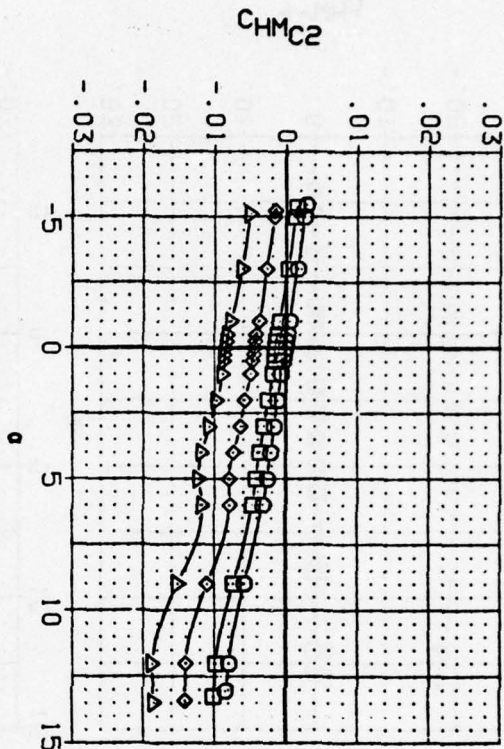


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
PHITAL=0 PHICND=0
= 1.51
(A)MACH

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (BXH077) ◇ AEDC W1A-C1A, CANARD CONTROL, BNSCST1
 (BXH078) △ AEDC W1A-C1A, CANARD CONTROL, BNSCST1
 (BXH079) △ AEDC W1A-C1A, CANARD CONTROL, BNSCST1
 (BXH080) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
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REFERENCE INFORMATION
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0020



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

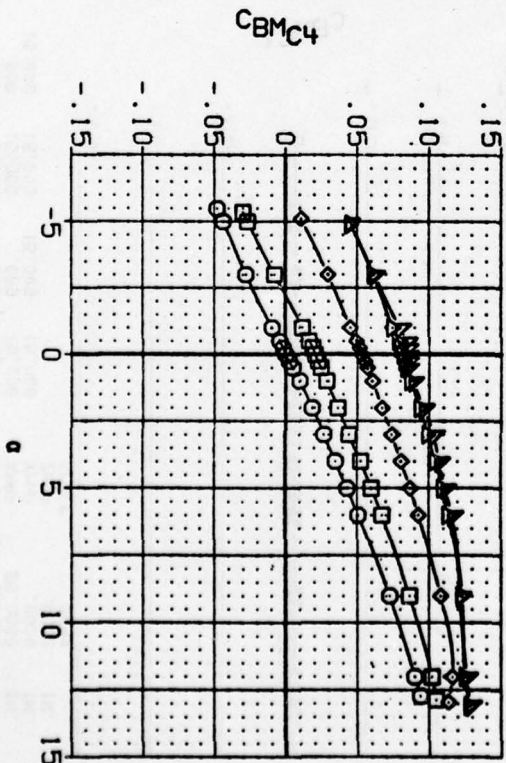
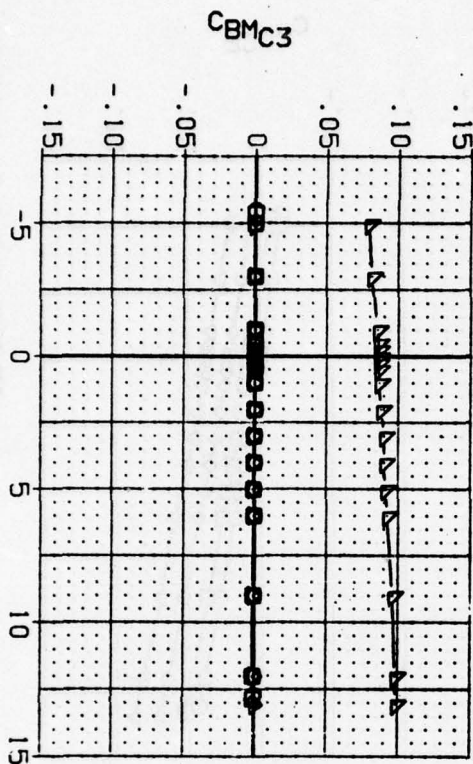
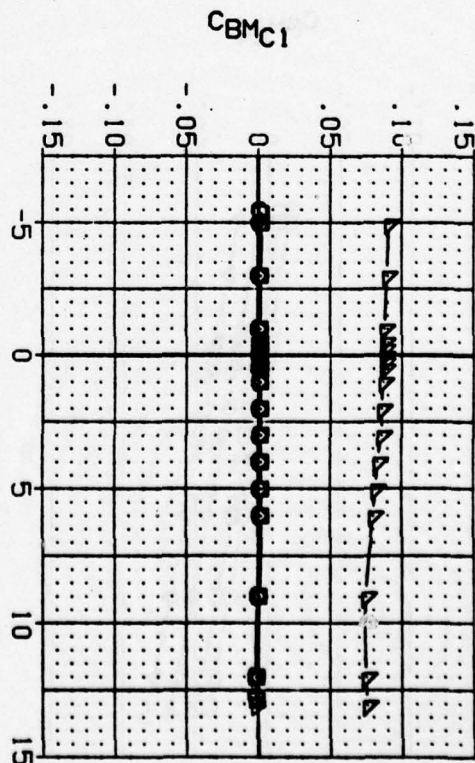
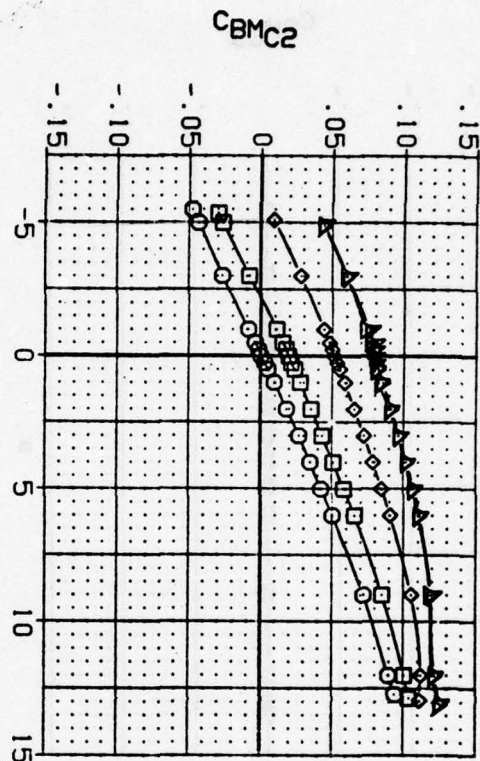
(BXH076) \square AEDC V41A-C1A, CANARD CONTROL, BMS6BT1
 (BXH077) \square AEDC V41A-C1A, CANARD CONTROL, BMS6BT1
 (BXH078) \square AEDC V41A-C1A, CANARD CONTROL, BMS6BT1
 (BXH079) \square AEDC V41A-C1A, CANARD CONTROL, BMS6BT1
 (BXH080) \square AEDC V41A-C1A, CANARD CONTROL, BMS6BT1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

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 SCALE .0000

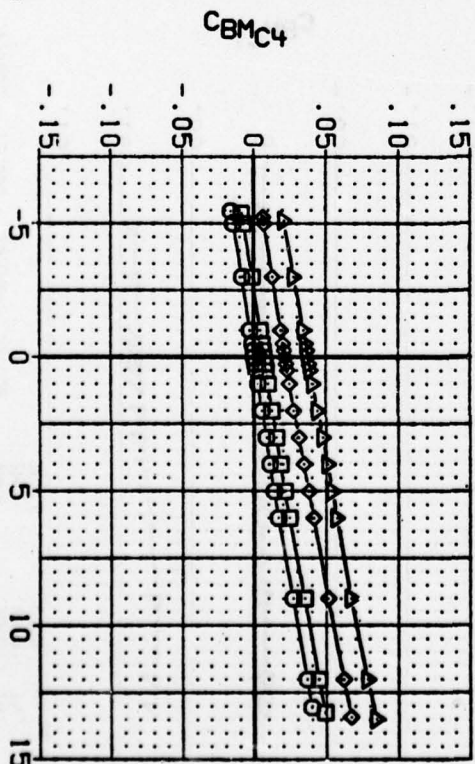
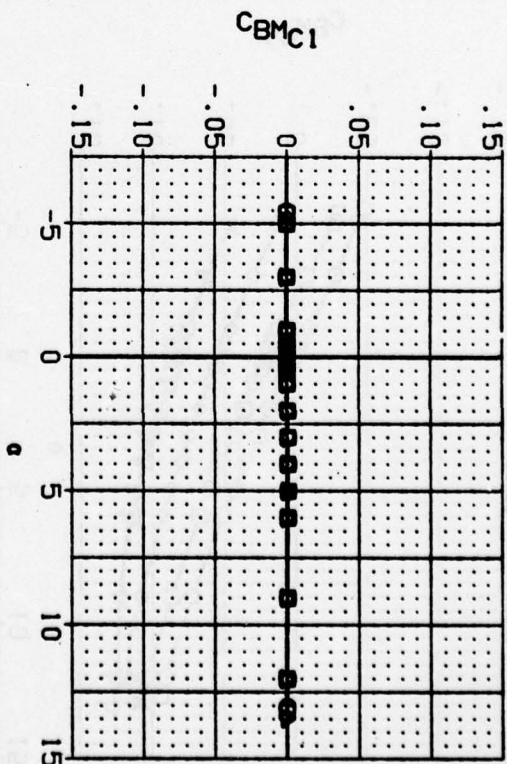


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$\Phi_{H1AL}=0$ $\Phi_{H1CND}=0$

(A) MACH = 1.51

REFERENCE INFORMATION				
DON1	DON2	DON3	DON4	
.000	.000	.000	.000	SREF 19.6350 50. IN.
.000	3.000	.000	3.000	LREF 5.0000 IN.
.000	9.000	.000	9.000	BREF 5.0000 IN.
.000	15.000	.000	15.000	XREF 26.0000 IN.
15.000	15.000	15.000	15.000	YREF .0000 IN.

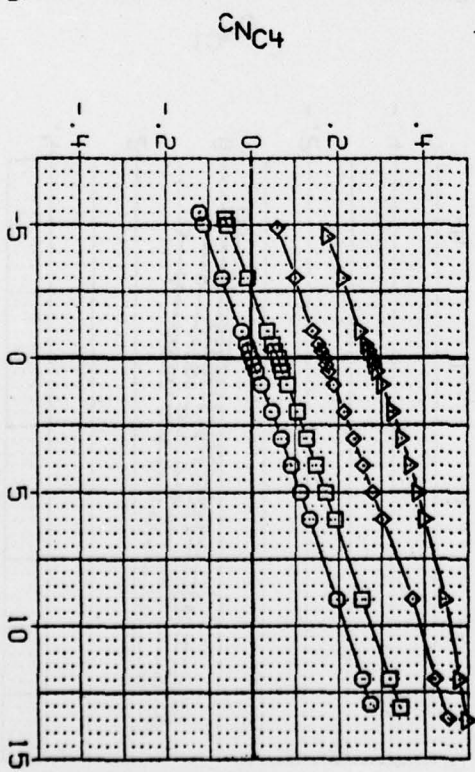
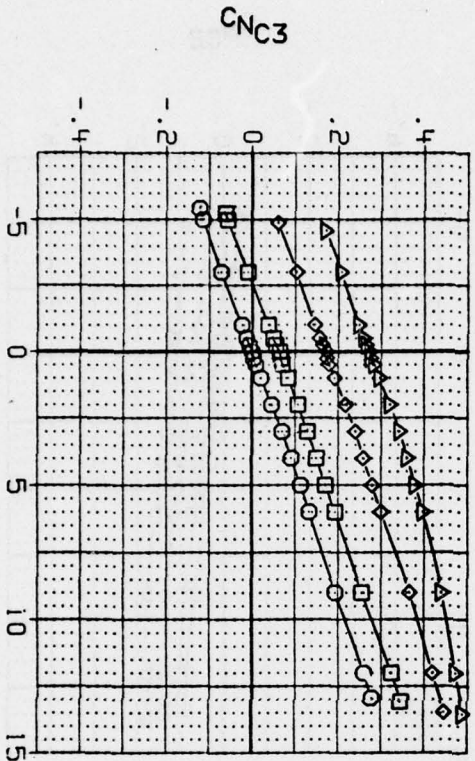
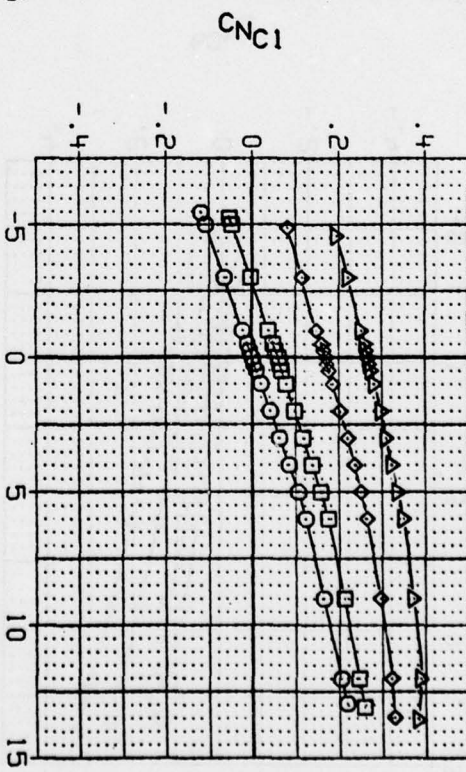
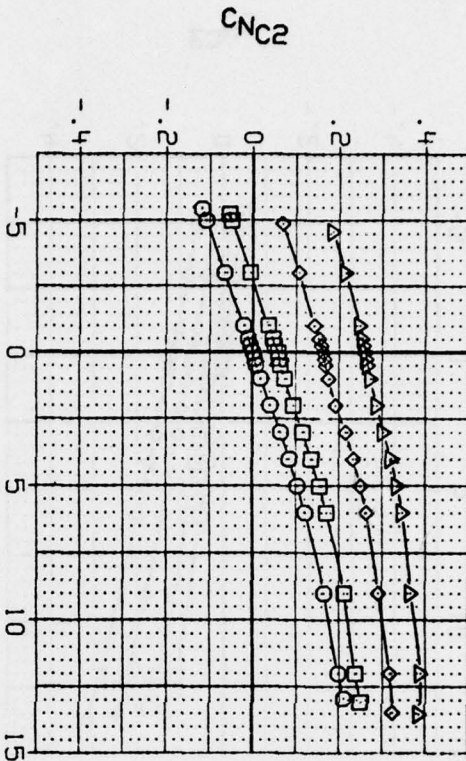


PAGE 140

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH081) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (AXH082) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (AXH083) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (AXH084) \triangle AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH081) \square AEDC V1A-C1A, CANARD CONTROL, BNCS011

(AXH082) \square AEDC V1A-C1A, CANARD CONTROL, BNCS011

(AXH083) \square AEDC V1A-C1A, CANARD CONTROL, BNCS011

(AXH084) \triangle AEDC V1A-C1A, CANARD CONTROL, BNCS011

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000

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9.000 9.000 9.000 9.000

15.000 15.000 15.000 15.000

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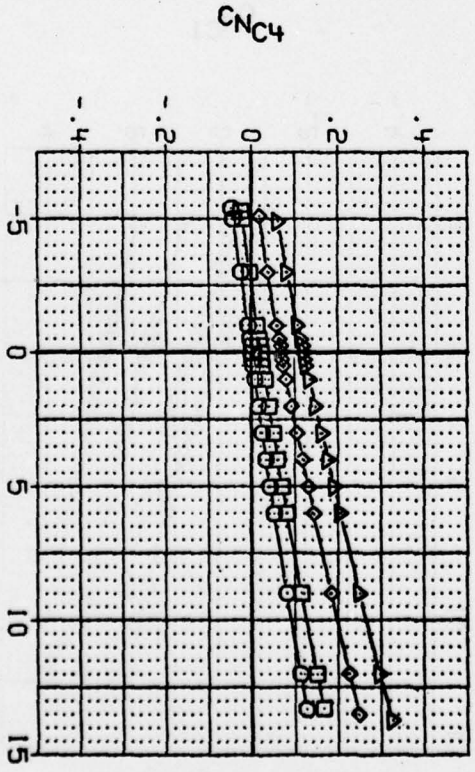
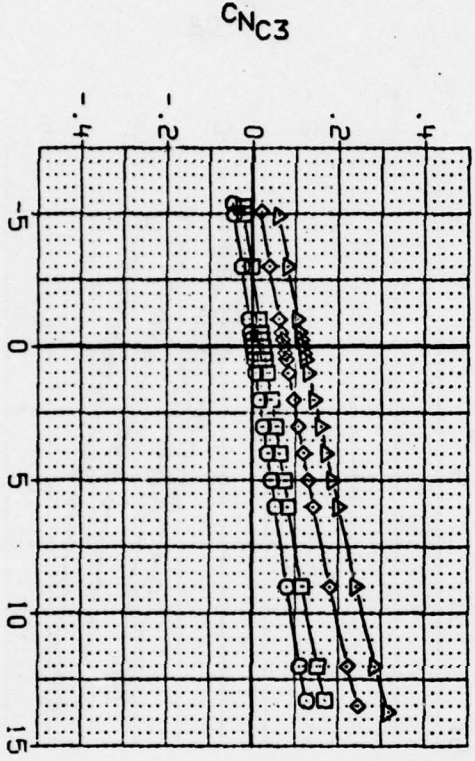
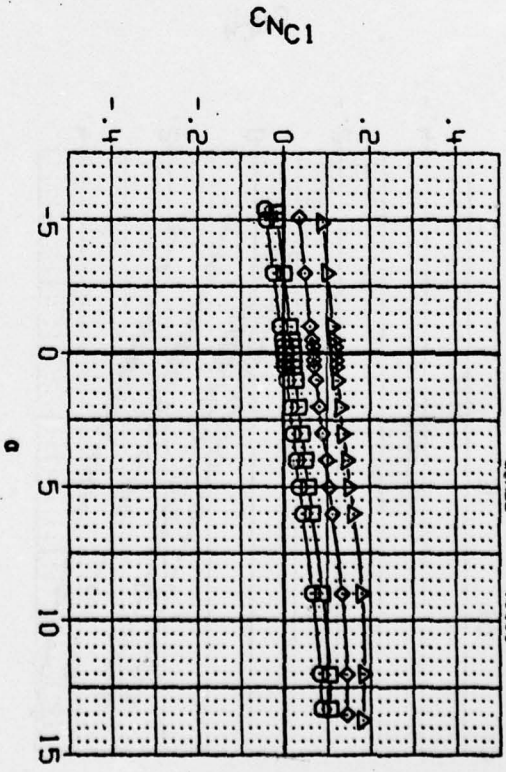
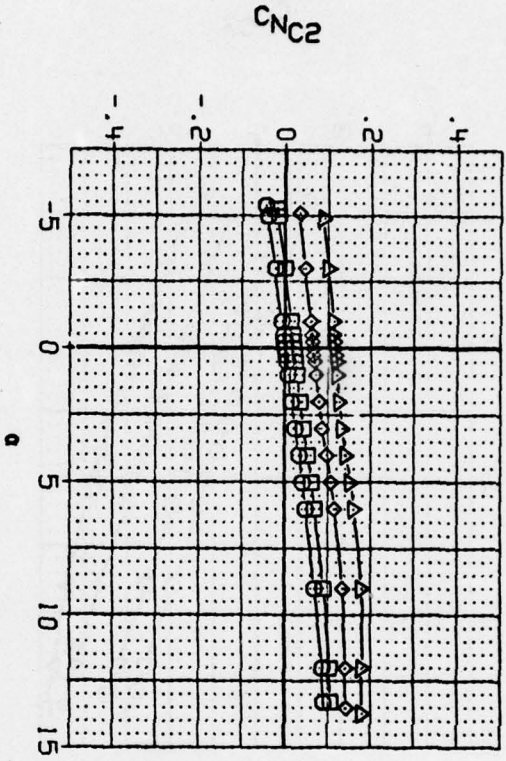
LREF 5.0000 IN.

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XREF 26.0000 IN.

YREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHIAL=0 PHICND=45

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

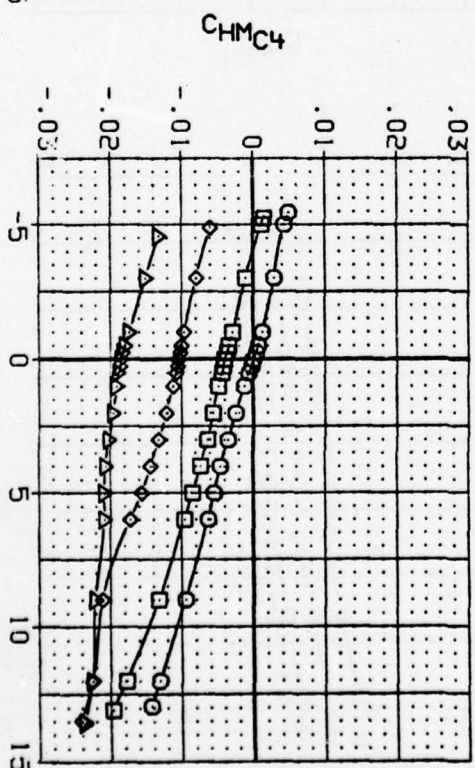
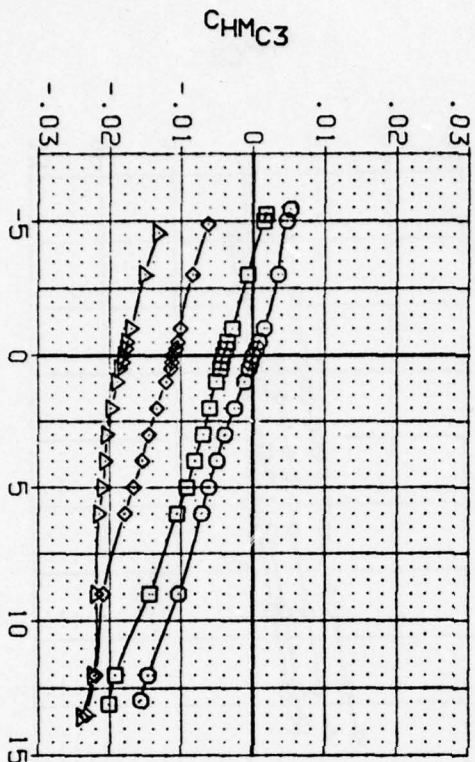
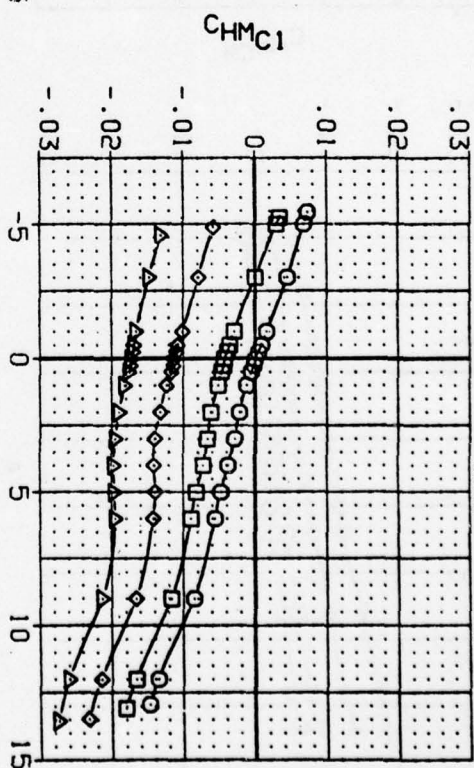
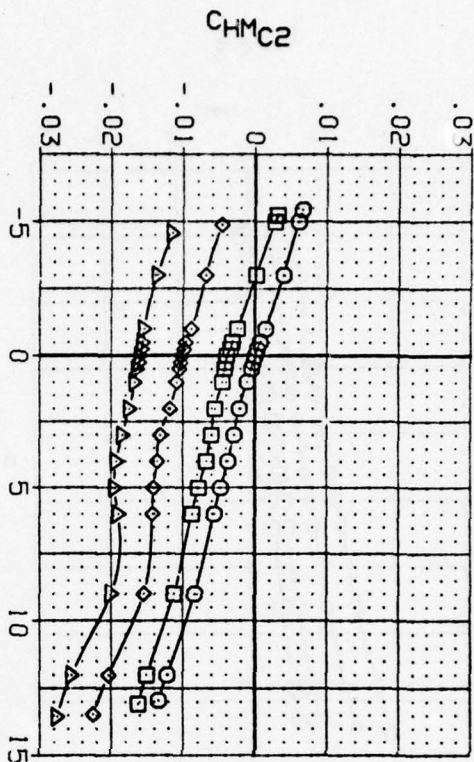
(BXH081) \square AEDC W1A-C1A, CANARD CONTROL, BNSC611
 (BXH082) \square AEDC W1A-C1A, CANARD CONTROL, BNSC611
 (BXH083) \square AEDC W1A-C1A, CANARD CONTROL, BNSC611
 (BXH084) \triangle AEDC W1A-C1A, CANARD CONTROL, BNSC611

DCND1 DCND2 DCND3 DCND4

3.000 3.000 3.000 3.000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YMRP 26.0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

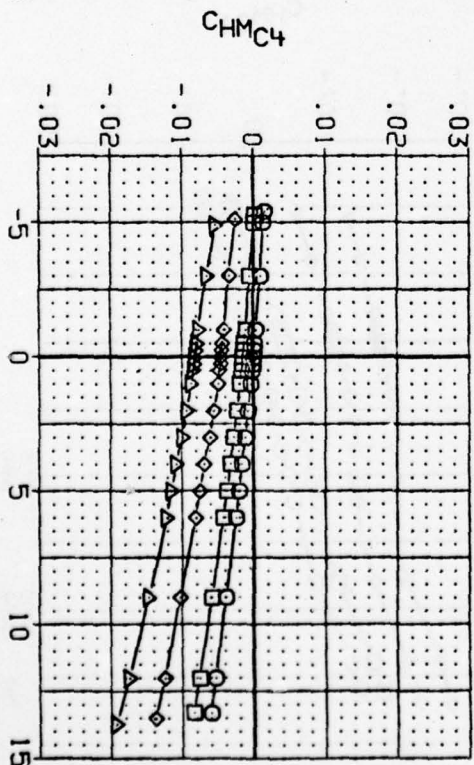
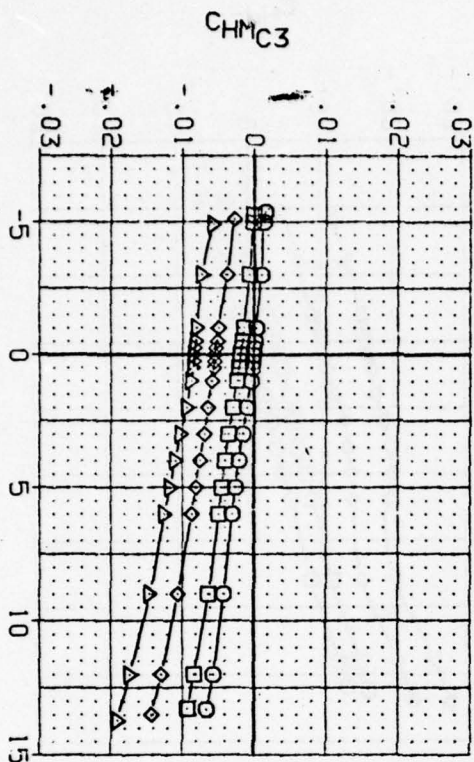
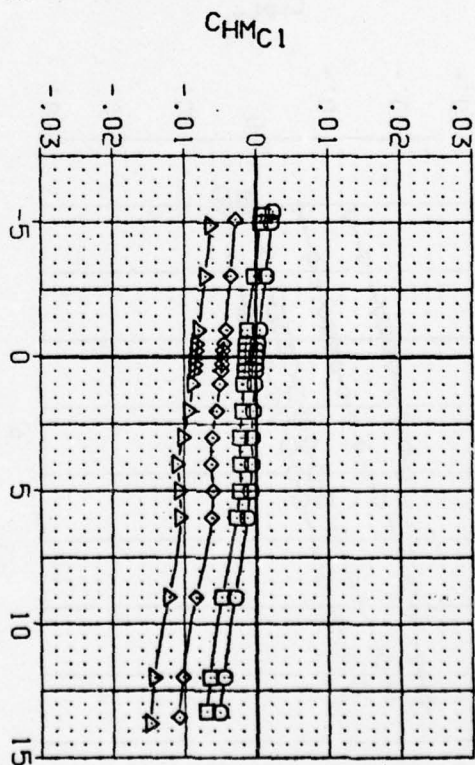
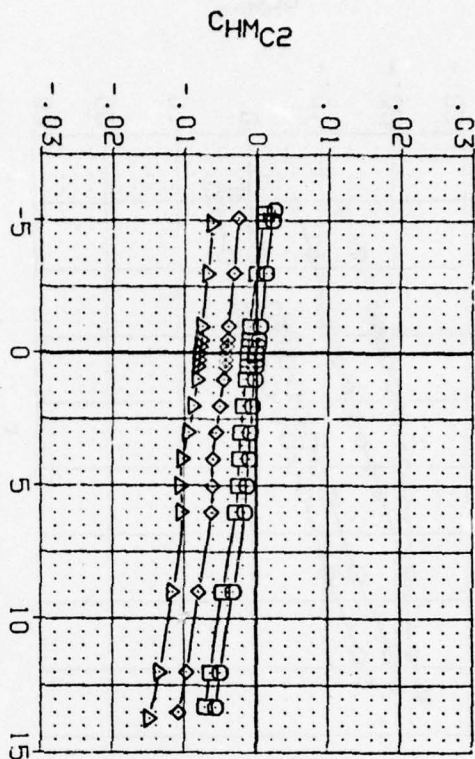
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 (BXH082) \square AEDC V41A-C1A, CANARD CONTROL, BNSCST1
 (BXH093) \triangle AEDC V41A-C1A, CANARD CONTROL, BNSCST1
 (BXH084) \triangle AEDC V41A-C1A, CANARD CONTROL, BNSCST1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

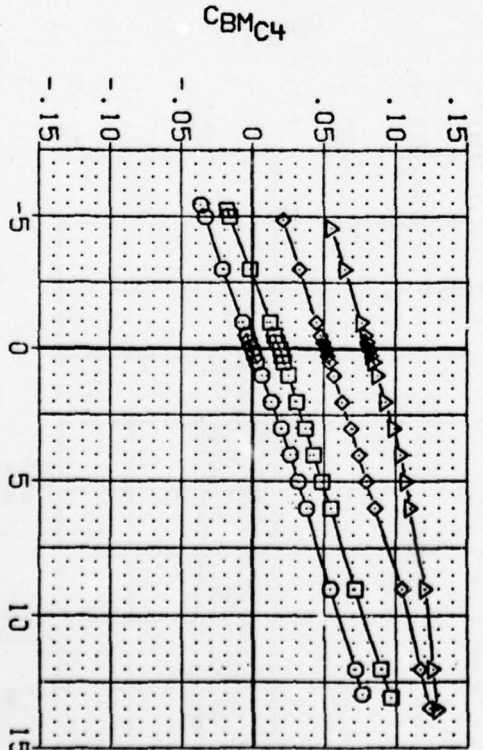
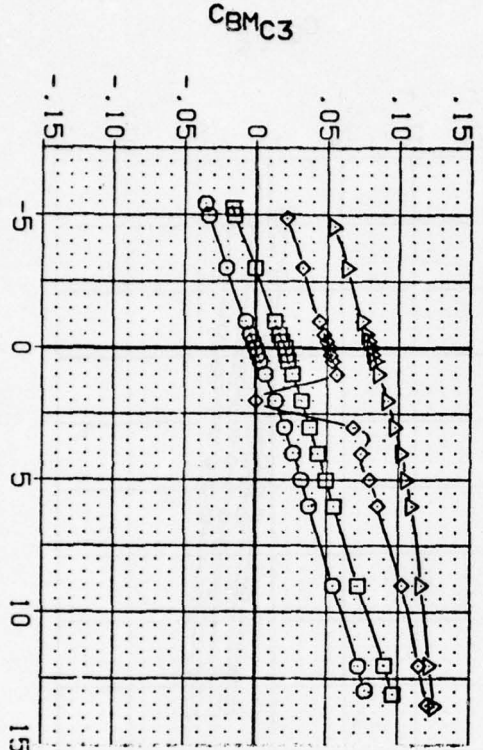
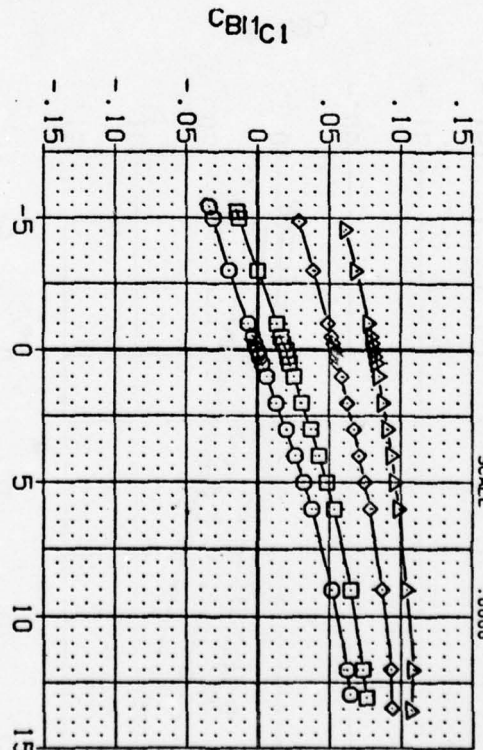
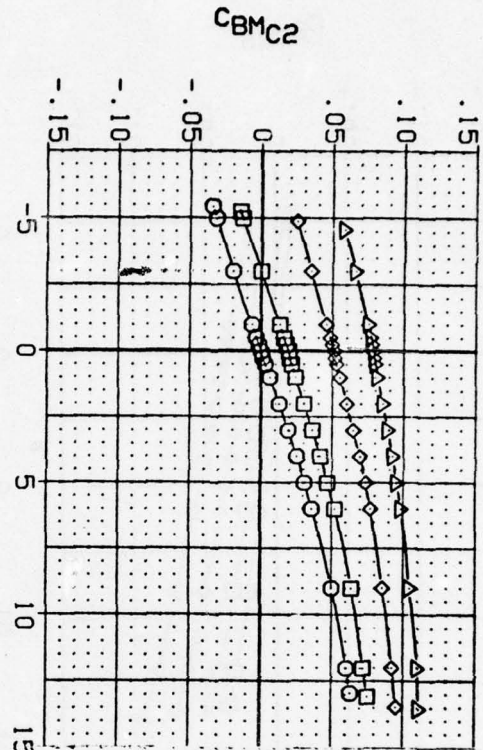
(BXH081) \square AEDC V41A-C1A, CANARD CONTROL, BNSC6T1
 (BXH082) \square AEDC V41A-C1A, CANARD CONTROL, BNSC6T1
 (BXH083) \square AEDC V41A-C1A, CANARD CONTROL, BNSC6T1
 (BXH084) Δ AEDC V41A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 25.0000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

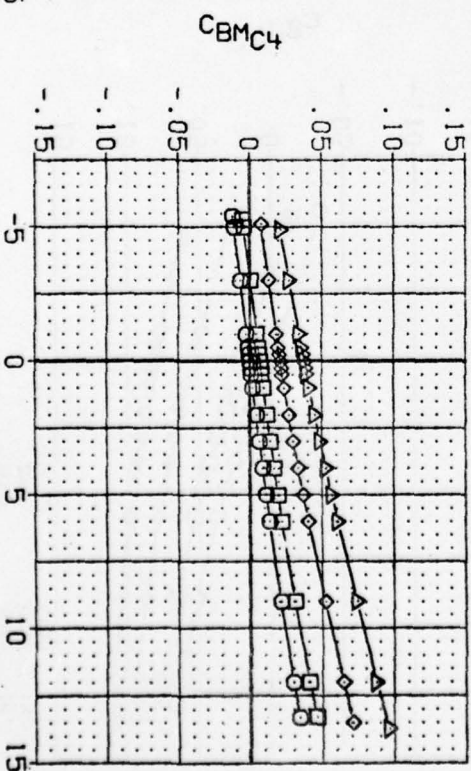
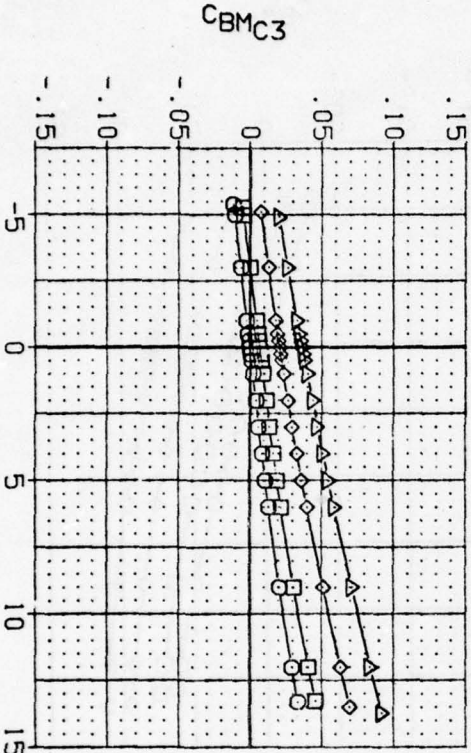
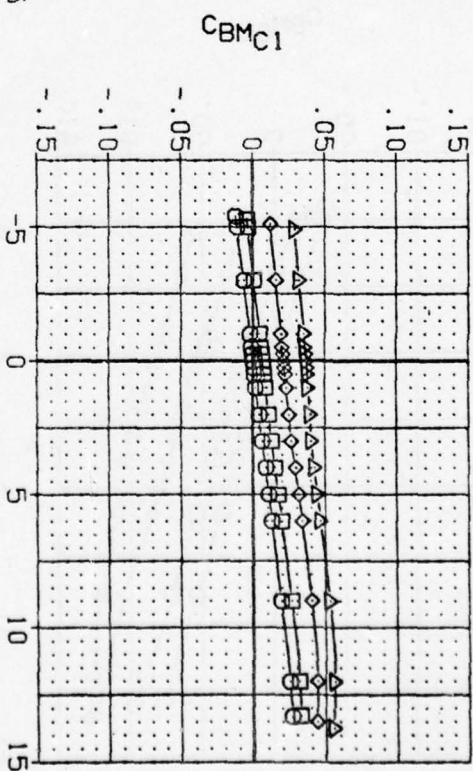
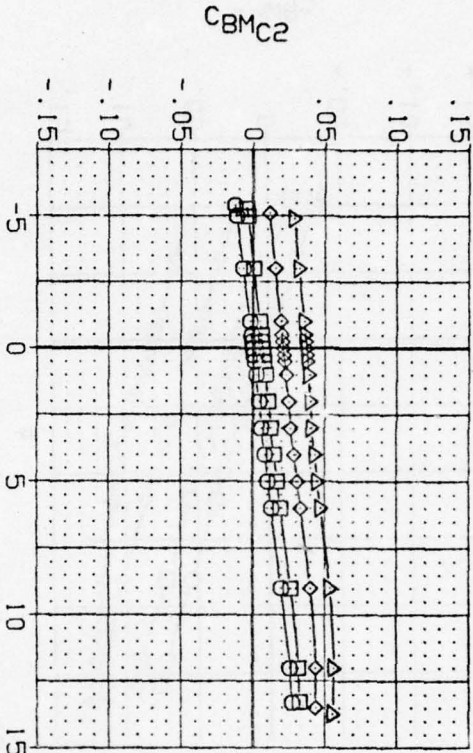
PHITAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH081) \square AEDC V41A-C1A, CANARD CONTROL, BNSCST1
 (BXH082) \square AEDC V41A-C1A, CANARD CONTROL, BNSCST1
 (BXH083) \square AEDC V41A-C1A, CANARD CONTROL, BNSCST1
 (BXH084) \triangle AEDC V41A-C1A, CANARD CONTROL, BNSCST1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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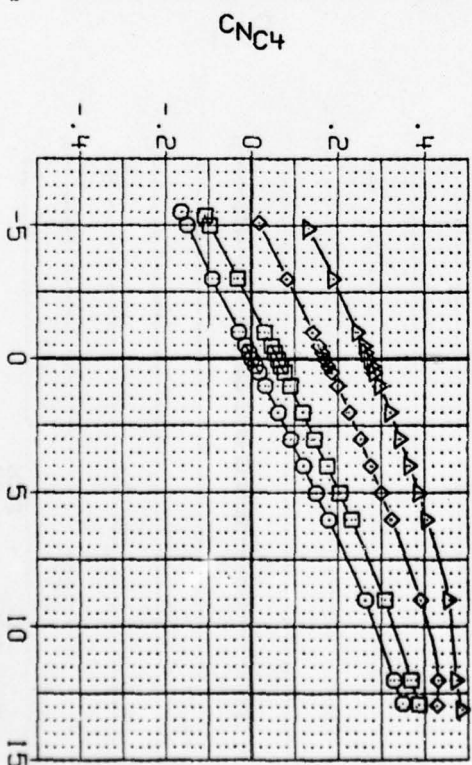
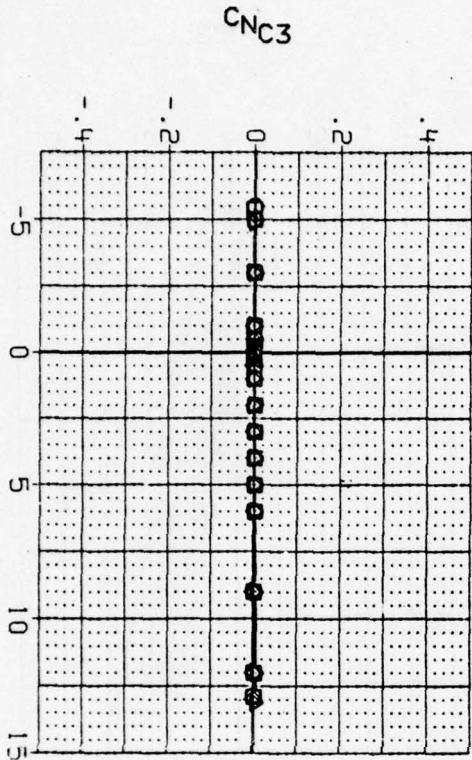
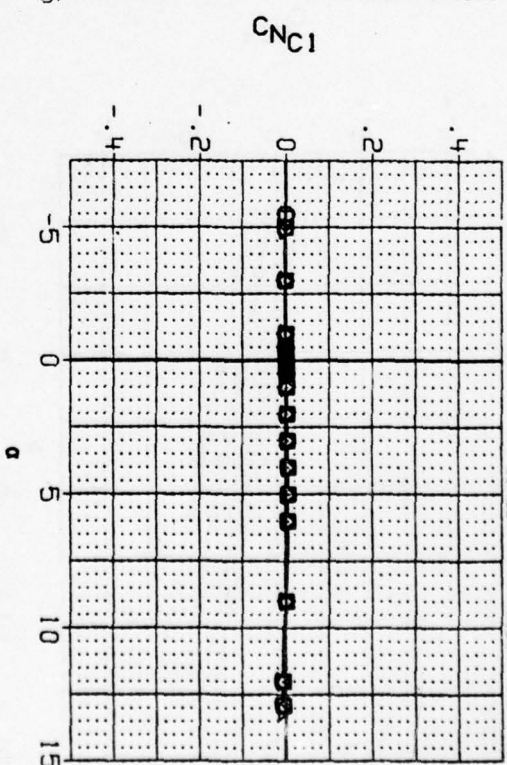
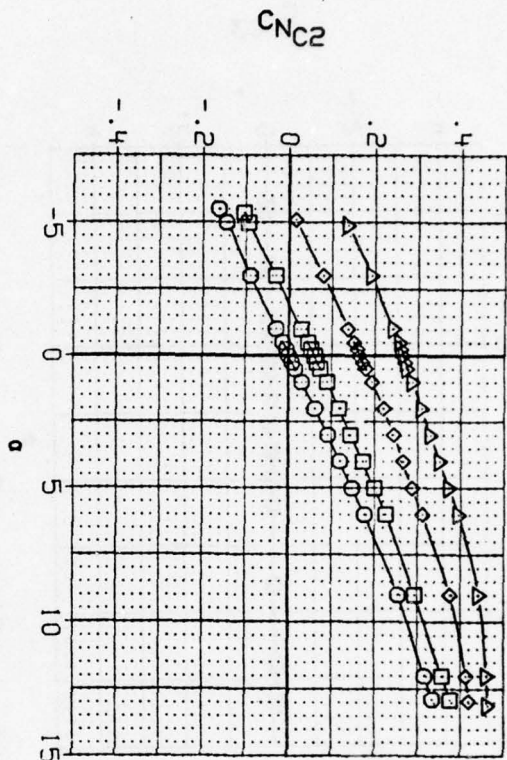


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHIAL=0$
 $PHICND=45$
 (B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH085) \square AEDC VMIA-CIA, CANARD CONTROL, BN6C6T1
 (AXH086) \square AEDC VMIA-CIA, CANARD CONTROL, BN6C6T1
 (AXH087) \square AEDC VMIA-CIA, CANARD CONTROL, BN6C6T1
 (AXH088) \triangle AEDC VMIA-CIA, CANARD CONTROL, BN6C6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 3.000
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REFERENCE INFORMATION
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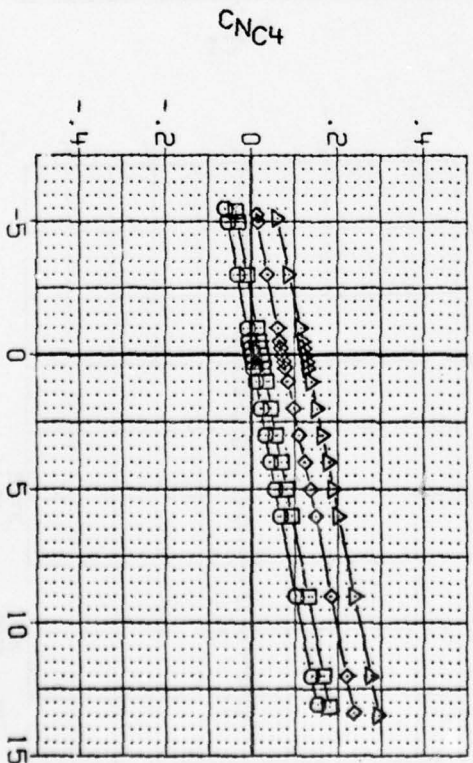
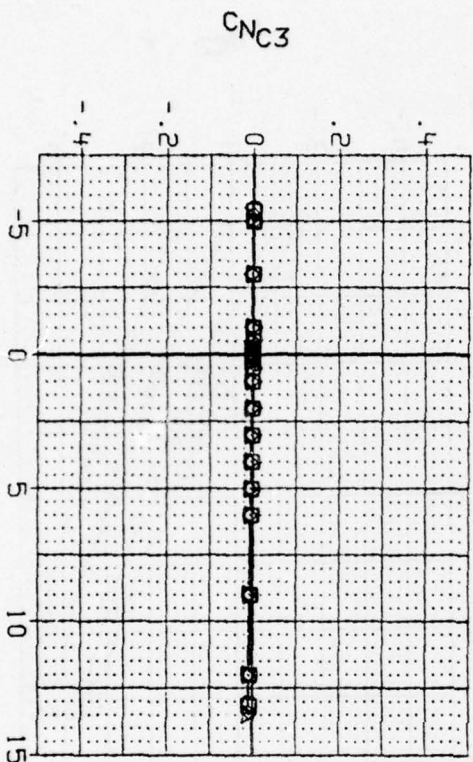
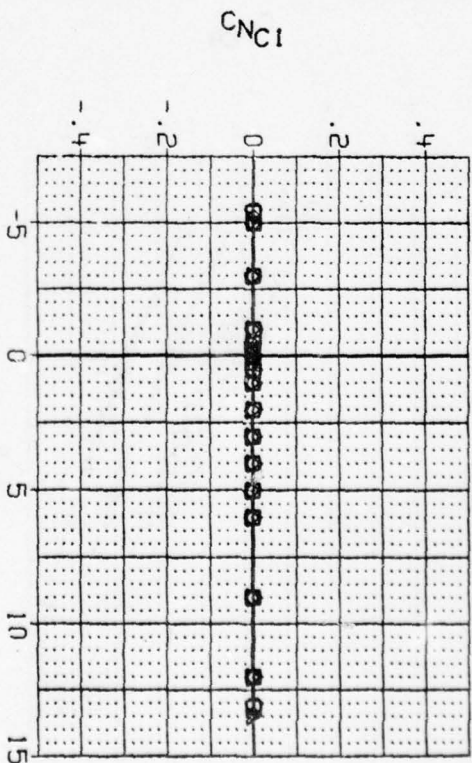
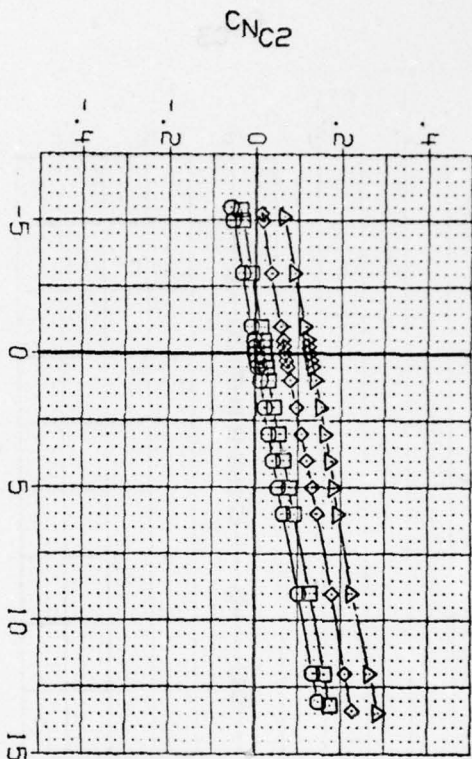


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=0$ $PHICND=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH085) □ AEDC VJ1A-C1A, CANARD CONTROL, BNSC611
 (AXH086) □ AEDC VJ1A-C1A, CANARD CONTROL, BNSC611
 (AXH087) △ AEDC VJ1A-C1A, CANARD CONTROL, BNSC611
 (AXH088) △ AEDC VJ1A-C1A, CANARD CONTROL, BNSC611

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 3.000
 .000 15.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (B) MACH = 3.01

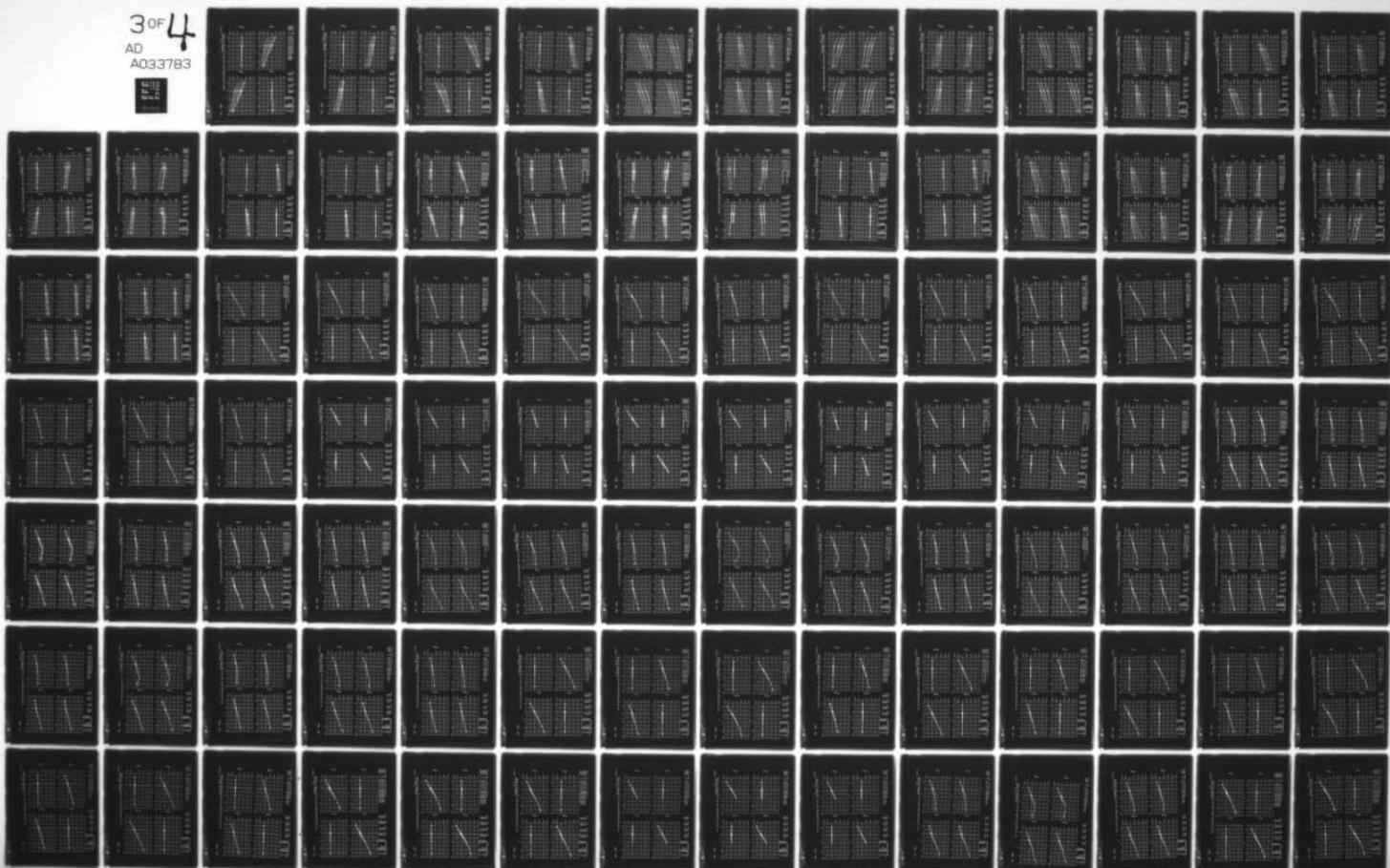
AD-A033 783

ARMY MISSILE RESEARCH DEVELOPMENT AND ENGINEERING LAB--ETC F/G 20/4
AN EXPERIMENTAL INVESTIGATION OF THE AERODYNAMIC CHARACTERISTIC--ETC(U)
OCT 76 J R BURT
RD-77-5

UNCLASSIFIED

NL

3 OF 4
AD
A033783



DATA SET SYMBOL CONFIGURATION DESCRIPTION

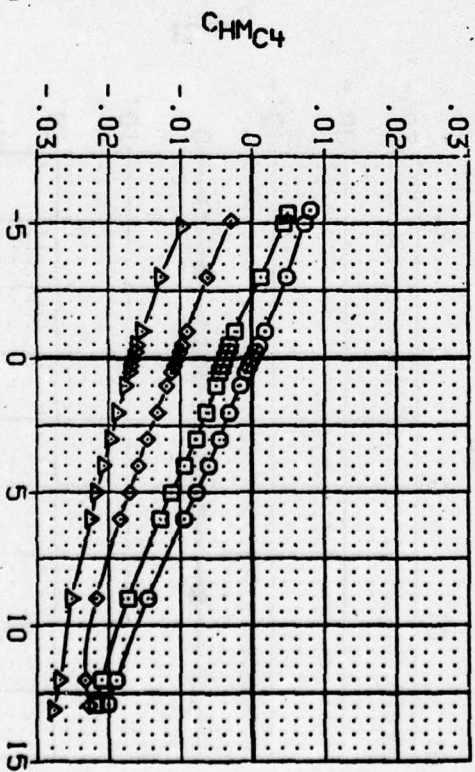
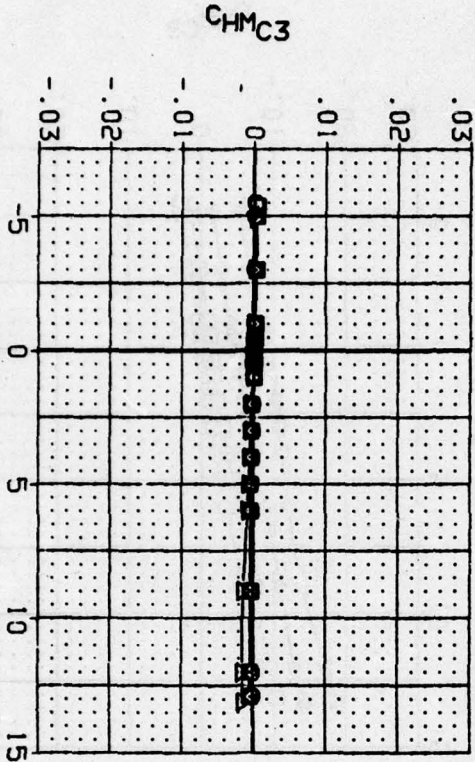
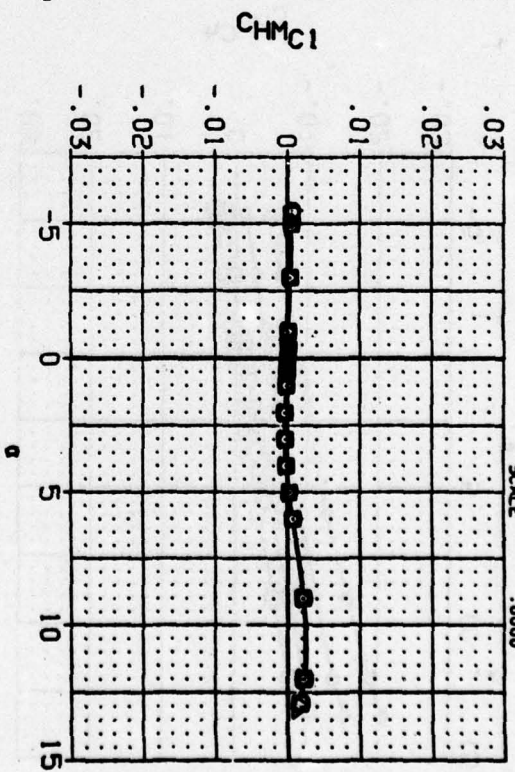
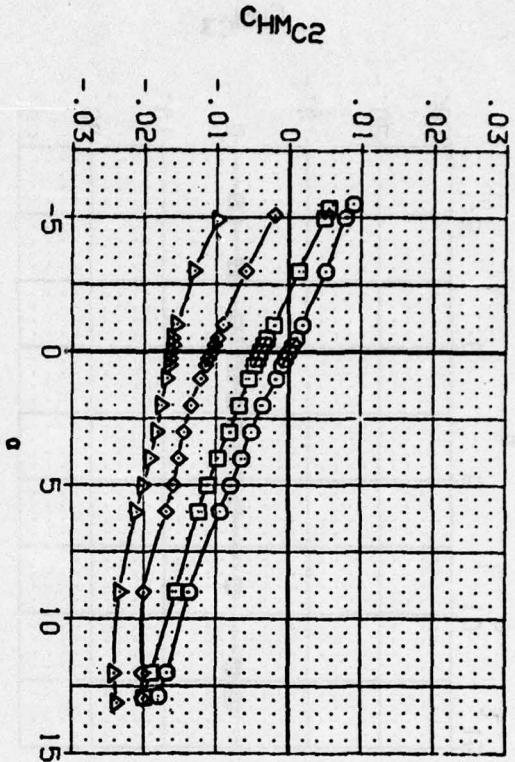
(BXH085) \square AEDC WJIA-C1A, CANARD CONTROL, BNGCST1
 (BXH086) \square AEDC WJIA-C1A, CANARD CONTROL, BNGCST1
 (BXH087) \square AEDC WJIA-C1A, CANARD CONTROL, BNGCST1
 (BXH088) Δ AEDC WJIA-C1A, CANARD CONTROL, BNGCST1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
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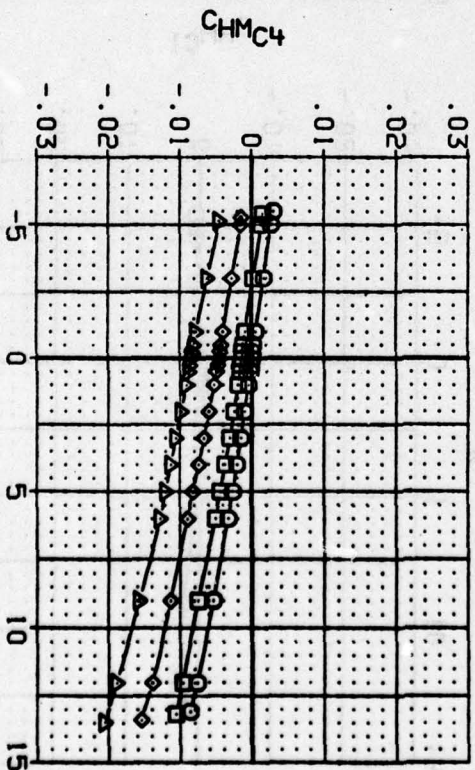
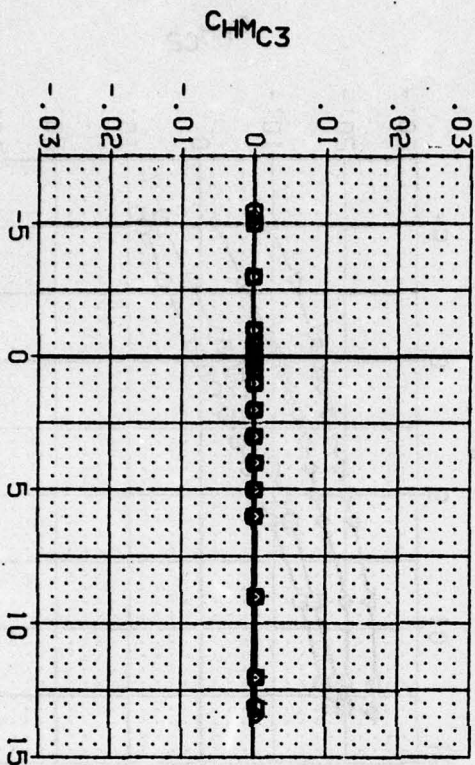
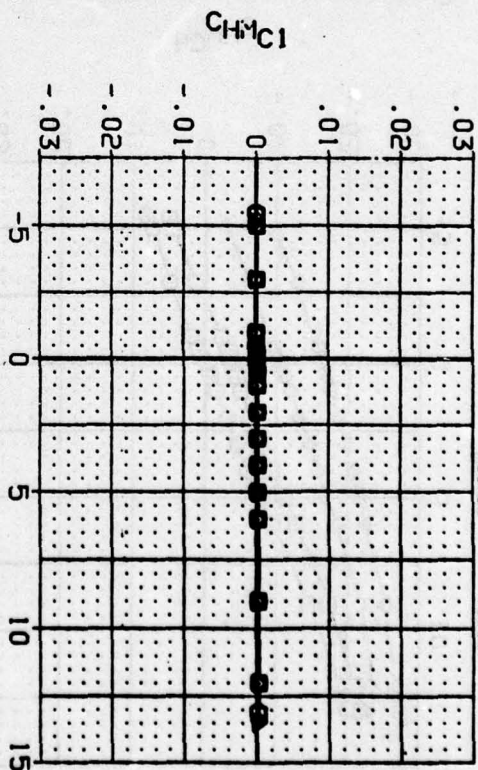
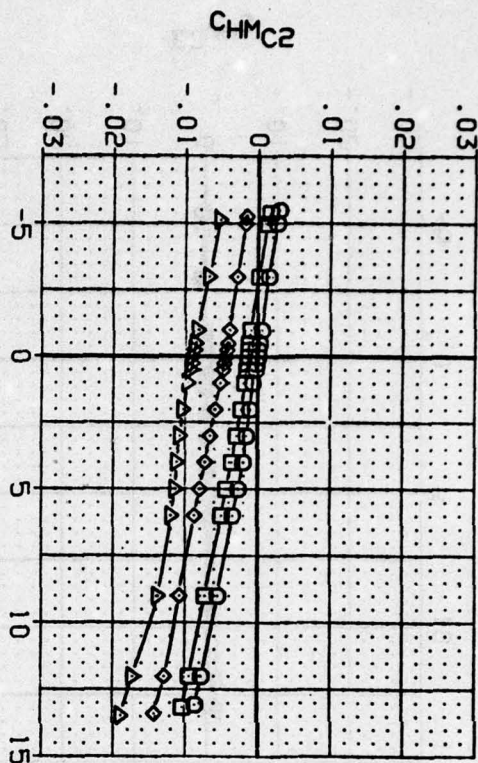


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$\Phi_{HIT} = 0$
 $\Phi_{HICND} = 0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BX-085) \square AEDC W1A-C1A, CANARD CONTROL, BNC6511
 (BX-085) \diamond AEDC W1A-C1A, CANARD CONTROL, BNC6511
 (BX-087) \triangle AEDC W1A-C1A, CANARD CONTROL, BNC6511
 (BX-088) \triangle AEDC W1A-C1A, CANARD CONTROL, BNC6511

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000
 REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0010 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHIAL=0$ $PHICND=0$
 (B) MACH = 3.01

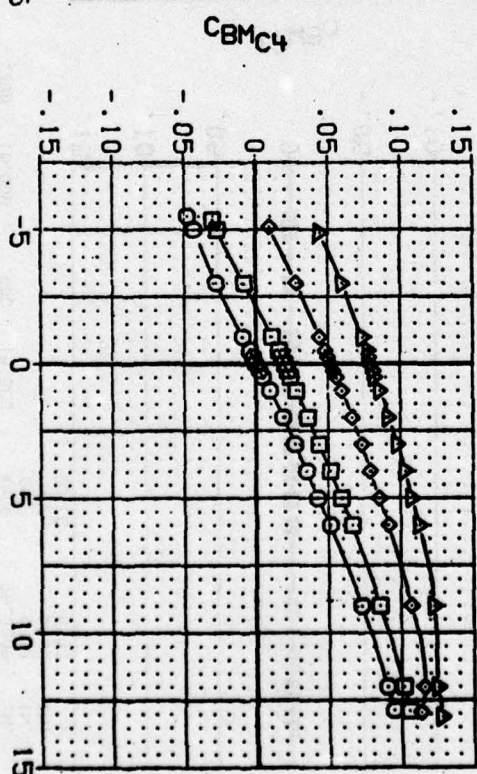
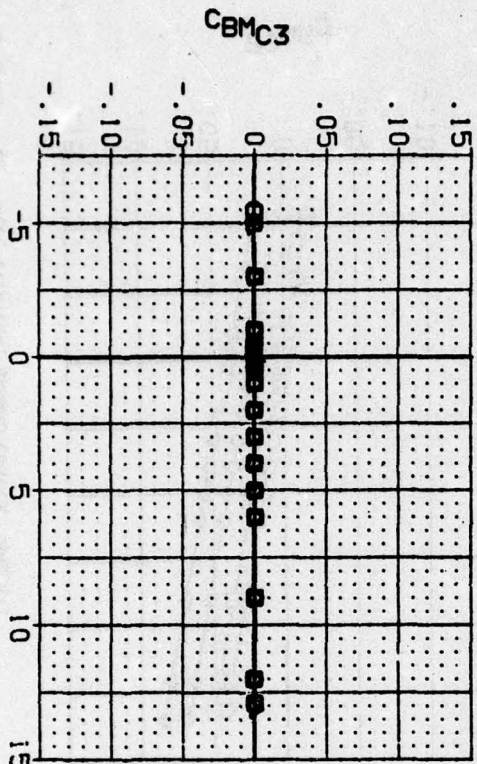
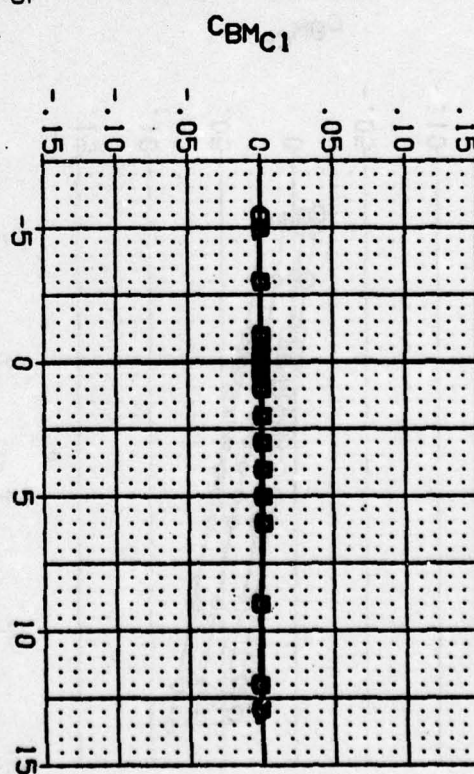
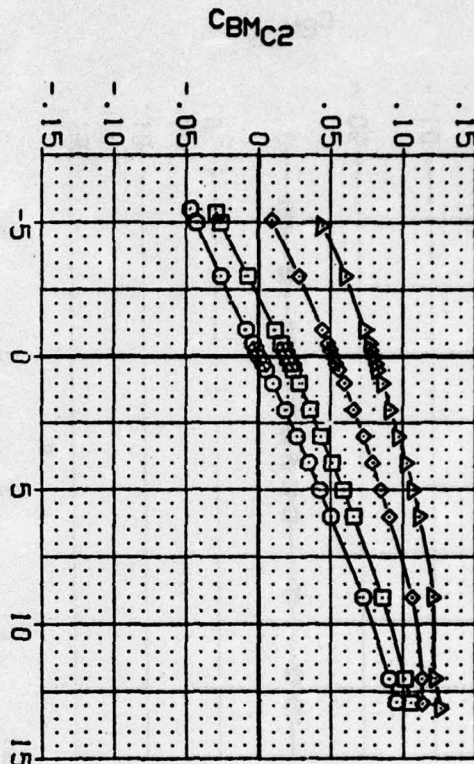
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH085)	○	AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
(BXH086)	□	AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
(BXH087)	△	AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
(BXH088)	△	AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1	DCND2	DCND3	DCND4
.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000

REFERENCE INFORMATION

SREF	19.6350	50. IN.
LBREF	5.0000	IN.
BRF	5.0000	IN.
YREF	26.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	

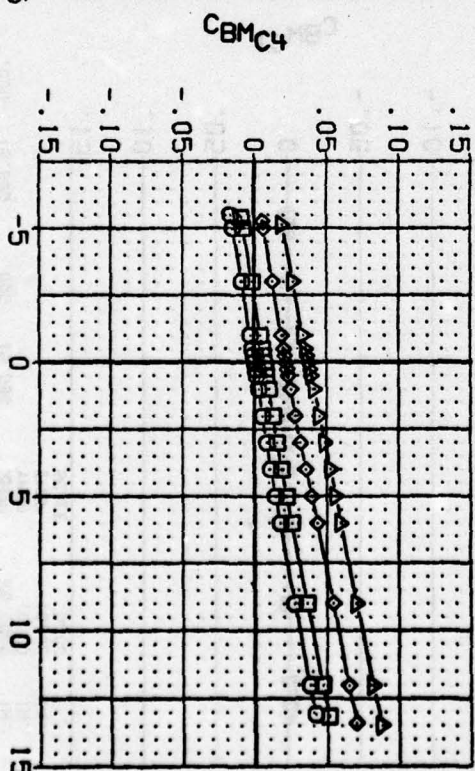
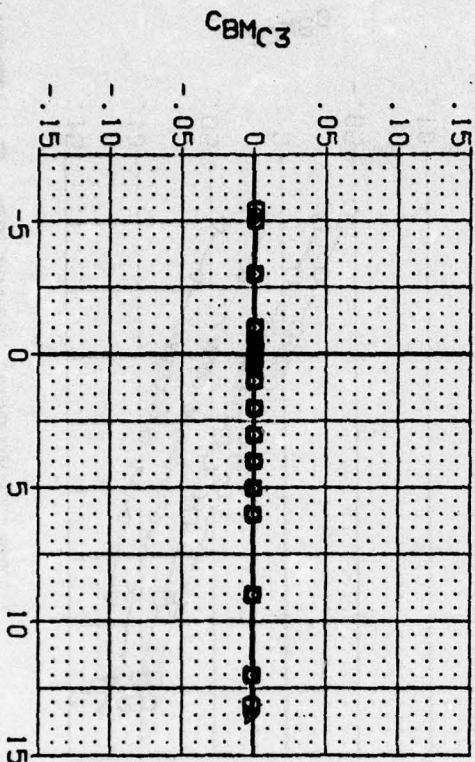
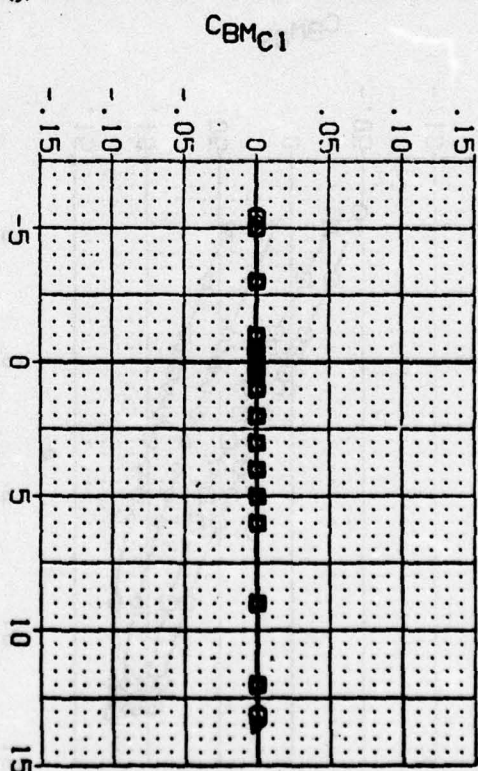
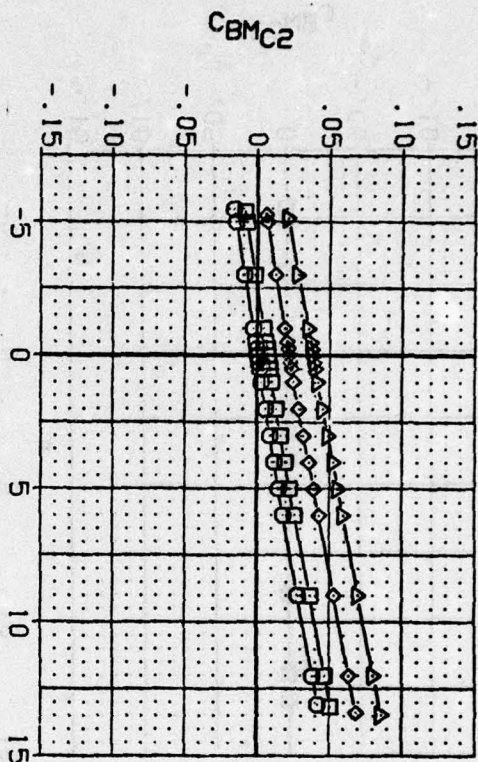


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{HITAL}=0$ $\Phi_{HICND}=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH085) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (BXH086) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (BXH087) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (BXH088) Δ AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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 BREF 5.0000 IN.
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 SCALE .0000

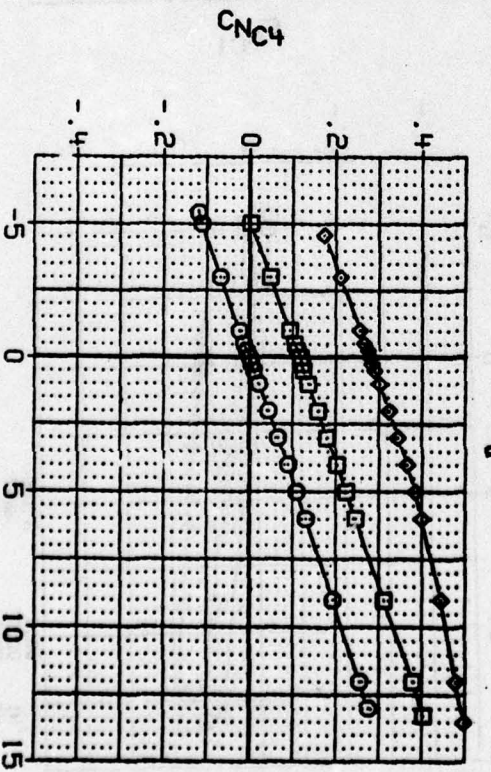
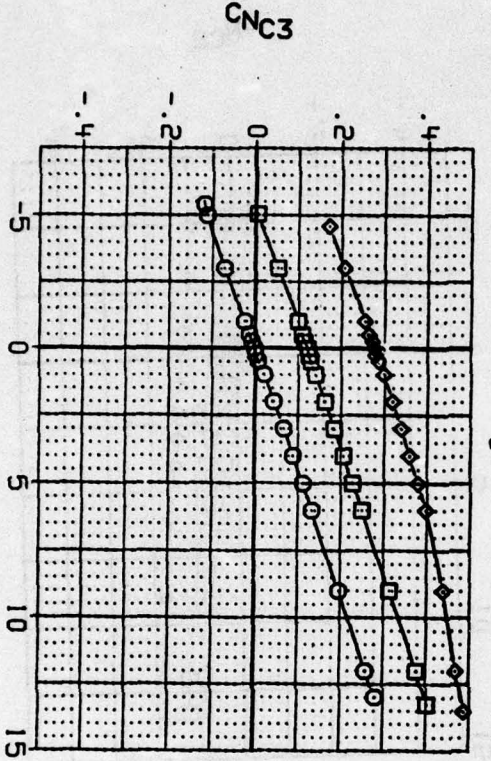
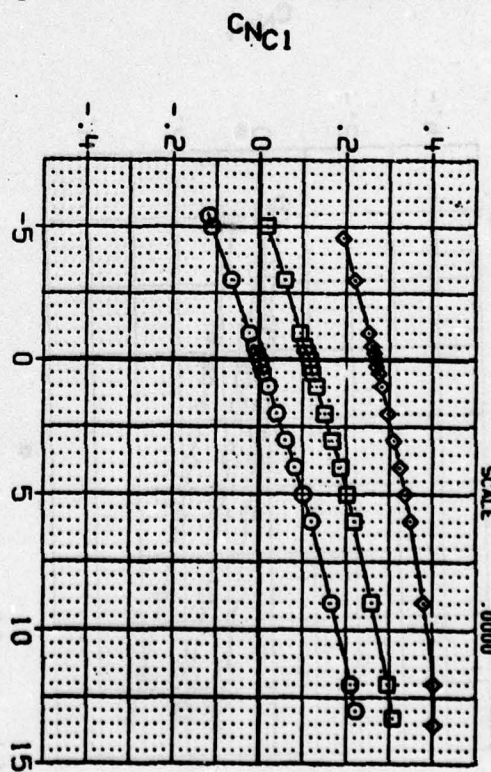
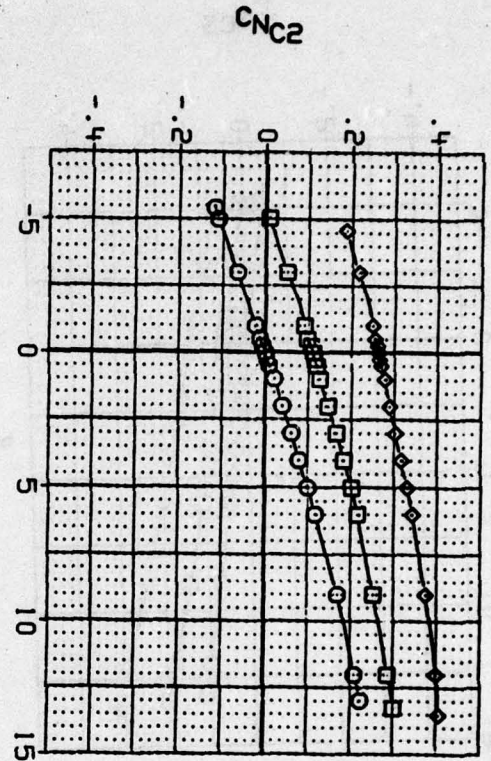


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=0$
 $PHICND=0$
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (A) (089) AEDC W1A-C1A, CANARD CONTROL, BNEC611
 (A) (091) AEDC W1A-C1A, CANARD CONTROL, BNEC611
 (A) (091) AEDC W1A-C1A, CANARD CONTROL, BNEC611

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 6.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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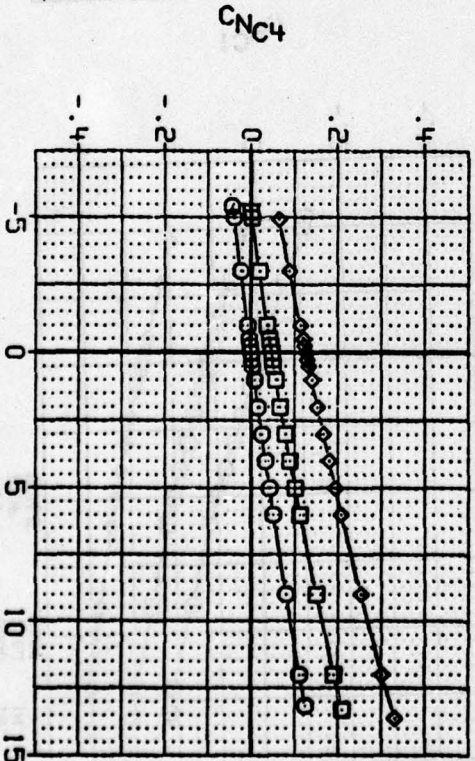
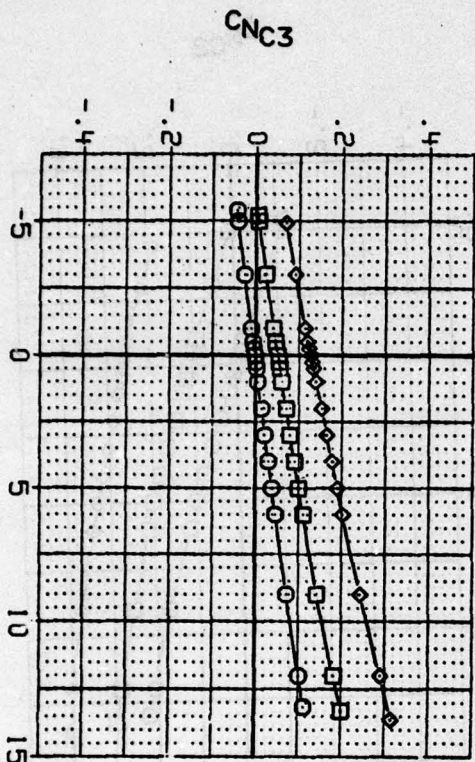
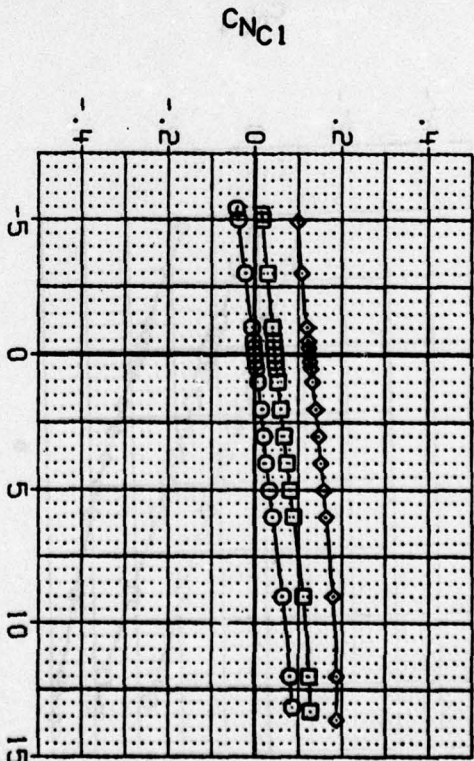
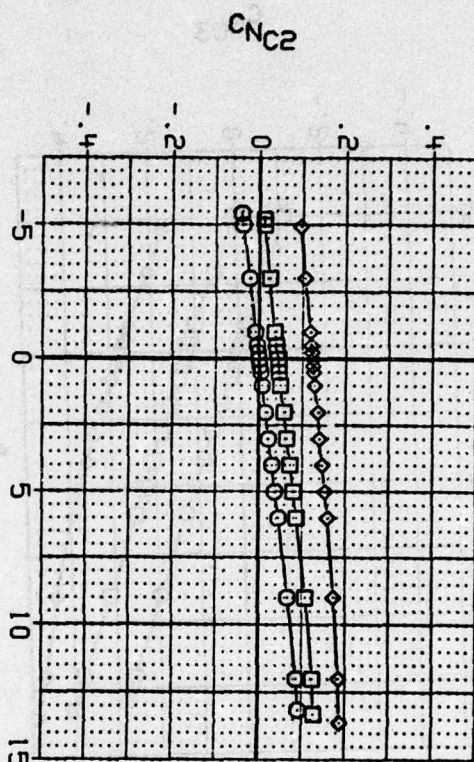


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH089) \square AEDC WJIA-CIA, CANARD CONTROL, BN6C6T1
 (AXH090) \diamond AEDC WJIA-CIA, CANARD CONTROL, BN6C6T1
 (AXH091) \square AEDC WJIA-CIA, CANARD CONTROL, BN6C6T1

DCND1 .000
 .000 6.000
 15.000
 DCND2 .000
 .000 6.000
 15.000
 DCND3 .000
 .000 6.000
 15.000
 DCND4 .000
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 15.000

REFERENCE INFORMATION
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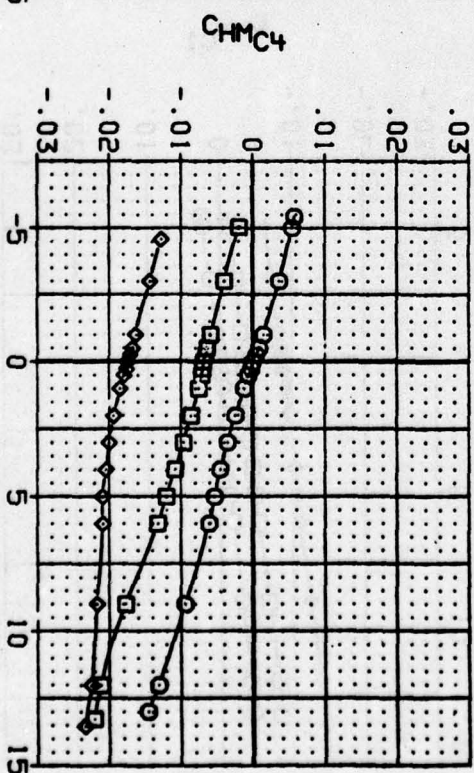
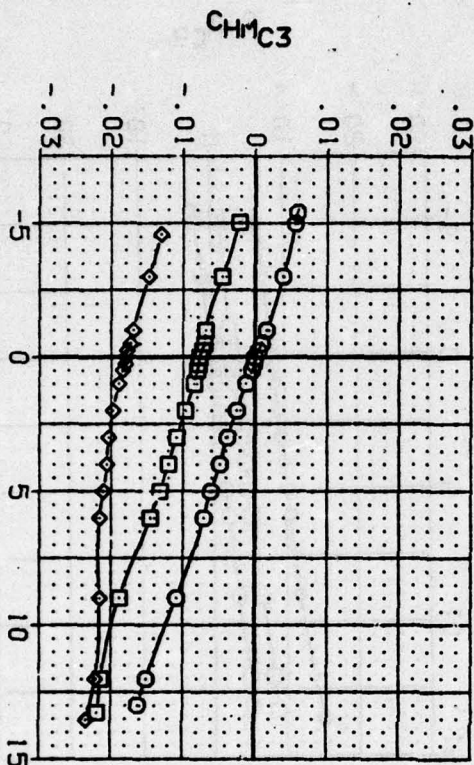
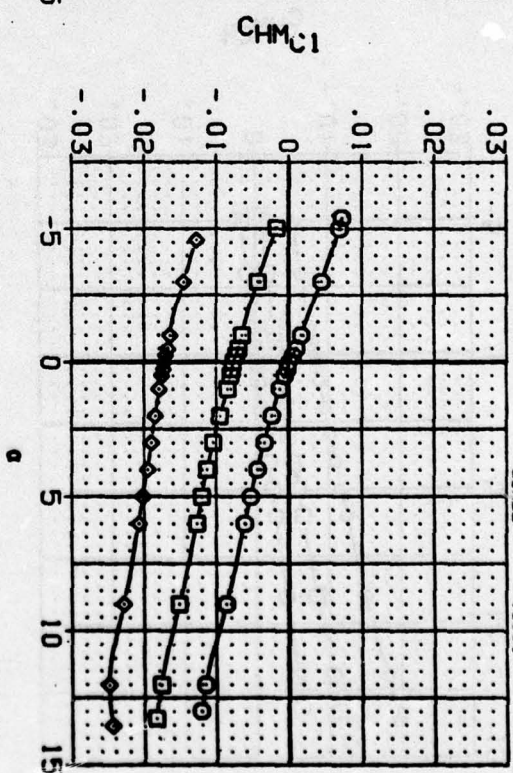
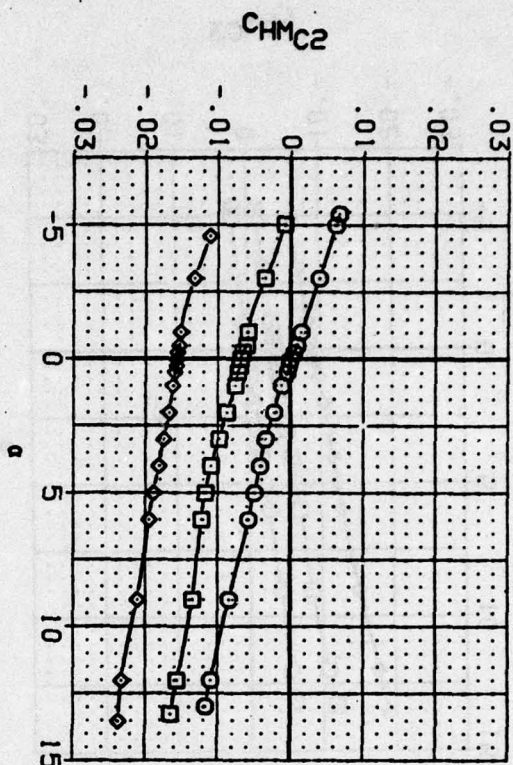


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=45
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH089) \square AEDC W1A-C1A, CANARD CONTROL, BNSCST1
 (BXH093) \square AEDC W1A-C1A, CANARD CONTROL, BNSCST1
 (BXH091) \diamond AEDC W1A-C1A, CANARD CONTROL, BNSCST1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 5.000 5.000 5.000 5.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 SCALE .0000

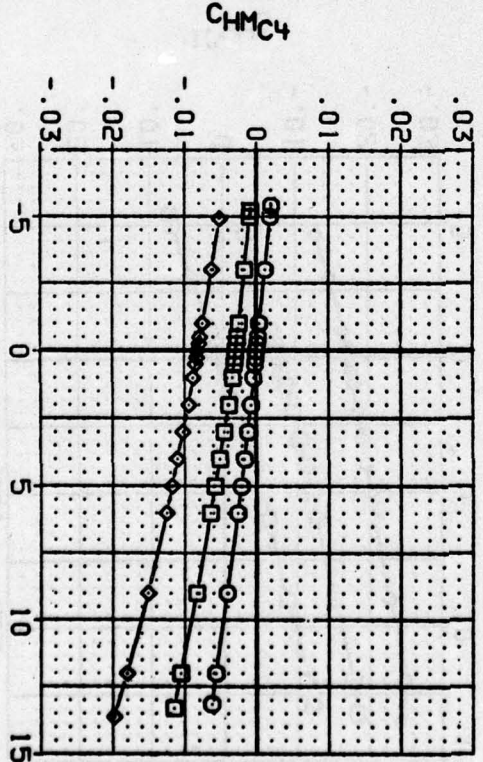
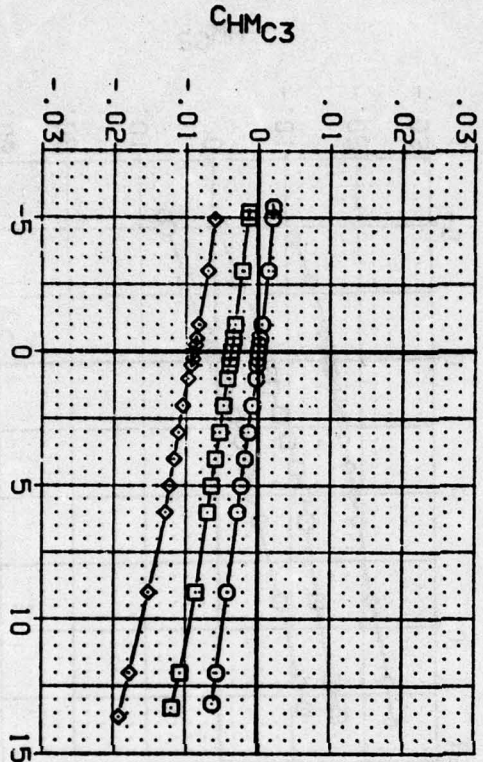
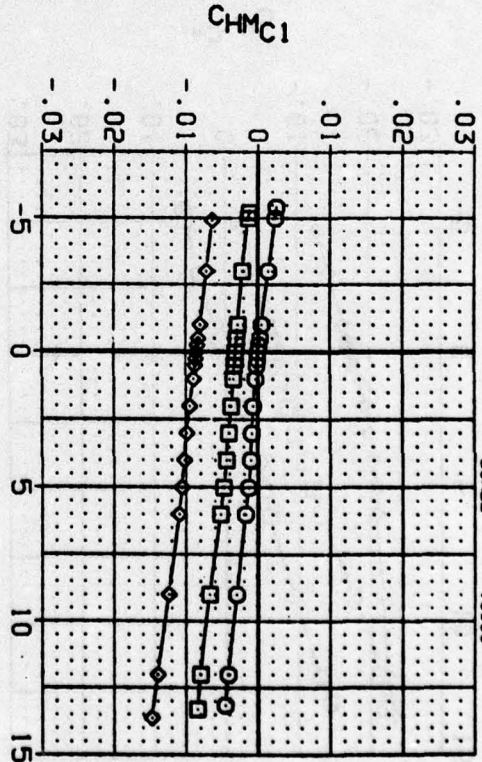
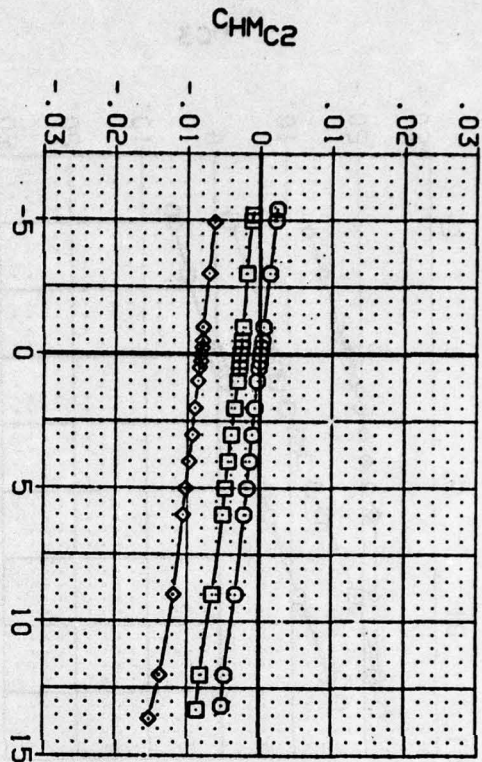


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHICAL=0 PHICND=45
 (A)MACH = 1.51

DATA SET SYMBOL. CONFIGURATION DESCRIPTION
 (BX-099) C AEDC W1A-C1A, CANARD CONTROL, BRSC611
 (BX-093) □ AEDC W1A-C1A, CANARD CONTROL, BRSC611
 (BX-091) AEDC W1A-C1A, CANARD CONTROL, BRSC611

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 6.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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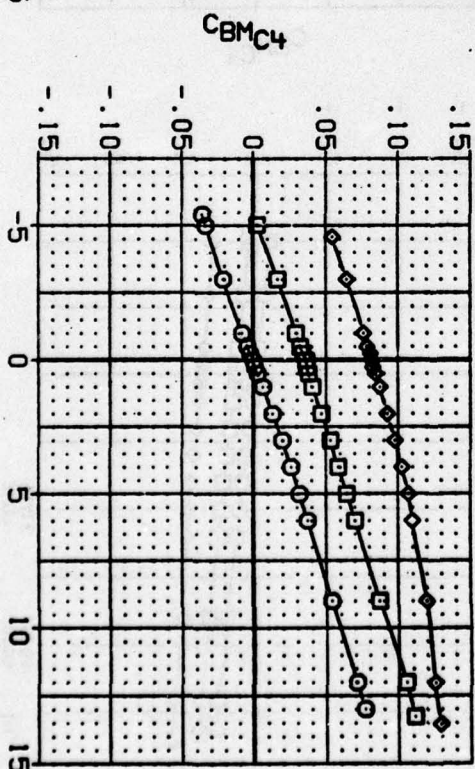
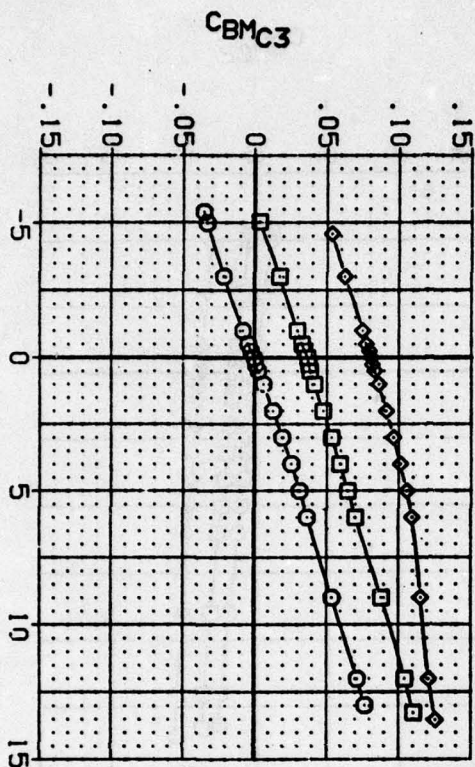
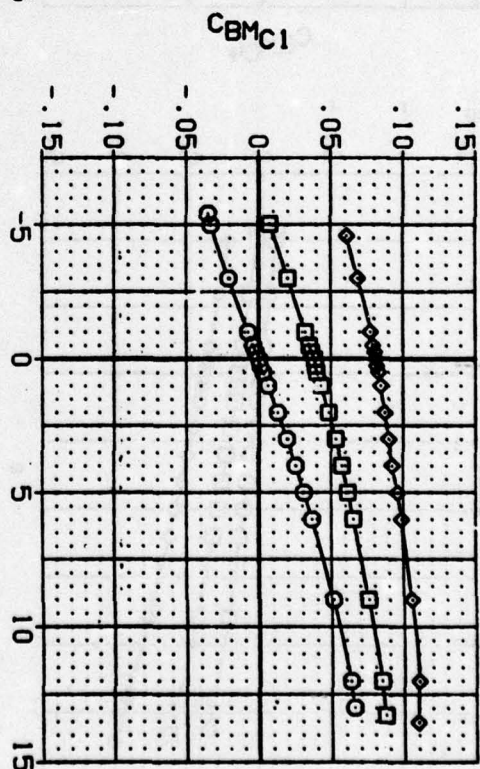
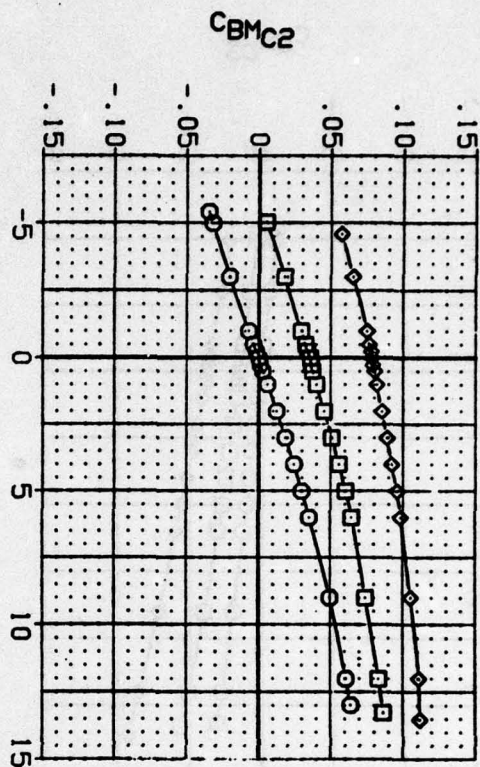


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=45
 (E) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH089) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (BXH090) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (BXH091) \diamond AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 5.000 5.000 5.000 5.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 SCALE .0000

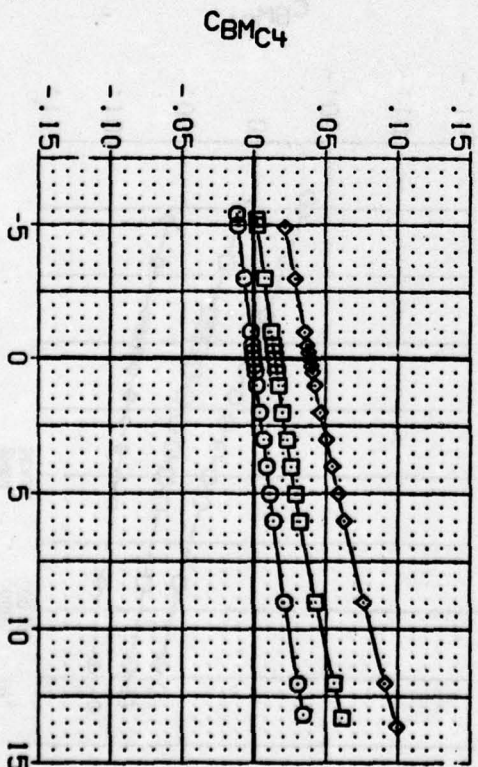
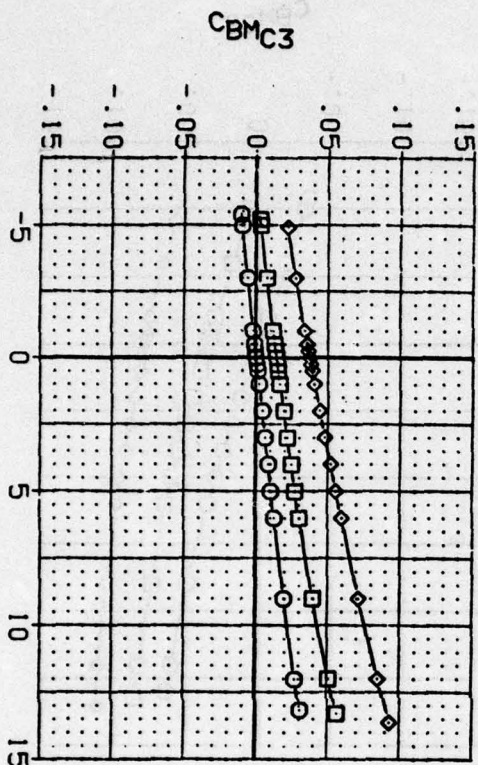
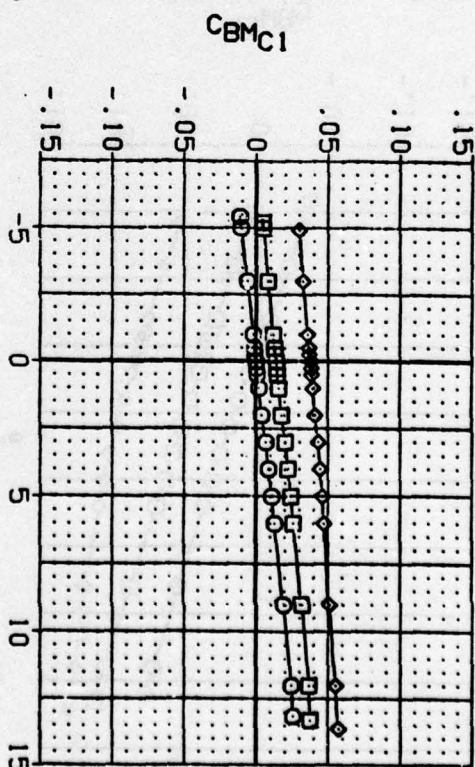
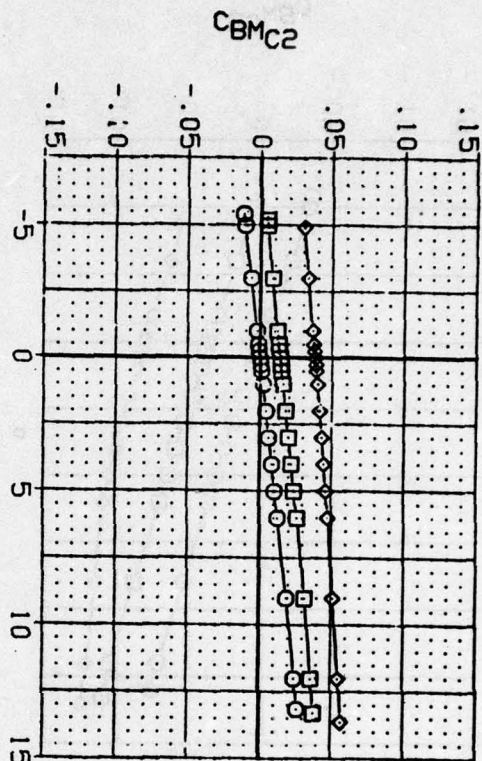


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHITAL=0$ $PHICND=45$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH089) \square AEDC V41A-C1A, CANARD CONTROL, BNC66T1
 (BXH090) \square AEDC V41A-C1A, CANARD CONTROL, BNC66T1
 (BXH091) \diamond AEDC V41A-C1A, CANARD CONTROL, BNC66T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .030 .000
 6.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SPC 19.6350 SQ. IN.
 LREF 5.0000 IN.
 GREF 5.0000 IN.
 YMRP 25.0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $PHIAL=0$ $PHICND=45$
 $(B)MACH = 3.01$

DATA SET SYMBOL CONFIGURATION DESCRIPTION

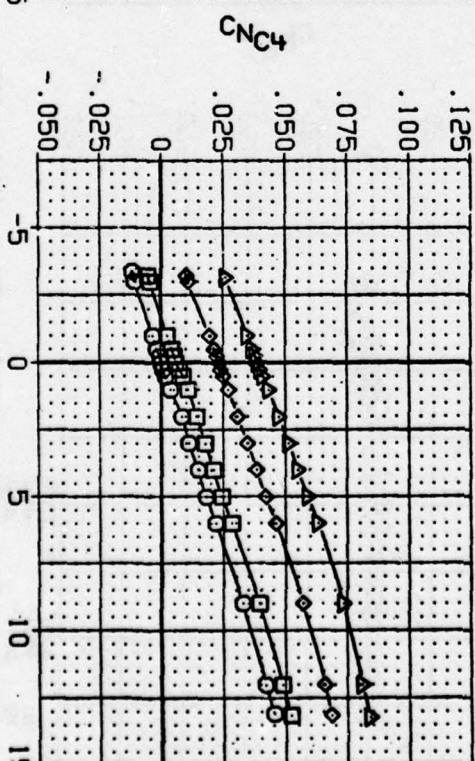
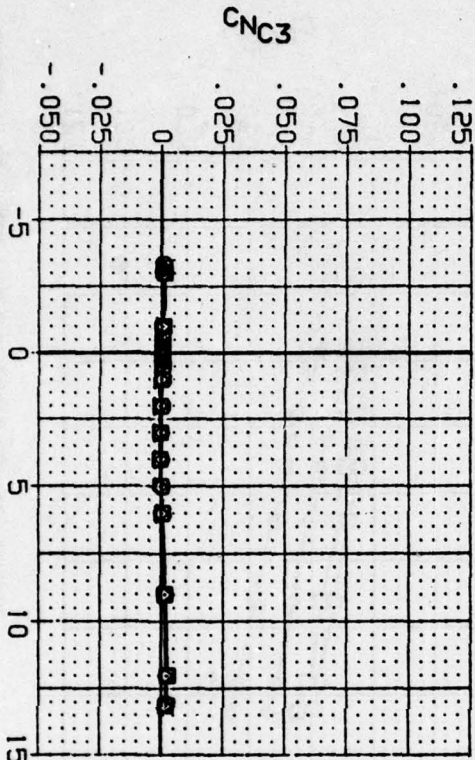
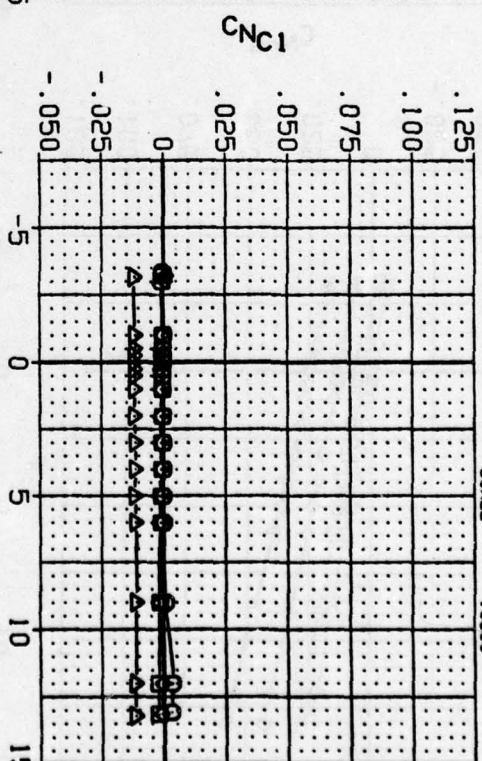
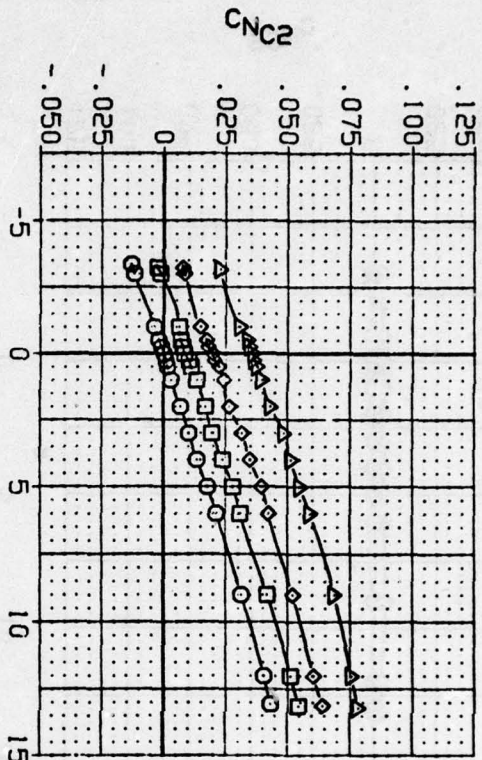
(AXH093) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH094) \diamond AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH095) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH096) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



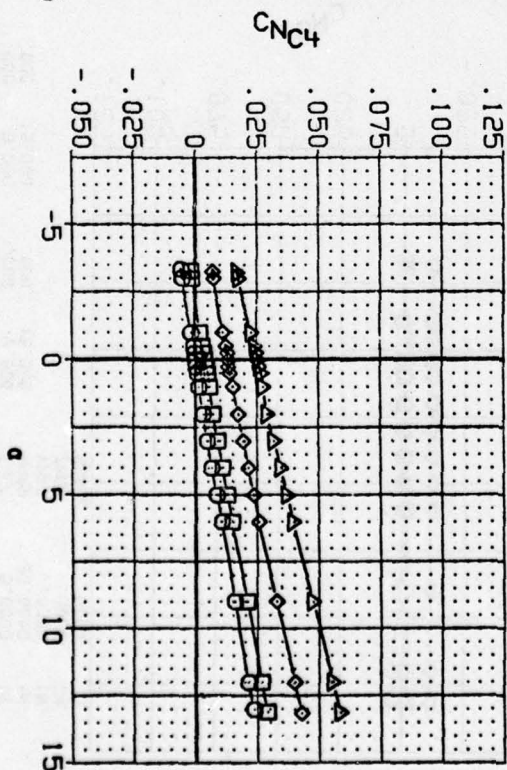
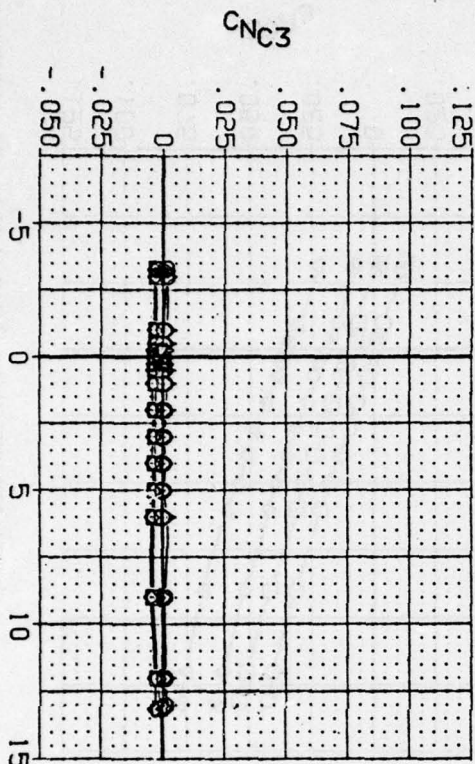
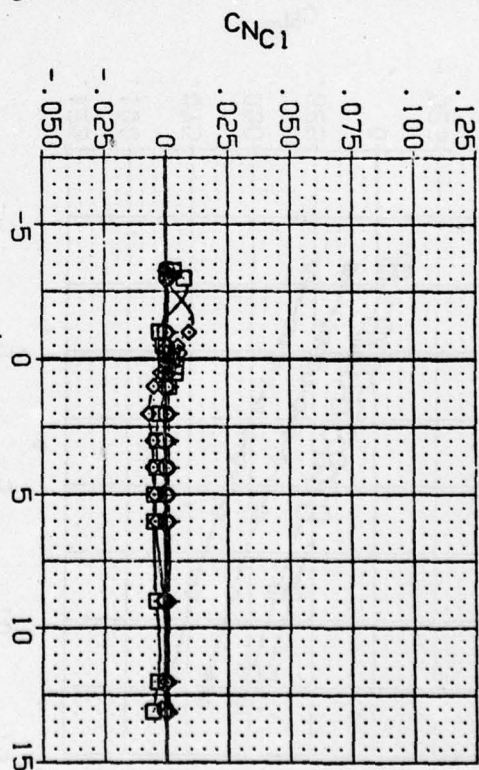
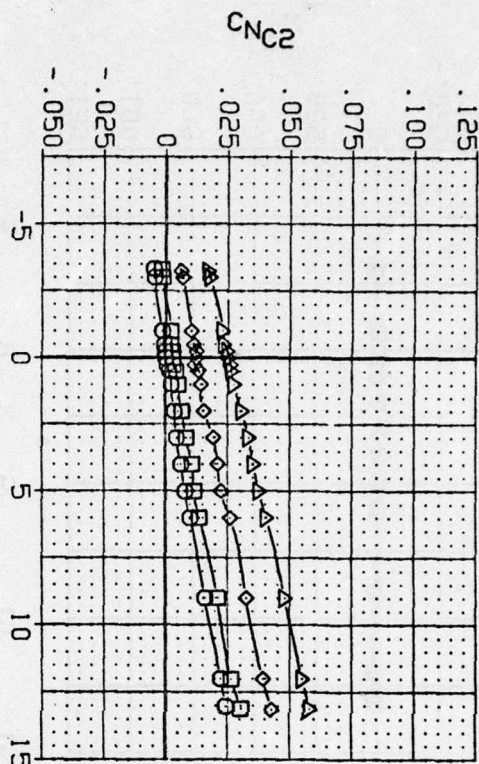
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHICAL=0
 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH093) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
 (AXH094) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
 (AXH095) \square AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
 (AXH096) \triangle AEDC V41A-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 3.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XHREF 26.0000 IN.
 YHREF .0000 IN.
 ZHREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHIAL=0 PHICND=0
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

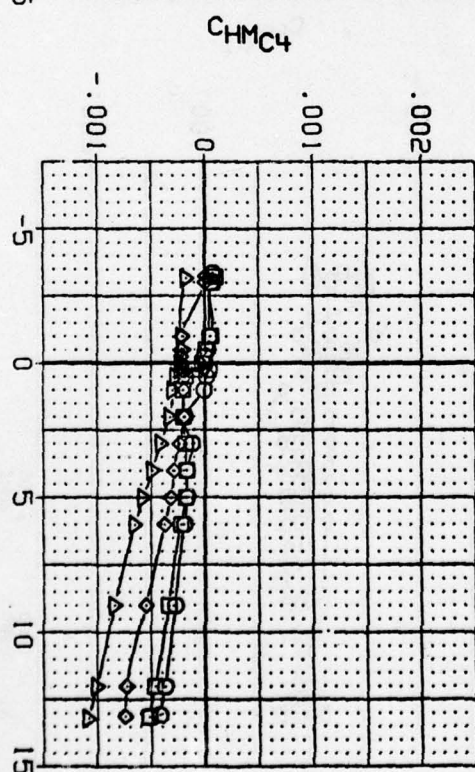
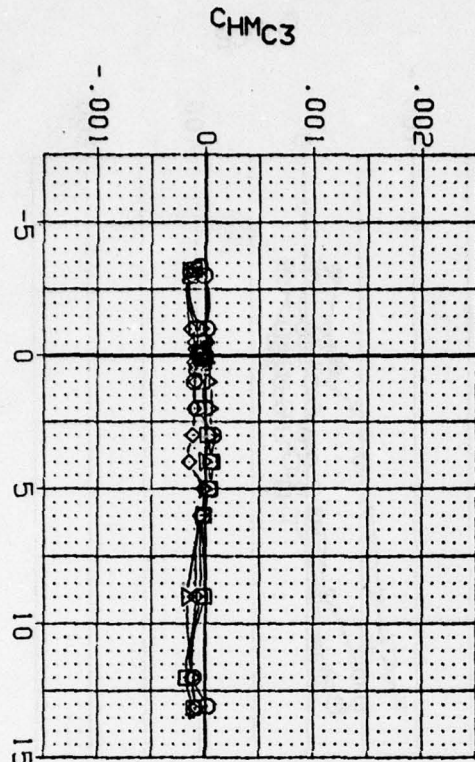
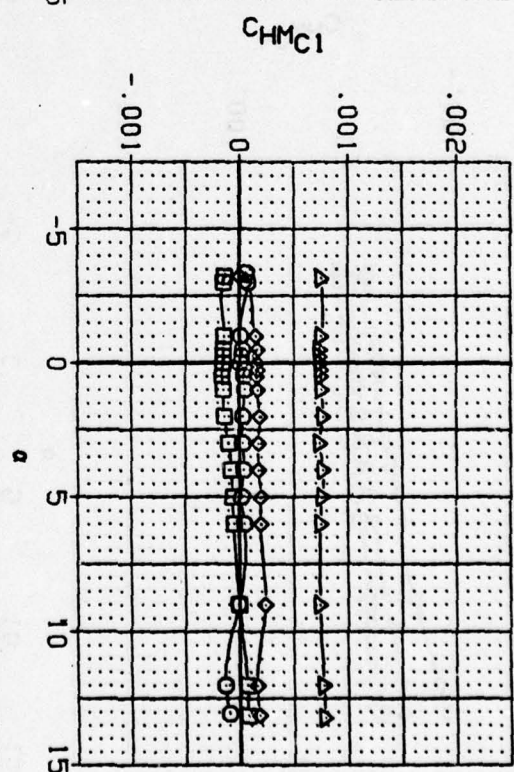
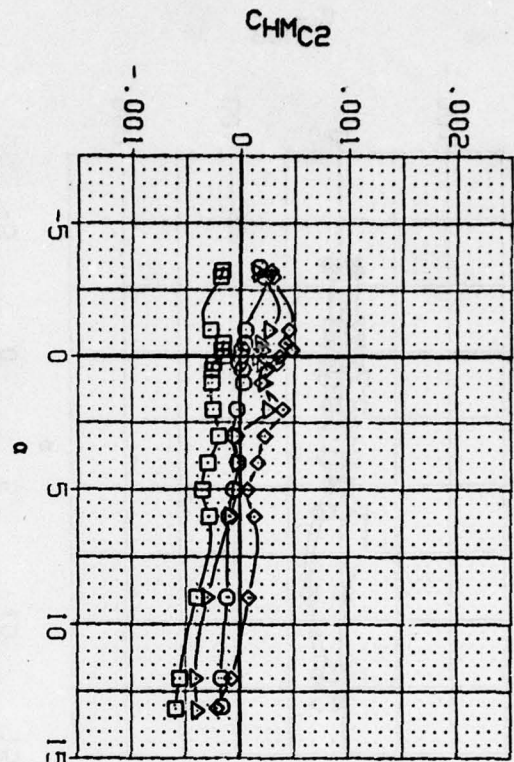
(BXH093) \square AEDC VMIA-CIA, CANARD CONTROL, BNIC1TR
 (BXH094) \square AEDC VMIA-CIA, CANARD CONTROL, BNIC1TR
 (BXH095) \square AEDC VMIA-CIA, CANARD CONTROL, BNIC1TR
 (BXH096) Δ AEDC VMIA-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XRRP 26.0000 IN.
 YRRP .0000 IN.
 ZRRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

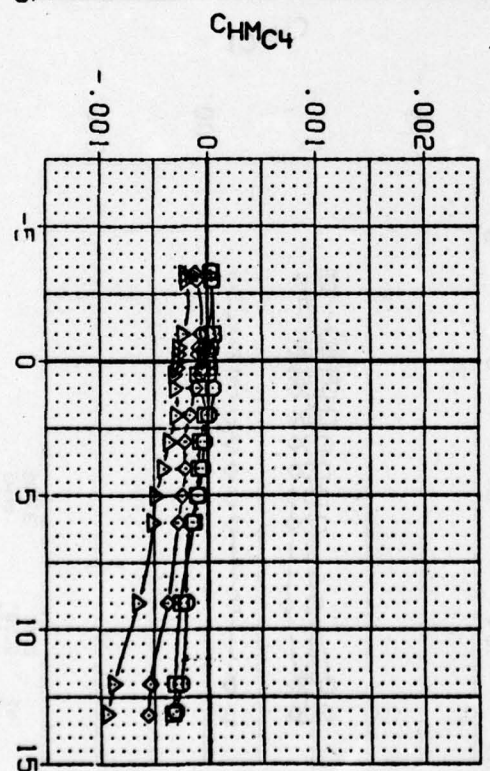
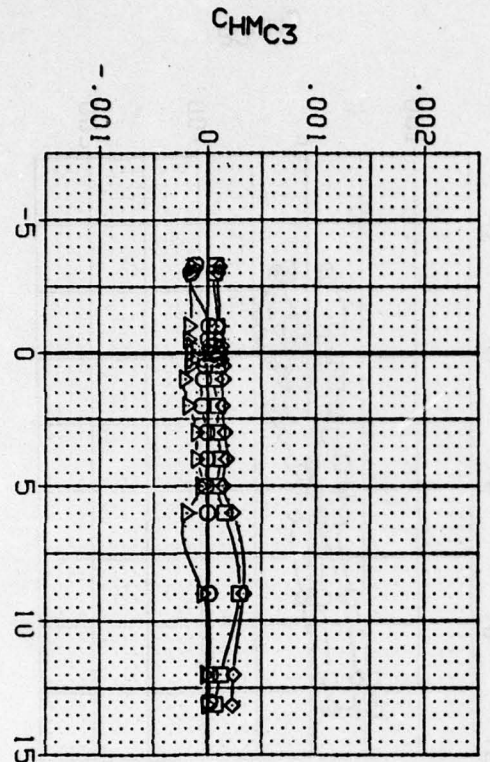
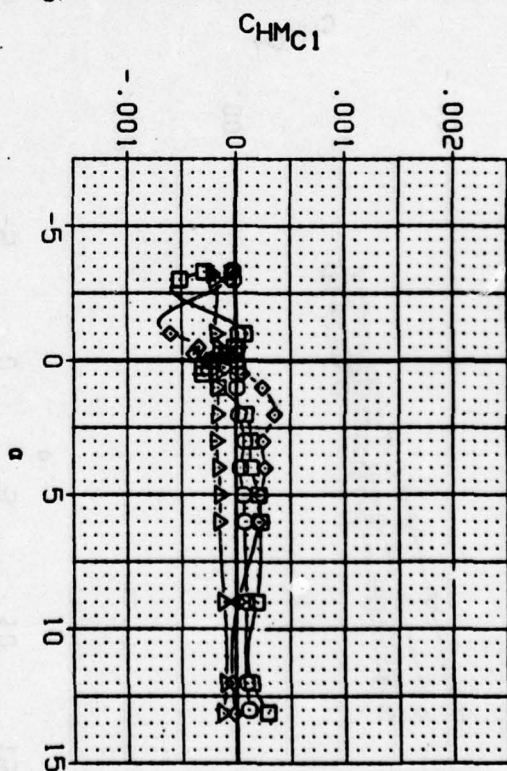
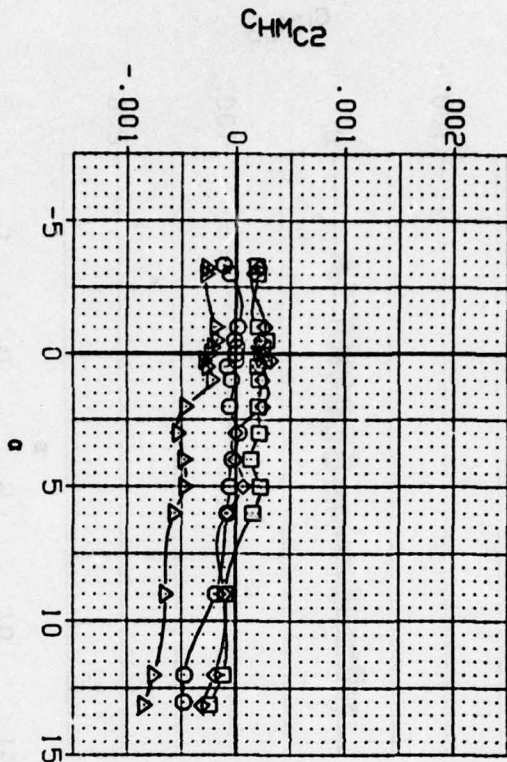
(BXH093) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH094) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH095) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH096) AEDC WIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

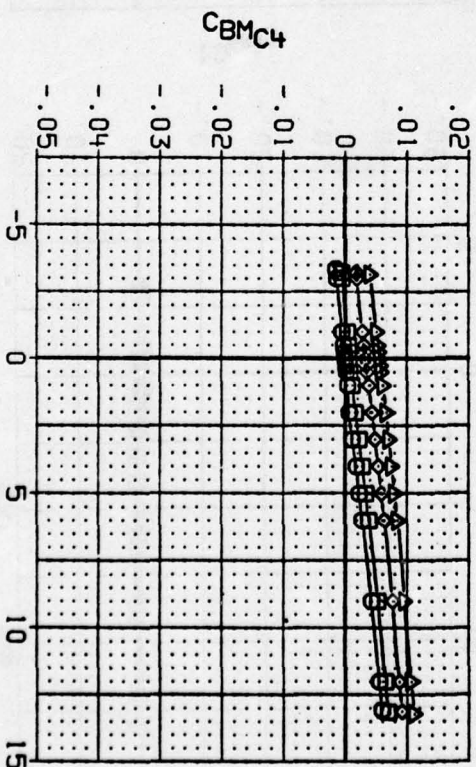
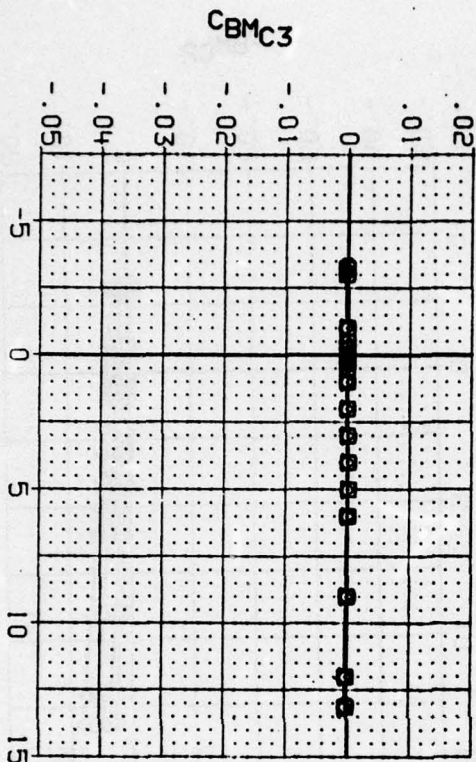
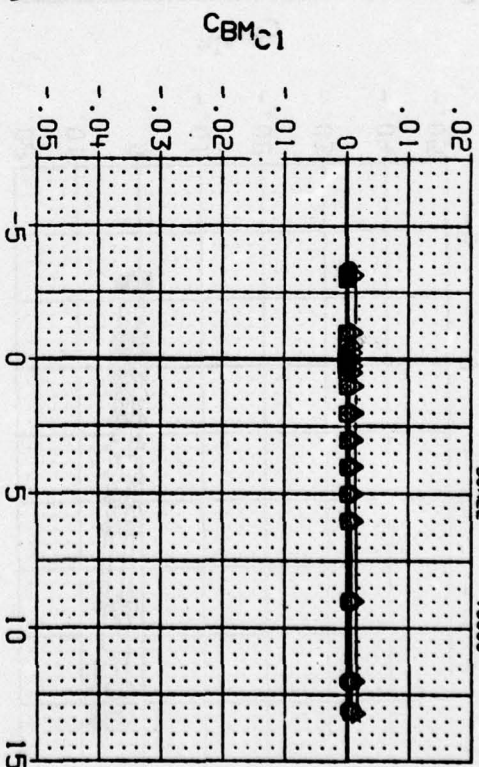
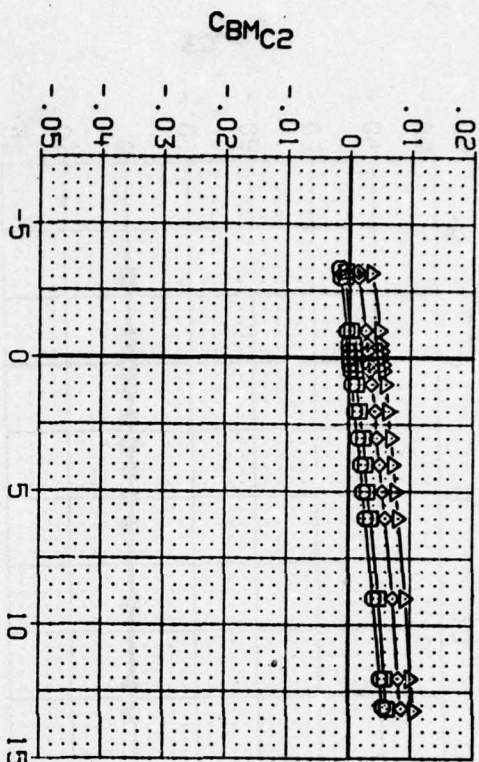
(BXH093) \square AEDC VAI-CIA, CANARD CONTROL, BNICITR
 (BXH094) \square AEDC VAI-CIA, CANARD CONTROL, BNICITR
 (BXH095) \square AEDC VAI-CIA, CANARD CONTROL, BNICITR
 (BXH096) Δ AEDC VAI-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

$\Phi_{HIAL}=0$ $\Phi_{HICND}=0$

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH093) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR

(BXH094) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR

(BXH095) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR

(BXH096) \triangle AEDC WJIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

.000 3.000 .000 3.000

.000 9.000 .000 9.000

.000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 50 IN.

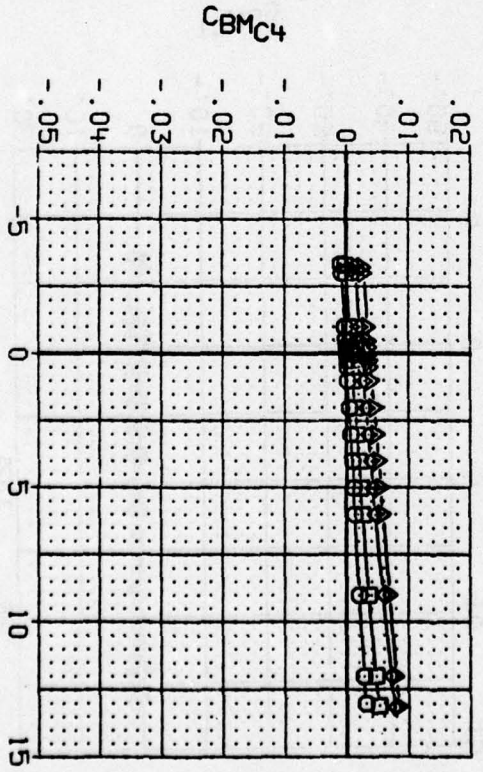
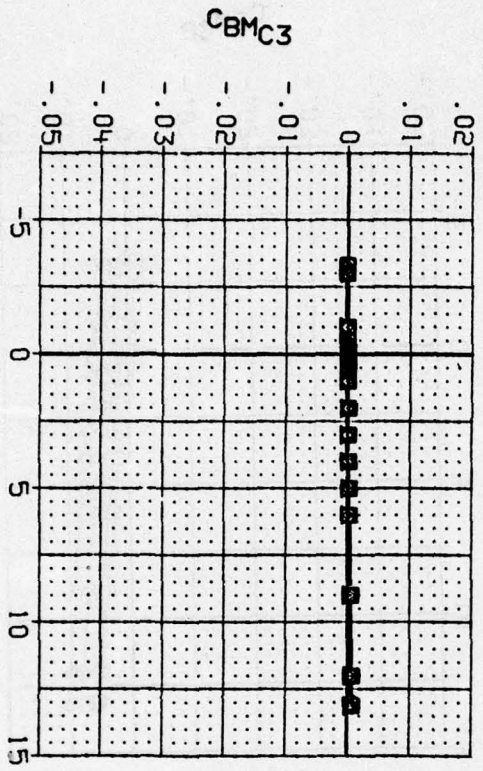
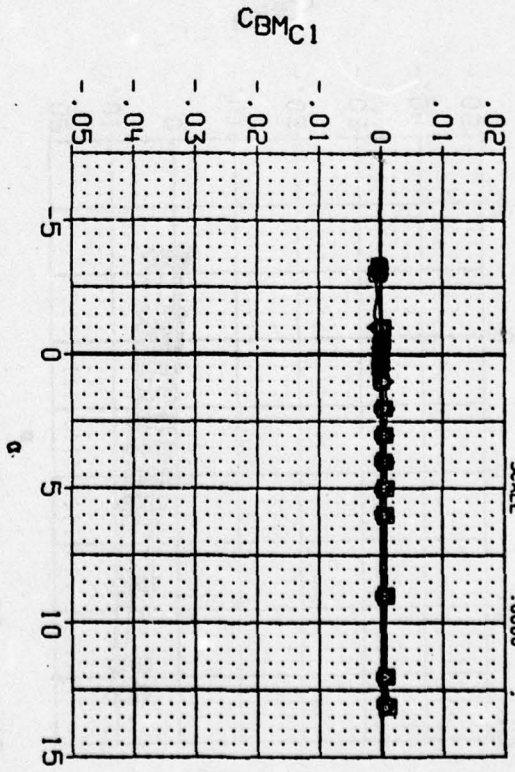
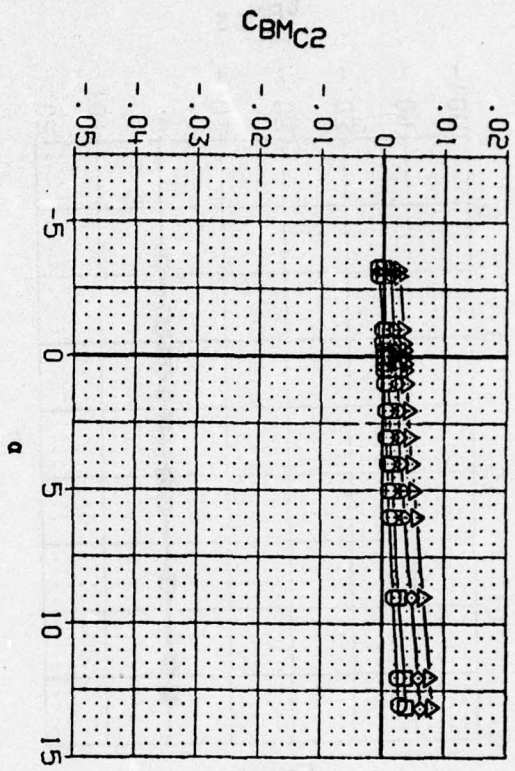
LREF 5.0000 IN.

BREF 5.0000 IN.

XREF 26.0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

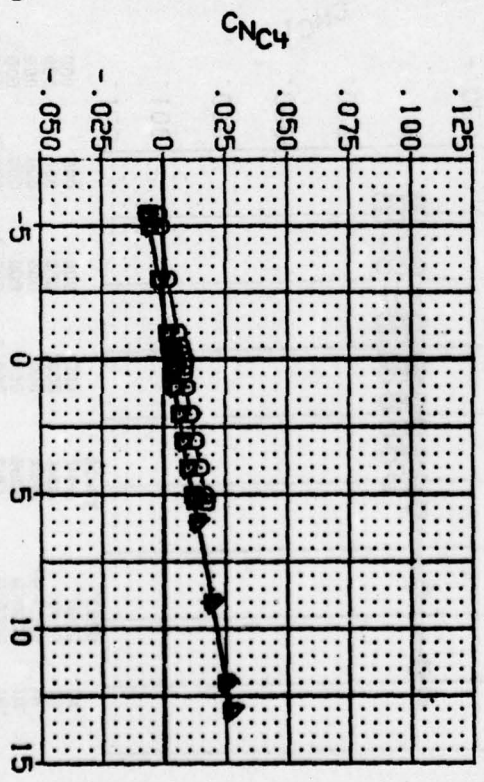
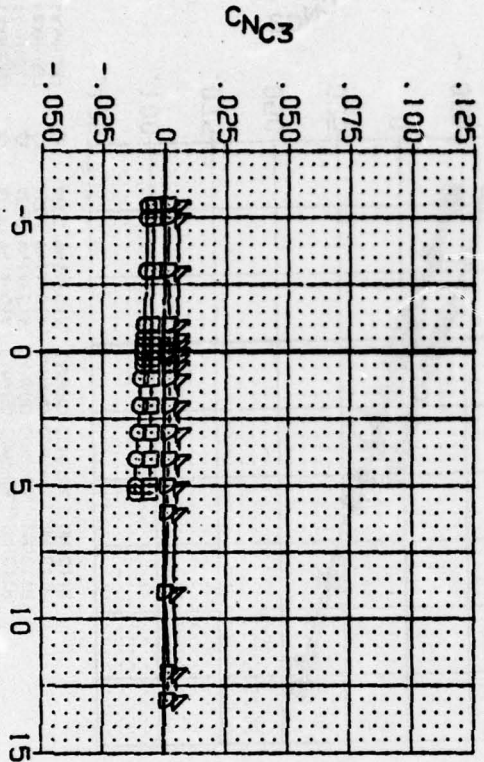
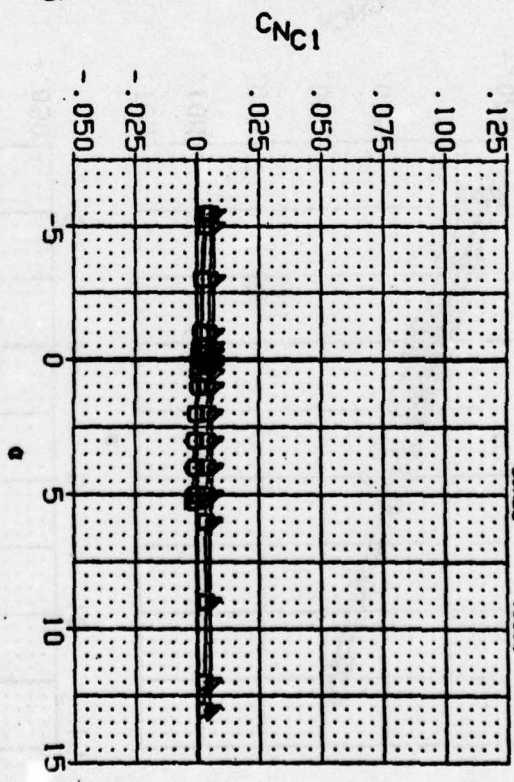
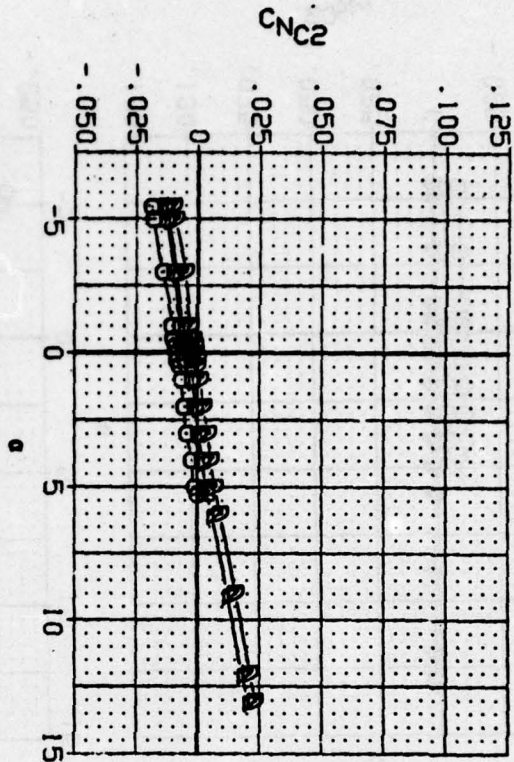
(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH057) AEDC WIA-CIA, CANARD CONTROL, BNIC1TR
 (AXH058) AEDC WIA-CIA, CANARD CONTROL, BNIC1TR
 (AXH059) AEDC WIA-CIA, CANARD CONTROL, BNIC1TR
 (AXH101) DATA NOT AVAILABLE
 (AXH102) AEDC WIA-CIA, CANARD CONTROL, BNIC1TR
 (AXH103) AEDC WIA-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4 REFERENCE INFORMATION

5.030 -5.000 -5.000 5.000 SREF 19.6350 SQ. IN.
 2.925 -2.000 -2.000 2.000 LREF 5.0000 IN.
 .000 .000 .000 .000 BREF 5.0000 IN.
 .000 .000 .000 .000 XREF 26.0000 IN.
 .000 .000 .000 .000 YREF .0000 IN.
 .000 .000 .000 .000 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

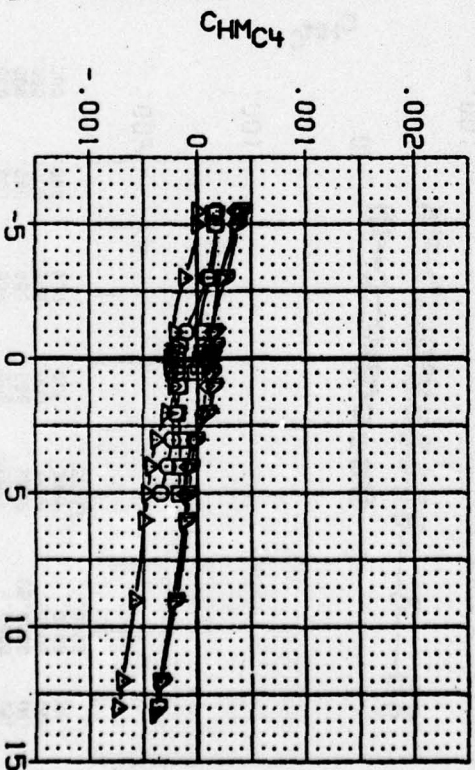
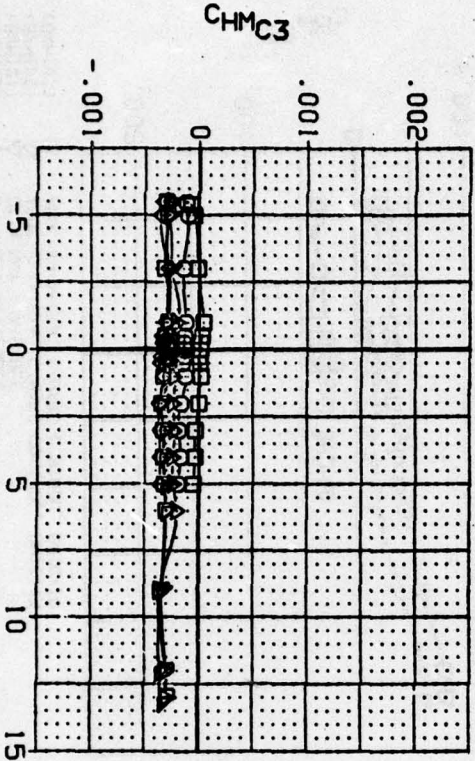
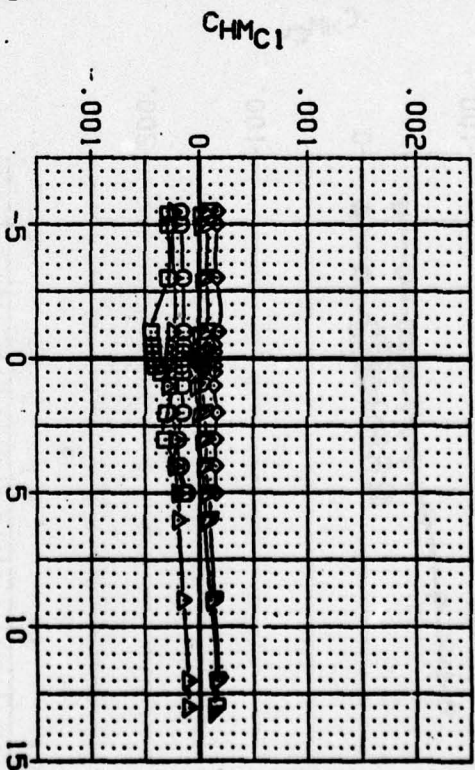
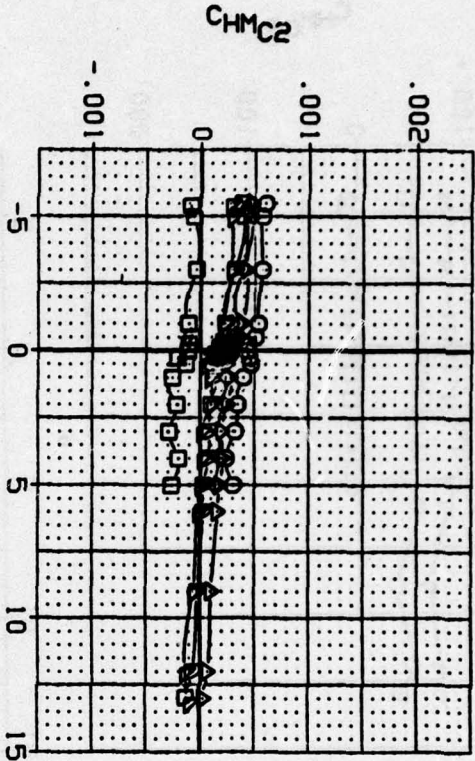
(BXH097) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH098) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH099) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH101) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH102) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BXH103) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

5.000 2.000 -5.000 -5.000 5.000
 2.000 -2.000 -2.000 2.000 2.000
 .000 .000 .000 .000 .000
 .000 .000 .000 .000 .000
 .000 .000 .000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHIAL=0 PHICND=0

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH097) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR

(BXH098) \diamond AEDC WIA-CIA, CANARD CONTROL, BNICITR

(BXH099) \triangle DATA NOT AVAILABLE

(BXH101) \square AEDC WIA-CIA, CANARD CONTROL, BNICITR

(BXH102) \diamond AEDC WIA-CIA, CANARD CONTROL, BNICITR

(BXH103) \triangle AEDC WIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

5.000 2.000 -5.000 -2.000 -5.000 2.000

.000 .000 .000 .000 .000 .000

.000 .000 .000 .000 .000 .000

.000 .000 .000 .000 .000 .000

SCALE

19.6350 50. IN.

5.0000 IN.

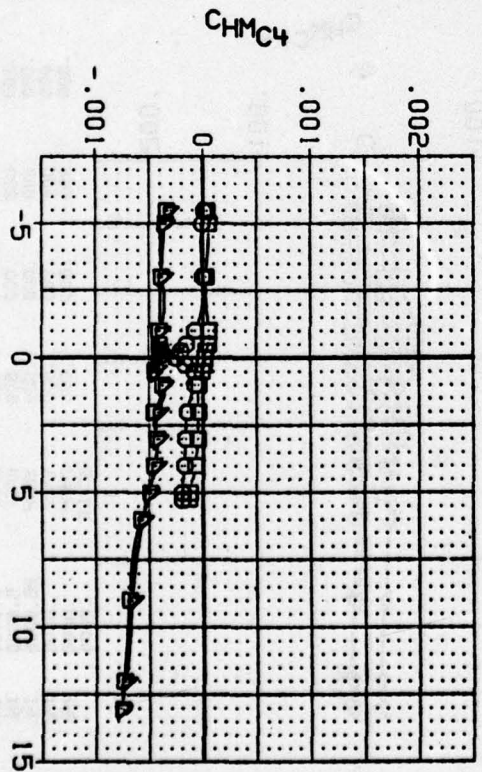
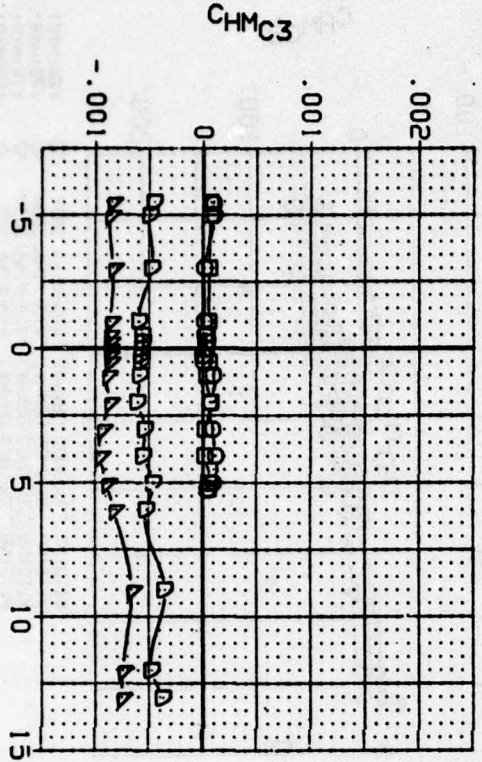
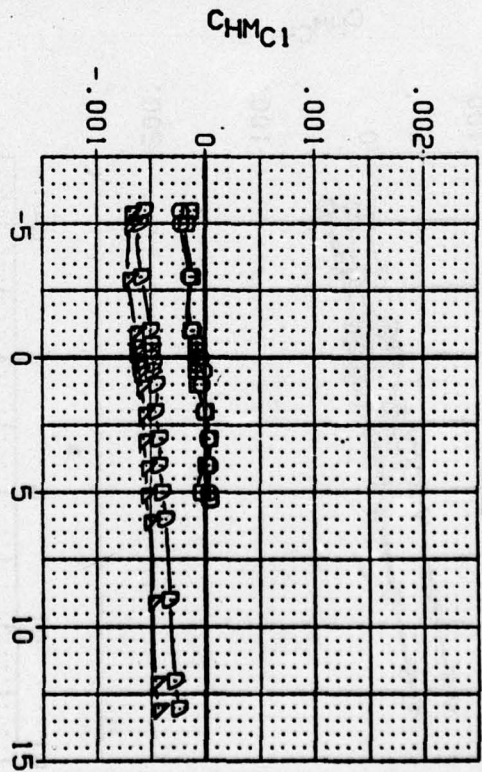
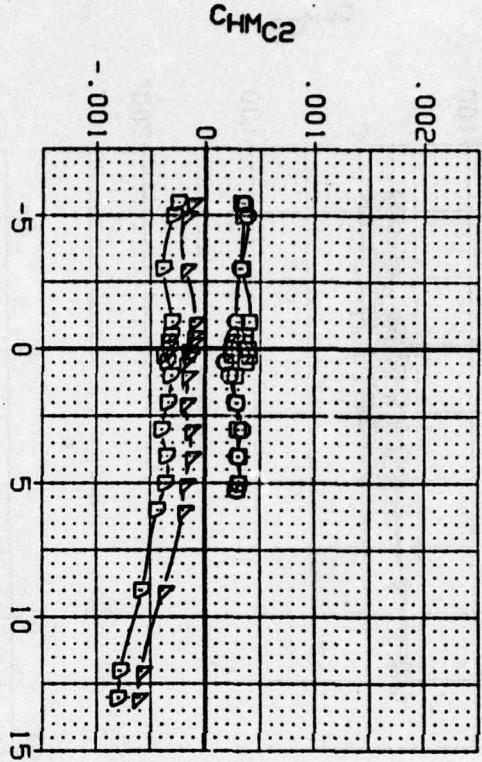
5.0000 IN.

25.0000 IN.

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.0000 IN.

.0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=0

(B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

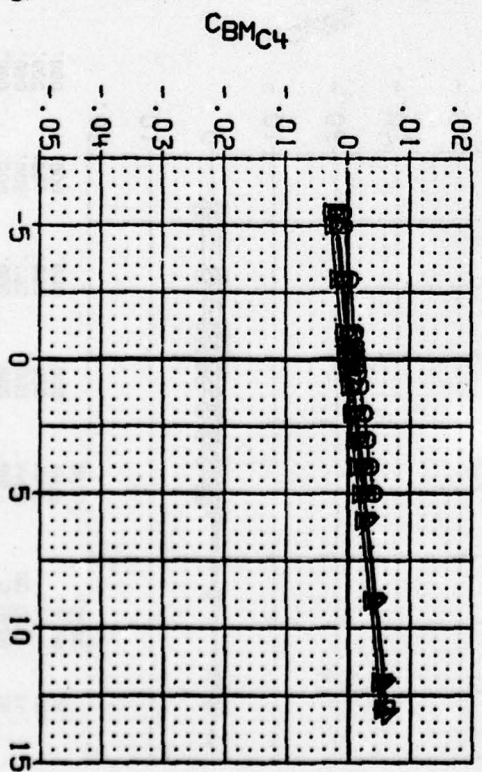
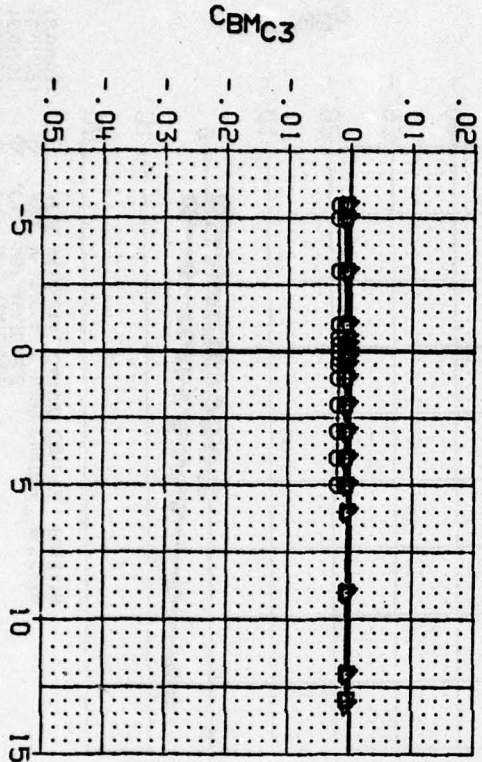
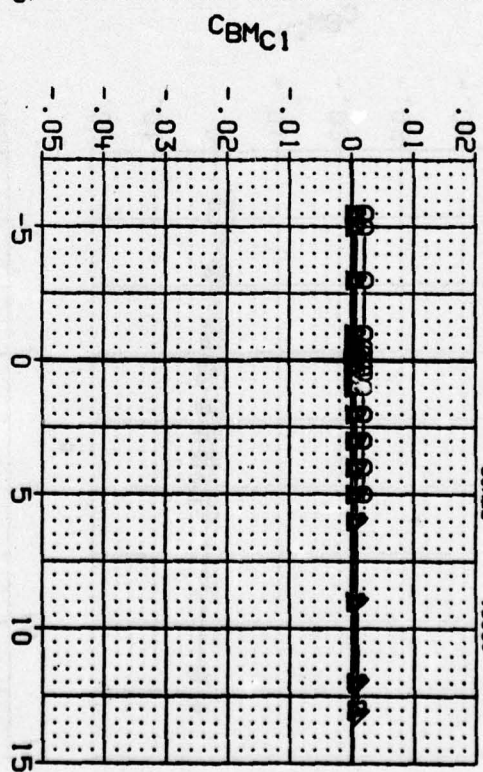
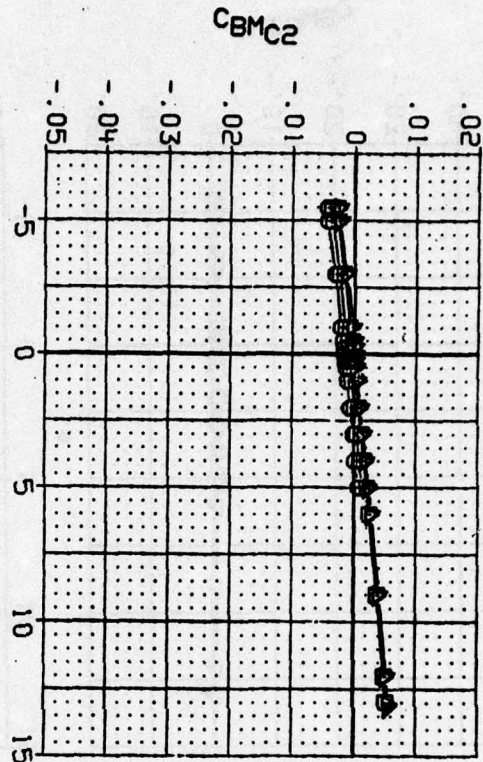
(BMH097) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BMH098) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BMH099) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BMH101) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BMH102) AEDC WIA-CIA, CANARD CONTROL, BNICITR
 (BMH103) AEDC WIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4

5.000 -5.000 -5.000 5.000
 2.000 -2.000 -2.000 2.000
 .000 .000 .000 .000
 .000 .000 .000 .000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

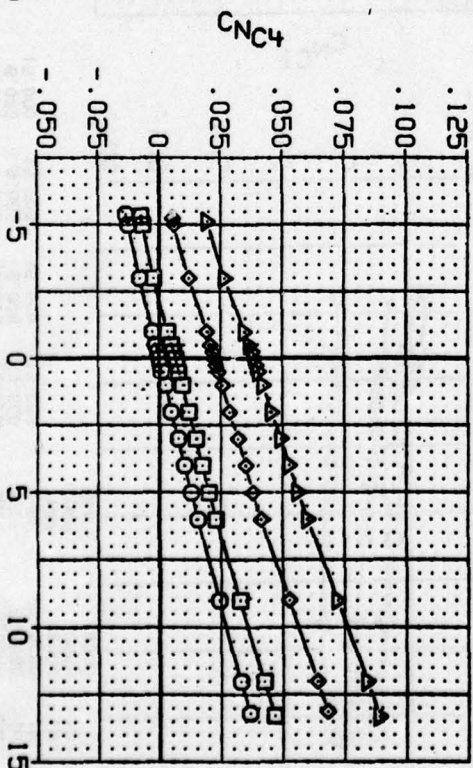
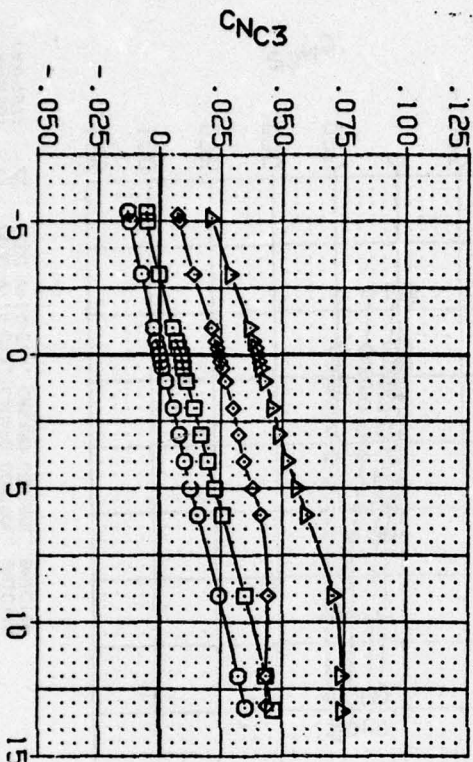
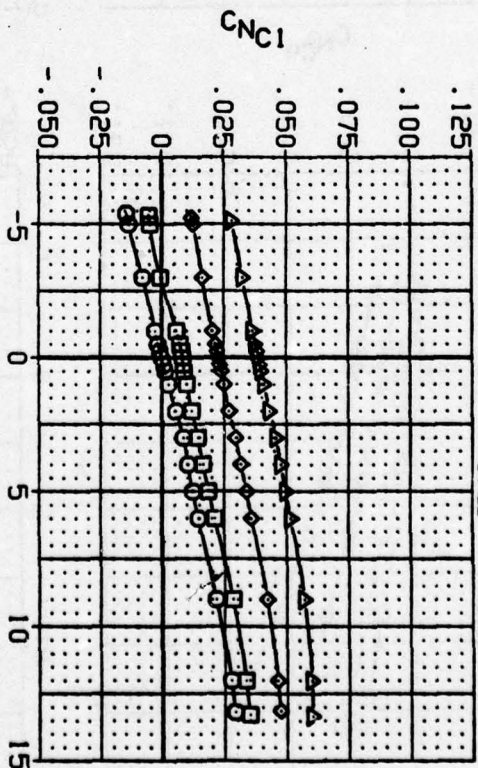
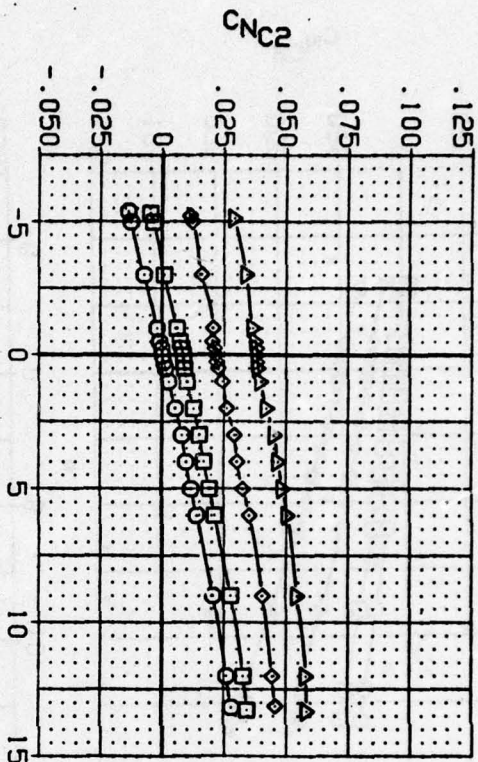
PHITAL=0 PHICND=0

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH104) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR
 (AXH105) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR
 (AXH106) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR
 (AXH107) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 $\Phi_{H1A} = 0$
 $\Phi_{H1CND} = 45$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

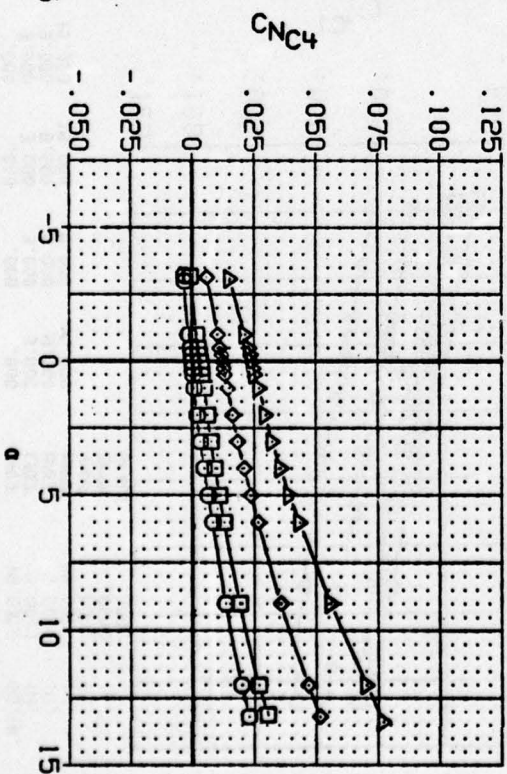
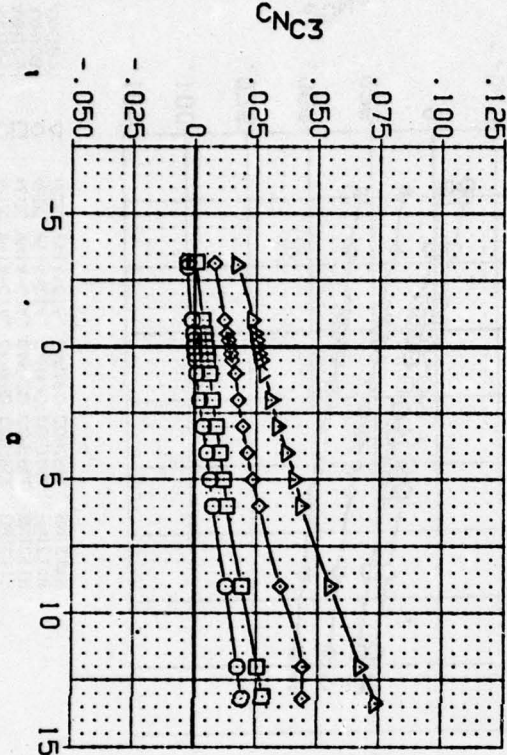
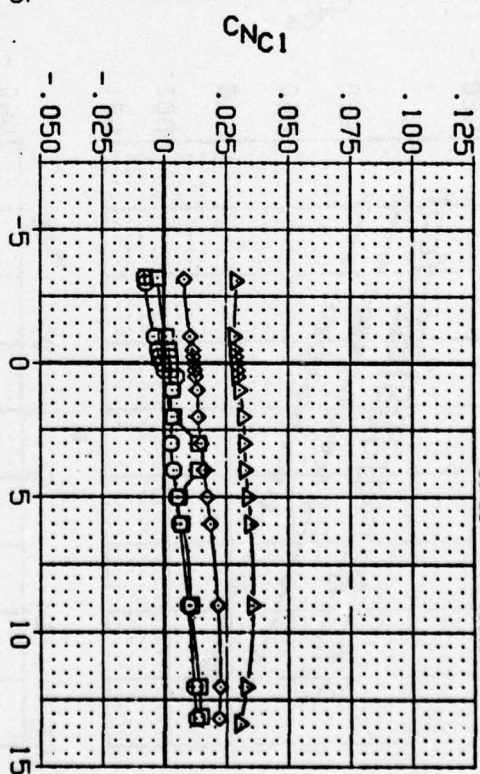
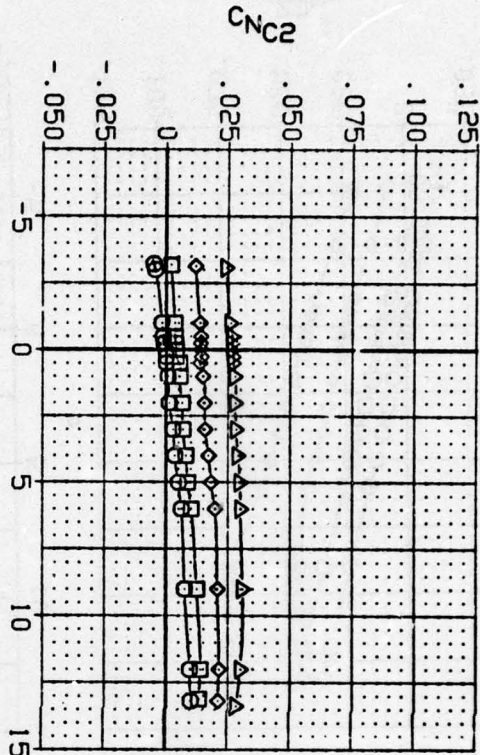
(AXH104) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH105) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH106) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (AXH107) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PH1AL=0 PHICND=45

(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

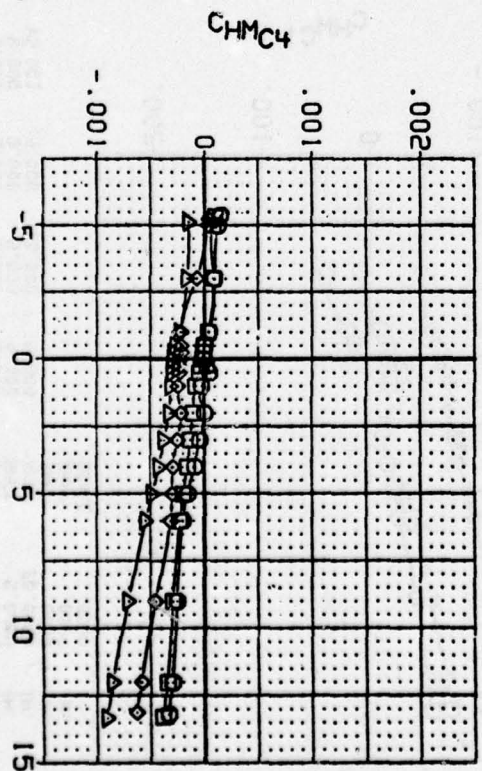
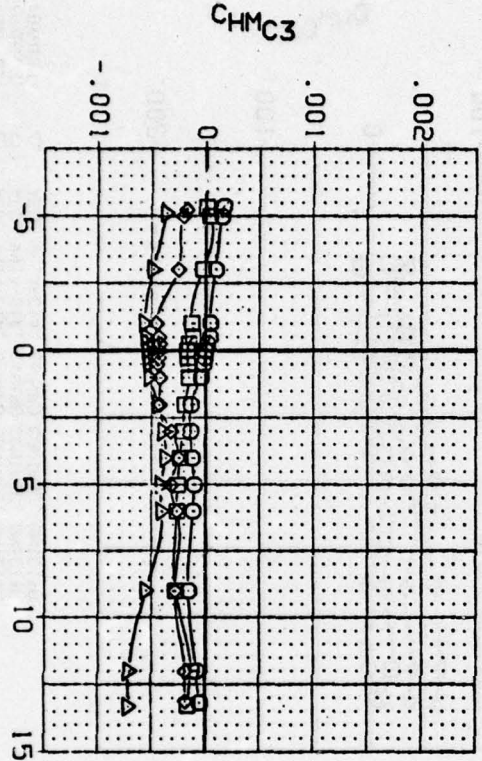
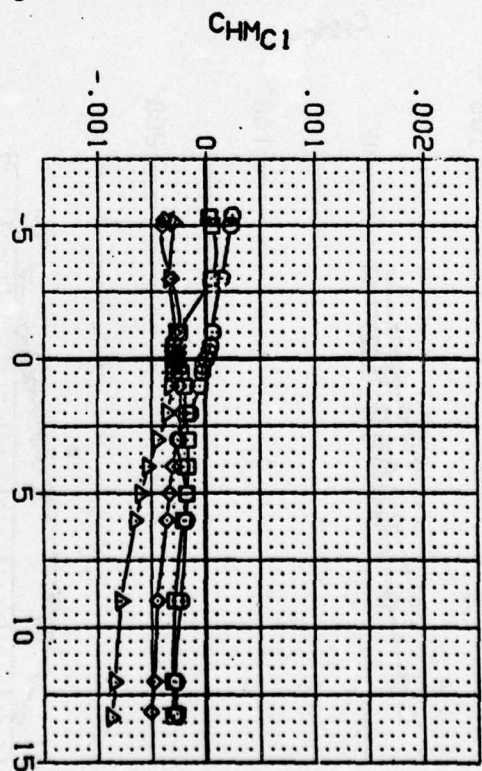
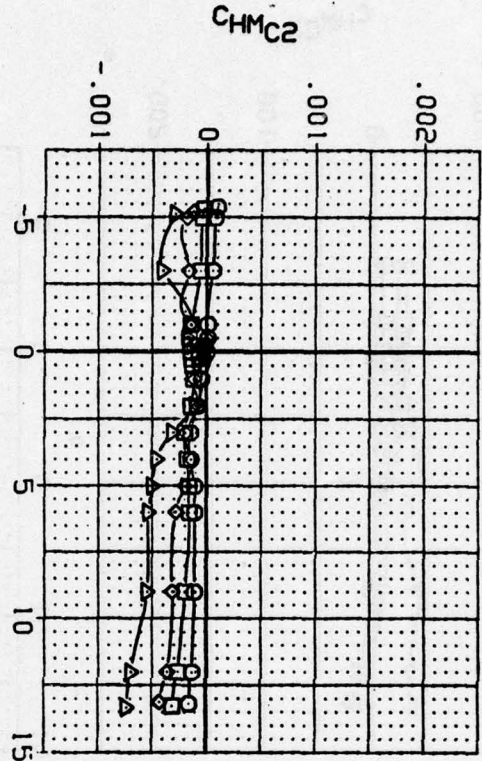
(BXH104) AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (BXH105) AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (BXH106) AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (BXH107) AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

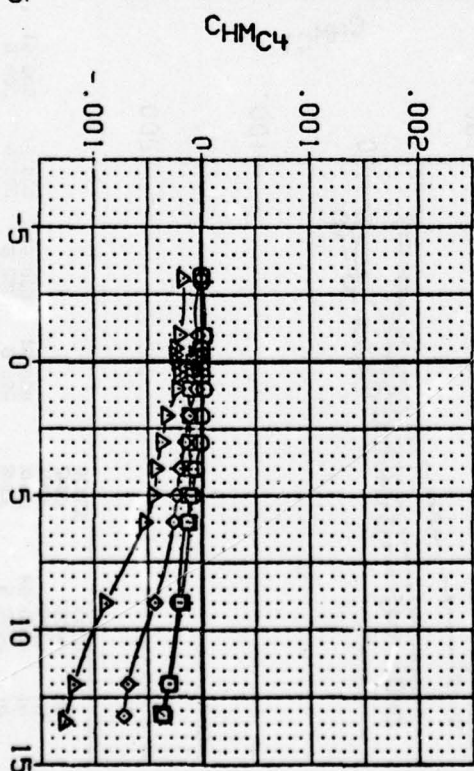
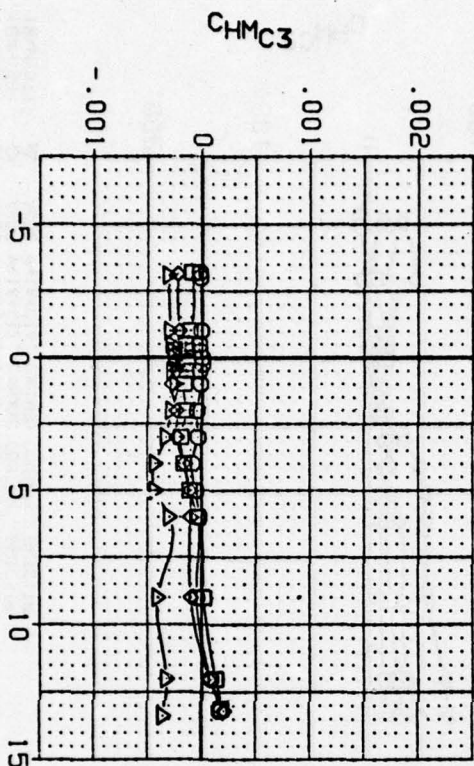
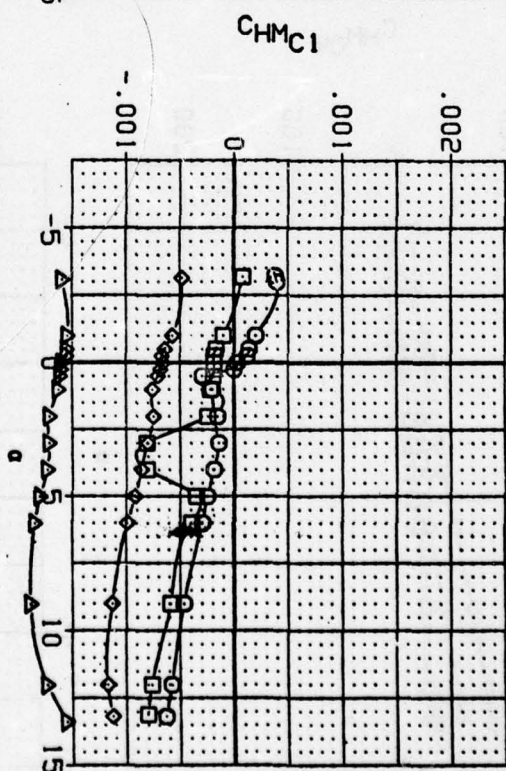
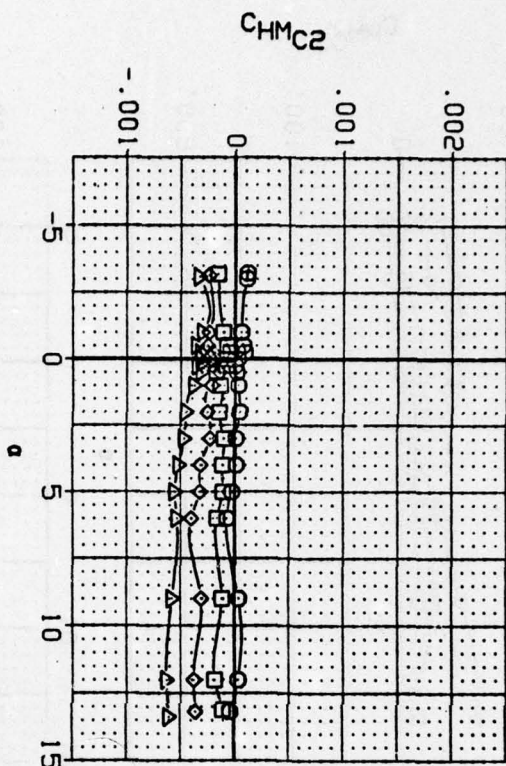
PH1AL=0 PH1CND=45

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (BXH104) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (BXH105) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (BXH105) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (BXH107) \triangle AEDC W1A-C1A, CANARD CONTROL, BNIC1TR

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD
 PHITAL=0 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

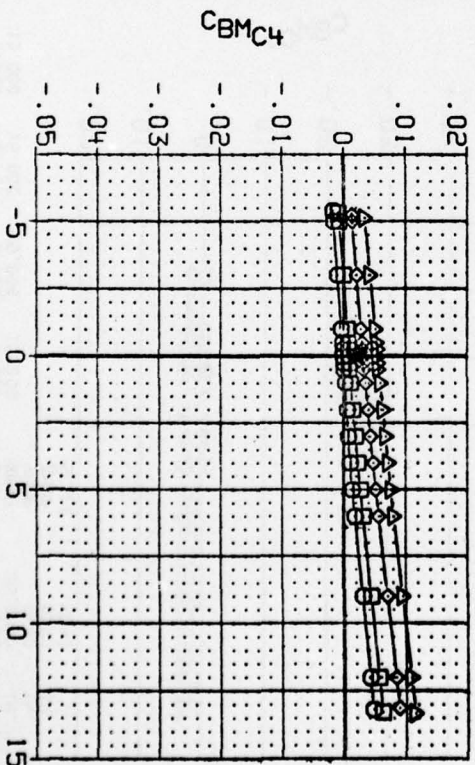
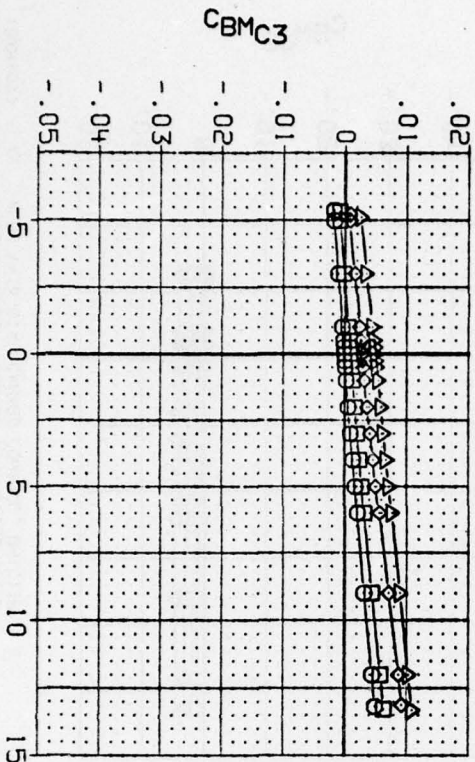
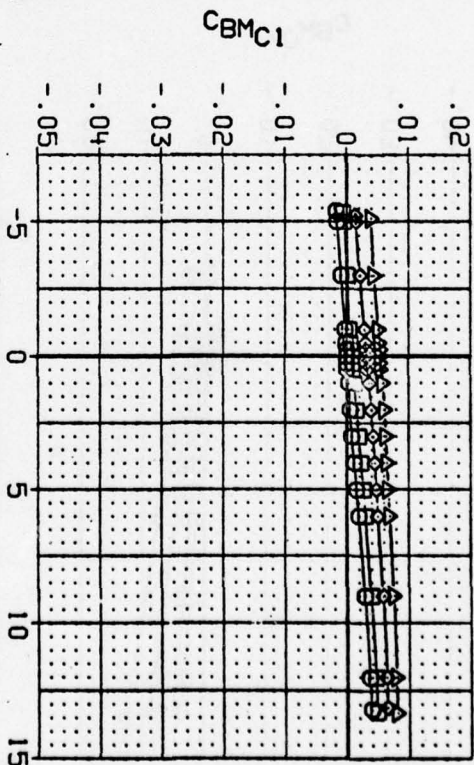
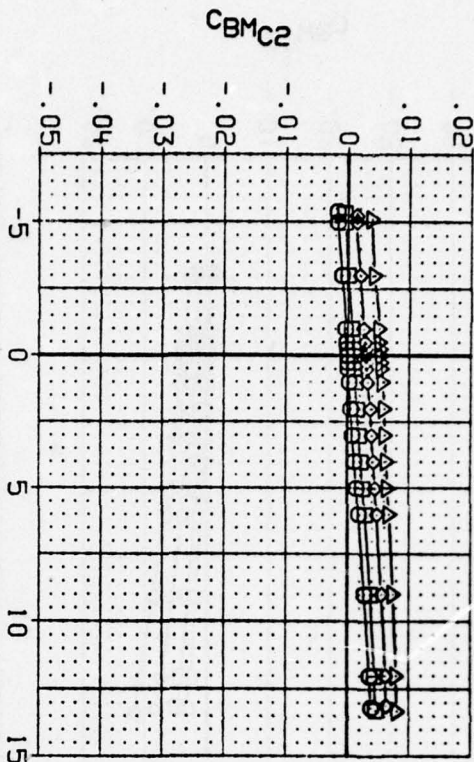
(BXH104) \square AEDC VAI-A-CIA, CANARD CONTROL, BNIC1TR
 (BXH105) \square AEDC VAI-A-CIA, CANARD CONTROL, BNIC1TR
 (BXH106) \square AEDC VAI-A-CIA, CANARD CONTROL, BNIC1TR
 (BXH107) \triangle AEDC VAI-A-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.5350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

PHITAL=0 PHICND=45

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

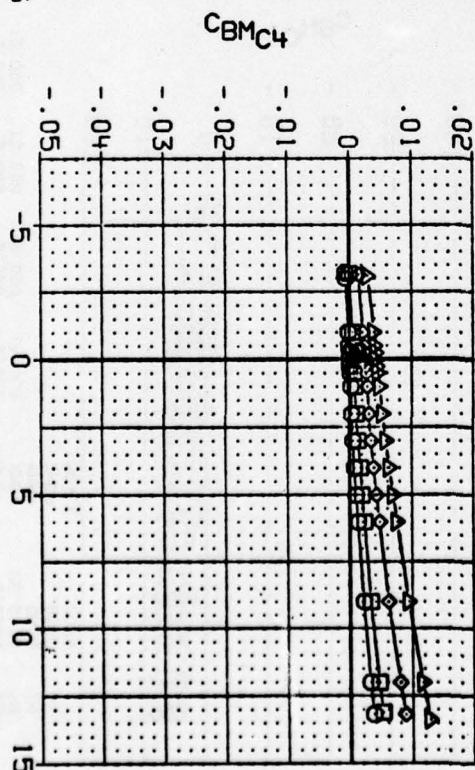
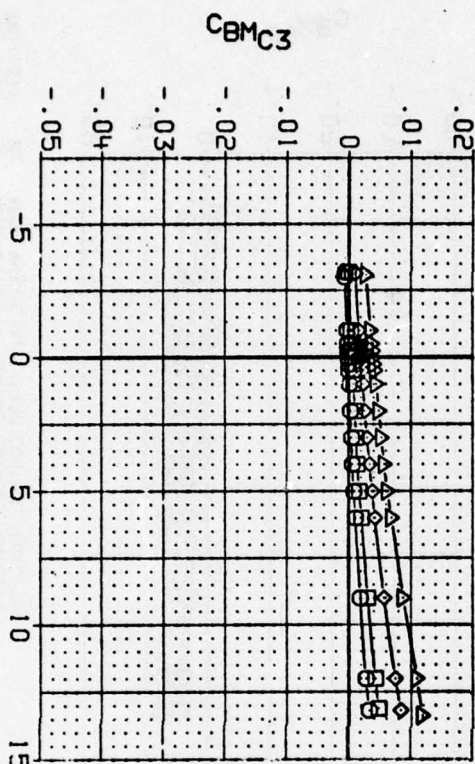
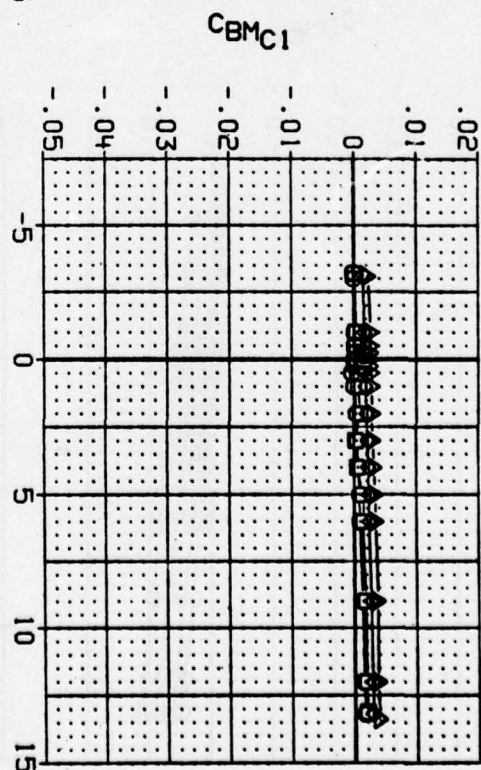
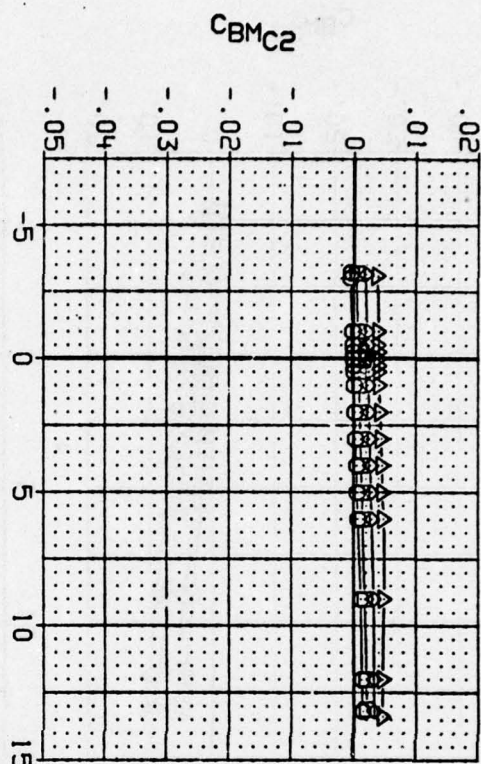
(BXH104)	○	AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
(BXH105)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
(BXH106)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
(BXH107)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000	.000	.000	.000
3.000	3.000	3.000	3.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION

SREF	19.6350	SO. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XTRP	26.0000	IN.
YTRP	.0000	IN.
ZTRP	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON CANARD

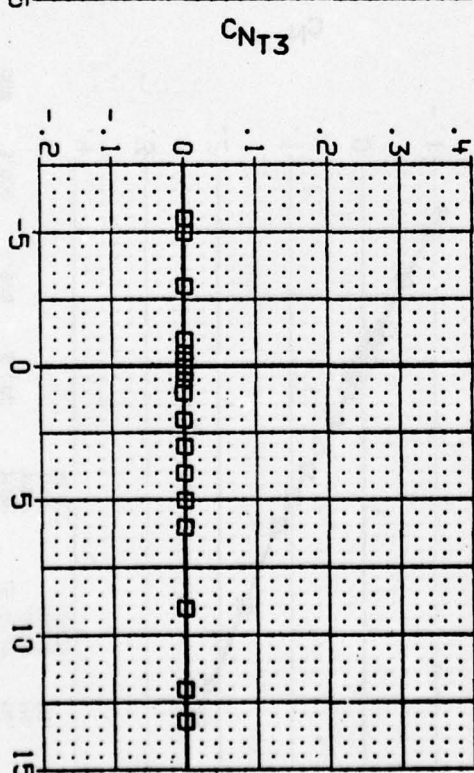
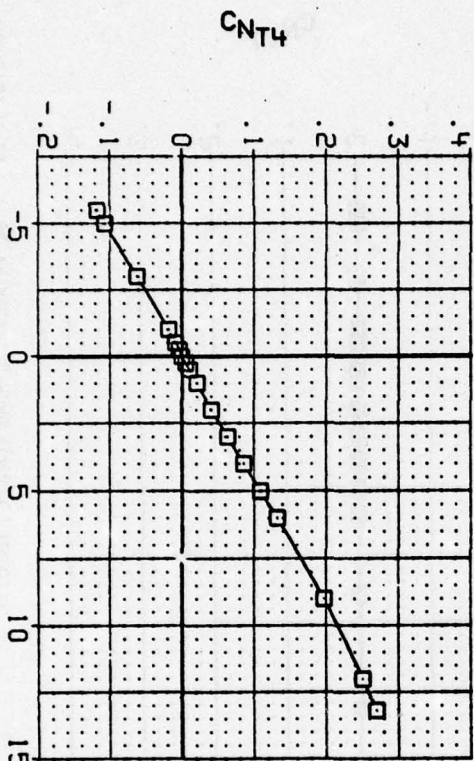
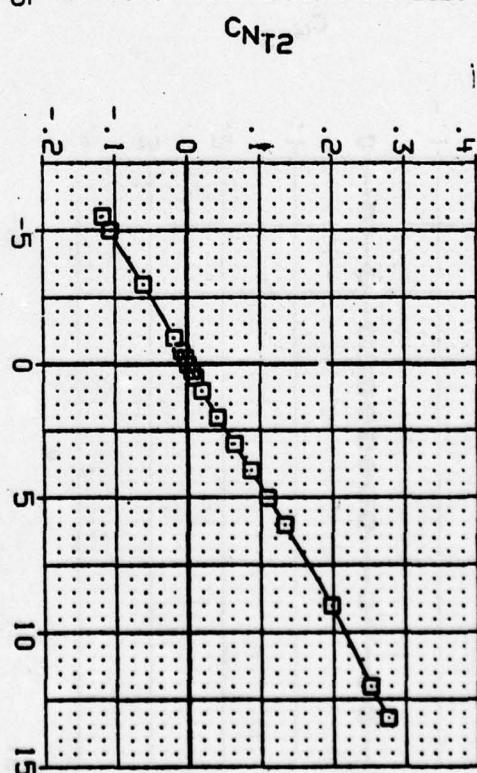
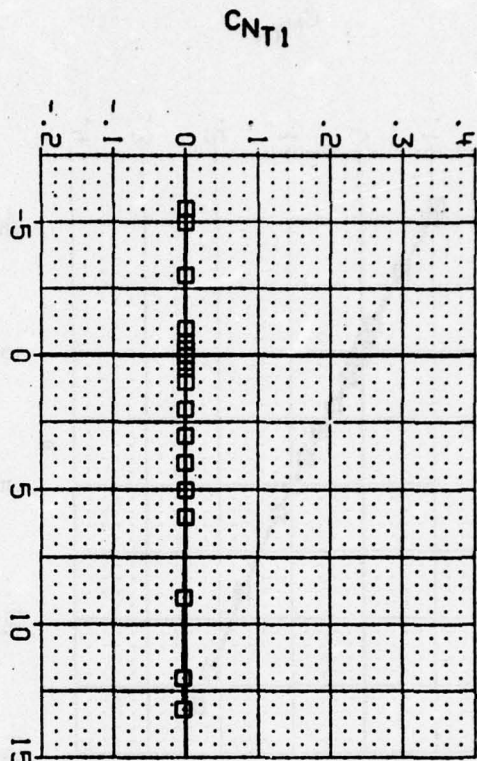
PHITAL=0 PHICND=45

(B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH016) DATA NOT AVAILABLE
 (AXH017) AEDC W/IA-CIA, CANARD CONTROL, BNICITI
 (AXH018) DATA NOT AVAILABLE
 (AXH019) DATA NOT AVAILABLE

DCND1 .000 DCND2 -3.000 DCND3 .000 DCND4 -3.000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

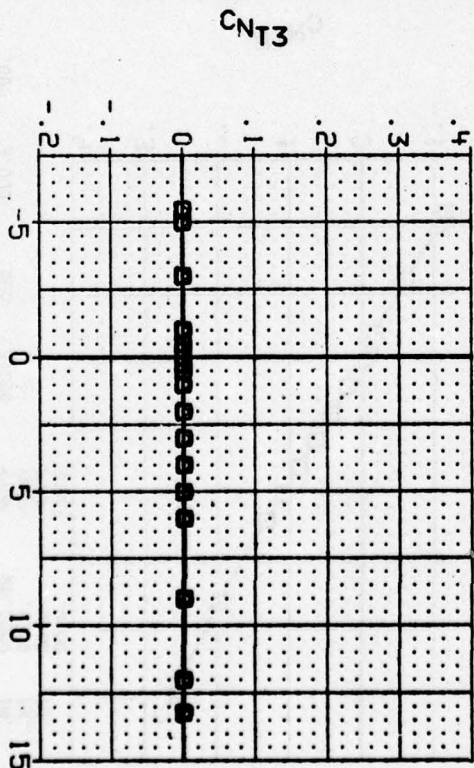
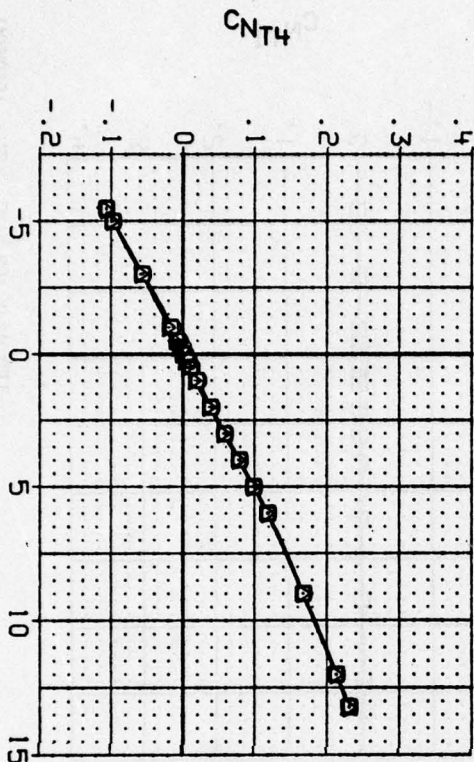
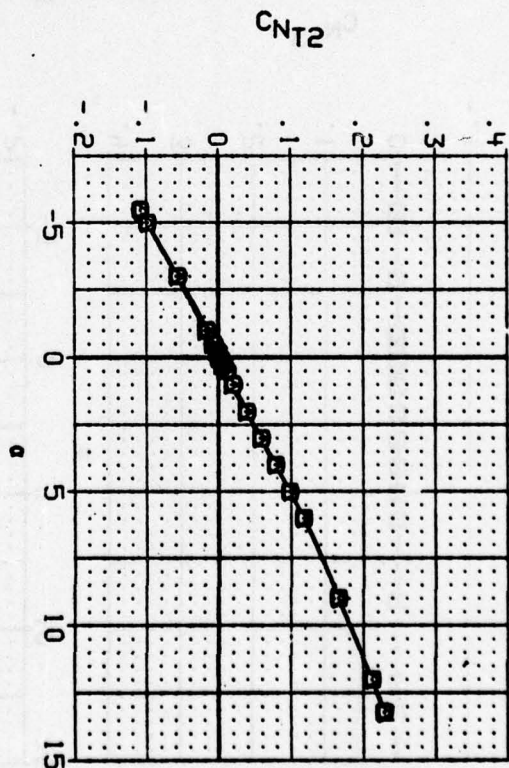
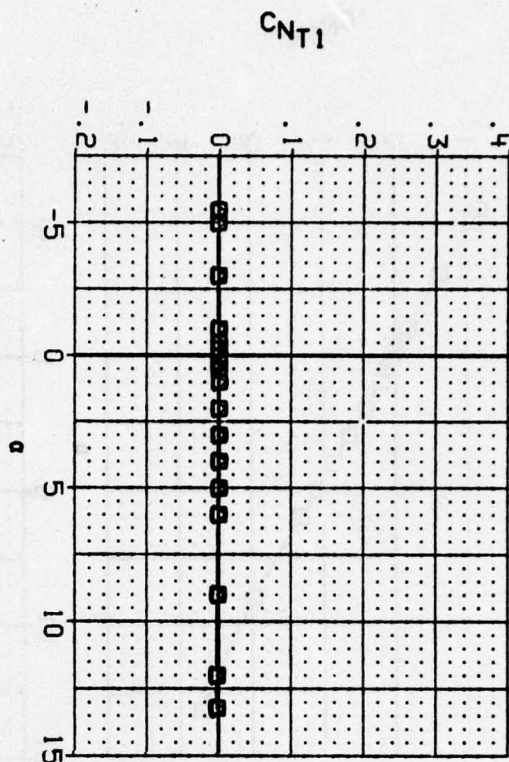


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITL=0$ $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH016) \square AEDC VJIA-CIA, CANARD CONTROL, BNICITI
 (AXH017) \square AEDC VJIA-CIA, CANARD CONTROL, BNICITI
 (AXH018) \square DATA NOT AVAILABLE
 (AXH019) Δ AEDC VJIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

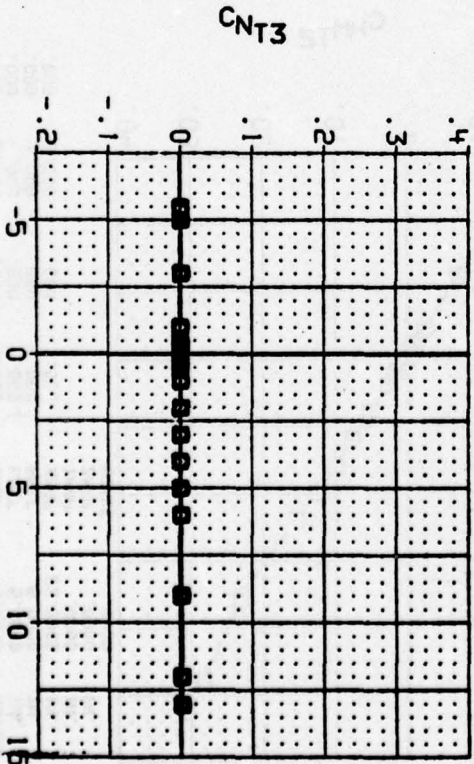
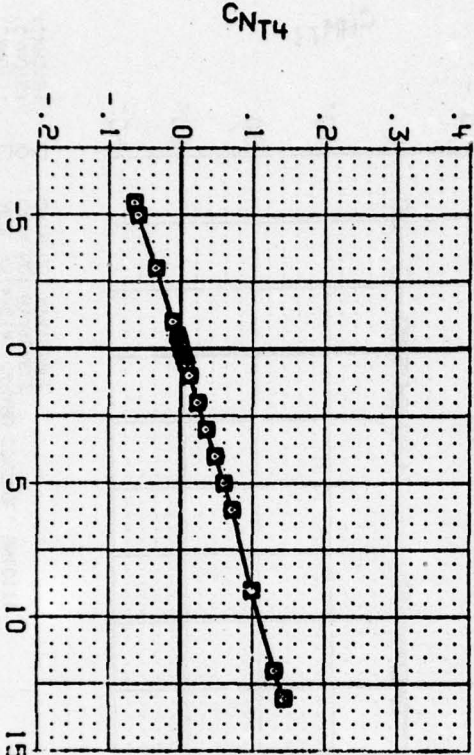
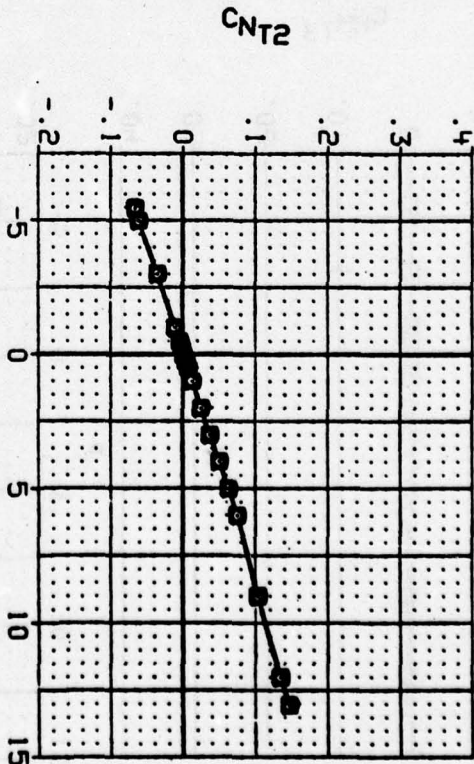
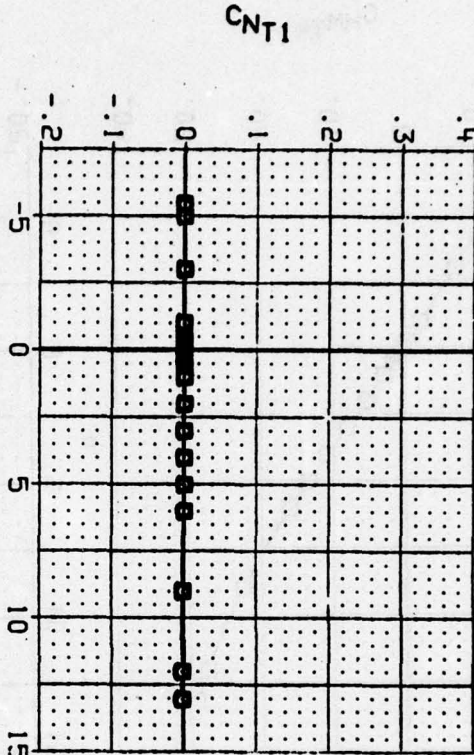


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBO. CONFIGURATION DESCRIPTION
 (AXH016) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (AXH017) \square AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (AXH018) \triangle AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (AXH019) \triangle AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 .000 DCND2 -3.000 DCND3 .000 DCND4 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 3.000
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YREF 26.0000 IN.
 ZREF .0000 IN.
 SCALE .0000 IN.

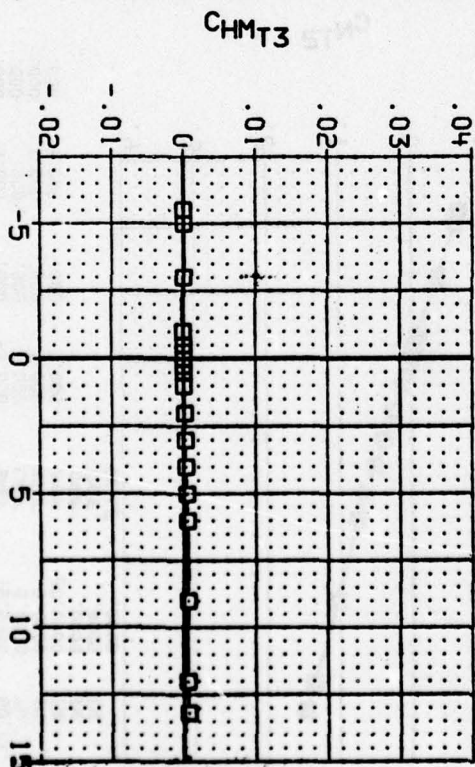
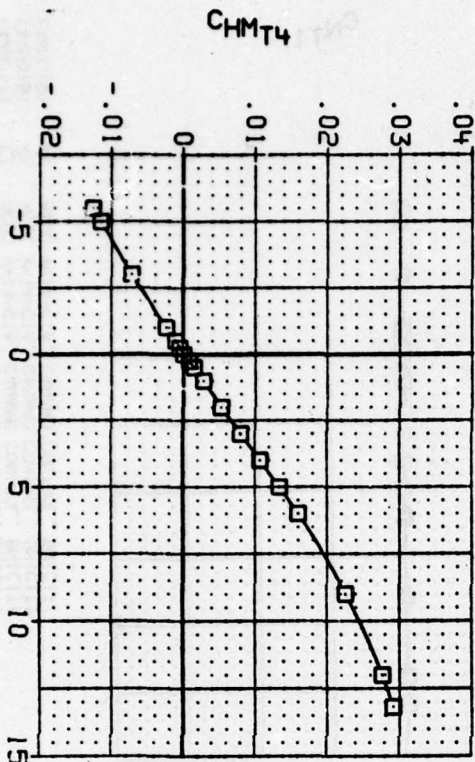
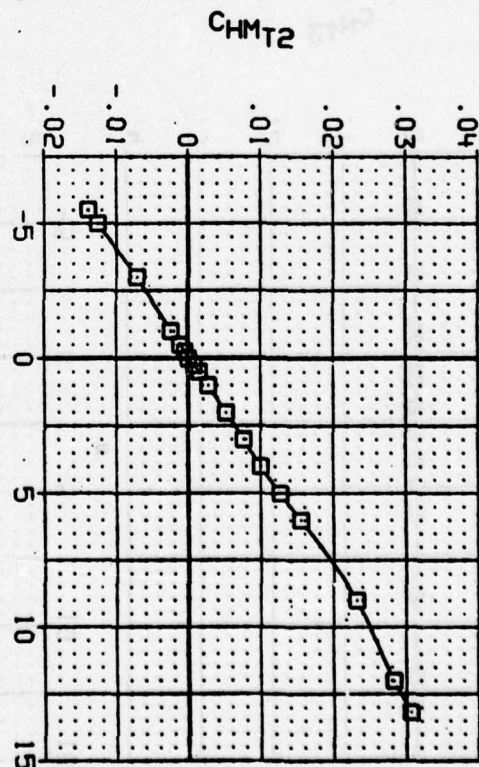
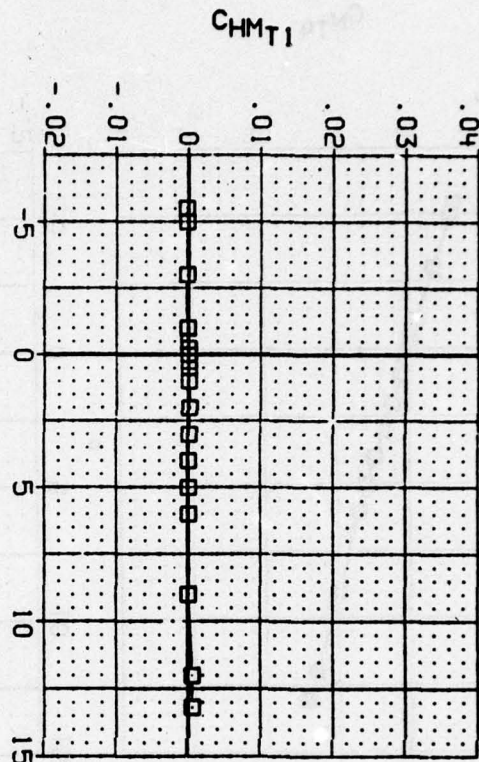


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=0$
 $(C)MACH = 4.52$

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH016) DATA NOT AVAILABLE
 (CXH017) AEDC WJIA-CIA, CANARD CONTROL, BNIC111
 (CXH018) DATA NOT AVAILABLE
 (CXH019) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 .000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 50 IN.
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 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

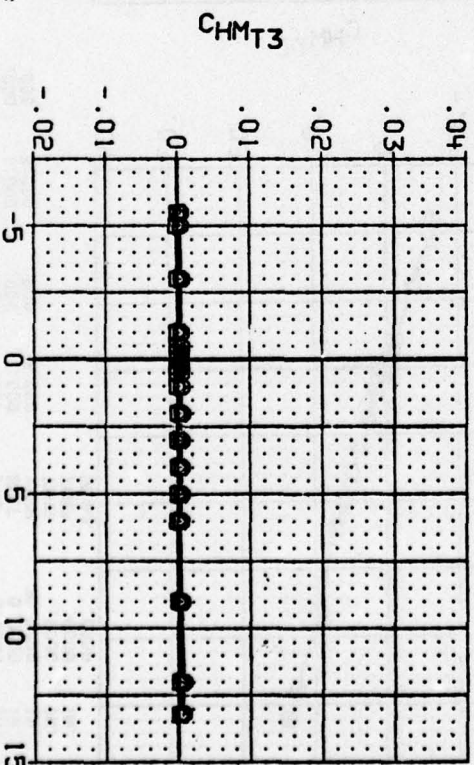
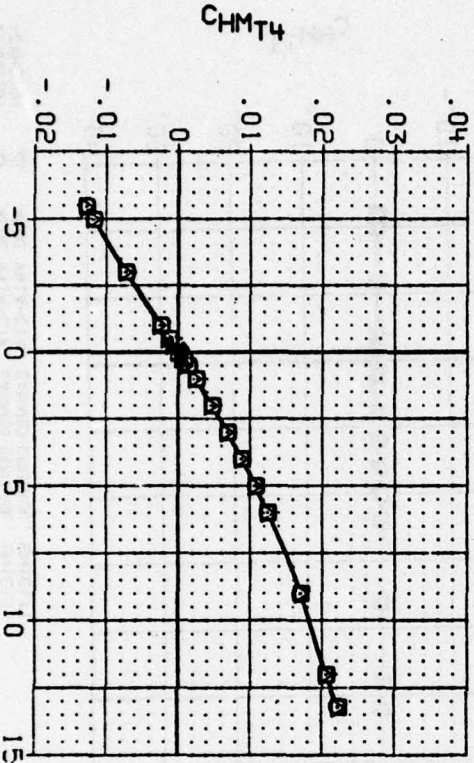
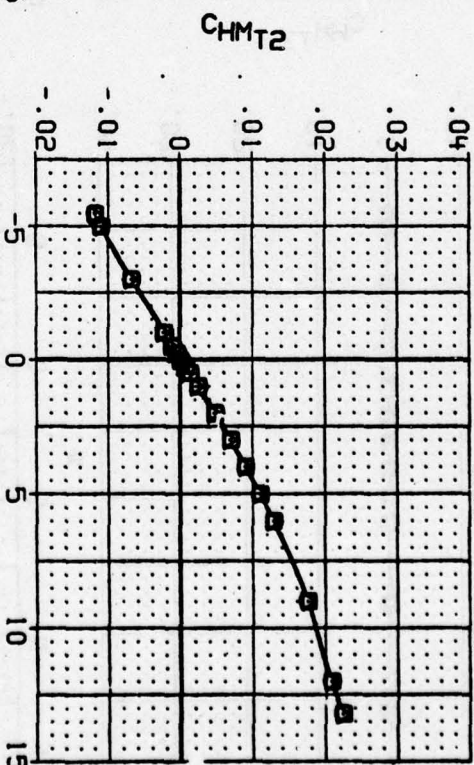
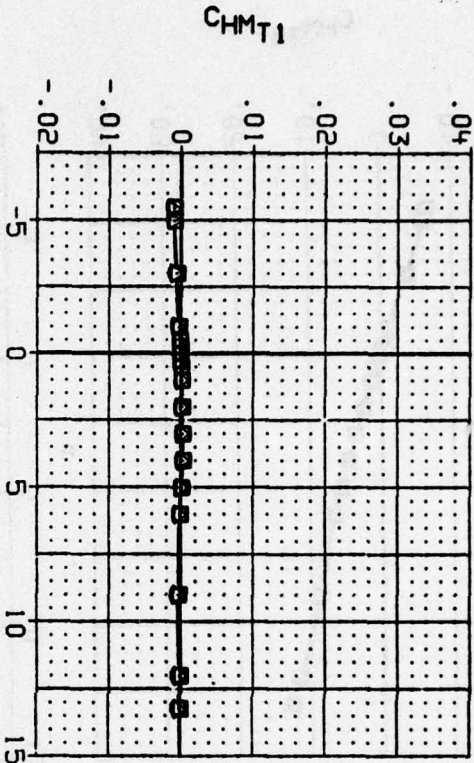


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH016) \square AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (CXH017) \square AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (CXH018) \square DATA NOT AVAILABLE
 (CXH019) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000

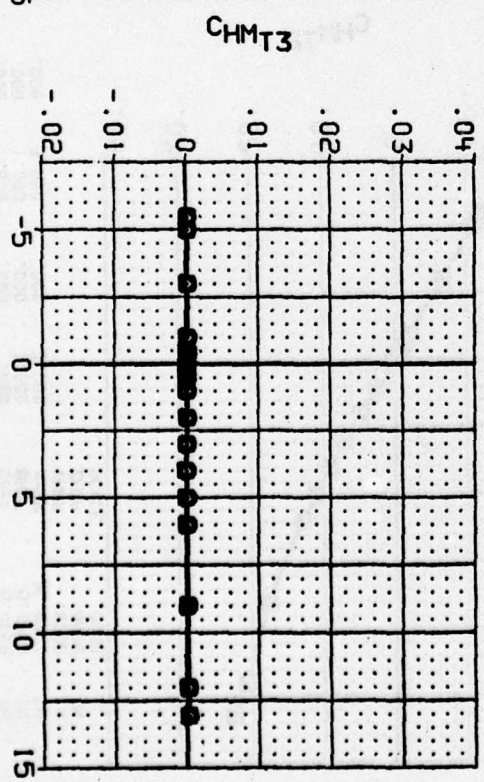
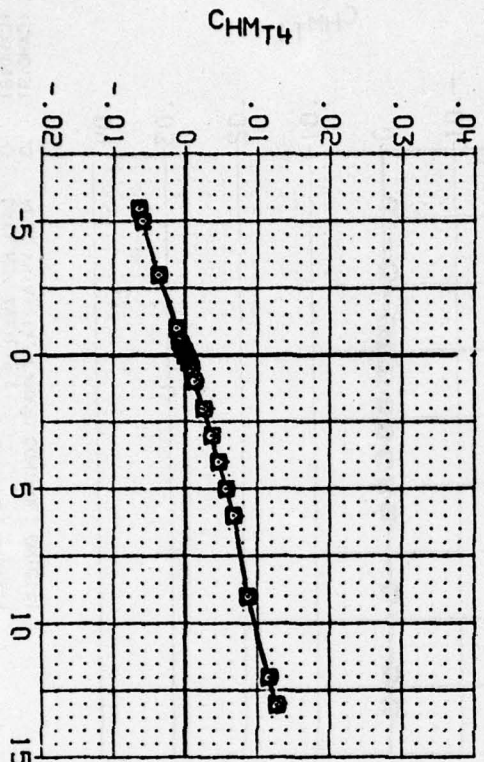
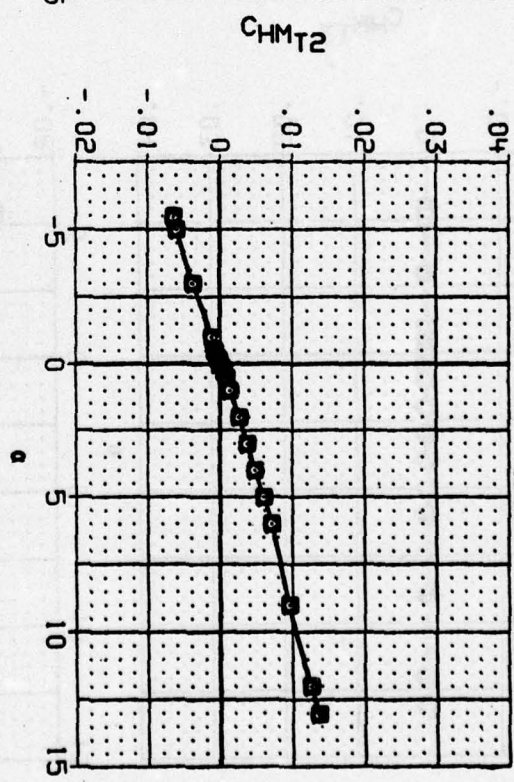
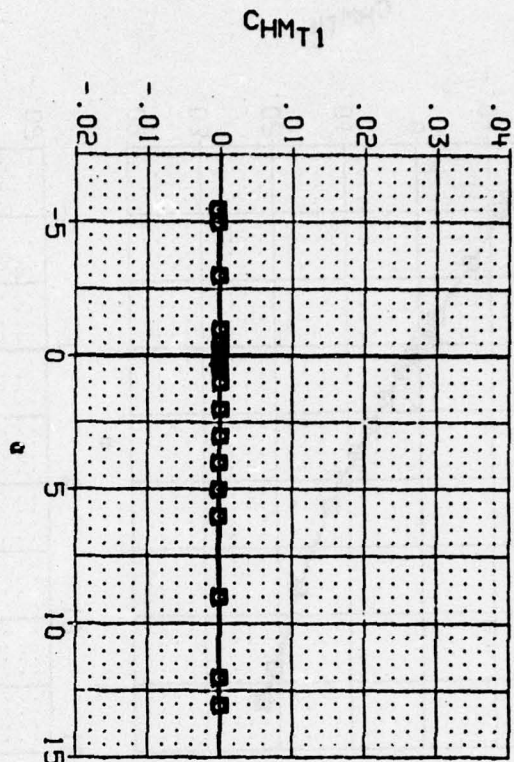


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=0
 (B) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CH016)	○	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(CH017)	□	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(CH018)	△	AEDC W1A-C1A, CANARD CONTROL, BNICITI
(CH019)	△	AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1	DCND2	DCND3	DCND4
.000	-3.000	.000	-3.000
.000	.000	.000	.000
.000	1.000	.000	1.000
.000	3.000	.000	3.000

REFERENCE INFORMATION	
SREF	19.6350
LBREF	5.0000
BRREF	5.0000
YRREF	26.0000
ZRREF	.0000
SCALE	.0000
	SO. IN.
	IN.
	IN.
	IN.

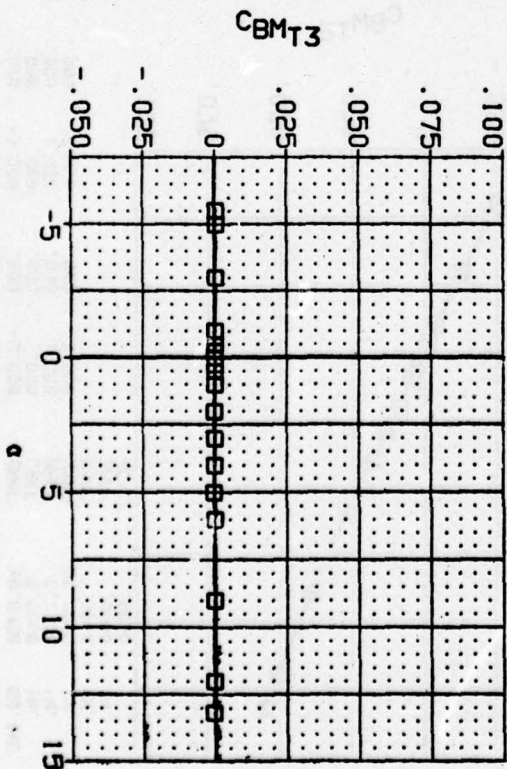
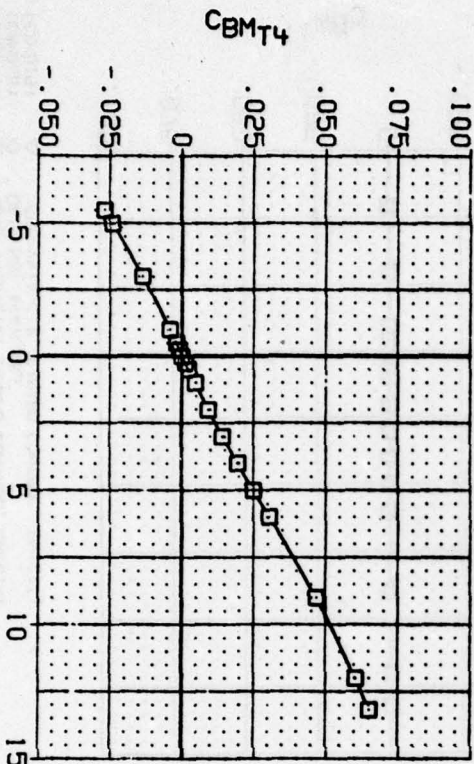
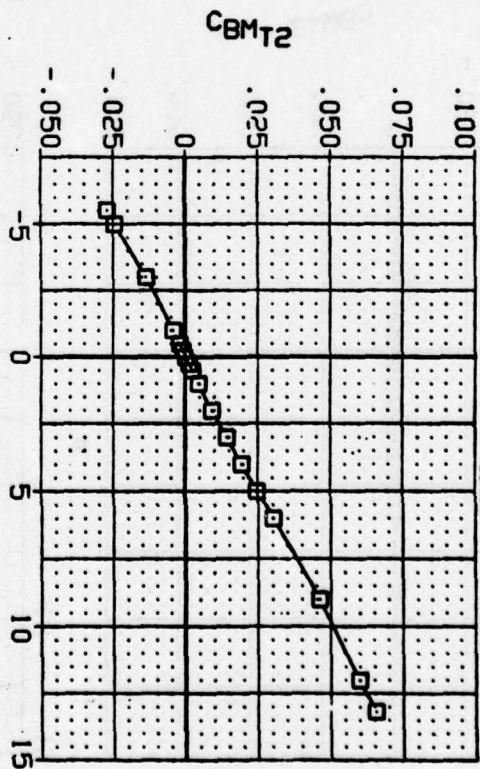
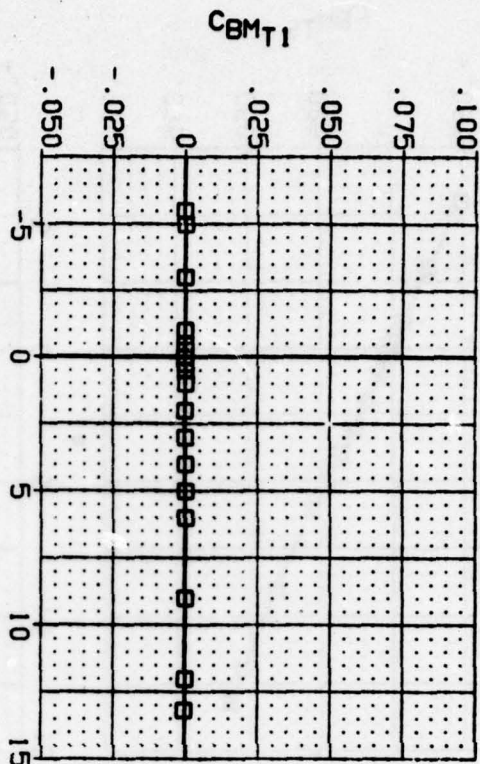


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CH016) DATA NOT AVAILABLE
 (CH017) AEDC WJIA-CIA, CANARD CONTROL, BNIC11
 (CH018) DATA NOT AVAILABLE
 (CH019) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000

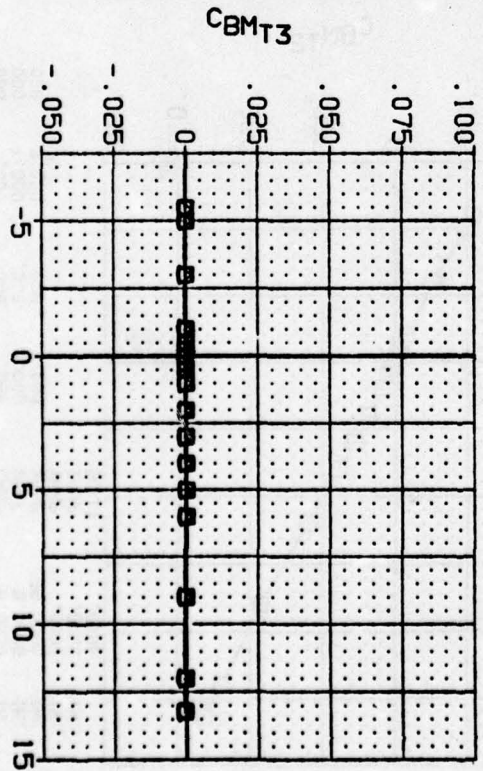
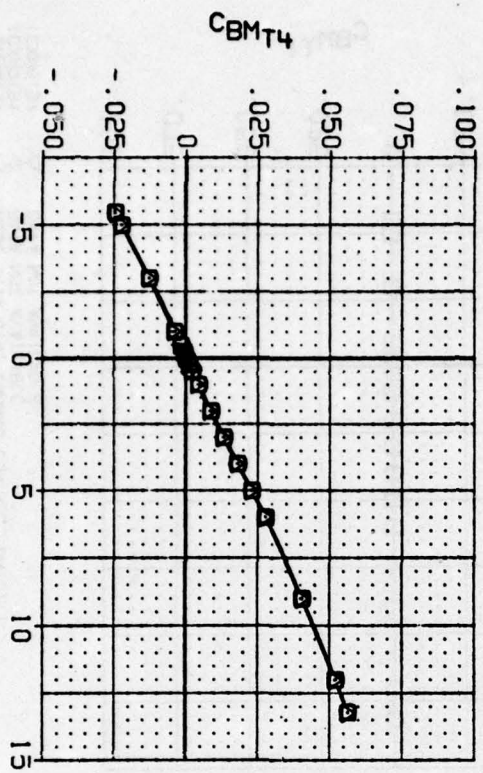
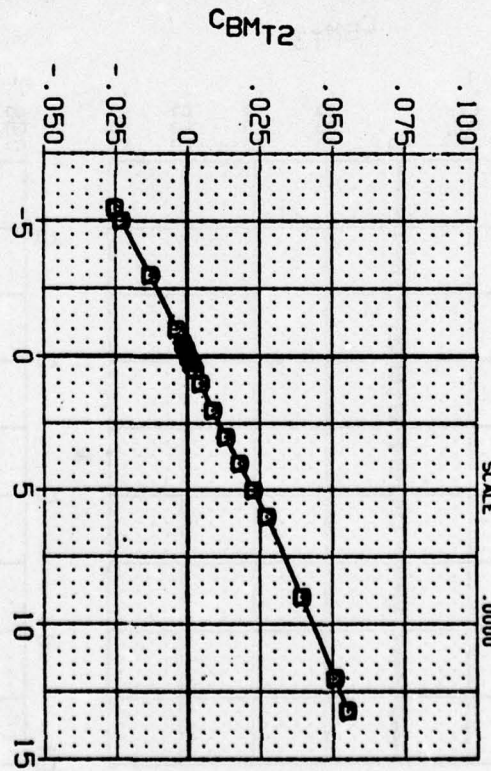
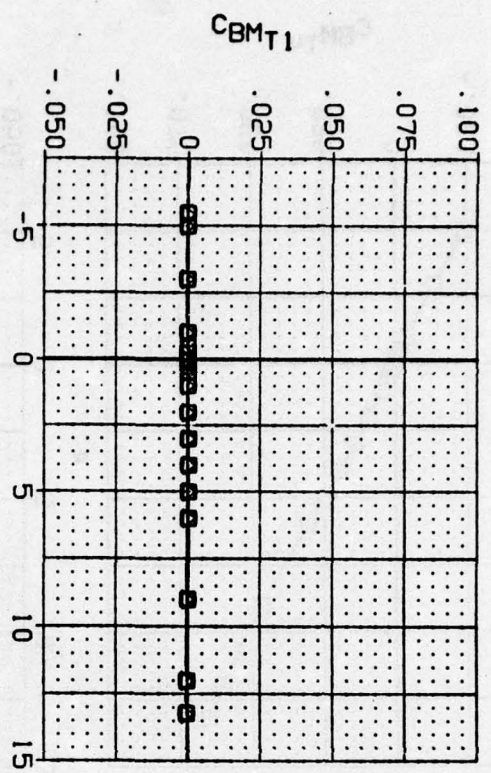


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH016) ☐ AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH017) ☐ AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH018) ☐ DATA NOT AVAILABLE
 (CXH019) ☒ AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

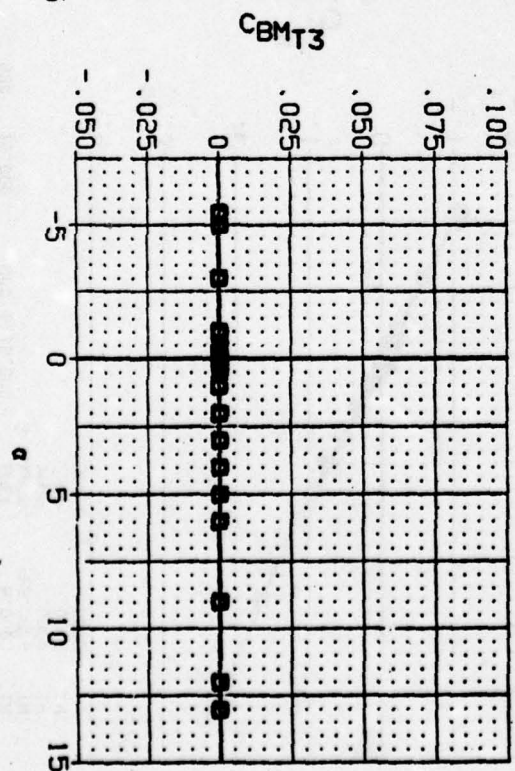
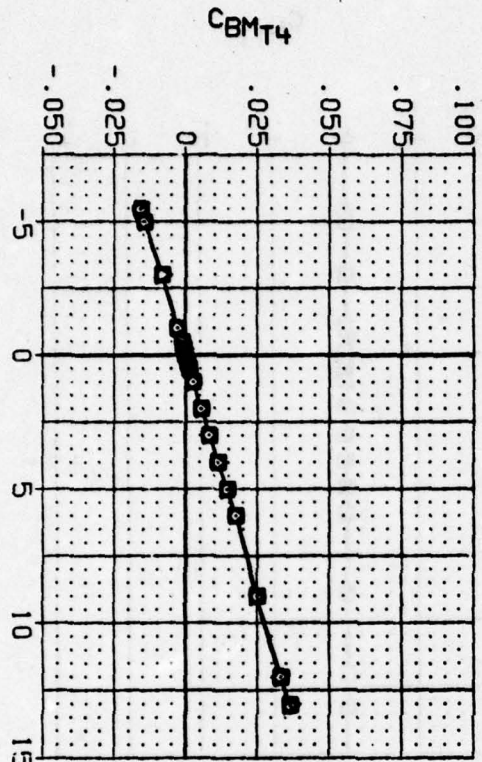
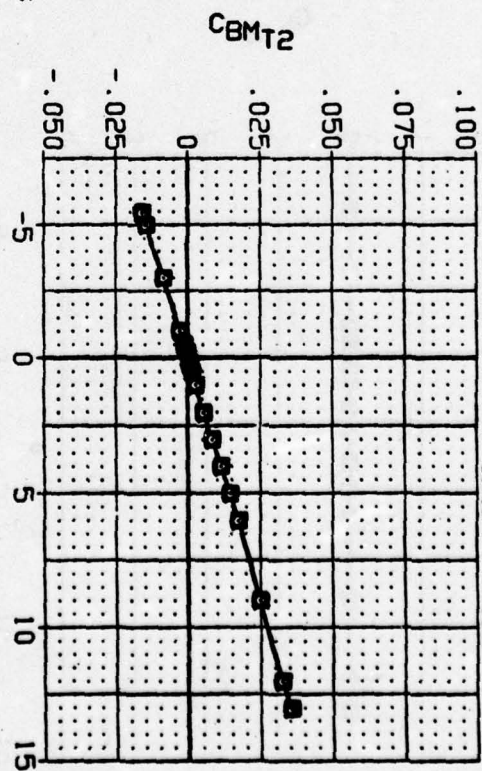
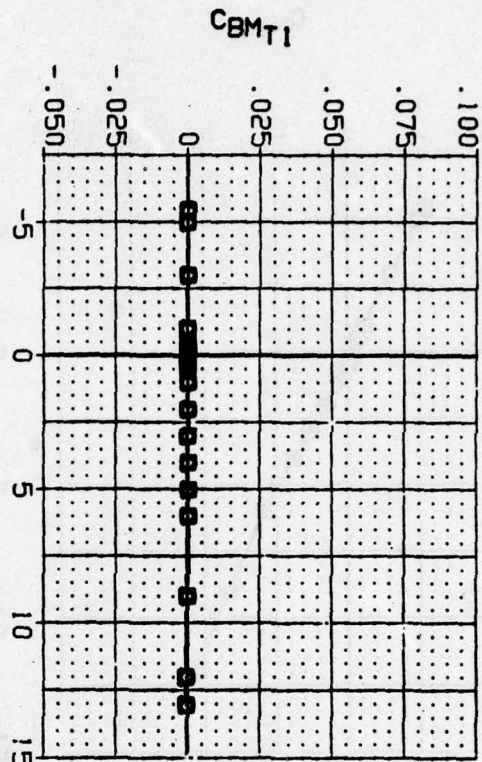


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH016) \square AEDC VM1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH017) \square AEDC VM1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH018) \square AEDC VM1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH019) \square AEDC VM1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

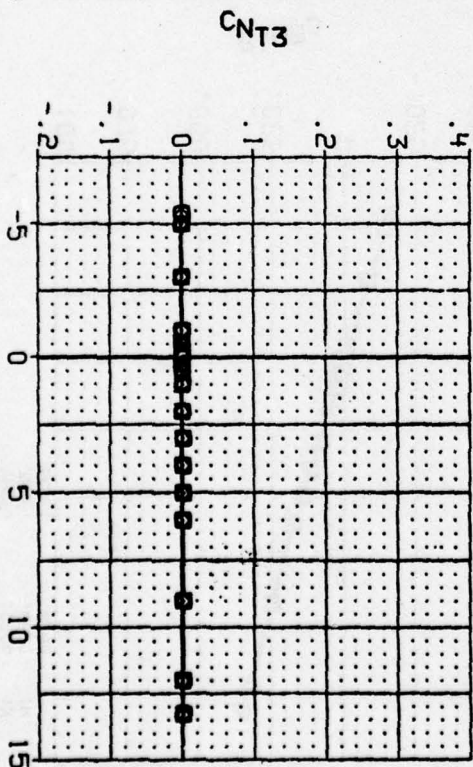
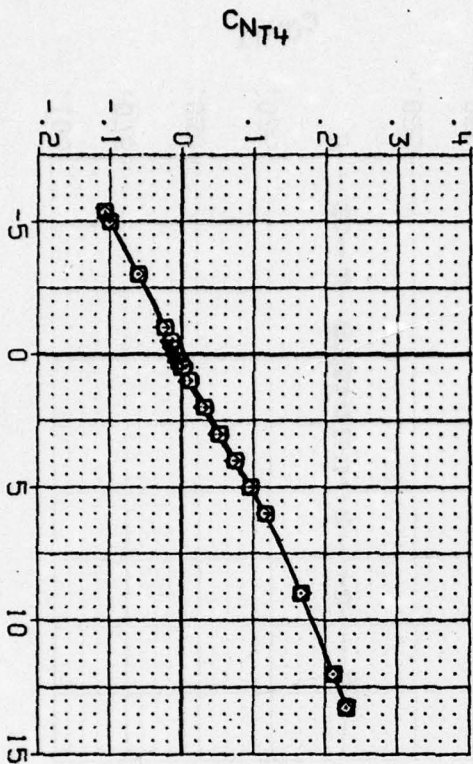
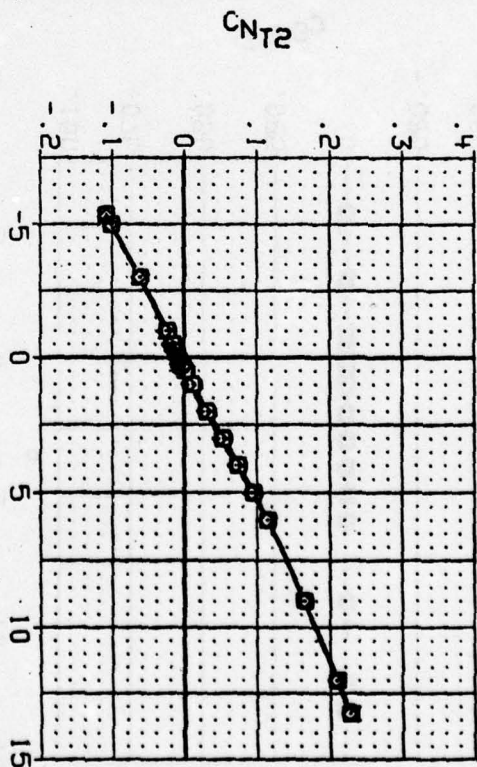
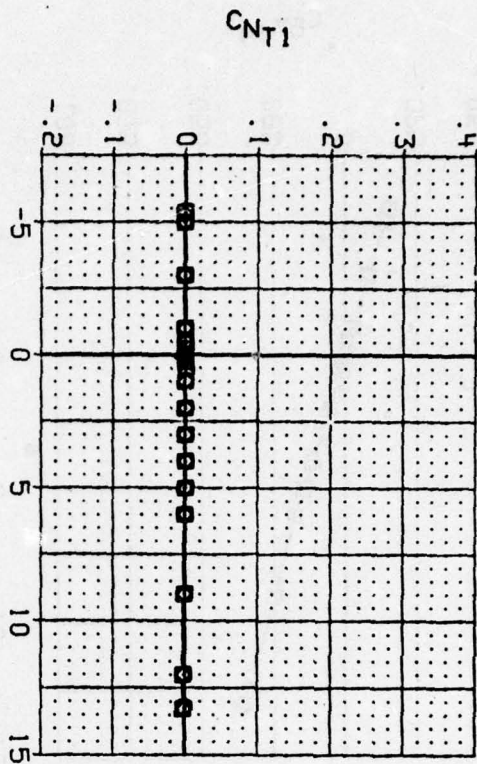


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH020) \square AEDC VMIA-C1A, CANARD CONTROL, BNIC111
 (AXH021) \square AEDC VMIA-C1A, CANARD CONTROL, BNIC111
 (AXH022) \square AEDC VMIA-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4
 .000 5.000 .000 5.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

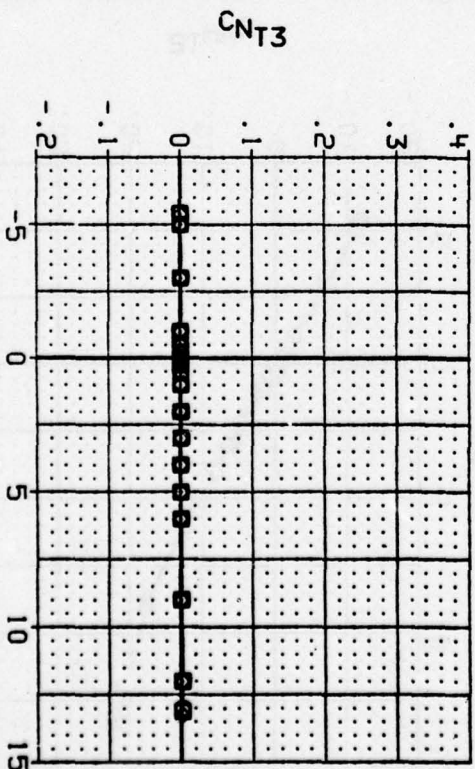
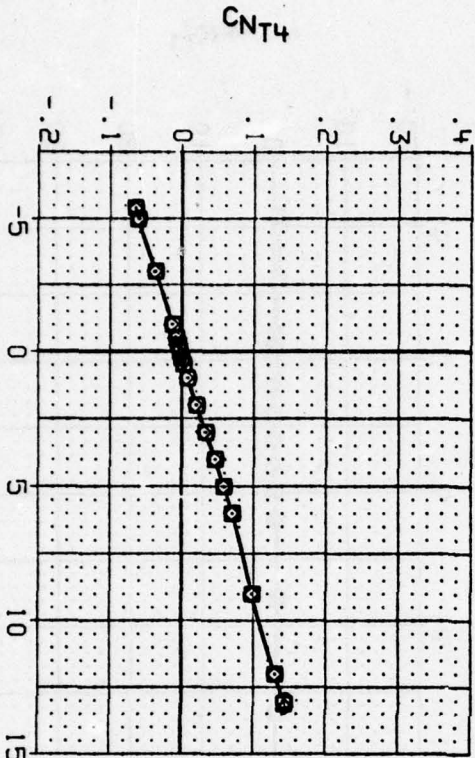
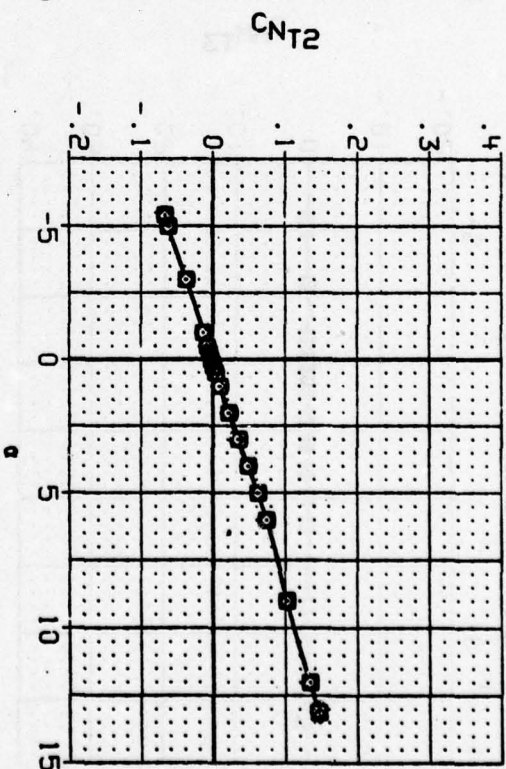
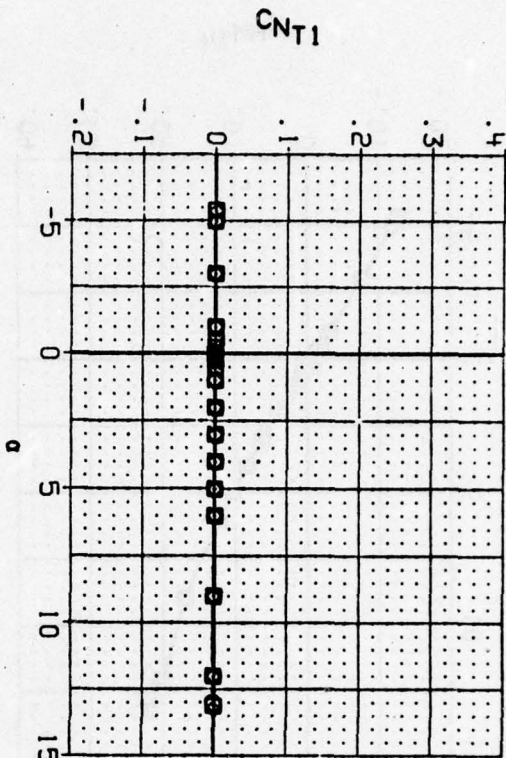


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\Phi_{HTAL}=0$ $\Phi_{HCND}=0$
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH020) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH021) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (AXH022) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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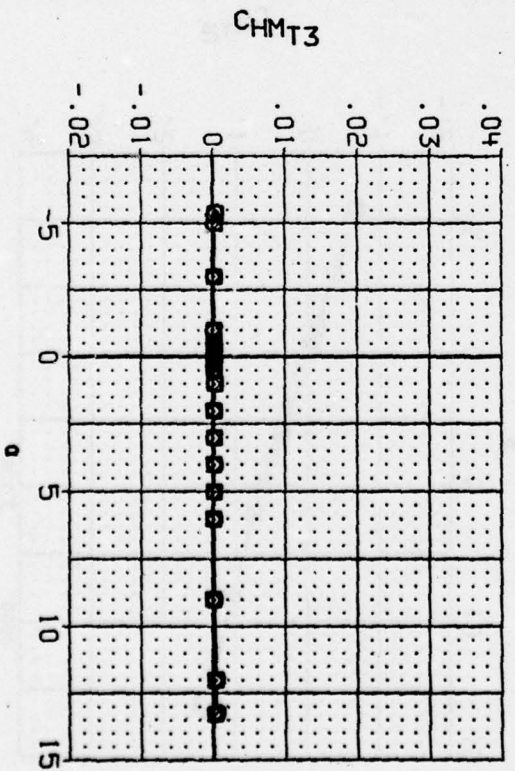
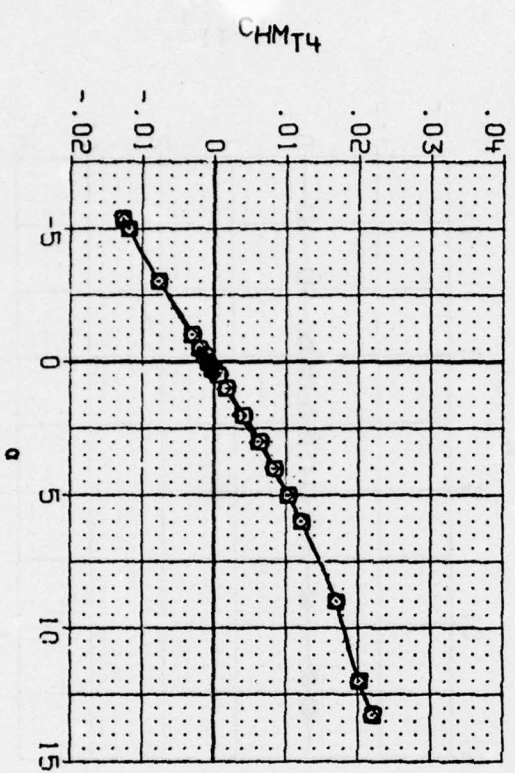
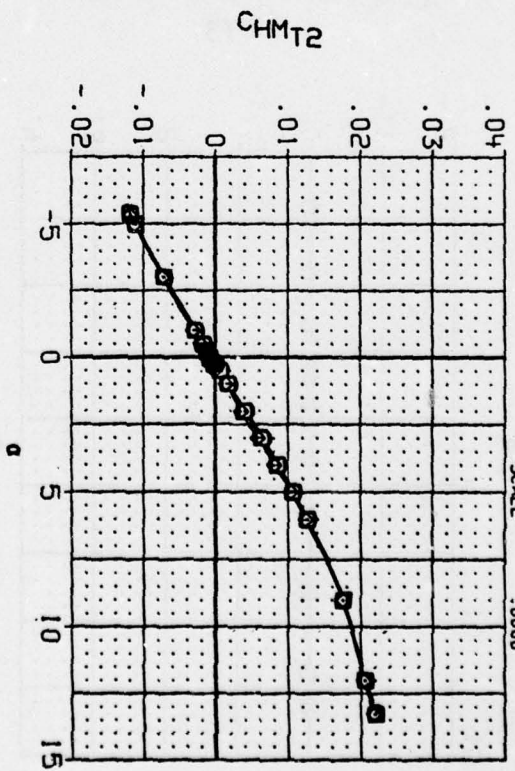
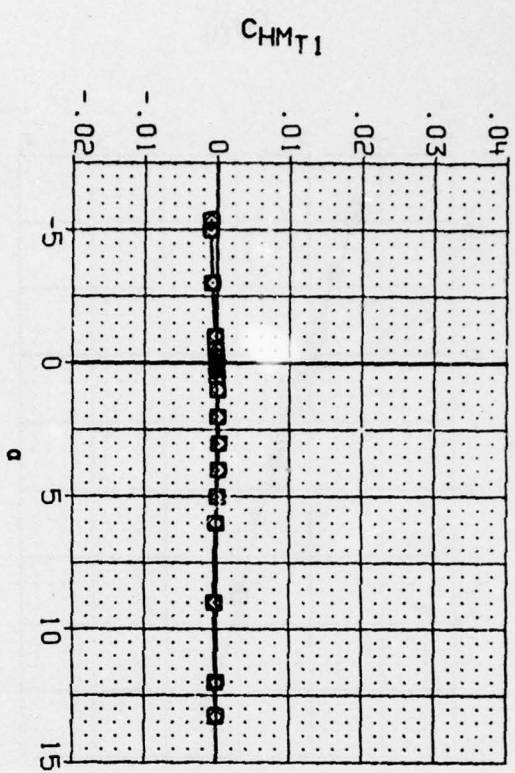


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH020) ☐ AEDC W41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH021) ☐ AEDC W41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH022) ☐ AEDC W41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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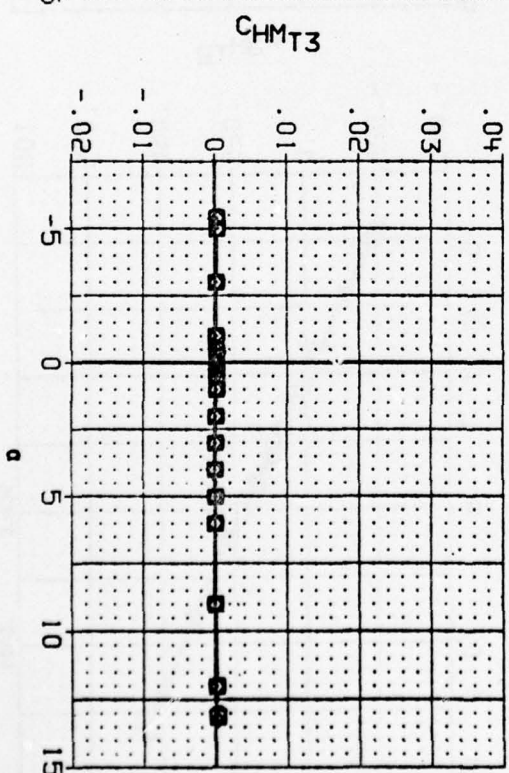
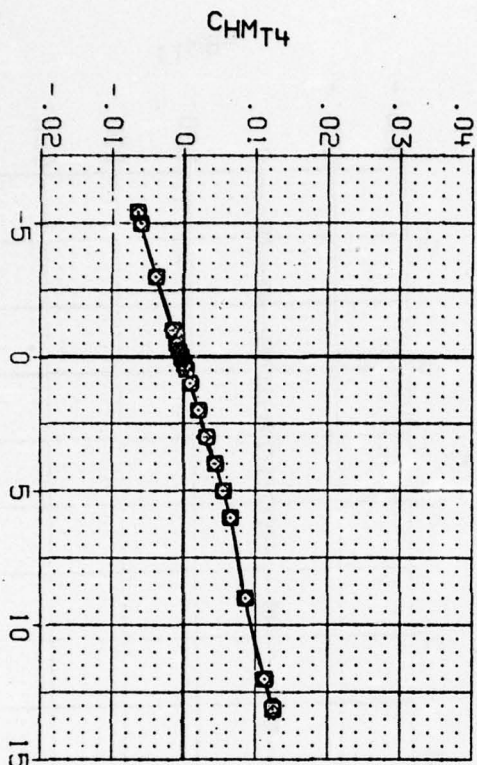
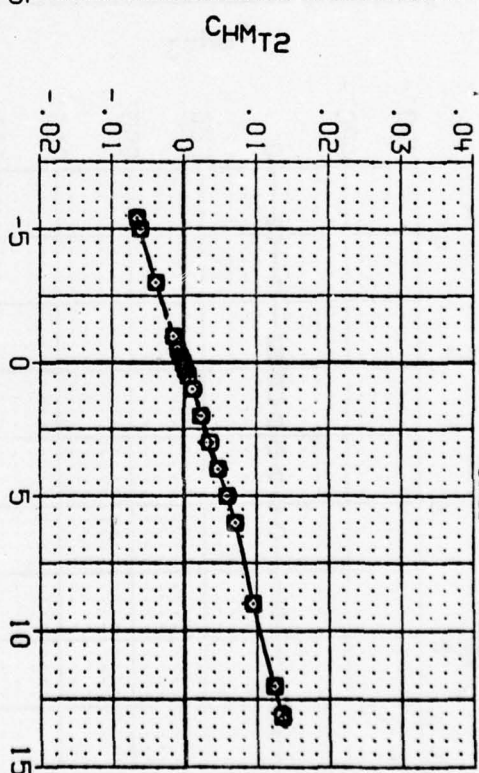
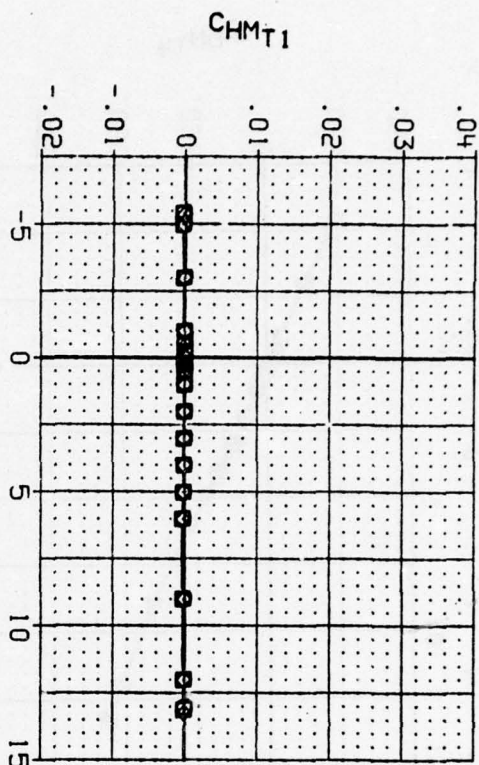


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH020) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (CXH021) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (CXH022) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 15.000
 .000 15.000 .000 15.000

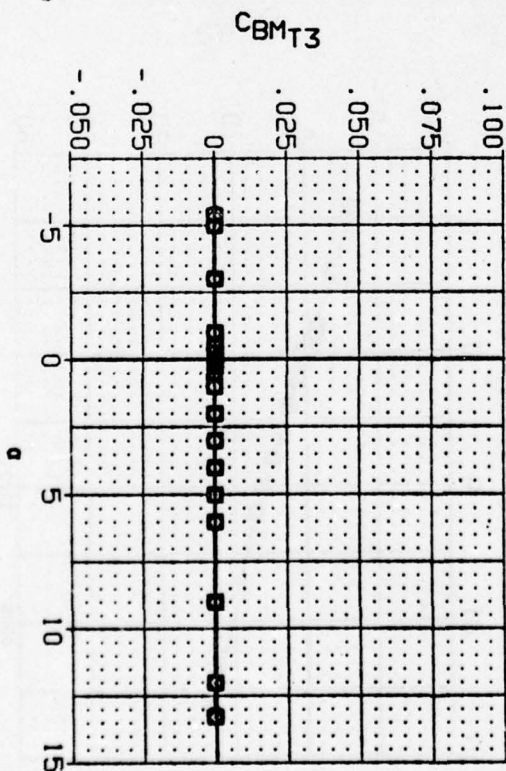
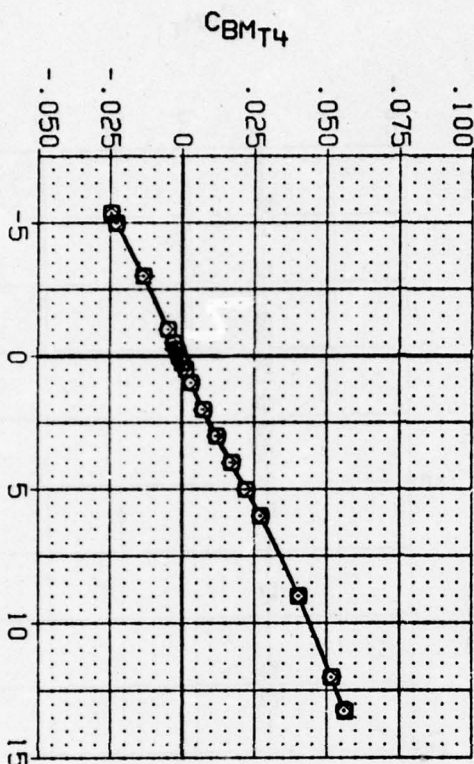
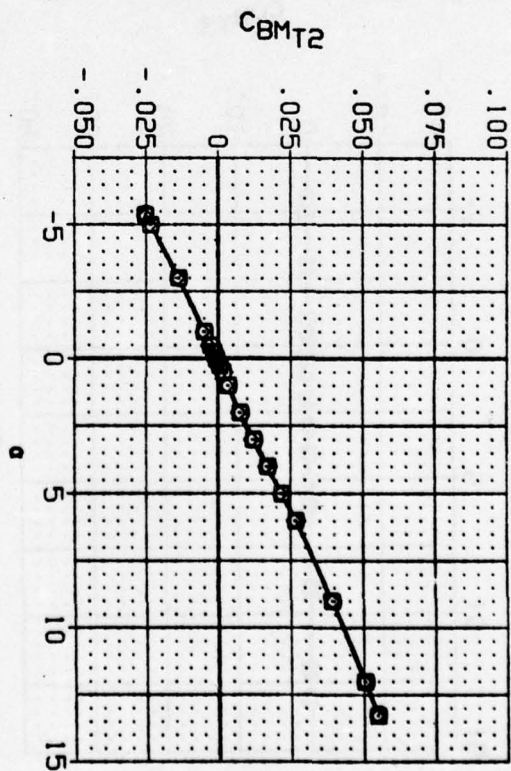
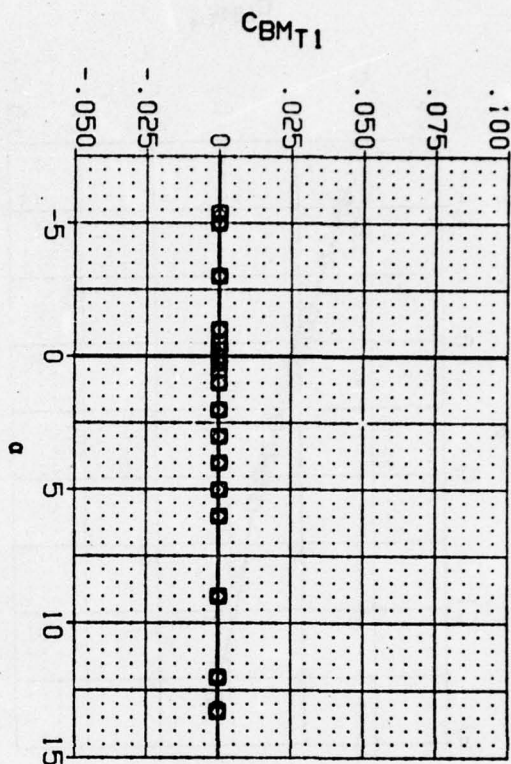
REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CH020) ☐ AEDC VNA-CIA, CANARD CONTROL, B11C1T1
 (CH021) ☐ AEDC VNA-CIA, CANARD CONTROL, B11C1T1
 (CH022) ☒ AEDC VNA-CIA, CANARD CONTROL, B11C1T1

DCND1 DCND2 DCND3 DCND4
 .000 5.000 .000 5.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000
 REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LBREF 5.0000 IN.
 BRREF 5.0000 IN.
 YHREF 26.0000 IN.
 ZHREF .0000 IN.
 SCALE .0000

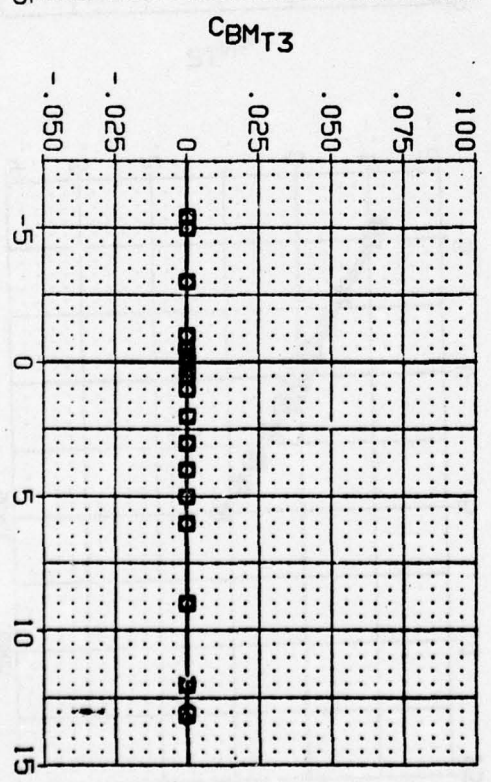
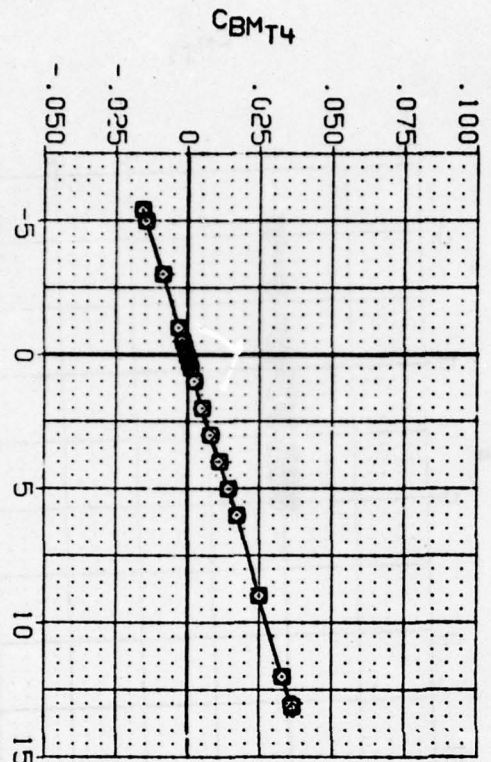
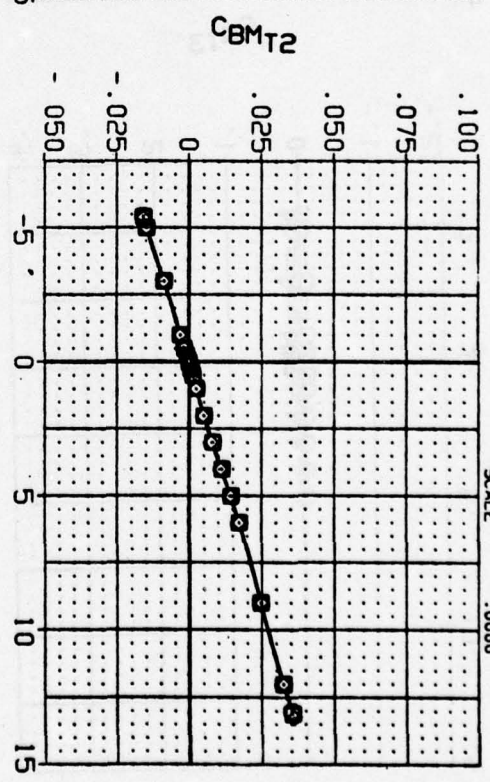
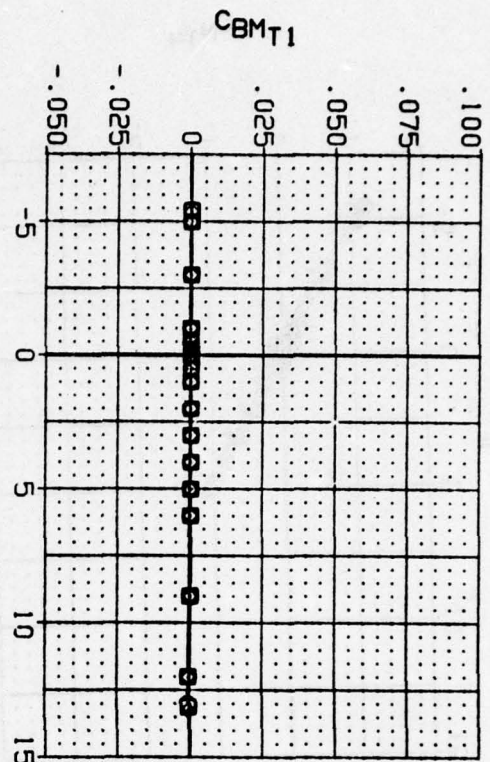


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH020) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (CXH021) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICITI
 (CXH022) ☒ AEDC WIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
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 SCALE .0000

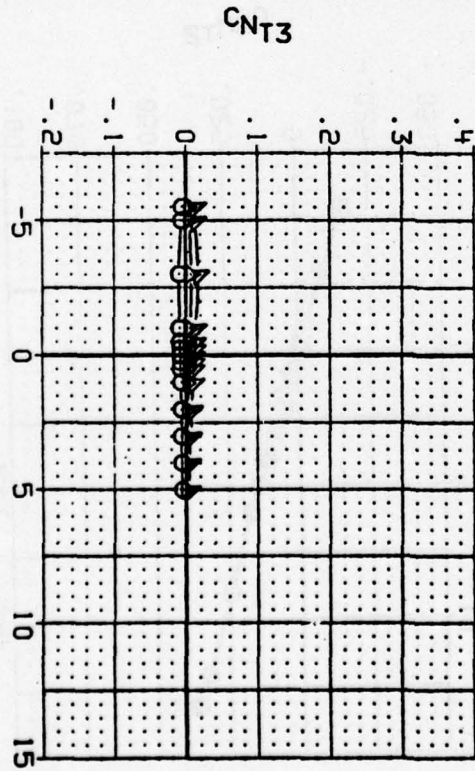
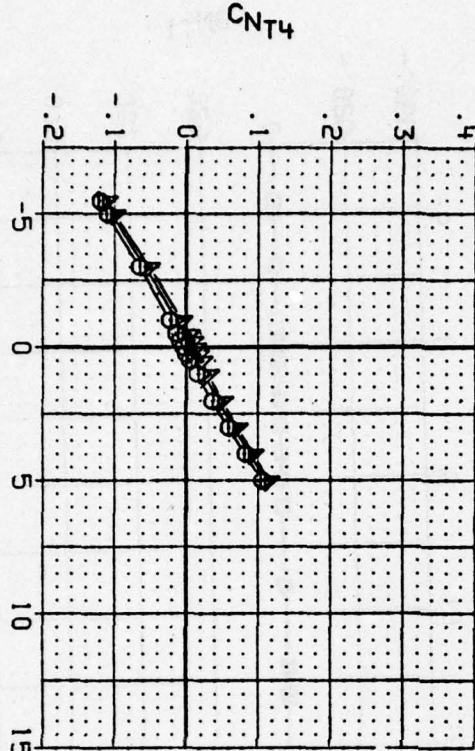
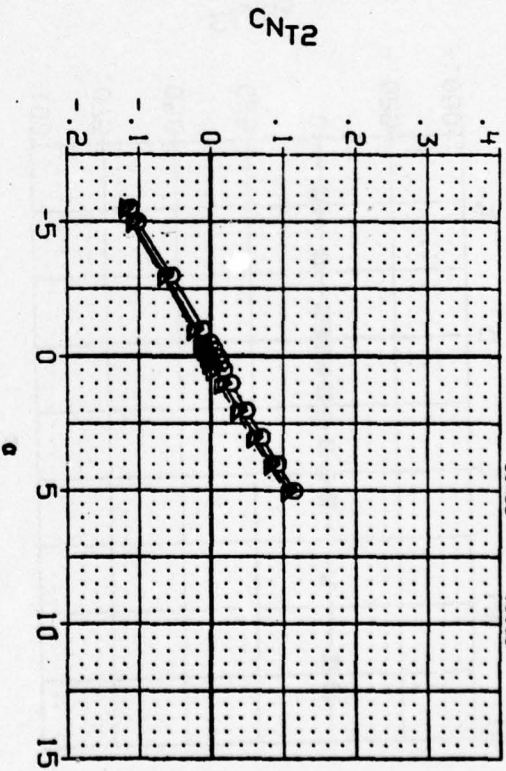
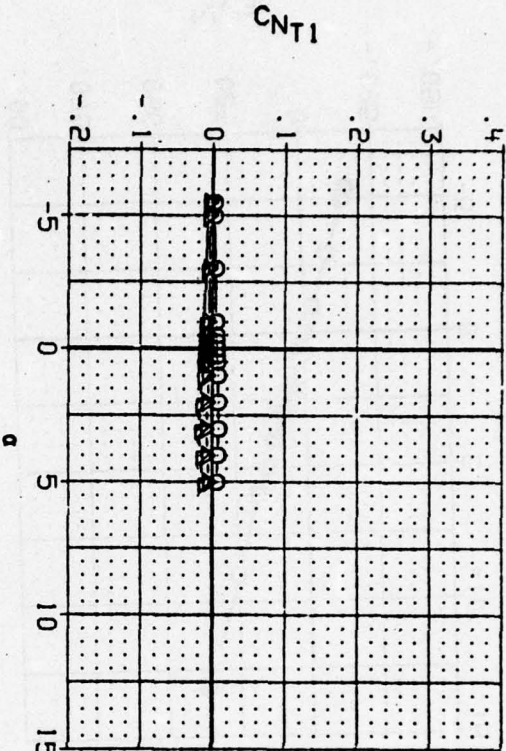


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH023) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH024) ☐ DATA NOT AVAILABLE
 (AXH025) ☐ DATA NOT AVAILABLE
 (AXH026) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH027) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 -1.500 -1.500 -1.500 -1.500
 1.000 1.000 1.000 1.000
 2.000 2.000 2.000 2.000
 5.000 5.000 5.000 5.000

REFERENCE INFORMATION
 SREF 19.6350 50 IN.
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 SCALE .0000

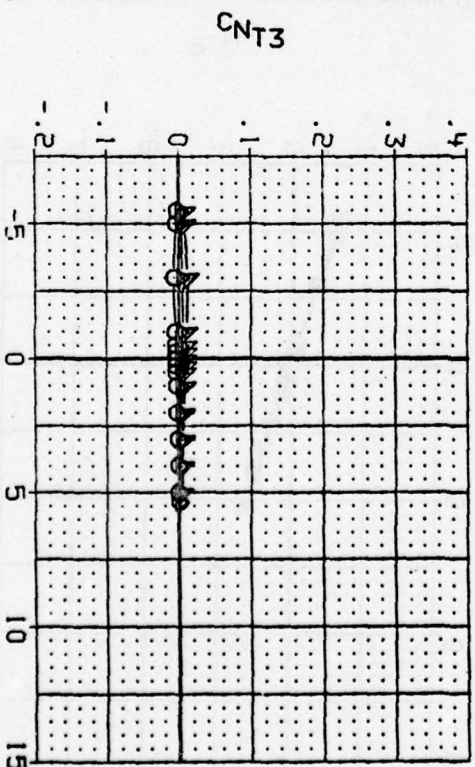
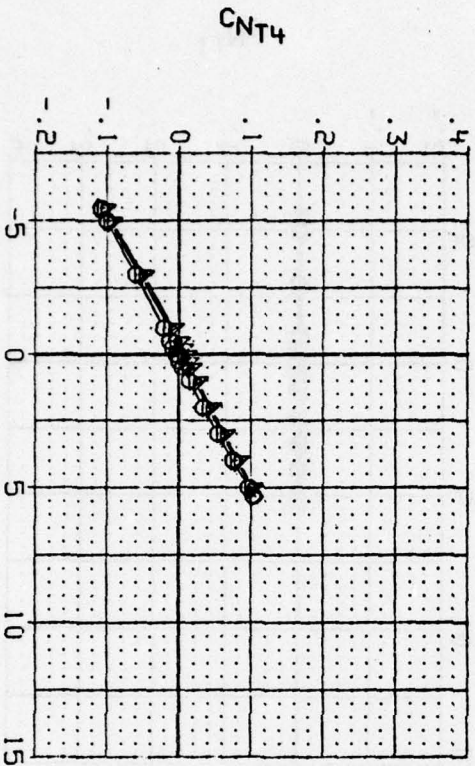
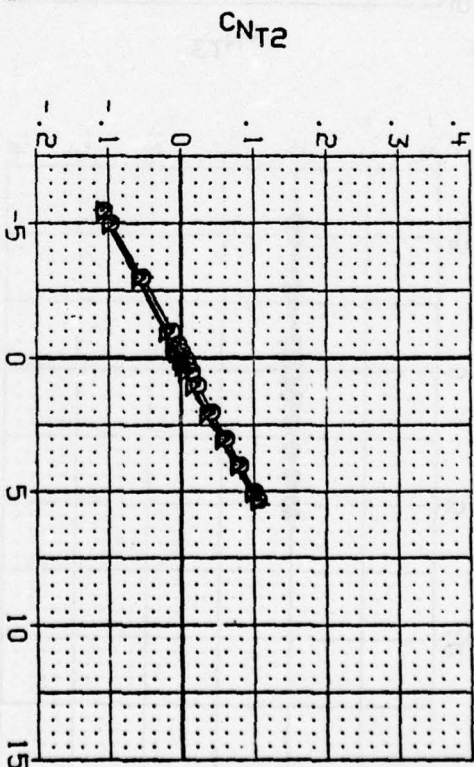
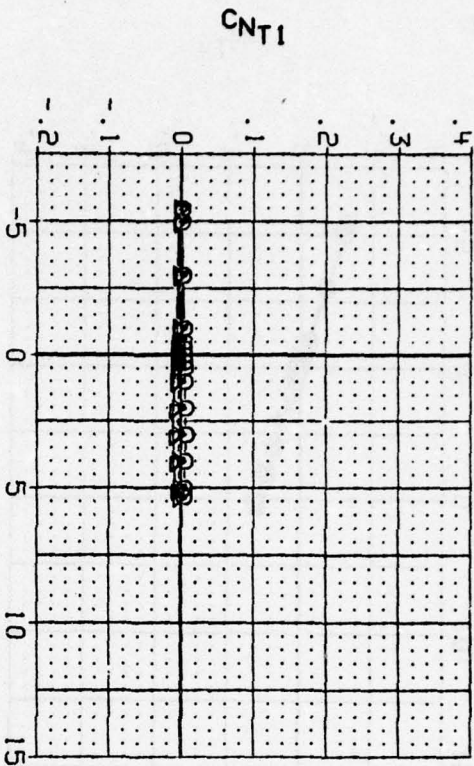


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH023) ☐ AEDC WIA-CIA, CANARD CONTROL, BNIC1T1
 (AXH024) ☐ DATA NOT AVAILABLE
 (AXH025) ☐ DATA NOT AVAILABLE
 (AXH026) ☐ AEDC WIA-CIA, CANARD CONTROL, BNIC1T1
 (AXH027) ☒ AEDC WIA-CIA, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 .500 -.500 -.500 .500
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
 SPEC 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH023) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH024) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH025) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH026) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH027) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 -3.000
 .500
 1.000
 2.000
 5.000

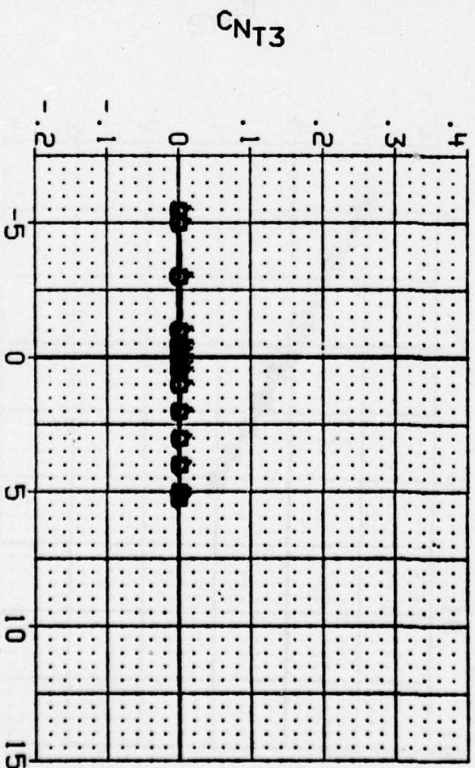
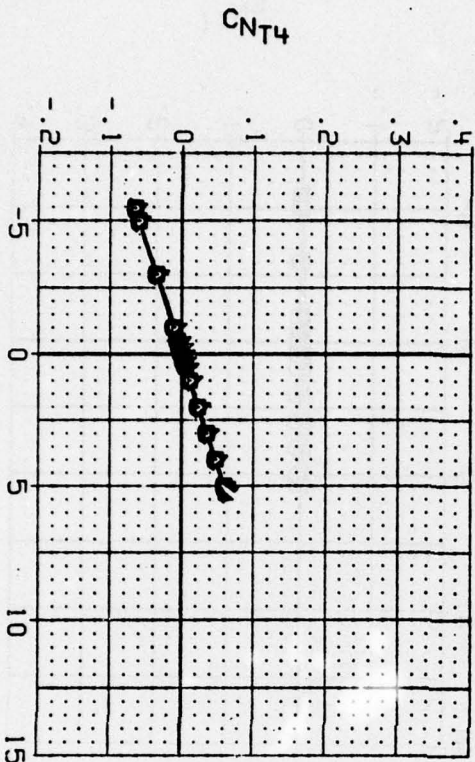
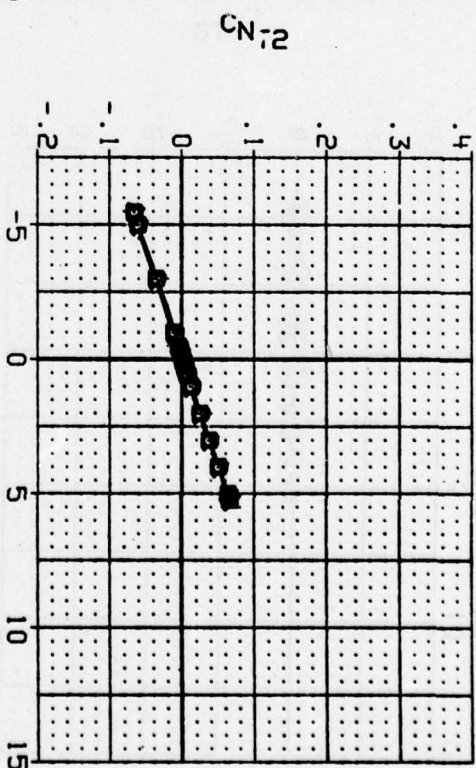
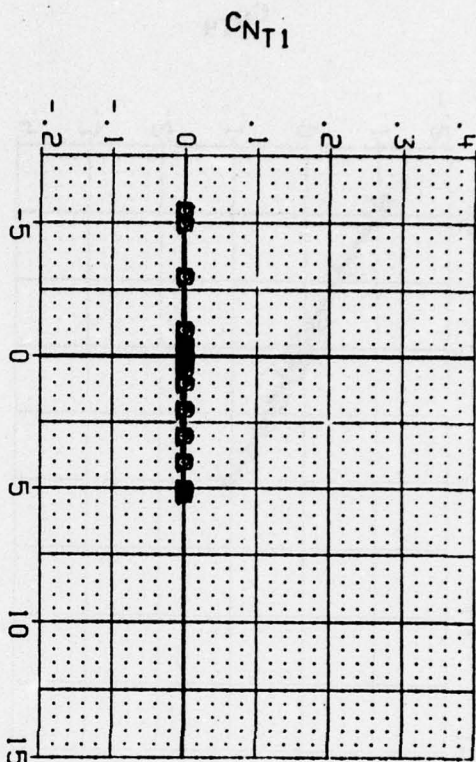
DCND2 3.000
 .500
 -1.000
 -2.000
 -5.000

DCND3 3.000
 .500
 -1.000
 -2.000
 -5.000

DCND4 -3.000
 .500
 1.000
 2.000
 5.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YREF 25.0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

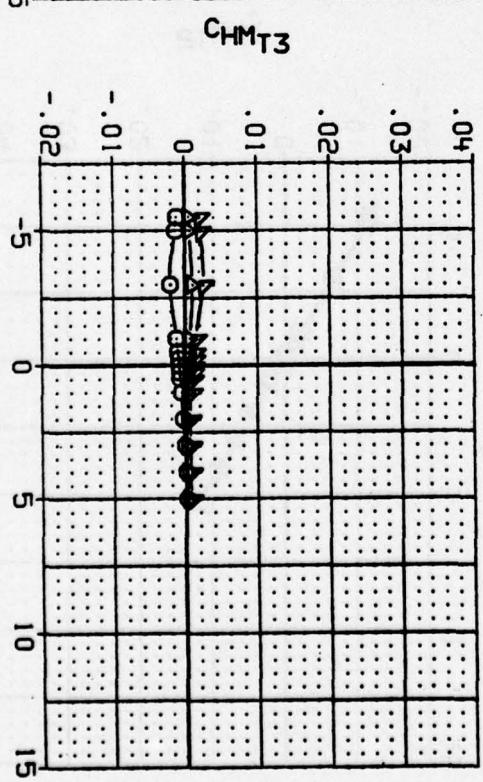
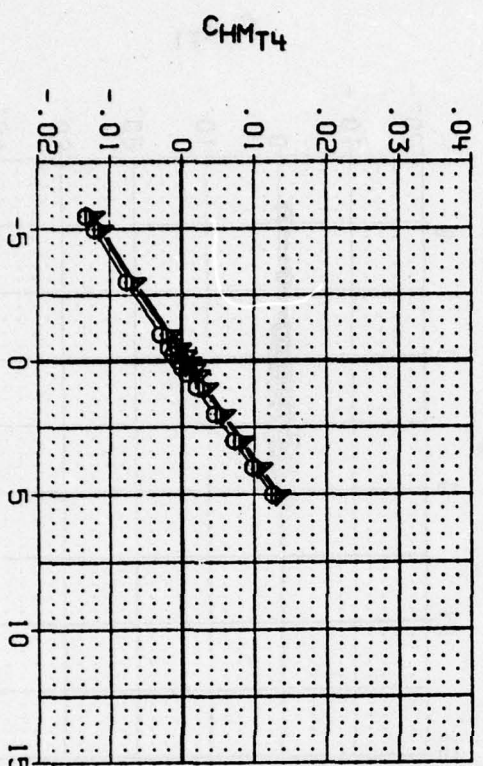
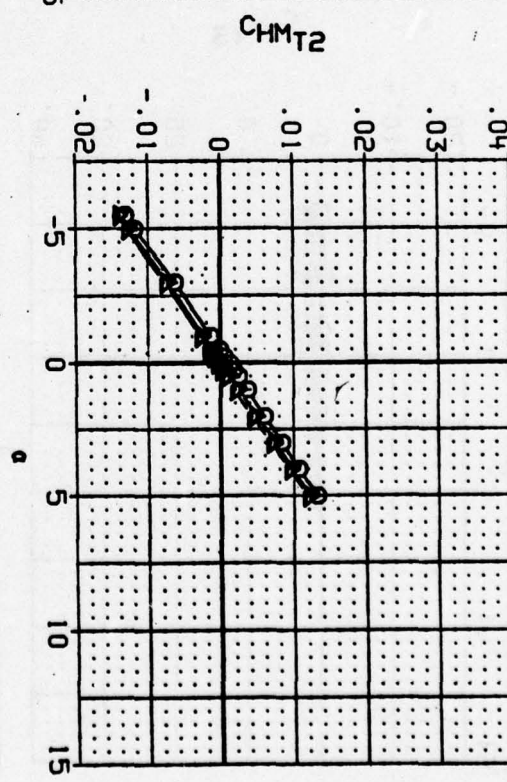
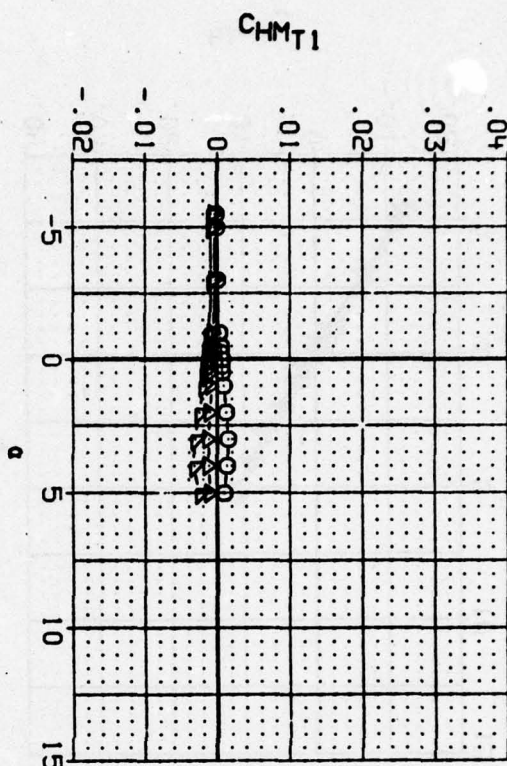
PHITAL=0 PHICND=0

(C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH023) \square AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (CXH024) \square DATA NOT AVAILABLE
 (CXH025) \square DATA NOT AVAILABLE
 (CXH026) \square AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (CXH027) \triangle AEDC VMA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 -3.000 -3.000
 .500 .500 .500 .500
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
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 BRREF 5.0000 IN.
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 SCALE .0000

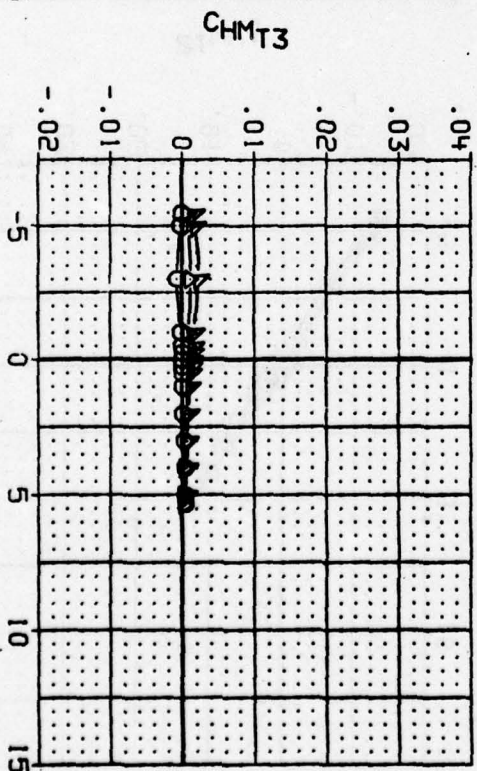
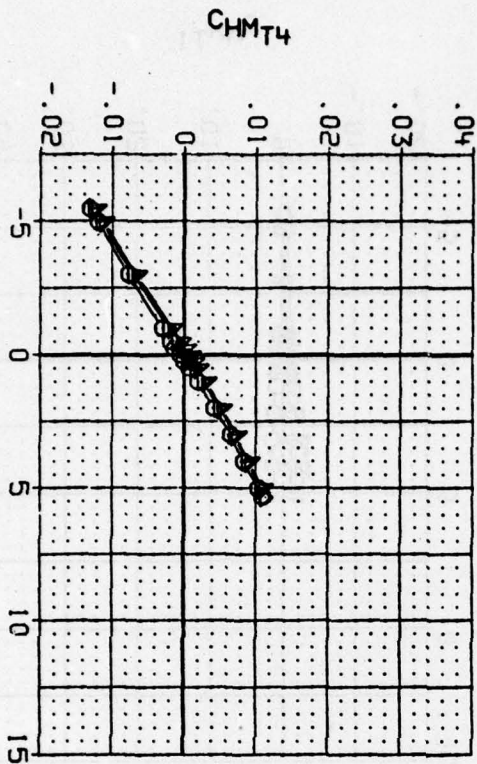
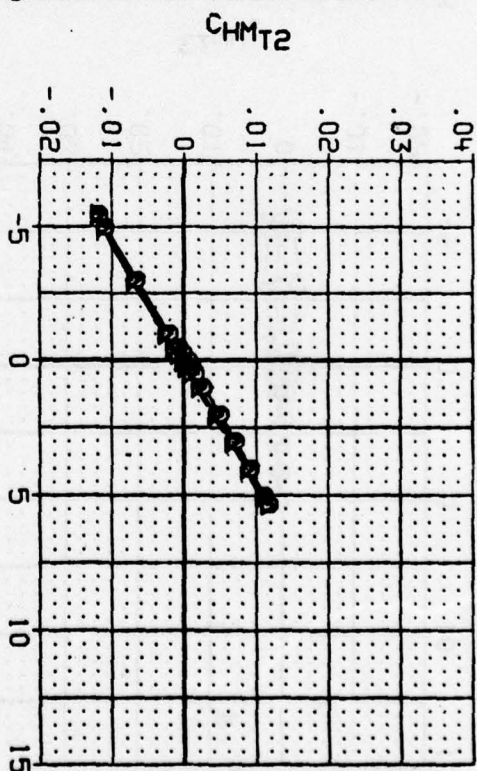
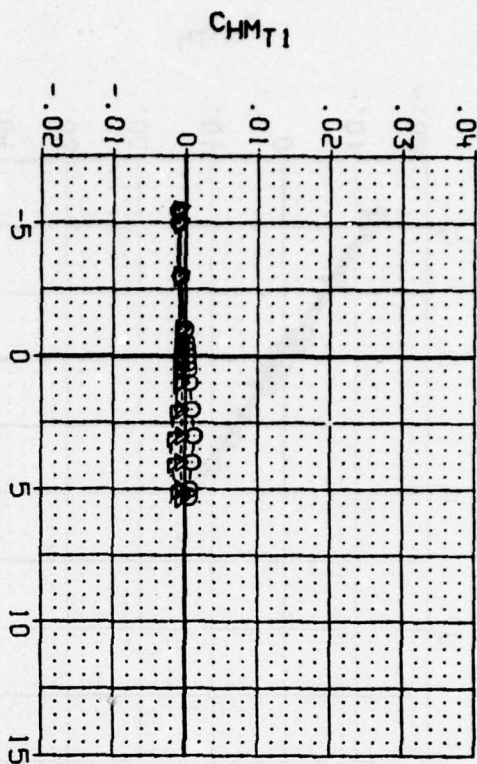


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITA=0$ $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH023) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICIT1
 (CXH024) ☐ DATA NOT AVAILABLE
 (CXH025) ☐ DATA NOT AVAILABLE
 (CXH026) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICIT1
 (CXH027) ☐ AEDC WIA-CIA, CANARD CONTROL, BNICIT1

DOIND1 DOIND2 DOIND3 DOIND4
 -3.000 3.000 -3.000 -3.000
 .500 .500 .500 .500
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
 SPEC 19.5350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF 10.0000 IN.
 ZREF 10.0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

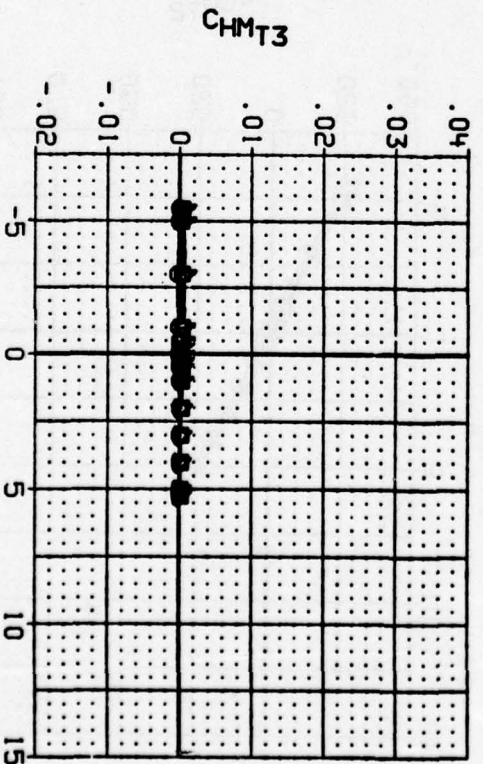
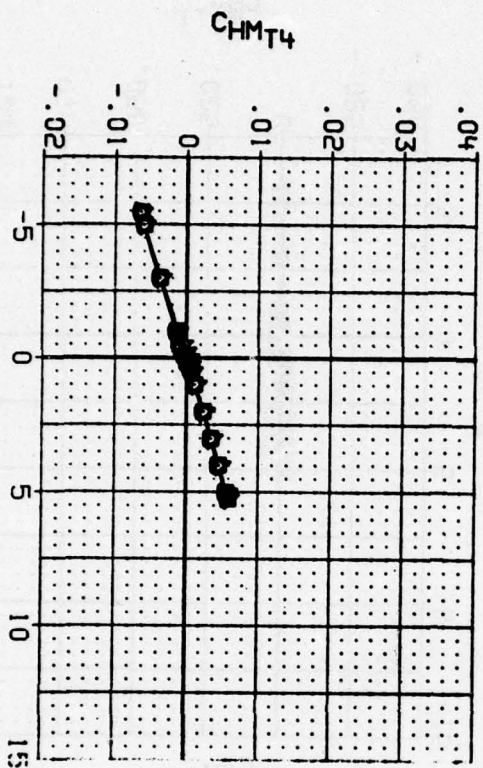
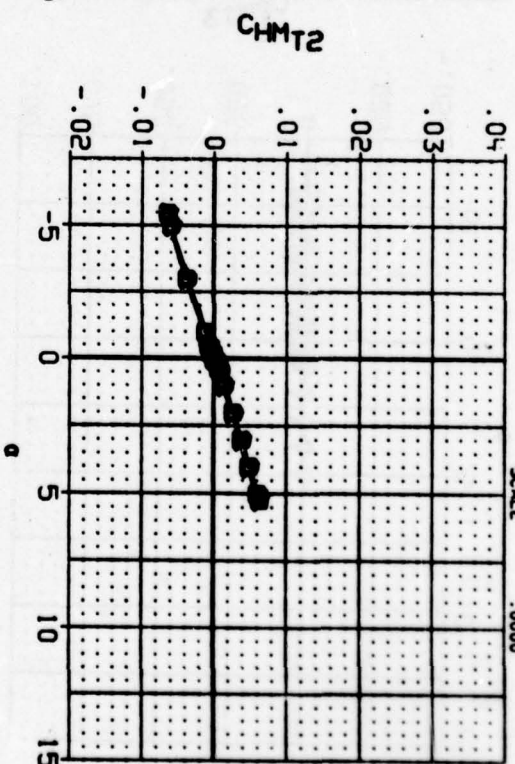
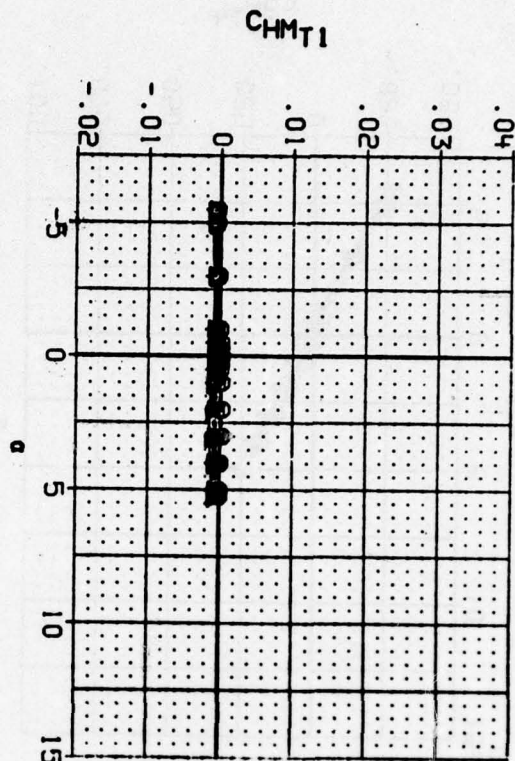
(CXH023) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (CXH024) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (CXH025) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (CXH026) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI
 (CXH027) \square AEDC W1A-C1A, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 -3.000 -3.000
 -5.000 -5.000 -5.000 -5.000
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
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 YREF .0000 IN.
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 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

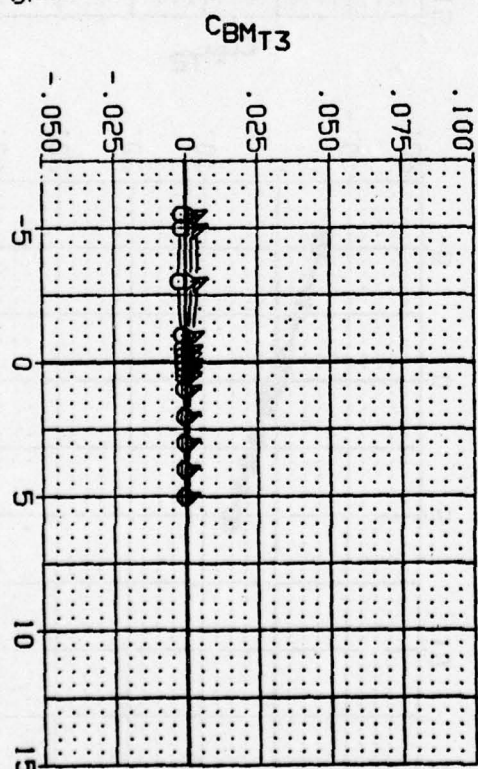
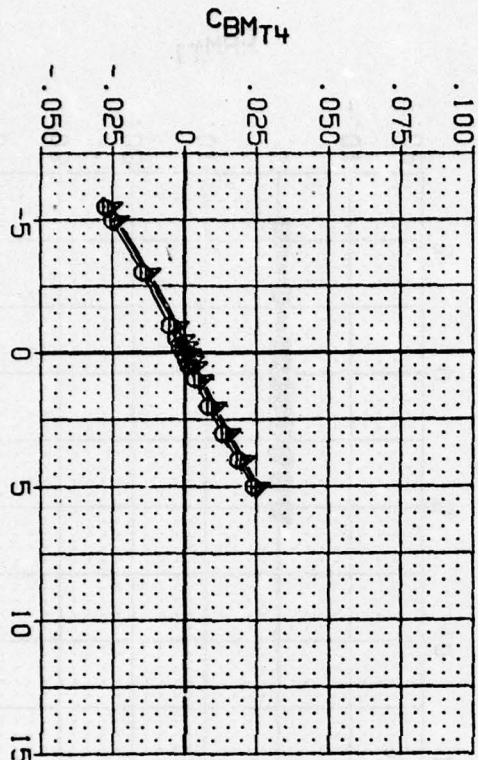
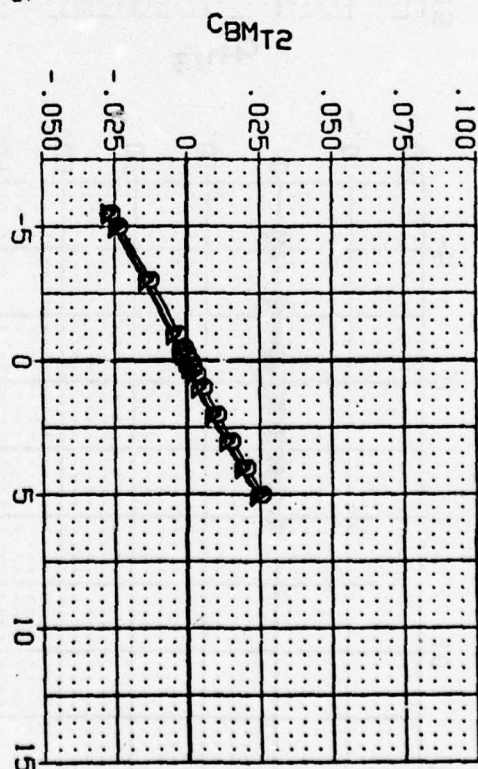
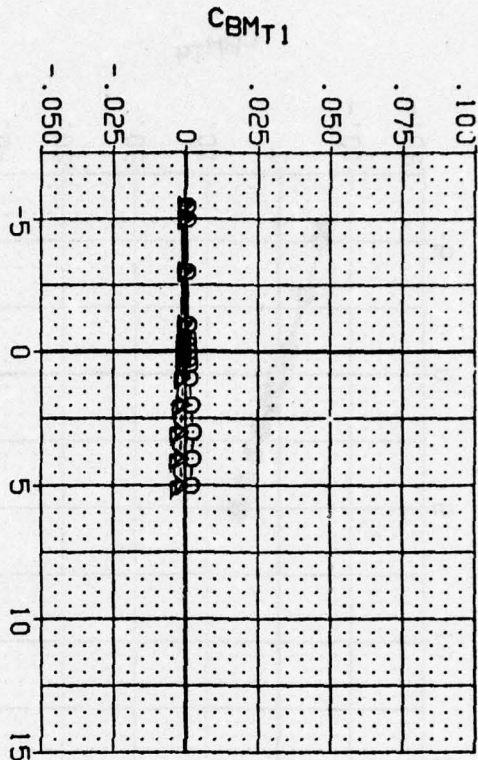
$PHITAL=0$ $PHICND=0$

(C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH023) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (CXH024) ☐ DATA NOT AVAILABLE
 (CXH025) ☐ DATA NOT AVAILABLE
 (CXH026) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC11
 (CXH027) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC11

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 1.500 -1.500 -1.500 1.500
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000

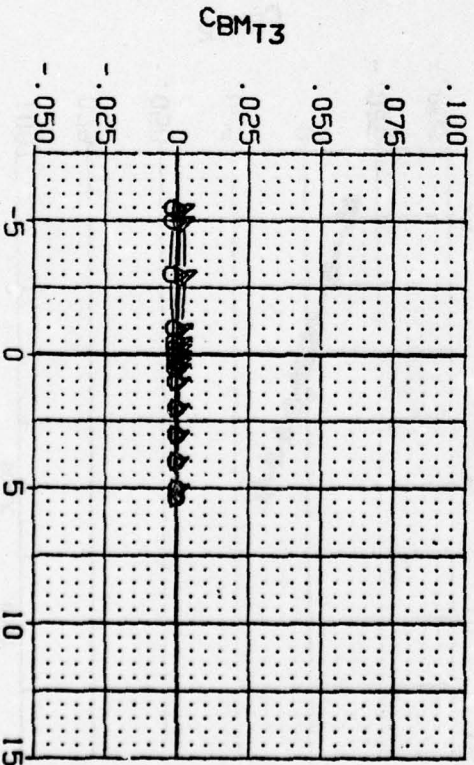
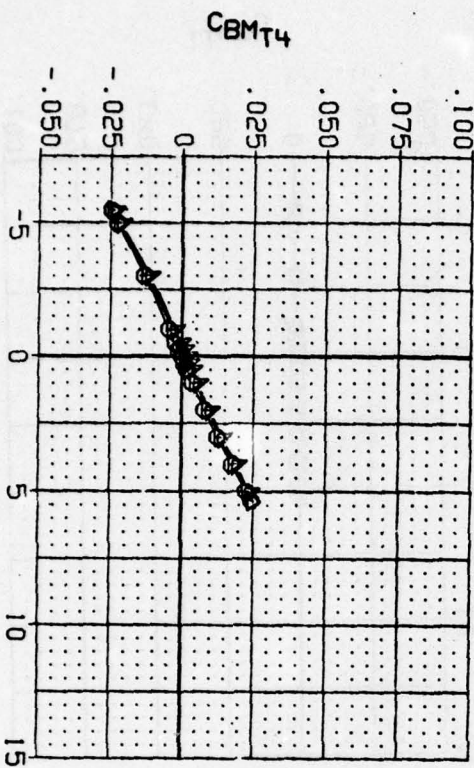
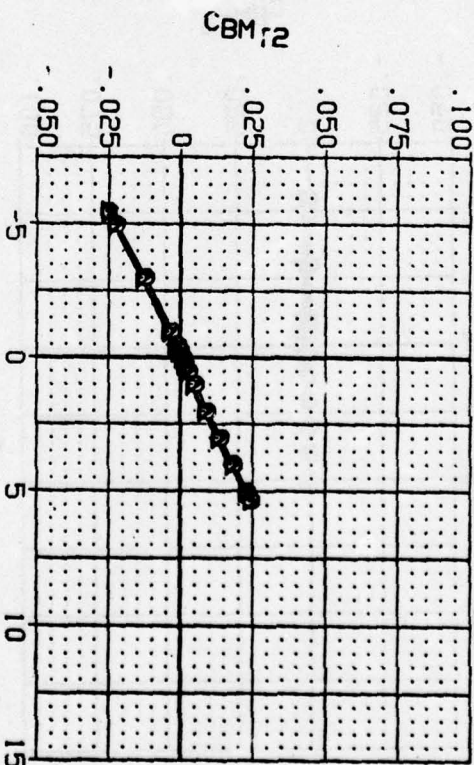
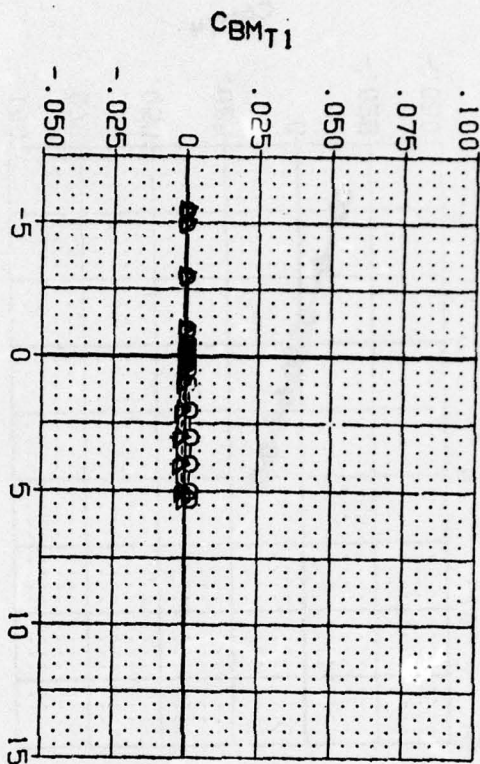


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH023) ☐ AECC W/A-CIA, CANARD CONTROL, BNICIT1
 (CXH024) ☐ DATA NOT AVAILABLE
 (CXH025) ☐ DATA NOT AVAILABLE
 (CXH026) ☐ AECC W/A-CIA, CANARD CONTROL, BNICIT1
 (CXH027) ☐ AECC W/A-CIA, CANARD CONTROL, BNICIT1

DCND1 DCND2 DCND3 DCND4
 -3.000 3.000 3.000 -3.000
 .500 .500 .500 .500
 1.000 -1.000 -1.000 1.000
 2.000 -2.000 -2.000 2.000
 5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\Phi_{HITAIL}=0$ $\Phi_{HICND}=0$
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CXH023) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(CXH024) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(CXH025) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(CXH026) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

(CXH027) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000 3.000 3.000 -3.000

1.500 -1.500 -1.500 1.500

2.000 -2.000 -2.000 2.000

5.000 -5.000 -5.000 5.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

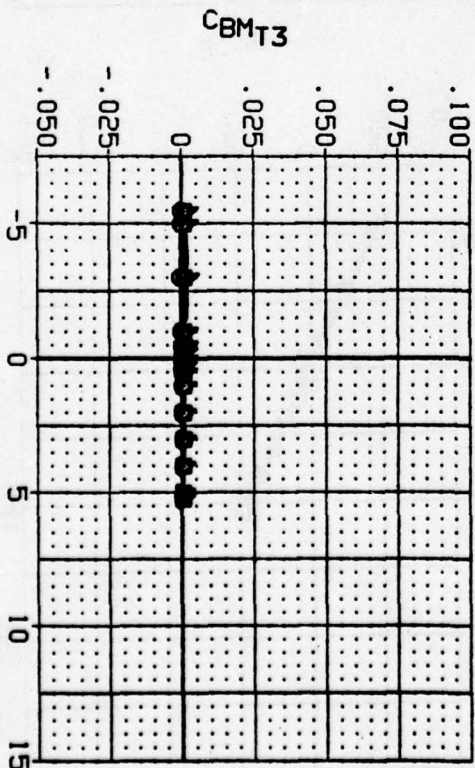
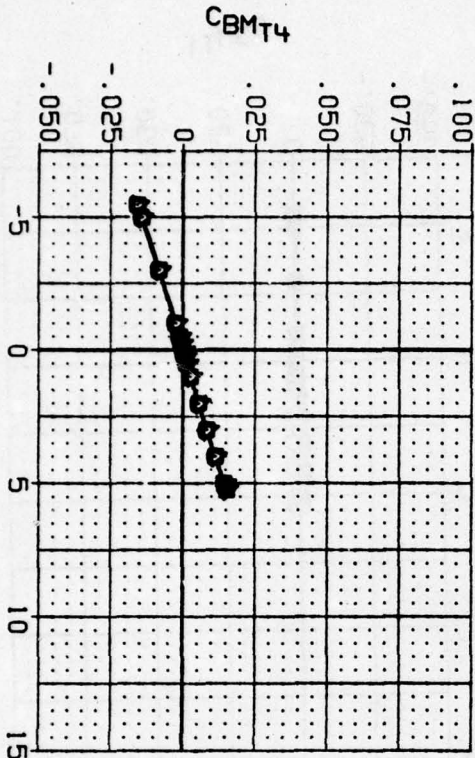
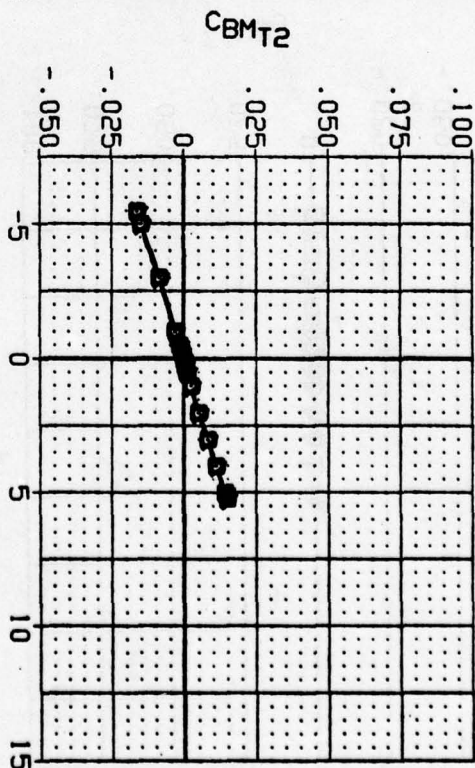
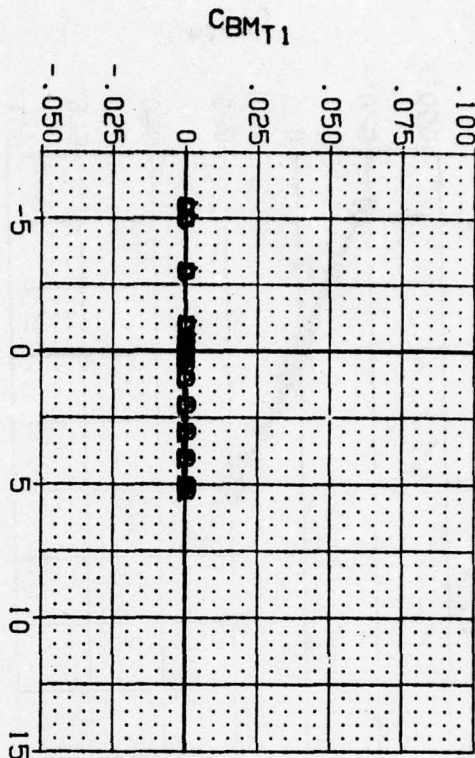
LBREF 5.0000 IN.

BRF 5.0000 IN.

YREF 26.0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

(C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

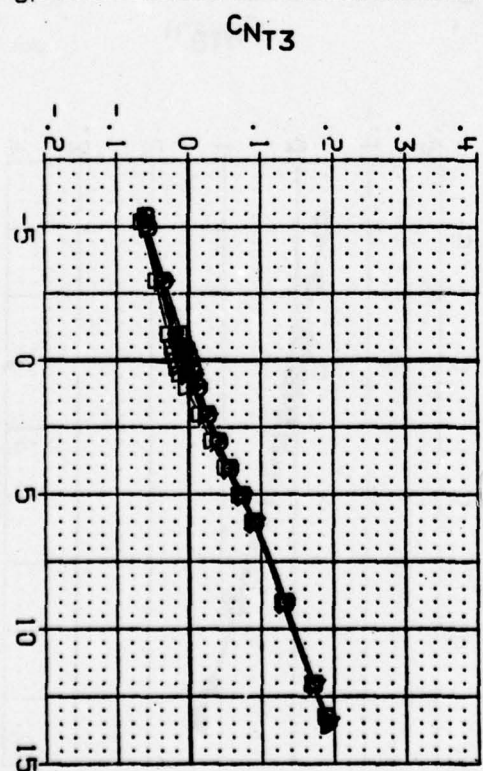
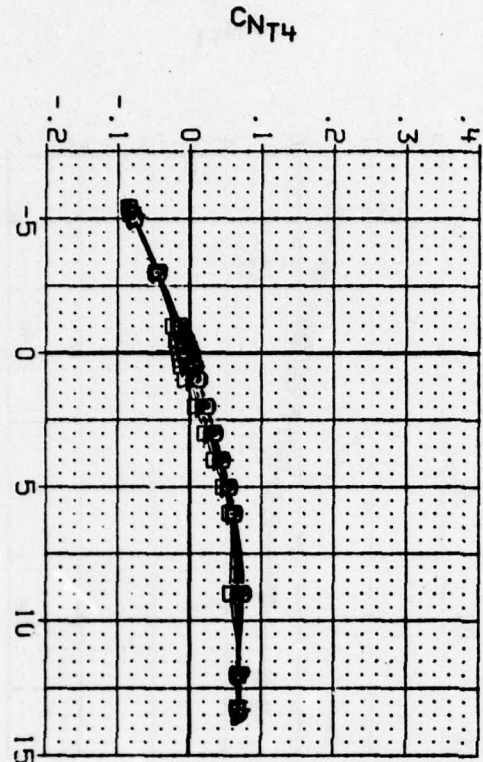
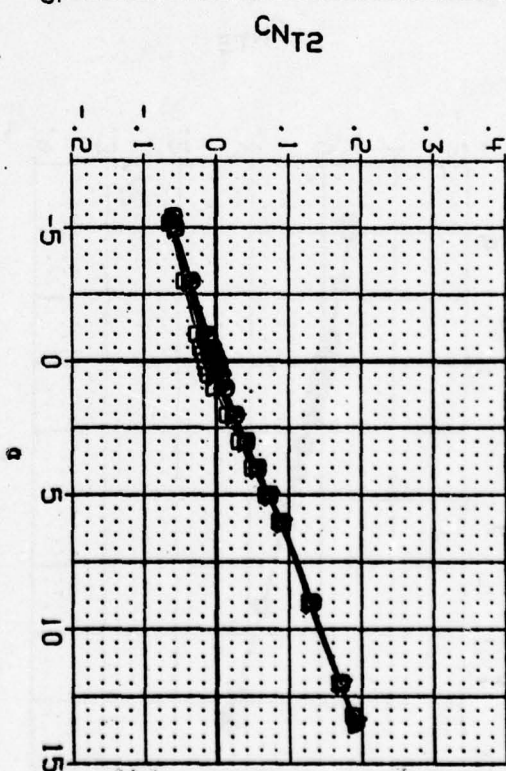
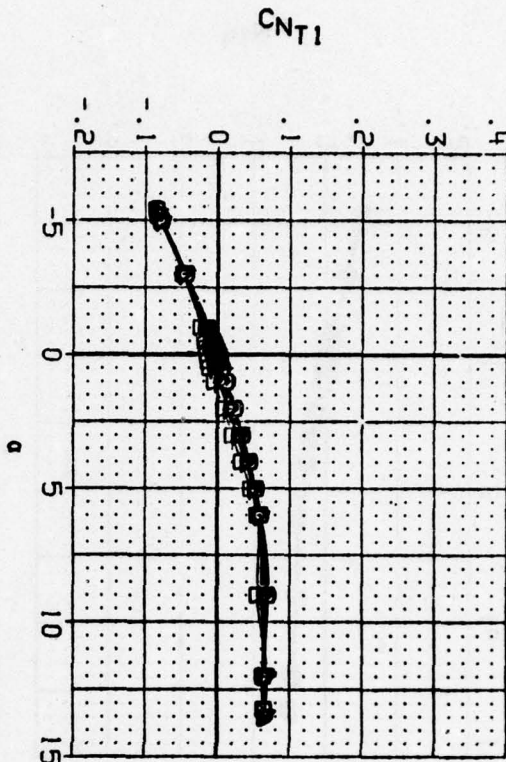
(AXH028)	○	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH029)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH030)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH031)	▽	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH032)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(AXH033)	◊	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000	-3.000	-3.000	-3.000
3.000	3.000	3.000	3.000
6.000	6.000	6.000	6.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION

SREF	19.6350	SO. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	

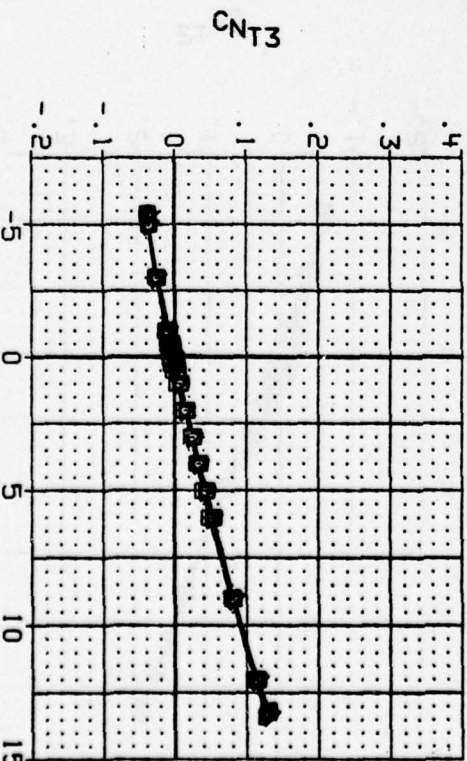
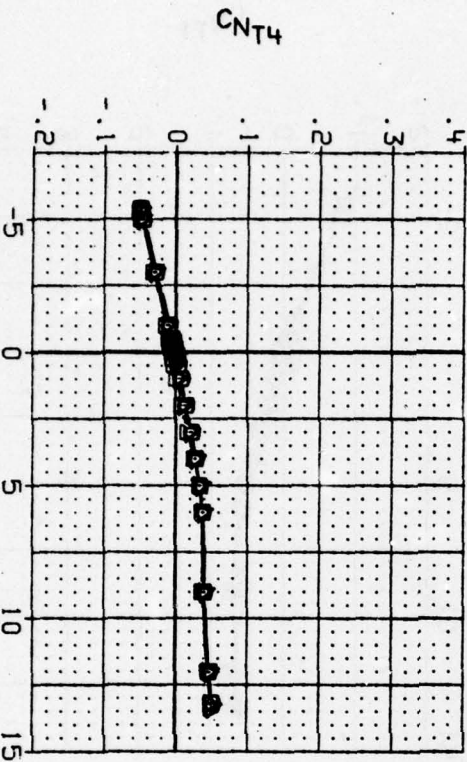
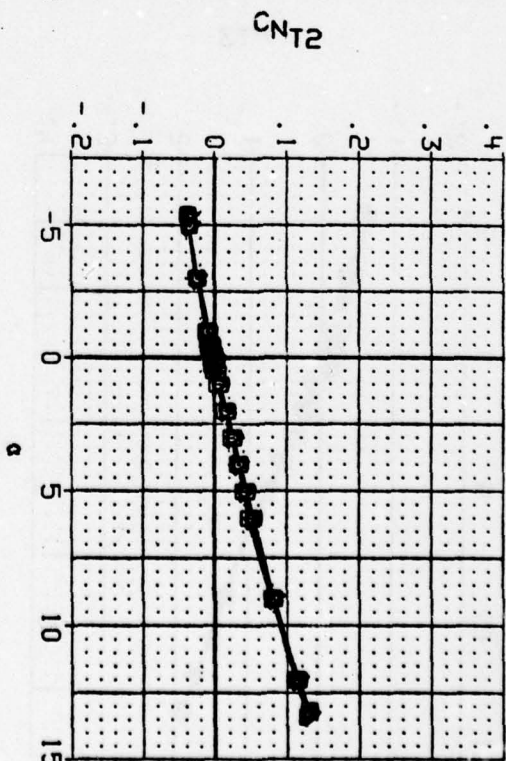
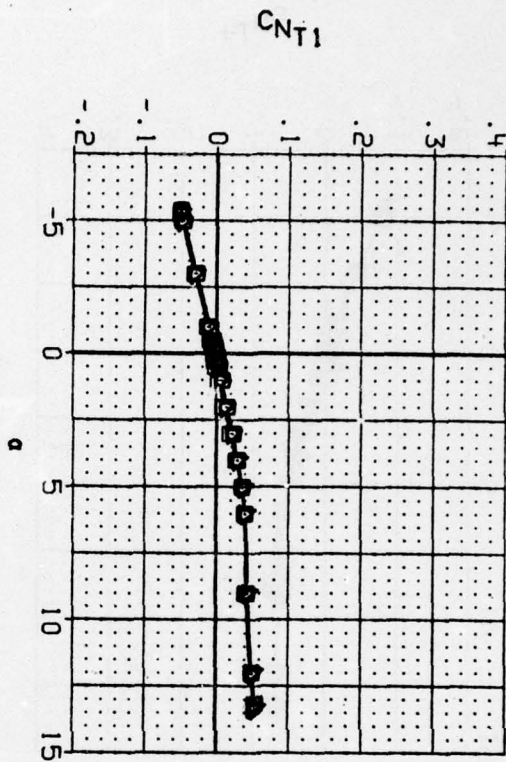


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH028) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH029) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH030) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH031) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH032) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1
 (AXH033) AEDC V1A-C1A, CANARD CONTROL, BNIC1T1

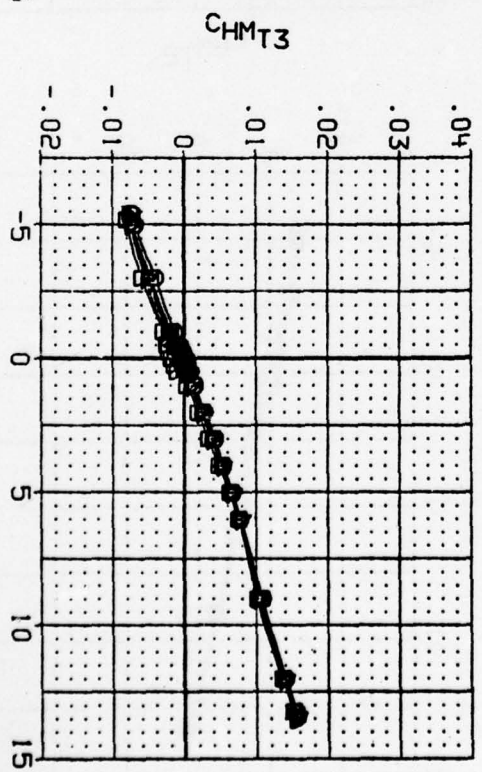
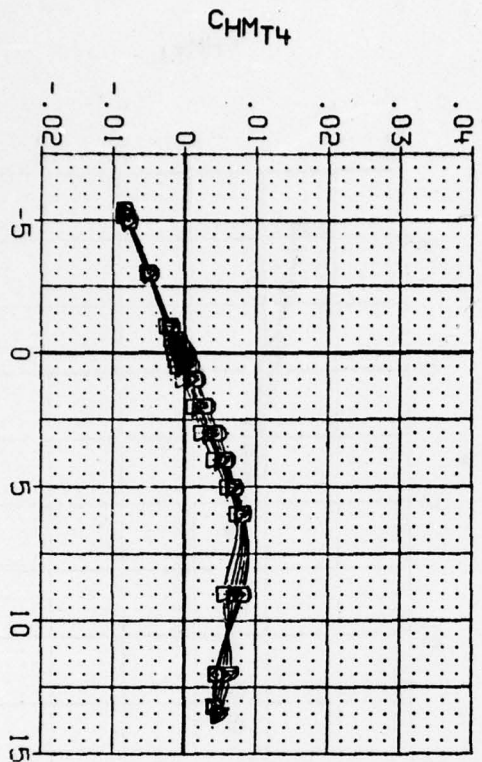
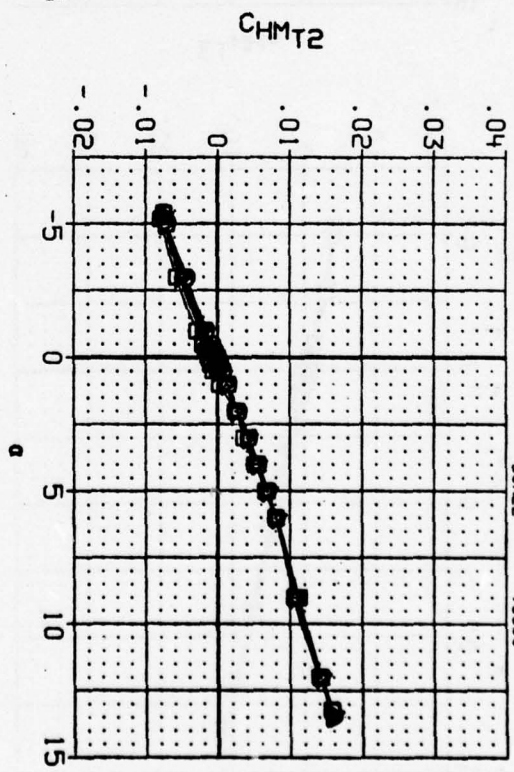
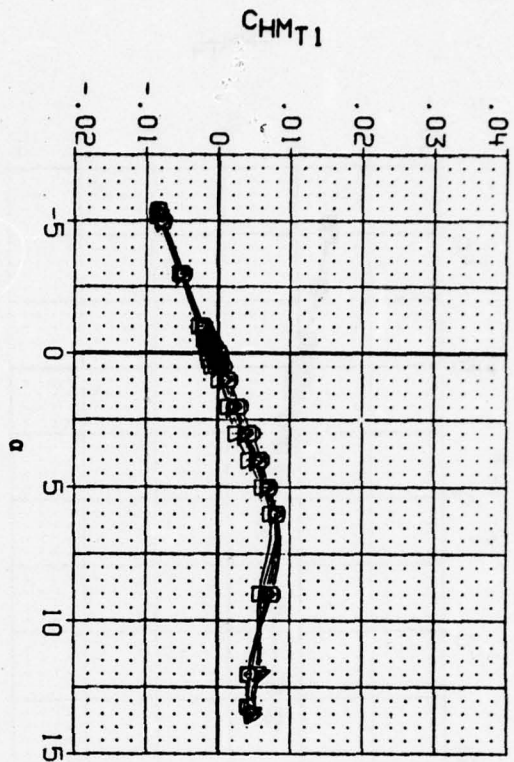
DCND1 DCND2 DCND3 DCND4 REFERENCE INFORMATION
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 .000 .000 .000 .000 LREF 5.0000 IN.
 3.000 3.000 3.000 3.000 BREF 26.0000 IN.
 6.000 6.000 6.000 6.000 YMRP .0000 IN.
 9.000 9.000 9.000 9.000 ZMRP .0000 IN.
 15.000 15.000 15.000 15.000 SCALE .0000 IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=45
 (B) MACH = 4.52

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH028)	□	AEDC V4A-CIA, CANARD CONTROL, BNICITI
(CXH029)	◇	AEDC V4A-CIA, CANARD CONTROL, BNICITI
(CXH030)	△	AEDC V4A-CIA, CANARD CONTROL, BNICITI
(CXH031)	▽	AEDC V4A-CIA, CANARD CONTROL, BNICITI
(CXH032)	○	AEDC V4A-CIA, CANARD CONTROL, BNICITI
(CXH033)	●	AEDC V4A-CIA, CANARD CONTROL, BNICITI

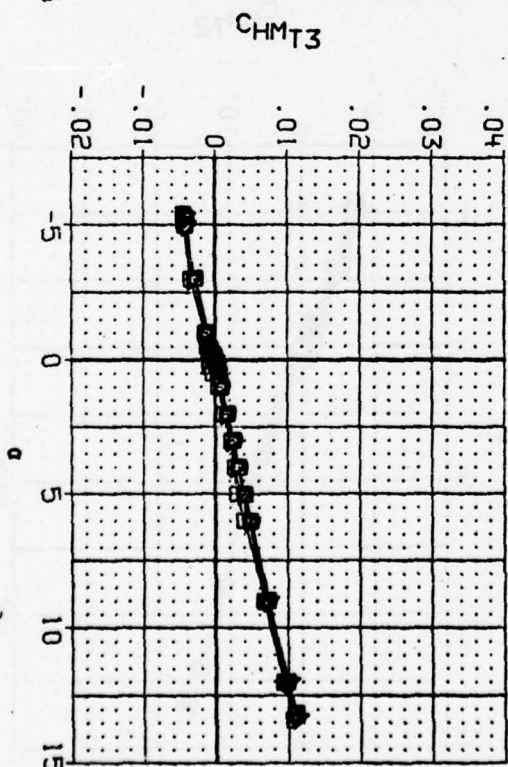
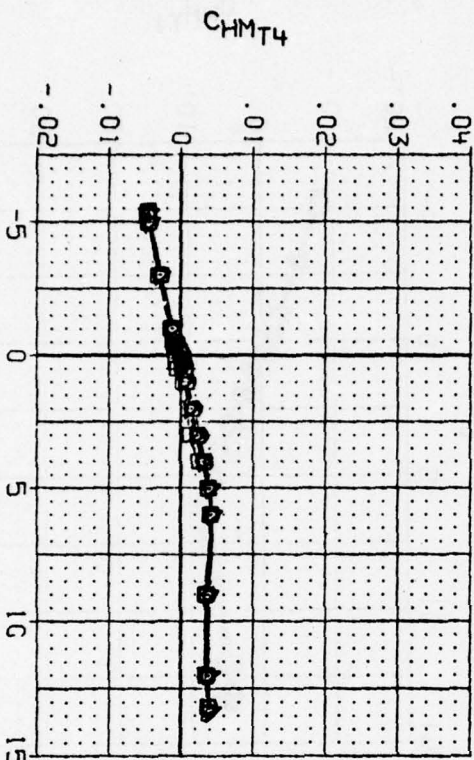
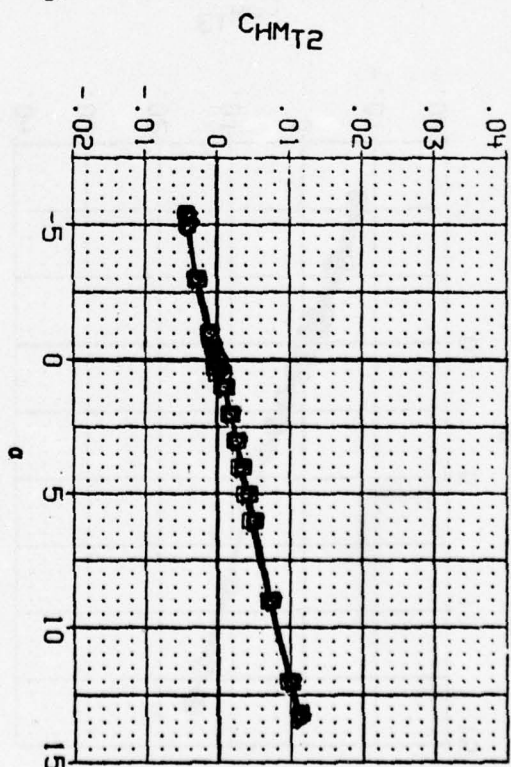
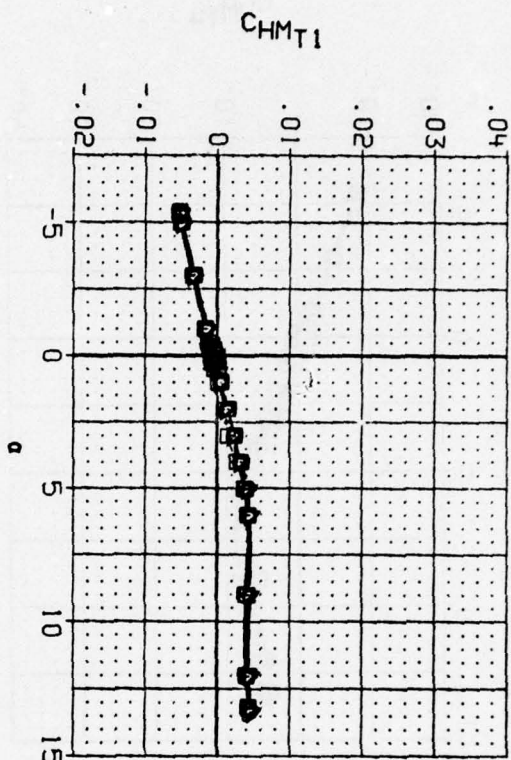
DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
-3.000	-3.000	-3.000	-3.000	SRET 19.6350 SQ.IN.
.000	.000	.000	.000	LREF 5.0000 IN.
3.000	3.000	3.000	3.000	BREF 5.0000 IN.
6.000	6.000	6.000	6.000	XREF 26.0000 IN.
9.000	9.000	9.000	9.000	YREF .0000 IN.
15.000	15.000	15.000	15.000	ZREF .0000 IN.
				SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=45
 (A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH028)	○	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH029)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH030)	△	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH031)	▽	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH032)	◇	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH033)	◇	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ.IN.
3.000	3.000	3.000	3.000	LREF 5.0000 IN.
3.000	3.000	3.000	3.000	BREF 5.0000 IN.
6.000	6.000	6.000	6.000	XRRP 26.0000 IN.
9.000	9.000	9.000	9.000	YRRP .0000 IN.
15.000	15.000	15.000	15.000	ZRRP .0000 IN.
				SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

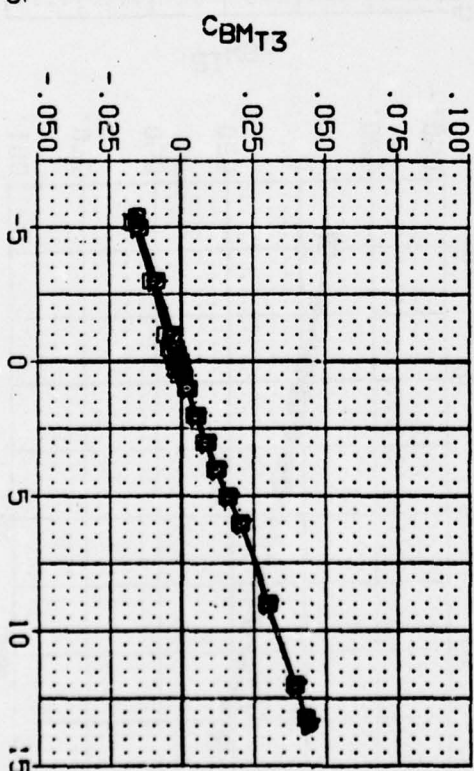
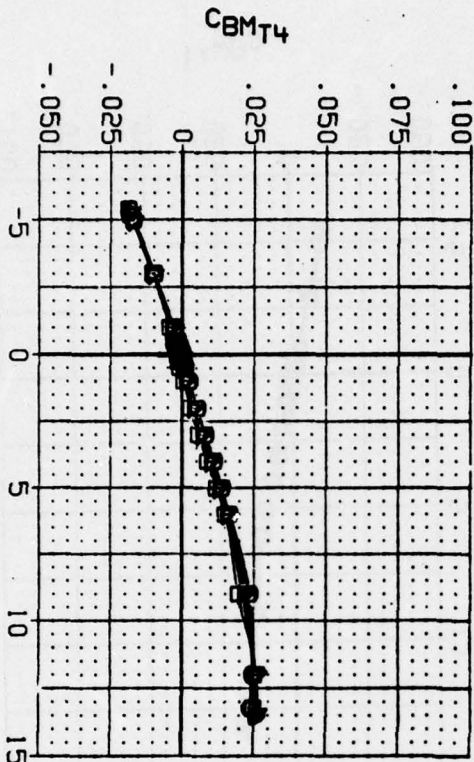
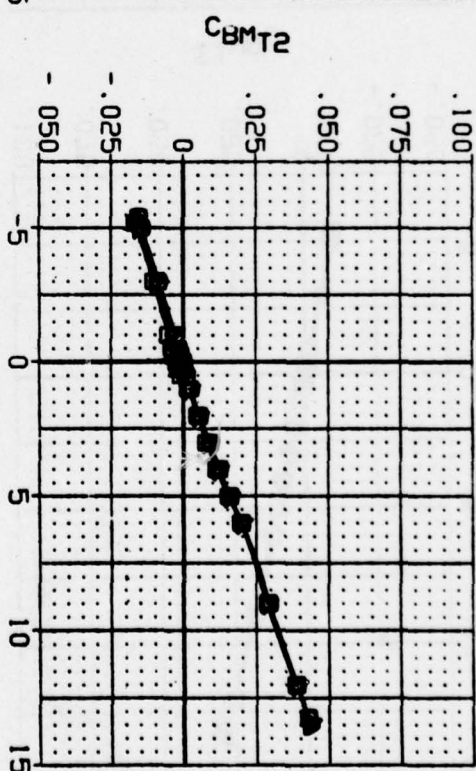
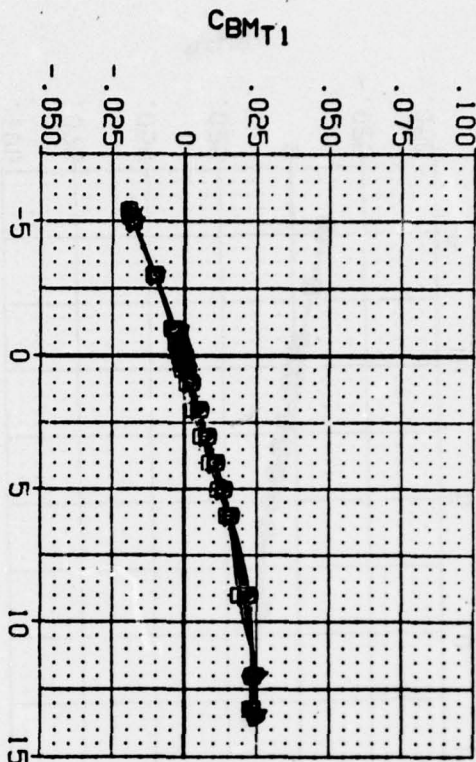
(CXH028)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH029)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH030)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH031)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH032)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
(CXH033)	□	AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4

-3.000	-3.000	-3.000	-3.000
3.000	3.000	3.000	3.000
6.000	6.000	6.000	6.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION

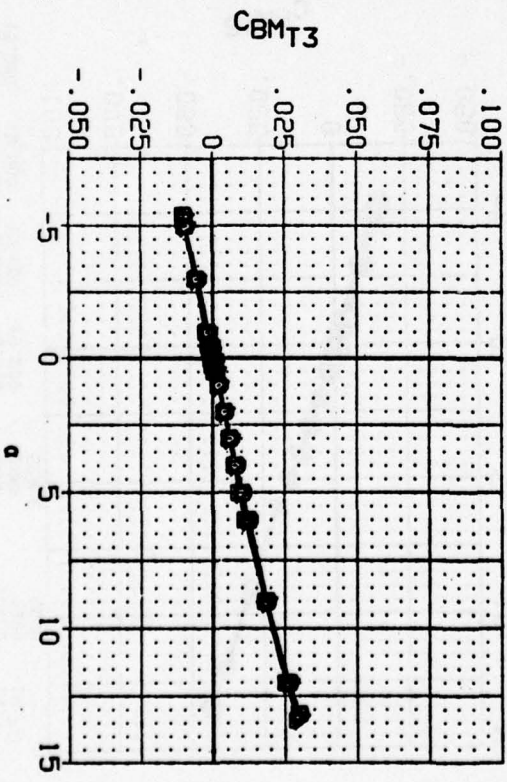
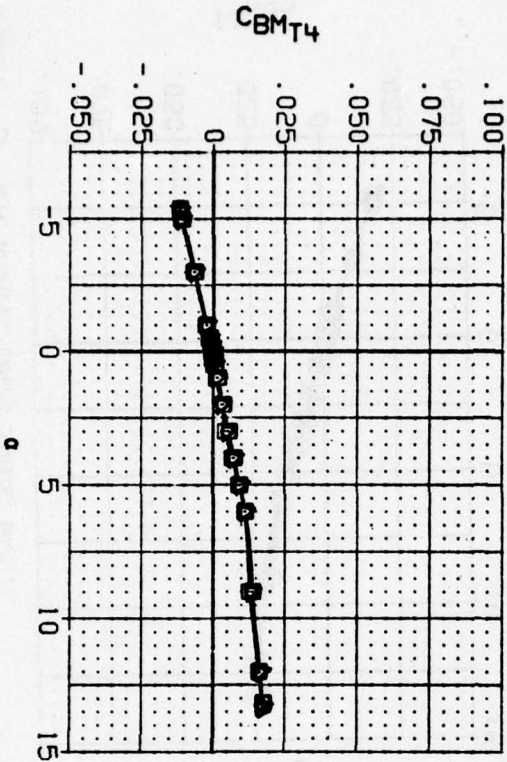
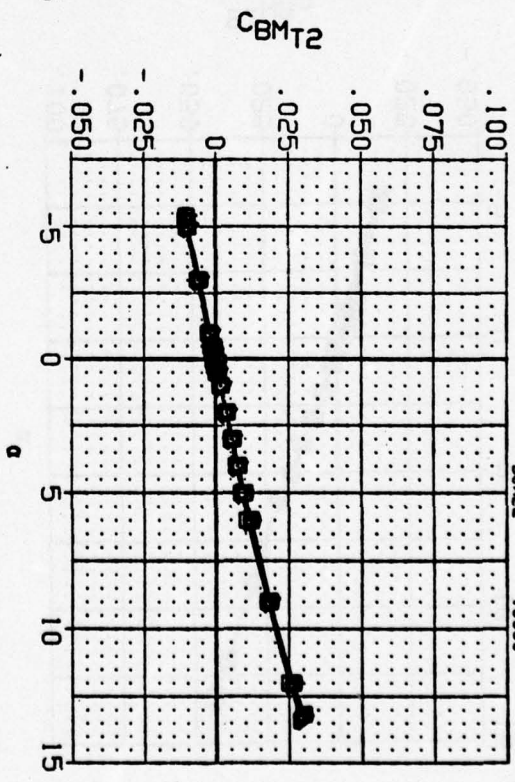
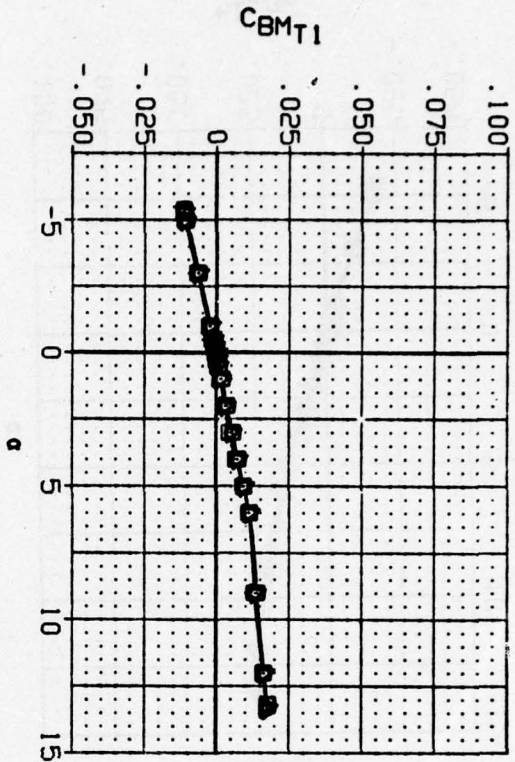
SREF	19.6350	50. IN.
LBREF	5.0000	IN.
BRREF	5.0000	IN.
YREF	26.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=45$
 (A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH028)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC111
(CXH029)	○	AEDC V41A-C1A, CANARD CONTROL, BNIC111
(CXH030)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC111
(CXH031)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC111
(CXH032)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC111
(CXH033)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
-3.000	-3.000	-3.000	-3.000	SREF 19.6350 SQ. IN.
.000	.000	.000	.000	LREF 5.0000 IN.
3.000	3.000	3.000	3.000	BREF 5.0000 IN.
6.000	6.000	6.000	6.000	WREF 26.0000 IN.
9.000	9.000	9.000	9.000	YREF .0000 IN.
15.000	15.000	15.000	15.000	ZREF .0000 IN.
				SCALE .0000

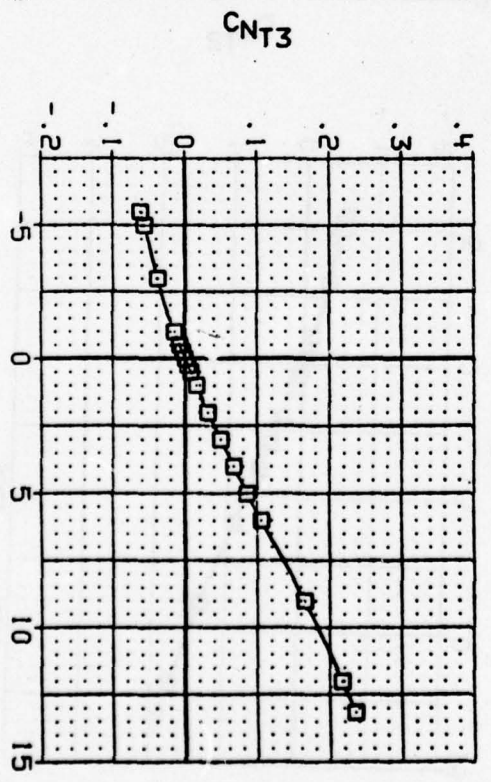
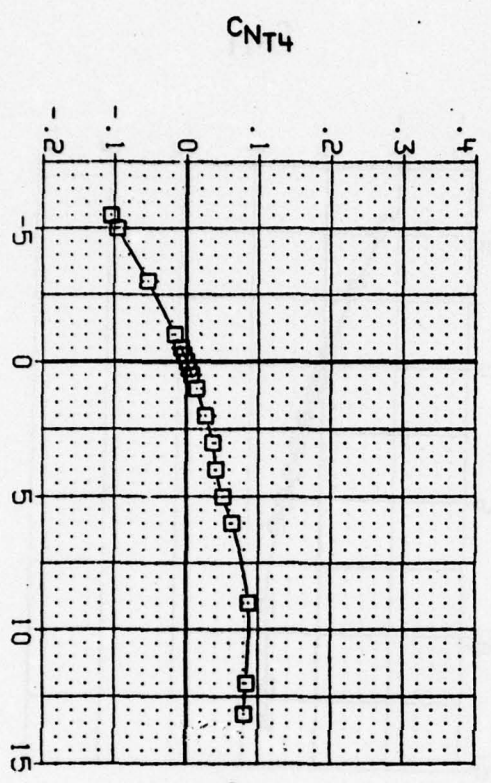
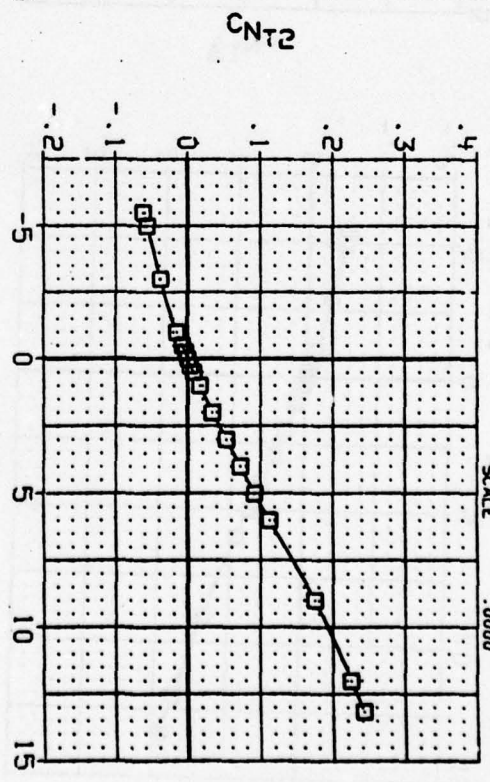
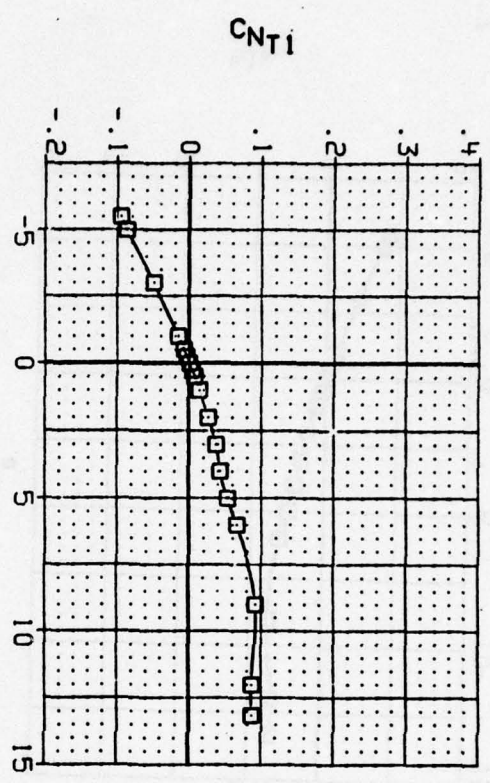


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH034) □ DATA NOT AVAILABLE
 (AXH035) □ AEDC VMA-CIA, CANARD CONTROL, BNIC111
 (AXH036) □ DATA NOT AVAILABLE
 (AXH037) △ DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 .000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000

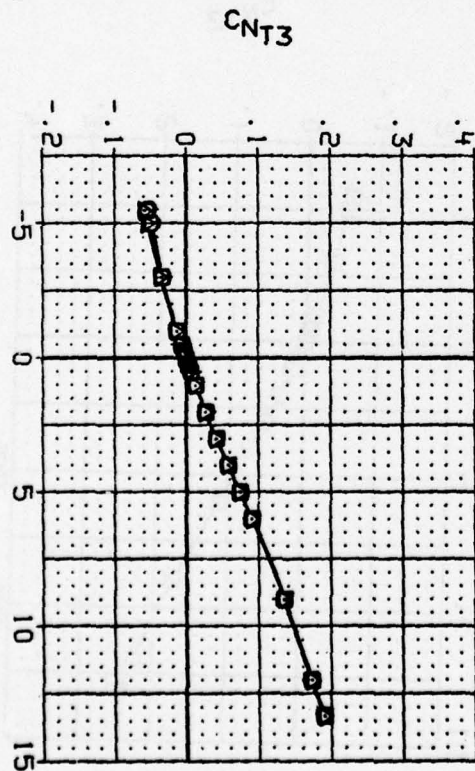
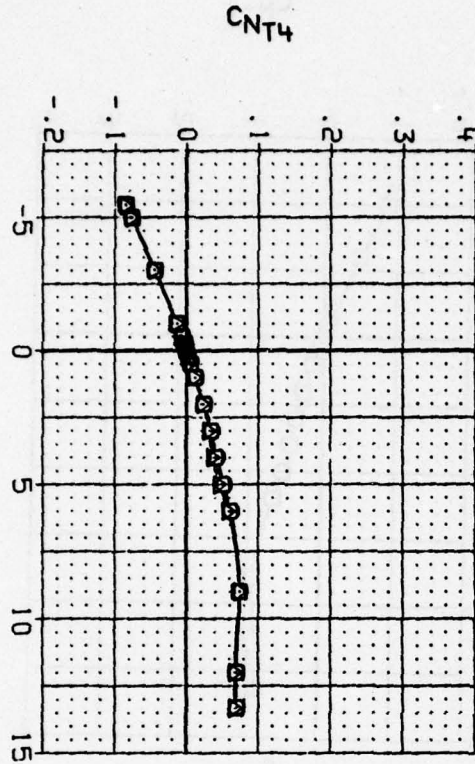
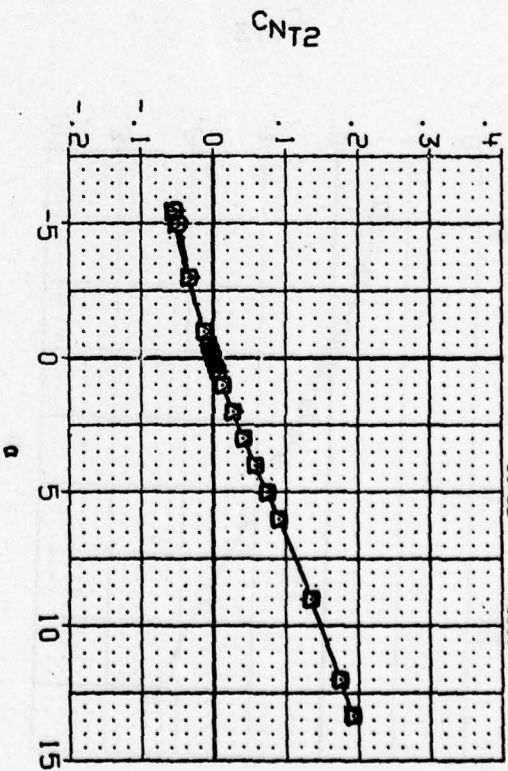
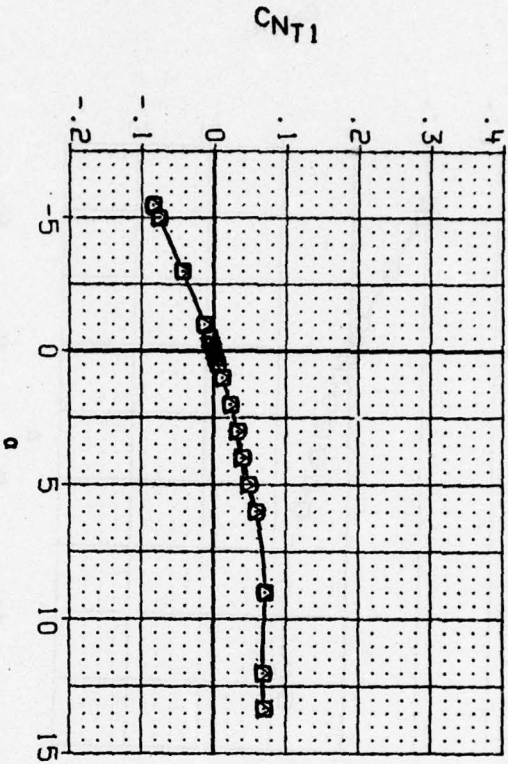


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAL=45$ $PHICND=0$
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH034) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH035) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (AXH036) ☐ DATA NOT AVAILABLE
 (AXH037) ☐ AEDC VMA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

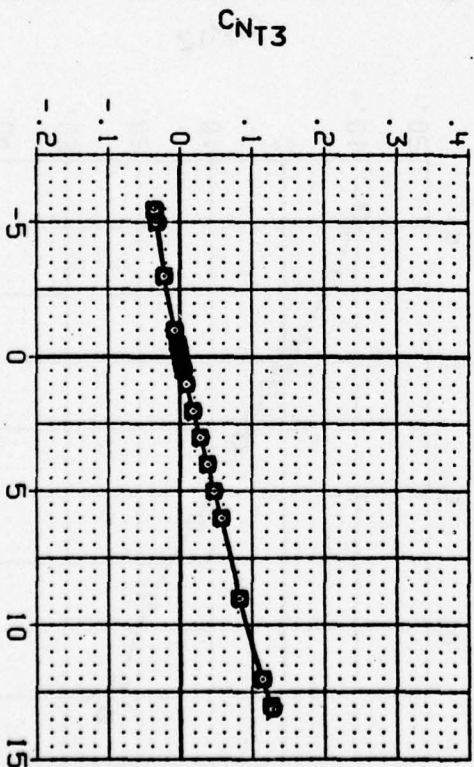
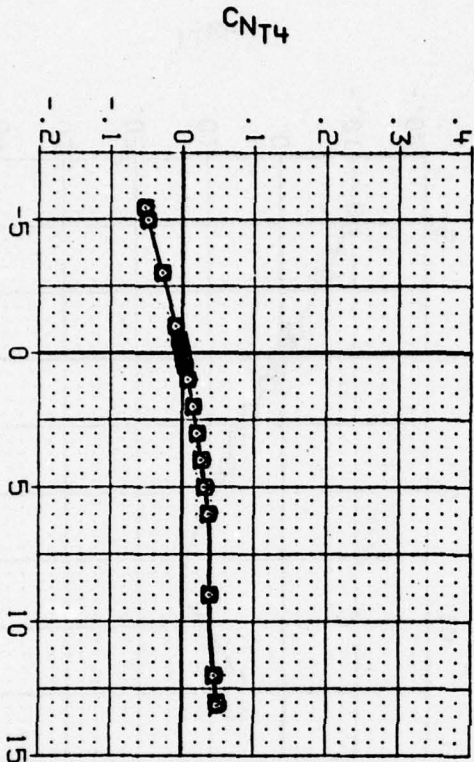
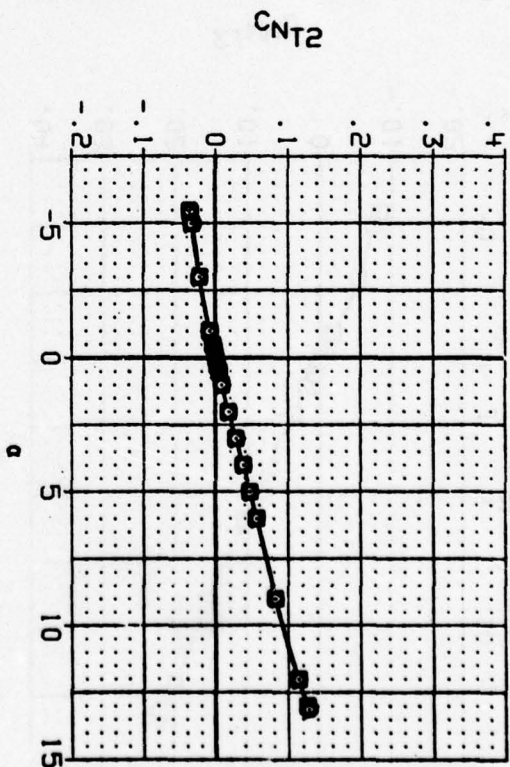
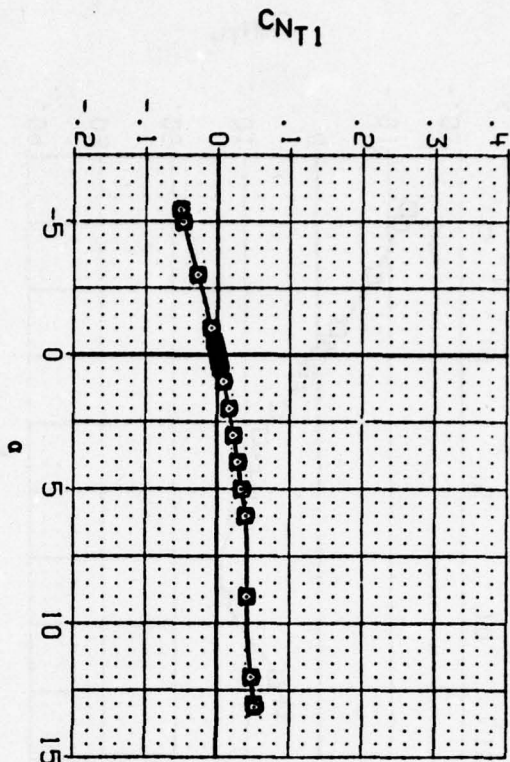
REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (B)MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH035) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (AXH035) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (AXH036) \square AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (AXH037) Δ AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 .000 DCND2 -3.000 DCND3 .000 DCND4 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000
 REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000

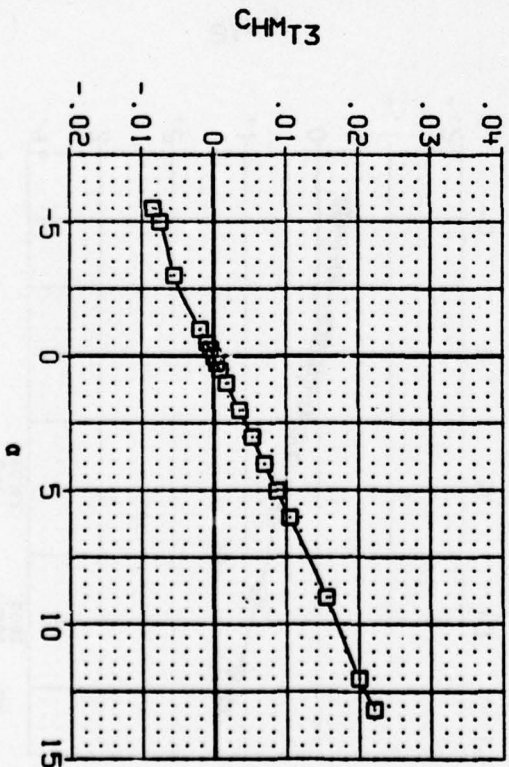
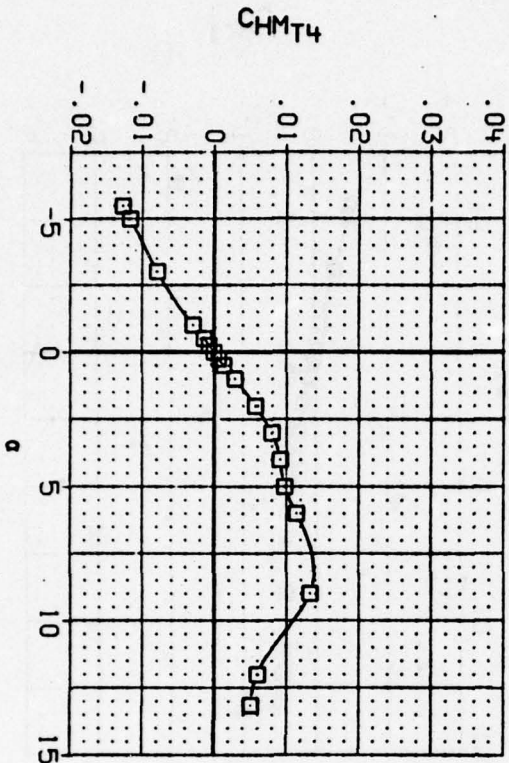
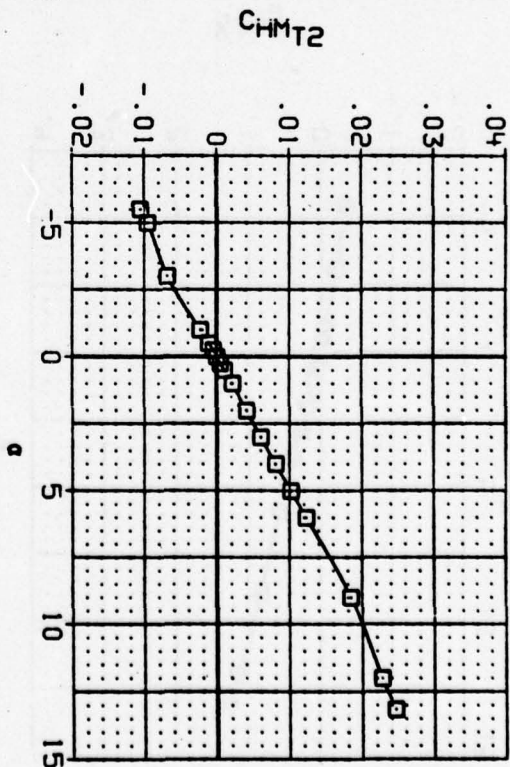
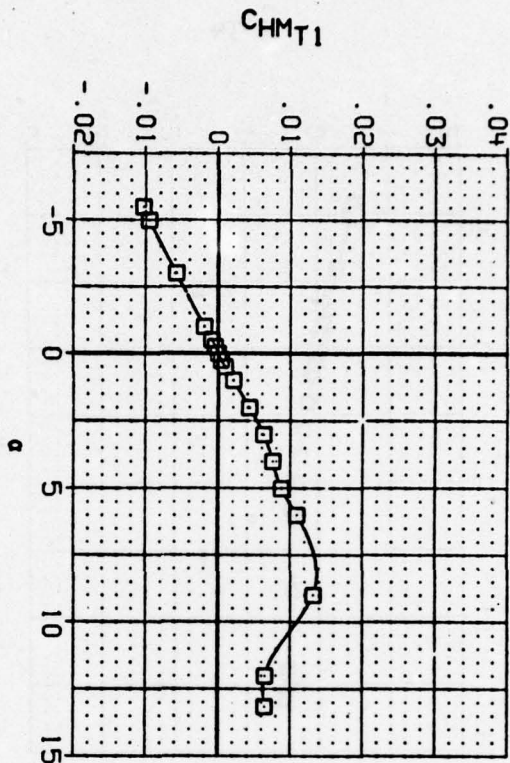


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (C) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH034) DATA NOT AVAILABLE
 (CXH035) AEDC W-1A-CIA, CANARD CONTROL, BNIC171
 (CXH036) DATA NOT AVAILABLE
 (CXH037) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

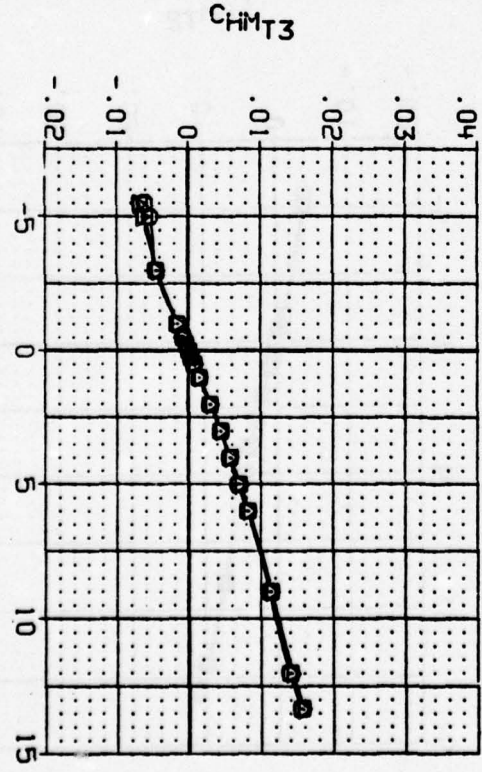
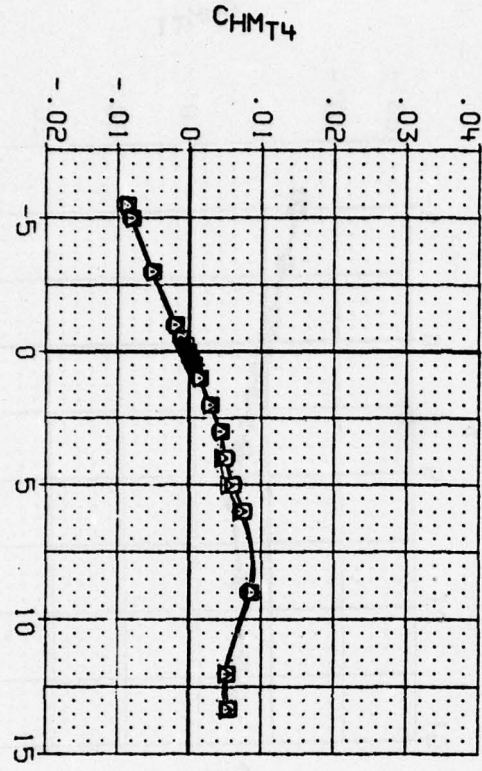
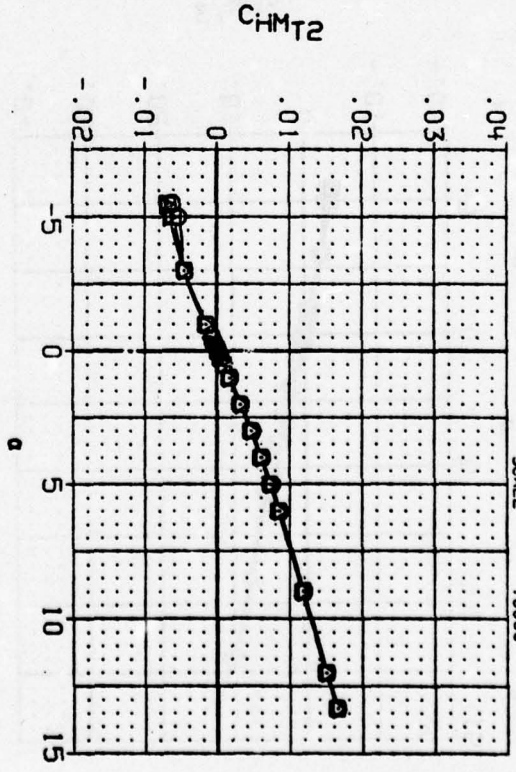
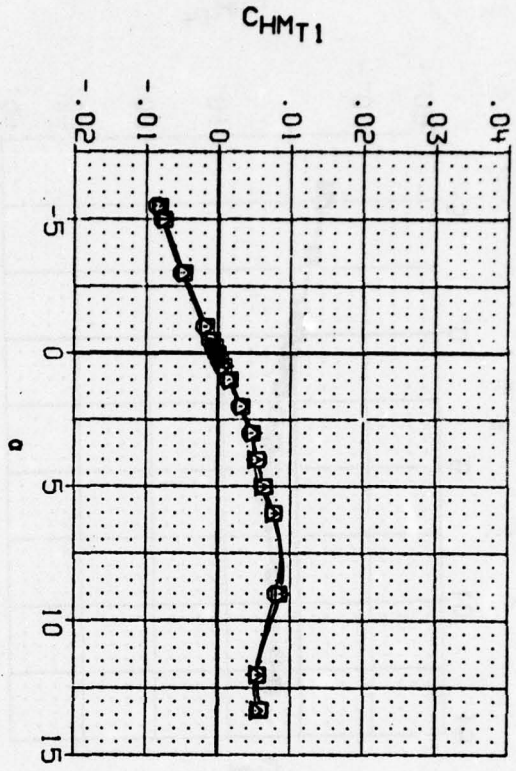
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 X-REF 25.0000 IN.
 Y-REF .0000 IN.
 Z-REF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH034) \square AEDC VMIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH035) \square AEDC VMIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH036) \square DATA NOT AVAILABLE
 (CXH037) Δ AEDC VMIA-CIA, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4 REFERENCE INFORMATION
 .000 -3.000 .000 -3.000 SREF 19.6350 SQ. IN.
 .000 .000 .000 .000 LREF 5.0000 IN.
 .000 1.000 .000 1.000 BREF 5.0000 IN.
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 SCALE .0000

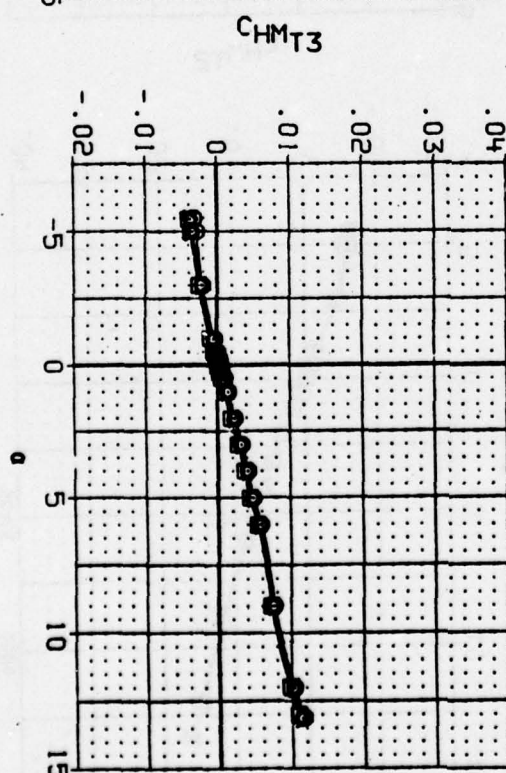
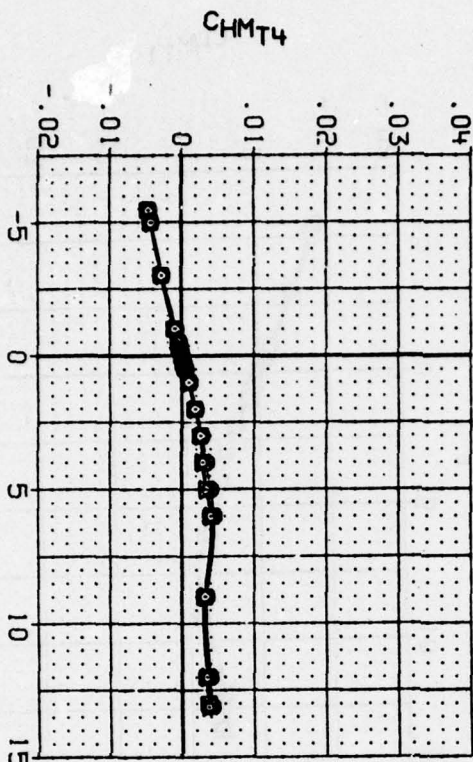
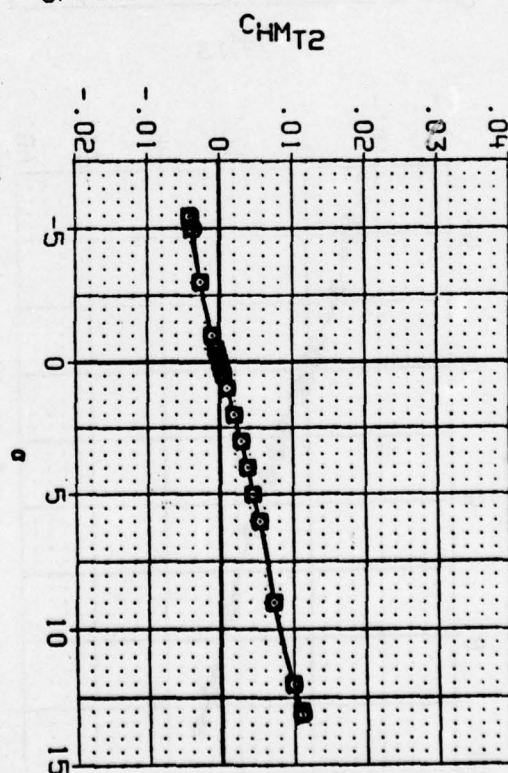
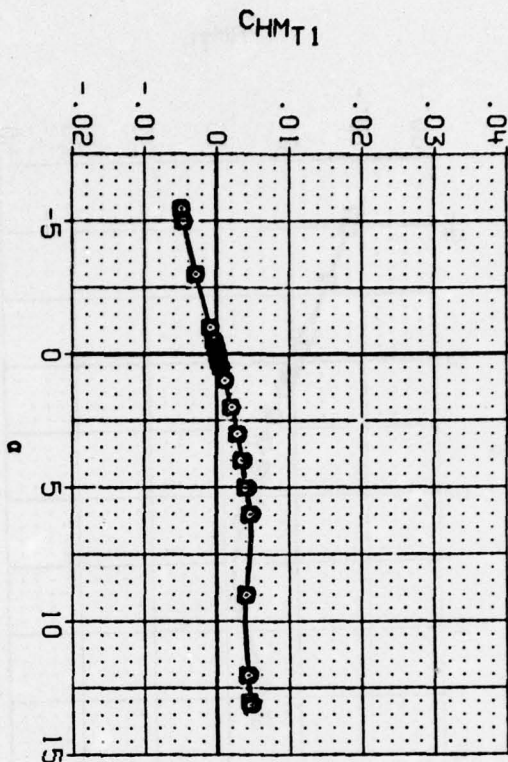


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH034) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH035) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH036) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH037) \square AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
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 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

REFERENCE INFORMATION
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000

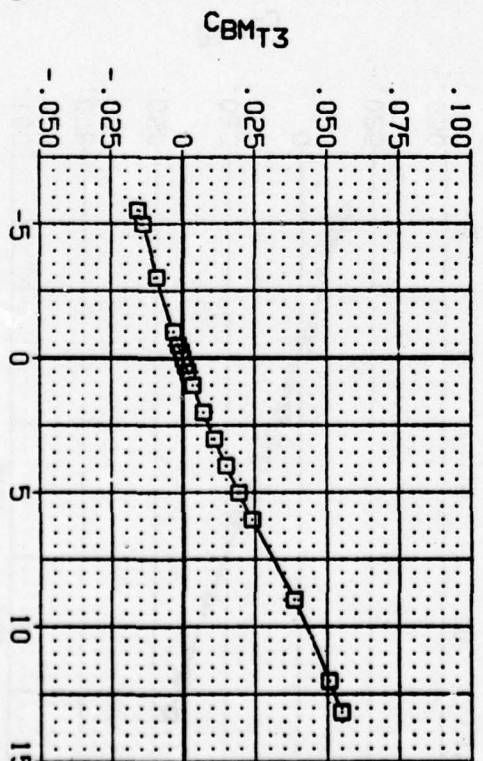
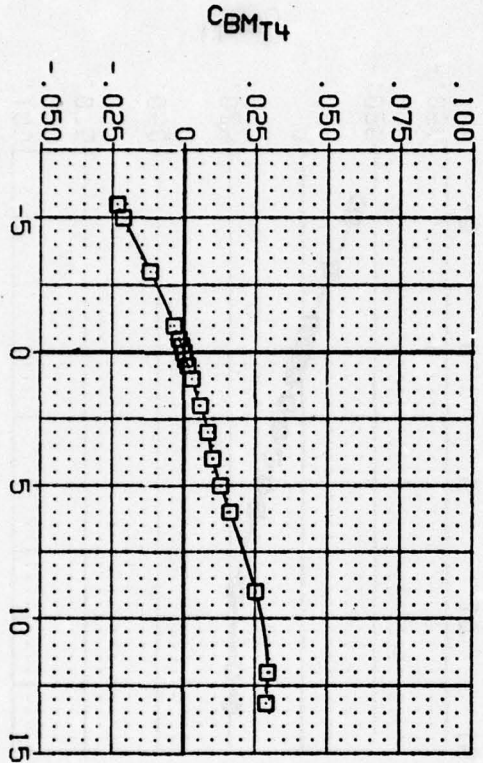
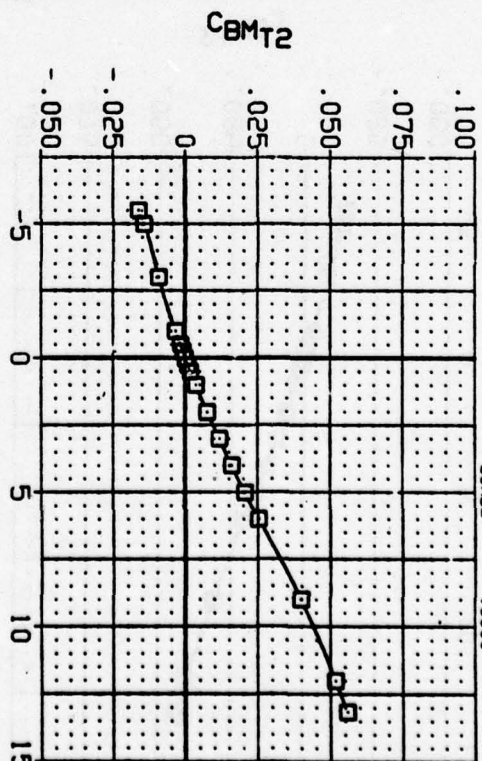
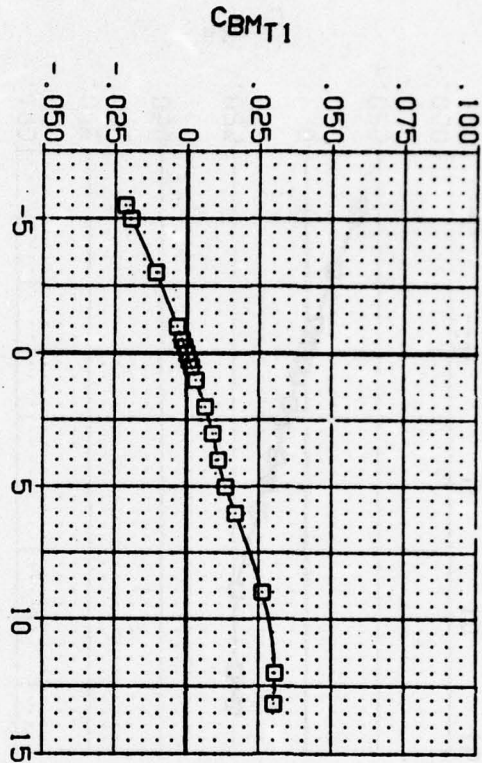


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=45 PHICND=0
 (C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH031) DATA NOT AVAILABLE
 (CXH033) AEDC WIA-CIA, CANARD CONTROL, BNIC11
 (CXH036) DATA NOT AVAILABLE
 (CXH037) DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
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 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000

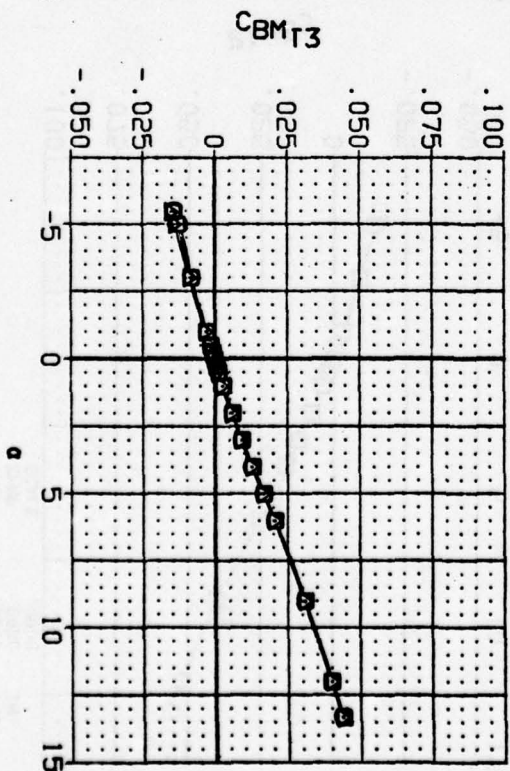
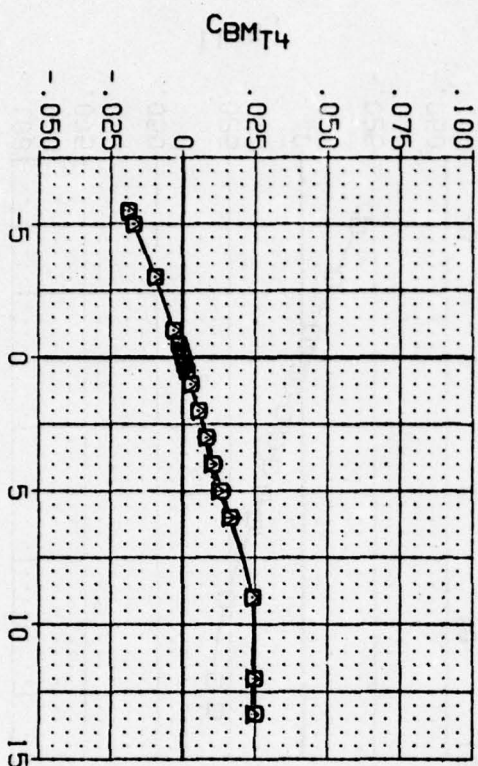
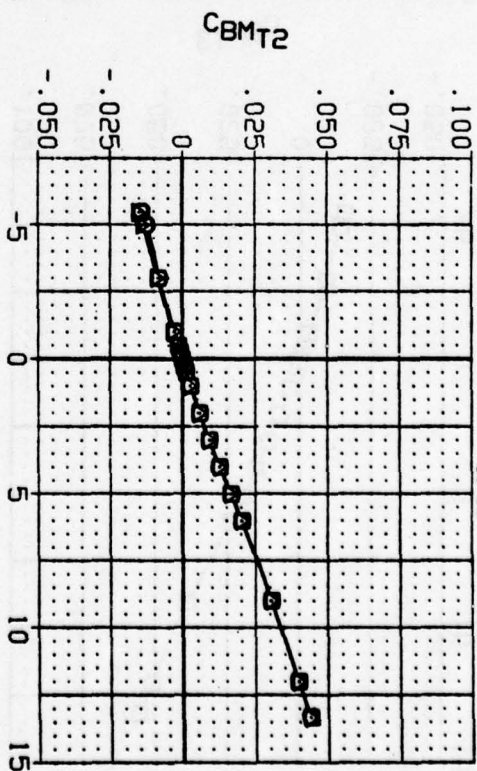
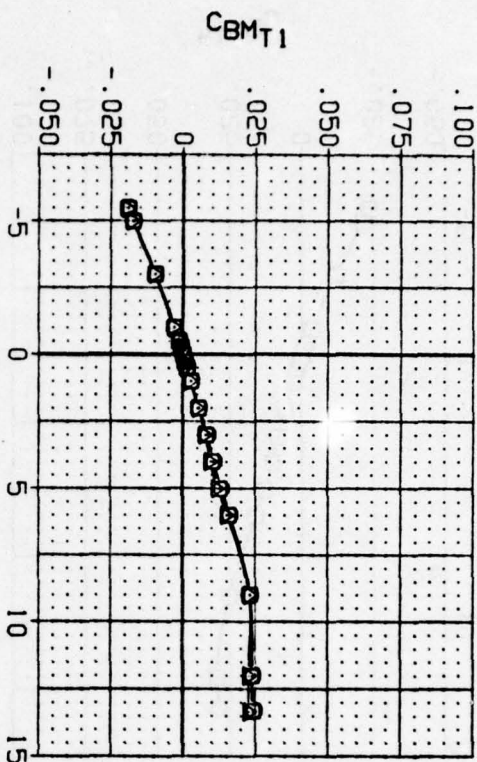
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 LREF 5.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH031) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH033) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1
 (CXH036) DATA NOT AVAILABLE
 (CXH037) AEDC W1A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
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 .000 .000 .000 .000
 .000 1.000 .000 1.000
 .000 3.000 .000 3.000
 REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (B) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(C)H034) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITI

(C)H035) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITI

(C)H036) \square AEDC WJIA-CIA, CANARD CONTROL, BNICITI

(C)H037) \triangle AEDC WJIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

.000 -3.000 .000 -3.000

.000 .000 .000 1.000

.000 1.000 .000 3.000

.000 .000 .000 3.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

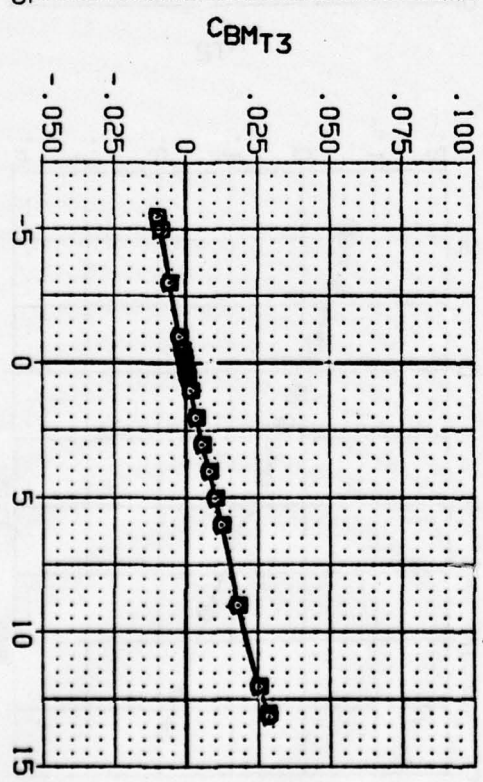
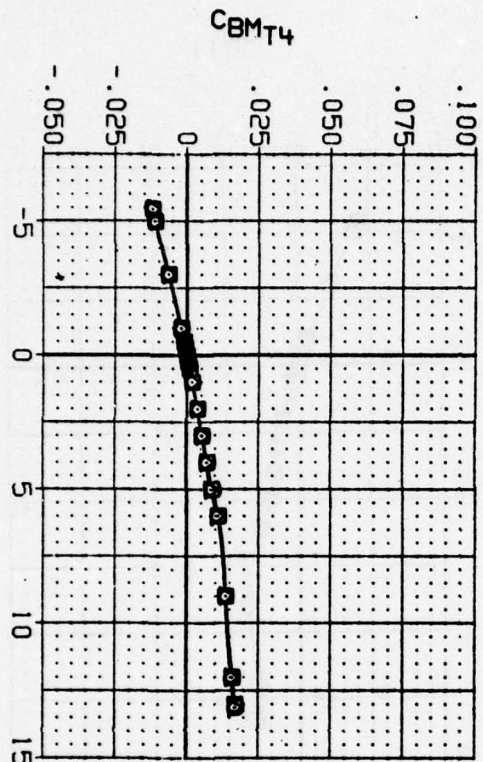
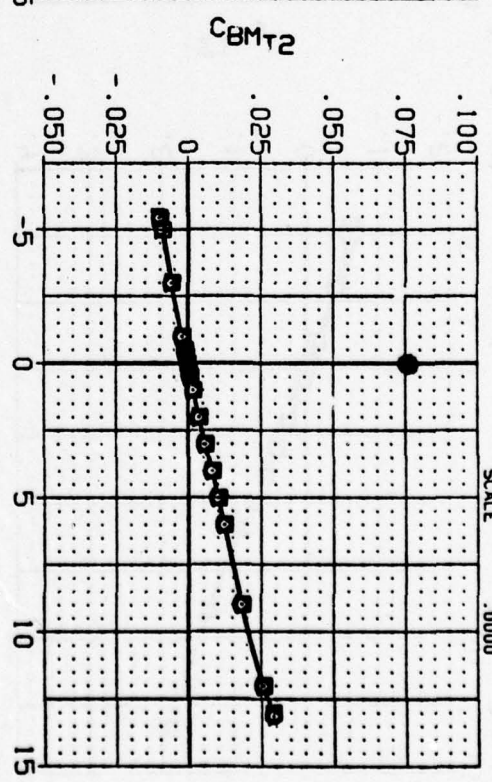
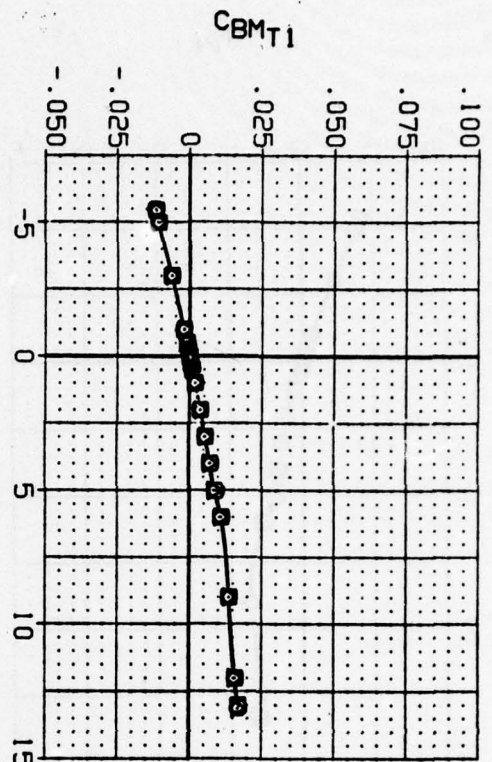
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SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

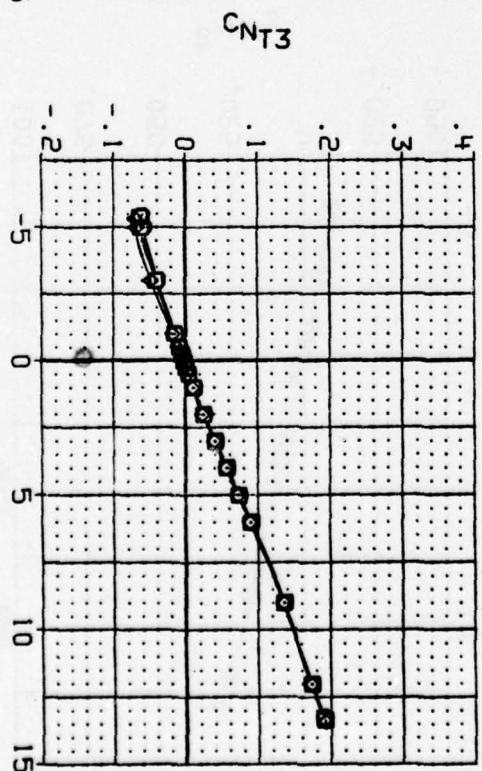
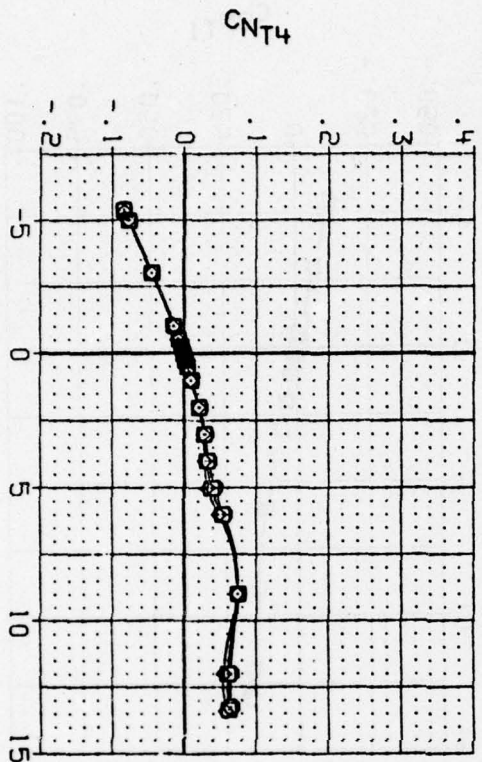
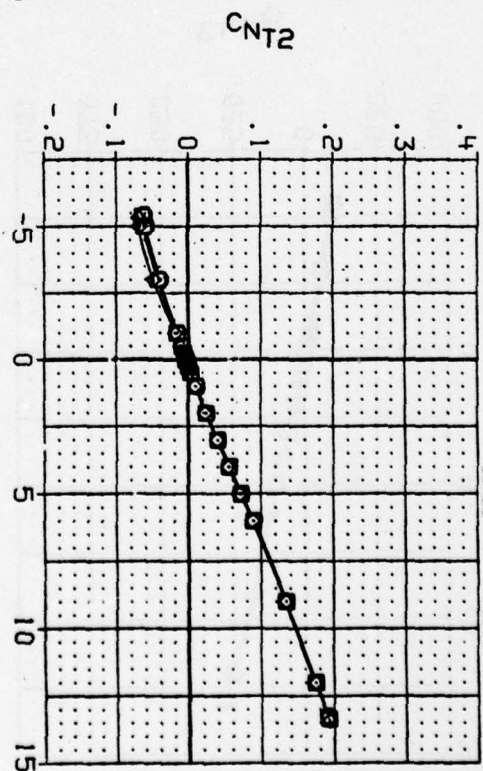
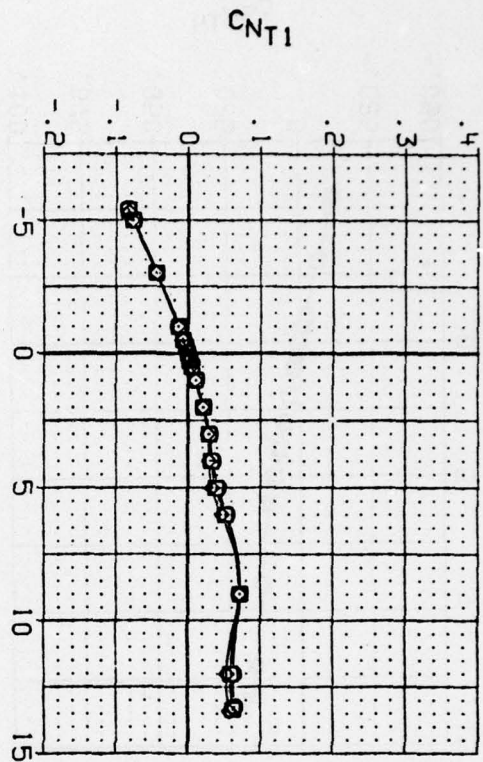
PHITAL=45 PHICND=0

(C)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH039) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH039) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH040) \square AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
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 ZREF .0000 IN.
 SCALE .0000



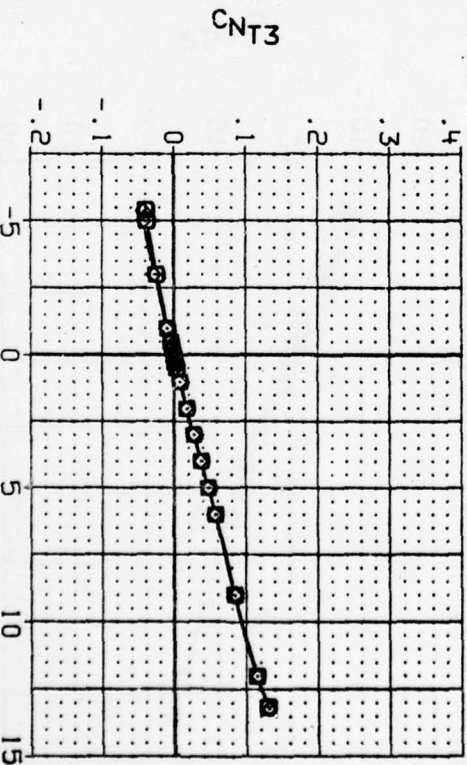
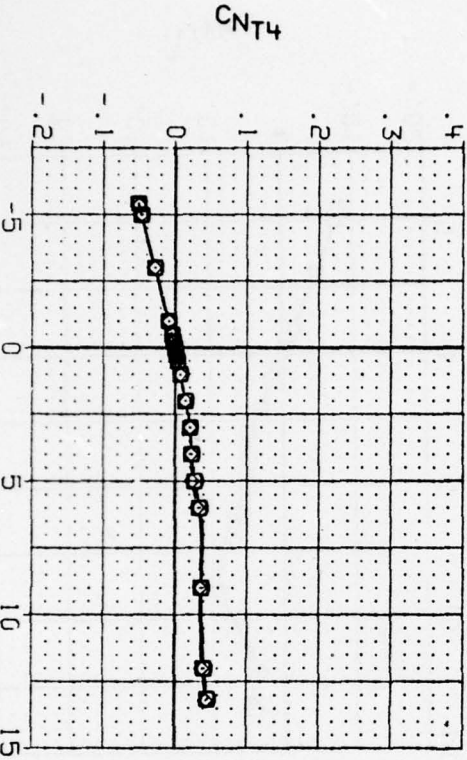
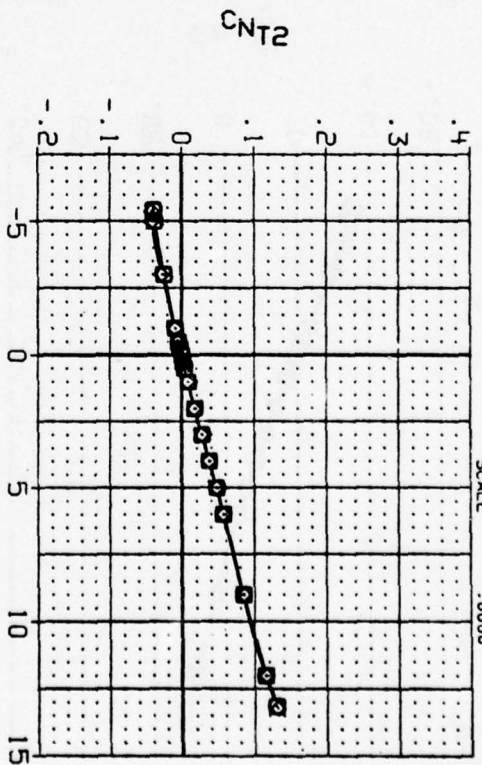
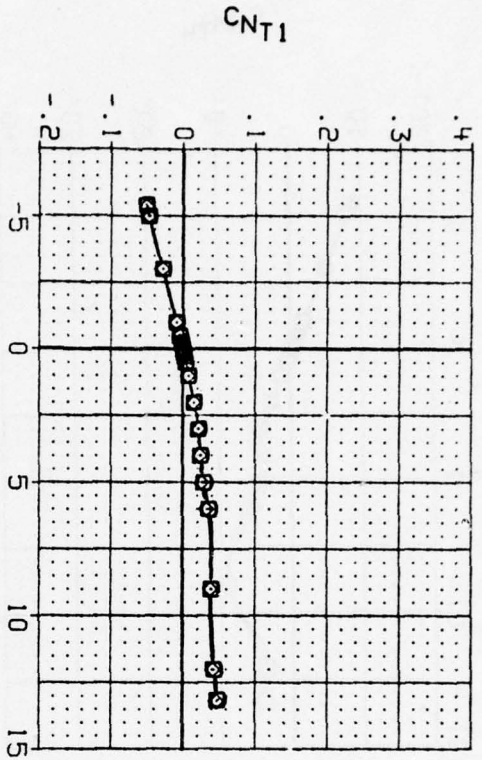
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=45 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL

CONFIGURATION DESCRIPTION
 (AXH038) AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH039) AEDC V41A-C1A, CANARD CONTROL, BNIC111
 (AXH040) AEDC V41A-C1A, CANARD CONTROL, BNIC111

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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 SCALE .0000

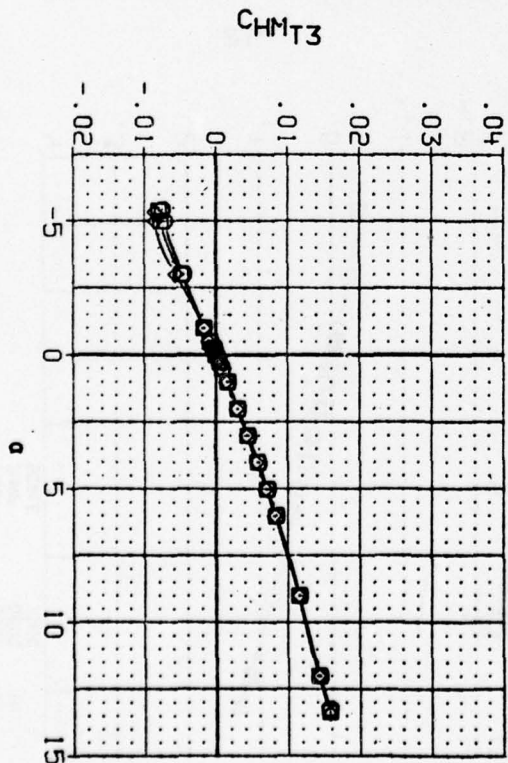
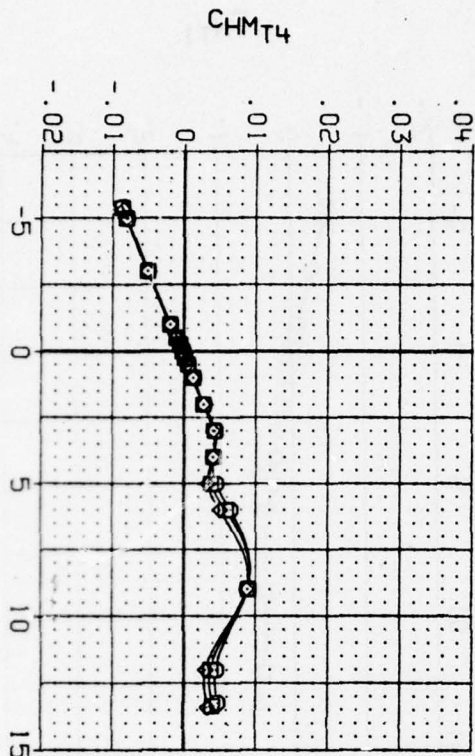
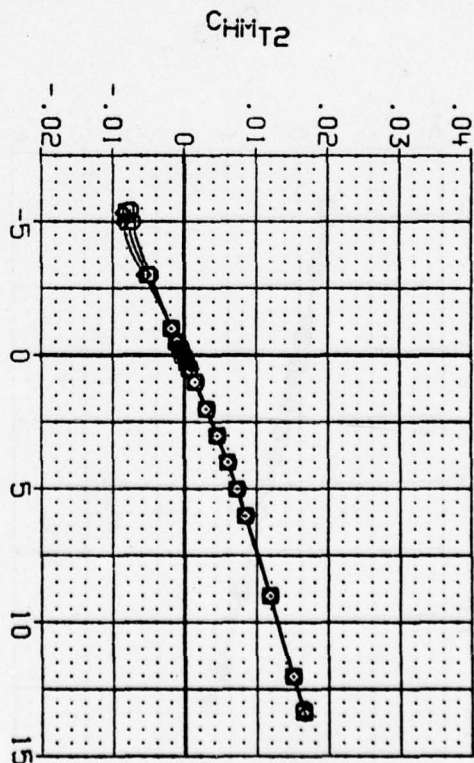
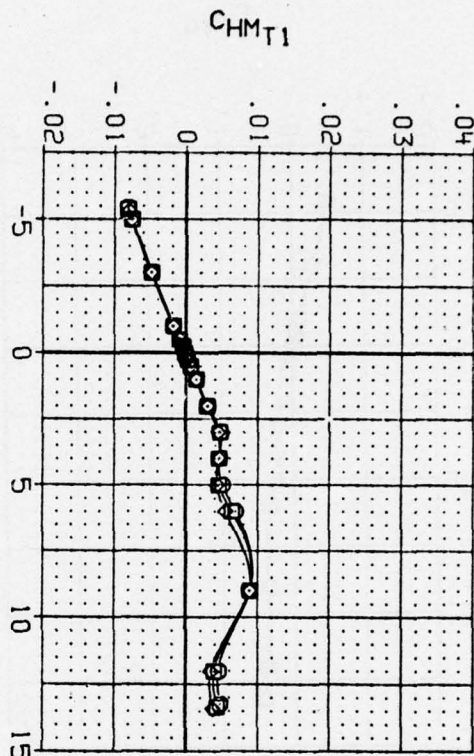


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH038) \square AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH039) \square AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1
 (CXH040) \square AEDC WJIA-CIA, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
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 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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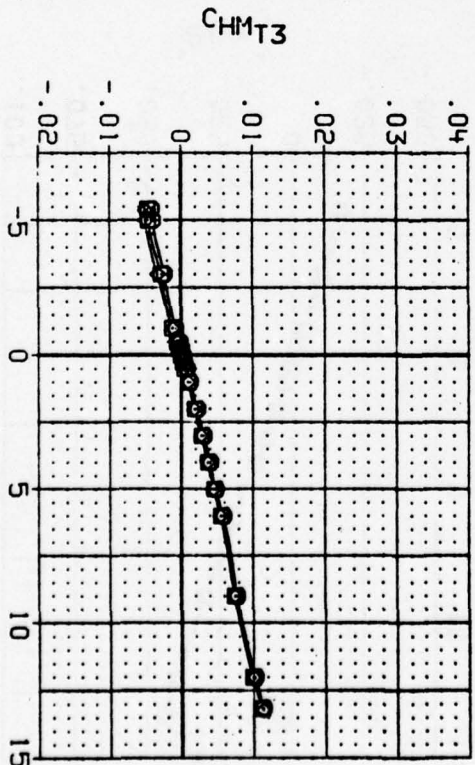
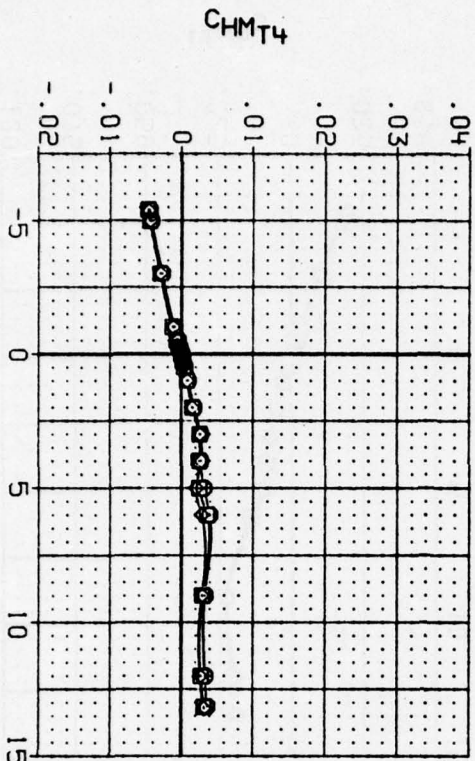
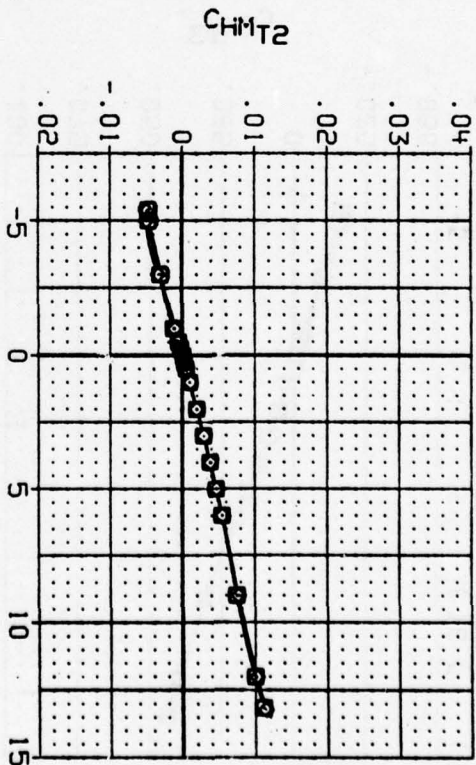
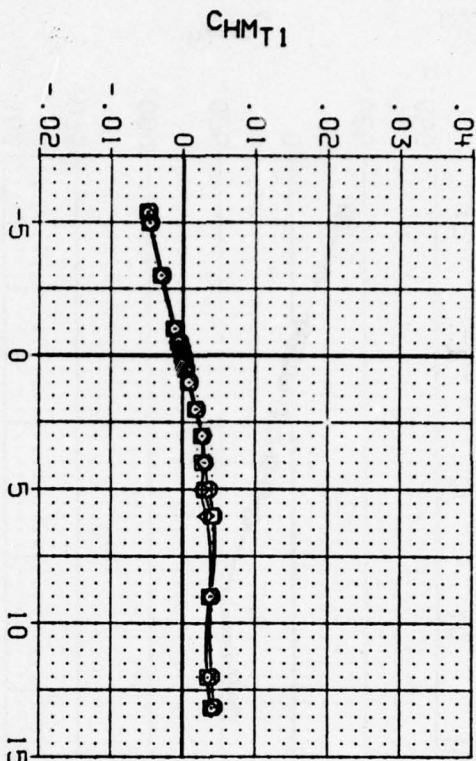


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH038) ☐ AEDC VJIA-CIA, CANARD CONTROL, BNICITI
 (CXH039) ☐ AEDC VJIA-CIA, CANARD CONTROL, BNICITI
 (CXH040) ☐ AEDC VJIA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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 BREF 5.0000 IN.
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 SCALE .0000

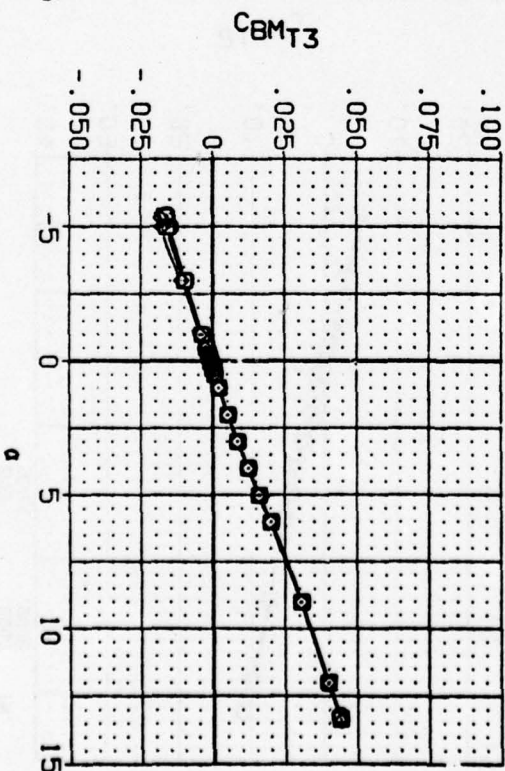
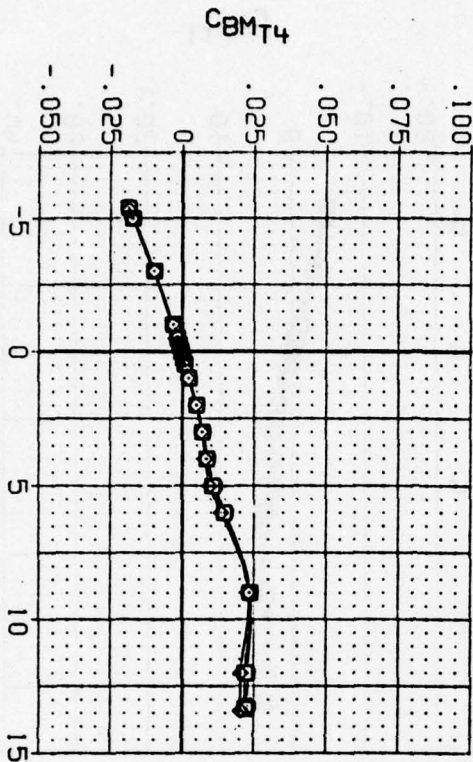
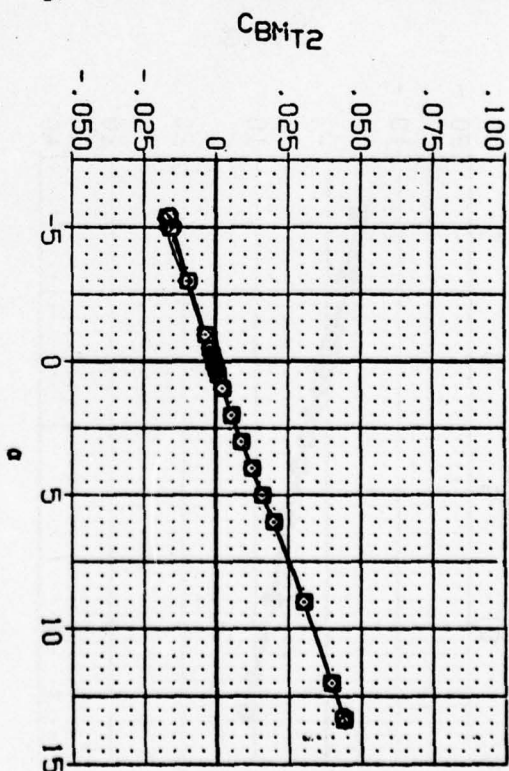
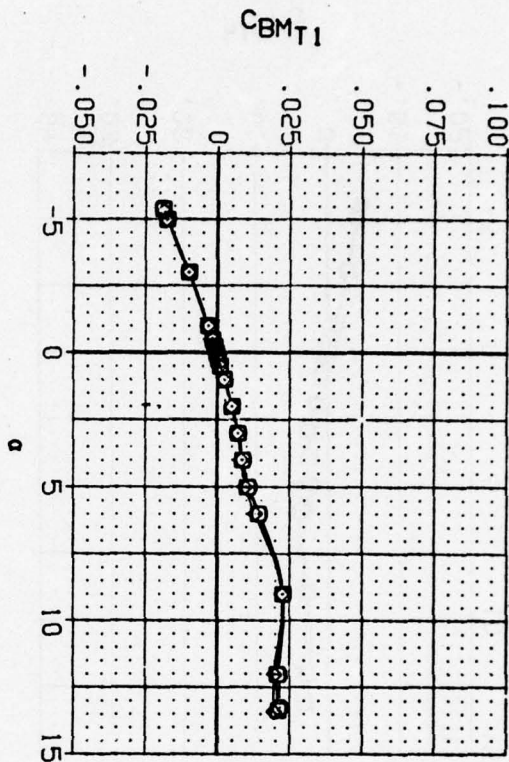


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=45 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH038) \square AEDC V4IA-CIA, CANARD CONTROL, BNICITI
 (CXH039) \square AEDC V4IA-CIA, CANARD CONTROL, BNICITI
 (CXH040) \square AEDC V4IA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SPEC 19.6350 50. IN.
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 BREF 5.0000 IN.
 XTRP 26.0000 IN.
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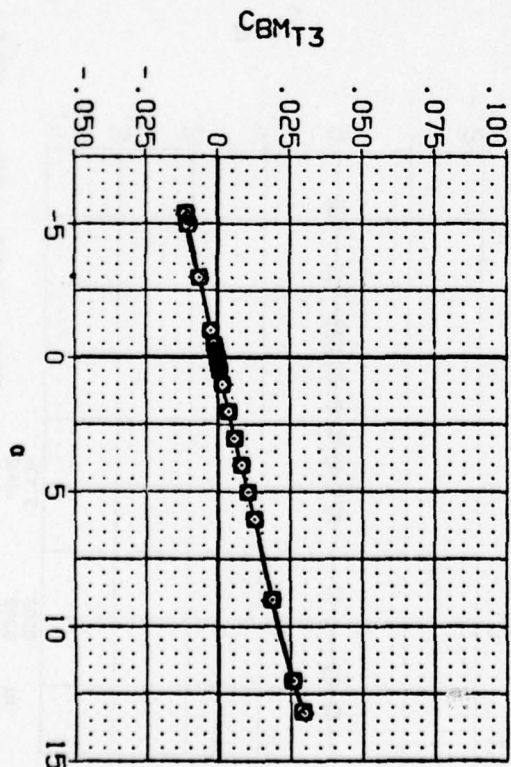
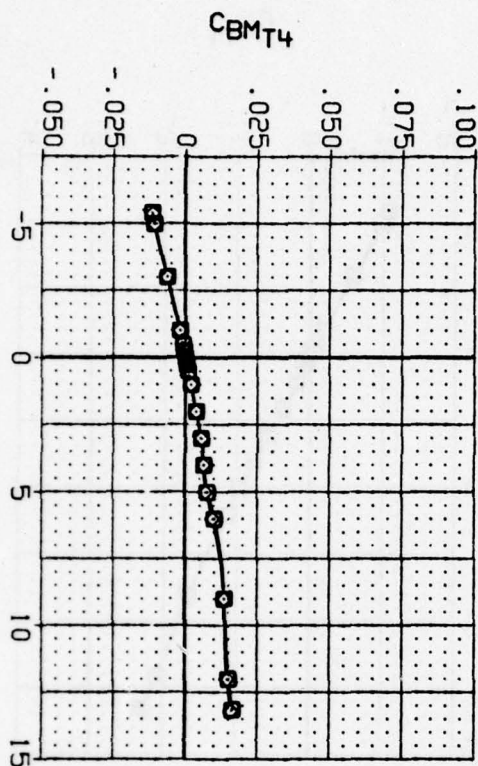
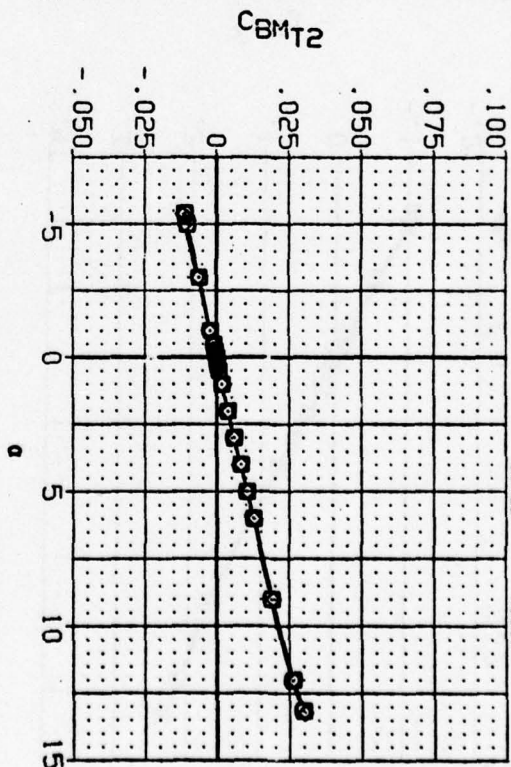
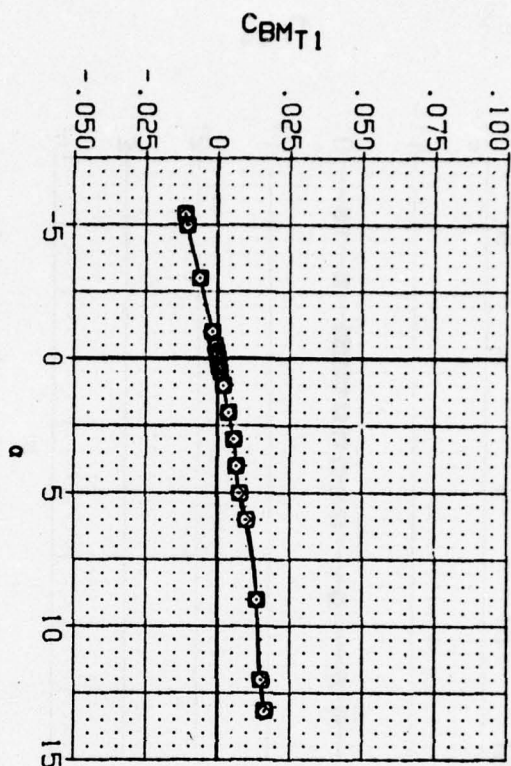


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=45 PHICND=0
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH038) ☐ AEDC V41A-C1A, CANARD CONTROL, BRICIT1
 (CXH039) ☐ AEDC V41A-C1A, CANARD CONTROL, BRICIT1
 (CXH040) ☐ AEDC V41A-C1A, CANARD CONTROL, BRICIT1

DCND1 DCND2 DCND3 DCND4
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
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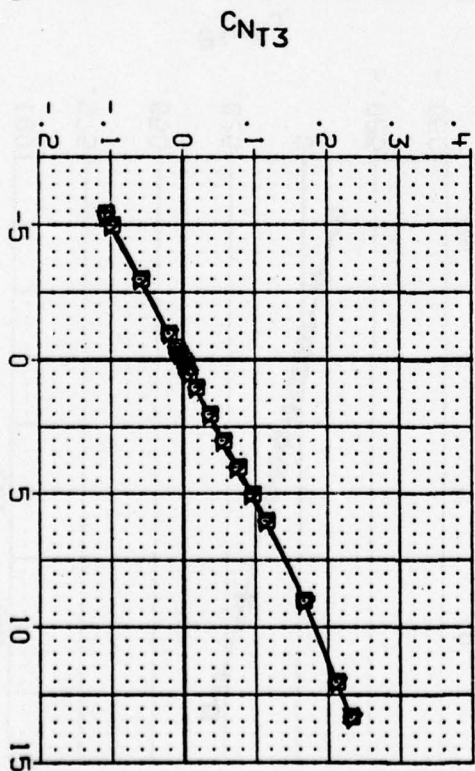
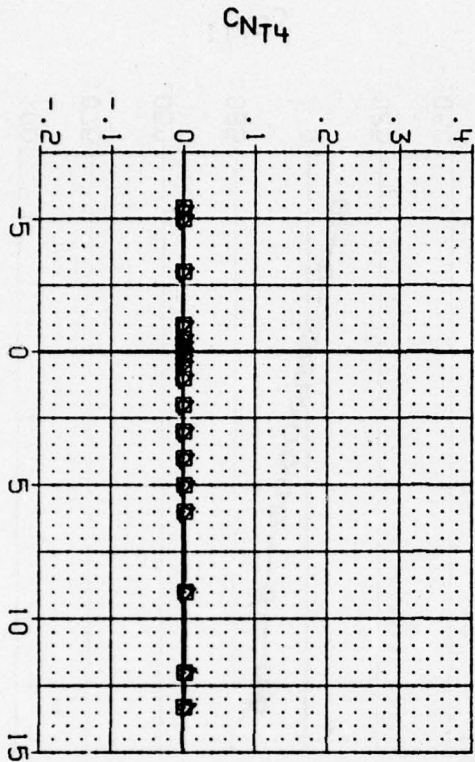
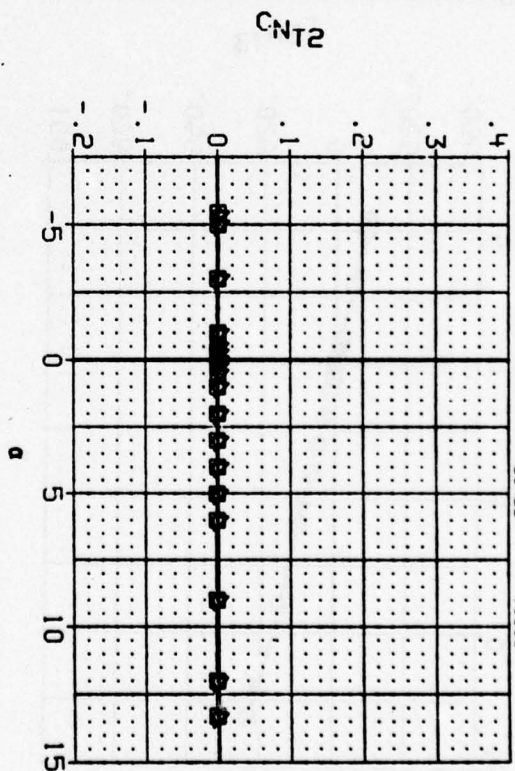
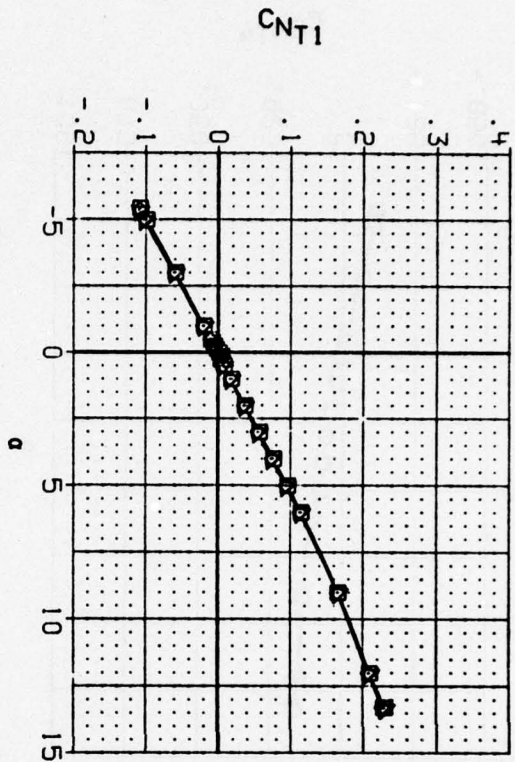


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=45 PHICND=0
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH041) DATA NOT AVAILABLE
 (AXH042) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH043) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH044) DATA NOT AVAILABLE
 (AXH045) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 5.000 -5.000 -3.000 -3.000 1.000 1.000 3.000
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 DCND3 -5.000 -3.000 -3.000 1.000 1.000 3.000
 DCND4 5.000 -3.000 -3.000 1.000 1.000 3.000

REFERENCE INFORMATION
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

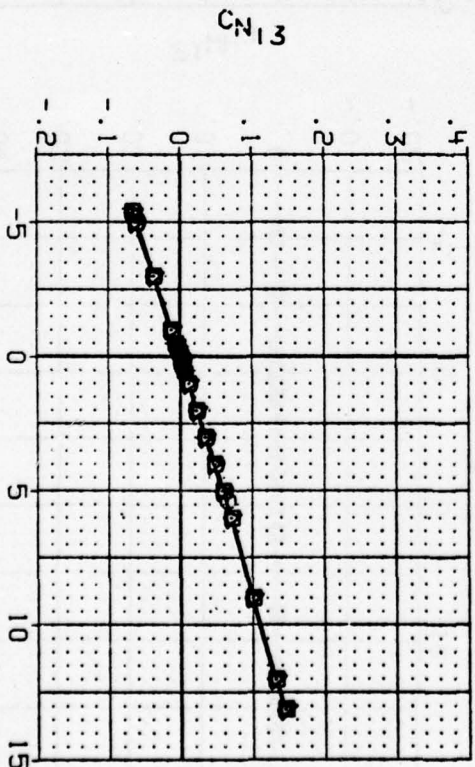
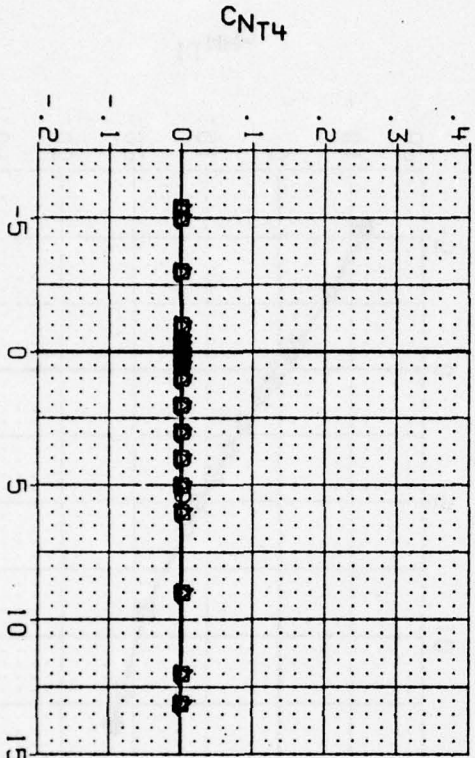
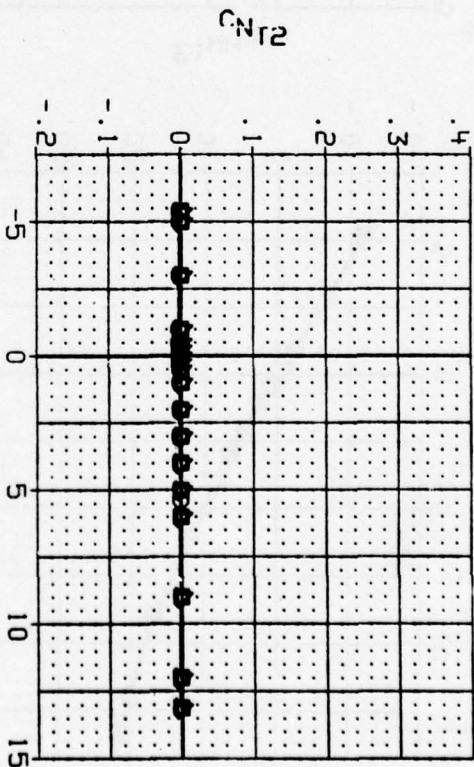
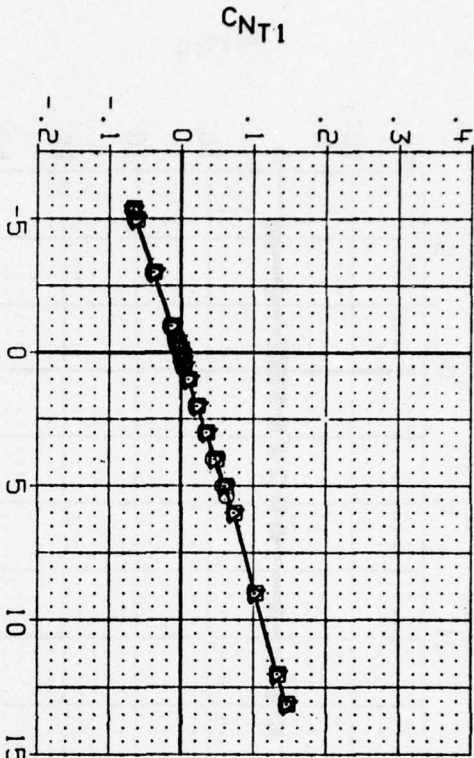
(AXH041) \square AEDC V41A-C1A, CANARD CONTROL, BNICIT1
 (AXH042) \square AEDC V41A-C1A, CANARD CONTROL, BNICIT1
 (AXH043) \square AEDC V41A-C1A, CANARD CONTROL, BNICIT1
 (AXH044) \square AEDC V41A-C1A, CANARD CONTROL, BNICIT1
 (AXH045) \square AEDC V41A-C1A, CANARD CONTROL, BNICIT1

DCND1 DCND2 DCND3 DCND4

5.000 -5.000 -5.000 5.000
 -3.000 -3.000 -3.000 -3.000
 1.000 .000 .000 1.000
 3.000 1.000 3.000 3.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XRRP 26.0000 IN.
 YRRP .0000 IN.
 ZRRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=45 PHICND=45

(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

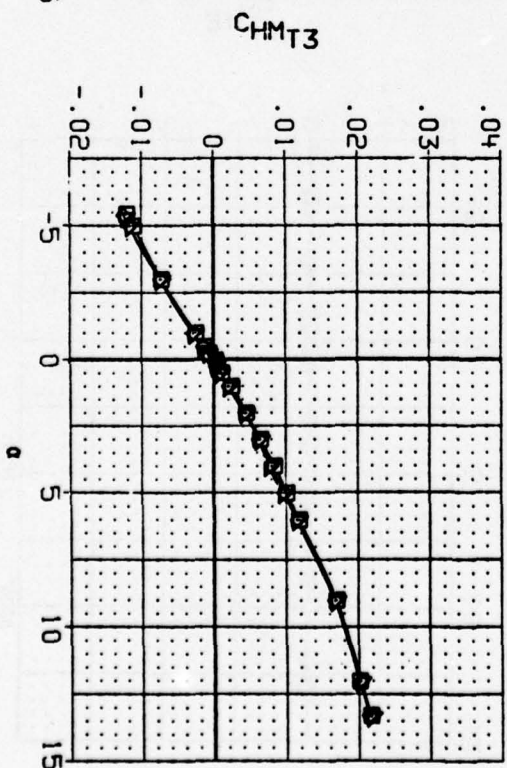
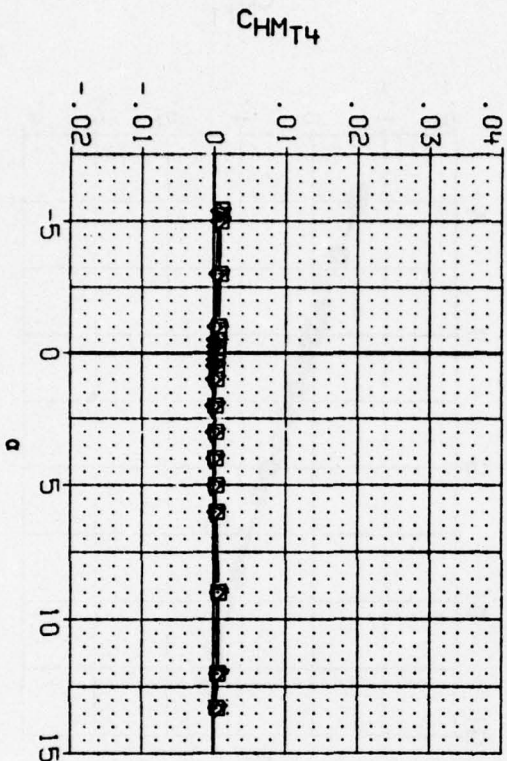
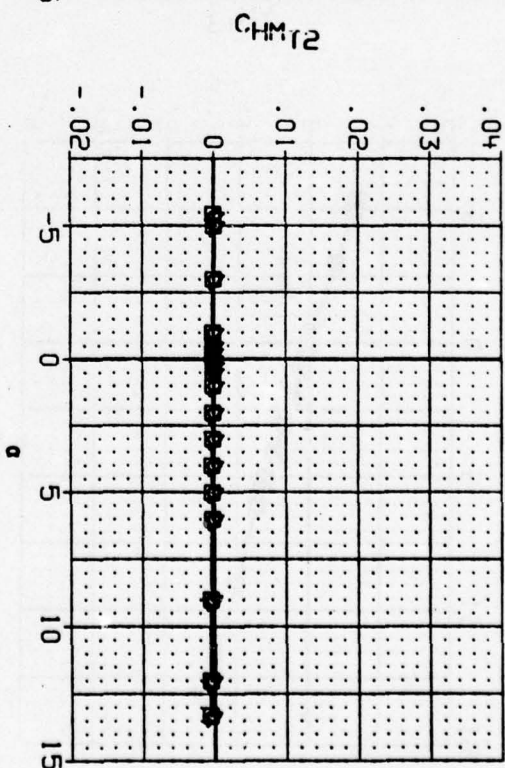
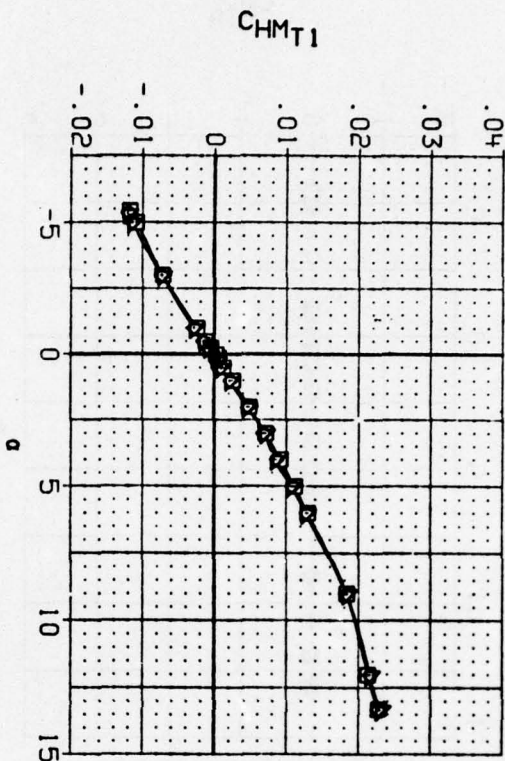
(CXH0+1) DATA NOT AVAILABLE
 (CXH0+2) AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (CXH0+3) AEDC VMA-CIA, CANARD CONTROL, BNICITI
 (CXH0+4) DATA NOT AVAILABLE
 (CXH0+5) AEDC VMA-CIA, CANARD CONTROL, BNICITI

DCND1 DCND2 DCND3 DCND4

5.000 -5.000 -5.000 5.000
 -3.000 -3.000 -3.000 -3.000
 .000 .000 .000 .000
 1.000 1.000 1.000 1.000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



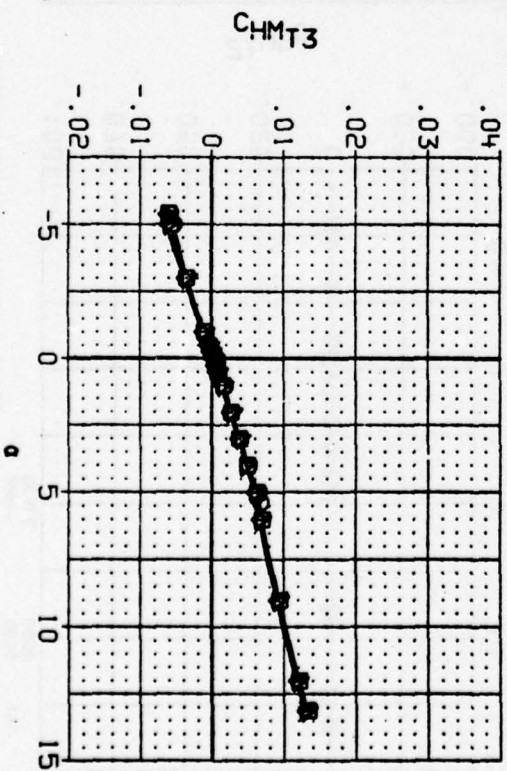
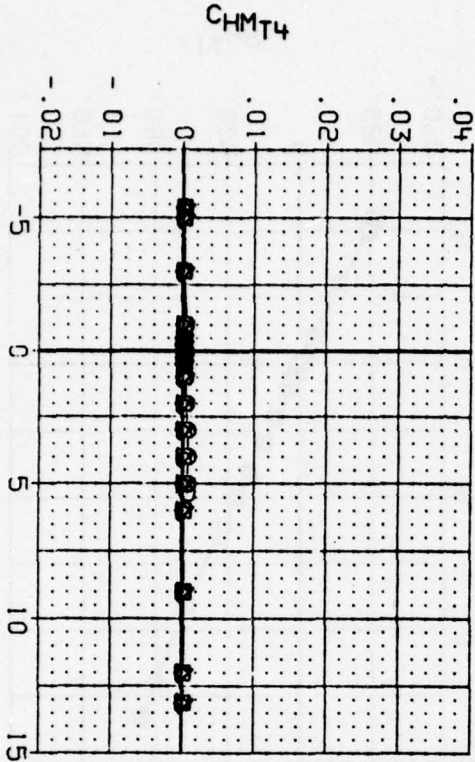
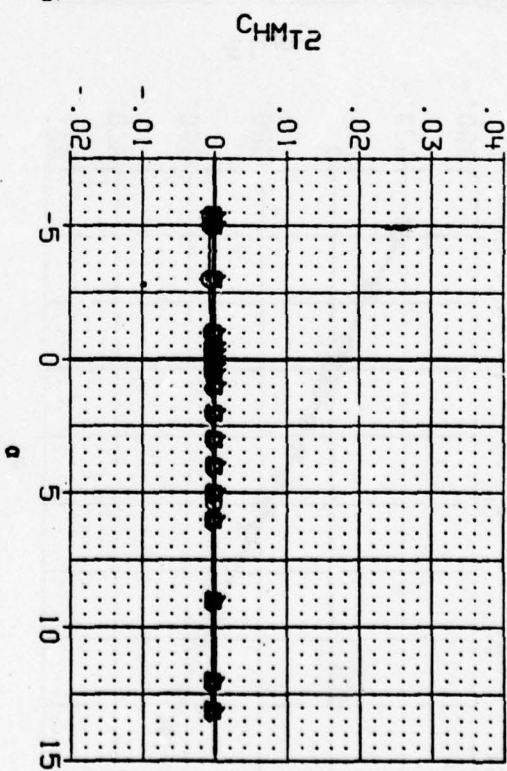
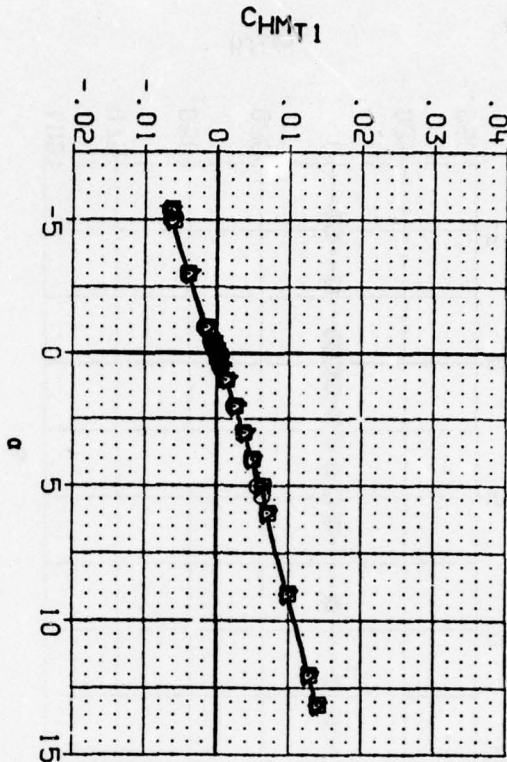
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=45 PHICND=45

(A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH041)	○	AEDC V4IA-CIA, CANARD CONTROL, BNICITI
(CXH042)	◇	AEDC V4IA-CIA, CANARD CONTROL, BNICITI
(CXH043)	△	AEDC V4IA-CIA, CANARD CONTROL, BNICITI
(CXH044)	△	AEDC V4IA-CIA, CANARD CONTROL, BNICITI
(CXH045)	△	AEDC V4IA-CIA, CANARD CONTROL, BNICITI

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
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-3.000	-3.000	-3.000	-3.000	LREF 5.0000 IN.
.000	.000	.000	.000	BREF 5.0000 IN.
1.000	1.000	1.000	1.000	XREF 26.0000 IN.
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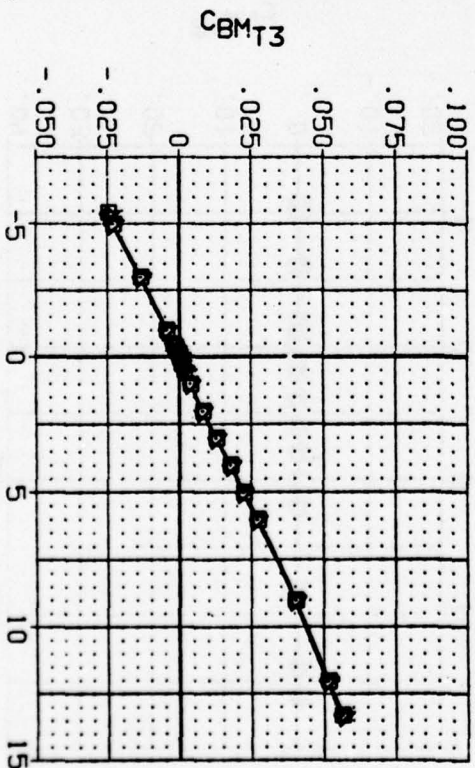
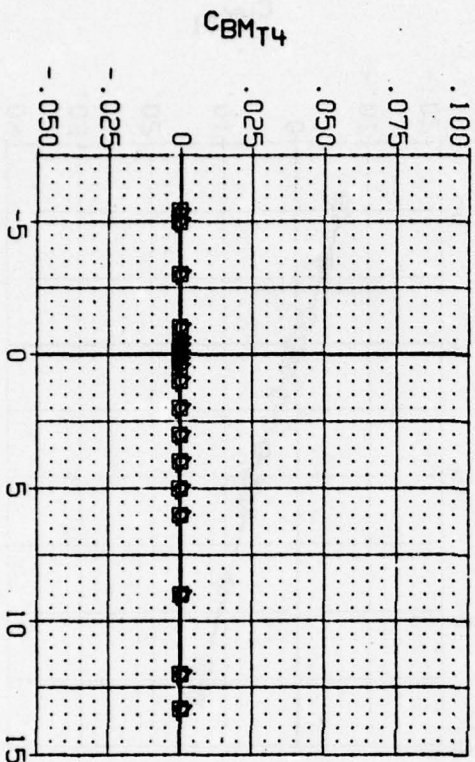
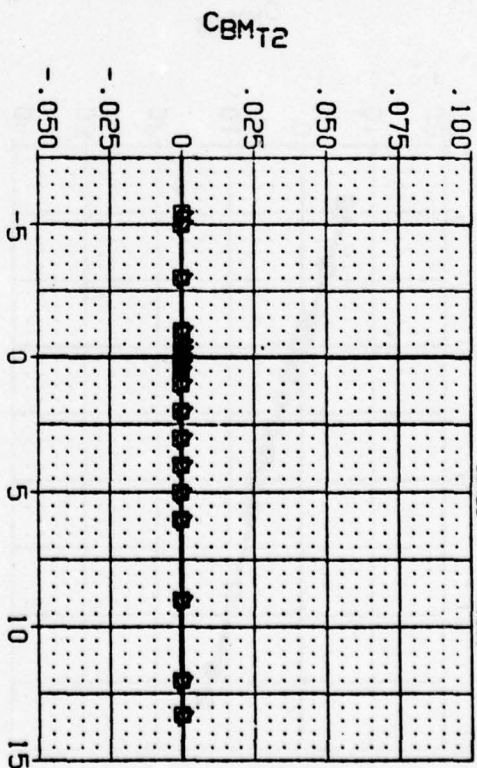
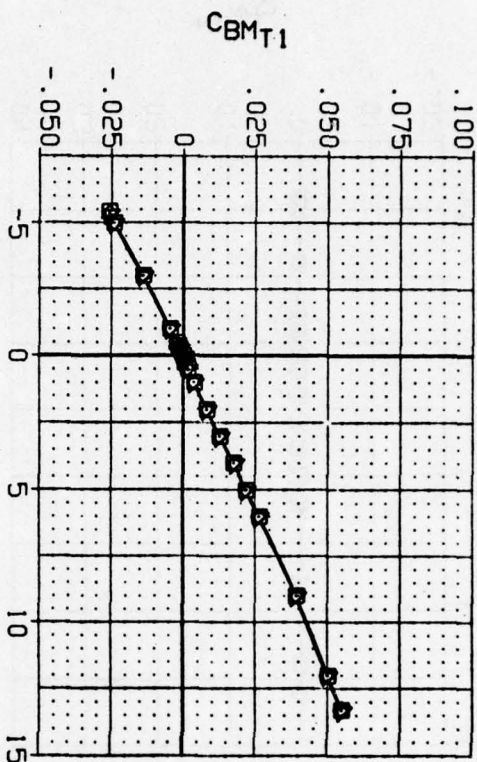


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH041) DATA NOT AVAILABLE
 (CXH042) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH043) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH044) DATA NOT AVAILABLE
 (CXH045) AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 DCND2 DCND3 DCND4
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 3.000 3.000 3.000 3.000

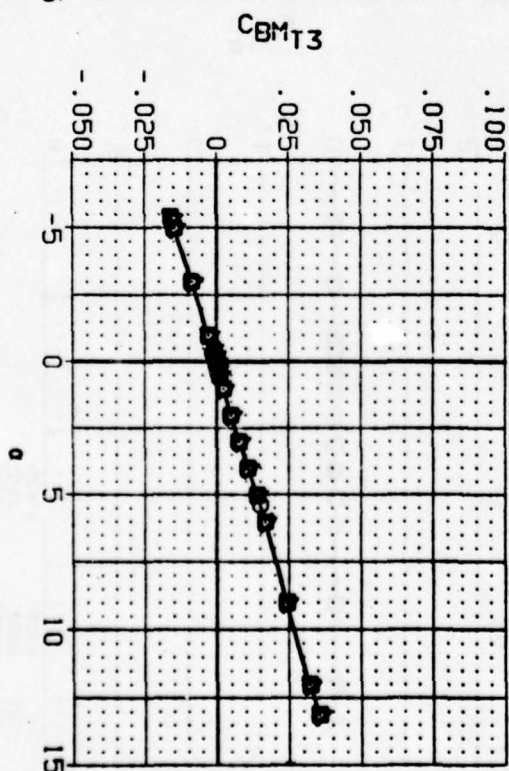
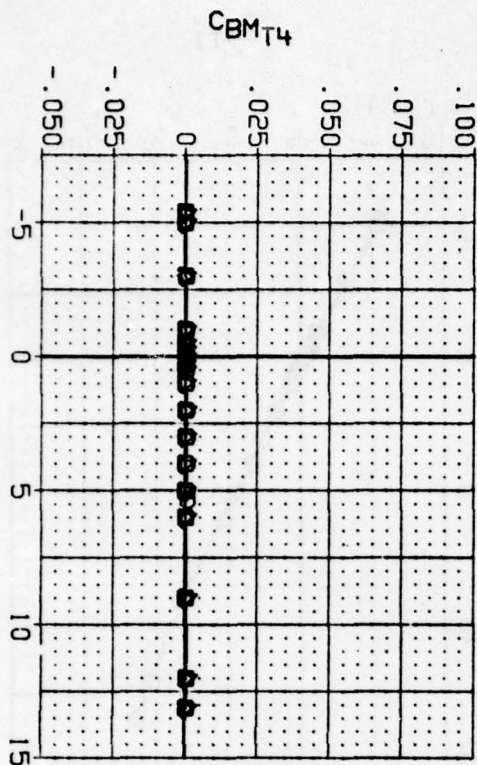
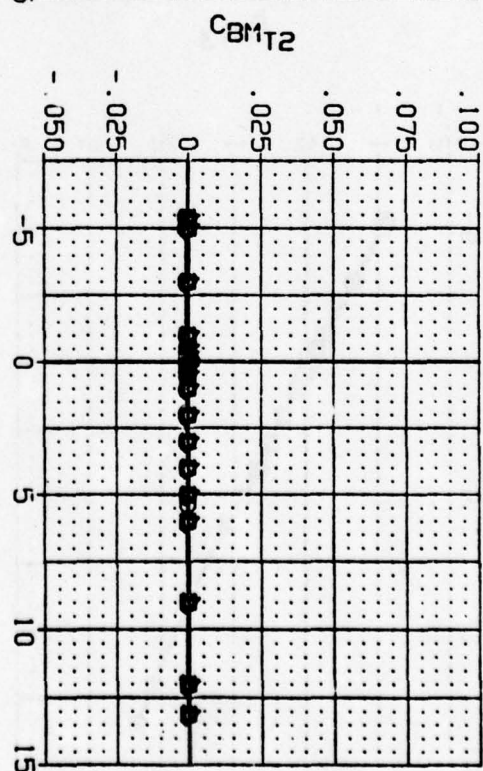
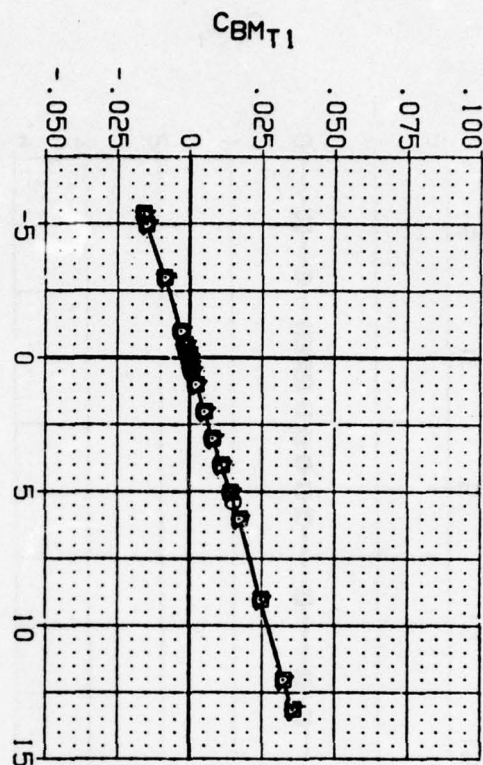
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH041)	○	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(CXH042)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(CXH043)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(CXH044)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
(CXH045)	▽	AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
5.000	-5.000	-5.000	-5.000	SREF 19.6350 SQ. IN.
-3.000	-3.000	-3.000	-3.000	LREF 5.0000 IN.
1.000	1.000	1.000	1.000	BREF 5.0000 IN.
3.000	3.000	3.000	3.000	XMRP 26.0000 IN.
				YMRP .0000 IN.
				ZMRP .0000 IN.
				SCALE .0000

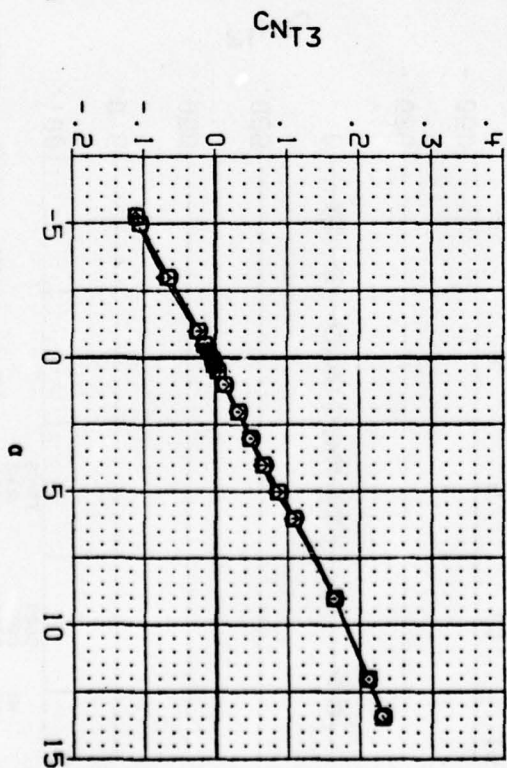
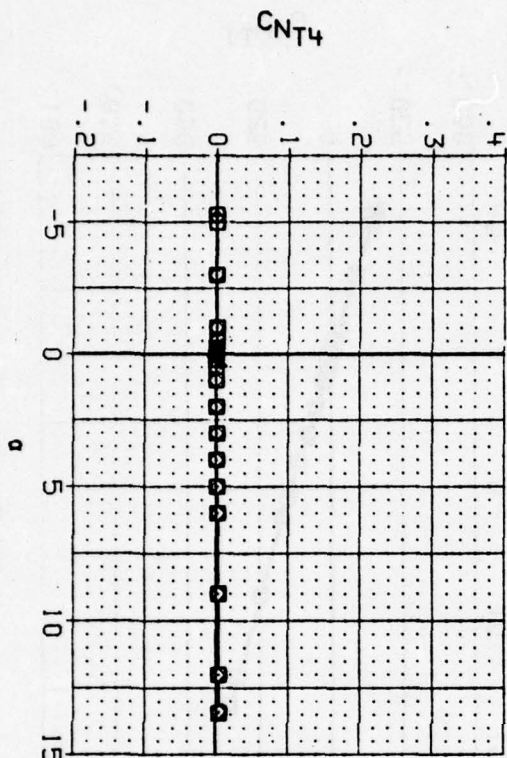
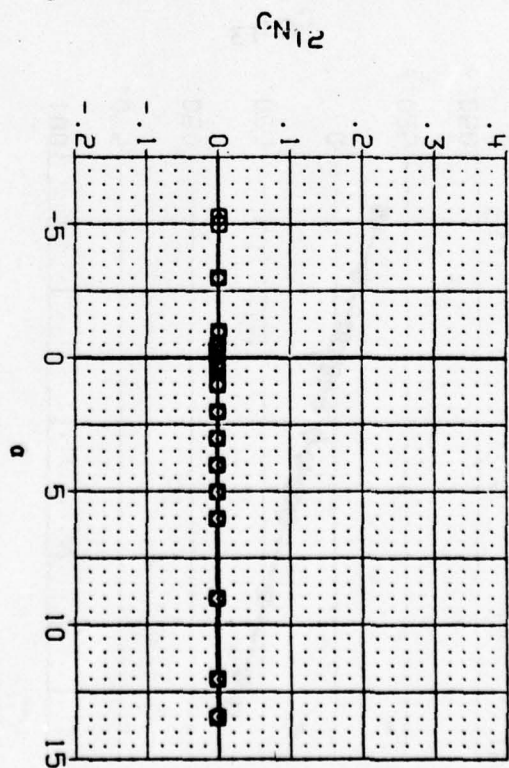
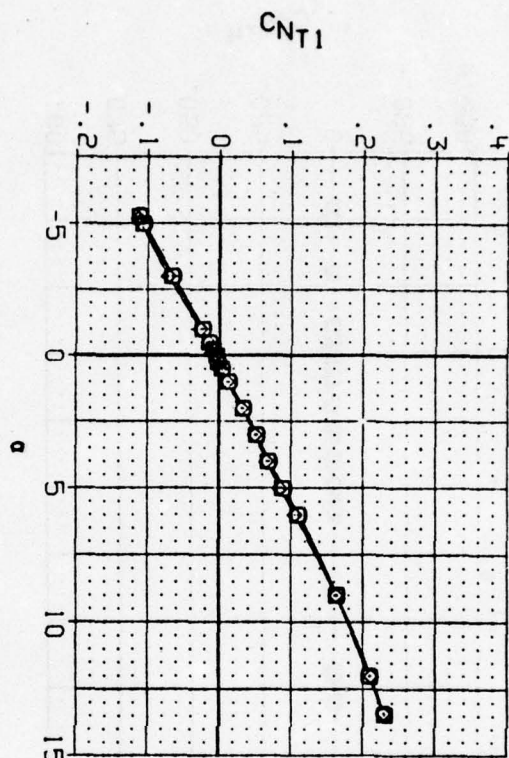


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (B)MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH046) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH047) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (AXH048) \diamond AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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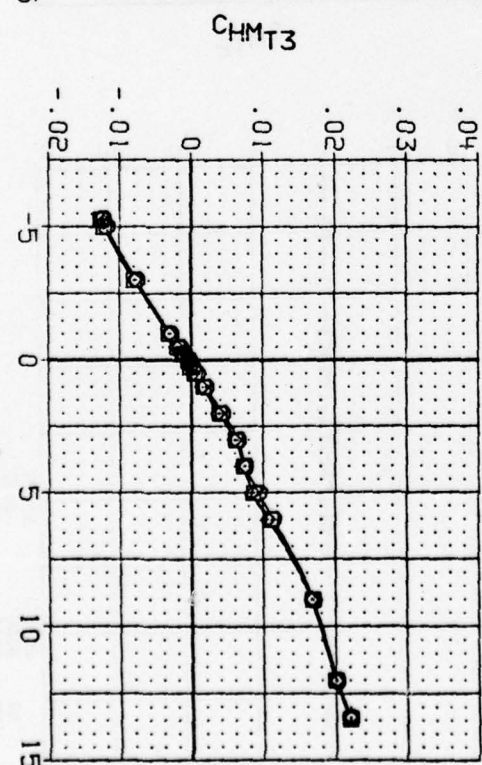
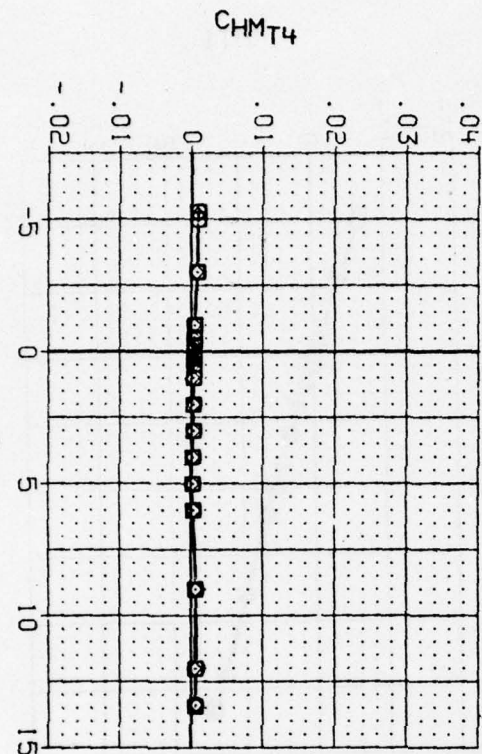
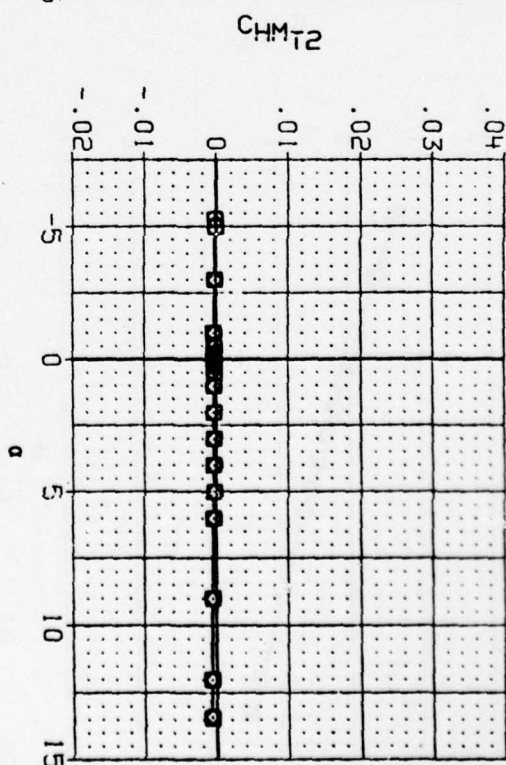
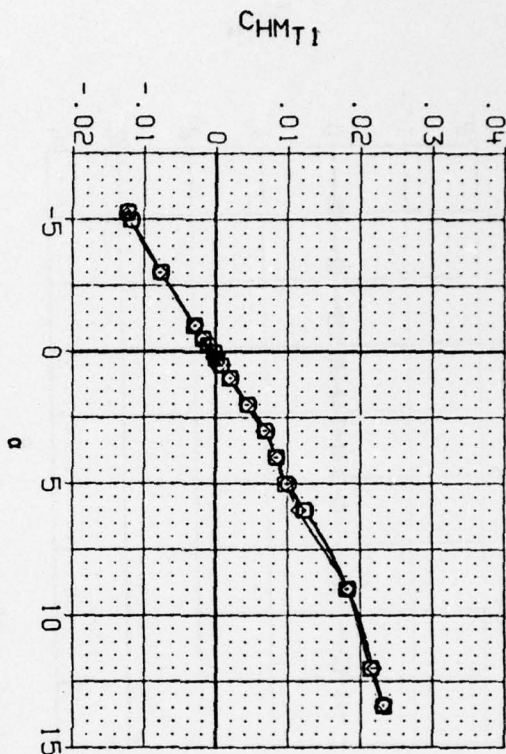


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAL=45$
 $PHICND=45$
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXHD-6) ☐ AEDC VM-A-CIA CANARD CONTROL BNICIT1
 (CXHD-7) ☐ AEDC VM-A-CIA CANARD CONTROL BNICIT1
 (CXHD-8) ☐ AEDC VM-A-CIA CANARD CONTROL BNICIT1

DCND1 6.000 6.000 6.000 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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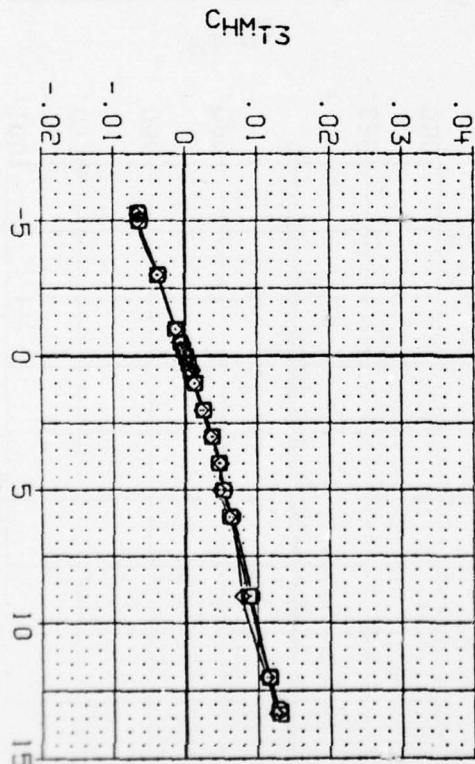
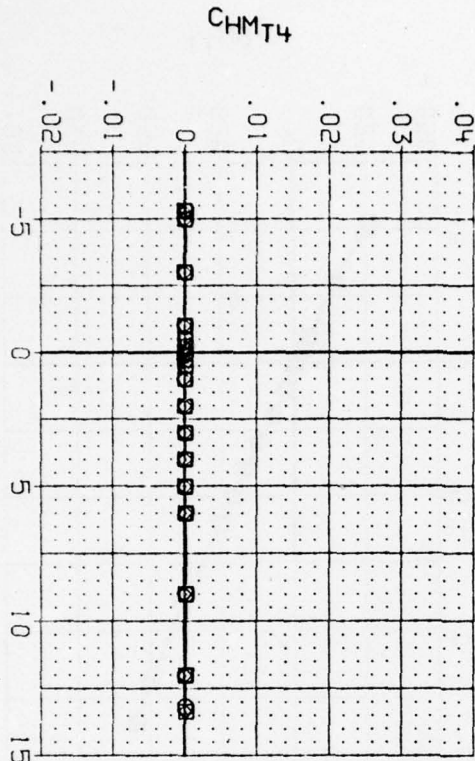
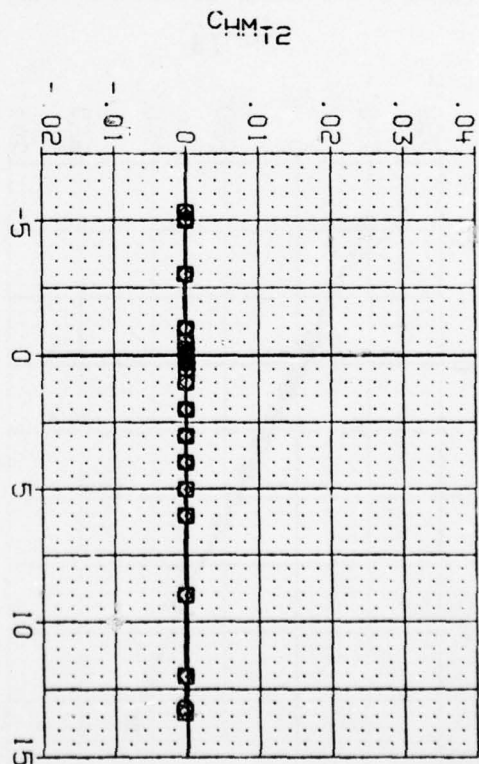
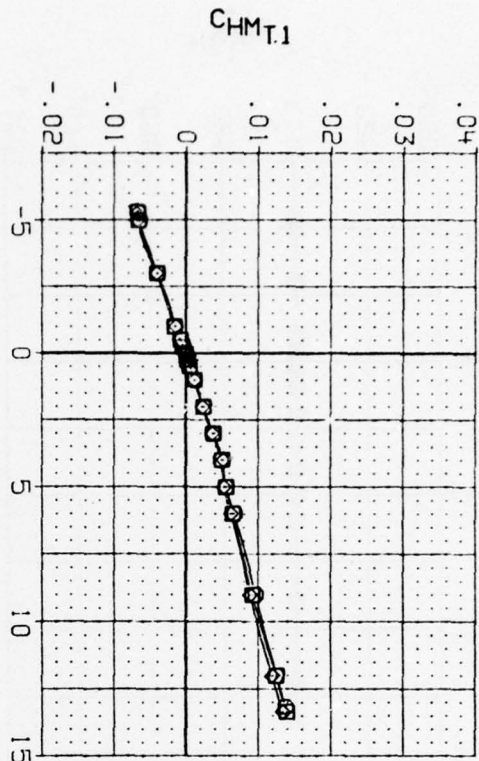


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH046) ☐ AEDC V4-A-C1A, CANARD CONTROL, BN(C1T1)
 (CXH047) ☐ AEDC V4-A-C1A, CANARD CONTROL, BN(C1T1)
 (CXH048) ☐ AEDC V4-A-C1A, CANARD CONTROL, BN(C1T1)

DCND1 5.000 DCND2 5.000 DCND3 5.000 DCND4 5.000
 5.000 9.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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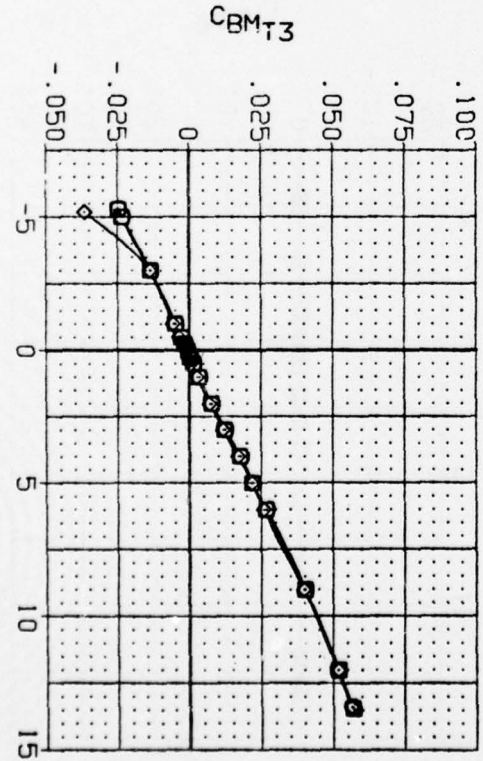
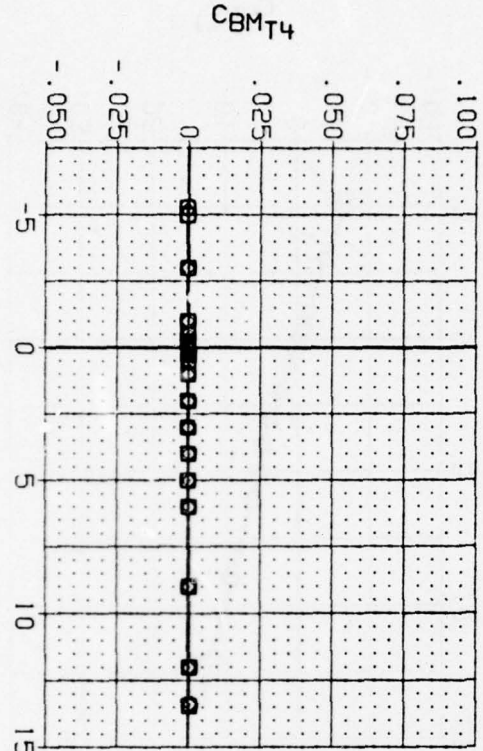
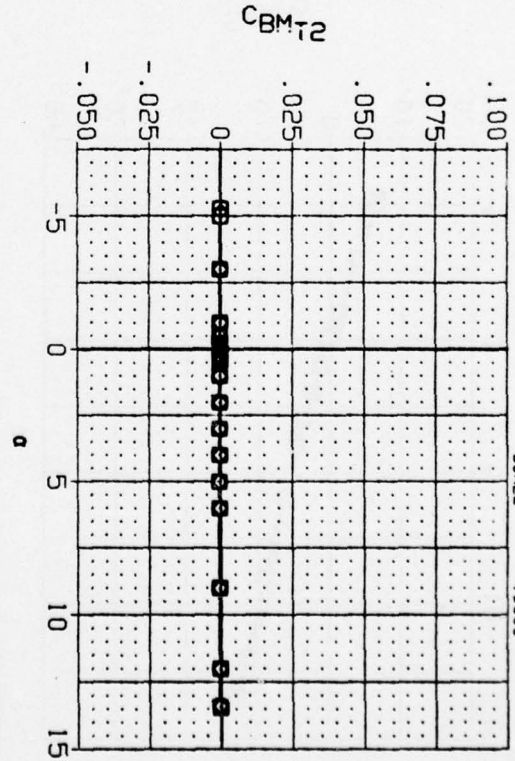
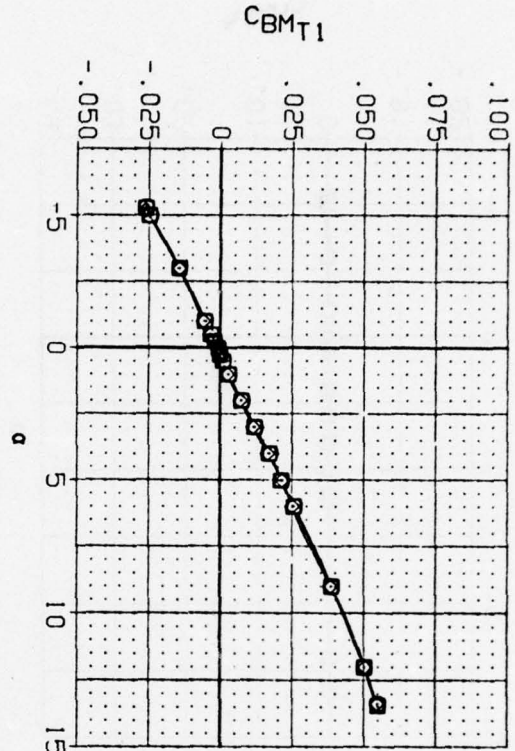


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH046) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (CXH047) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICITI
 (CXH048) ☒ AEDC V41A-C1A, CANARD CONTROL, BNICITI

DCND1 6.000 DCND2 6.000 DCND3 6.000 DCND4 6.000
 9.000 9.000 9.000 9.000
 5.000 15.000 15.000 15.000

REFERENCE INFORMATION
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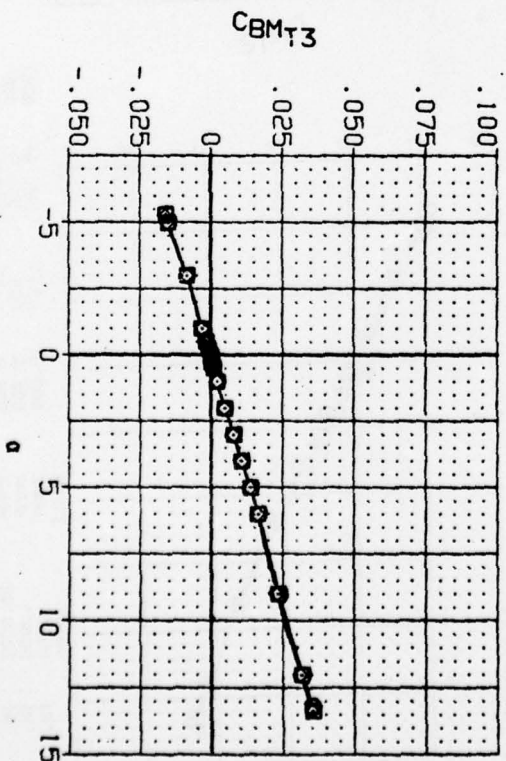
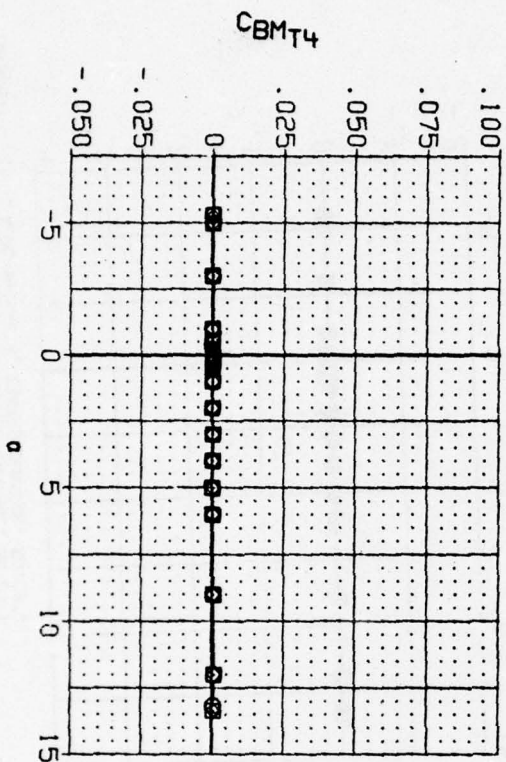
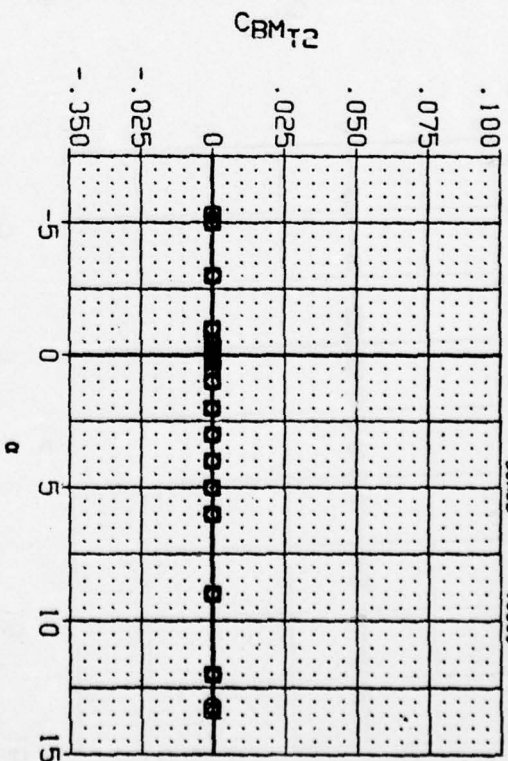
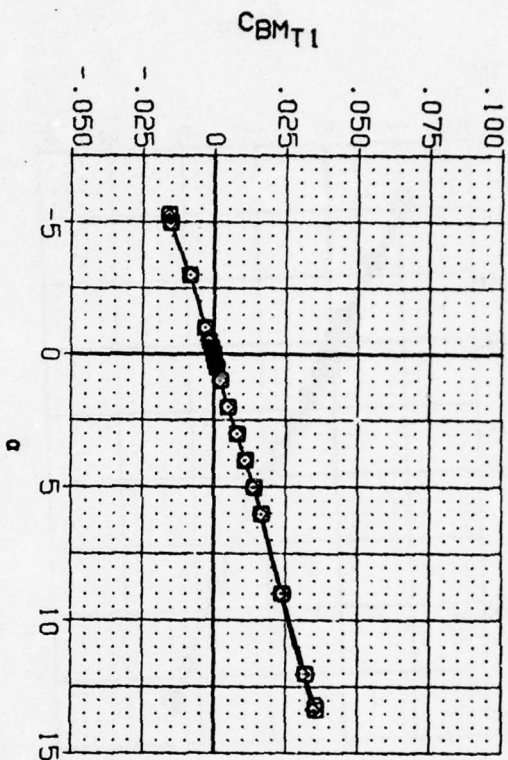


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH046) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH047) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC1T1
 (CXH048) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC1T1

DCND1 6.000 DCND2 6.000 DCND3 6.000 DCND4 6.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=45 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

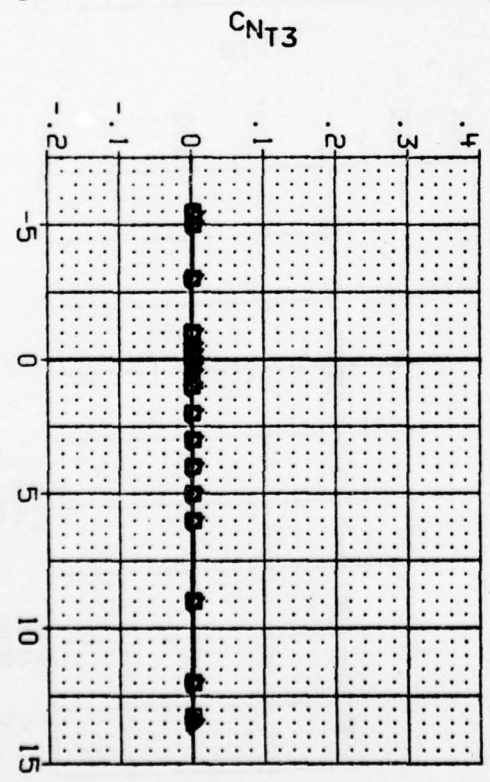
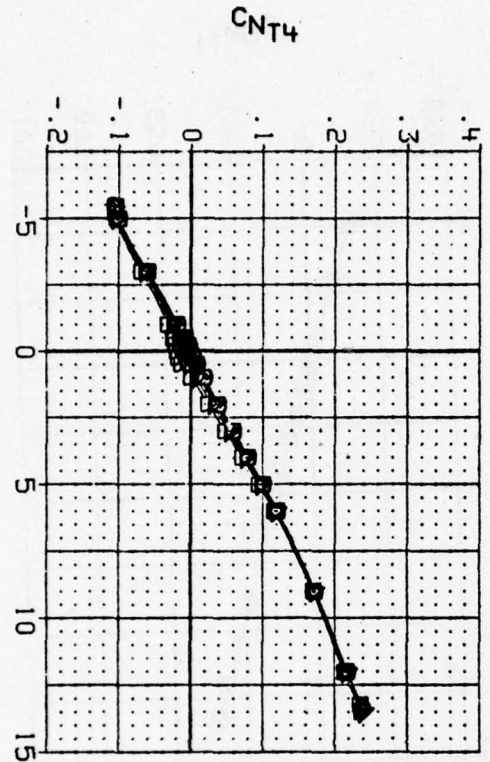
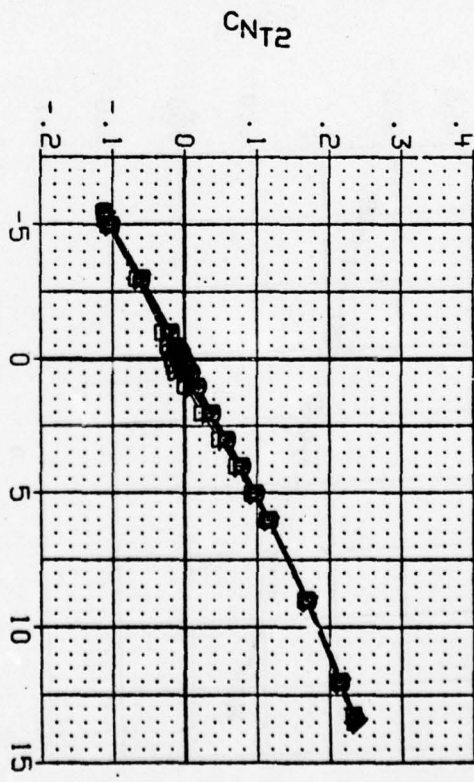
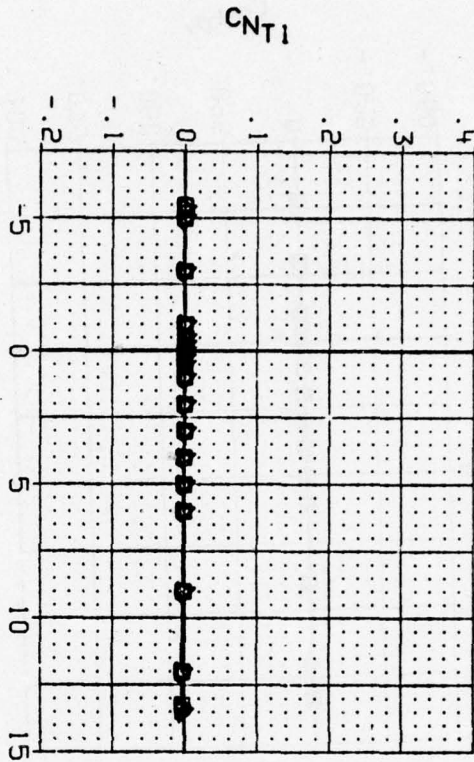
(AXH049) AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (AXH050) AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (AXH051) AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (AXH052) AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (AXH053) AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (AXH054) AEDC V41A-C1A, CANARD CONTROL, BNICST1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 1.000 .000 .000
 .000 3.000 .000 1.000
 .000 6.000 .000 3.000
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REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
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 SCALE .0000



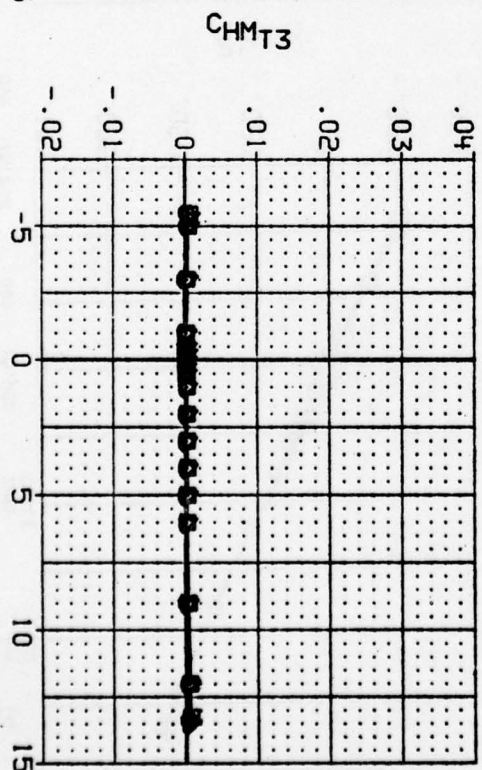
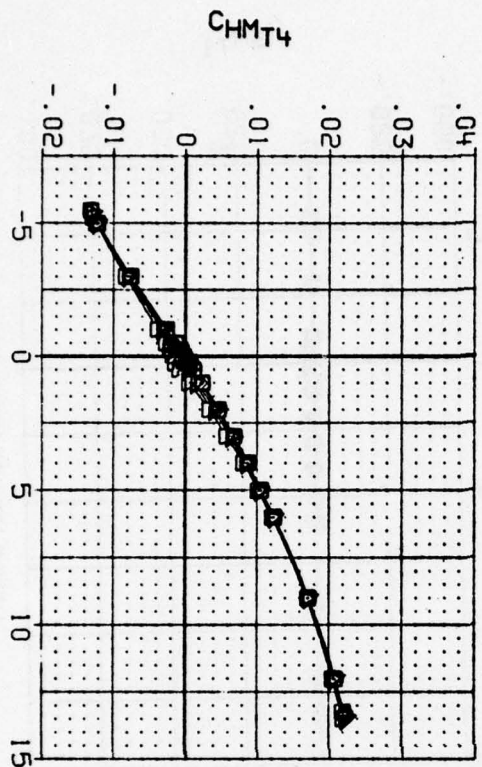
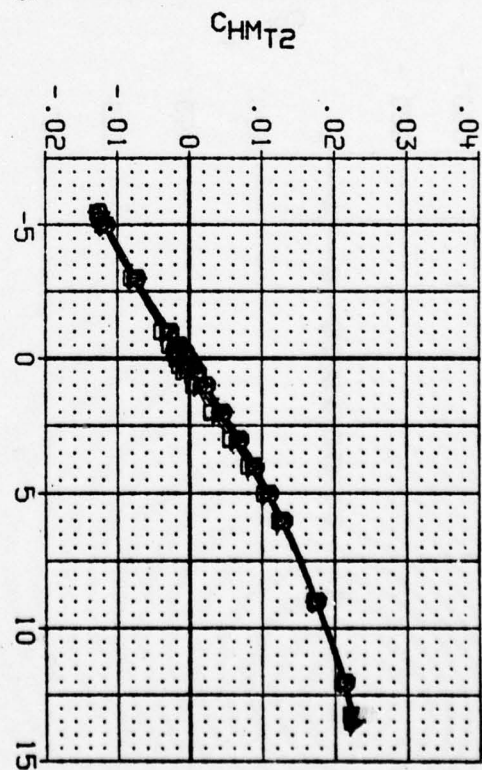
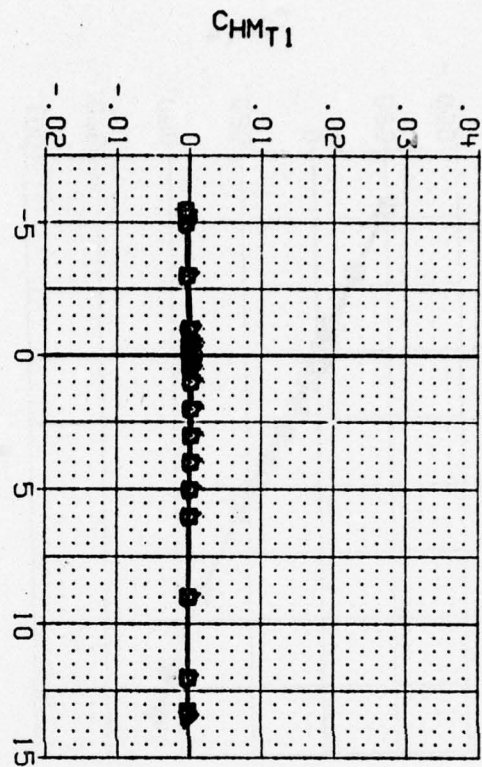
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

(A) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH049)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH050)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH051)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH052)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH053)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH054)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1

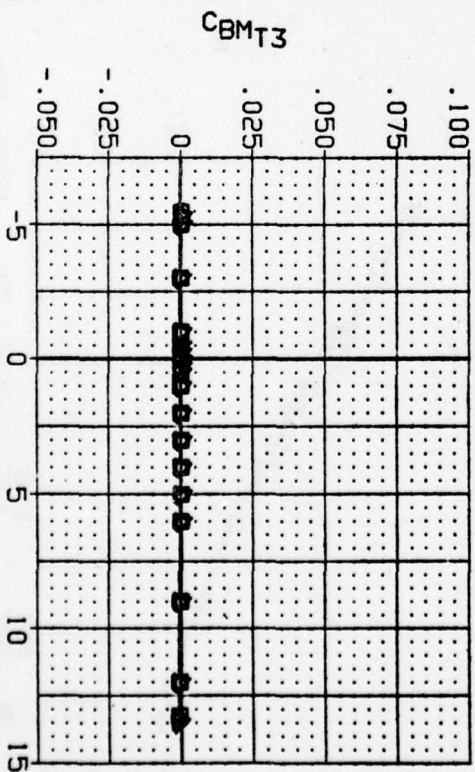
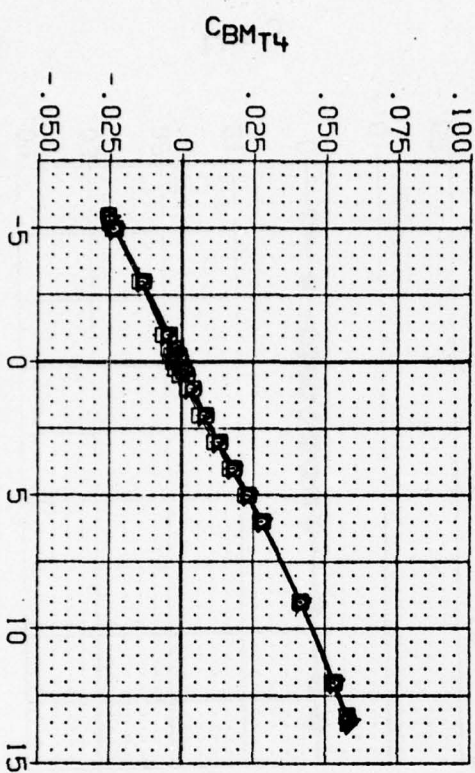
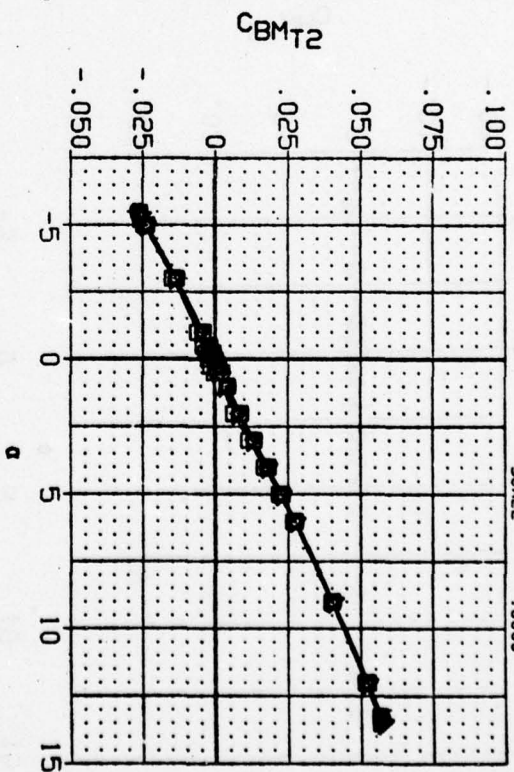
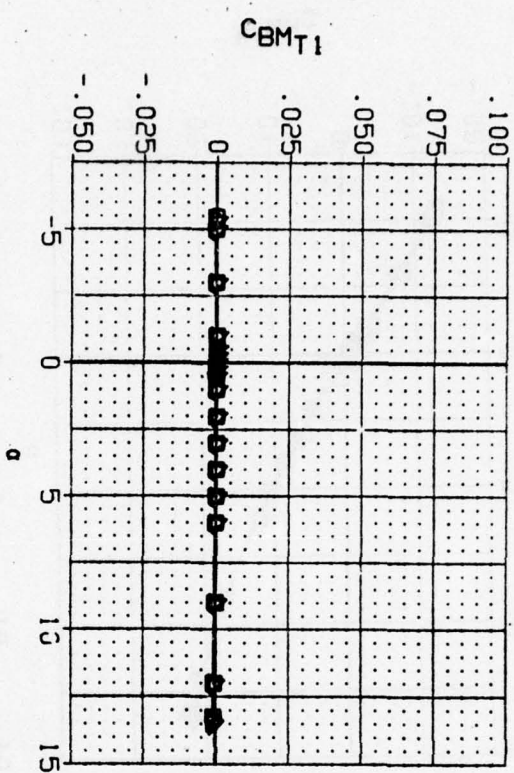
DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
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.000	3.000	.000	.000	BREF 26.0000 IN.
.000	6.000	.000	.000	YREF .0000 IN.
.000	9.000	.000	.000	ZREF .0000 IN.
.000	15.000	.000	.000	SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (A) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH09)	□	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH050)	◇	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH051)	△	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH052)	▽	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH053)	◊	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1
(CXH054)	◈	AEDC V41A-C1A, CANARD CONTROL, BNIC3T1

DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
.000	.000	.000	.000	SREF 19.6350 SQ. IN.
.000	1.000	.000	.000	LREF 5.0000 IN.
.000	3.000	.000	.000	BREF 5.0000 IN.
.000	6.000	.000	.000	YREF 26.0000 IN.
.000	9.000	.000	.000	ZREF .0000 IN.
.000	15.000	.000	.000	SCALE .0000 IN.

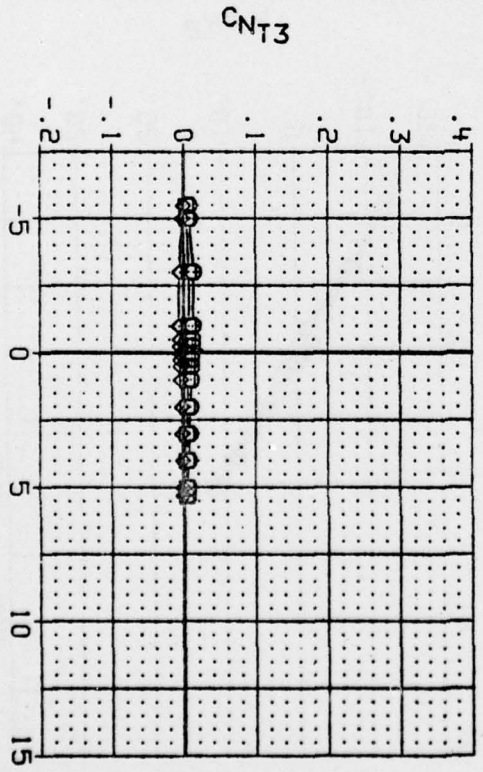
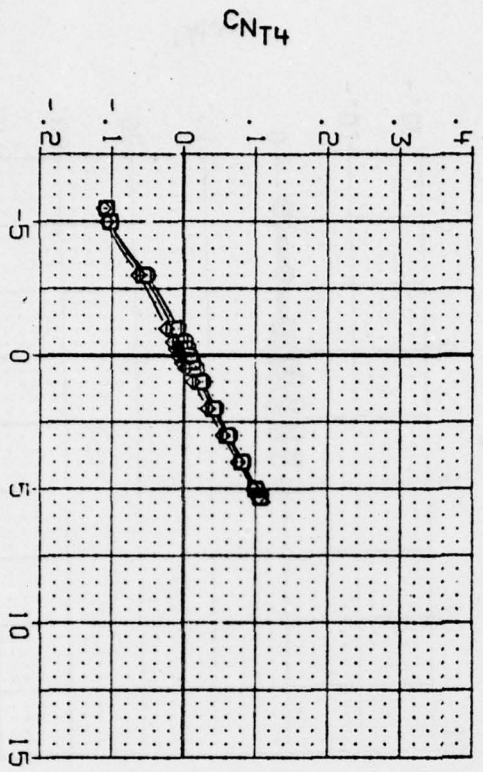
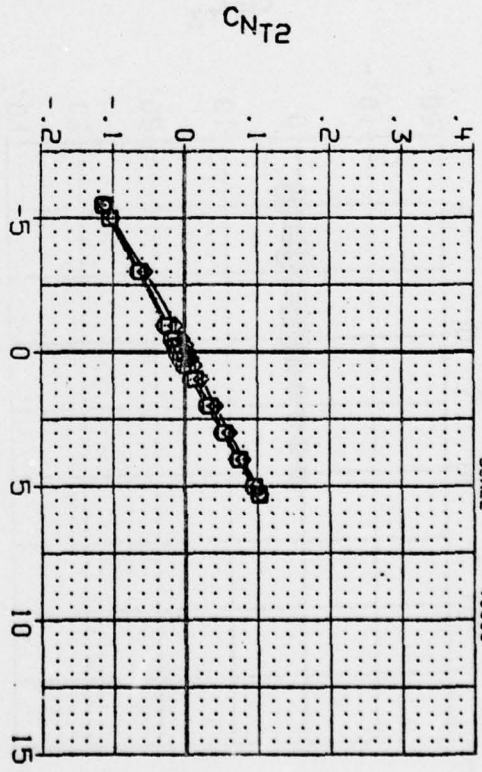
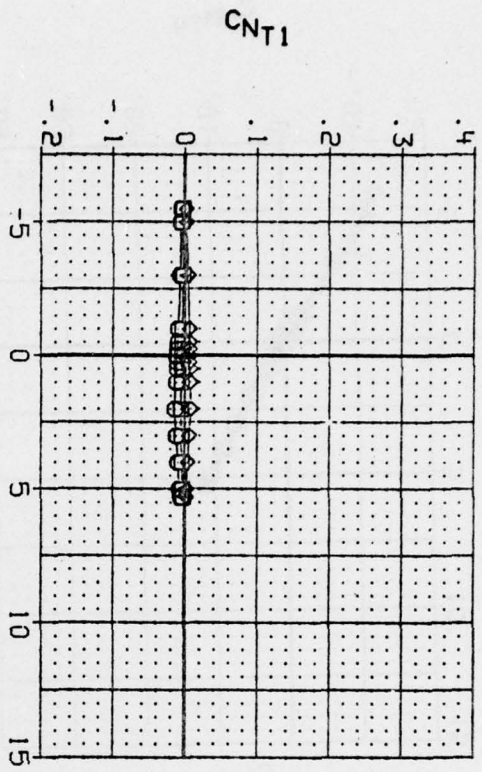


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH055) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH056) \square AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (AXH057) \diamond AEDC W1A-C1A, CANARD CONTROL, BNIC3T1

DCND1 5.000
 2.000
 -3.000
 DCND2 -5.000
 -2.000
 3.000
 DCND3 -5.000
 -2.000
 3.000
 DCND4 5.000
 2.000
 -3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
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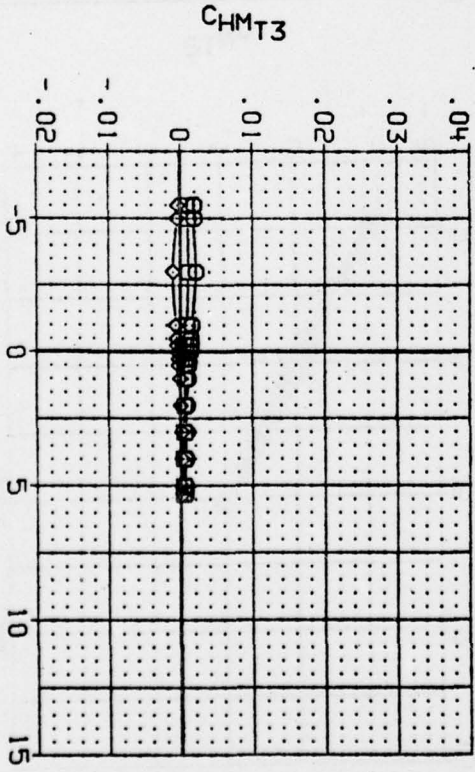
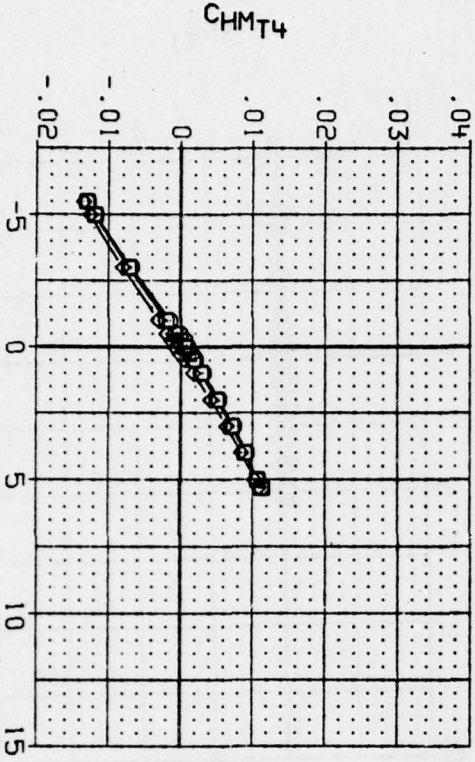
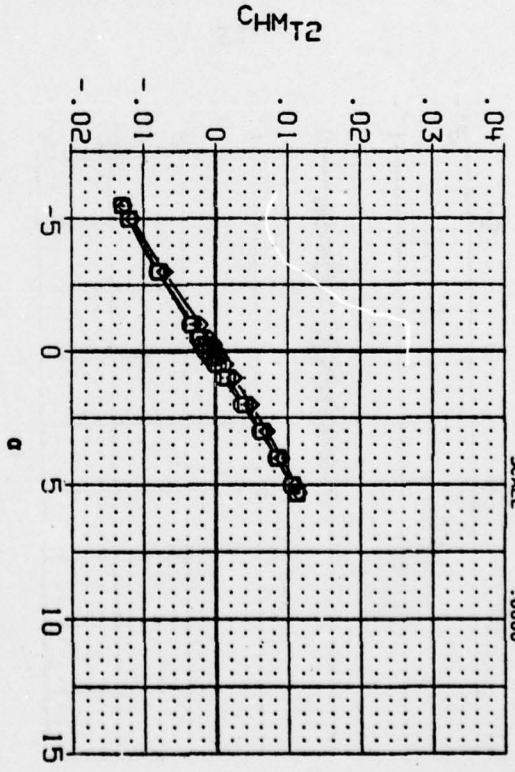
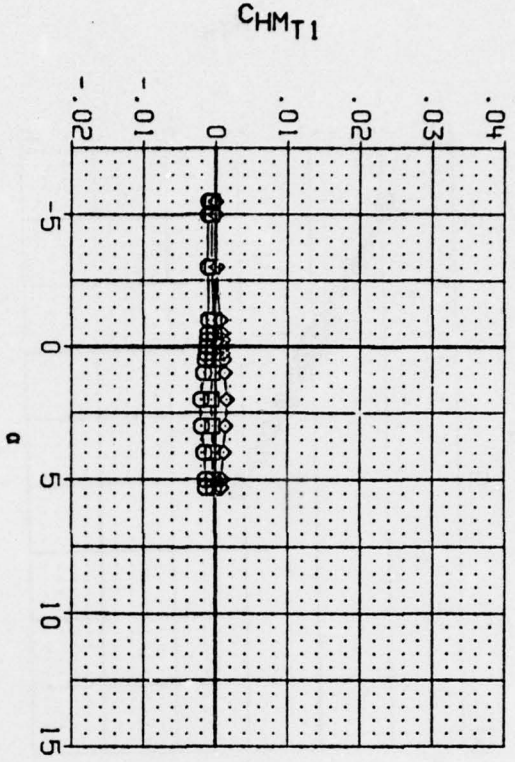


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\text{PHITAIL}=0$ $\text{PHICND}=0$
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CH055) ☐ AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (CH056) ☐ AEDC W1A-C1A, CANARD CONTROL, BNIC3T1
 (CH057) ☒ AEDC W1A-C1A, CANARD CONTROL, BNIC3T1

DCND1 5.000 DCND2 -5.000 DCND3 -5.000 DCND4 5.000
 2.000 -2.000 -2.000 2.000
 -3.000 3.000 3.000 -3.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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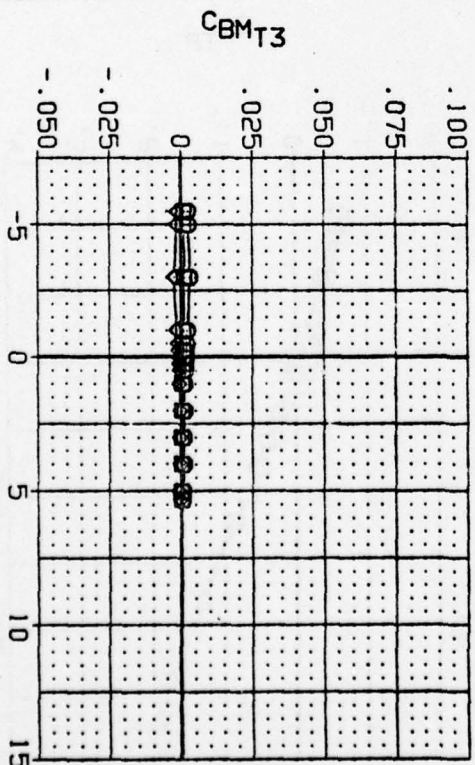
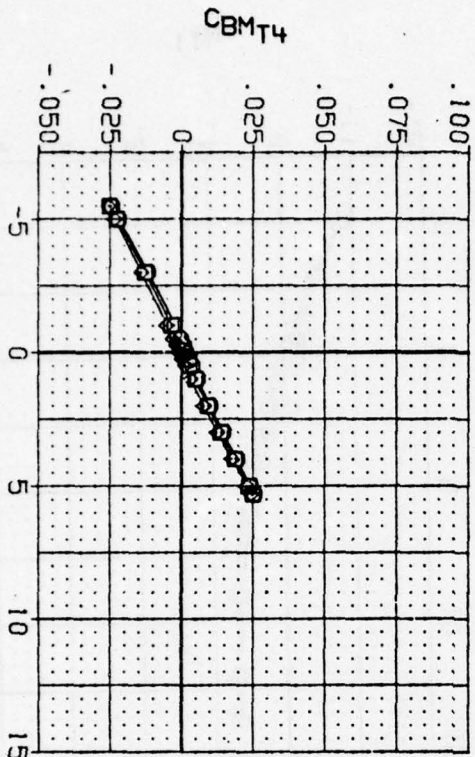
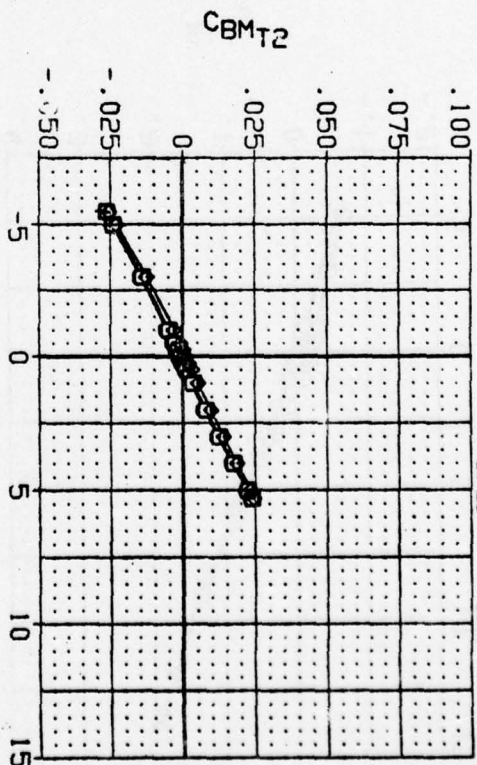
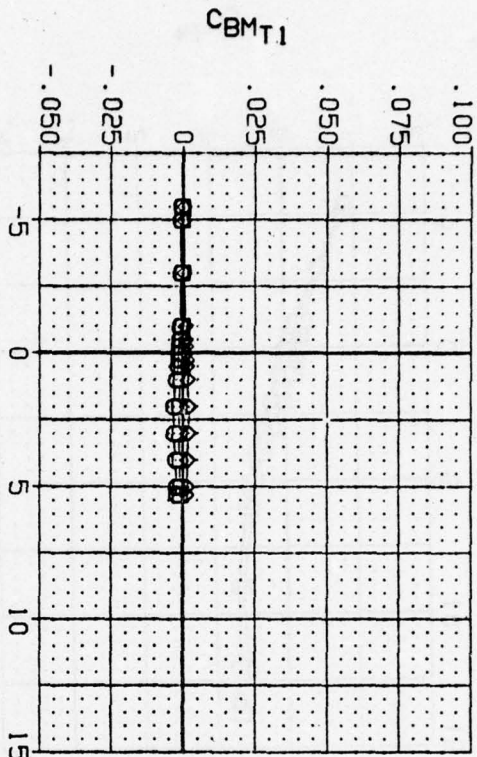


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (CXH056) ☐ AEDC V41A-C1A, CANARD CONTROL, BNICST1
 (CXH057) ☒ AEDC V41A-C1A, CANARD CONTROL, BNICST1

DCND1 5.000 DCND2 -5.000 DCND3 -5.000 DCND4 5.000
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 -3.000 3.000 3.000 -3.000

REFERENCE INFORMATION
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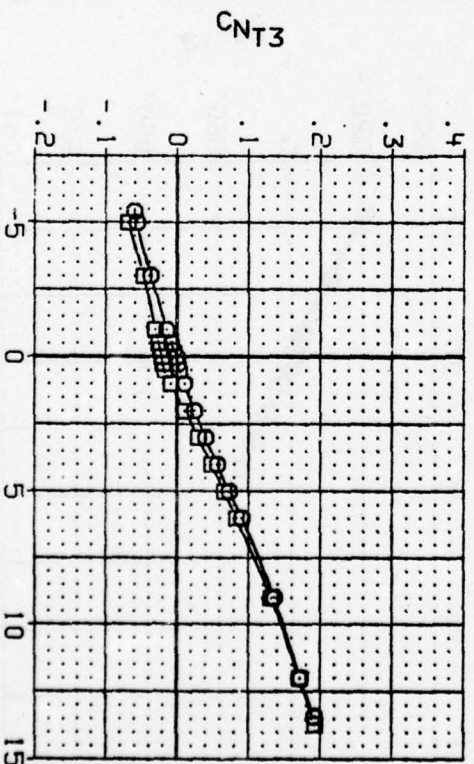
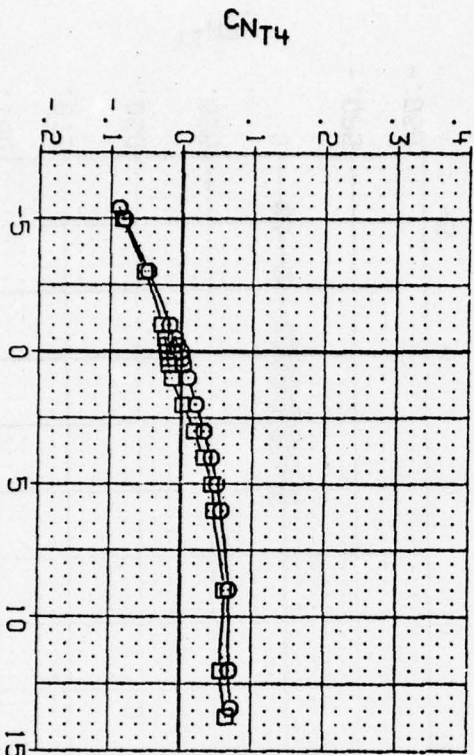
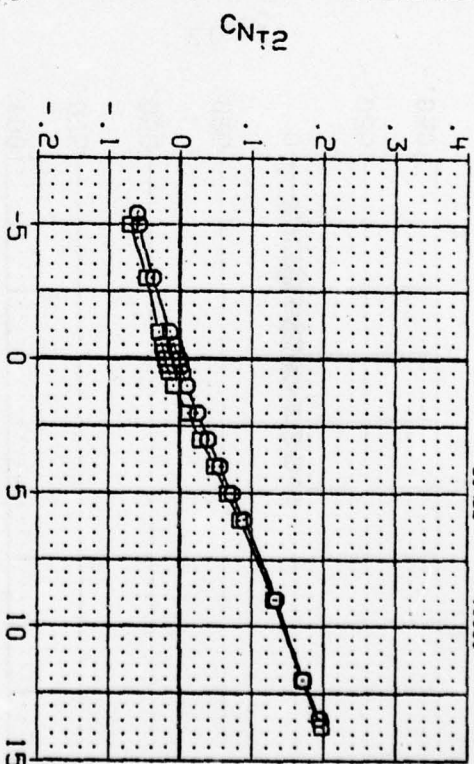
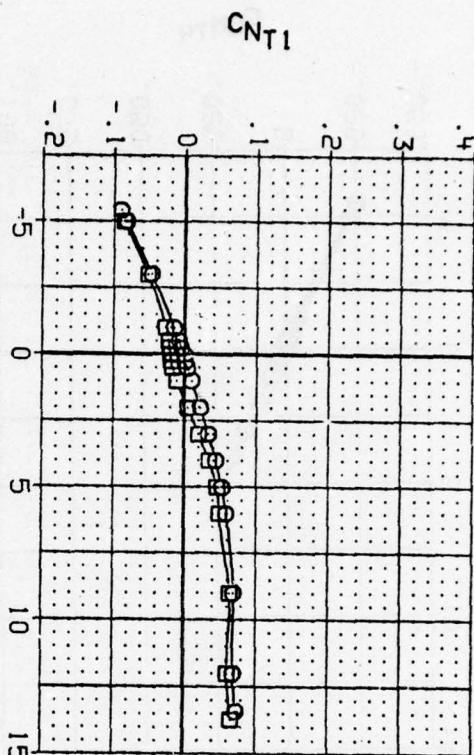


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
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 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH058) ☐ AEDC V1A-C1A, CANARD CONTROL, BN1C3T1
 (AXH059) ☐ AEDC V1A-C1A, CANARD CONTROL, BN1C3T1

COND1 COND2 COND3 COND4
 .000 .000 .000 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
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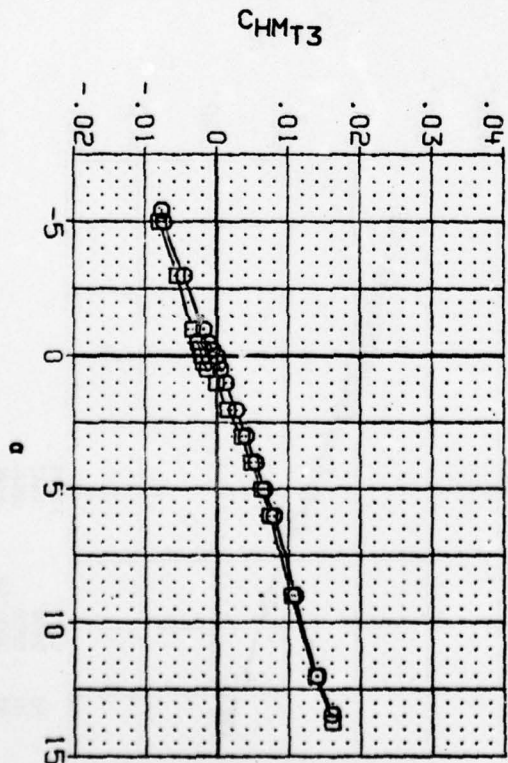
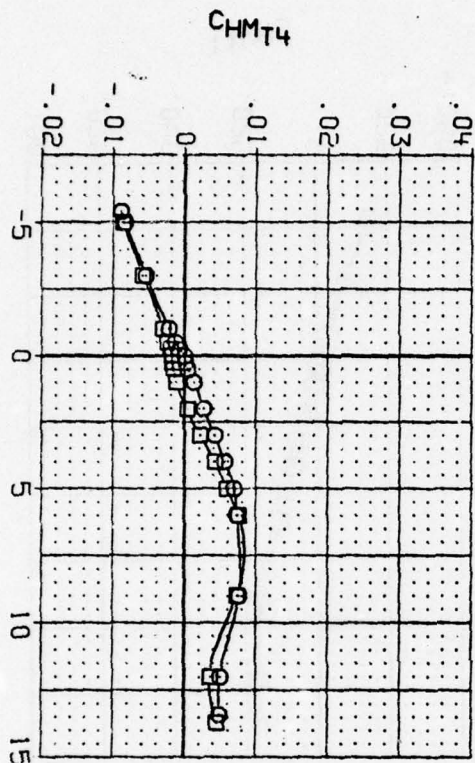
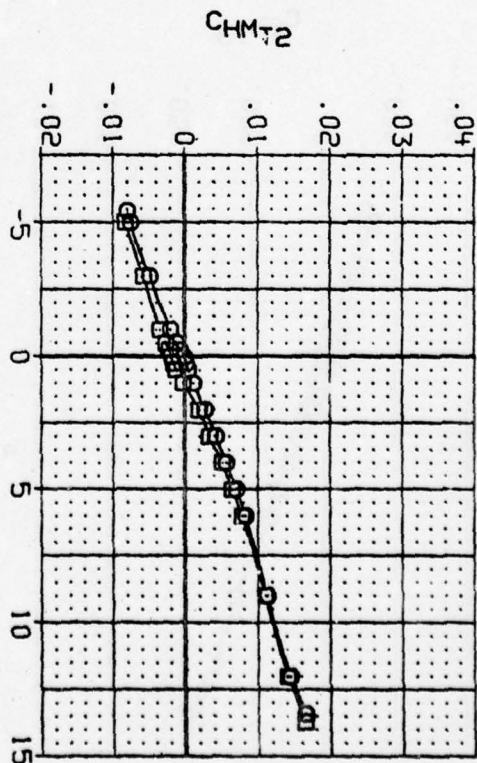
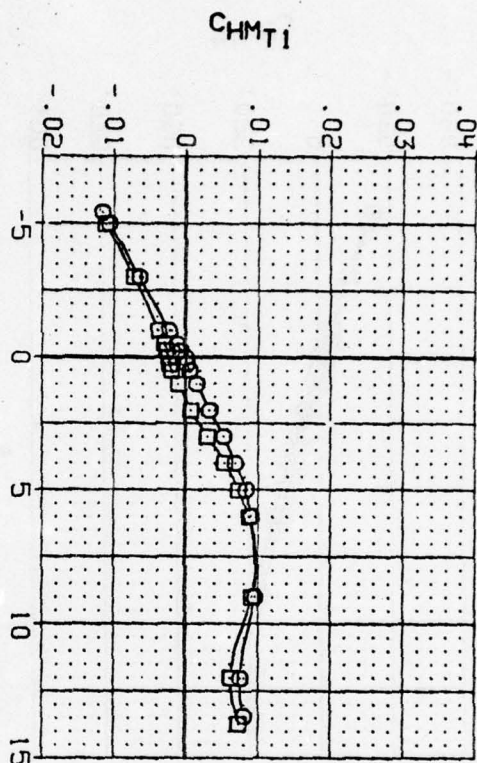


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH058) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC311
 (CXH059) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC311

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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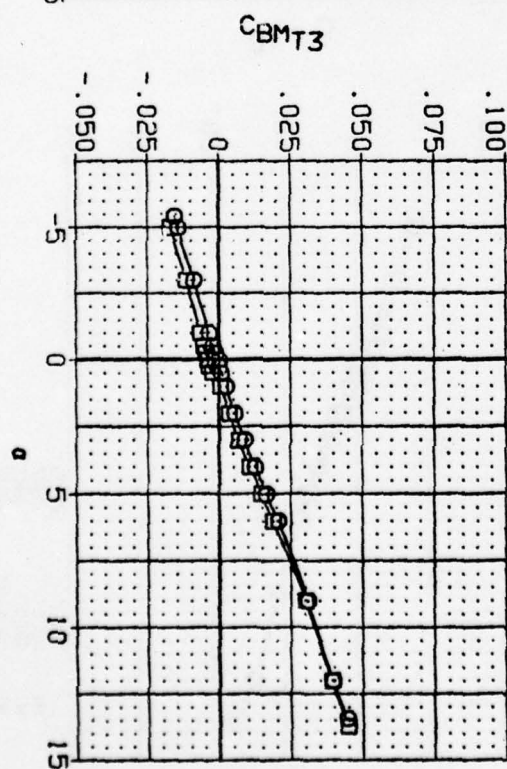
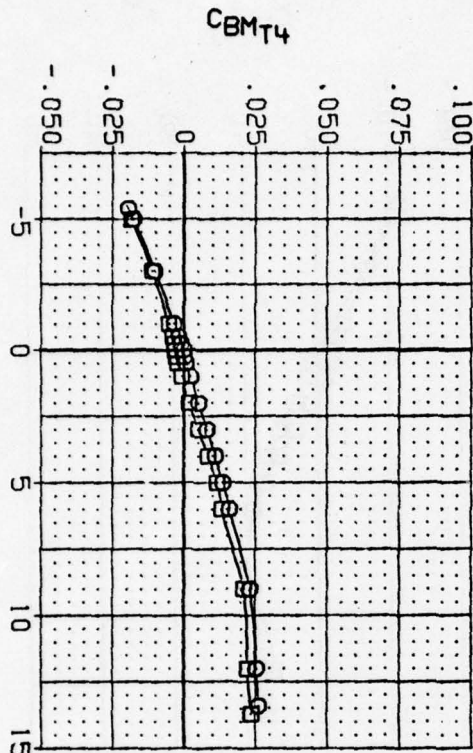
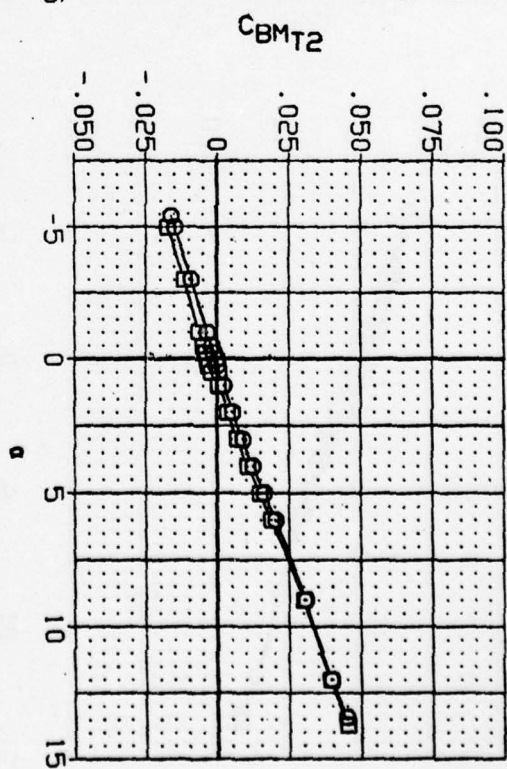
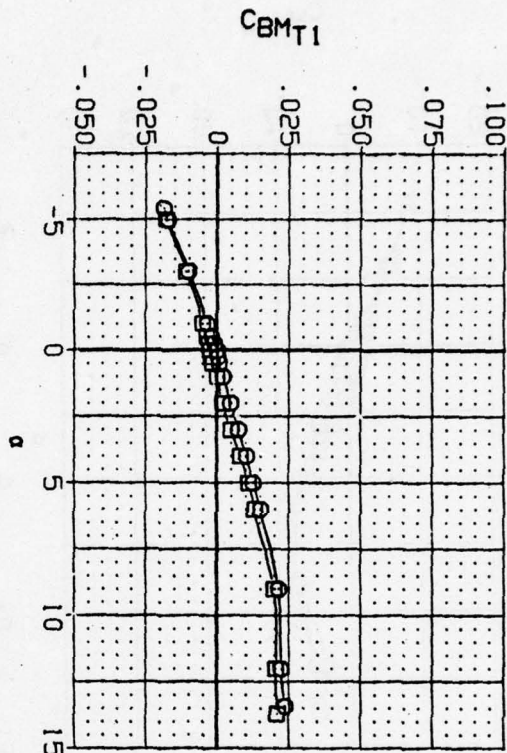


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=45
 (A) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CH058) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC371
 (CH059) ☐ AEDC V41A-C1A, CANARD CONTROL, BNIC371

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SPEC 19.6350 SQ. IN.
 LREF 5.0000 IN.
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 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
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EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
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 (A)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

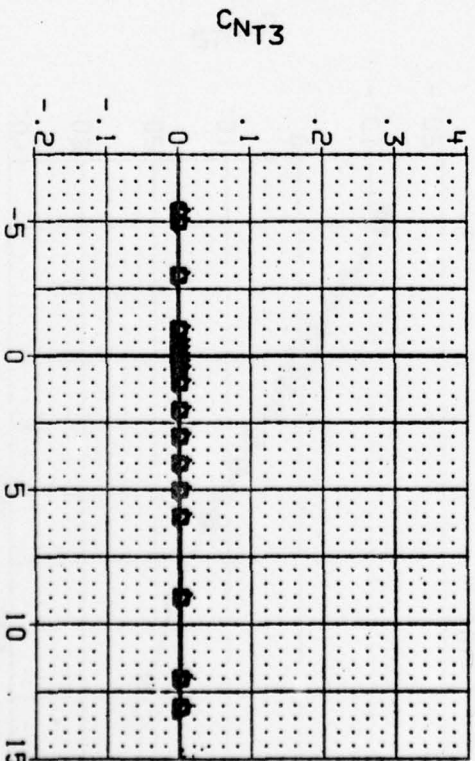
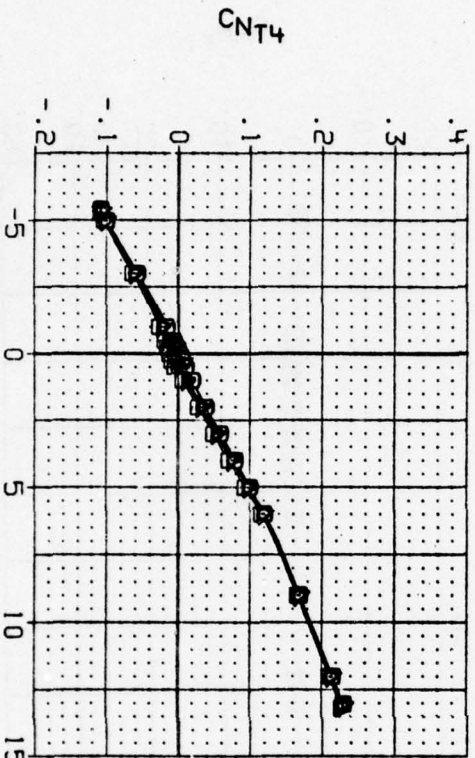
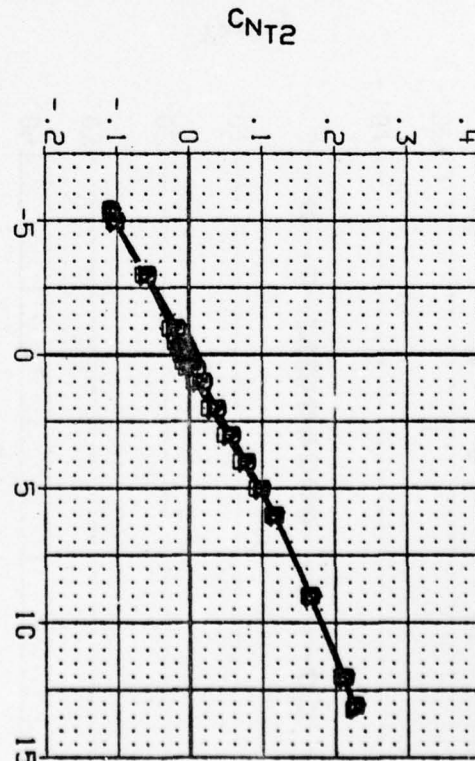
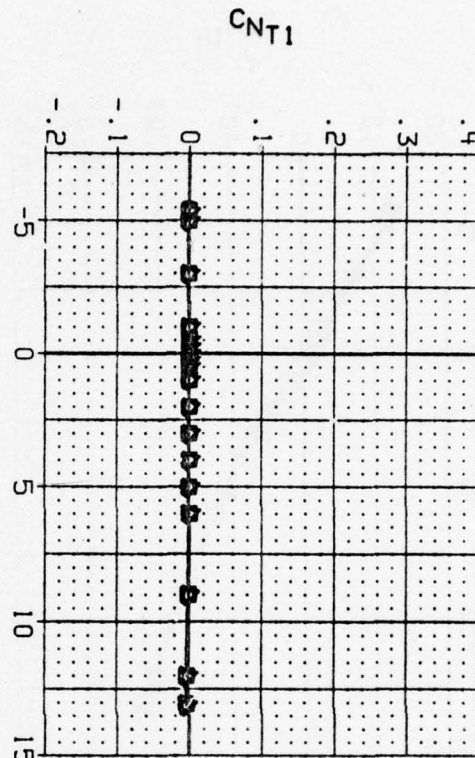
(AXH061) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111
 (AXH062) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111
 (AXH063) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111
 (AXH064) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111
 (AXH065) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111
 (AXH066) \square AEDC VAI-A-CIA, CANARD CONTROL, BNEC111

DCND1 DCND2 DCND3 DCND4

.000 -3.000 .000 -3.000
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 6.000 .000 6.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
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EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

(A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

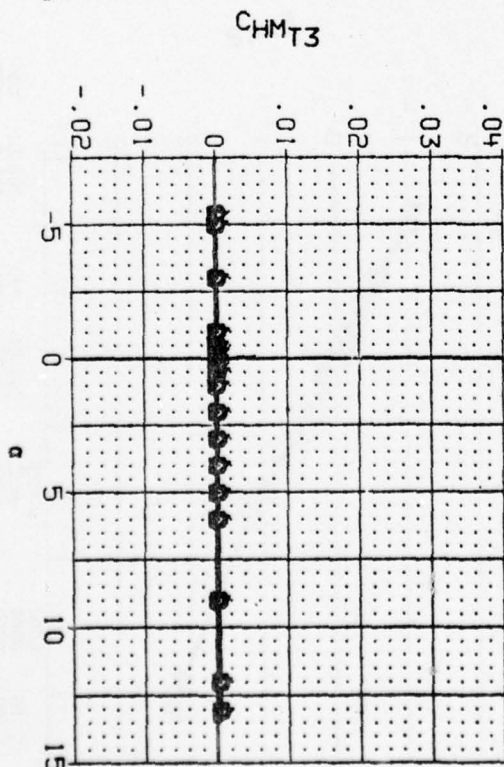
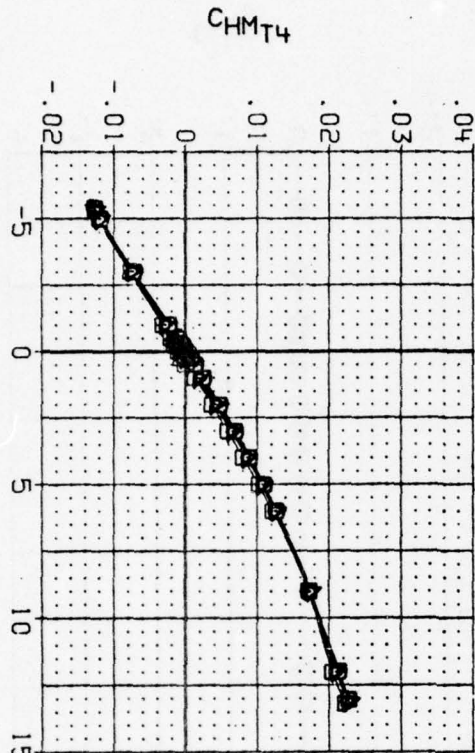
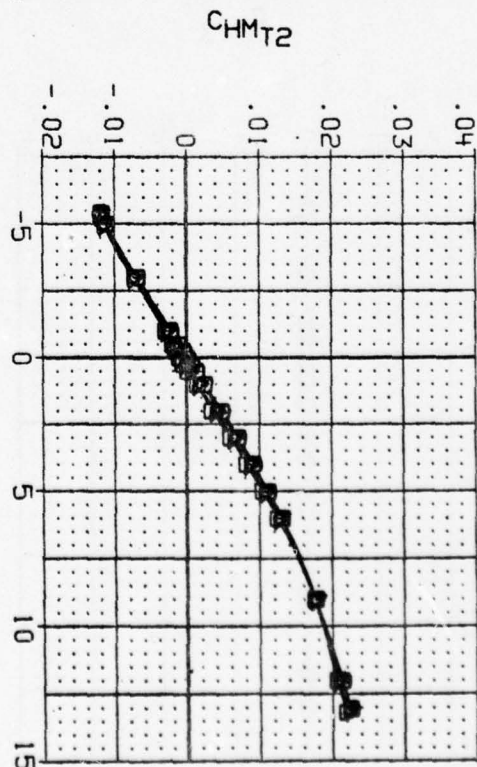
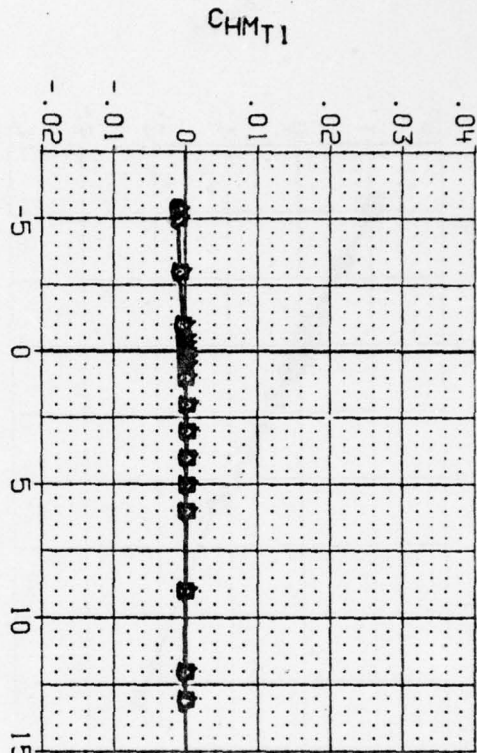
(CXH051)	□	AEDC V41A-C1A, CANARD CONTROL, BNEC111
(CXH052)	◇	AEDC V41A-C1A, CANARD CONTROL, BNEC111
(CXH053)	△	AEDC V41A-C1A, CANARD CONTROL, BNEC111
(CXH054)	▽	AEDC V41A-C1A, CANARD CONTROL, BNEC111
(CXH055)	□	AEDC V41A-C1A, CANARD CONTROL, BNEC111
(CXH056)	◇	AEDC V41A-C1A, CANARD CONTROL, BNEC111

DCND1 DCND2 DCND3 DCND4

.000	-3.000	.000	-3.000
.000	.000	.000	.000
.000	3.000	.000	3.000
.000	6.000	.000	6.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000

REFERENCE INFORMATION

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XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=0$
 (A) MACH = 3.00

AD-A033 783 ARMY MISSILE RESEARCH DEVELOPMENT AND ENGINEERING LAB--ETC F/G 20/4
AN EXPERIMENTAL INVESTIGATION OF THE AERODYNAMIC CHARACTERISTIC--ETC(U)
OCT 76 J R BURT

ARMY MISSILE RESEARCH DEVELOPMENT AND ENGINEERING LAB--ETC F/G 20/4
AN EXPERIMENTAL INVESTIGATION OF THE AERODYNAMIC CHARACTERISTIC--ETC(U)
OCT 76 J R BURT

RD-77-5

NL

AD
A033783

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A033783



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DATE
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2-77

DATE
FILMED

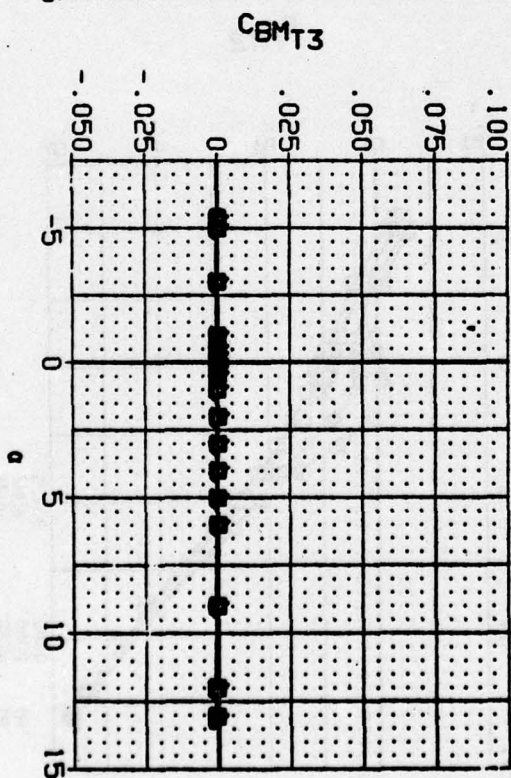
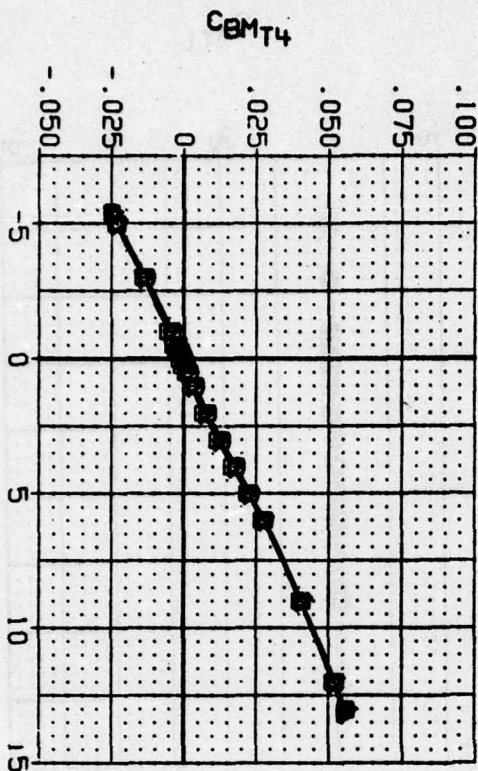
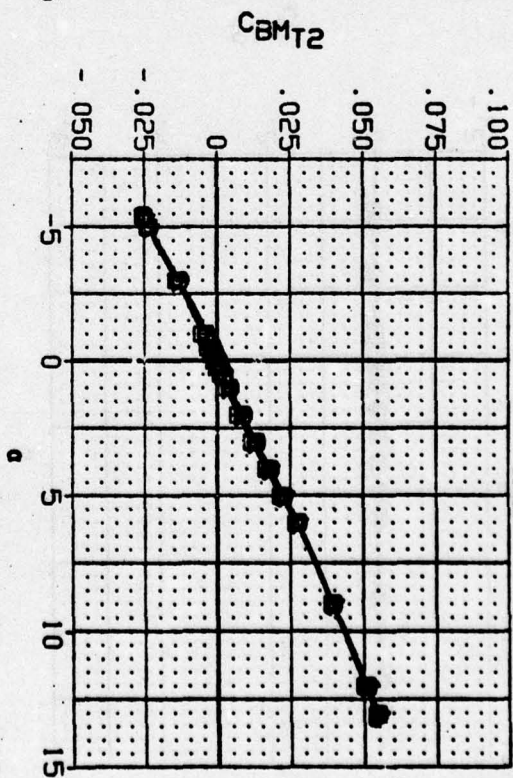
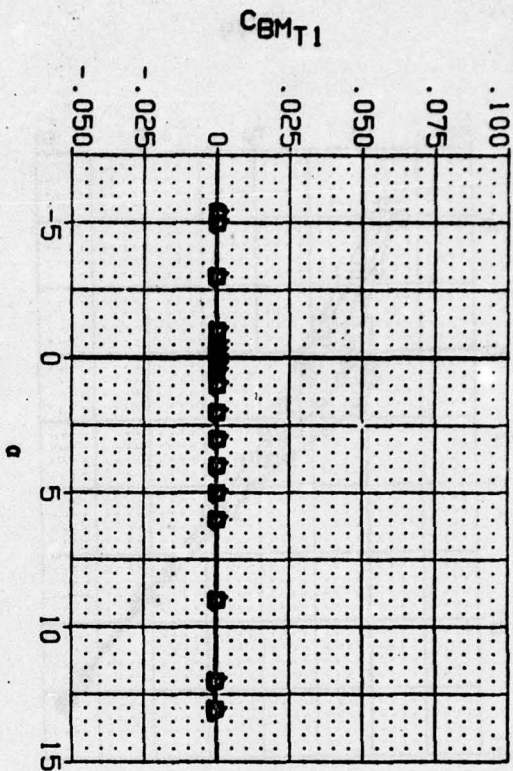
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DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CH061)	□	AEDC W1A-C1A, CANARD CONTROL, BRAC111
(CH062)	◇	AEDC W1A-C1A, CANARD CONTROL, BRAC111
(CH063)	△	AEDC W1A-C1A, CANARD CONTROL, BRAC111
(CH064)	▽	AEDC W1A-C1A, CANARD CONTROL, BRAC111
(CH065)	□	AEDC W1A-C1A, CANARD CONTROL, BRAC111
(CH066)	◇	AEDC W1A-C1A, CANARD CONTROL, BRAC111

REFERENCE INFORMATION

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.000	6.000	.000	6.000	5.0000	5.0000	5.0000	5.0000	.0000	IN.
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EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

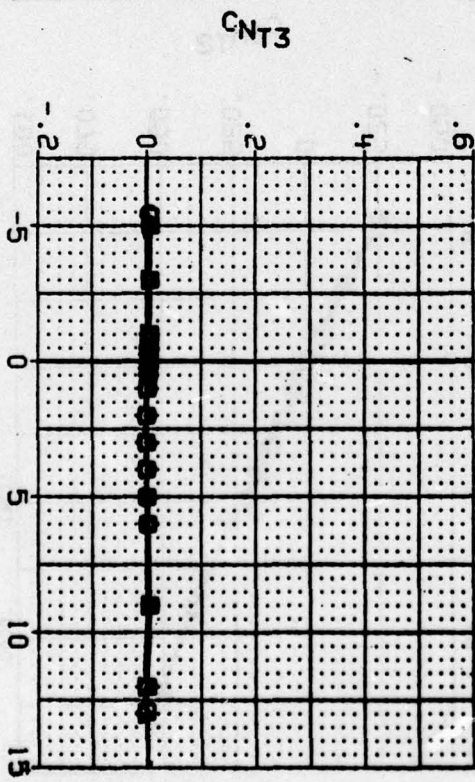
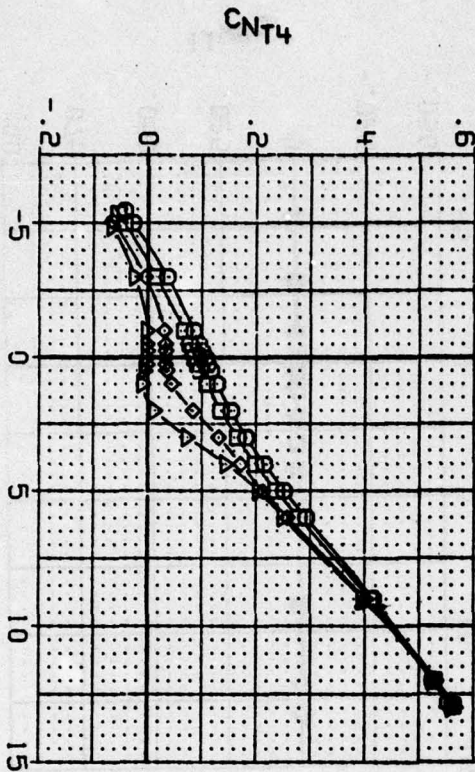
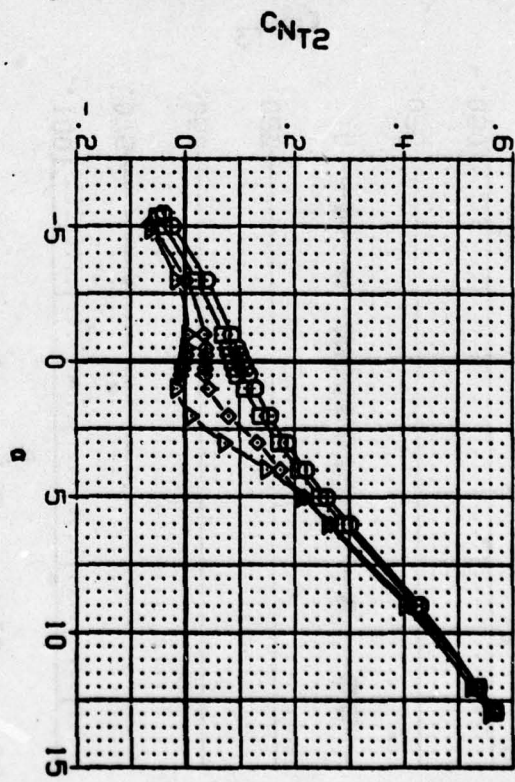
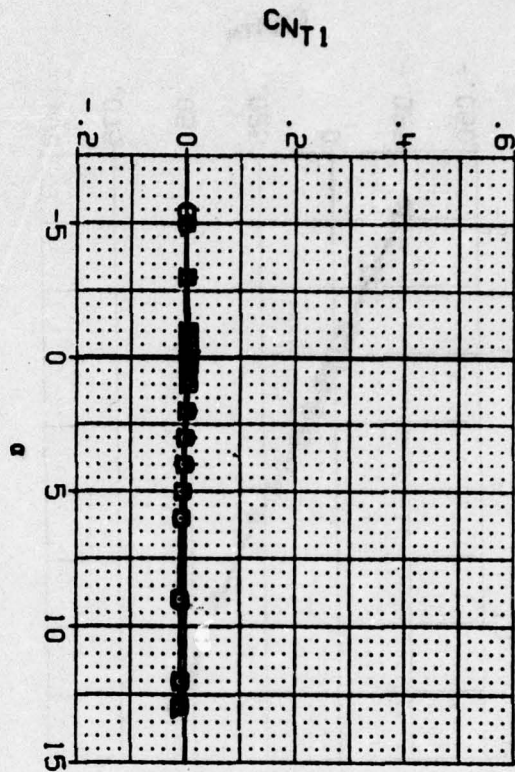
PHITAL=0 PHICND=0

(A) MACH = 3.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH067) \square AEDC W1A-C1A, CANARD CONTROL, BM/C6T1
 (AXH068) \square AEDC W1A-C1A, CANARD CONTROL, BM/C6T1
 (AXH069) \square AEDC W1A-C1A, CANARD CONTROL, BM/C6T1
 (AXH070) \triangle AEDC W1A-C1A, CANARD CONTROL, BM/C6T1

DOCD1 DOCD2 DOCD3 DOCD4
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
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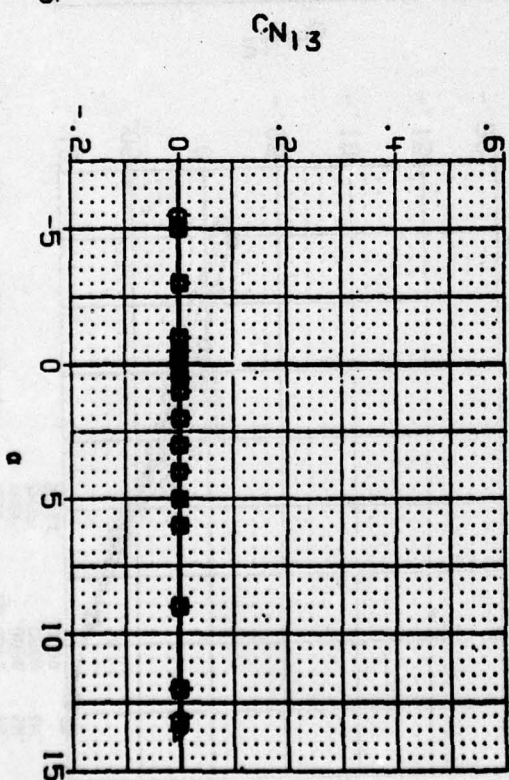
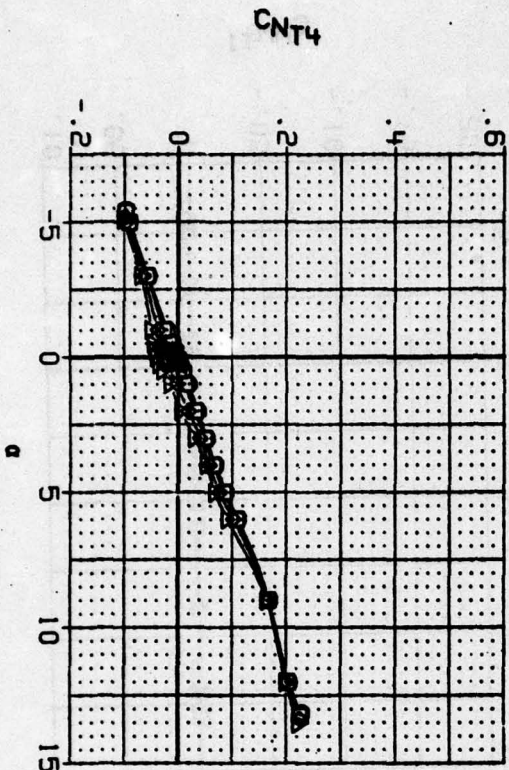
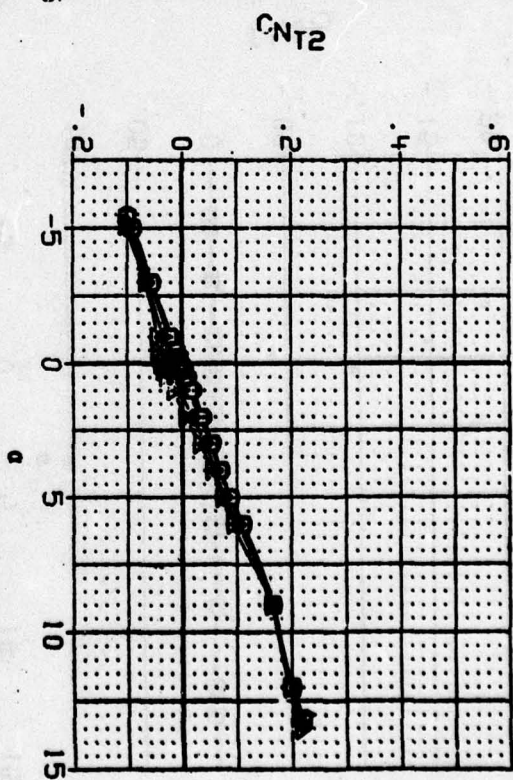
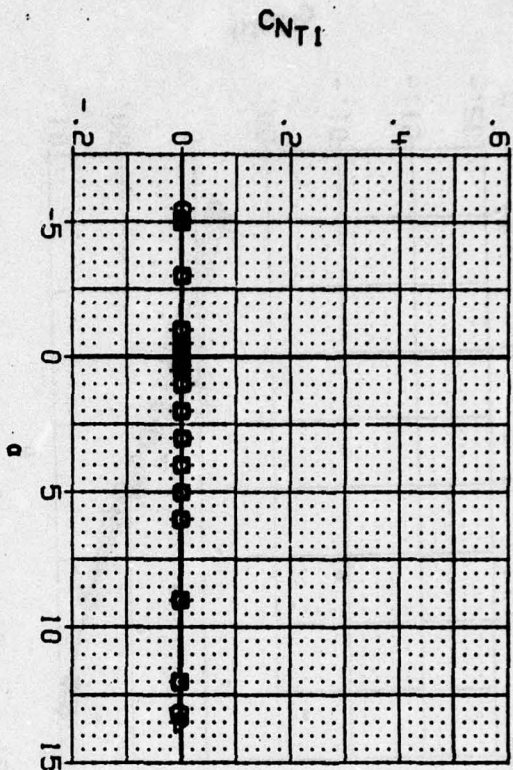


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\Phi_{H1A}=0$ $\Phi_{H1C}=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH067) \square AEDC W1A-C1A, CANARD CONTROL, BMMCS11
 (AXH068) \square AEDC W1A-C1A, CANARD CONTROL, BMMCS11
 (AXH069) \square AEDC W1A-C1A, CANARD CONTROL, BMMCS11
 (AXH070) Δ AEDC W1A-C1A, CANARD CONTROL, BMMCS11

DDND1 DDND2 DDND3 DDND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 3.000
 .000 15.000 .000 9.000
 .000 .000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CXX057) \square AEDC W1A-C1A, CANARD CONTROL, BNC6T1

(CXX058) \square AEDC W1A-C1A, CANARD CONTROL, BNC6T1

(CXX059) \triangle AEDC W1A-C1A, CANARD CONTROL, BNC6T1

(CXX070) \triangle AEDC W1A-C1A, CANARD CONTROL, BNC6T1

DN01 DN02 DN03 DN04

.000 .000 .000 .000

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.000 9.000 .000 .000

.000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

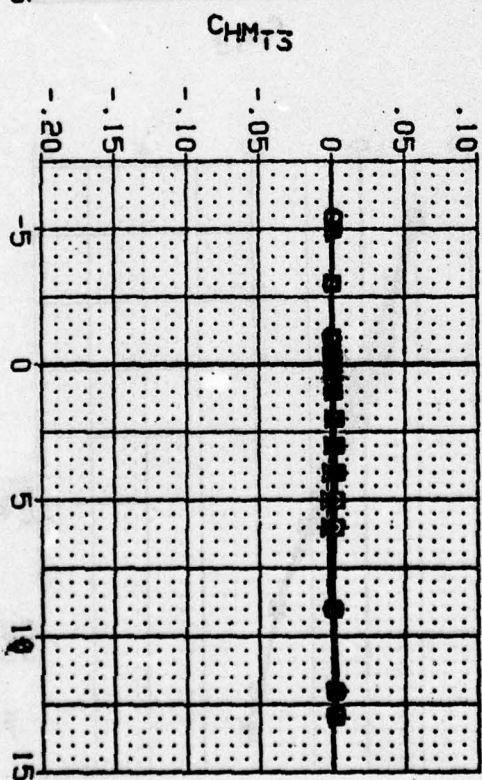
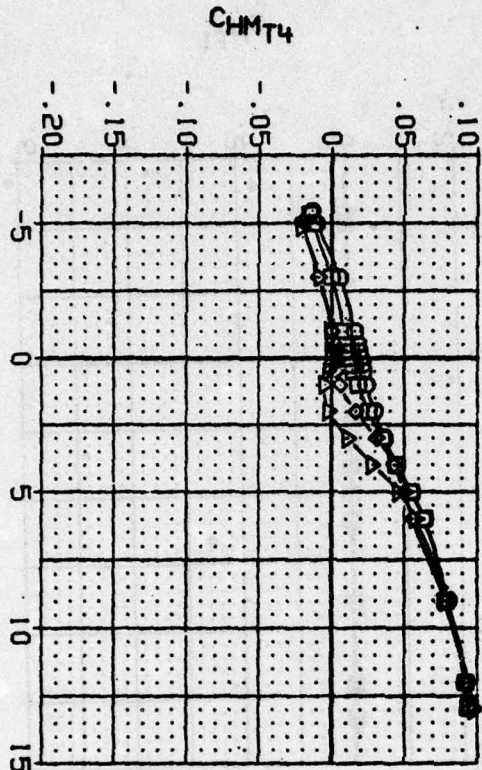
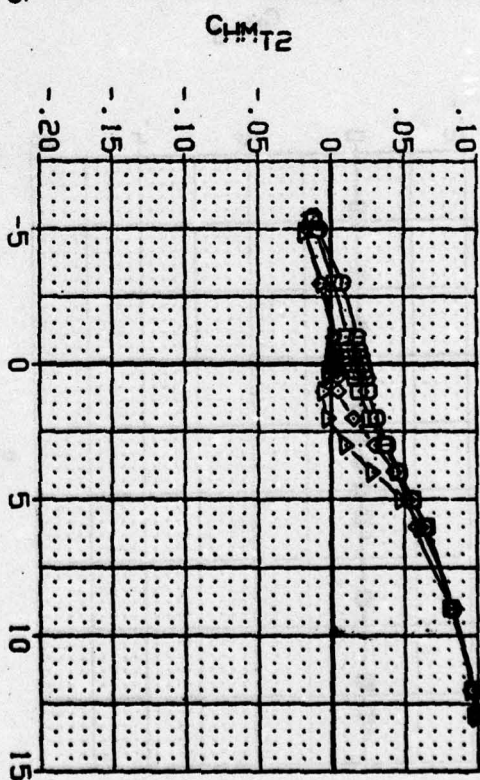
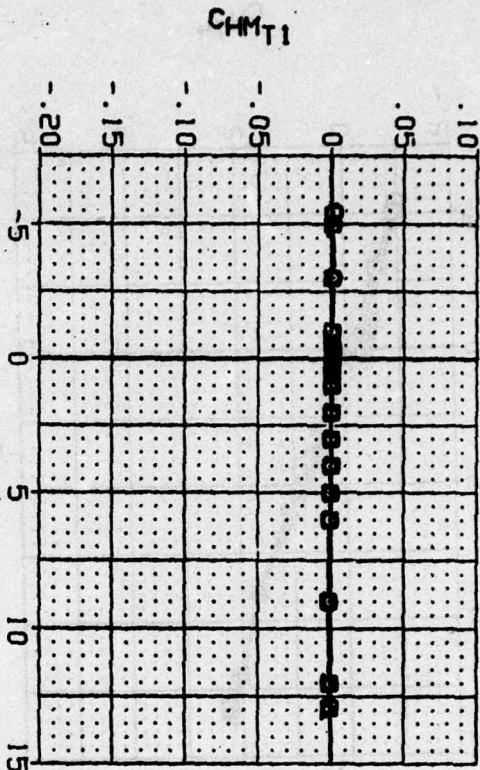
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WREF 25.0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CH067) \square AEDC WJA-CIA, CANARD CONTROL, BMMCT1

(CH068) \square AEDC WJA-CIA, CANARD CONTROL, BMMCT1

(CH069) \square AEDC WJA-CIA, CANARD CONTROL, BMMCT1

(CH070) \triangle AEDC WJA-CIA, CANARD CONTROL, BMMCT1

DO01 DO02 DO03 DO04

.000 .000 .000 .000

.000 3.000 .000 3.000

.000 .000 .000 3.000

.000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

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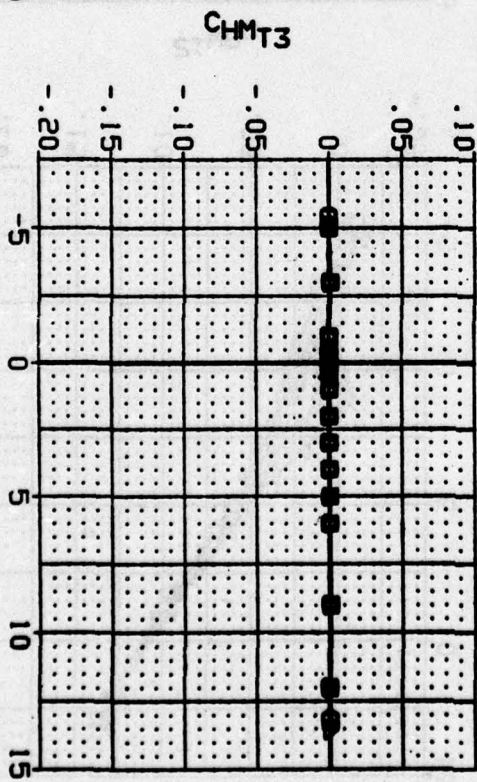
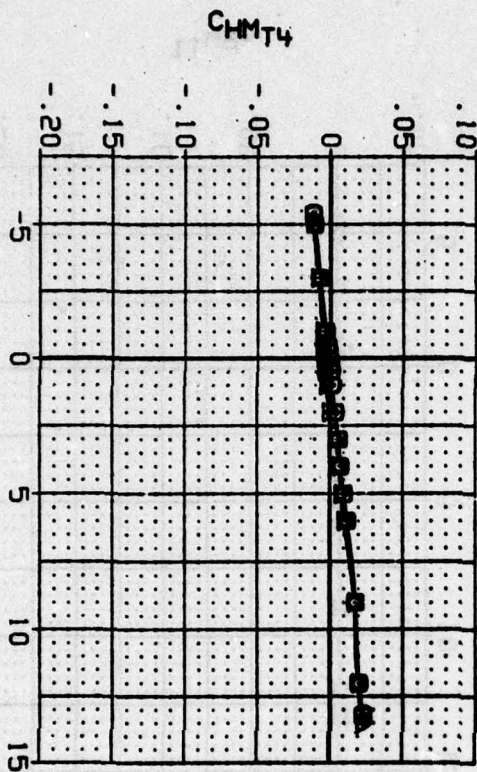
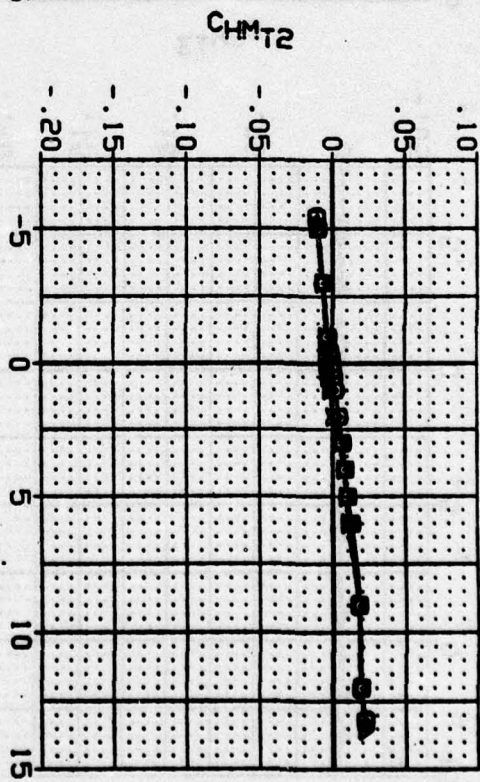
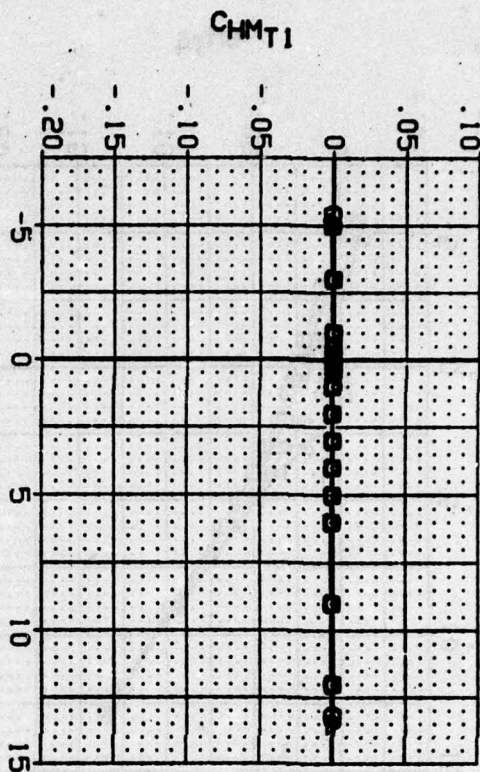
BRF 5.0000 IN.

XREF 26.0000 IN.

YREF .0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

$PHITAL=0$ $PHICND=0$

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

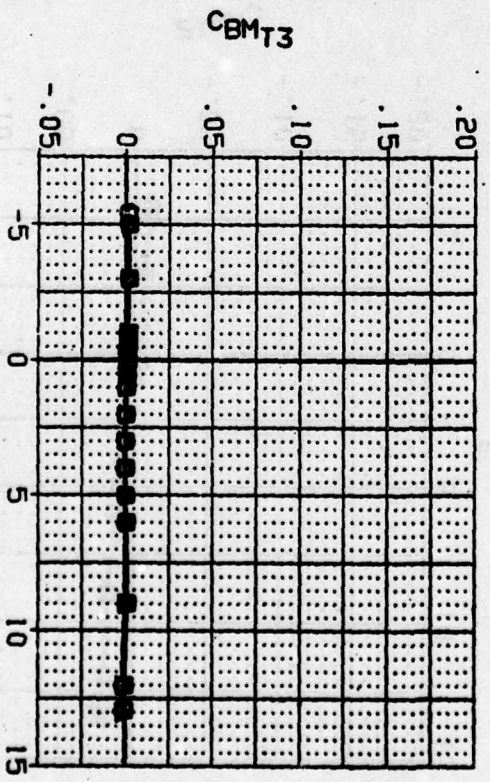
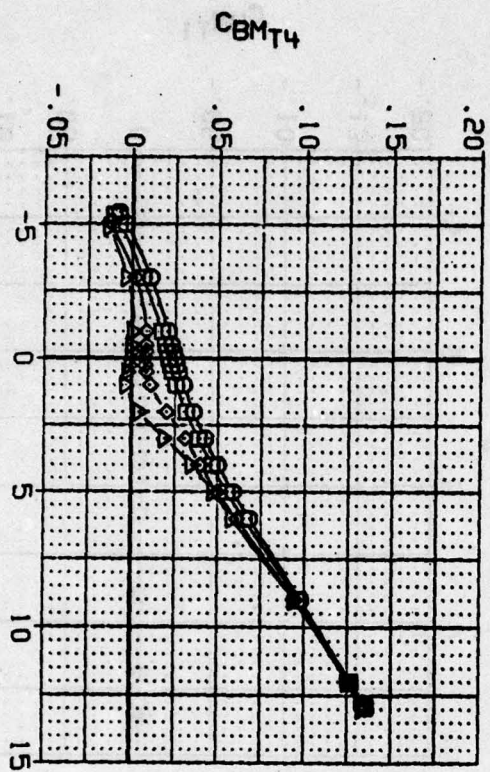
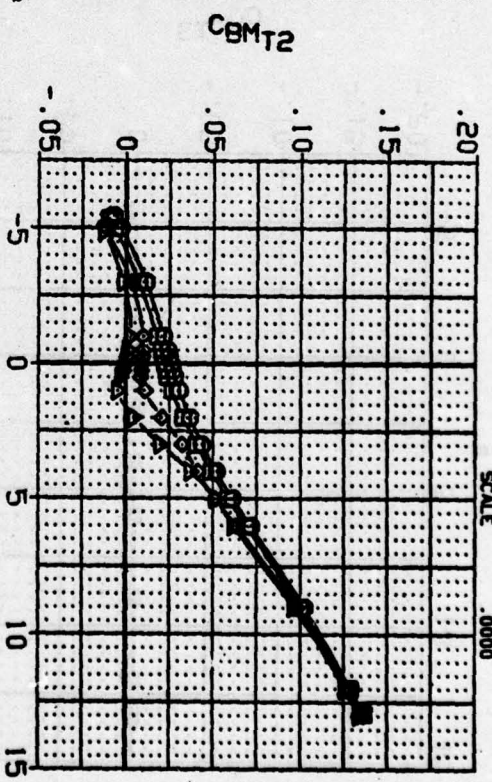
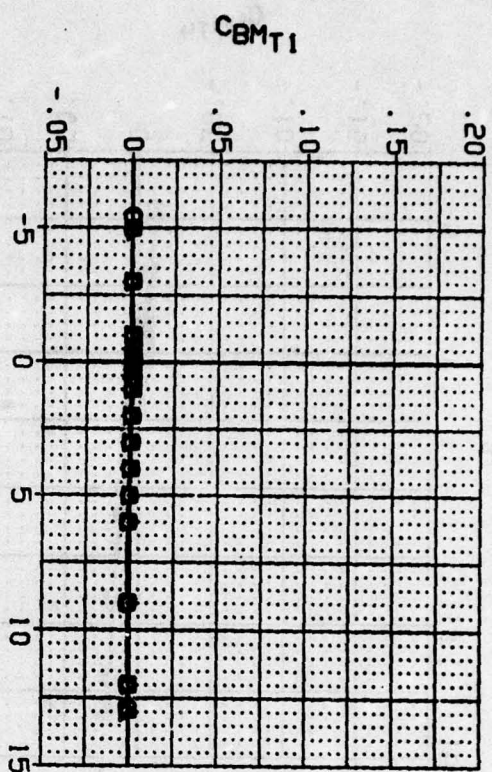
(CXH067)	○	AEDC WIA-CIA, CANARD CONTROL, BMH0671
(CXH068)	□	AEDC WIA-CIA, CANARD CONTROL, BMH0681
(CXH069)	◇	AEDC WIA-CIA, CANARD CONTROL, BMH0691
(CXH070)	△	AEDC WIA-CIA, CANARD CONTROL, BMH0701

DOCK1 DOCK2 DOCK3 DOCK4

.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000

REFERENCE INFORMATION

SEF	19.6350	50. IN.
LEF	5.0000	IN.
BREF	5.0000	IN.
XREF	25.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\Phi_{HIAL}=0$ $\Phi_{HICND}=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(C84067) \square AEDC W1A-C1A, CANARD CONTROL, BMICST1

(C84068) \square AEDC W1A-C1A, CANARD CONTROL, BMICST1

(C84069) \square AEDC W1A-C1A, CANARD CONTROL, BMICST1

(C84070) Δ AEDC W1A-C1A, CANARD CONTROL, BMICST1

DOCD1 DOCD2 DOCD3 DOCD4

.000 .000 .000 .000

.000 3.000 .000 3.000

.000 9.000 .000 9.000

.000 15.000 .000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

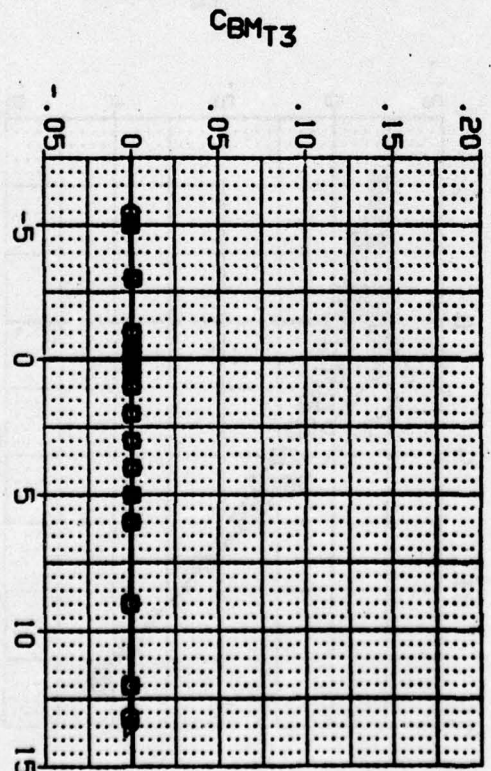
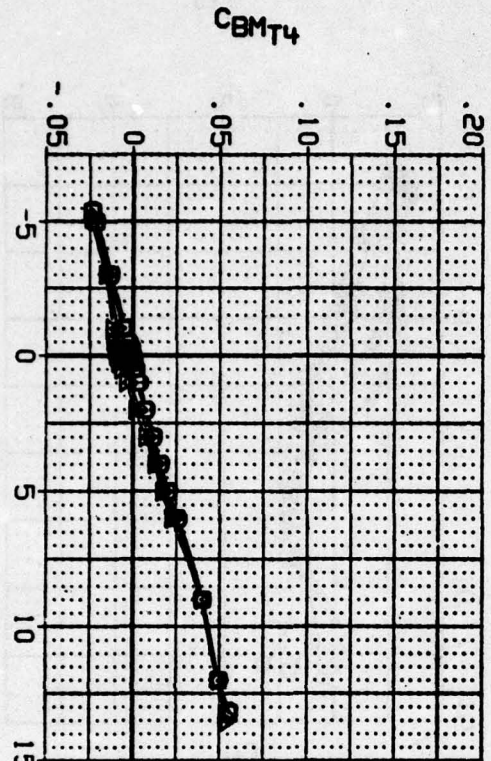
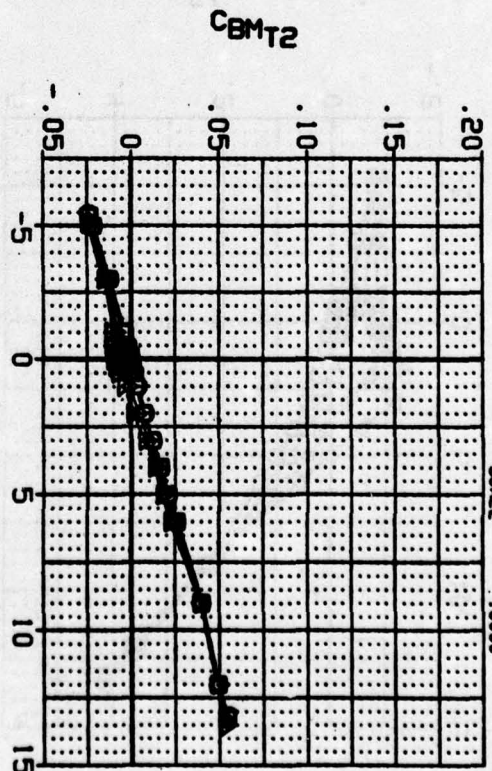
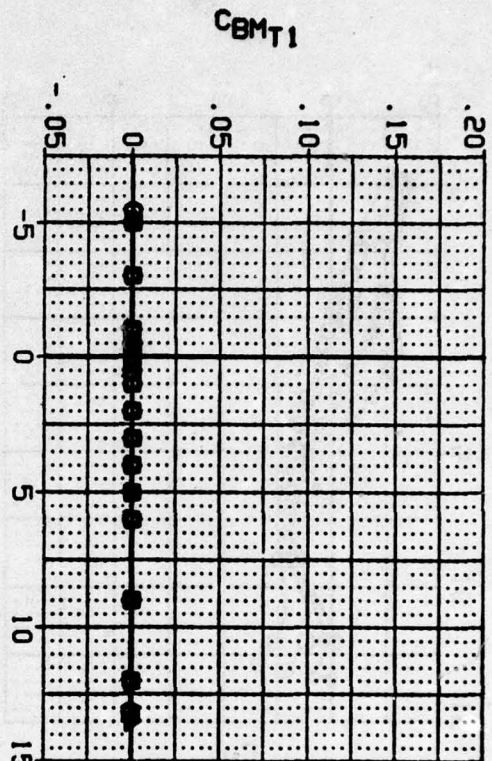
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WREF 26.0000 IN.

ZREF .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

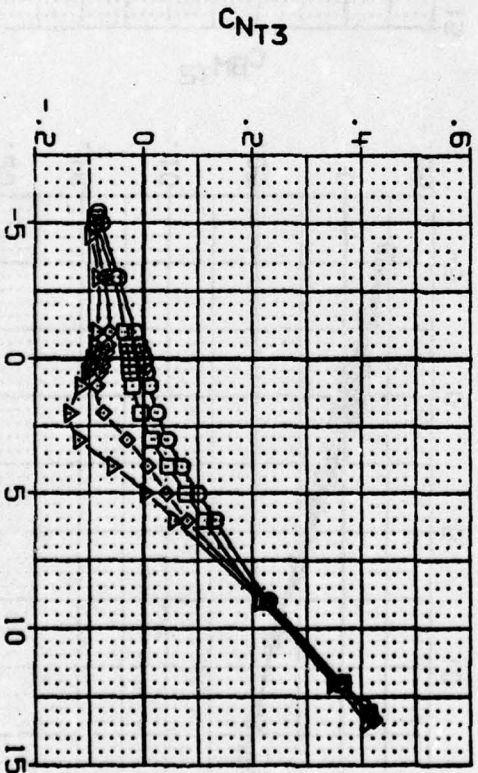
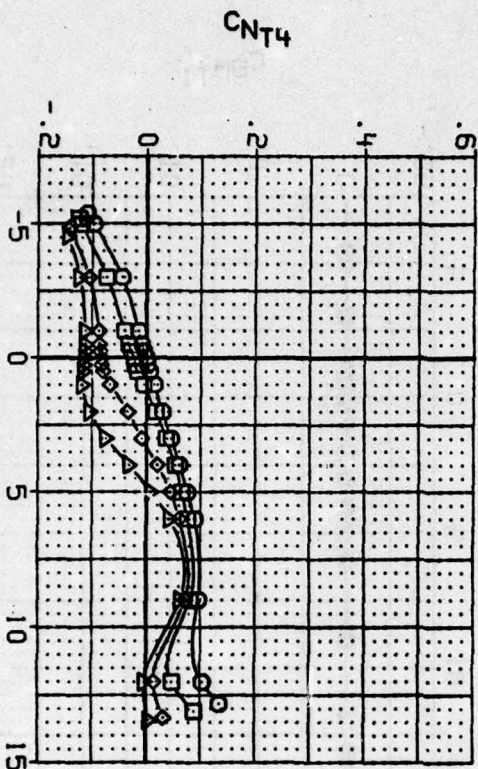
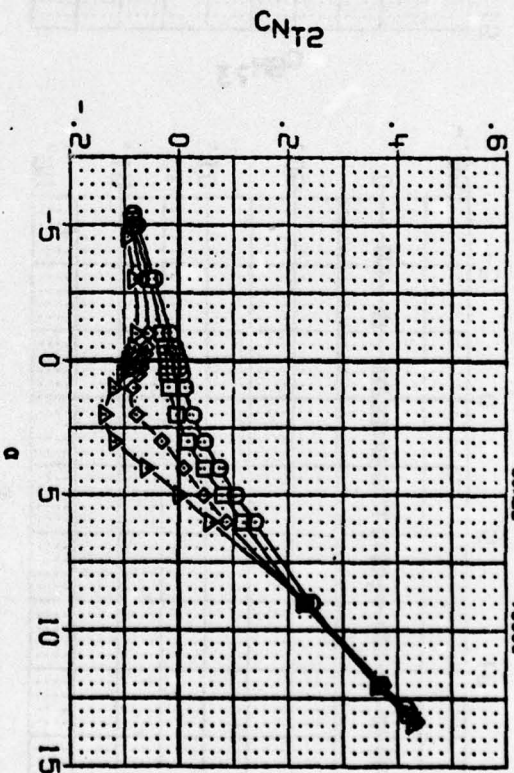
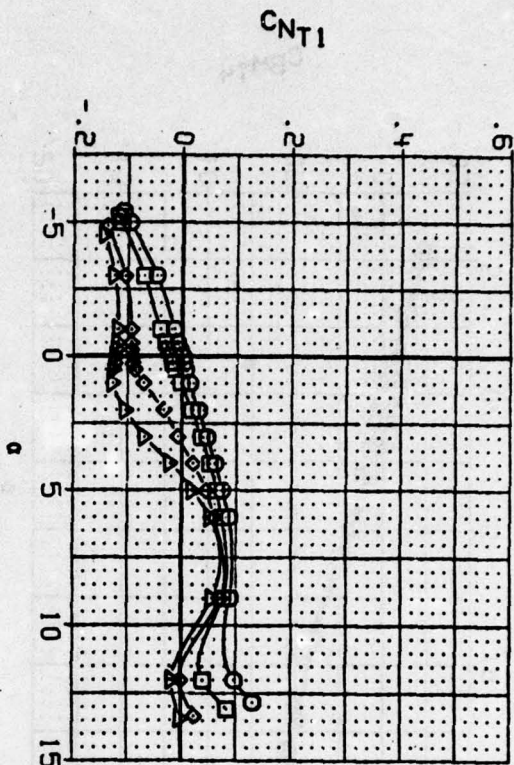
PHITAL=0 PHICND=0

(B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH072) \square AEDC WIA-CIA, CANARD CONTROL, BH4C6T1
 (AXH073) \square AEDC WIA-CIA, CANARD CONTROL, BH4C6T1
 (AXH074) \square AEDC WIA-CIA, CANARD CONTROL, BH4C6T1
 (AXH075) Δ AEDC WIA-CIA, CANARD CONTROL, BH4C6T1

DCND1 .000
 3.000
 9.000
 15.000
 DCND2 .000
 3.000
 9.000
 15.000
 DCND3 .000
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REFERENCE INFORMATION
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 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PH1AL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

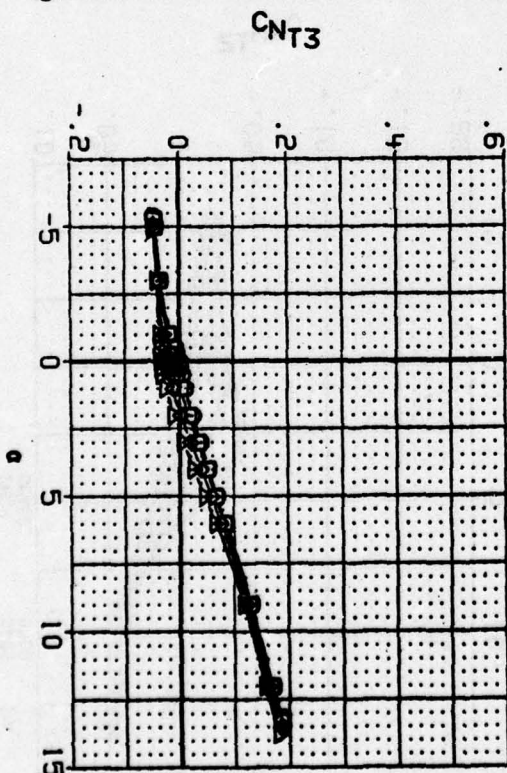
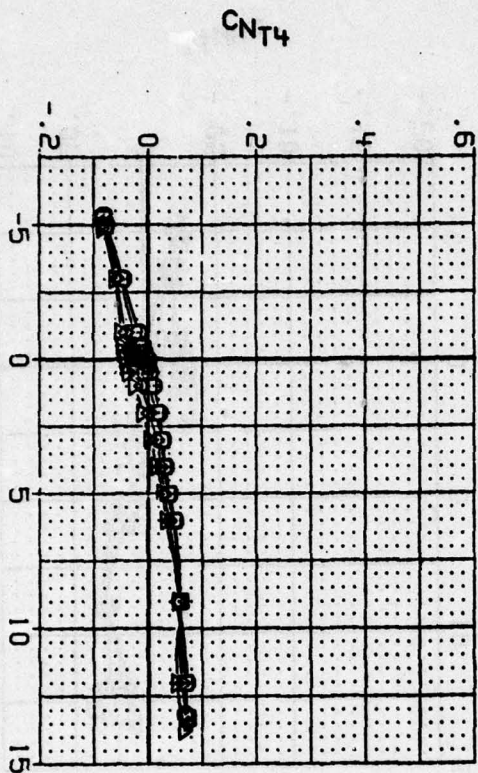
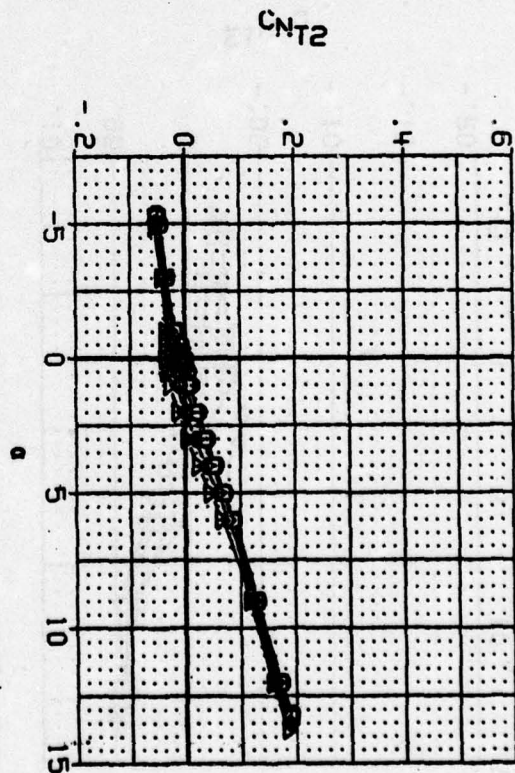
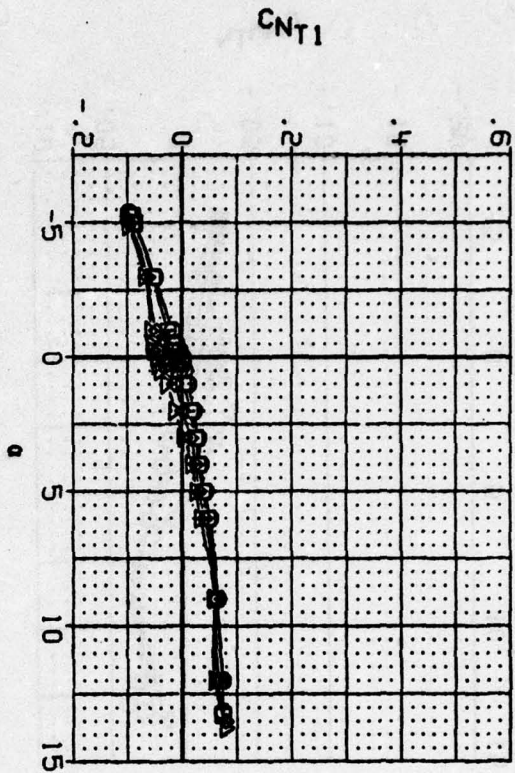
(A33472) □ AEDC W1A-C1A, CANARD CONTROL, BANCST1
 (A33473) ○ AEDC W1A-C1A, CANARD CONTROL, BANCST1
 (A33474) △ AEDC W1A-C1A, CANARD CONTROL, BANCST1
 (A33475) △ AEDC W1A-C1A, CANARD CONTROL, BANCST1

DCND1 DCND2 DCND3 DCND4

3.000 3.000 3.000 3.000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SEF 19.6350 SQ. IN.
 LREF 3.0000 IN.
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 YREF .0000 IN.
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 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

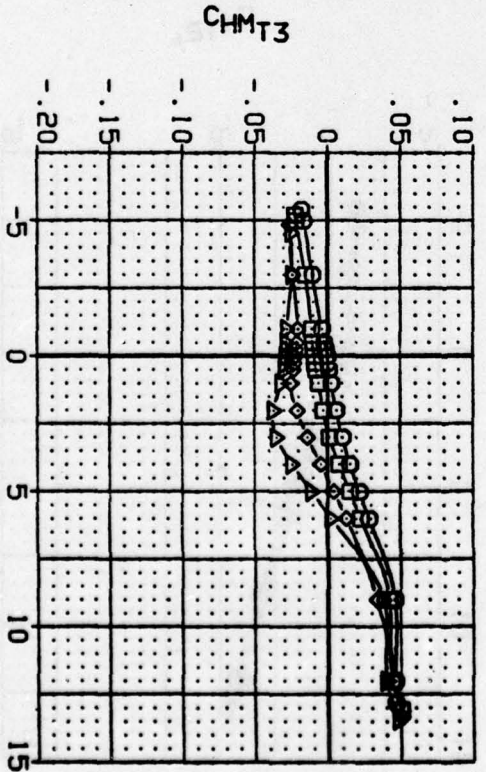
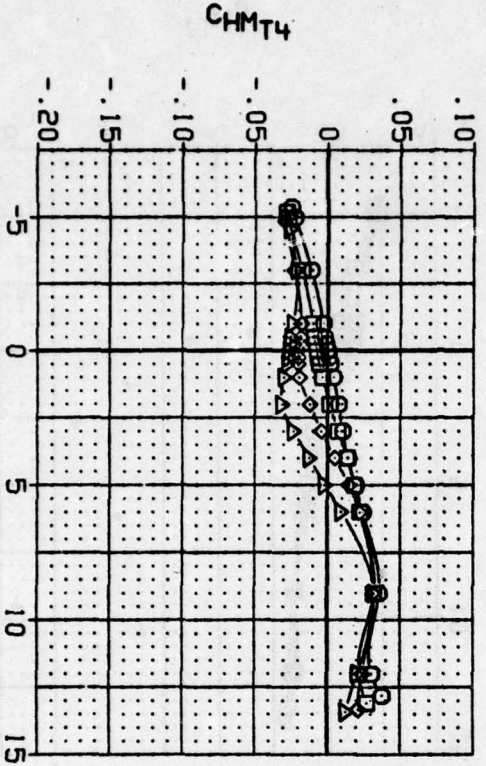
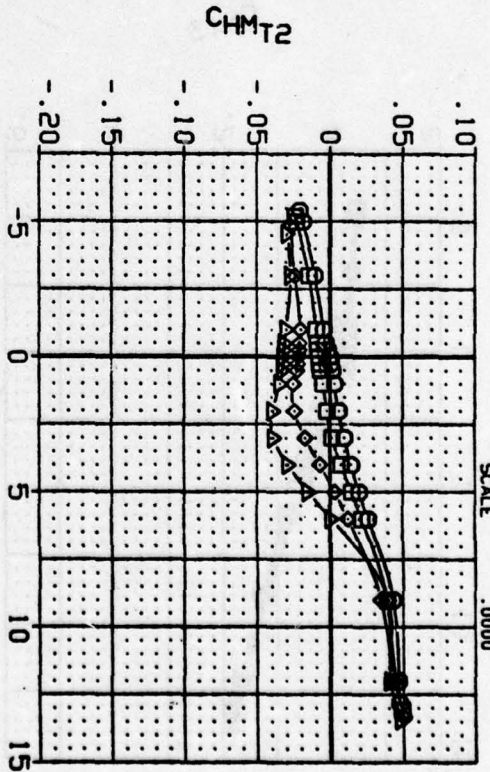
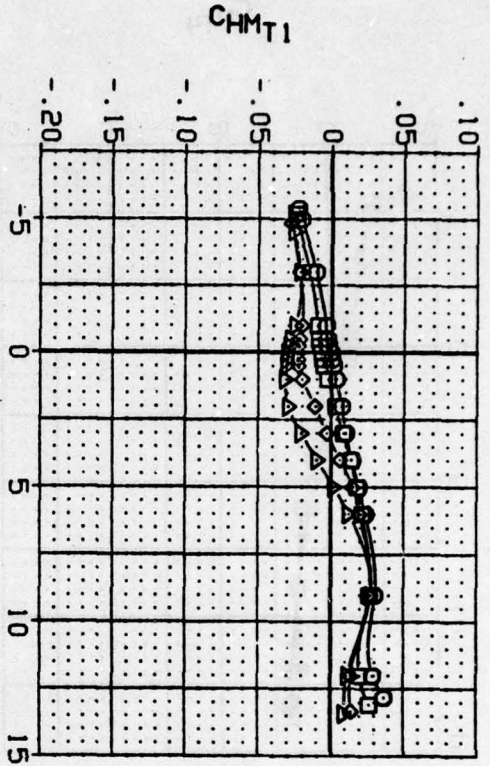
PHITAL=0 PHICND=45

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (CH073) \square AEDC W1A-C1A, CANARD CONTROL, BH4C6T1
 (CH074) \square AEDC W1A-C1A, CANARD CONTROL, BH4C6T1
 (CH075) \triangle AEDC W1A-C1A, CANARD CONTROL, BH4C6T1

COND1 COND2 COND3 COND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 BREF 5.0000 IN.
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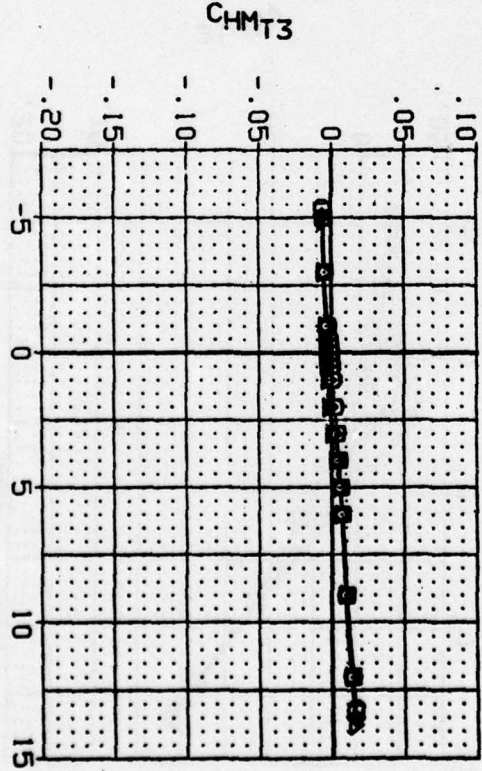
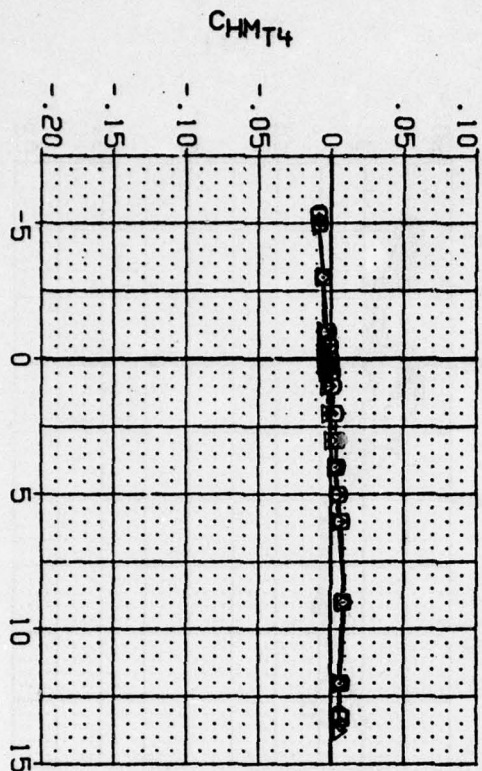
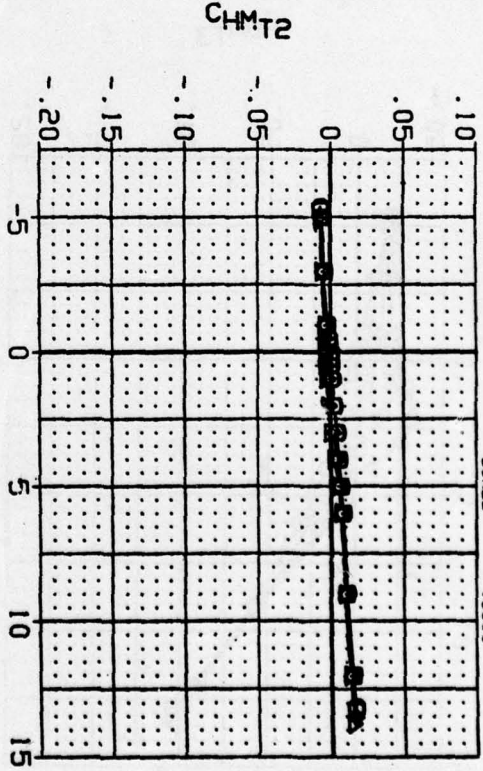
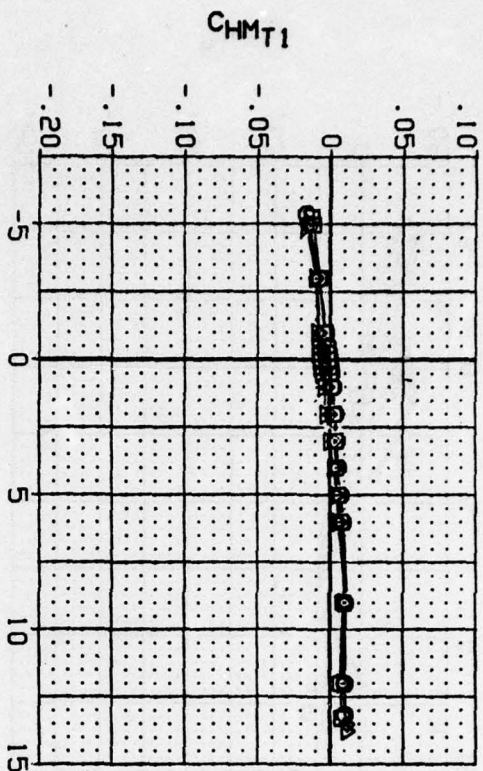


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=45$
 (A) MACH = 1.51

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CXH072)	○	AEDC W1A-C1A, CANARD CONTROL, BWHCST1
(CXH073)	□	AEDC W1A-C1A, CANARD CONTROL, BWHCST1
(CXH074)	◇	AEDC W1A-C1A, CANARD CONTROL, BWHCST1
(CXH075)	△	AEDC W1A-C1A, CANARD CONTROL, BWHCST1

DCND1	DCND2	DCND3	DCND4
.000	.000	.000	.000
3.000	3.000	3.000	3.000
9.000	9.000	9.000	9.000
15.000	15.000	15.000	15.000

REFERENCE INFORMATION	
SREF	19.6350
LREF	5.0000
BREF	5.0000
XREF	25.0000
YREF	.0000
ZREF	.0000
SCALE	.0000
	IN.



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=45
 (B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

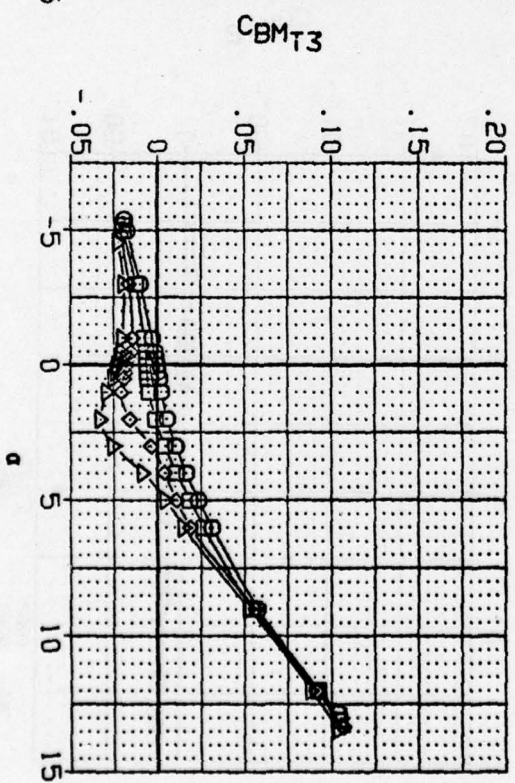
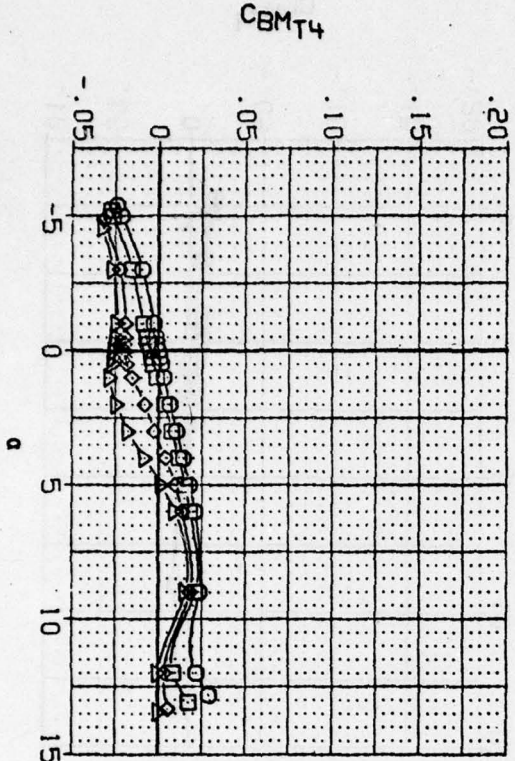
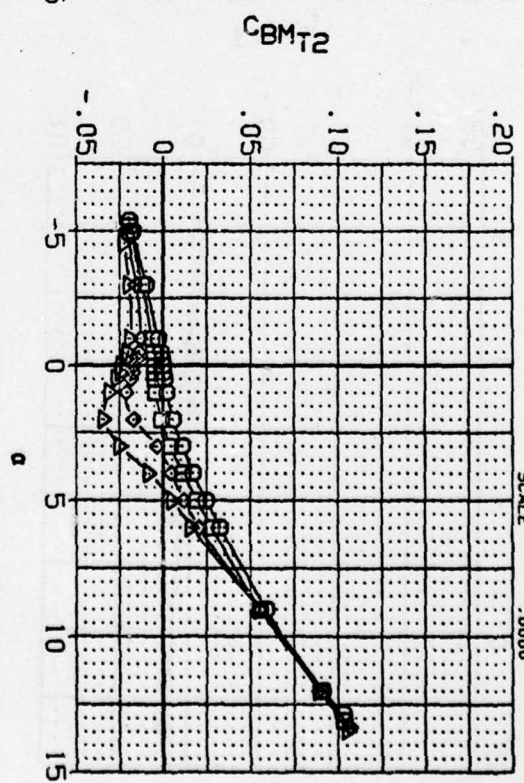
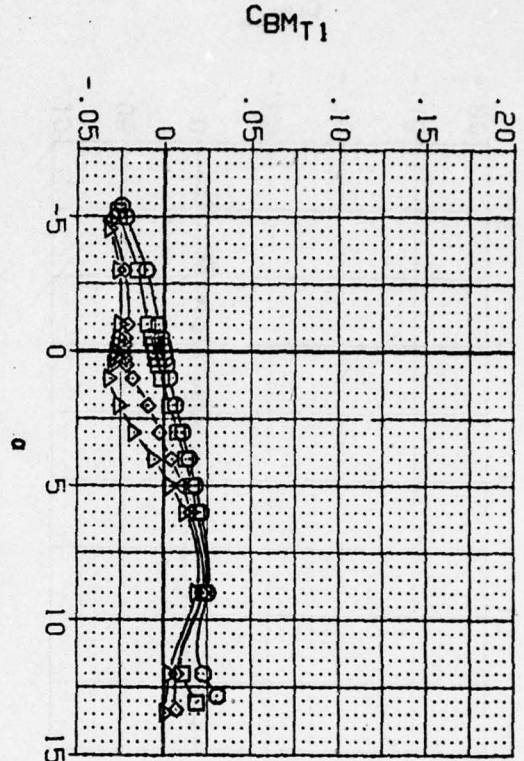
(CXH072) \square AEDC V41A-C1A, CANARD CONTROL, BN4C6T1
 (CXH073) \square AEDC V41A-C1A, CANARD CONTROL, BN4C6T1
 (CXH074) \square AEDC V41A-C1A, CANARD CONTROL, BN4C6T1
 (CXH075) \triangle AEDC V41A-C1A, CANARD CONTROL, BN4C6T1

DCND1 DCND2 DCND3 DCND4

3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 IN.
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 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

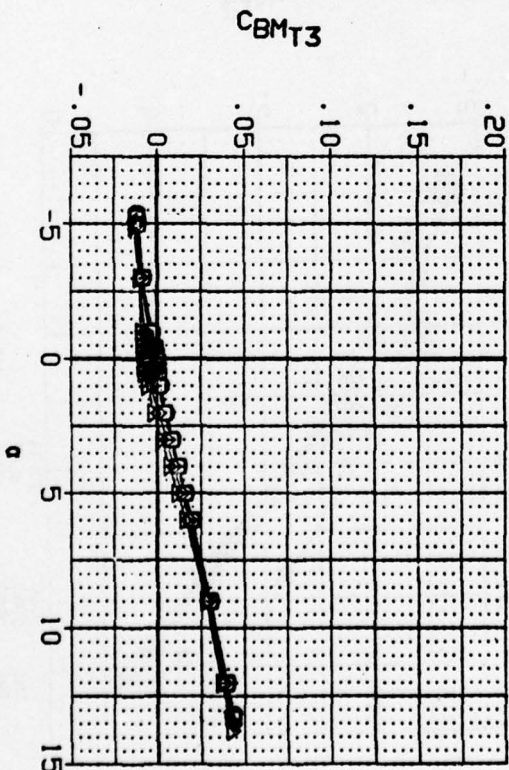
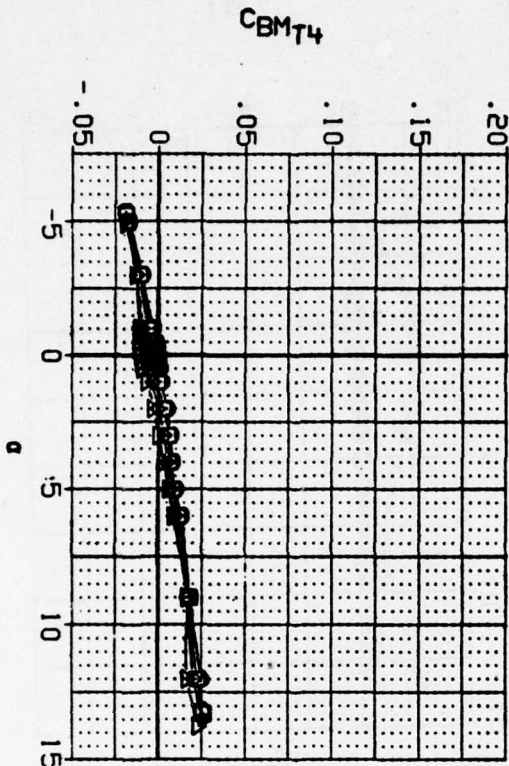
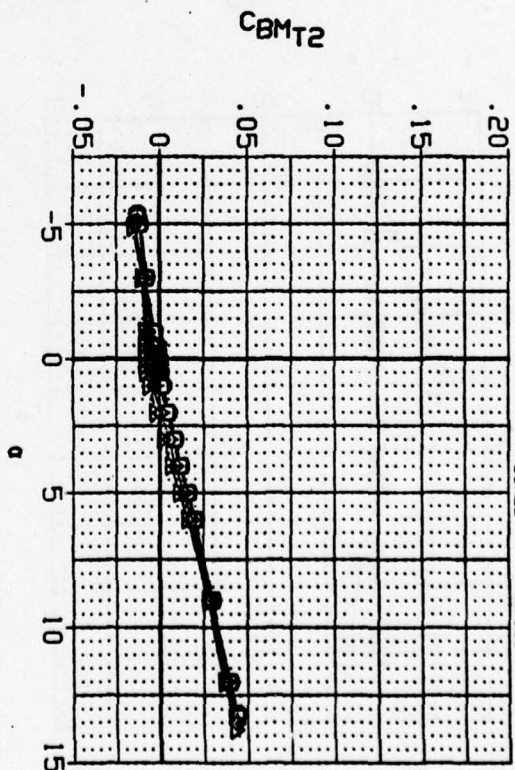
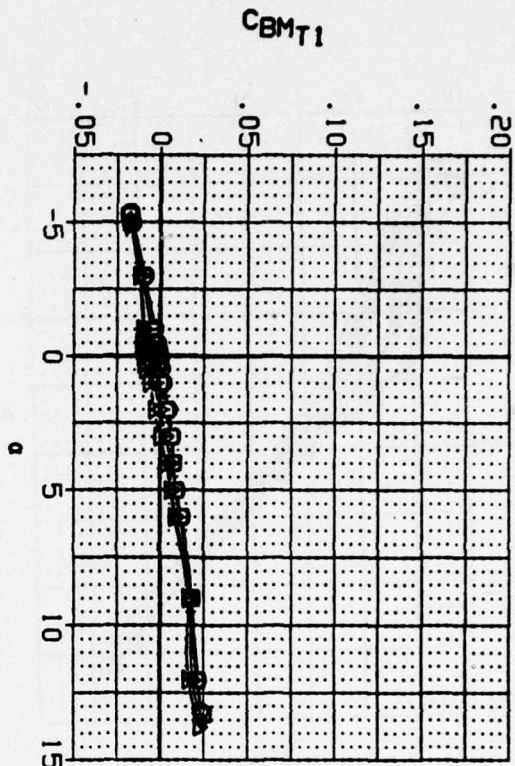
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(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (CH073) \square AEDC W1A-C1A, CANARD CONTROL, BRAC611
 (CH074) \square AEDC W1A-C1A, CANARD CONTROL, BRAC611
 (CH075) \triangle AEDC W1A-C1A, CANARD CONTROL, BRAC611

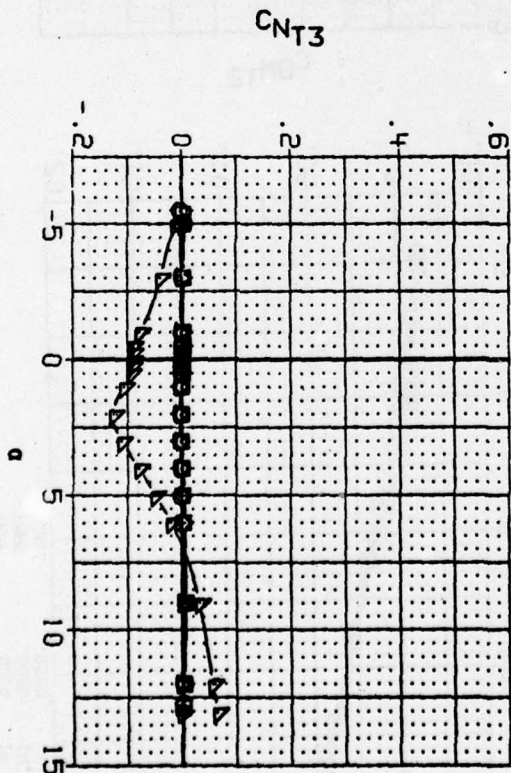
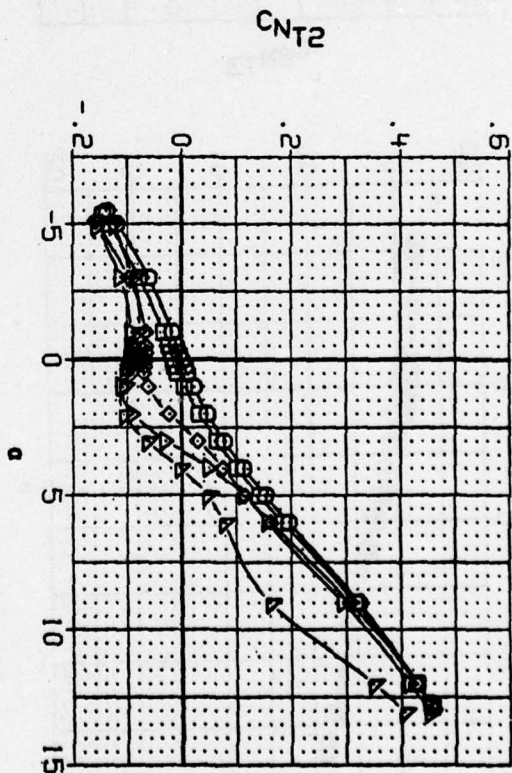
DCND1 DCND2 DCND3 DCND4
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 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 SCALE .0000

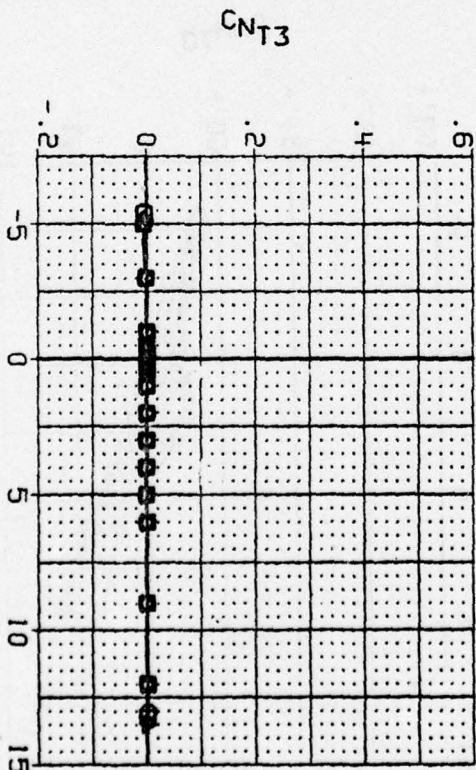
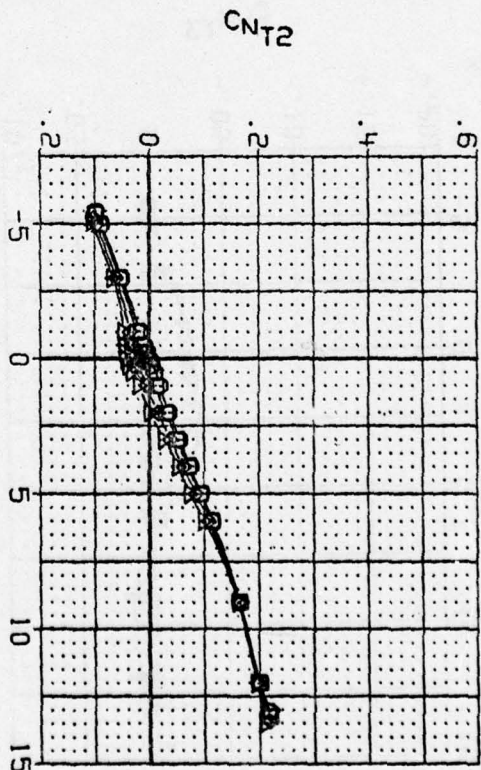


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $\Phi_{TAIL}=0$
 $\Phi_{CAN}=45$
 (B) MACH = 3.01

REFERENCE INFORMATION		
SRF	19.6350	50. IN.
LRF	5.0000	IN.
BRF	5.0000	IN.
YRFP	25.0000	IN.
ZRFP	.0000	IN.
SCALE	.0000	IN.



REFERENCE INFORMATION		
SREF	19.6350	SO. IN
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XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	IN.



DATA SET SYMBOL: CONFIGURATION DESCRIPTION

(CXH076) \square AEDC W1A-C1A, CANARD CONTROL, BRSCST1
 (CXH077) \square AEDC W1A-C1A, CANARD CONTROL, BRSCST1
 (CXH078) \square AEDC W1A-C1A, CANARD CONTROL, BRSCST1
 (CXH079) \square AEDC W1A-C1A, CANARD CONTROL, BRSCST1
 (CXH080) \square AEDC W1A-C1A, CANARD CONTROL, BRSCST1

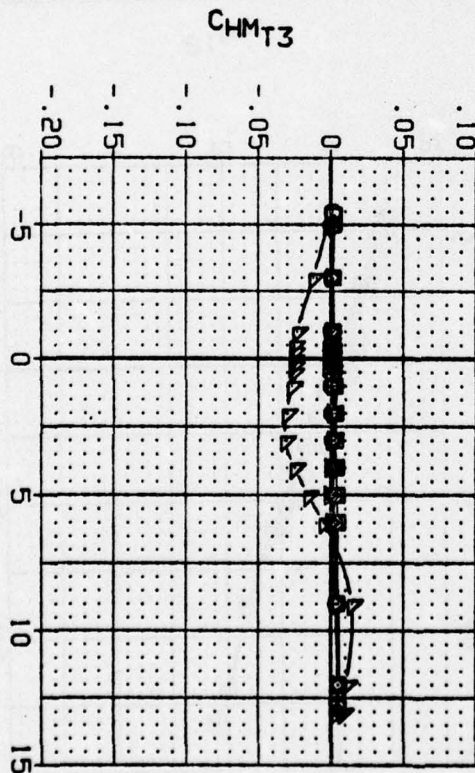
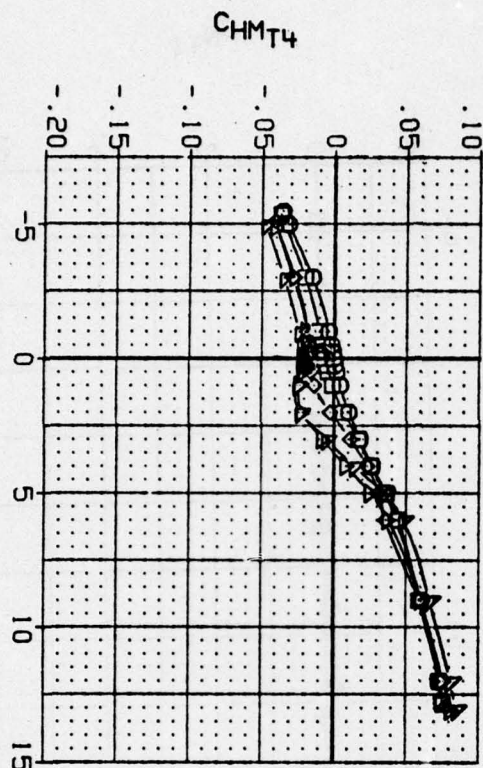
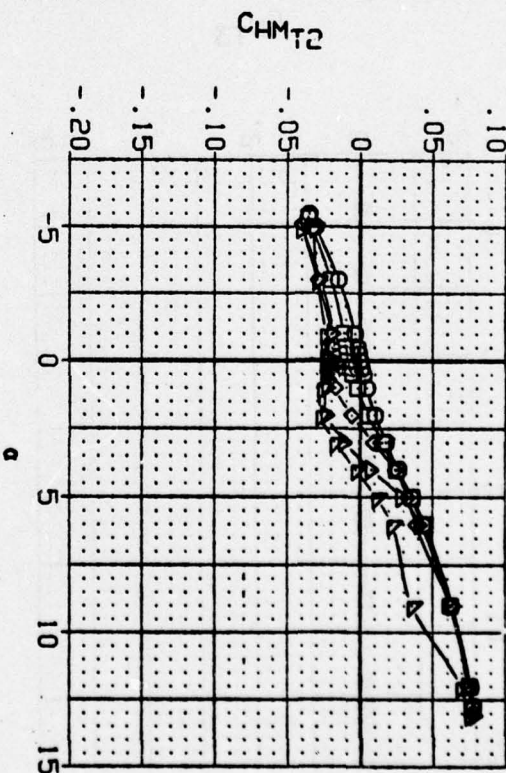
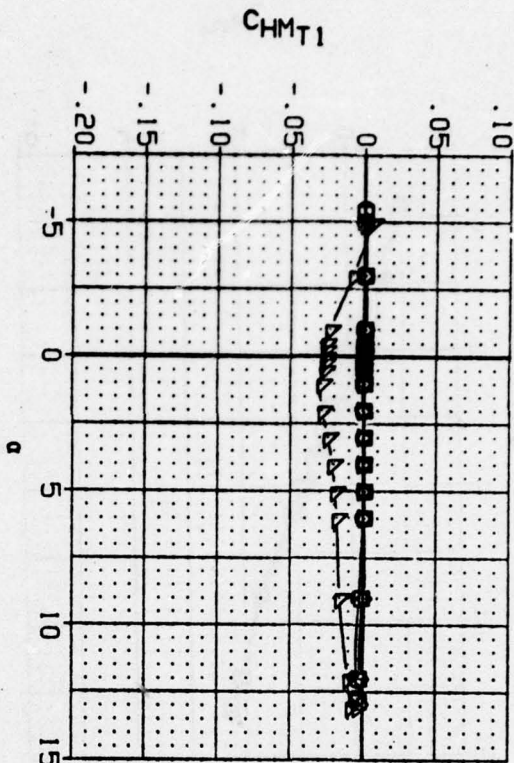
DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 .000 3.000 .000 .000
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 .000 15.000 .000 .000
 .000 15.000 .000 .000
 .000 15.000 .000 .000

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SCALE .0000

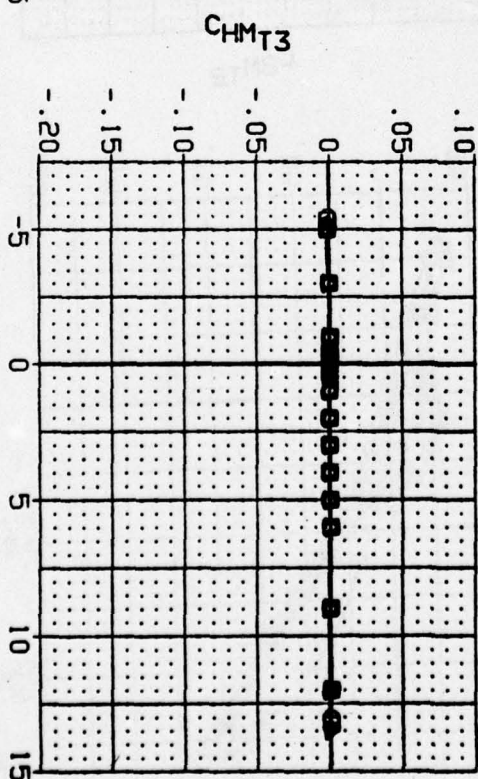
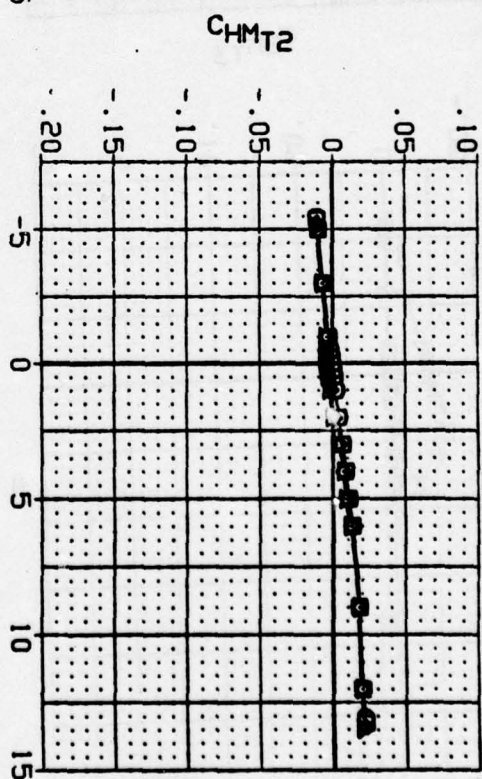


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

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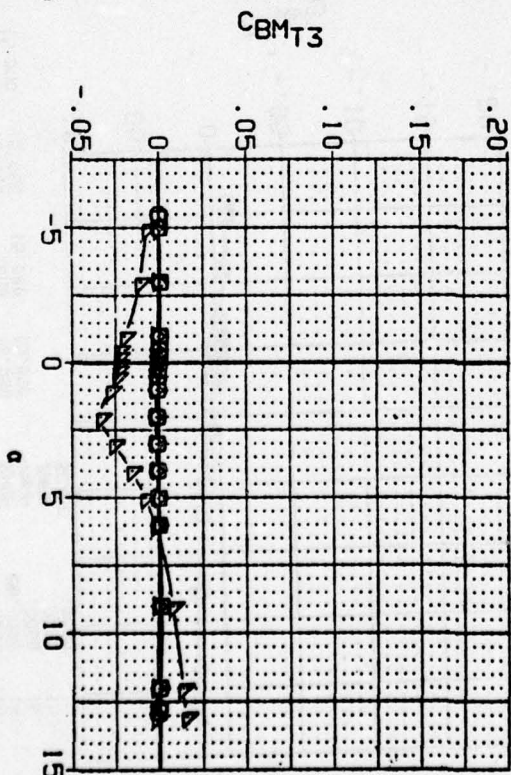
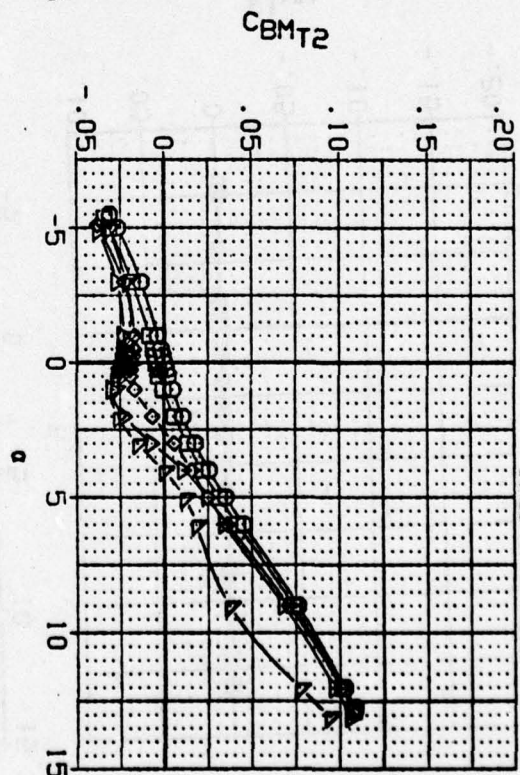
(A) MACH = 1.51

REFERENCE INFORMATION		
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LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	26.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0000	



PAGE 261

REFERENCE INFORMATION		
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LREF	5.0000	IN.
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SCALE	.0000	

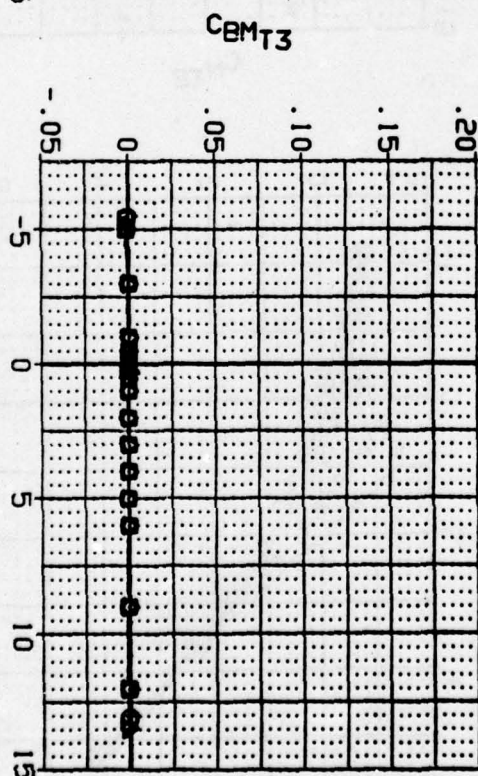
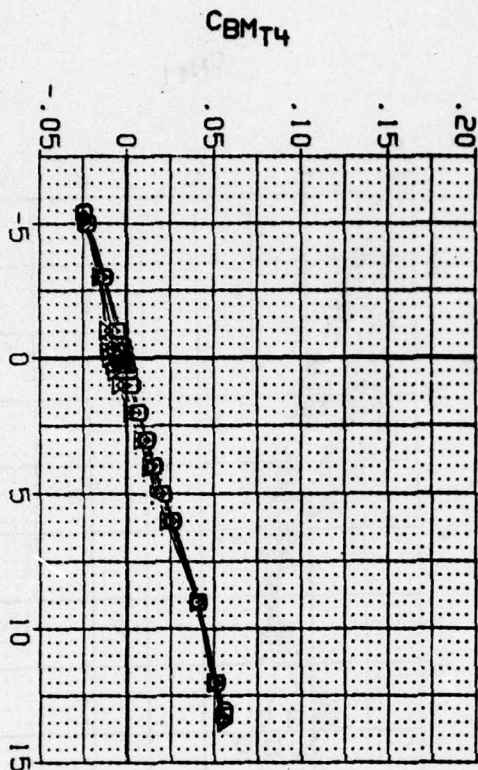
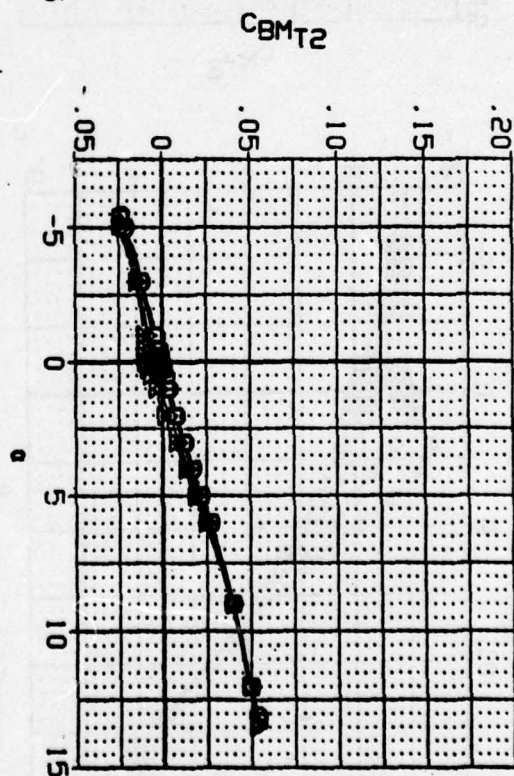
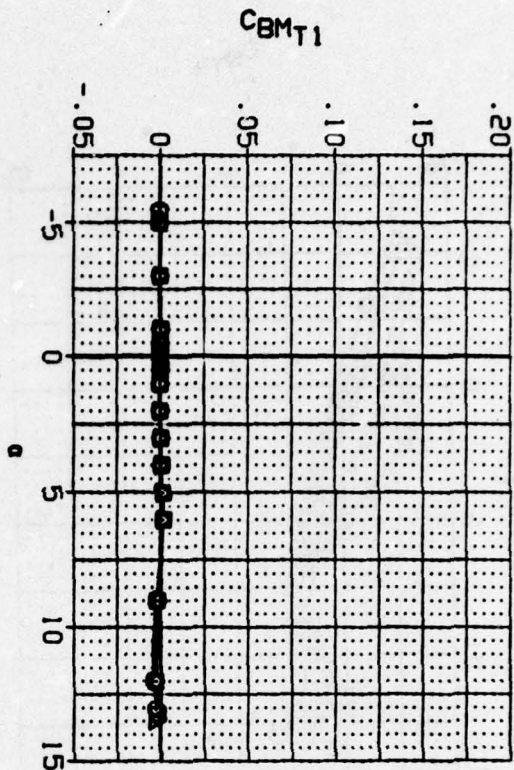


PAGE 262

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH076) ☐ AEDC V/A-CIA, CANARD CONTROL, BNSC6T1
 (CXH077) ☐ AEDC V/A-CIA, CANARD CONTROL, BNSC6T1
 (CXH078) ☐ AEDC V/A-CIA, CANARD CONTROL, BNSC6T1
 (CXH079) ☐ AEDC V/A-CIA, CANARD CONTROL, BNSC6T1
 (CXH080) ☒ DATA NOT AVAILABLE

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000
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REFERENCE INFORMATION
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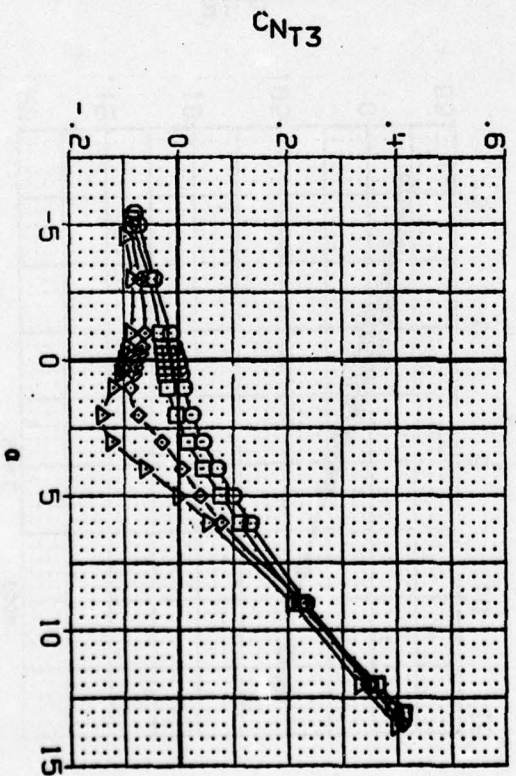
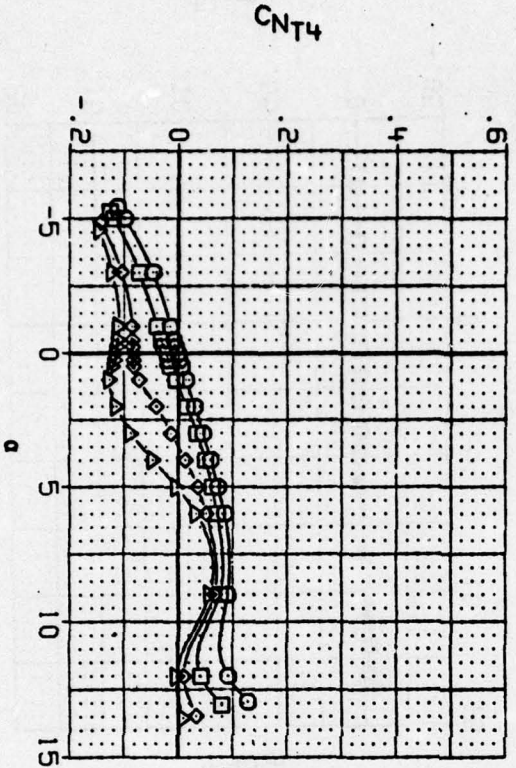
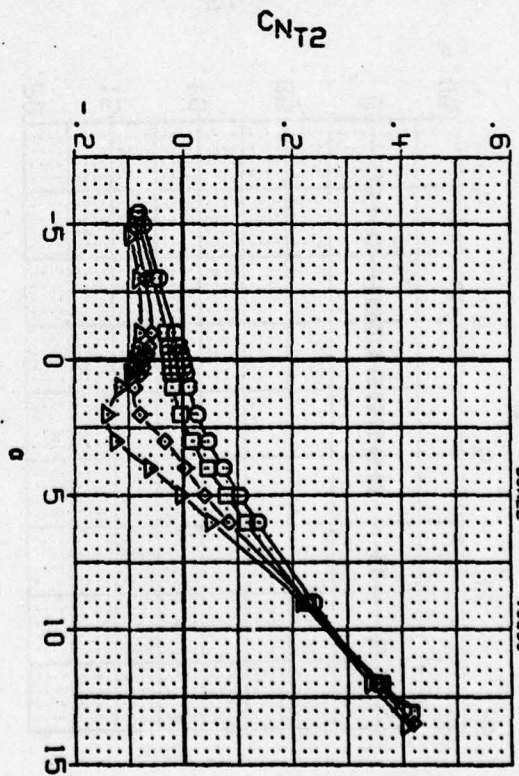
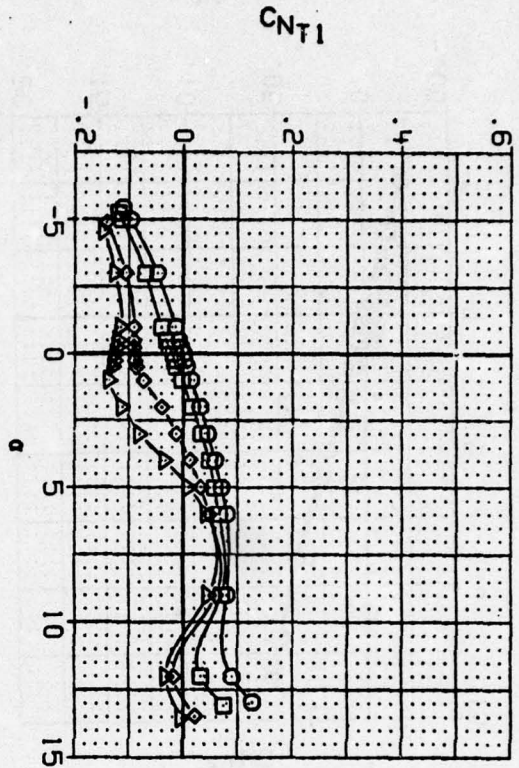


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (AXH082) \square AEDC W1A-C1A, CANARD CONTROL, BN5C6T1
 (AXH083) \square AEDC W1A-C1A, CANARD CONTROL, BN5C6T1
 (AXH084) \triangle AEDC W1A-C1A, CANARD CONTROL, BN5C6T1

DCND1 .000
 .000
 3.000
 9.000
 15.000
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REFERENCE INFORMATION
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 SCALE .0000

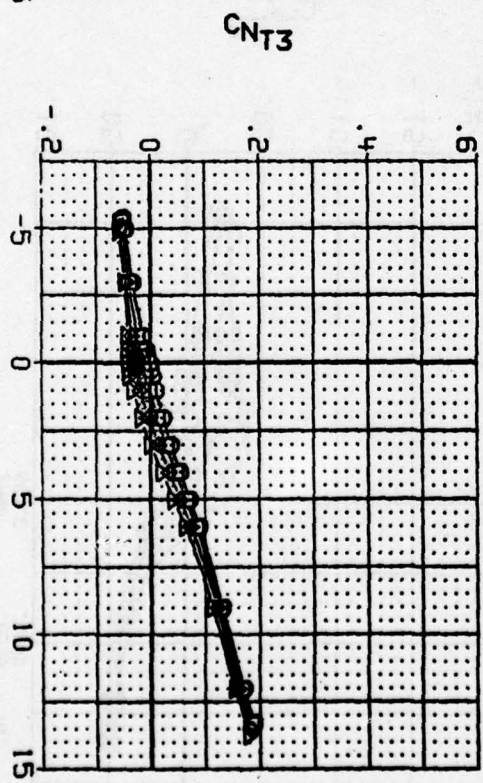
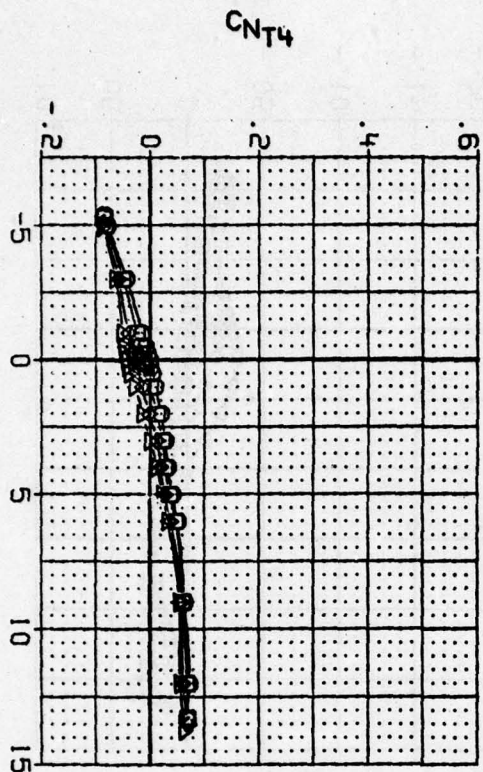
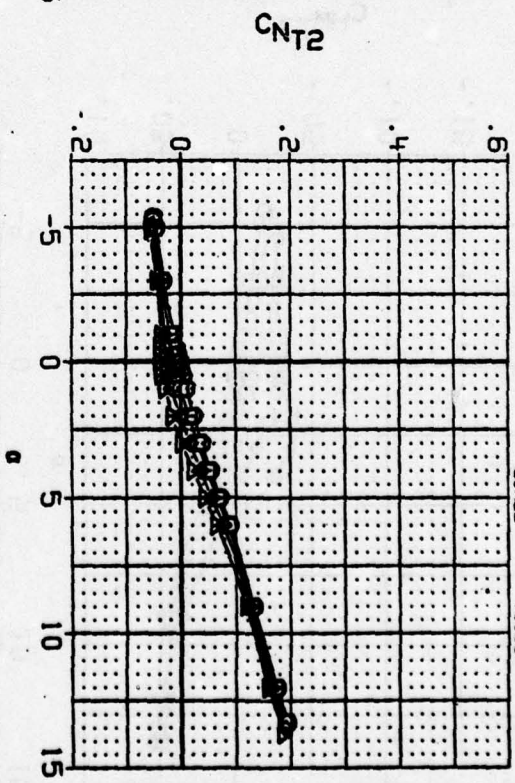
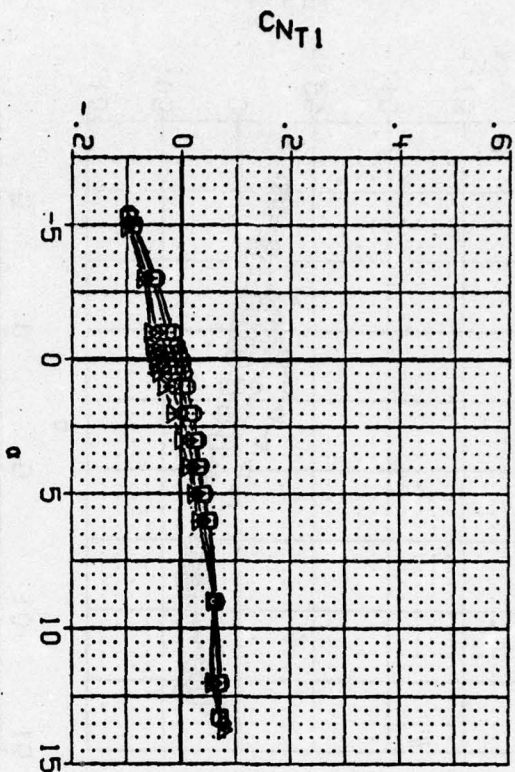


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH081) \square AEDC W-1A-C1A, CANARD CONTROL, BNC66T1
 (AXH082) \square AEDC W-1A-C1A, CANARD CONTROL, BNC66T1
 (AXH083) \square AEDC W-1A-C1A, CANARD CONTROL, BNC66T1
 (AXH084) \square AEDC W-1A-C1A, CANARD CONTROL, BNC66T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=45
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

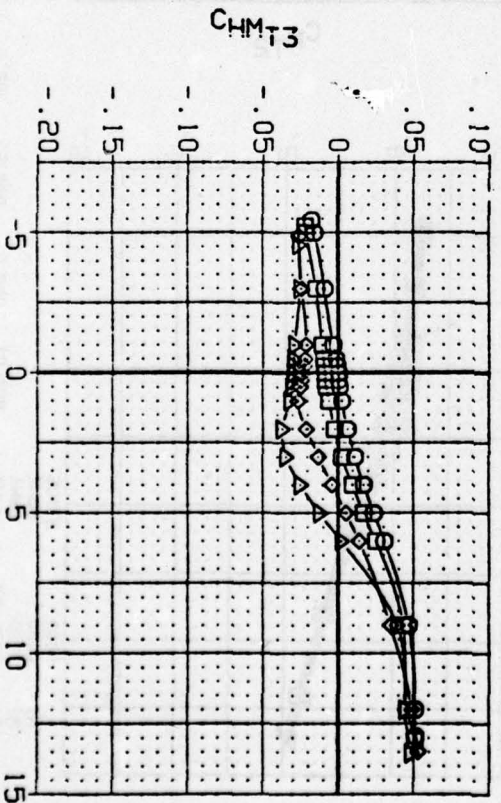
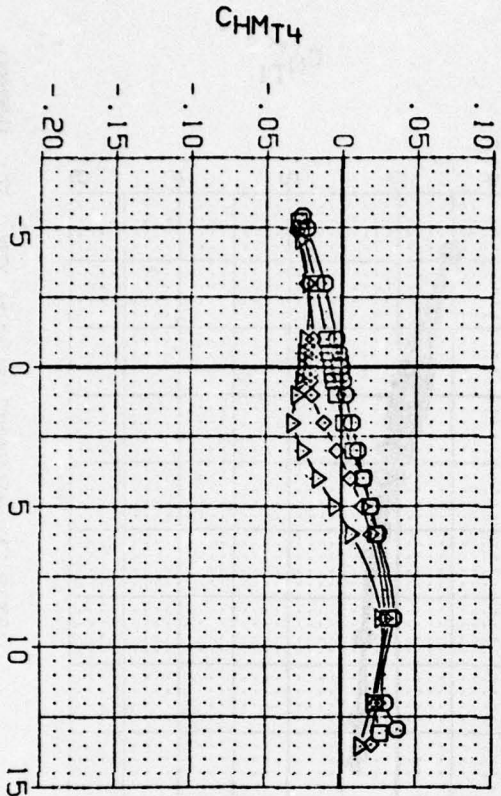
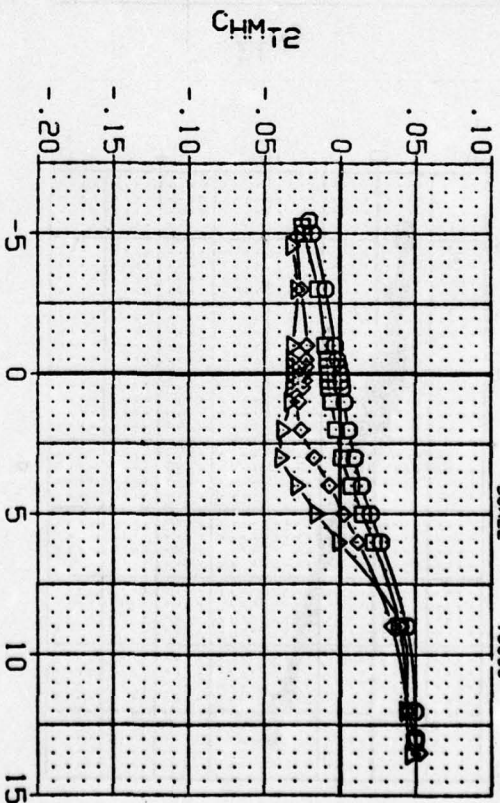
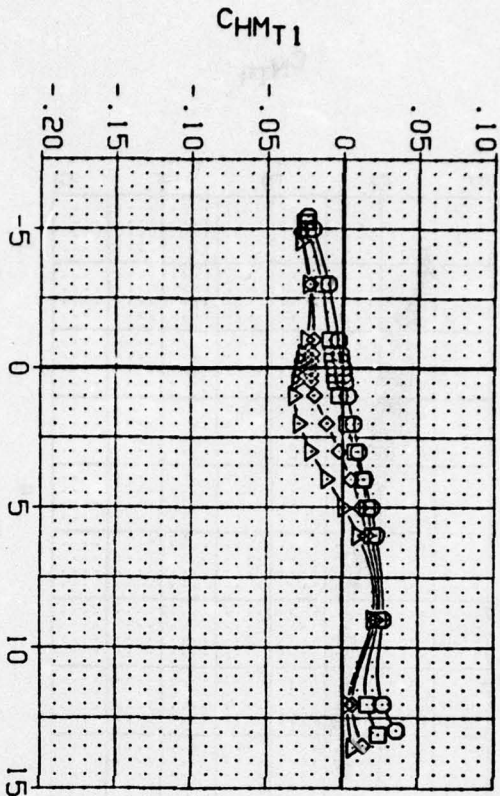
(CXH081) \square AEDC V41A-C1A, CANARD CONTROL, BN5C6T1
 (CXH082) \square AEDC V41A-C1A, CANARD CONTROL, BN5C6T1
 (CXH083) \triangle AEDC V41A-C1A, CANARD CONTROL, BN5C6T1
 (CXH084) \triangle AEDC V41A-C1A, CANARD CONTROL, BN5C6T1

DCND1 DCND2 DCND3 DCND4

3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

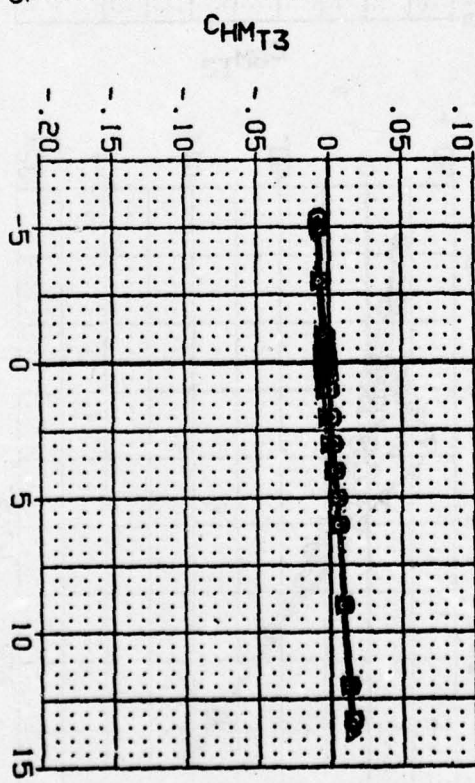
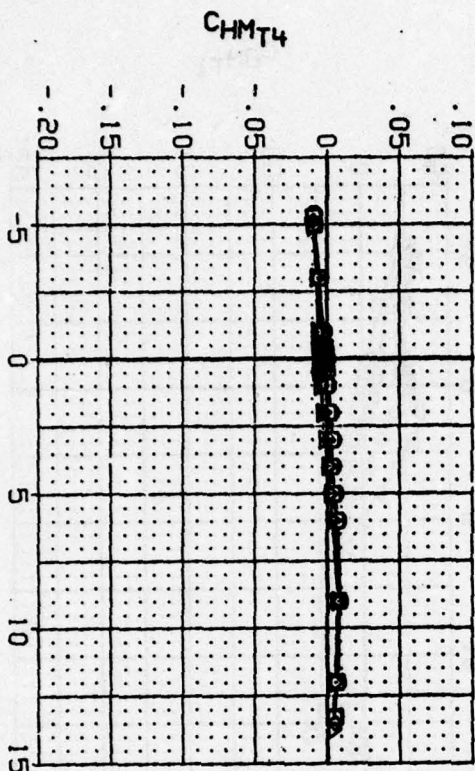
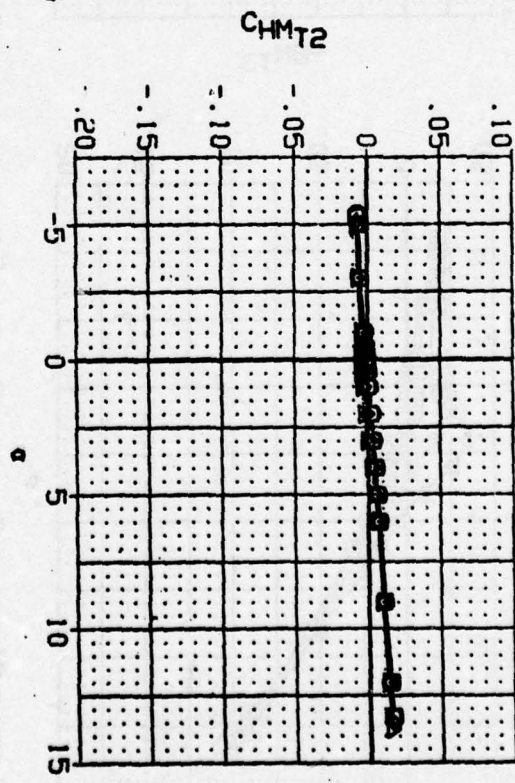
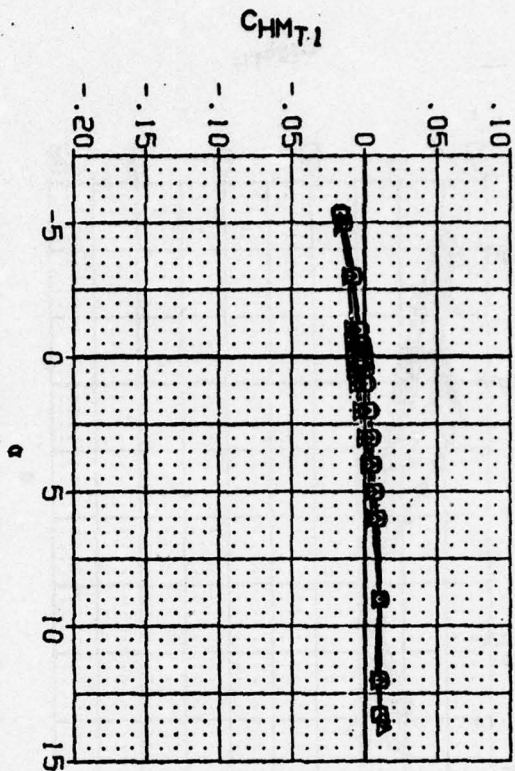
(CXH081) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (CXH082) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (CXH083) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1
 (CXH084) Δ AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=45

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CXH081) \square AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

(CXH082) \diamond AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

(CXH083) \triangle AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

(CXH084) \triangle AEDC W1A-C1A, CANARD CONTROL, BNSC6T1

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

3.000 3.000 3.000 3.000

9.000 9.000 9.000 9.000

15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

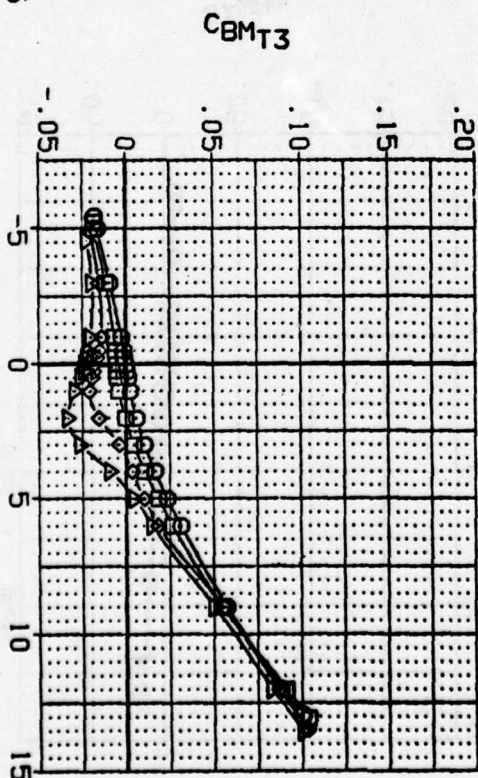
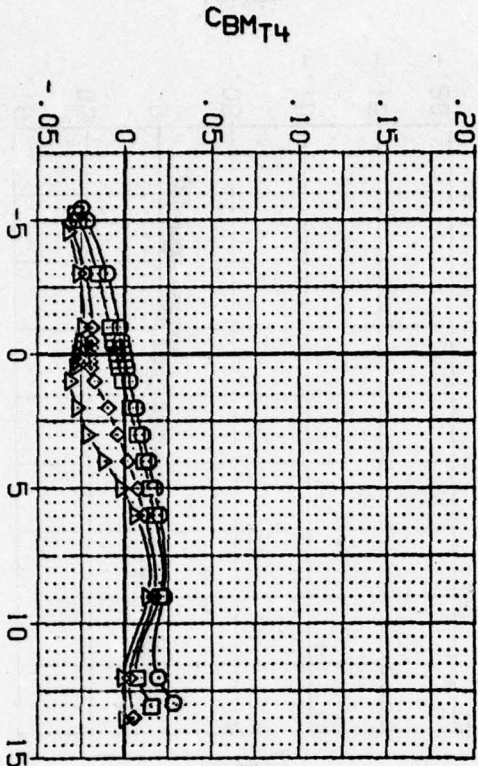
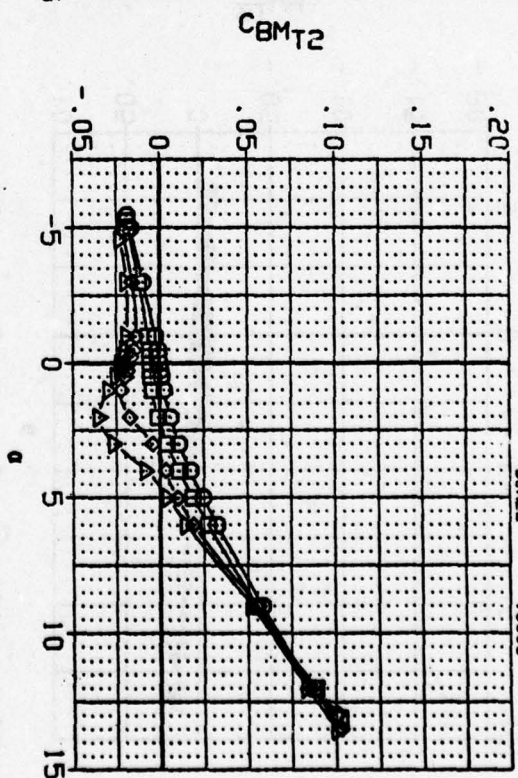
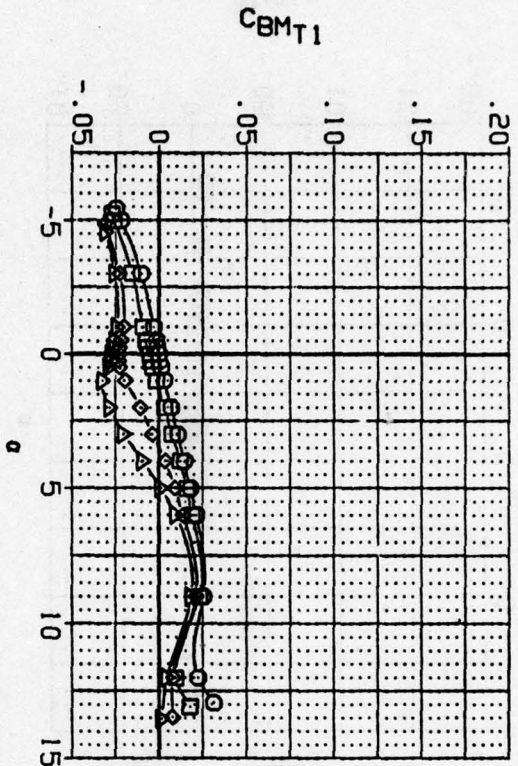
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ZREF 0.0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

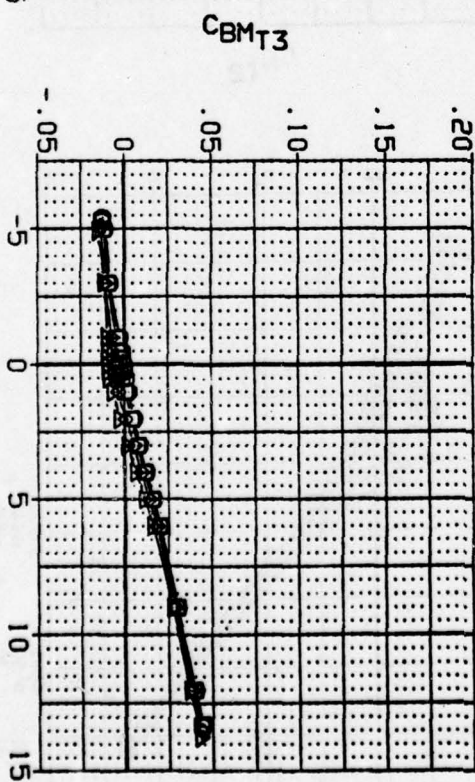
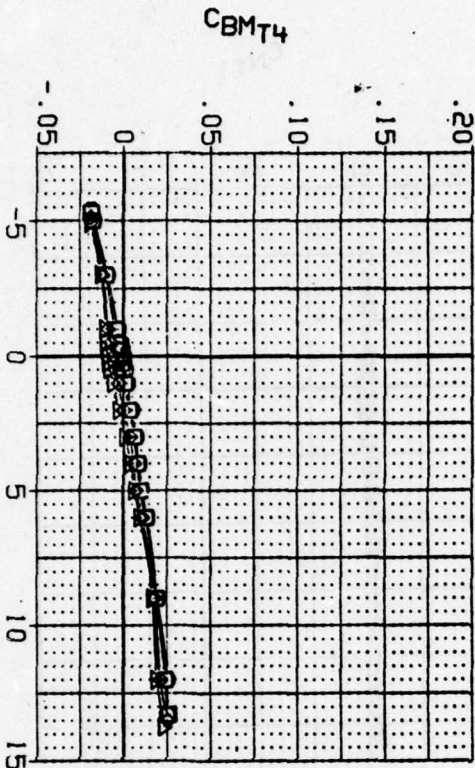
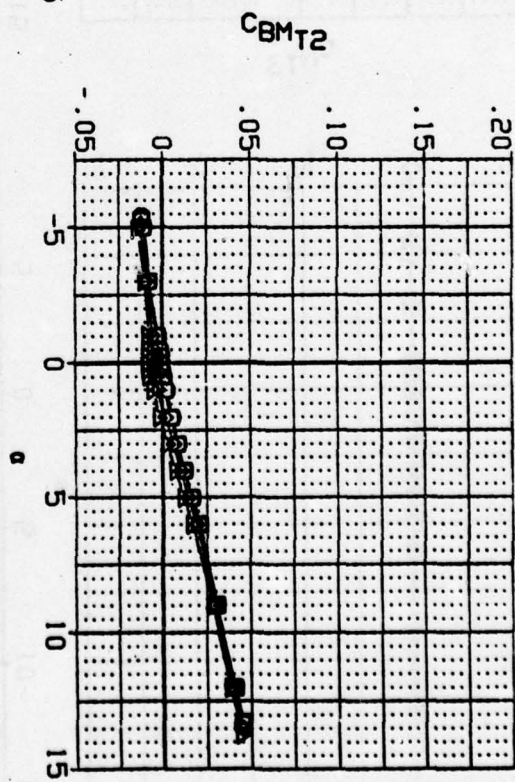
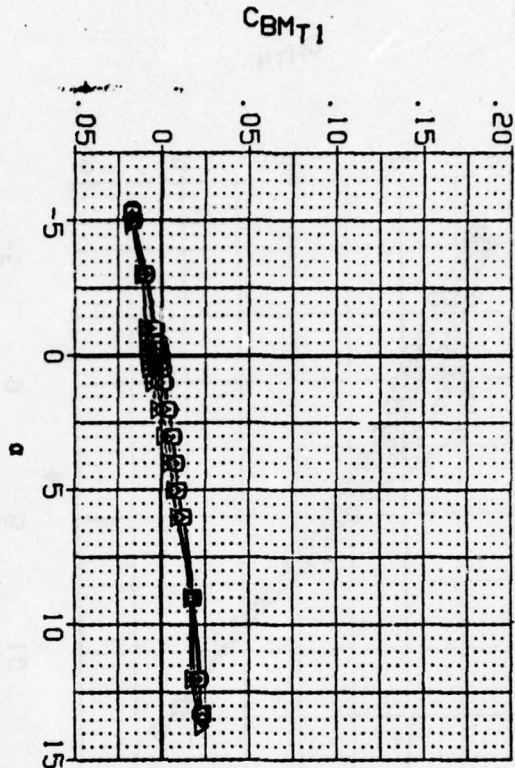
PHICAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CH081) \square AEDC W41A-C1A, CANARD CONTROL, BMS6GT1
 (CH082) \square AEDC W41A-C1A, CANARD CONTROL, BMS6GT1
 (CH083) \square AEDC W41A-C1A, CANARD CONTROL, BMS6GT1
 (CH084) \square AEDC W41A-C1A, CANARD CONTROL, BMS6GT1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
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 BREF 5.0000 IN.
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 SCALE .0000

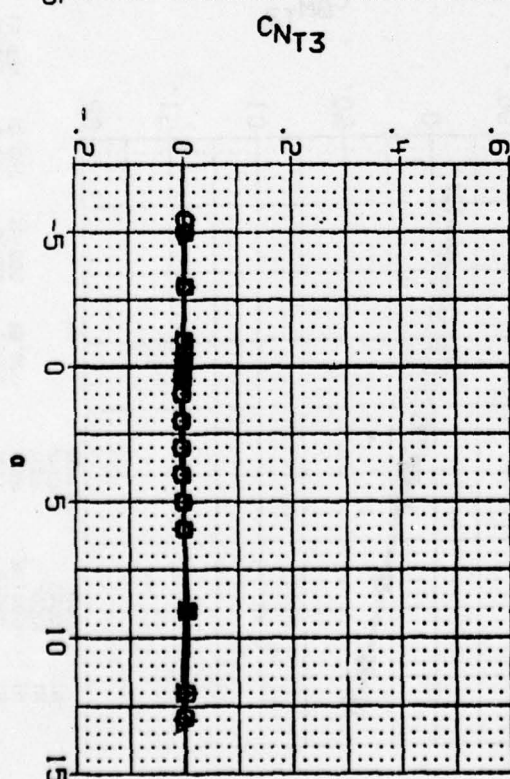
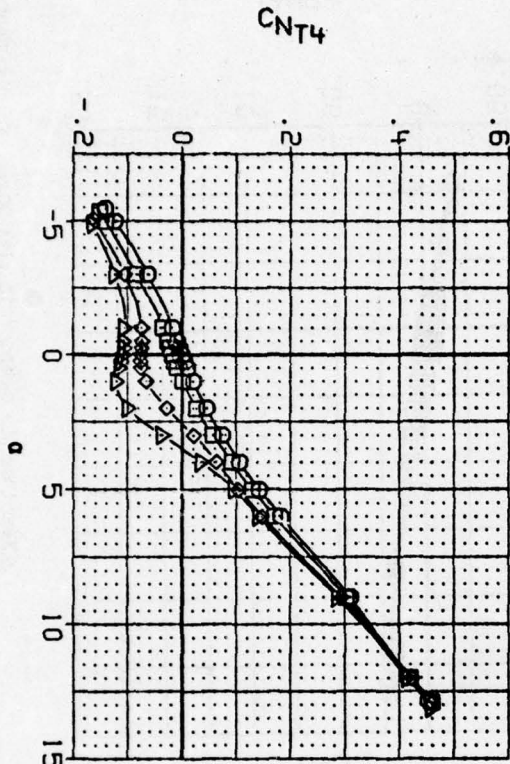
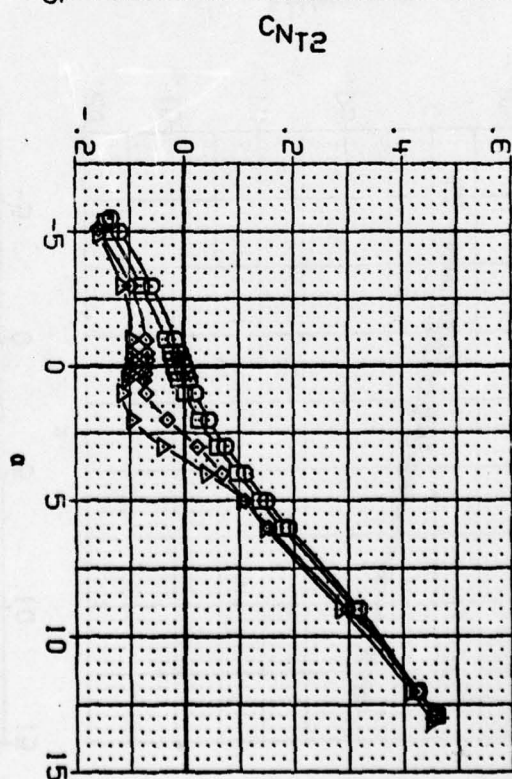
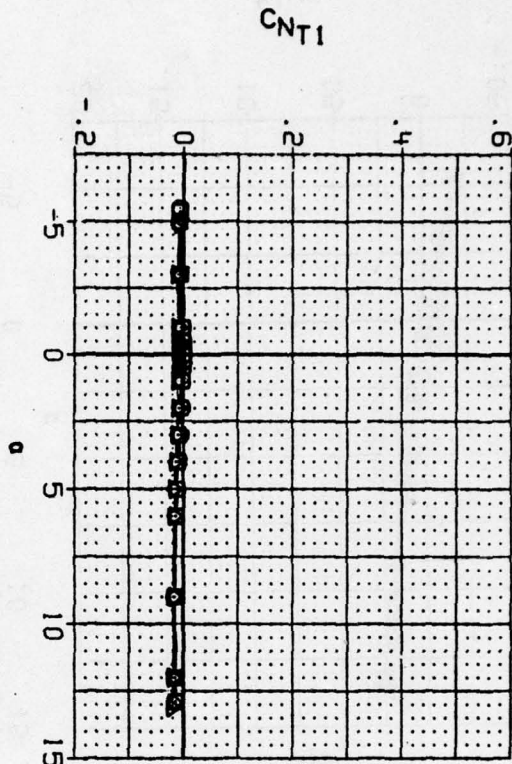


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=45
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH085) \square AEDC W1A-C1A, CANARD CONTROL, BN6C6T1
 (AXH086) \square AEDC W1A-C1A, CANARD CONTROL, BN6C6T1
 (AXH087) \square AEDC W1A-C1A, CANARD CONTROL, BN6C6T1
 (AXH088) \triangle AEDC W1A-C1A, CANARD CONTROL, BN6C6T1

DCND1 .000 DCND2 .000 DCND3 .000 DCND4 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
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 SCALE .0000

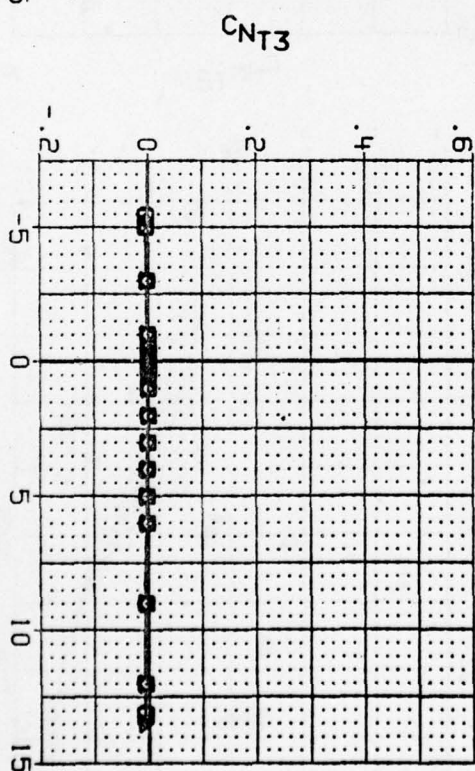
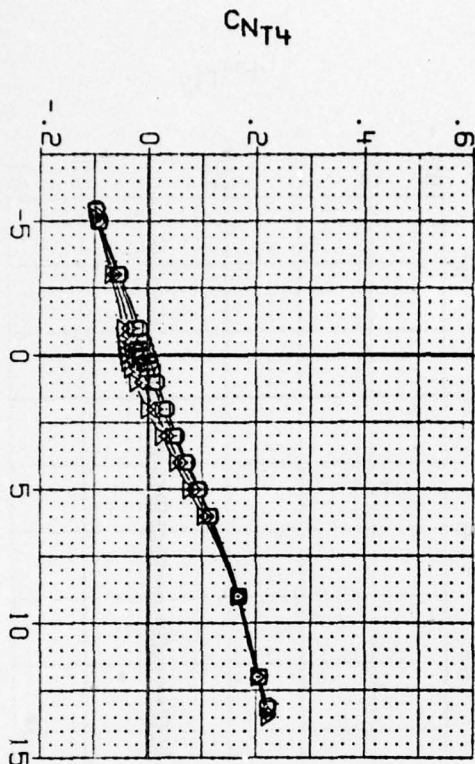
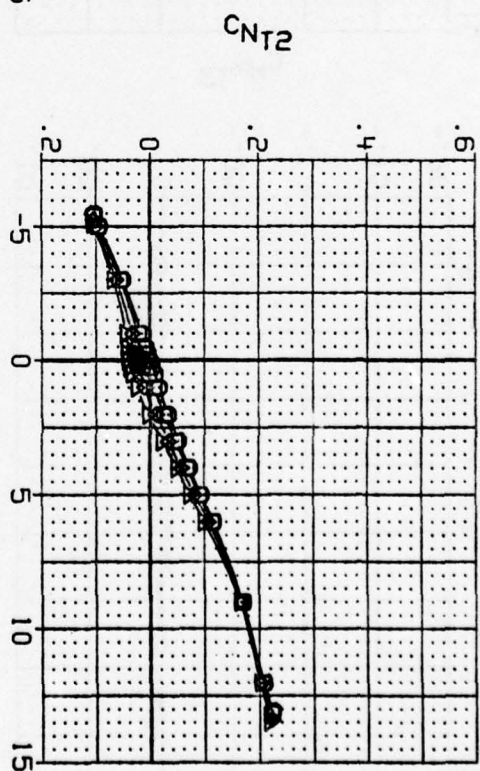
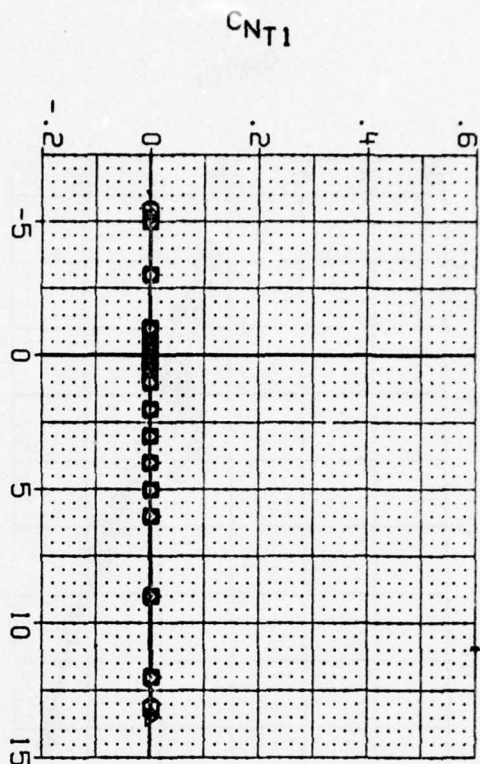


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAL=0 PHICND=0
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH085) □ AEDC W/IA-CIA, CANARD CONTROL, BNGC6T11
 (AXH086) □ AEDC W/IA-CIA, CANARD CONTROL, BNGC6T11
 (AXH087) △ AEDC W/IA-CIA, CANARD CONTROL, BNGC6T11
 (AXH088) △ AEDC W/IA-CIA, CANARD CONTROL, BNGC6T11

COND1 COND2 COND3 COND4
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHIAL=0 PHICND=0
 (B)MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

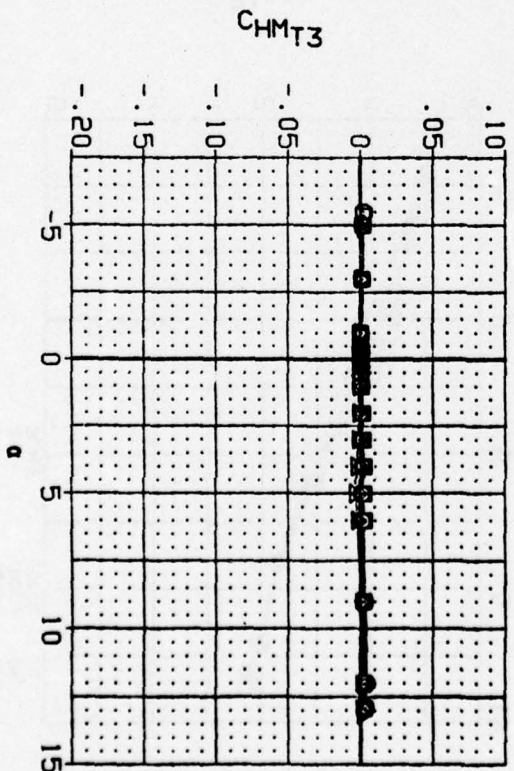
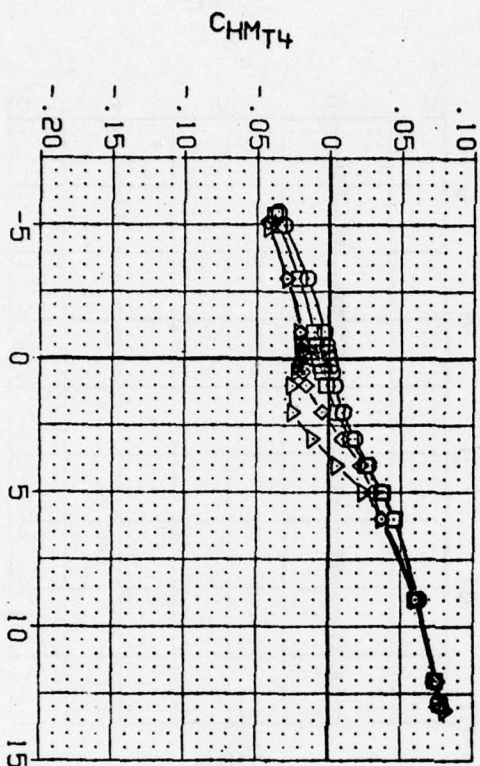
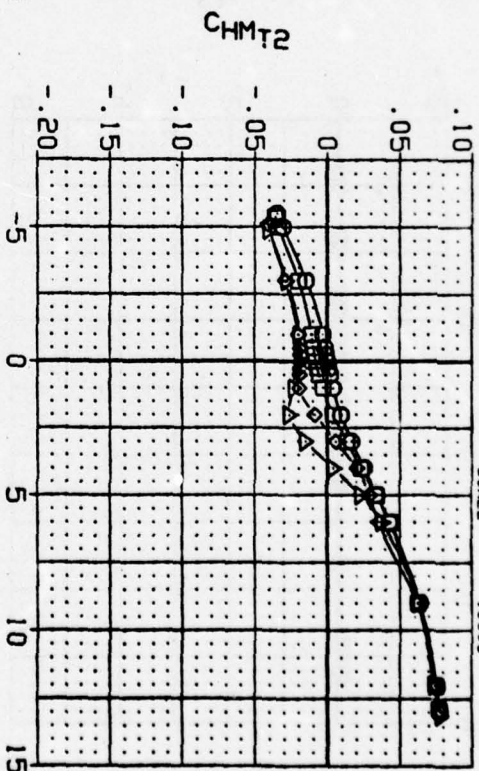
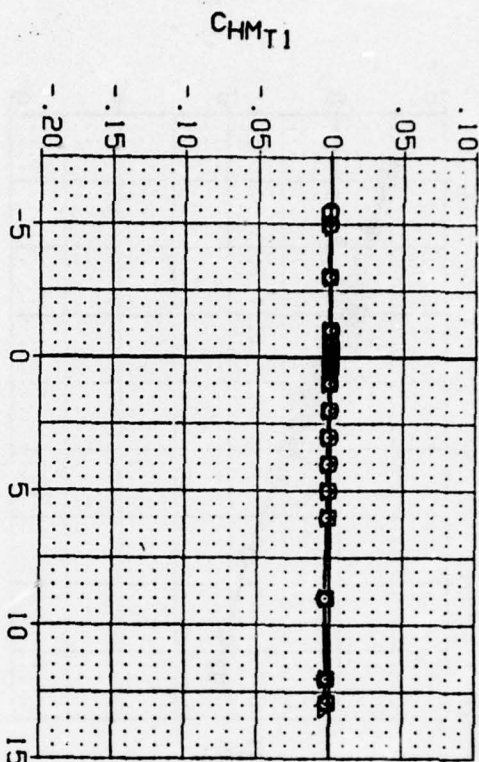
(CXH085)	□	AEDC W1A-C1A, CANARD CONTROL, BNGC6T1
(CXH086)	○	AEDC W1A-C1A, CANARD CONTROL, BNGC6T1
(CXH087)	△	AEDC W1A-C1A, CANARD CONTROL, BNGC6T1
(CXH088)	×	AEDC W1A-C1A, CANARD CONTROL, BNGC6T1

COND1 DCOND2 DCOND3 DCOND4

.000	.000	.000	.000
.000	3.000	.000	.000
.000	9.000	.000	3.000
.000	15.000	.000	9.000
		.000	15.000

REFERENCE INFORMATION

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YREF	26.0007	IN.
ZREF	.0007	IN.
SCALE	.00	IN.

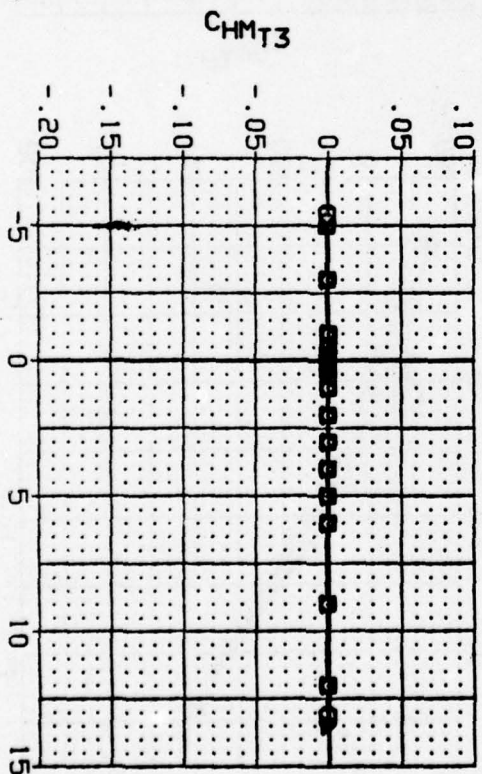
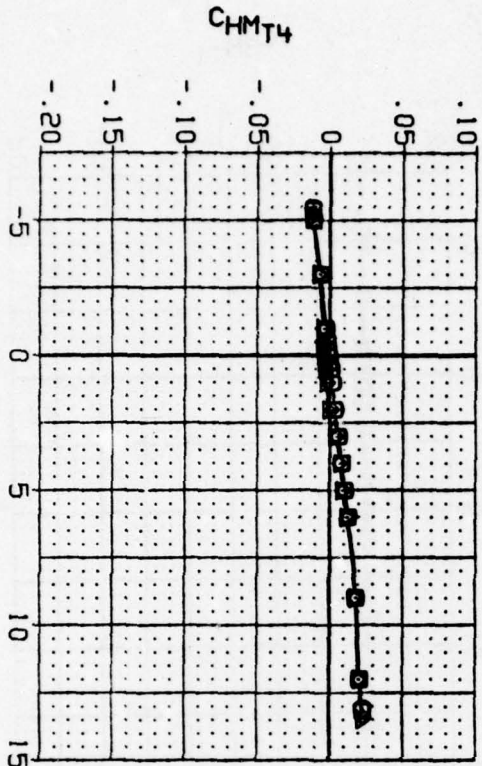
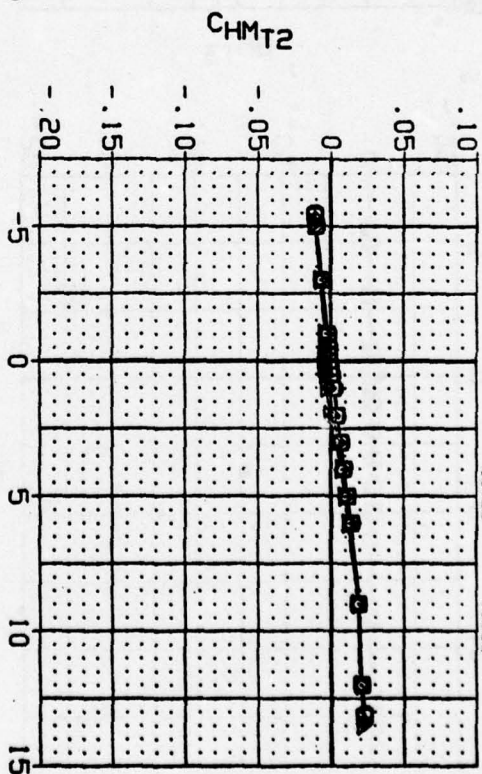
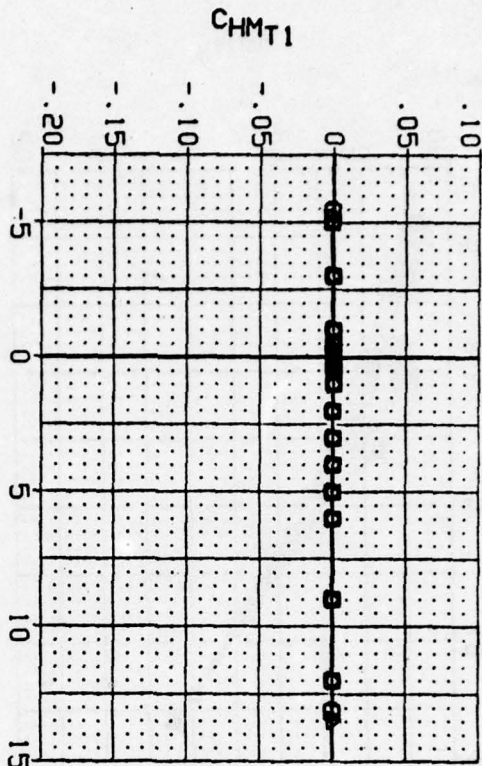


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAL=0$ $PHICND=0$
 (A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH085) \square AEDC V41A-C1A, CANARD CONTROL, BNC6ST1
 (CXH086) \square AEDC V41A-C1A, CANARD CONTROL, BNC6ST1
 (CXH087) \square AEDC V41A-C1A, CANARD CONTROL, BNC6ST1
 (CXH088) Δ AEDC V41A-C1A, CANARD CONTROL, BNC6ST1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 .000 3.000 .000 3.000
 .000 9.000 .000 9.000
 .000 15.000 .000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAL=0$ $PHICND=0$
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

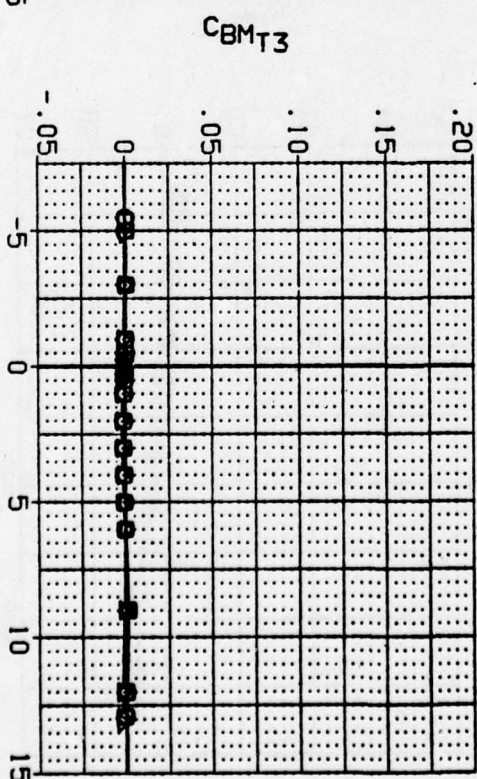
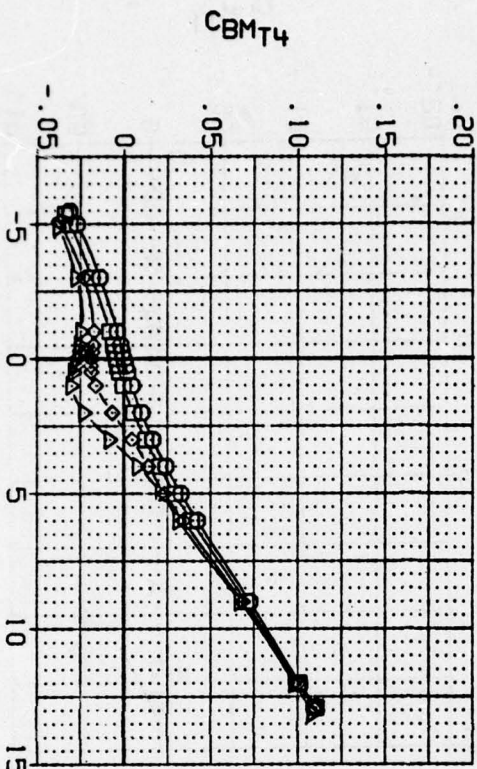
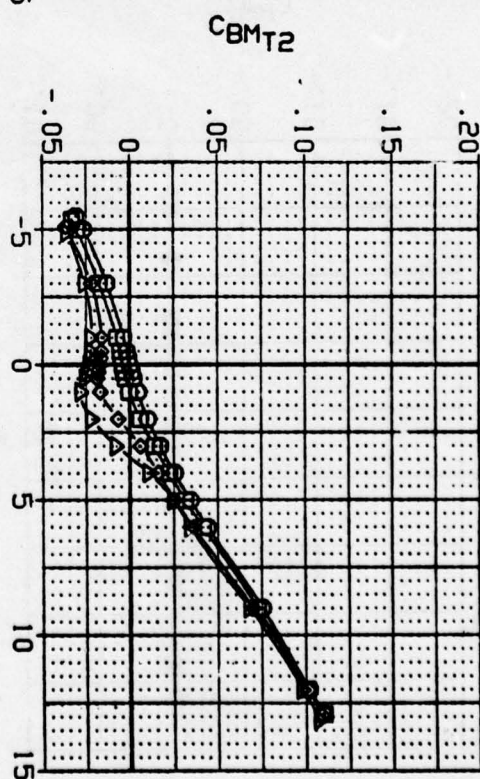
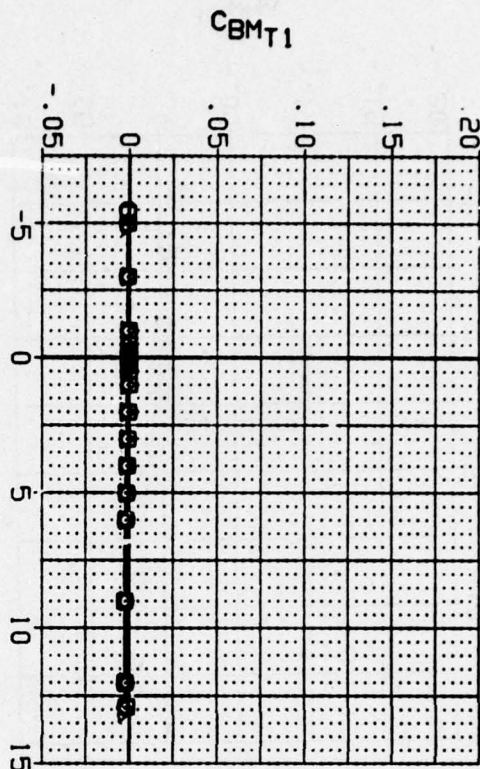
(CXH085)	○	AEDC W1A-C1A, CANARD CONTROL, BNC65T1
(CXH086)	○	AEDC W1A-C1A, CANARD CONTROL, BNC65T1
(CXH087)	△	AEDC W1A-C1A, CANARD CONTROL, BNC65T1
(CXH088)	△	AEDC W1A-C1A, CANARD CONTROL, BNC65T1

DCND1 DCND2 DCND3 DCND4

.000	.000	.000	.000
.000	3.000	.000	3.000
.000	9.000	.000	9.000
.000	15.000	.000	15.000

REFERENCE INFORMATION

SREF	19.6350	50. IN.
LREF	5.0000	IN.
BREF	5.0000	IN.
XREF	20.0000	IN.
YREF	.0000	IN.
ZREF	.0000	IN.
SCALE	.0030	



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION

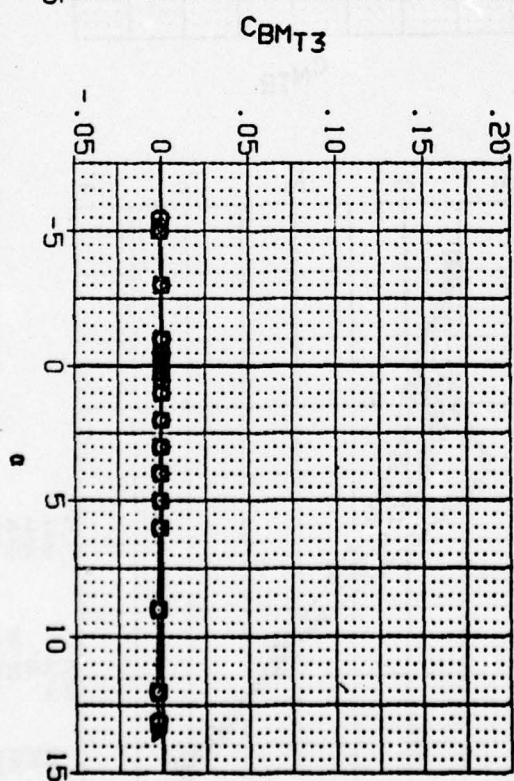
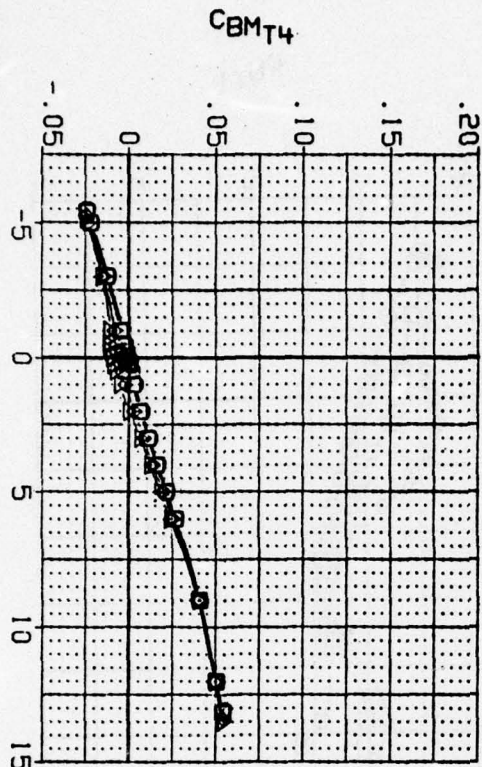
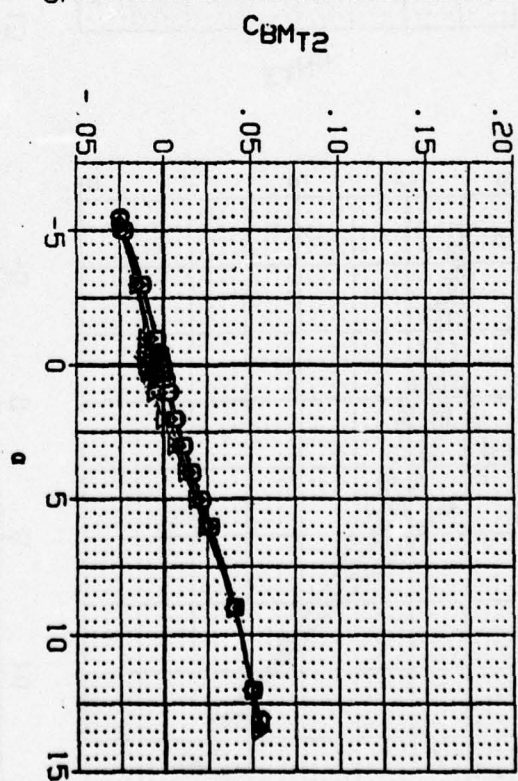
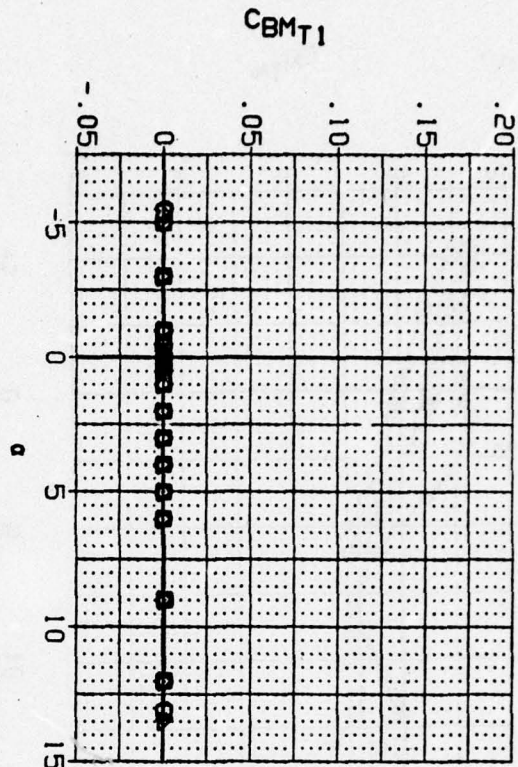
(CXH085) \square AEDC WJIA-CIA, CANARD CONTROL, BMS6GT1
 (CXH086) \square AEDC WJIA-CIA, CANARD CONTROL, BMS6GT1
 (CXH087) \square AEDC WJIA-CIA, CANARD CONTROL, BMS6GT1
 (CXH088) \triangle AEDC WJIA-CIA, CANARD CONTROL, BMS6GT1

COND1 COND2 COND3 COND4

.000 .000 .000 .000
 .000 3.000 .000 .000
 .000 9.000 .000 .000
 .000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YMRP 26.0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

$PHITL=0$ $PHICND=0$

(B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION

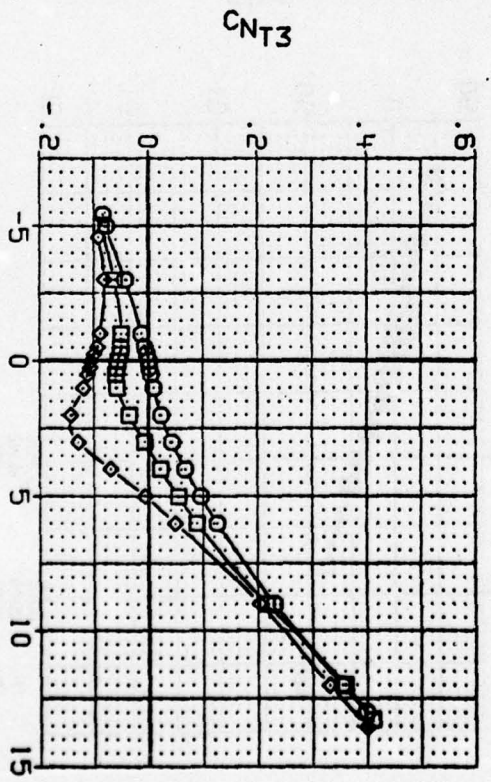
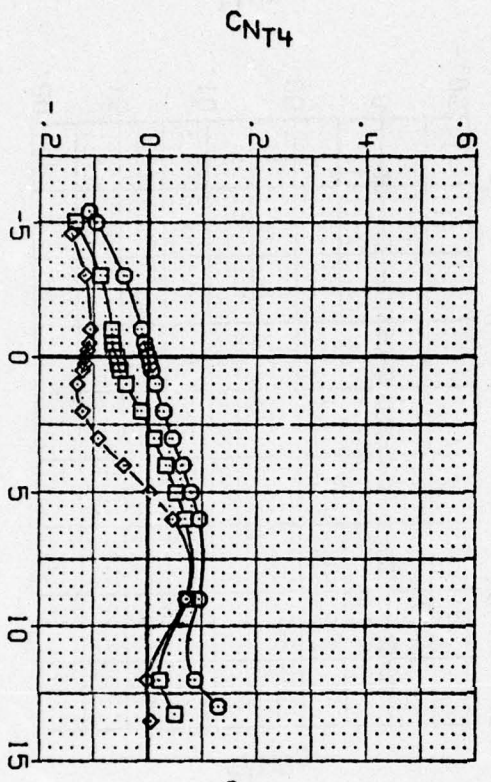
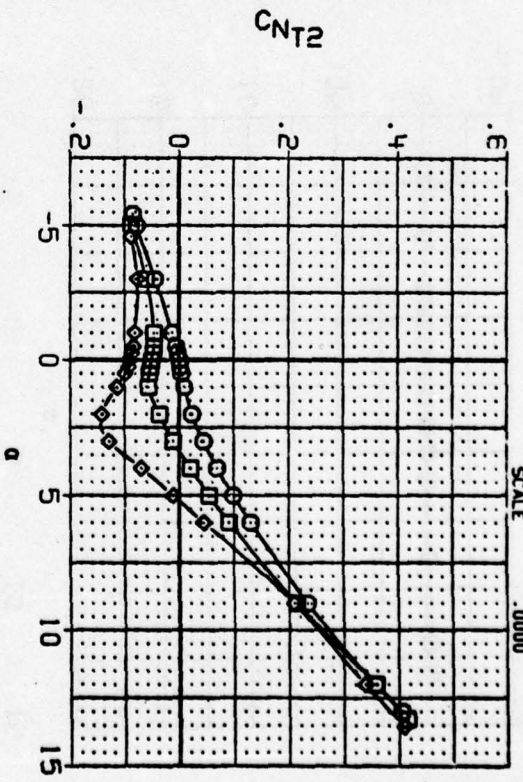
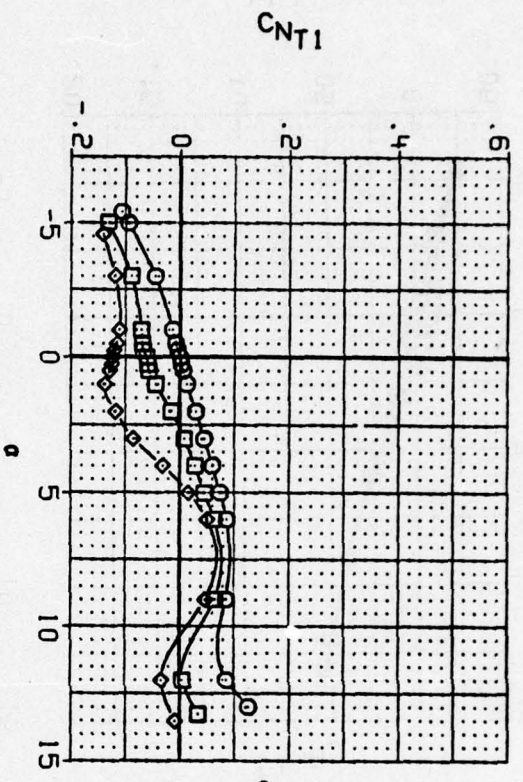
(AXH089) \square AEDC W1A-C1A, CANARD CONTROL, BNEC6T1
 (AXH090) \square AEDC W1A-C1A, CANARD CONTROL, BNEC6T1
 (AXH091) \diamond AEDC W1A-C1A, CANARD CONTROL, BNEC6T1

DCND1 DCND2 DCND3 DCND4

6.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.0000 IN.
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 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

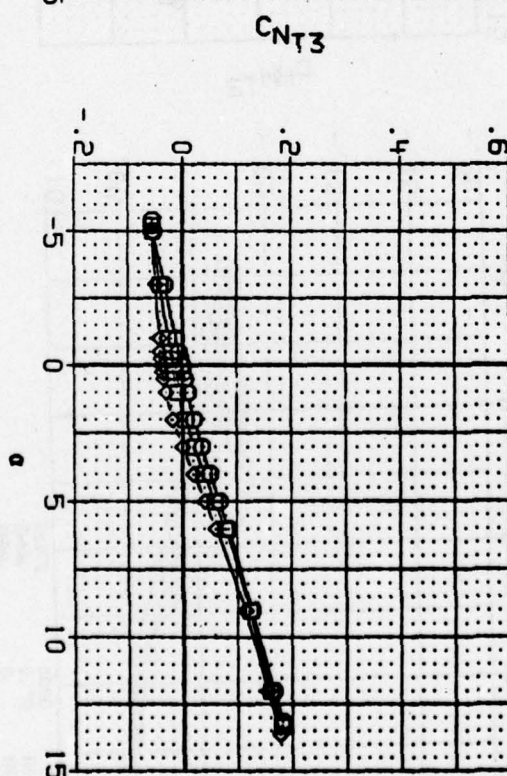
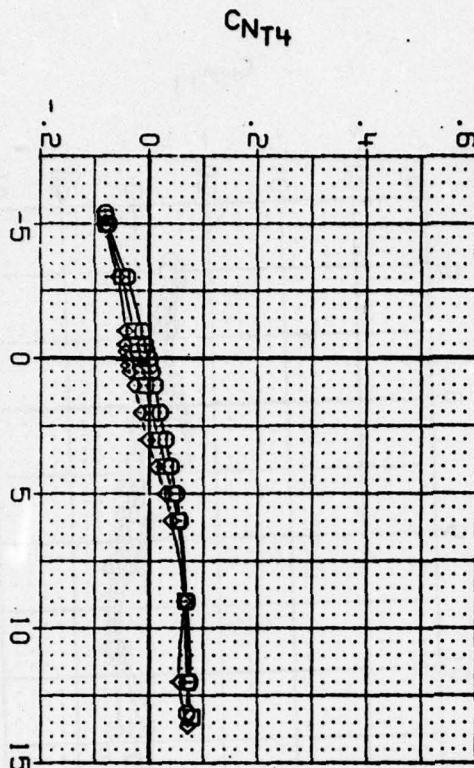
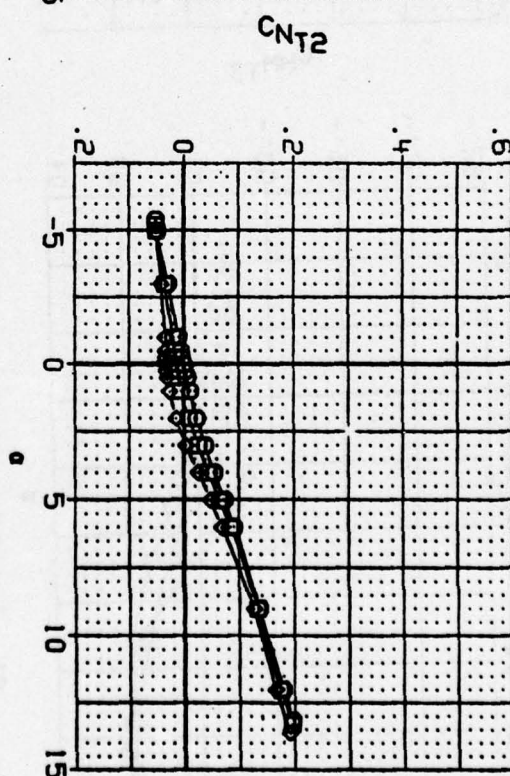
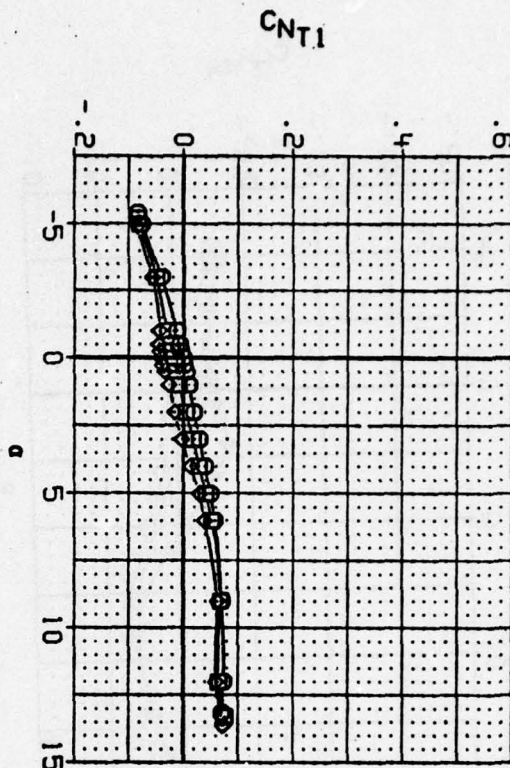
PHITAL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (AXH089) ☐ AEDC W1A-C1A, CANARD CONTROL, BNGC611
 (AXH090) ☐ AEDC W1A-C1A, CANARD CONTROL, BNGC611
 (AXH091) ☐ AEDC W1A-C1A, CANARD CONTROL, BNGC611

DCND1 .000
 .000
 6.000
 15.000
 DCND2 .000
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 15.000
 DCND3 .000
 .000
 6.000
 15.000
 DCND4 .000
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 6.000
 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000

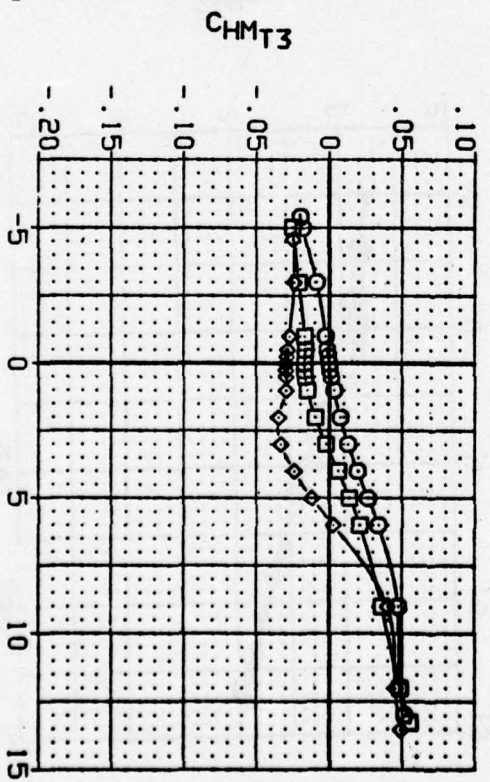
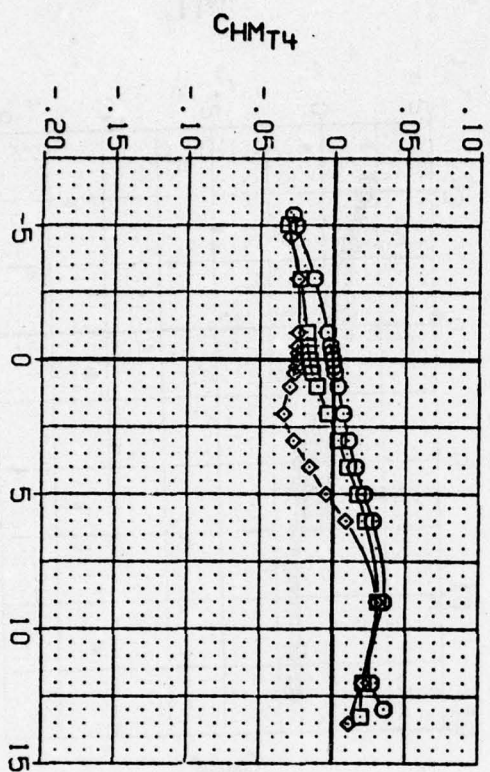
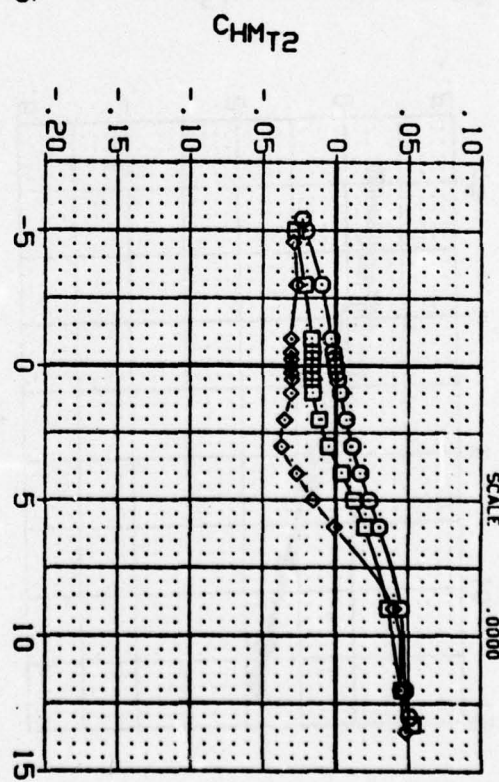
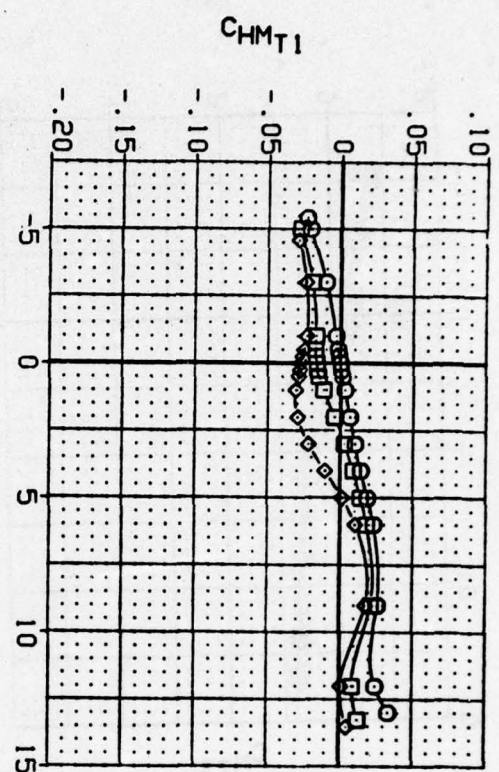


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITL=0$
 $PHICND=45$
 (B) MACH = 3.01

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (C9H089) \square AEDC W1A-C1A, CANARD CONTROL, BNEC6T1
 (C9H090) \square AEDC W1A-C1A, CANARD CONTROL, BNEC6T1
 (C9H091) \diamond AEDC W1A-C1A, CANARD CONTROL, BNEC6T1

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 5.000 6.000 6.000 6.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 25.00 IN.
 YREF 0.00 IN.
 ZREF 0.00 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITL=0 PHICND=45
 (A) MACH = 1.51

DATA SET SYMBOL

(CXH099)
(CXH09J)
(CXH09I)

CONFIGURATION DESCRIPTION

AEDC W1A-C1A, CANARD CONTROL, BNSCGT1
AEDC W1A-C1A, CANARD CONTROL, BNSCGT1
AEDC W1A-C1A, CANARD CONTROL, BNSCGT1

DCND1

.000
6.000
15.000

DCND2

.000
6.000
15.000

DCND3

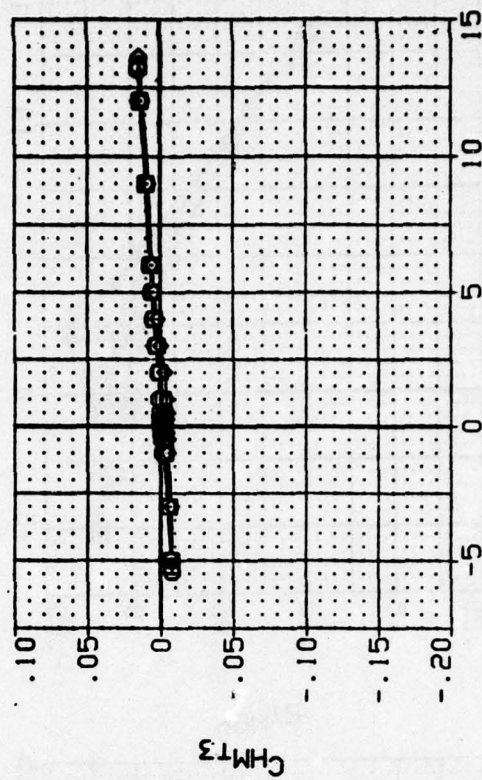
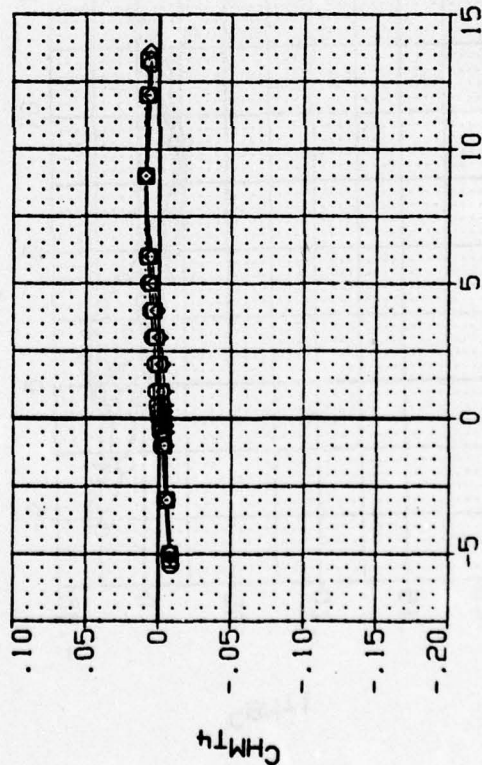
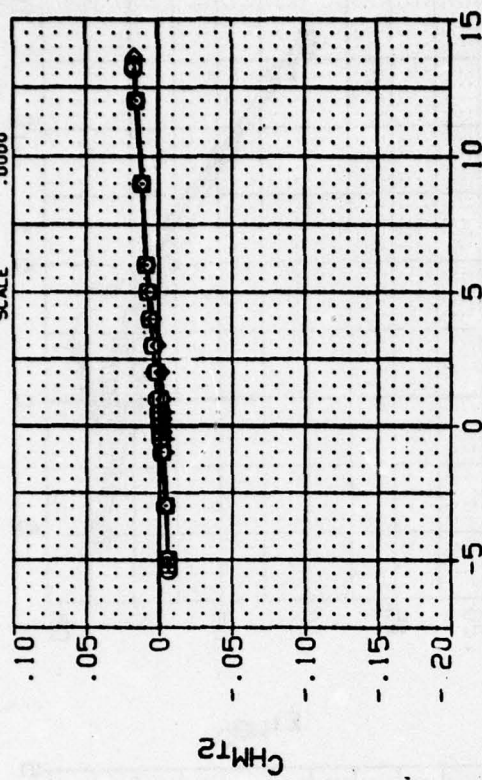
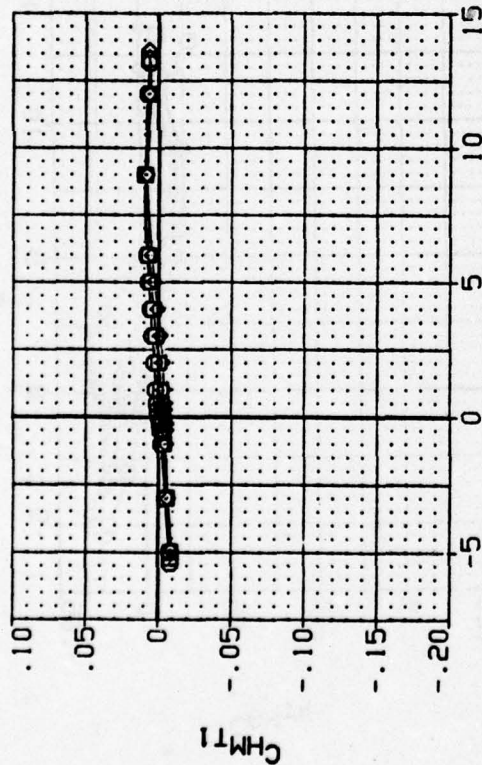
.000
6.000
15.000

DCND4

.000
6.000
15.000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAIL=0 PHICND=45

(B) MACH = 3.01

DATA SET SYMBOL

(CXH089)
(CXH090)
(CXH091)

CONFIGURATION DESCRIPTION

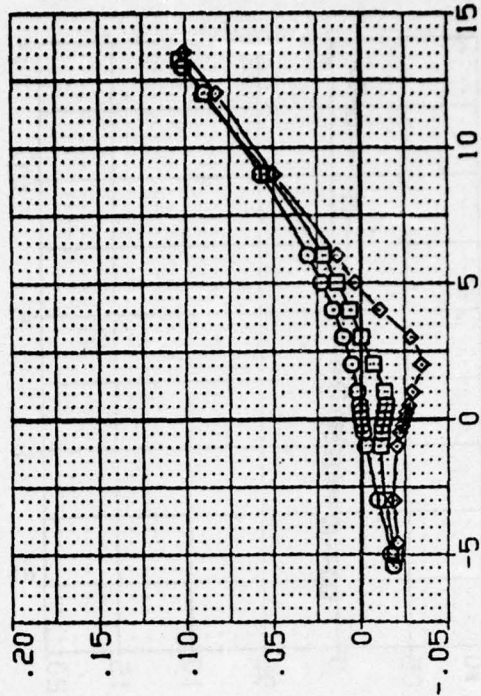
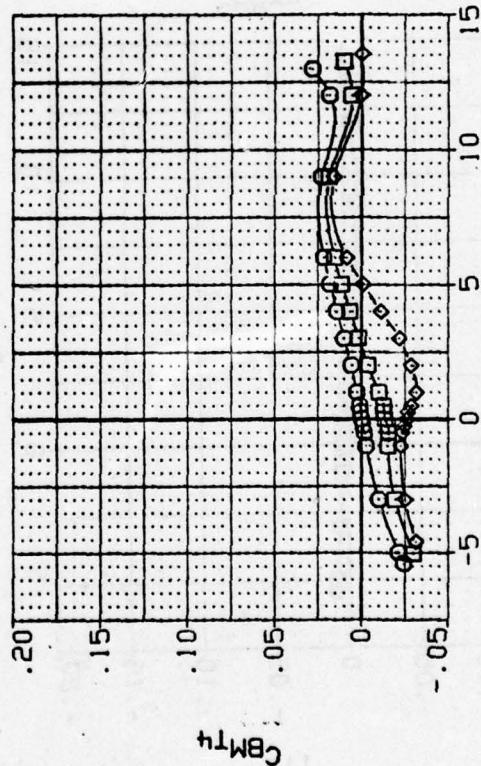
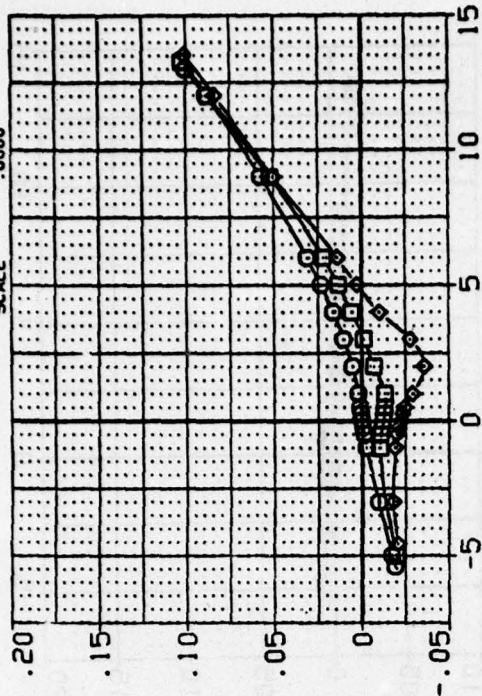
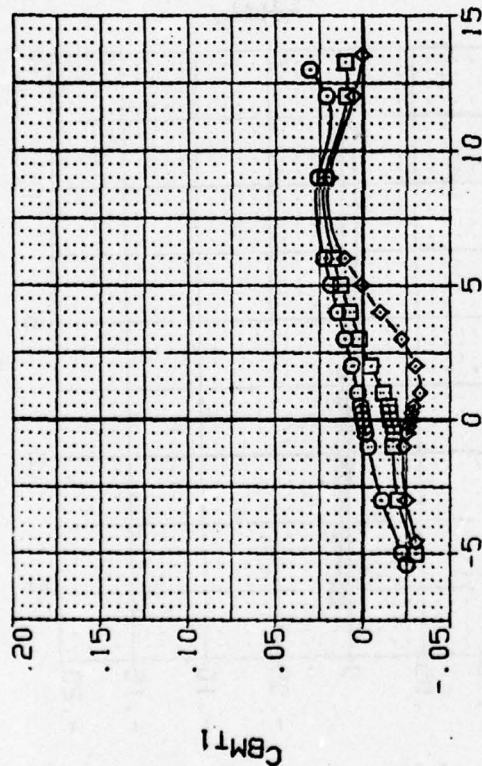
AEDC W1A-C1A, CANARD CONTROL, BNCST1
AEDC W1A-C1A, CANARD CONTROL, BNCST1
AEDC W1A-C1A, CANARD CONTROL, BNCST1

CONFIDENCE

CONFIDENCE
CONFIDENCE
CONFIDENCE

REFERENCE INFORMATION

SREF 19.6350 50. IN.
LREF 5.0000 10. IN.
BREF 5.0000 10. IN.
XREF 26.0000 10. IN.
YREF .000 10. IN.
ZREF .000 10. IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAIL=0 PHICND=45

(A) MACH = 1.51

DATA SET SYMBOL

(CXH089)
(CXH090)
(CXH091)

CONFIGURATION DESCRIPTION

AEDC W-1A-CIA, CANARD CONTROL, BNEC6T1
AEDC W-1A-CIA, CANARD CONTROL, BNEC6T1
AEDC W-1A-CIA, CANARD CONTROL, BNEC6T1

DCND1

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15.000

DCND2

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.000
15.000

DCND3

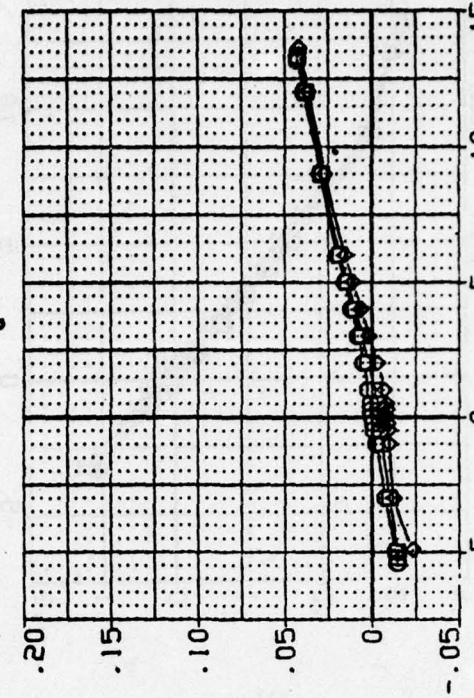
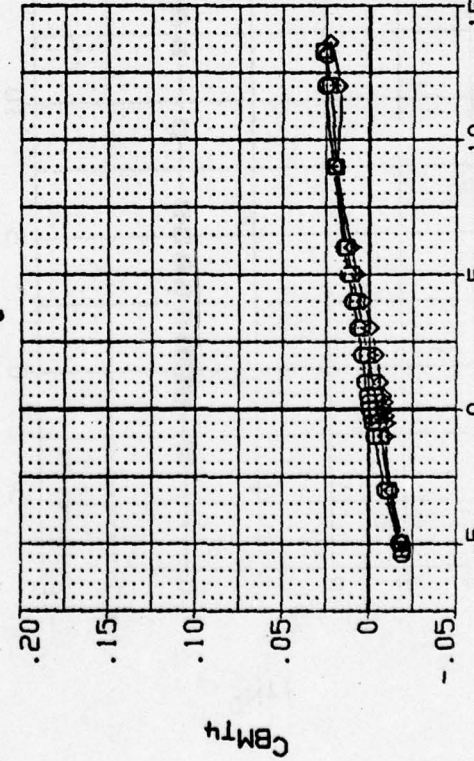
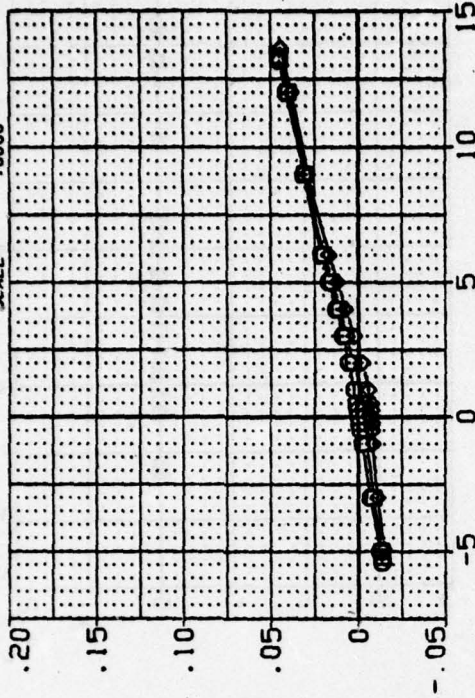
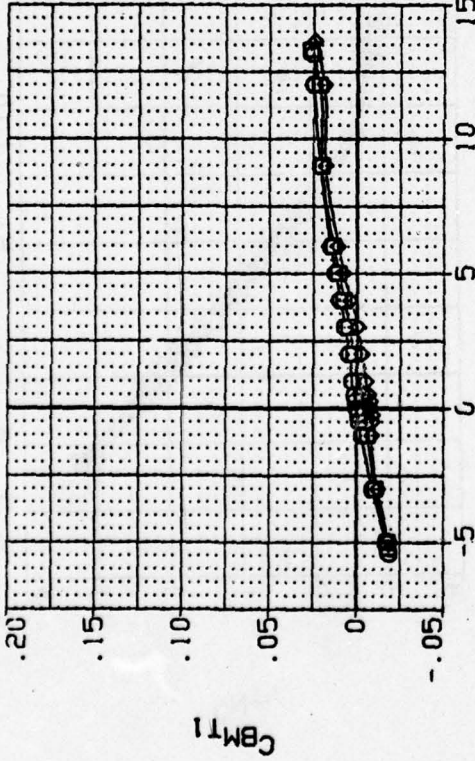
.000
.000
15.000

DCND4

.000
.000
15.000

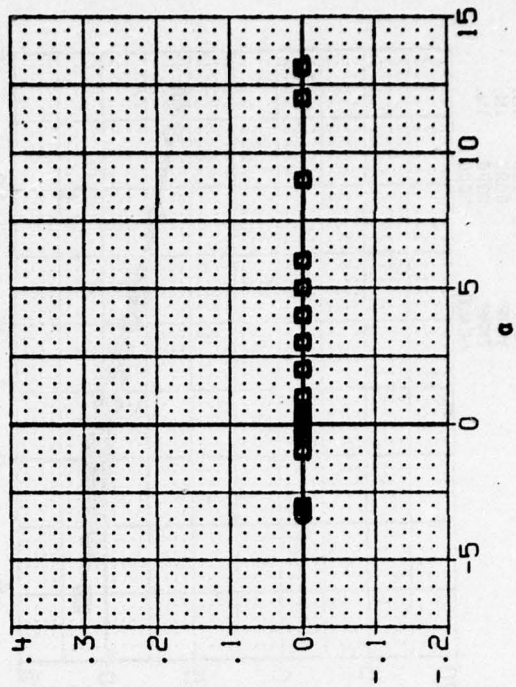
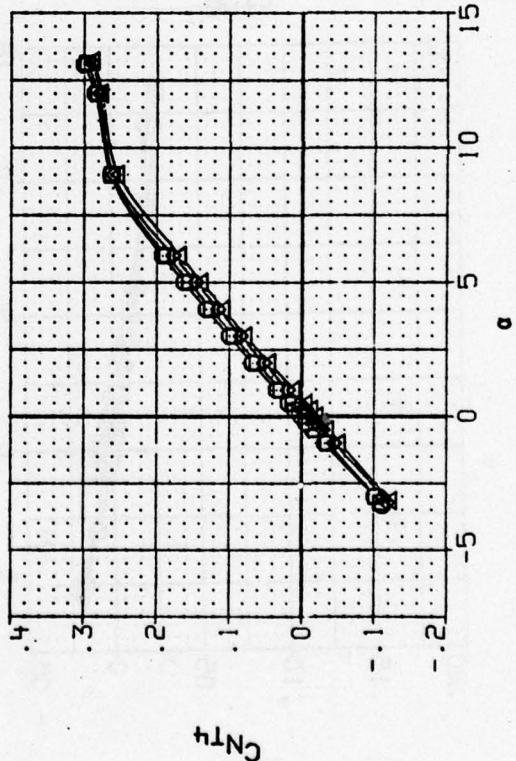
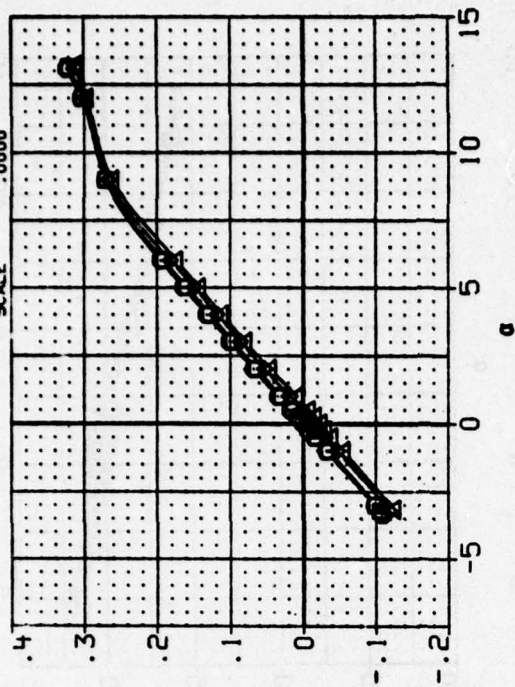
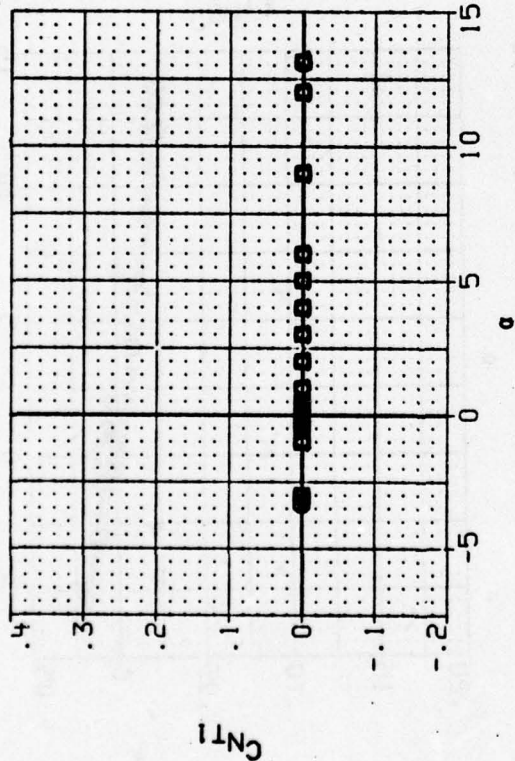
REFERENCE INFORMATION

SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XREF 26.0000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAIL=0 PHICND=45
(B) MACH = 3.01

DATA SET	SYMBOL	CONFIGURATION DESCRIPTION	DCND1	DCND2	DCND3	DCND4	REFERENCE INFORMATION
(AXH093)	○	AEDC W4IA-CIA, CANARD CONTROL, BNICITR	.000	.000	.000	.000	SREF 19.6350 50. IN.
(AXH094)	□	AEDC W4IA-CIA, CANARD CONTROL, BNICITR	.000	3.000	.000	3.000	LREF 5.0000 IN.
(AXH095)	◇	AEDC W4IA-CIA, CANARD CONTROL, BNICITR	.000	9.000	.000	9.000	BREF 5.000 IN.
(AXH096)	△	AEDC W4IA-CIA, CANARD CONTROL, BNICITR	.000	15.000	.000	15.000	XREF 26.00 IN.
							YREF .00 IN.
							ZREF .0000 IN.
							SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 $PHITAIL=0$ $PHICND=0$
 (A) MACH = 2.50

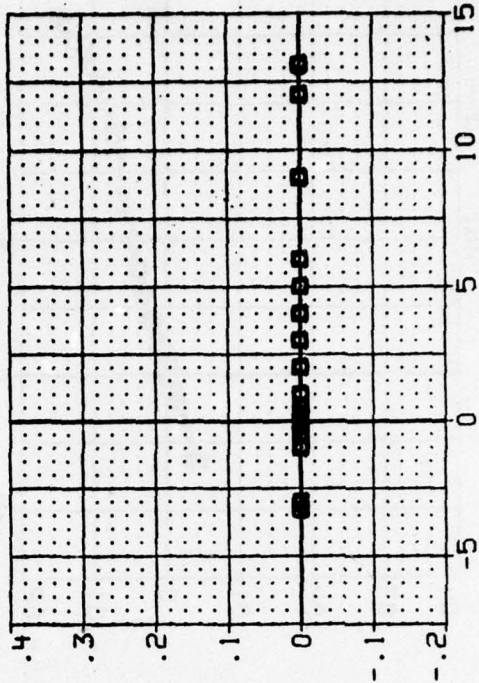
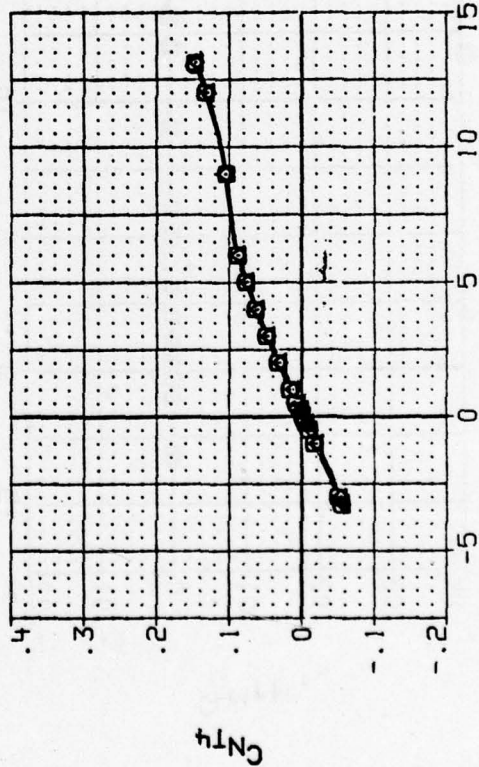
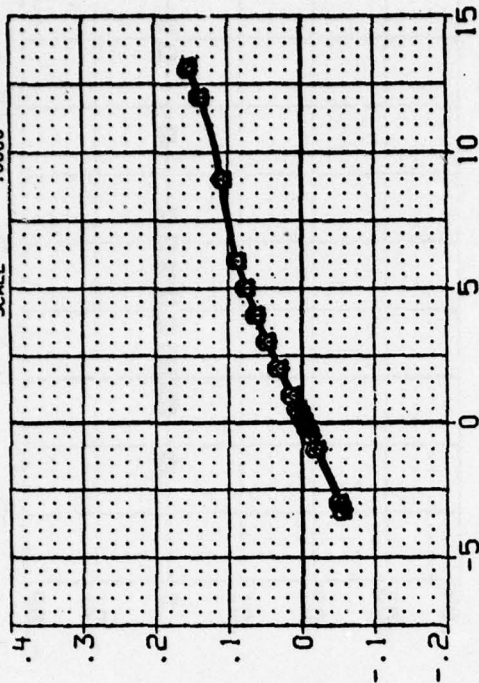
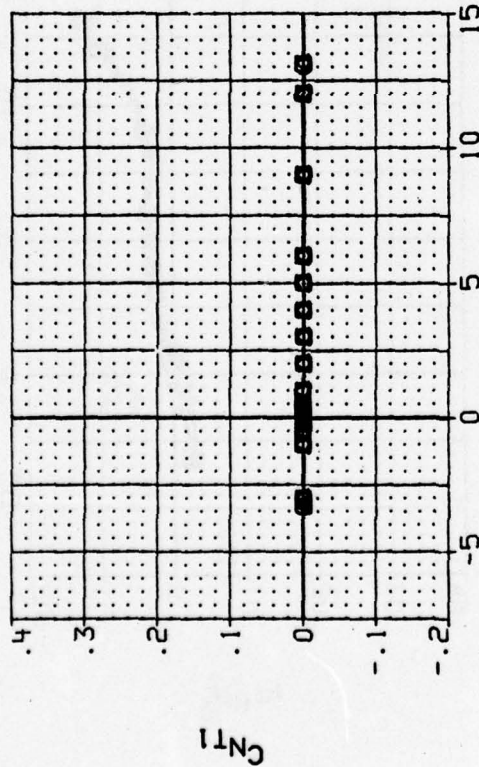
DATA SET SYMBOL
(AXH093)
(AXH094)
(AXH095)
(AXH096)

CONFIGURATION DESCRIPTION
AEDC V41A-CIA, CANARD CONTROL.
AEDC V41A-CIA, CANARD CONTROL.
AEDC V41A-CIA, CANARD CONTROL.
AEDC V41A-CIA, CANARD CONTROL.

BNIC1TR
BNIC1TR
BNIC1TR
BNIC1TR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAIL=0 PHICND=0

(B)MACH = 4.52

DATA SET SYMBOL
(CXH093)
(CXH094)
(CXH095)
(CXH096)

CONFIGURATION DESCRIPTION

AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR

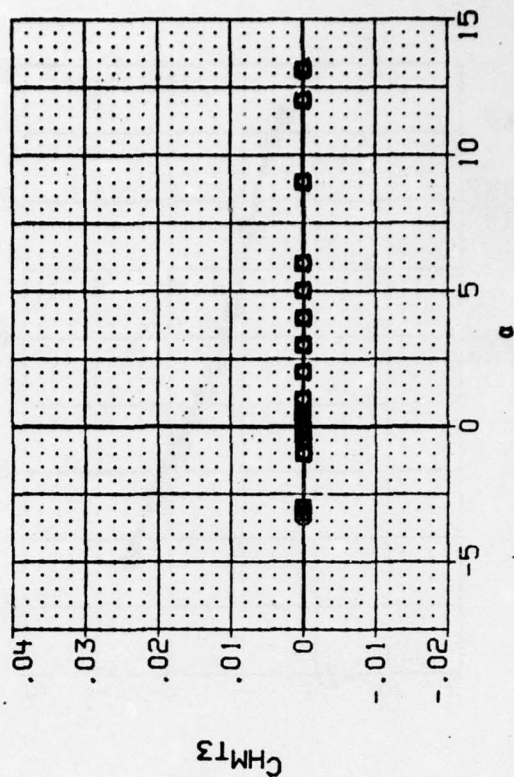
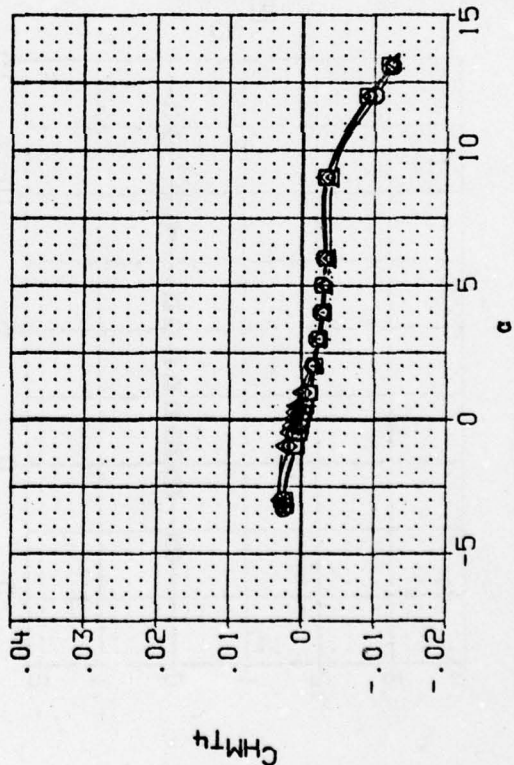
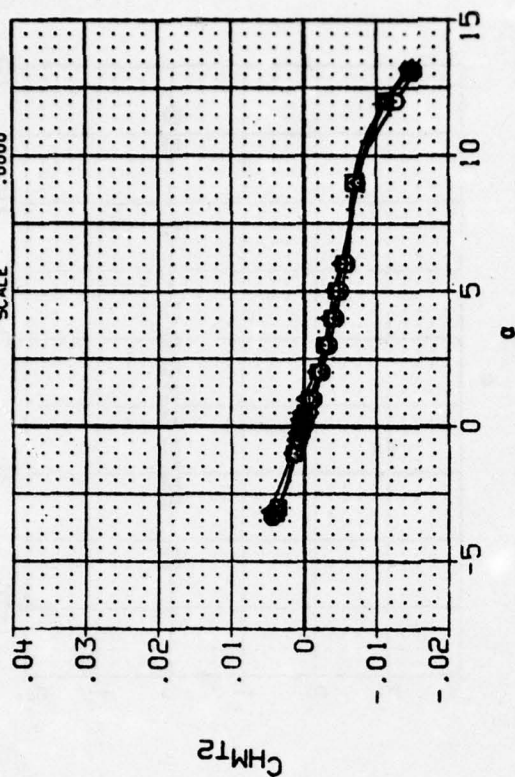
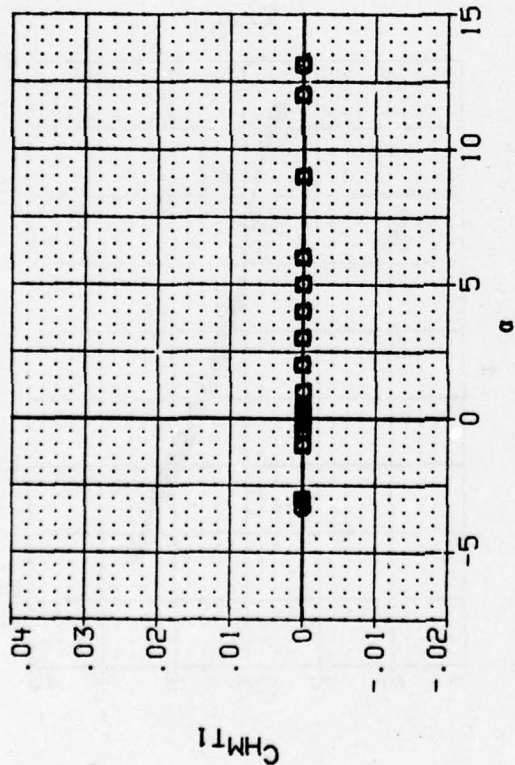
DCND1
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DCND2
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DCND3
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DCND4
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REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XREF 26.000 IN.
YREF .001 IN.
ZREF .001 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAIL=0 PHICND=0

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(CXH093) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

(CXH094) \square AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

(CXH095) \diamond AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

(CXH096) \triangle AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

.000 .000 .000 .000

.000 3.000 .000 .000

.000 .000 .000 .000

.000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

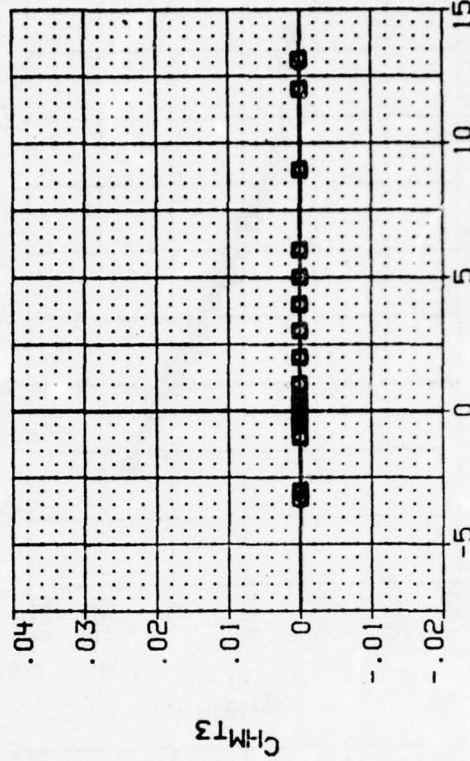
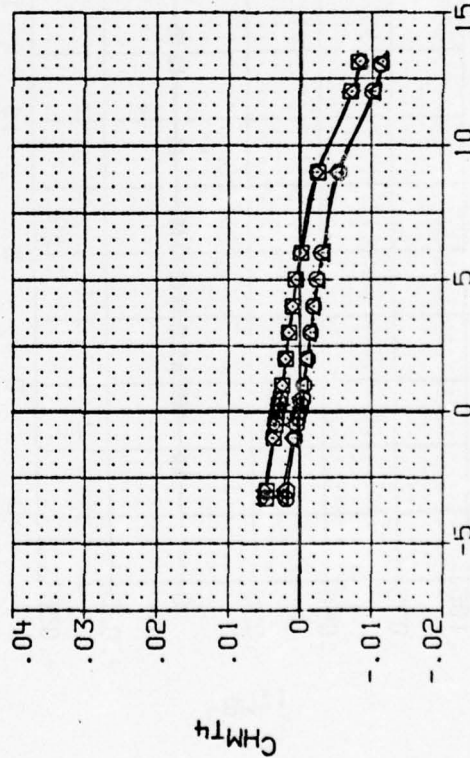
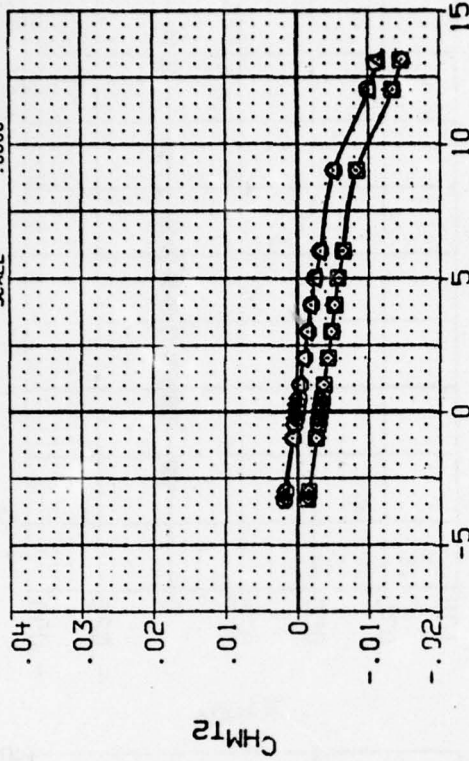
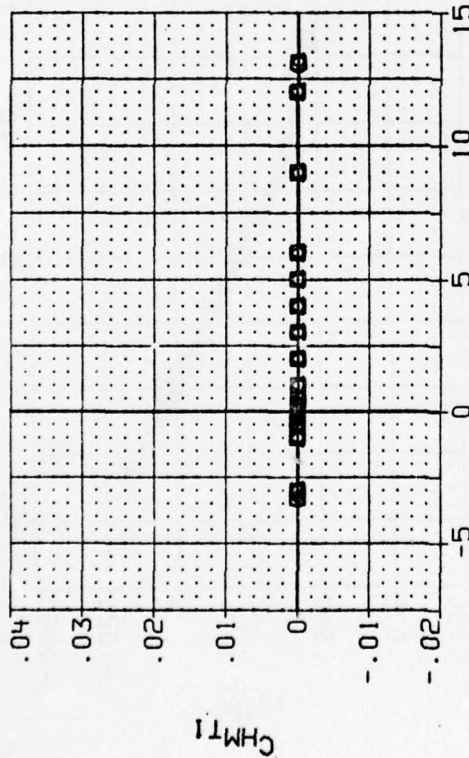
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAL=0 PHICND=0

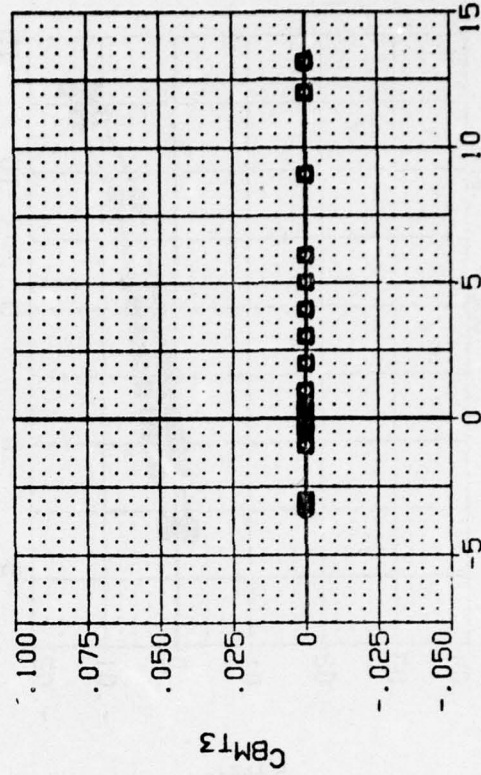
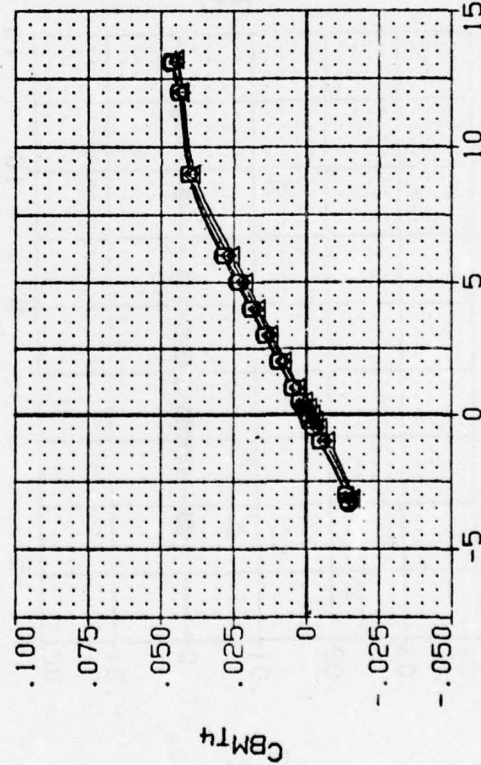
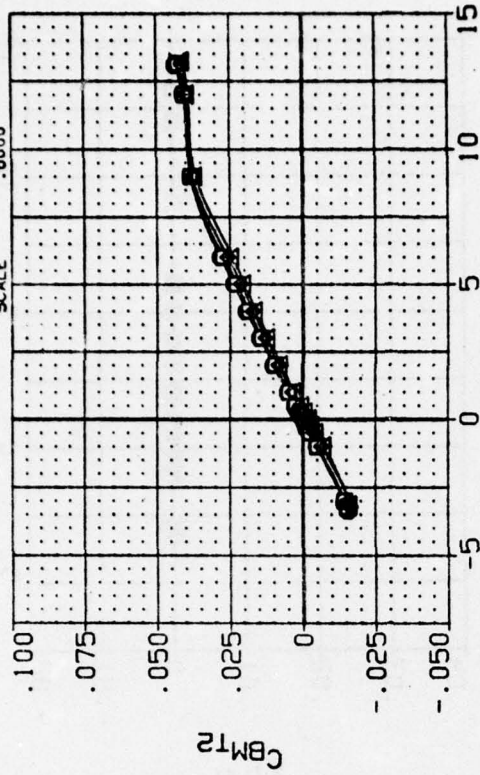
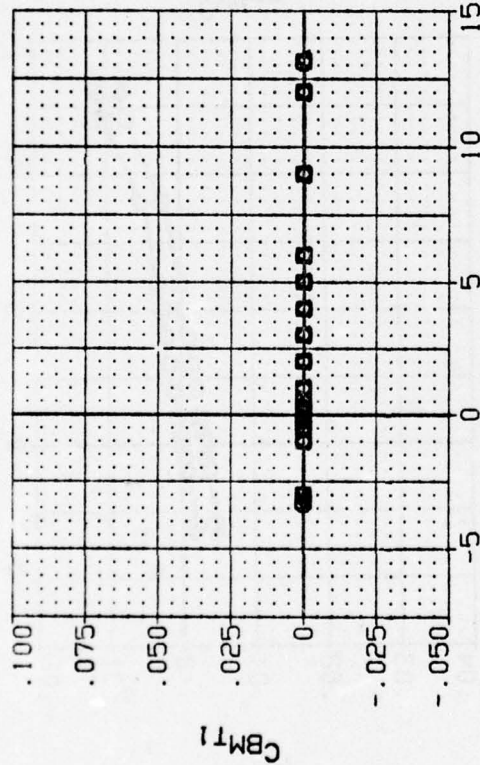
(B) MACH = 4.52

DATA SET SYMBOL
(CXH093) □
(CXH094) ◇
(CXH095) △

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAL=0 PHICND=0

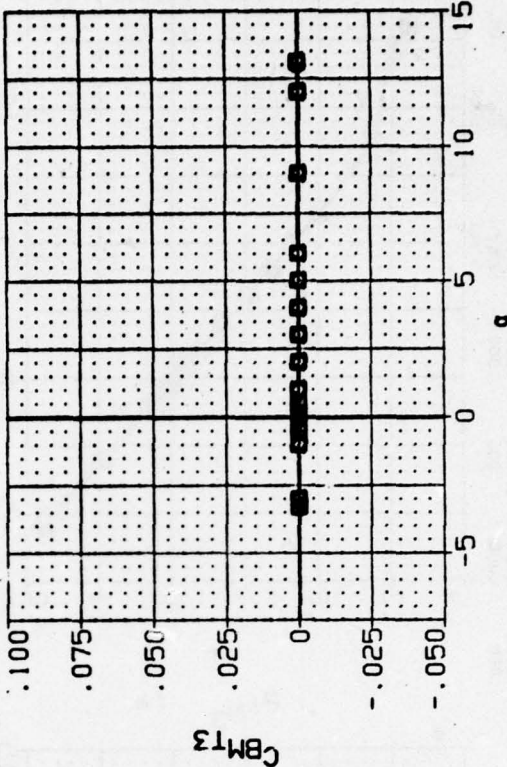
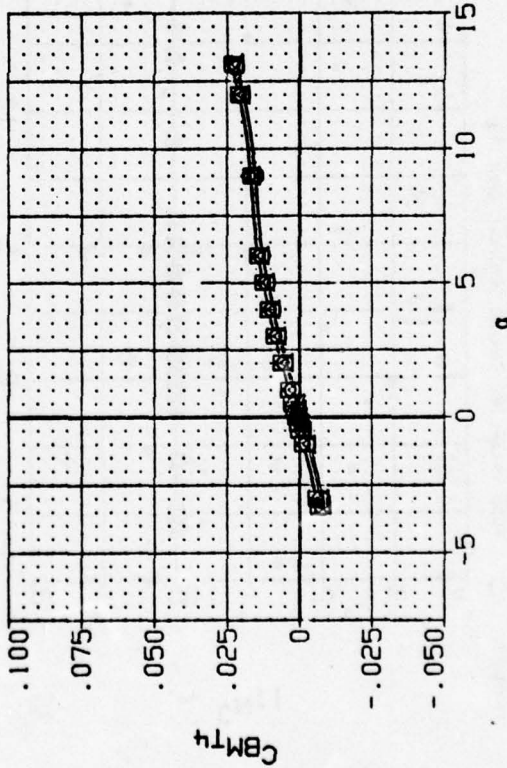
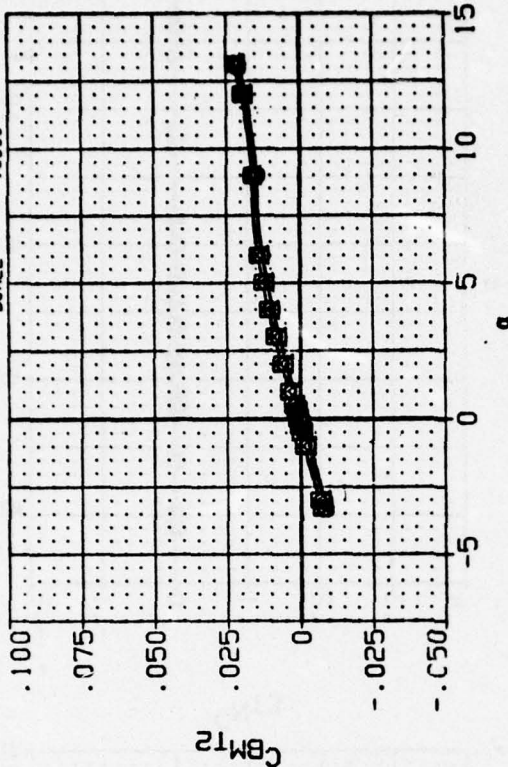
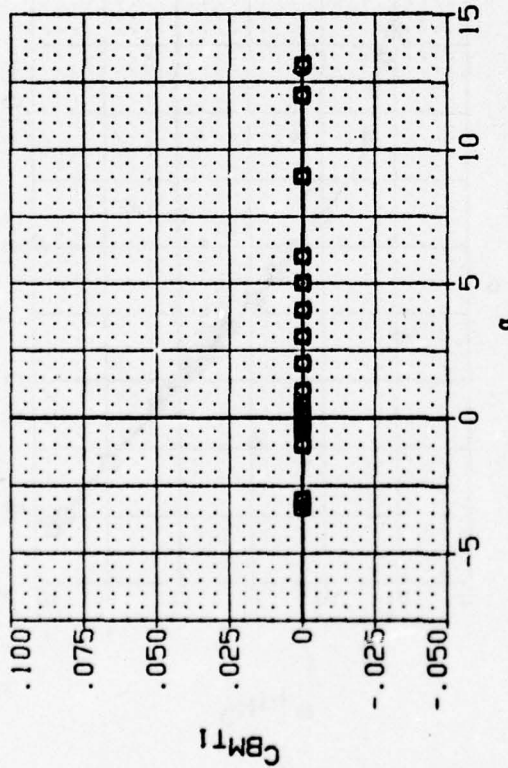
(A) MACH = 2.50

DATA SET SYMBOL
(CXH093) □
(CXH094) ◇
(CXH095) △
(CXH096) △

CONFIGURATION DESCRIPTION
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
.000 3.000 .000 3.000
.000 9.000 .000 9.000
.000 15.000 .000 15.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



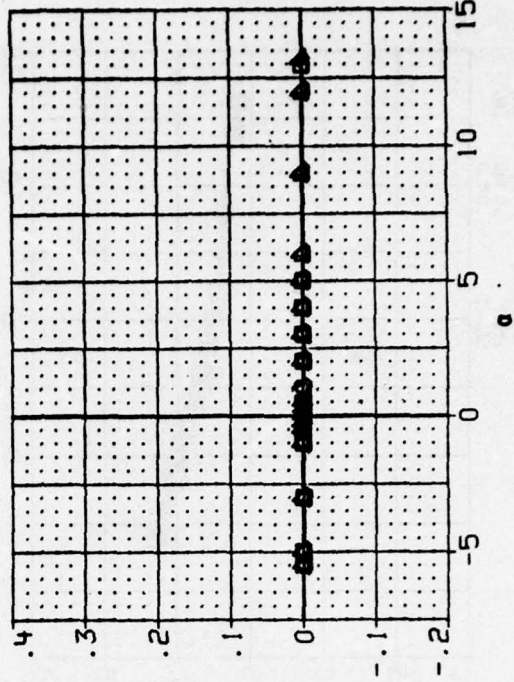
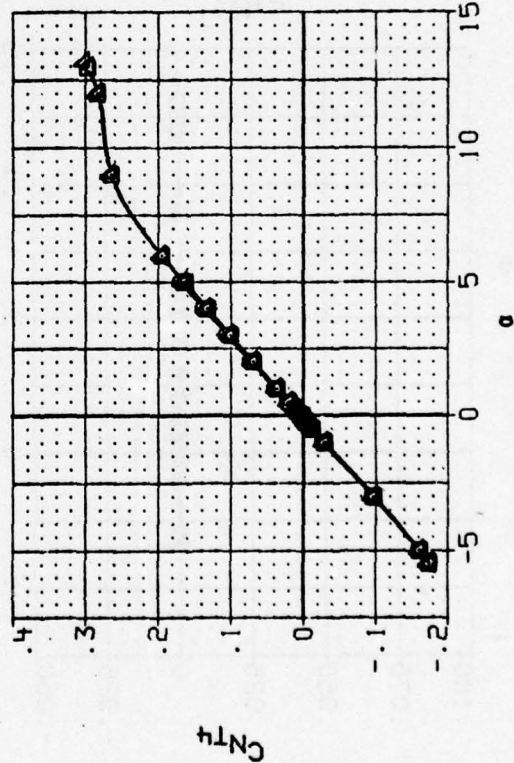
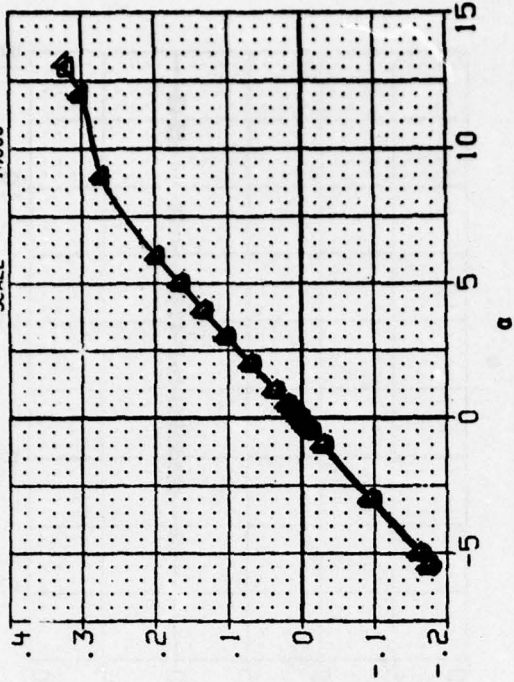
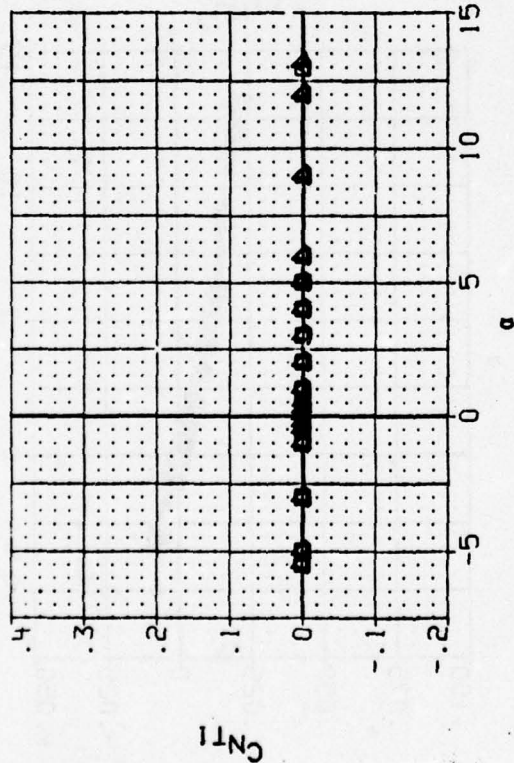
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAI=0 PHICND=0
(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH097) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH098) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH099) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH100) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH101) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH102) AEDC V41A-CIA. CANARD CONTROL. BNICITR
 (AXH103) AEDC V41A-CIA. CANARD CONTROL. BNICITR

DCND1 DCND2 DCND3 DCND4

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.0000 IN.
 YMRP .000 IN.
 ZMRP .000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAI=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH097) \square AEDC V4IA-CIA, CANARD CONTROL, BNICITR

(AXH098) \square AEDC V4IA-CIA, CANARD CONTROL, BNICITR

(AXH099) \diamond DATA NOT AVAILABLE

(AXH101) \triangle DATA NOT AVAILABLE

(AXH102) \triangle AEDC V4IA-CIA, CANARD CONTROL, BNICITR

(AXH103) \triangle AEDC V4IA-CIA, CANARD CONTROL, BNICITR

DCND1 5.000
2.000
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DCND2 -5.000
-2.000
.000
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DCND3 -5.000
-2.000
.000
.000
.000
.000

DCND4 5.000
2.000
.000
.000
.000
.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

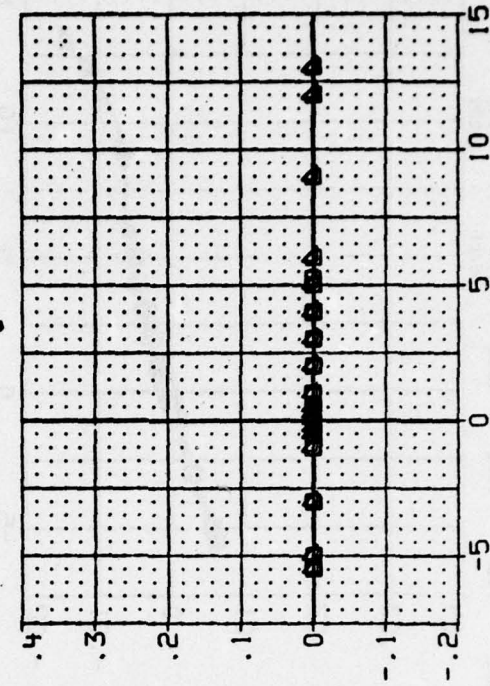
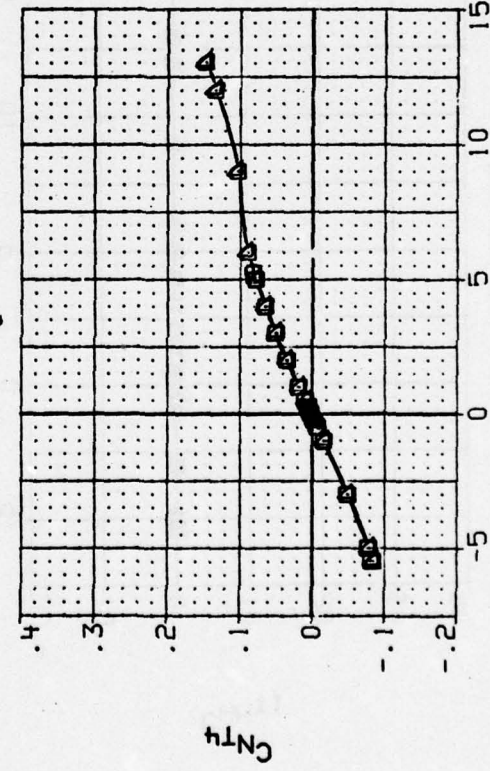
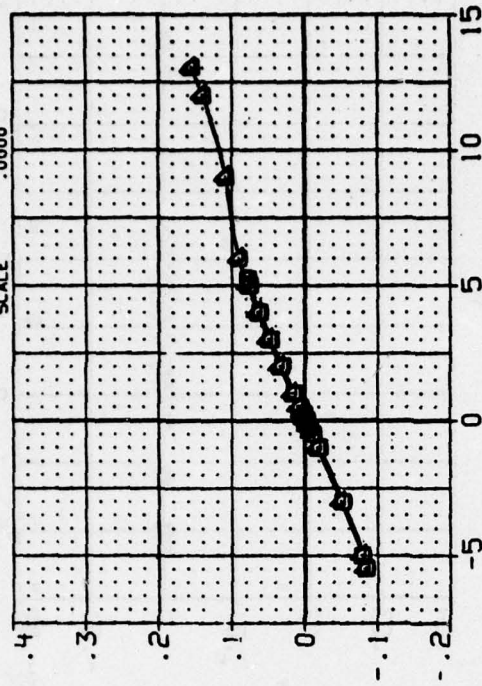
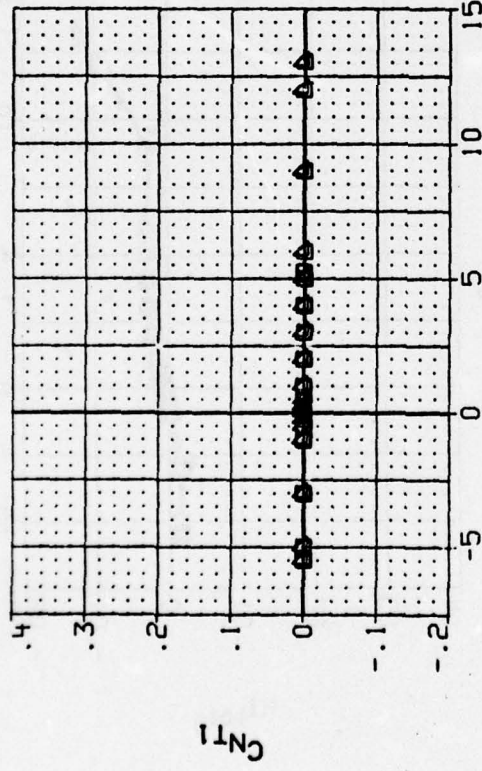
BREF 5.0000 IN.

XPRP 26.0000 IN.

YPRP .0000 IN.

ZPRP .0000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHICAL=0 PHICND=0

(B) MACH = 4.52

DATA SET SYMBOL

(CXH097) □

(CXH098) ◇

(CXH099) △

(CXH101) ▽

(CXH102) ▽

(CXH103) ▽

CONFIGURATION DESCRIPTION

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

5.000 -5.000 -5.000 5.000

2.000 -2.000 -2.000 2.000

.000 .000 .000 .000

.000 .000 .000 .000

.000 .000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

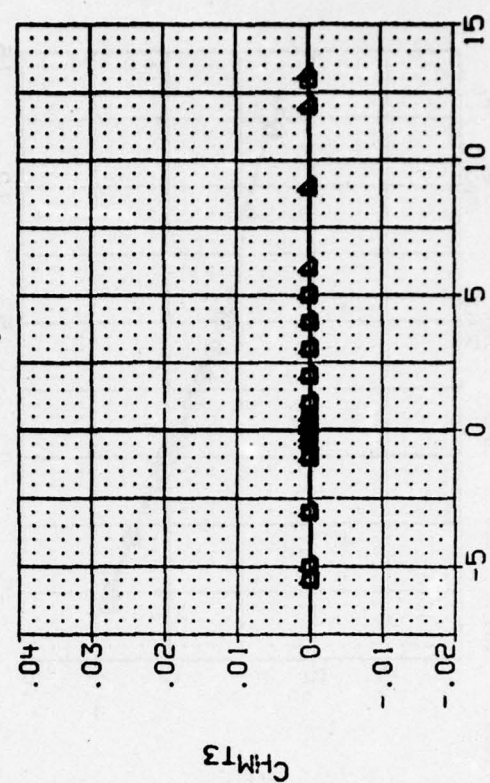
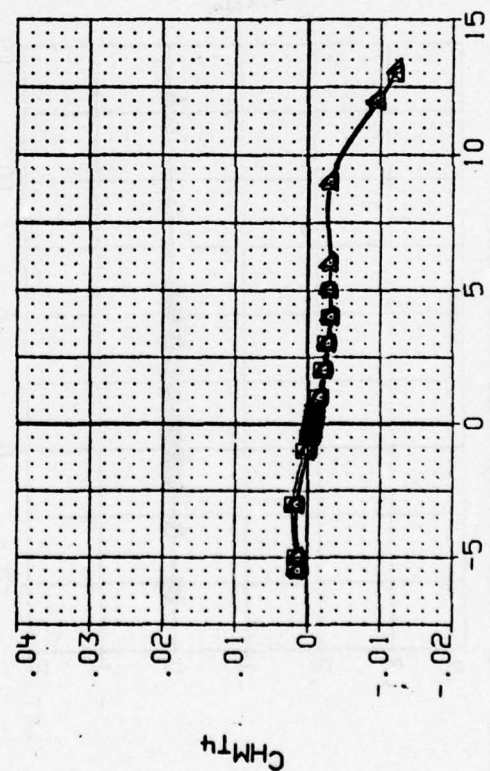
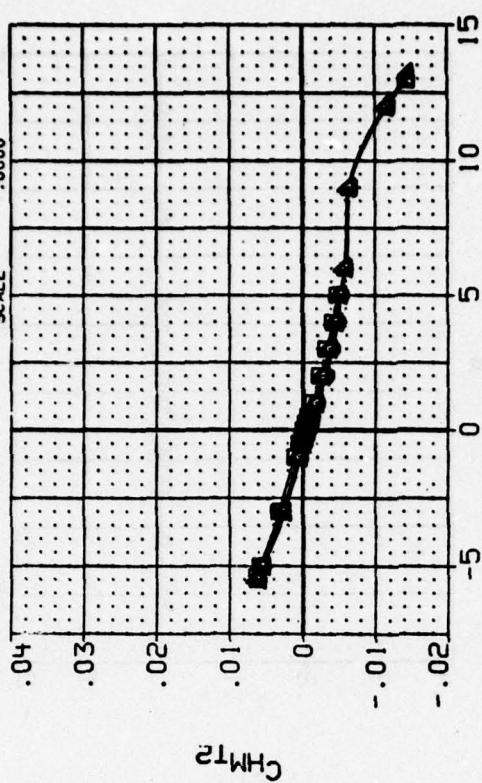
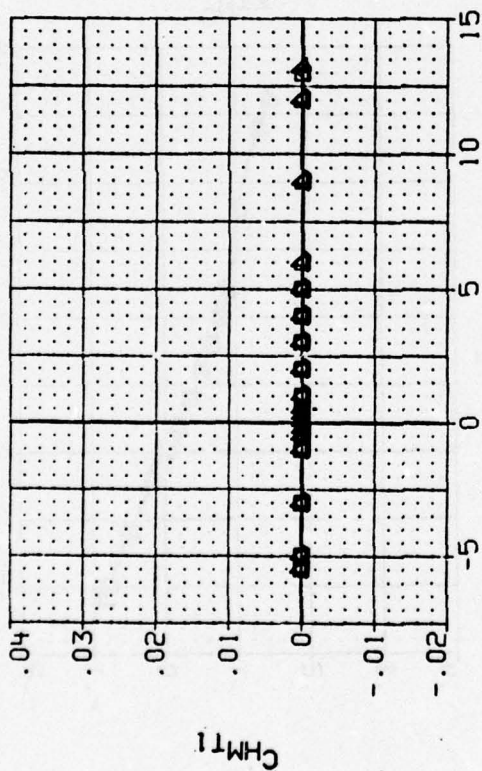
BREF 5.0000 IN.

XPRP 26.000 IN.

YPRP .000 IN.

ZPRP .000 IN.

SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAIL=0 PHICND=0

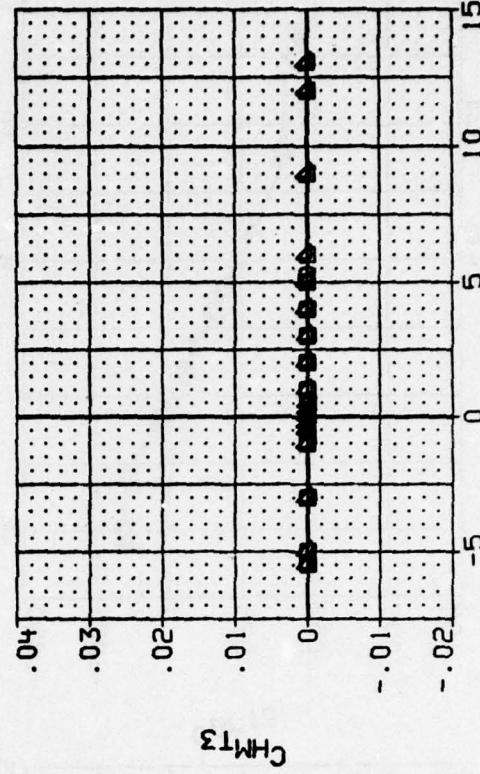
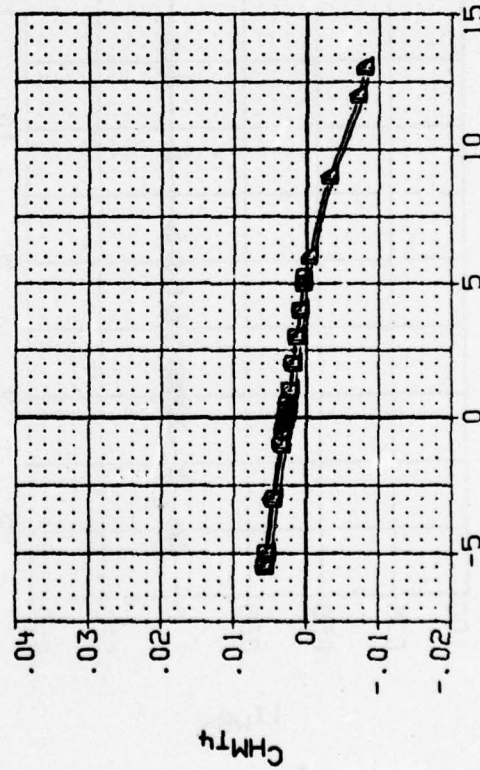
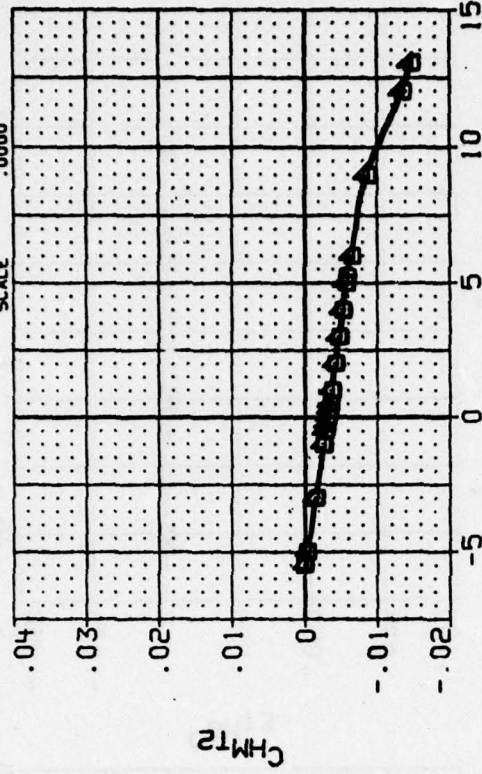
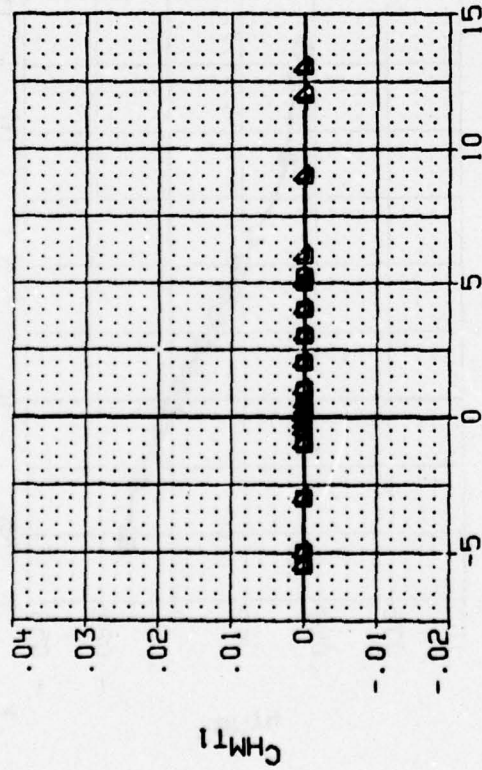
(A) MACH = 2.50

DATA SET SYMBOL
(CXH097) □
(CXH098) □
(CXH099) ◇
(CXH101) △
(CXH102) ▽
(CXH103) ▽

CONFIGURATION DESCRIPTION
AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR
DATA NOT AVAILABLE
AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
5.000 -5.000 -5.000 5.000
2.000 -2.000 -2.000 2.000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000
.000 .000 .000 .000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAL=0 PHICND=0

(B) MACH = 1.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

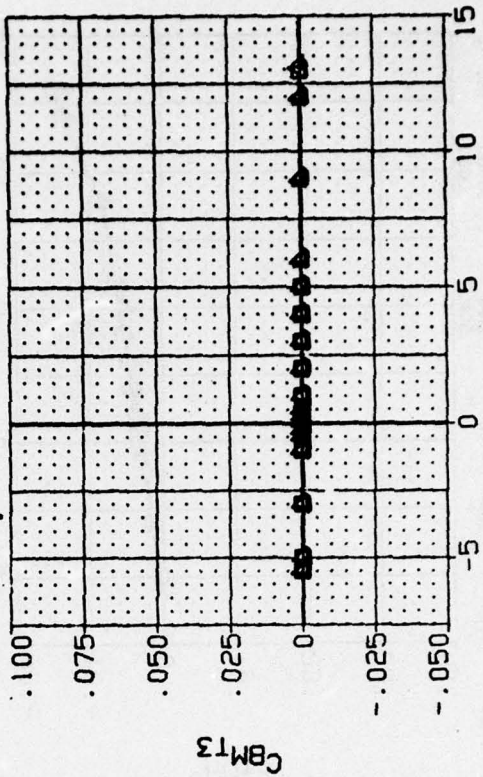
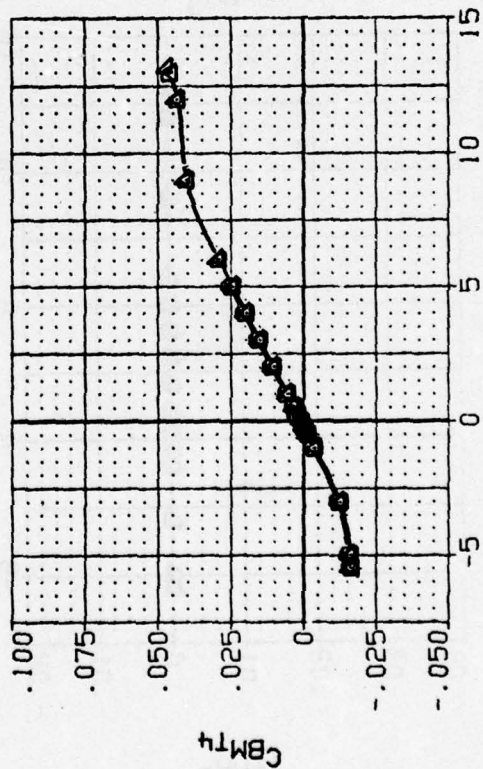
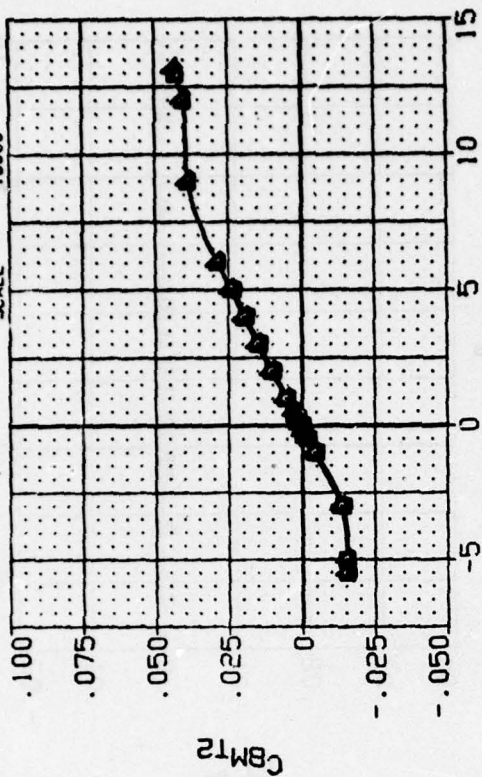
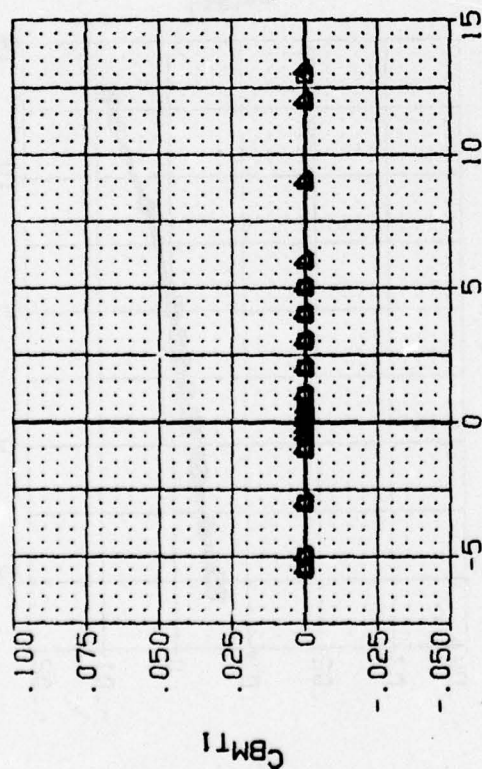
(CXH097) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (CXH098) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (CXH099) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (CXH101) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (CXH102) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
 (CXH103) AEDC W1A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4

5.000 -5.000 -5.000 5.000
 2.000 -2.000 -2.000 2.000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 10. IN.
 BREF 5.0000 10. IN.
 XREF 26.0000 10. IN.
 YREF .0000 10. IN.
 ZREF .0000 10. IN.
 SCALE .0000

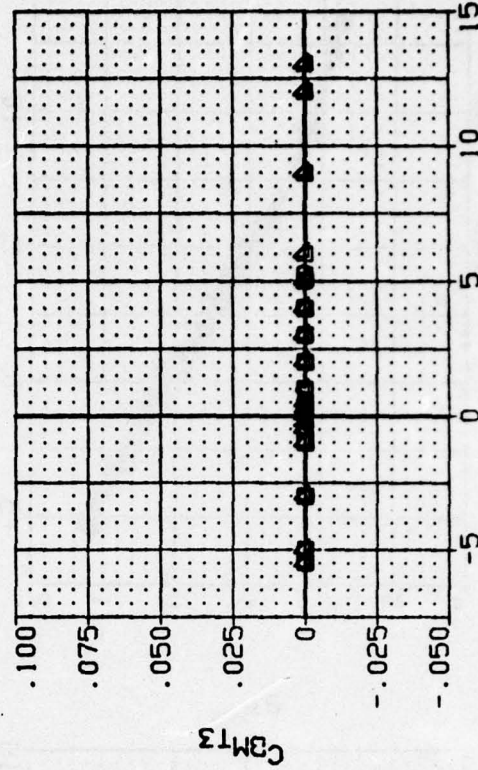
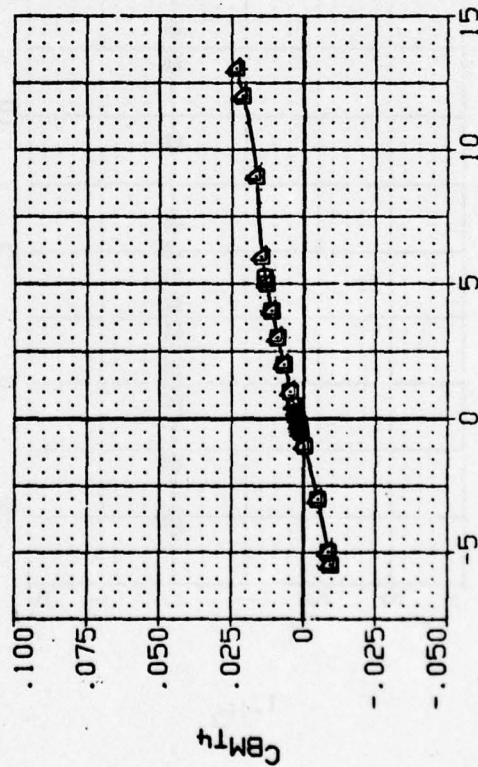
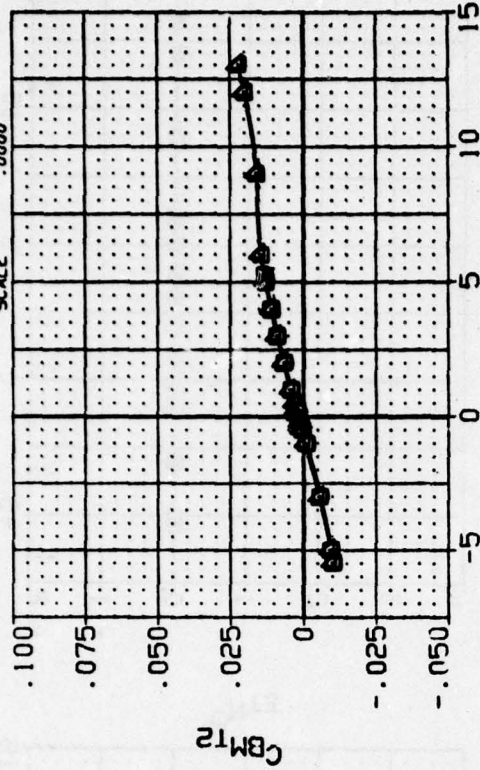
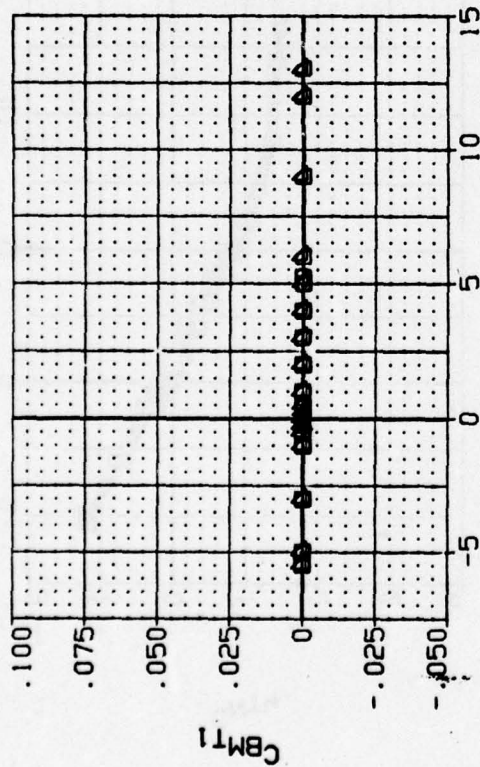


EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH097) □ AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (CXH098) ◇ AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (CXH099) △ DATA NOT AVAILABLE
 (CXH101) ▽ DATA NOT AVAILABLE
 (CXH102) ▽ AEDC V41A-C1A, CANARD CONTROL, BNIC1TR
 (CXH103) ▽ AEDC V41A-C1A, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4
 5.000 -5.000 -5.000 5.000
 2.000 -2.000 -2.000 2.000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.0000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000



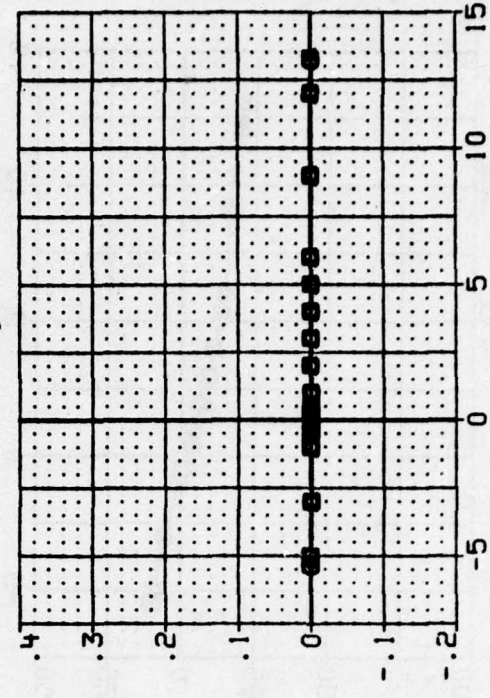
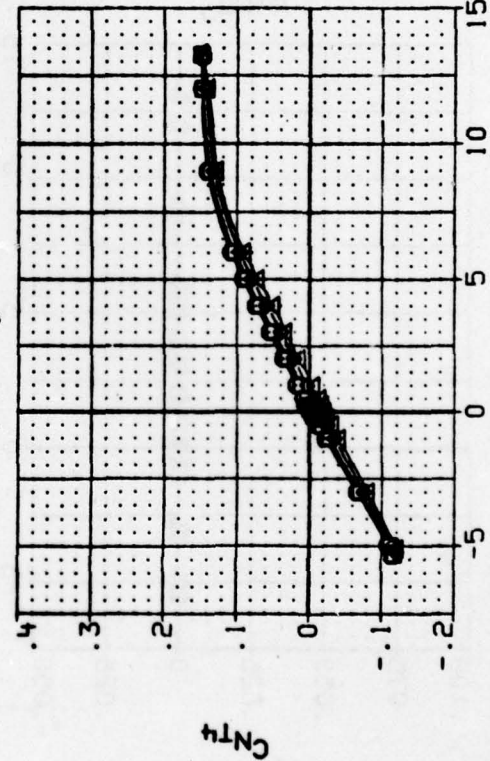
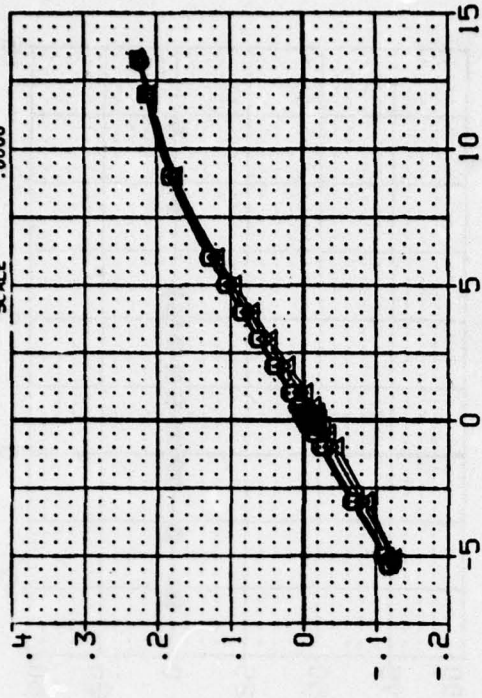
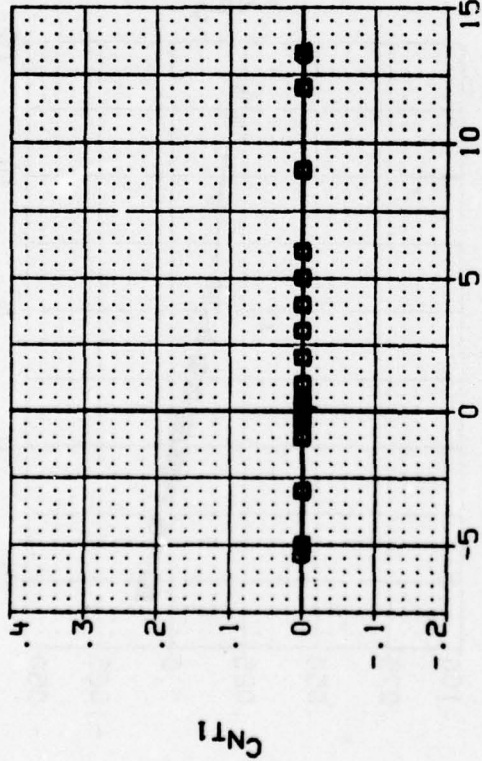
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=0
 (B) MACH = 4.52

DATA SET SYMBOL
 (AXH104)
 (AXH105)
 (AXH106)
 (AXH107)

CONFIGURATION DESCRIPTION
 AEDC WTA-CIA, CANARD CONTROL, BNICITR
 AEDC WTA-CIA, CANARD CONTROL, BNICITR
 AEDC WTA-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XRRP 26.000 IN.
 YRRP .0000 IN.
 ZRRP .0000 IN.
 SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL

PHITAIL=0 PHICND=45

(A) MACH = 2.50

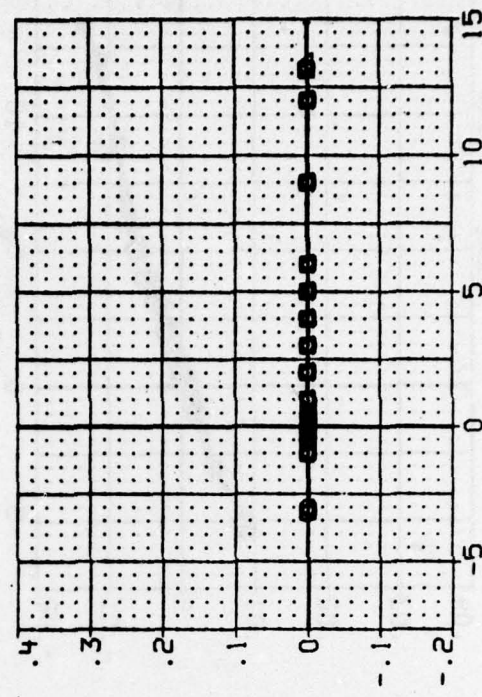
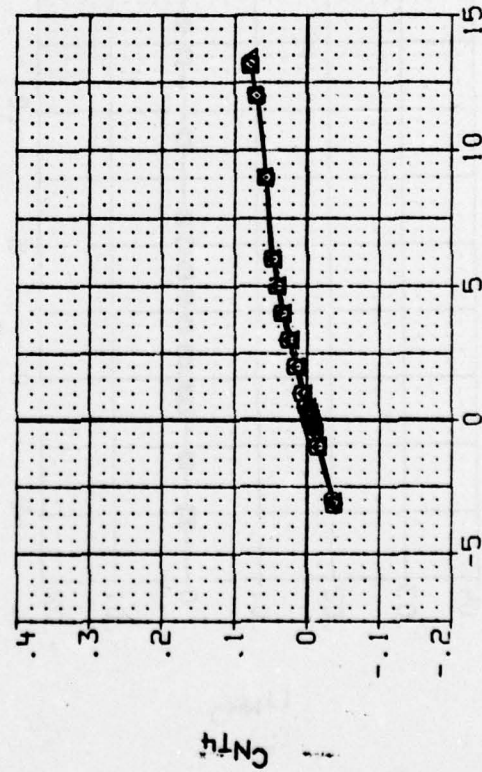
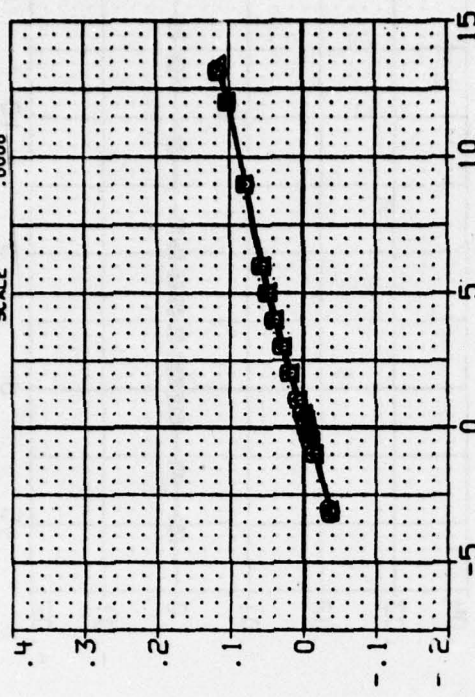
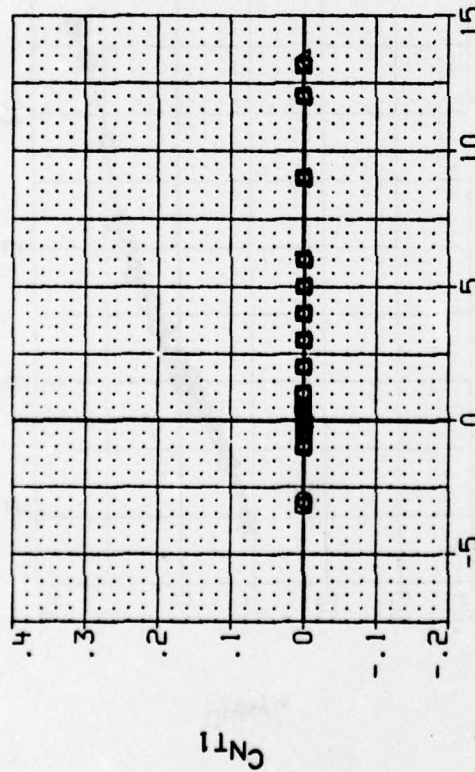
DATA SET SYMBOL
(AXH104)
(AXH105)
(AXH106)
(AXH107)

CONFIGURATION DESCRIPTION
AEDC VN1A-CIA, CANARD CONTROL
AEDC VN1A-CIA, CANARD CONTROL
AEDC VN1A-CIA, CANARD CONTROL

BNIC1TR
BNIC1TR
BNIC1TR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
3.000 3.000 3.000 3.000
9.000 9.000 9.000 9.000
15.000 15.000 15.000 15.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAIL=0 PHICND=45
(B)MACH = 4.52

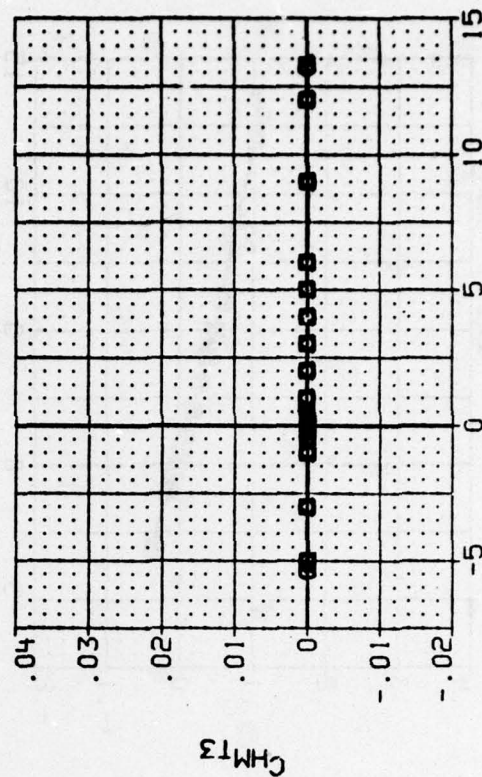
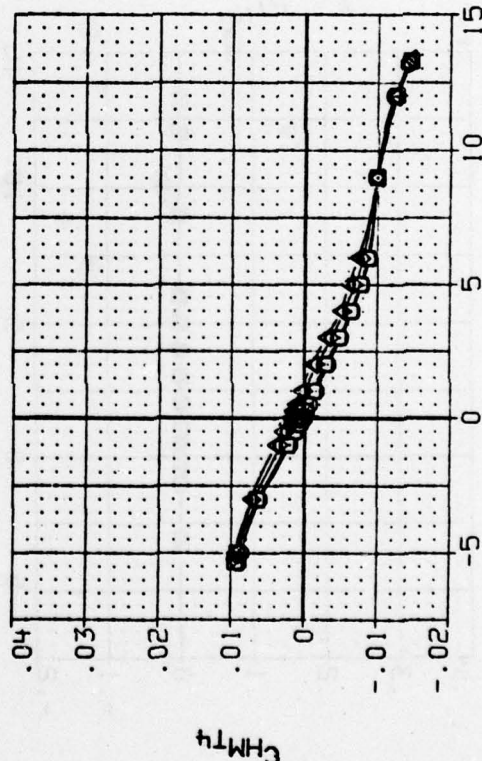
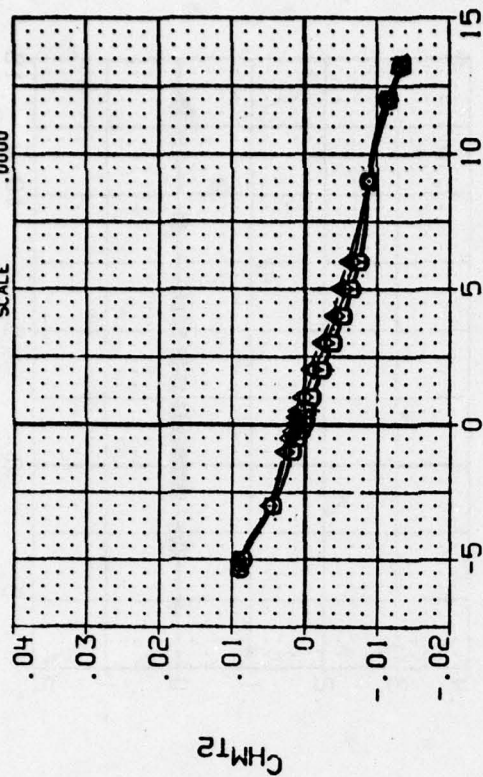
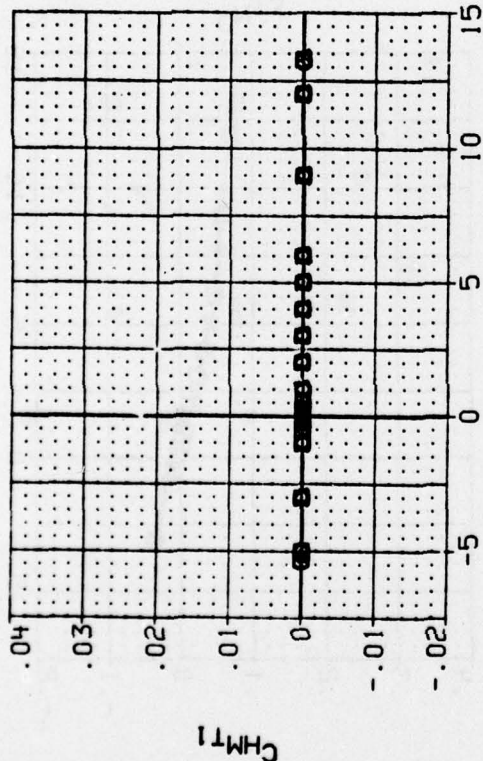
DATA SET SYMBOL
(CXH104)
(CXH105)
(CXH106)
(CXH107)

CONFIGURATION DESCRIPTION

AEDC V-1A-CIA, CANARD CONTROL, BNICITR
AEDC V-1A-CIA, CANARD CONTROL, BNICITR
AEDC V-1A-CIA, CANARD CONTROL, BNICITR
AEDC V-1A-CIA, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
3.000 3.000 3.000 3.000
9.000 15.000 15.000 15.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .000 IN.
ZMRP .000 IN.
SCALE .0000



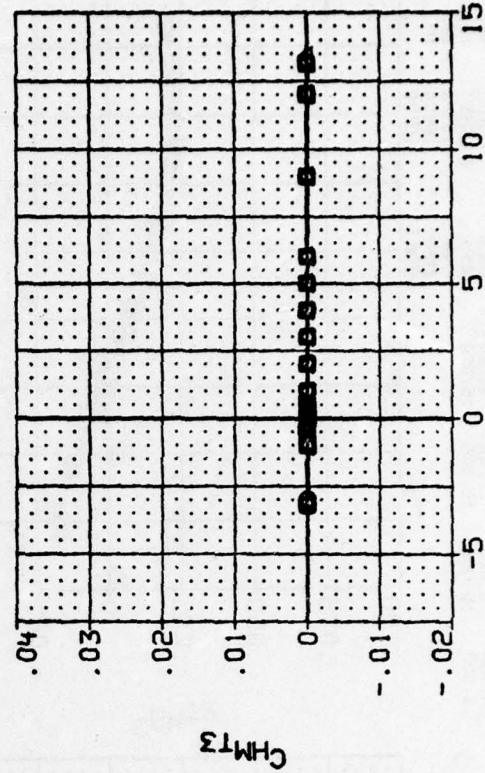
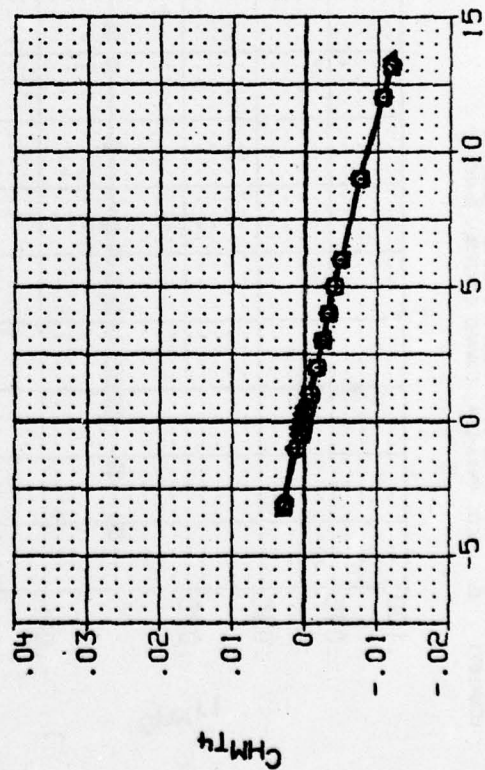
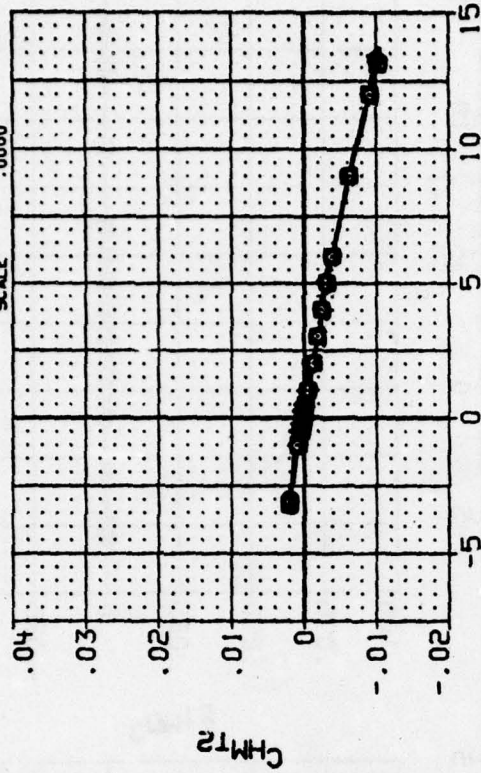
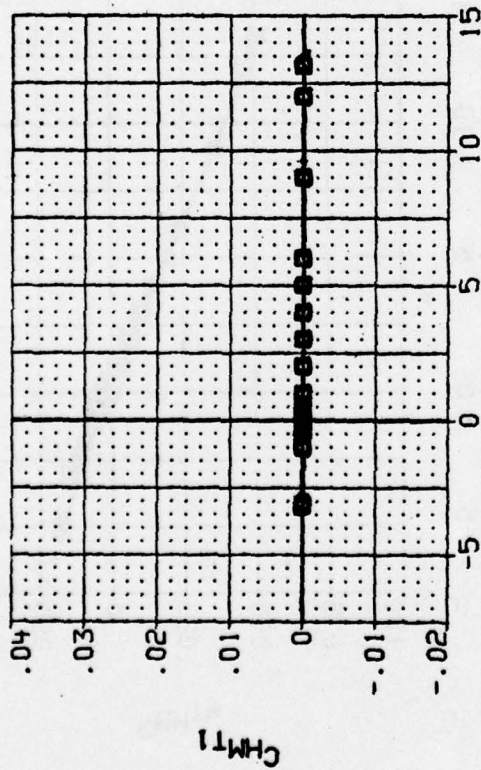
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAL=0 PHICND=45

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CXH104) O AEDC V41A-C1A, CANARD CONTROL, BNICITR
 (CXH105) □ AEDC V41A-C1A, CANARD CONTROL, BNICITR
 (CXH106) △ AEDC V41A-C1A, CANARD CONTROL, BNICITR
 (CXH107) ◇ AEDC V41A-C1A, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
 .000 .000 .000 .000
 3.000 3.000 3.000 3.000
 9.000 9.000 9.000 9.000
 15.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XREF 26.0000 IN.
 YREF .0000 IN.
 ZREF .0000 IN.
 SCALE .0000



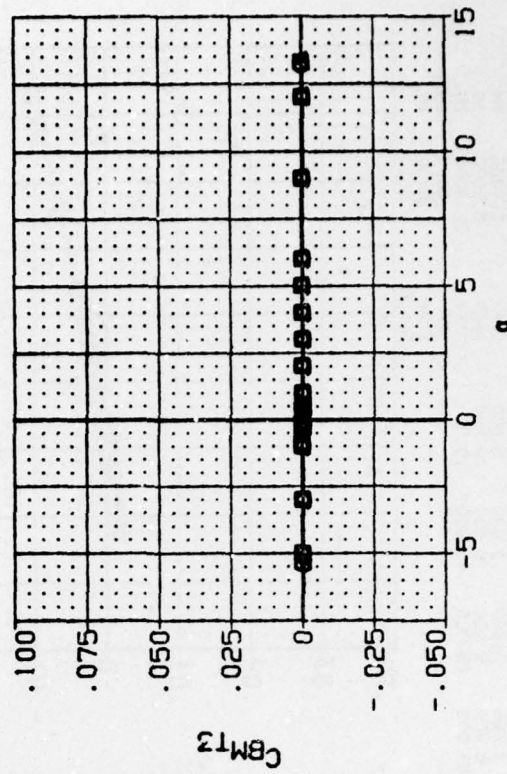
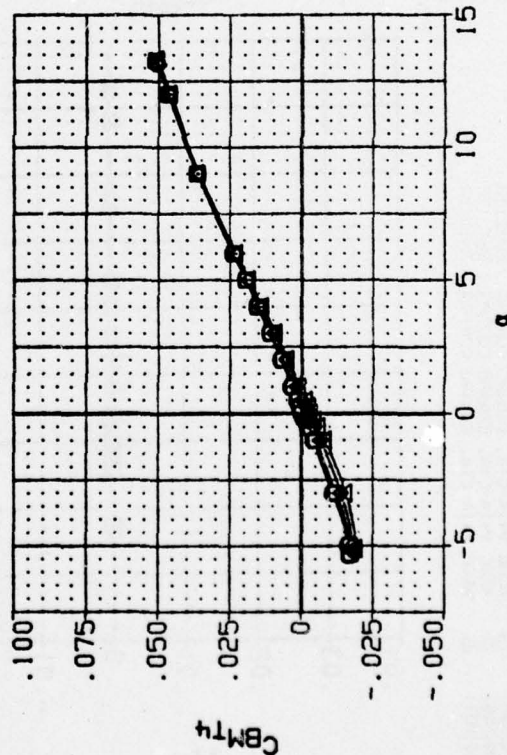
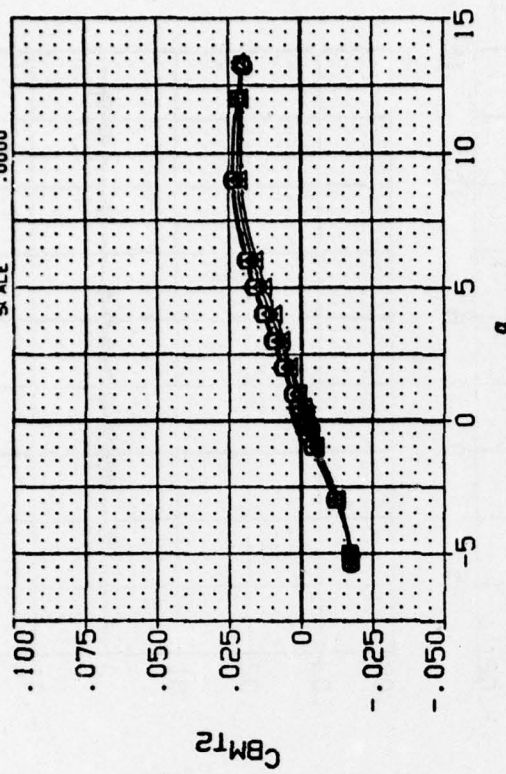
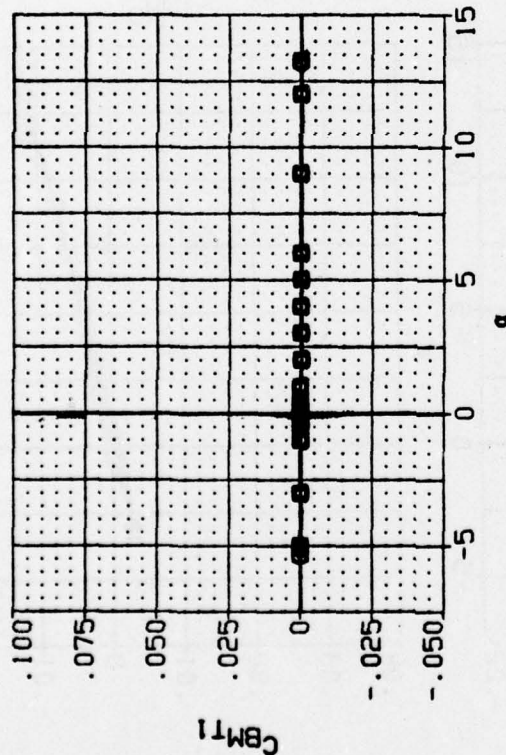
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
 PHITAIL=0 PHICND=45
 (B) MACH = 4.52

DATA SET SYMBOL
(CXH104) □
(CXH105) ◇
(CXH106) △
(CXH107)

CONFIGURATION DESCRIPTION
AEDC V1A-C1A, CANARD CONTROL, BNICITR
AEDC V1A-C1A, CANARD CONTROL, BNICITR
AEDC V1A-C1A, CANARD CONTROL, BNICITR
AEDC V1A-C1A, CANARD CONTROL, BNICITR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
2.000 3.000 3.000 3.000
9.000 9.000 1.000 15.000
15.000 15.000 15.000 15.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.000 IN.
BREF 5.000 IN.
XREF 25.000 IN.
YREF .0000 IN.
ZREF .0000 IN.
SCALE .0000



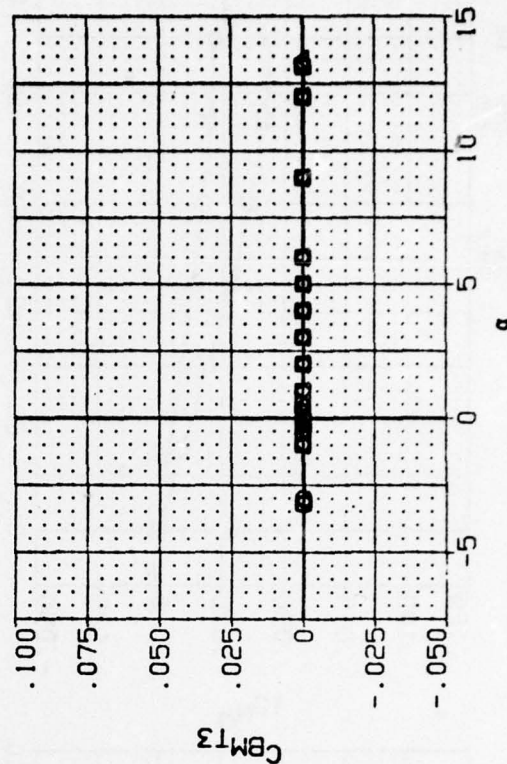
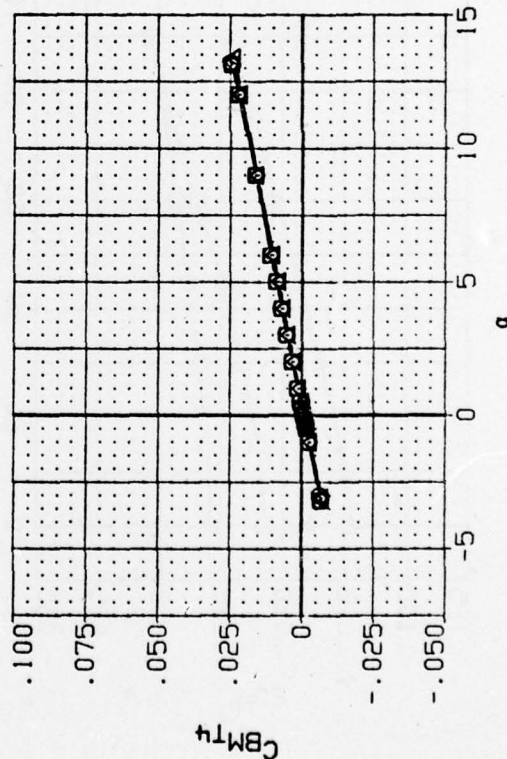
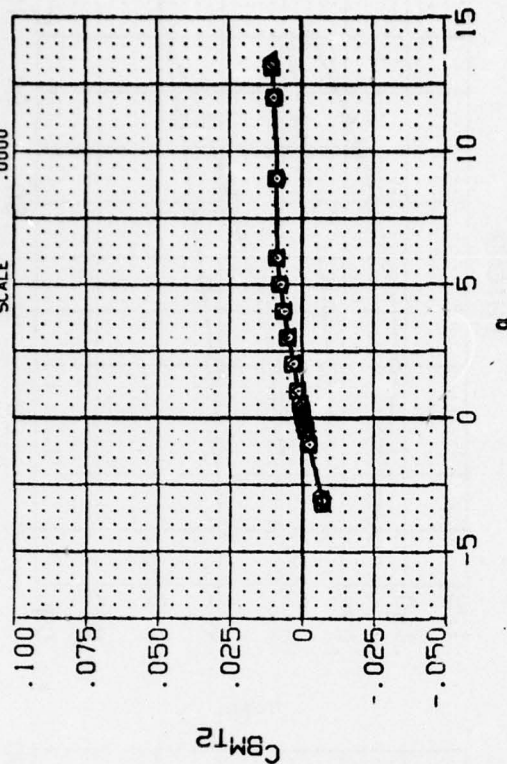
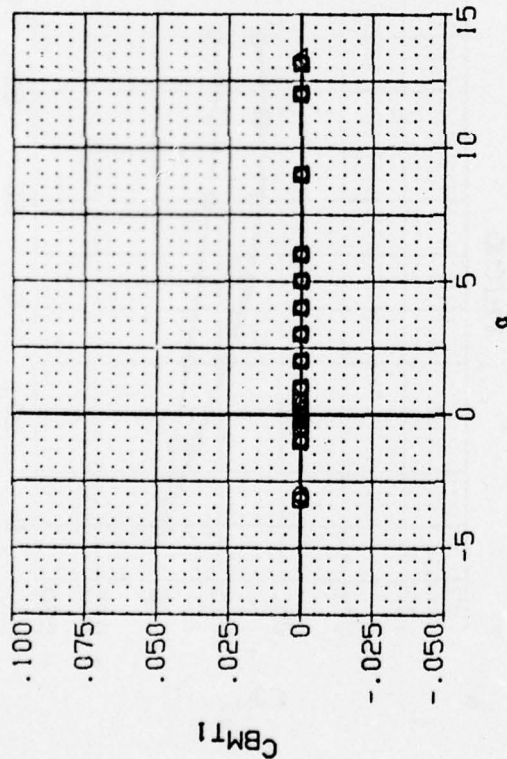
EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAL=0 PHICND=45
(A) MACH = 2.50

DATA SET SYMBOL
(CXH104)
(CXH105)
(CXH106)
(CXH107)

CONFIGURATION DESCRIPTION
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR
AEDC V41A-CIA, CANARD CONTROL, BNIC1TR

DCND1 DCND2 DCND3 DCND4
.000 .000 .000 .000
3.000 3.000 3.000 3.000
9.000 9.000 9.000 9.000
15.000 15.000 15.000 15.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF ANGLE OF ATTACK AND CANARD DEFLECTION ON TAIL
PHITAIL=0 PHICND=45
(B) MACH = 4.52

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH108) \square AEDC V41A-C1A, CANARD CONTROL, BNICITR

(AXH109) \diamond AEDC V41A-C1A, CANARD CONTROL, BNICITR

(AXH110) \triangle AEDC V41A-C1A, CANARD CONTROL, BNICITR

(AXH111) \square AEDC V41A-C1A, CANARD CONTROL, BNICITR

ALPHA DCND2 DCND3 DCND4

6.000 .000 .000 .000

12.000 .000 .000 .000

12.000 15.000 .000 .000

12.000 15.000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 IN.

LREF 5.0000 IN.

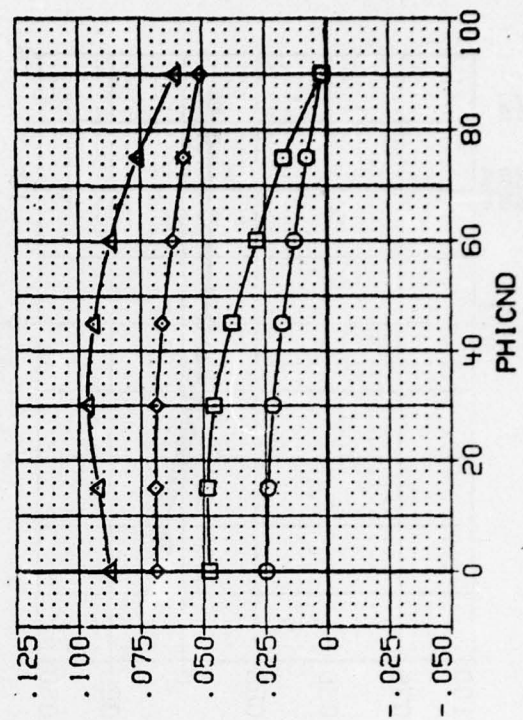
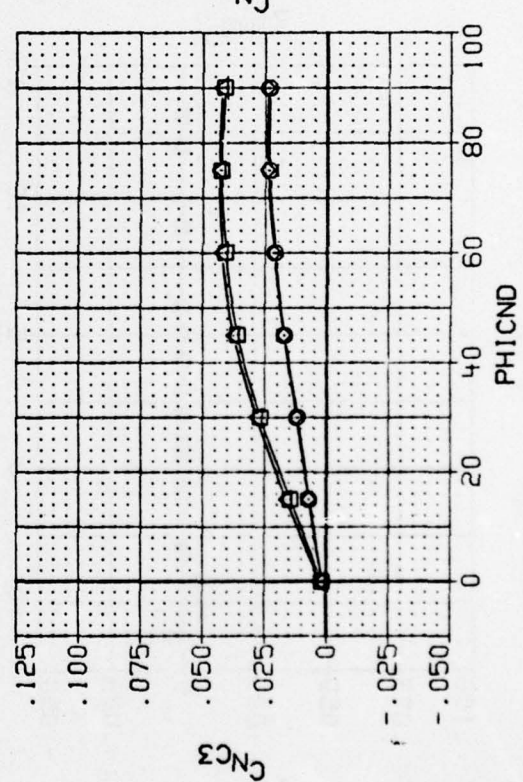
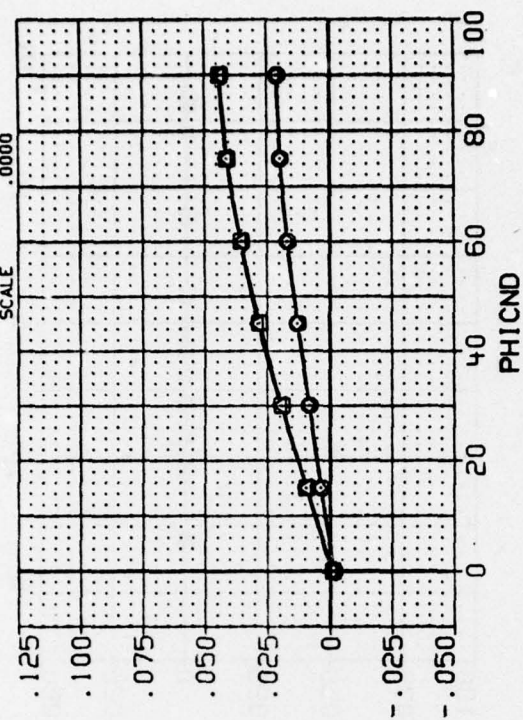
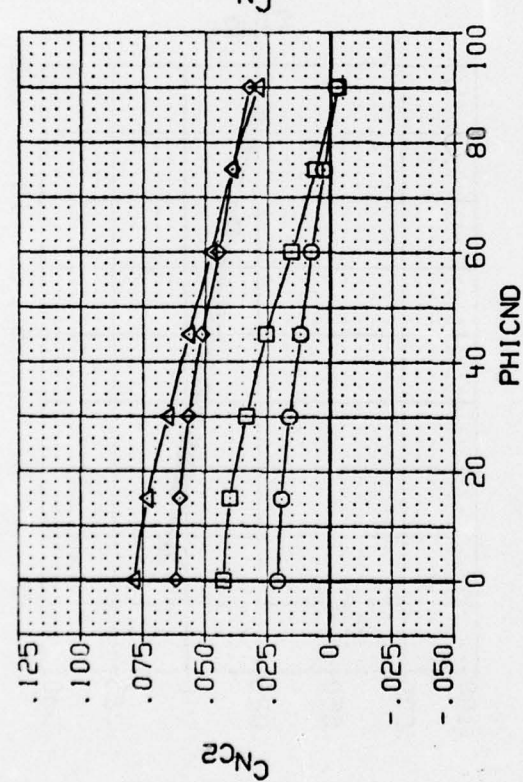
BREF 5.0000 IN.

XMRP 26.0000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



CANARD AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS

PHITAL=0 DCND1=0

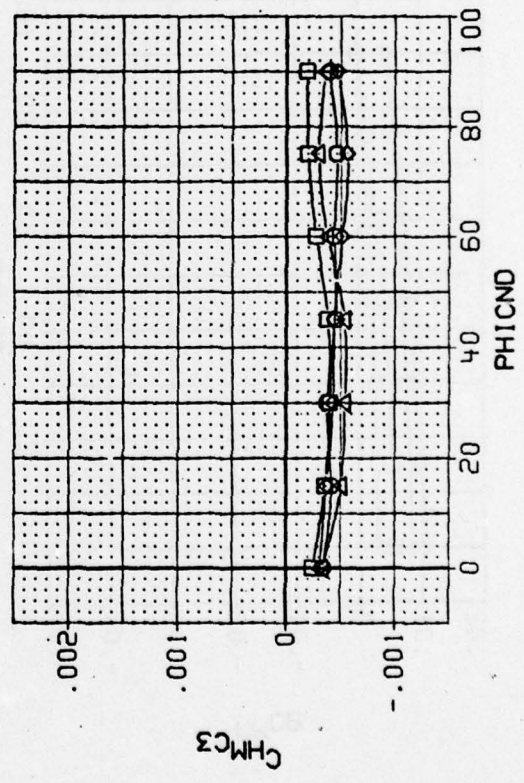
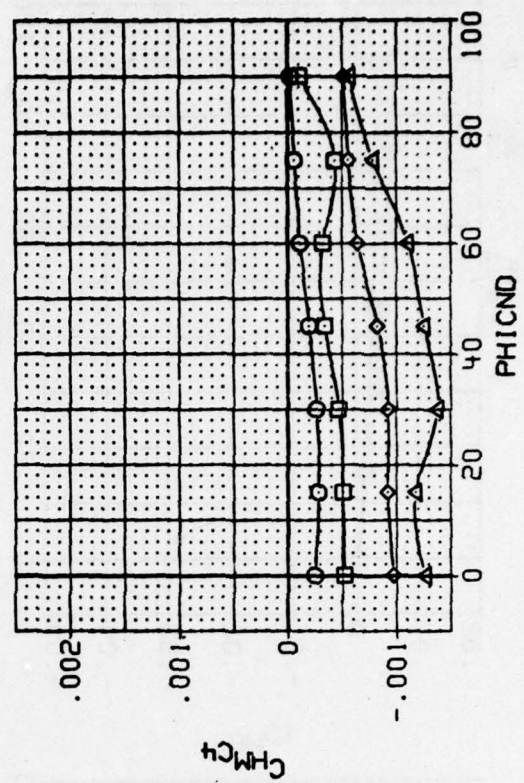
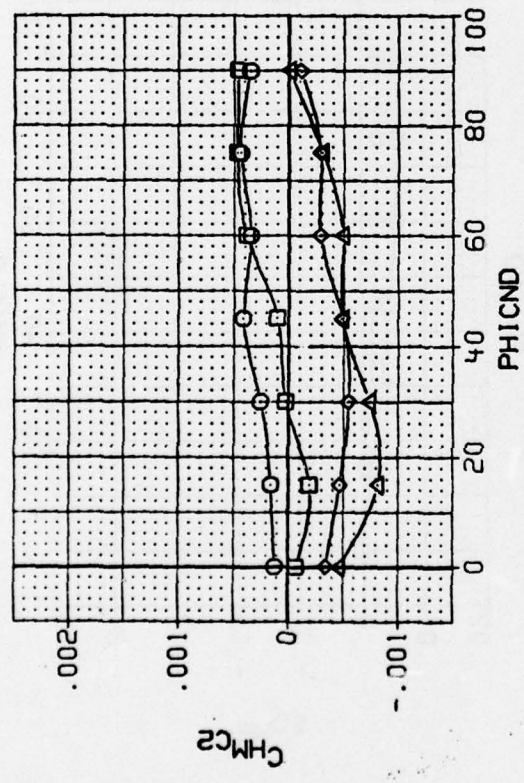
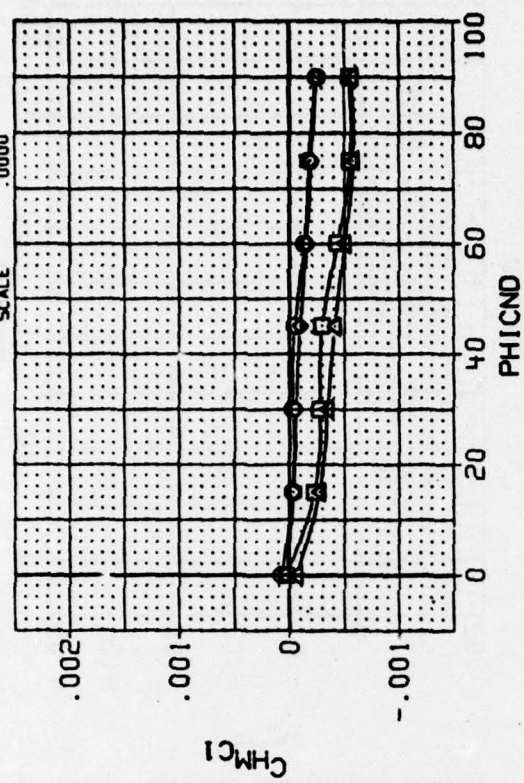
(A) MACH = 2.50

DATA SET SYMBOL
 (BXH108) ○
 (BXH109) □
 (BXH110) ◇
 (BXH111) △

CONFIGURATION DESCRIPTION
 AEDC V11A-C1A, CANARD CONTROL, BNIC1TR
 AEDC V11A-C1A, CANARD CONTROL, BNIC1TR
 AEDC V11A-C1A, CANARD CONTROL, BNIC1TR
 AEDC V11A-C1A, CANARD CONTROL, BNIC1TR

ALPHA DCND2 DCND3 DCND4
 6.000 .000 .000 .000
 12.000 .000 .000 .000
 15.000 15.000 15.000 15.000
 12.000 15.000 15.000 15.000

REFERENCE INFORMATION
 SREF 19.5350 50. IN.
 LREF 5.0000 IN.
 SREF 5.0000 IN.
 XTRP 26.0000 IN.
 YTRP .0000 IN.
 ZTRP .0000 IN.
 SCALE .0000



CANARD AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS
 PHITAL=0 DCND1=0
 (A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(BXH10B) \square AEDC V41A-C1A, CANARD CONTROL, ENIC1TR

(BXH10S) \square AEDC V41A-C1A, CANARD CONTROL, ENIC1TR

(BXH110) \square AEDC V41A-C1A, CANARD CONTROL, ENIC1TR

(BXH111) \square AEDC V41A-C1A, CANARD CONTROL, ENIC1TR

ALPHA DCND2 DCND3 DCND4

6.000 .000 .000

2.000 .000 .000

5.000 15.000

2.000 15.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

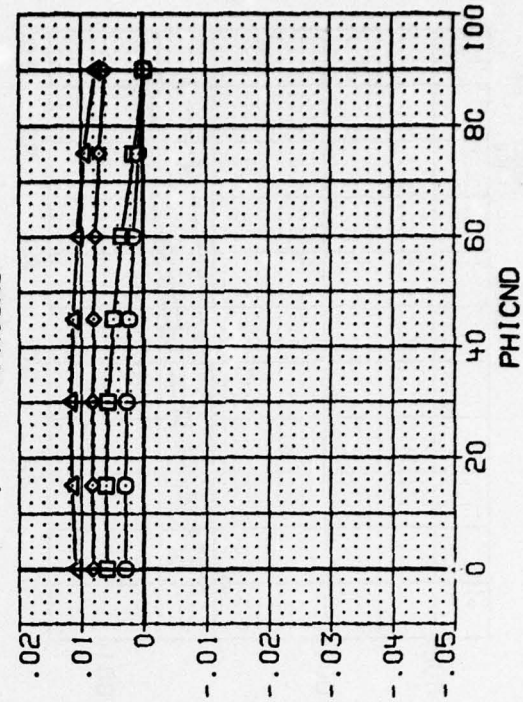
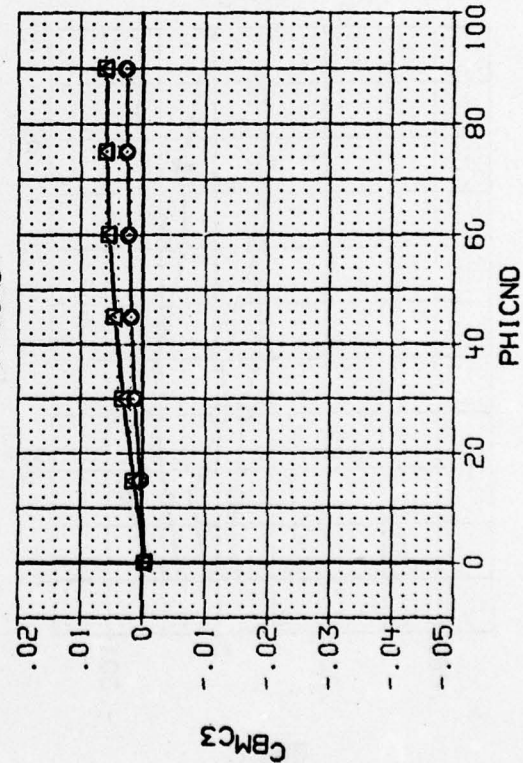
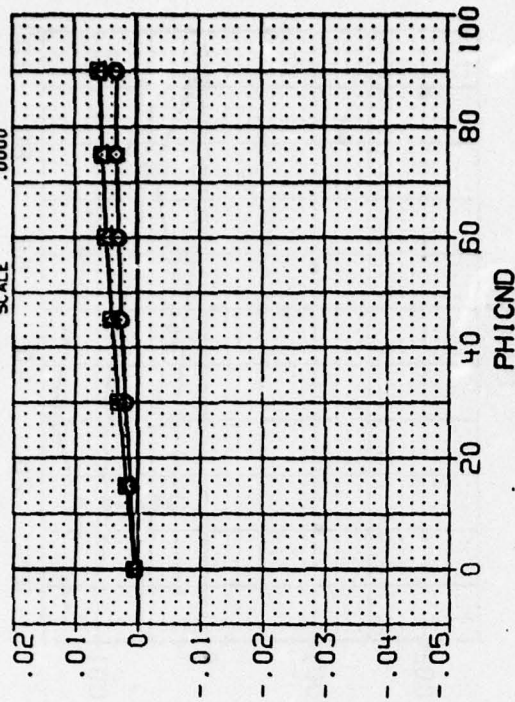
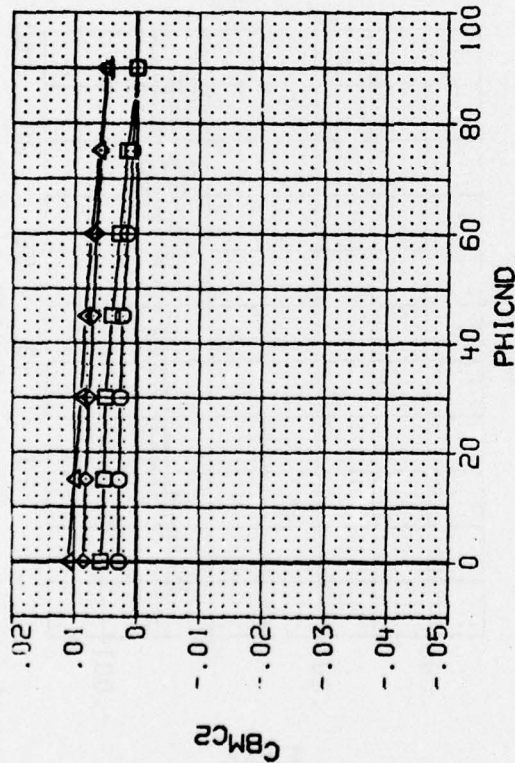
BREF 5.0000 IN.

XREF 26.0000 IN.

YREF .0000 IN.

ZREF .0000 IN.

SCALE



CANARD AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS

$PHITAL=0$ $DCND1=0$

(A) MACH = 2.50

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(AXH108) \square AEDC V11A-C1A, CANARD CONTROL, BNICITR

(AXH109) \square AEDC V11A-C1A, CANARD CONTROL, BNICITR

(AXH110) \square AEDC V11A-C1A, CANARD CONTROL, BNICITR

(AXH111) \square AEDC V11A-C1A, CANARD CONTROL, BNICITR

ALPHA DCND2 DCND3 DCND4

5.000 .000 .000 .000

12.000 .000 .000 .000

15.000 .000 .000 .000

15.000 .000 .000 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

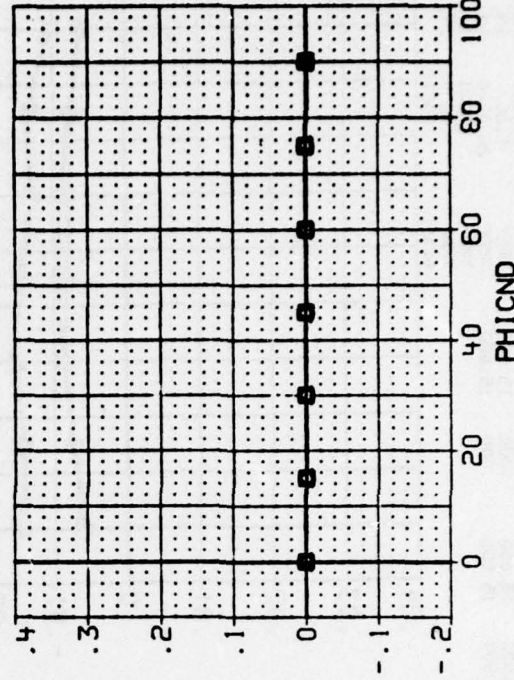
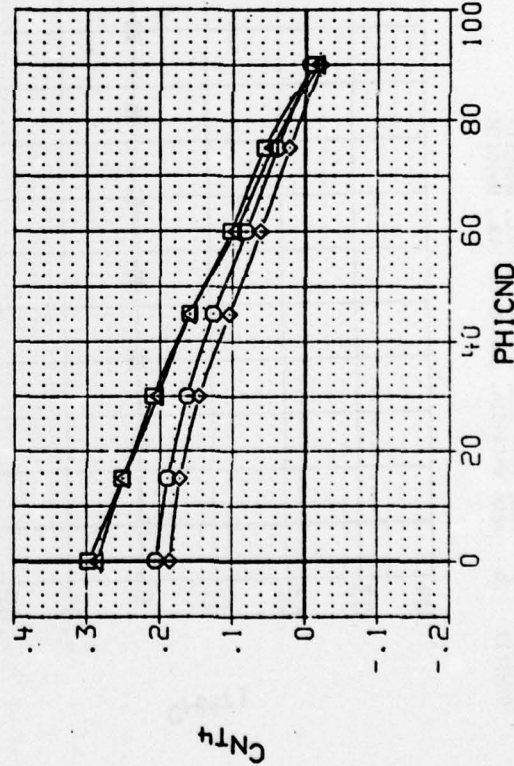
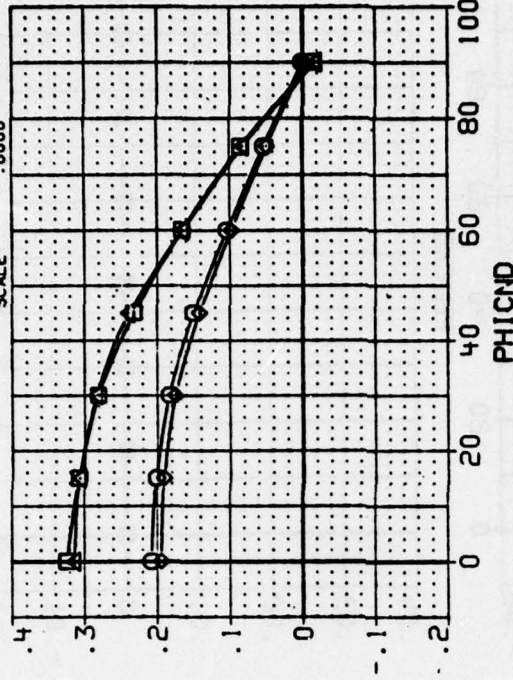
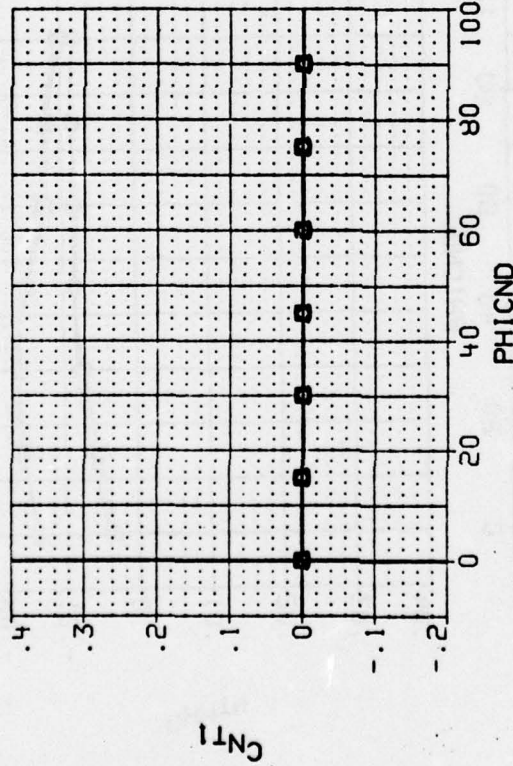
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XPRP 26.0000 IN.

YPRP .0000 IN.

ZPRP .0000 IN.

SCALE .0000



TAIL AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS

$PHITAIL=0$ $DCND1=0$

(A) MACH = 2.50

DATA SET SYMBOL

(CXH102)
(CXH103)
(CXH110)
(CXH111)

CONFIGURATION DESCRIPTION

AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
AEDC W1A-C1A, CANARD CONTROL, BNIC1TR
AEDC W1A-C1A, CANARD CONTROL, BNIC1TR

ALPHA

6.000
12.000
15.000
15.000

DCND2

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15.000
15.000

DCND3

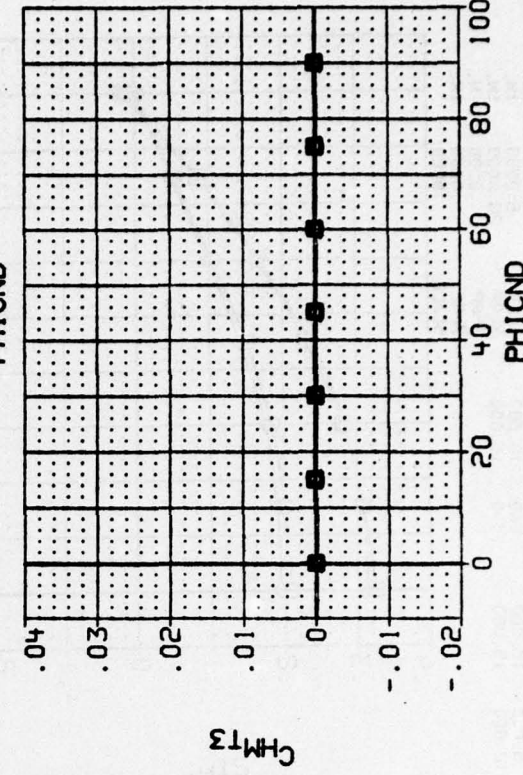
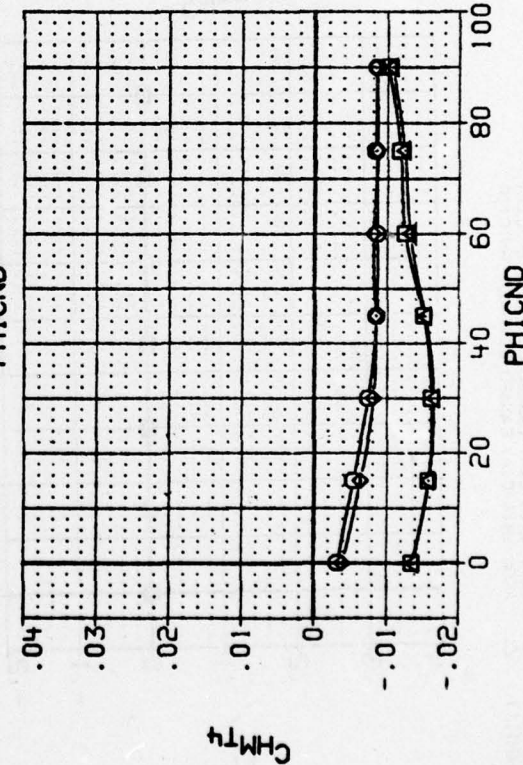
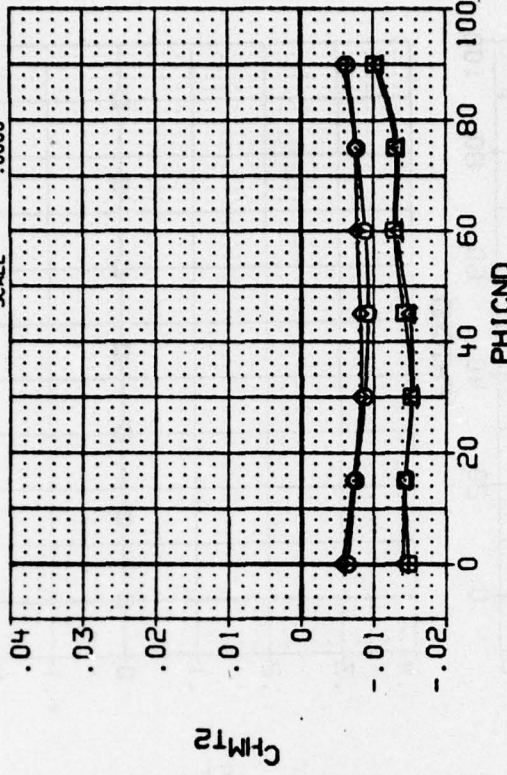
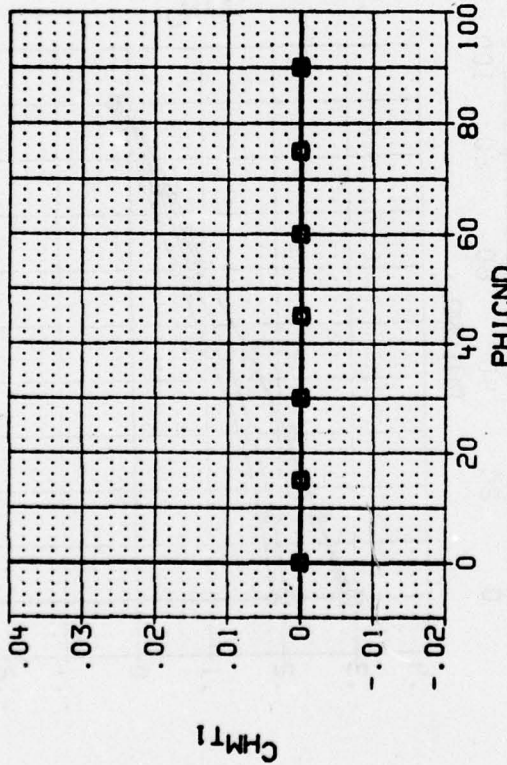
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DCND4

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15.000
15.000

REFERENCE INFORMATION

SREF 19.635 IN.
LREF 5.00 IN.
BREF 5.00 IN.
XMRP 26.0000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



TAIL AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS
PHITAIL=0 DCND1=0

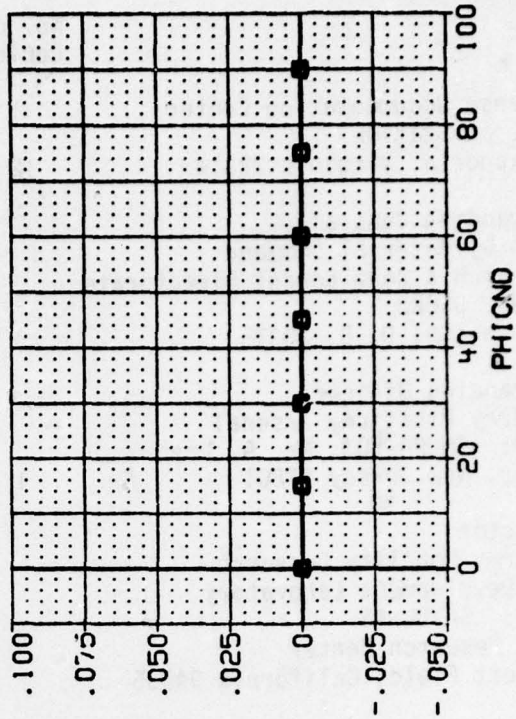
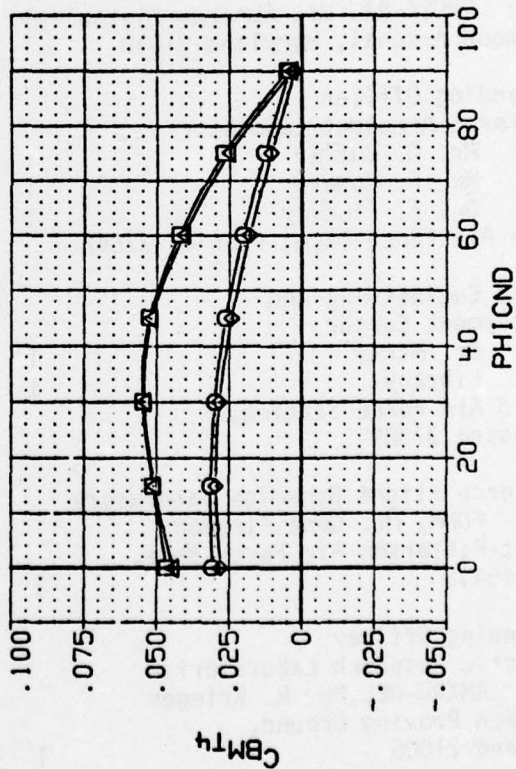
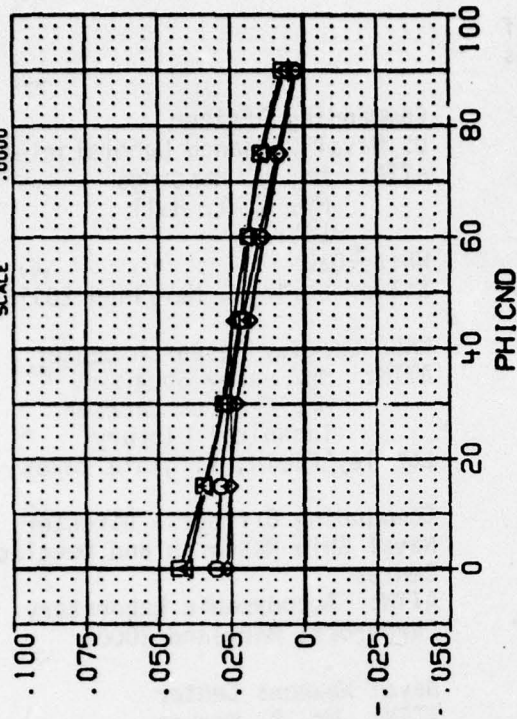
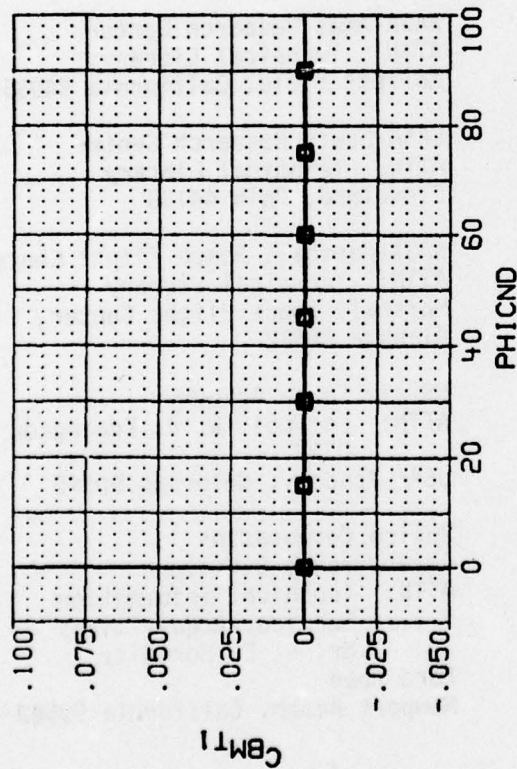
(A) MACH = 2.50

DATA SET SYMBOL
(CXH108) □
(CXH109) ◇
(CXH110) △
(CXH111)

CONFIGURATION DESCRIPTION
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AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR
AEDC V41A-C1A, CANARD CONTROL, BNICITR

ALPHA DCND2 DCND3 DCND4
6.000 .000 .000 .000
12.000 .000 .000 .000
15.000 15.000 15.000 15.000
12.000 15.000 15.000 15.000

REFERENCE INFORMATION
SREF 19.6350 SQ. IN.
LREF 5.0000 IN.
SREF 5.0000 IN.
XPRP 26.0000 IN.
YPRP .0000 IN.
ZPRP .0000 IN.
SCALE



TAIL AERODYNAMIC CHARACTERISTICS FOR VARYING ROLL POSITIONS
PHITAL=0 DCND1=0
(A) MACH = 2.50

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